Spatial Spectral Analysis of Elastic Waves in a Three-Dimensional Truss Structure

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B.S., Oceanography, Seoul National University, Korea (1993)

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Abstract

The MIT Truss Damping group has been studying the application of truss-like structures to support the internal components of underwater vehicles. A truss structure is able to reduce the vibrational energy transmitted from internal noise sources to the hull by scattering and multi-paths. In a real application a truss will mount equipment at many points and this truss will be attached to the hull at various points.

Spatial spectral analysis of the total elastic wave field is conducted to find the phase relationship of the total elastic wave field at the joints of a three-dimensional truss. A single-input/multiple-output system is excited by broadband white noise at one base strut and measured by accelerometers at joints and on base struts along the axis of the truss. The phase of the cross-spectra shows an approximately linear phase relationship as a function of frequency. However, the frequency-wavenumber spectrum shows that the wavenumbers at the joints of the truss are evenly distributed in high inertance bands (approximately between -10dB and 10dB re 1/kg). This is due to reverberation, and can be regarded as phase randomness.

A measurement of the major early arrival times is also conducted by measuring the impulse response at joints with a transient input. The results show that the total elastic wave field travels with approximately constant group delay in the high frequency bands and there is no preferential wave type distribution during propagation. Energy decay rates of the total elastic wave field depends upon position and frequency. The mean-free-paths of energy build up and energy decay are about 6 times the total shortest path in the truss.

The frequency domain results show that with a stationary input the wave propagates, as measured on the joints of the truss, both in positive and negative axial directions. The constant group delay of wave propagation at high frequency bands with a transient input in time domain will disappear after the truss becomes fully energized. The multi-paths on the truss causes phase randomness of the output at the joints over all frequency bands due to reverberation.

Thesis Supervisor: Ira Dyer
Title: Weber-Shaughnessy Professor of Ocean Engineering
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Dedication

To my parents
To study is to know,

to know is to understand,

to understand is to appreciate,


to appreciate is to love.
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Chapter 1

Introduction

Truss structures have been used in many areas from civil structures to space structures, not only because of their low construction and maintenance cost, but also due to their structural advantage. Recently, the US Navy has been interested in truss-like structures as a new design concept for the internal component of marine vehicles, in an attempt to reduce the propagation of noise generated from machinery to the hull.

Truss structures are able to reduce the vibrational energy transmitted from internally mounted rotating and reciprocating noise sources to the hull by scattering and attenuation as a result of multi-paths. A truss mounting, contrary to the conventional marine vehicle mounting system, can simultaneously improve acoustic and structural performance, as well as lower construction and maintenance costs, and shorten refit times. It is believed that passive damping treatment applied to the truss could maximize attenuation of vibration, which is a crucial requirement for a marine vehicle operation. A schematic showing a truss as an internal structure to a marine vehicle is shown in Fig. 1-1.

1.1 Previous Work at MIT

The MIT truss program on machinery cradle damping began in early 1993 with support from the Office of Naval Research (ONR) and the Advanced Research Program
Agency (ARPA). The MIT truss was designed and constructed by Heath[1]. He explored the dynamic behavior of the truss and showed that the major mechanisms of attenuation are scattering and strut radiation.

Branthonne[2] investigated the application of dynamic absorbers to the truss. His result shows that these treatments are very effective in attenuating the radiated noise from the truss, and he suggested that the assumption of equipartition of energy distribution might be incorrect. Energy partitioning in the truss was investigated by Machens and Dyer[3] and the result shows the existence of energy equilibrium in the truss instead of energy equipartition.

Research on optimized joint designs for a truss was conducted by Frink[4]. His result shows that pinned joints effectively maximize noise attenuation. Air borne noise radiation of the truss was studied by Chiasson[5] numerically and experimentally. His results show the ability to predict truss radiation from knowledge of its geometry.

Numerical wave propagation of the truss was studied by Guo[6] using a global-matrix method. He showed that it is possible to study large-scale complicated truss structures with numerical methods.
1.2 Motivation and Approach

In a real marine vehicle application, a truss could be mounted with machinery, equipment as well as compartments and will be attached to the hull at various points over the truss. The vibration can be coupled destructively or constructively through multi-paths and can propagate to the hull even though high attenuation is achieved by truss mounting isolators. Thus it becomes important to understand the characteristic response of the truss at these hull attachment points due to the forcing applied by the machinery at various points within the truss.

In this study I explore the spatial spectral analysis of the elastic waves in a truss structure. Spatial spectral analysis of the total elastic wave field, the sum of flexural, compressional, and torsional waves, measures the dynamic behavior of a truss with multiple attachment points and equipment loadings. With this approach I investigate the response characteristics of the truss by making measurement at the joints.

Multiple equipment loadings in a truss and its attachment points to a hull can be modeled as a multiple-input/multiple-output system. However, instead of using this complicated model, a simplified single-input/multiple-output model is used in this study for convenience. This model is equivalent to an ideal single machine mounting with many attachment points to the hull.

1.3 MIT Truss Design

The MIT truss is composed of eleven square-based pyramids made of aluminum 6061 T6 struts and joints. The nominal dimension of the structure is $4.7\times0.84\times0.80m$ and its weight is approximately $16kg$. This configuration could be roughly a $1/15$ scale of a marine vehicle internal structure. However, it is designed to investigate the dynamics of a fully three-dimensional truss and this is the reason it does not look like a practical internal structure.
The truss consists of 109 struts and 35 joints. There are three different strut lengths, 54 long diagonals (810mm), 11 horizontal diagonals (700mm) and 44 short base struts (484mm), made of 12.7mm outer diameter tubes with wall thickness of 1.65mm. The struts and joints are connected by epoxy to strengthen their connectivity, and the numbers of struts per joint vary from three to eight. A typical joint design schematic is shown Fig. 1-2. To facilitate reporting of experiments, these joints are labeled and a global coordinate system is assigned. Fig. 1-3 shows these labels and a perspective of the MIT truss with the coordinate system. More detailed characteristics of the MIT truss are shown in Heath[1].

Figure 1-2: Typical joint design.

Figure 1-3: Label of joints and perspective view of truss

1.4 Experimental Setup

The following experimental setup is that commonly used for frequency spectrum measurement and early arrival time measurement, except for the way the input signal
is provided. A schematic of the test setup is shown in Fig. 1-4. The MIT truss is hung from a wooden support structure by bungie cords. This provides good isolation to model a free-free boundary condition above \(10Hz\).

The truss is excited in the \(Z\) direction by a Br"uel & Kjær (B&K) mini-shaker type 4810 at the base strut \(a4c4\) 16.5\(cm\) (6\(\frac{1}{2}\)”) from the joint \(a4\). This \(a4c4\) strut is selected instead of the very end of the truss structure to avoid unfavorable boundary effects. The shaker position at the strut does not have any symmetry since the length of the \(a4c4\) strut is 70.8\(cm\) (27\(\frac{7}{8}\)”) and the forcing location is closer to the joint \(a4\).

A B&K impedance head type 8001 is used to measure the applied force and acceleration, and this is connected to a data acquisition system via two Piezotronics PCB MOD462A charge amplifiers. Twenty two 0.5\(g\) Kistler Picotron type 8616 accelerometers are used to measure the accelerations at the \(b\) struts and \(c\) joints. The accelerometers at the \(b\) struts are located 36.2\(cm\) (14\(\frac{1}{4}\)”) from the \(c\) joints, slightly off from the center of the \(b\) struts, to avoid lower mode’s nodes and anti-nodes. The positions of the shaker and accelerometers are shown in Fig. 1-5.

For the frequency spectrum measurement, band limited white noise (0~25\(kHz\)) with 30\(V_{pk}\) is provided to the shaker by a Hewlett Packard (HP) 3562A dynamic signal analyzer driven by a Wilcoxon Research (WR) PA7 power amplifier. For the early arrival time measurement, a periodic impulse-like signal is generated with a period of 1\(Hz\) by using a HP3325A synthesizer/function generator and a WaveTek 2\(MHz\) function generator model 20.

Data are acquired by a HP VXI acquisition system and processed by an I-DEAS signal processing software package. The frequency band from 0 to 25.6\(kHz\) is sampled for the frequency spectrum measurement with 100 ensemble averages. For the early arrival time measurement, time frame length of 0.125\(sec\) is used to avoid the overlap and reverberation of signal arrival. The sampling frequency is 65,536\(Hz\) and 1.2207% pretrigger with 10% positive trigger is used to avoid unfavorable time delay effects which may contaminate the data. The time domain data are acquired with 100 ensemble averages to smooth the fluctuations inherent in any one test.
Figure 1-4: Schematic of test setup.

Figure 1-5: Positions of shaker and accelerometers.
1.5 Overview of Thesis

In Chapter 2, I investigate the spatial spectrum of the total elastic wave field at the joints of the MIT truss and analyze the result by an array signal processing method. The results show that spatial coherence at the joints is high except in stop bands caused by the periodicity of the truss, and an approximately linear phase relationship is observed at each joint from the phase of the cross-spectra. However, the frequency-wavenumber spectrum in high inertance spectrum bands show that there is no directional preference of the elastic wave propagation through the joints of the truss due to reverberation.

In Chapter 3, I analyze the early arrival times of the elastic waves using a transient input. To enhance data resolution, the Gaussian bandpass filtered impulse response function, rather than cross-correlation, is measured to find the major early arrival times. The results show that the elastic wave energy arrives with approximately constant group delay in high frequency bands with a transient input, but, the distribution of wave types becomes random. The decay rates of the elastic wave energy are not constant due to wave coupling properties of the truss.

In Chapter 4, I summarize the major results of this study and provide suggestions for future work. The wavenumber distribution of the truss is broad over all frequency bands in the frequency-wavenumber spectrum. This is due to the multi-paths of the truss structure which cause phase randomness of the cross-spectra by reverberation.
Chapter 2

Spatial Cross-Spectrum of Elastic Waves

2.1 Spectral Analysis

Previous research at MIT was confined to radiation, scattering and vibration of the truss. Moreover, much of the research in related fields was done on wave propagation in spatially periodic structures with multiple coupling of waves. Both theoretical and computational approaches were used and can be found in references such as [7], [8], [9], [10], [11], [12], [13], [14].

However, in a real application of the truss as an internal structure of a marine vehicle, equipment will be mounted at many points through the truss and eventually this truss structure will be attached to the hull at various points. Thus, it is important to understand the characteristic response of noise and vibration from these mounting points to the hull attachment points, since noise can propagate through multi-paths of a truss structure constructively or destructively.

The characteristic response at the joints of the truss and hull attachment points is hypothesized to be coherent in low frequency bands and more complicated with increasing frequency and distance due to the introduction of noise.
This hypothesis is based on the fact that the dynamic behavior of the MIT truss changes from global to local dynamics as suggested by Heath[1]. Heath postulated that the transition frequency occurs below 1kHz. Below 1kHz the MIT truss behaves as a single beam and above this frequency the truss behavior is more complicated and local. A numerical study of mode shape of the truss was conducted by Branthonne[15] and his result shows that even at the frequencies below 100Hz, where the truss is fully energized, the mode shapes become very complicated.

To investigate the characteristic response of the joints of the truss I use spectral density functions defined below[16]. All the analysis in this study is done using the one-sided spectral density function. In practice we can extend measured positive frequency results to the negative domain due to symmetry. However, I use the two-sided spectra for analytical calculation since they are more convenient in the mathematics. The one-sided auto-spectral density and cross-spectral density functions are defined as

\[
G_{xx}(f) = 2S_{xx}(f) = 2 \int_{-\infty}^{\infty} R_{xx}(\tau)e^{-j2\pi f \tau} d\tau \quad f > 0, \quad (2.1)
\]

\[
G_{xy}(f) = 2S_{xy}(f) = 2 \int_{-\infty}^{\infty} R_{xy}(\tau)e^{-j2\pi f \tau} d\tau \quad f > 0. \quad (2.2)
\]

where \( R_{xx}(\tau) \) and \( R_{xy}(\tau) \) are auto-correlation function and cross-correlation function respectively and defined as

\[
R_{xx}(\tau) = \lim_{T \to \infty} \frac{1}{T} \int_{0}^{T} x(t)x(t+\tau)dt, \quad (2.3)
\]

\[
R_{xy}(\tau) = \lim_{T \to \infty} \frac{1}{T} \int_{0}^{T} x(t)y(t+\tau)dt. \quad (2.4)
\]

The coherence of total elastic wave field is defined as

\[
\gamma_{xy}^2(f) = \frac{|G_{xy}(f)|^2}{G_{xx}(f)G_{yy}(f)} \quad 0 \leq \gamma_{xy}^2(f) \leq 1. \quad (2.5)
\]
I use the $Z$ component of acceleration from the truss. However, these data not only include the flexural component but also the compressional and torsional components of acceleration, since these different wave types couple to each other via interactions at the joints.

As mentioned in Chapter 1, the many equipment loadings in a truss and its attachment points to a hull can be modeled as a multiple-input/multiple-output system, but, I use a simplified single-input/multiple-output model for convenience. This model is equivalent to an ideal single machinery mounting with many attachment points through a hull and can be realized by exciting the truss with a single shaker and measuring at multiple accelerometers over the truss. A schematic of single-input/multiple-output system is shown in Fig. 2-1. Truss structures are characterized by their multiple paths connected by many joints to maintain their shape. If a forcing source is applied at an arbitrary position of a truss and a response is measured at another point, it can be modeled as a single-input/single-output with multi-paths. A schematic of this system is shown in Fig. 2-2.

For convenience, nondispersive propagation through multiple paths of an input $x(t)$ from a stationary random process is considered first with an assumption of a

![Figure 2-1: Schematic of single-input/multiple-output system.](image-url)
uniform gain factor $H_k$, $k = 1, 2, 3, \cdots, r$. This will produce an output $y(t)$ such that

$$y(t) = H_1 x(t - \tau_1) + H_2 x(t - \tau_2) + \cdots + H_r x(t - \tau_r)$$

$$= \sum_{k=1}^{r} H_k x(t - \tau_k) \tag{2.6}$$

where $\tau_k$ is time delay caused by each path. With use of equations Eq. (2.4) and Eq. (2.3), the cross-correlation function between the input $x(t)$ and the output $y(t)$ becomes

$$R_{xy}(\tau) = \lim_{T \to \infty} \frac{1}{T} \int_0^T x(t) \left[ H_1 x(t - \tau_1 + \tau) + H_2 x(t - \tau_2 + \tau) + \cdots + H_r x(t - \tau_r + \tau) \right] dt$$

$$= \sum_{k=1}^{r} H_k R_{xx}(\tau - \tau_k). \tag{2.7}$$

The above equation shows that the cross-correlation between $x(t)$ and $y(t)$ appears as a sum of autocorrelation functions of $x(t)$ with offset by $\tau_k$ with each $H_k$, $k = 1, 2, 3, \cdots, r$. However, since the truss is composed of multiple paths we cannot get clear resolution of cross-correlation in the time domain.

The Fourier transform of the cross-spectral density becomes

$$G_{xy}(f) = \sum_{k=1}^{r} H_k G_{xx}(f)e^{-j2\pi f\tau_k}. \tag{2.8}$$
This can be expressed in complex polar notation as

$$G_{xy}(f) = |G_{xy}(f)| e^{-j\theta_{xy}(f)}$$

(2.9)

where

$$|G_{xy}(f)| = G_{xx}(f) \left[ \sum_{i=1}^{r} \sum_{k=1}^{r} H_i H_k \cos 2\pi f (\tau_i - \tau_k) \right]^{1/2}$$

(2.10)

$$\theta_{xy}(f) = \tan^{-1} \left[ \frac{\sum_{k=1}^{r} H_k \sin 2\pi f \tau_k}{\sum_{k=1}^{r} H_k \cos 2\pi f \tau_k} \right].$$

(2.11)

It can be shown from Eq. (2.10) that the magnitude of the cross-spectrum for multiple paths will produce destructive interference notches at frequency intervals of

$$\Delta f = \frac{1}{\tau_i - \tau_k}.$$  

(2.12)

The coherence function between the input and output in multiple paths can be derived with the autospectrum of $y(t)$. These are expressed as

$$R_{yy}(\tau) = \sum_{i=1}^{r} \sum_{k=1}^{r} H_i H_k R_{xx}(\tau + \tau_i - \tau_k)$$

(2.13)

$$G_{yy}(f) = G_{xx}(f) \sum_{i=1}^{r} \sum_{k=1}^{r} H_i H_k \cos 2\pi f (\tau_i - \tau_k).$$

(2.14)

From the above results the coherence function, $\gamma_{xy}^2(f)$, should be always one at all frequencies for an ideal case where there is no noise;

$$\gamma_{xy}^2(f) = \frac{|G_{xy}(f)|^2}{G_{xx}(f)G_{yy}(f)} = 1.$$  

(2.15)

However, introducing a non-uniform gain factor as a function of frequency, $H_k(f)$, and dispersive propagation will cause more complicated cross-spectrum phases.

The accelerations measured at the joints can be considered as a line array, and the spatial Fourier transform can be used to investigate the characteristic response of the truss at the joints in the frequency-wavenumber domain to determine the dispersion
relationship of traveling waves via these joints. The spatial Fourier transformation is defined as

\[
A(\omega, k) = \frac{1}{N} \sum_{n=0}^{N-1} a(\omega, n\Delta x)e^{-in\Delta x},
\]

(2.16)

where \( \omega \) is angular frequency, \( k \) is wavenumber, \( N \) is the number of sensors and \( \Delta x \) is the interval between sensors.

In this array processing, I imagine the acceleration at joints as the response from a virtual beam which is attached to the joints. In a real application this virtual beam can be replaced by a hull. Thus if the frequency-wavenumber properties of the hull are known then the multiplication of the frequency-wavenumber properties of the truss and hull can give the frequency-wavenumber characteristic of the combined two systems because of the Fourier transform properties. Hence, it is important to look at the frequency-wavenumber characteristic of the truss at the joints, which will be attached to the hull.

The array used for the MIT truss has 6 accelerometers and the length of the array is 3.917m, or 5 cells long. The spatial sampling rate is 0.783m and the Nyquist wavenumber, \( k_s \), becomes 4.021/m. The resolution, half the main lobe width, \( k_x = 2\pi/L \), of this array is 1.604. If a continuous-variable signal is band limited to frequencies below \( k_s \), then it can be periodically sampled without loss of information as long as the sampling period is less than \( \pi/k_s[17] \). However, because of the sparse spatial sampling, aliasing occurs at high frequencies for flexural, torsional, and compressional waves approximately above 52Hz, 1,986Hz, and 3,259Hz respectively if the phase speeds of elastic wave propagation through the virtual beam are the same as that in the struts. This bad aliasing is what really happens in the truss, and we cannot avoid aliasing since the interval between the joints is fixed.
2.2 Spatial Cross-Spectrum

First, I measured the spatial coherence of the total elastic wave field at various locations on the truss to find the linear dependence between the input and outputs. It is assumed that the spatial coherence is unity, $\gamma^2 = 1$, over the entire frequency band except in destructive interference notches.

The spatial coherence measured at the $c$ joints is shown in Fig. 2-3. As expected the coherence at each joint is nearly unity except in some wide notches at all frequencies, and irrespective of notches below $5kHz$. The coherence level decreases with increasing distance and this is due to the decrease of signal to noise ratio with increasing distance. Note that the coherence function can be expressed in terms of signal to noise ratio (SNR) as

$$\gamma^2 = \frac{SNR}{1 + SNR}. \quad (2.17)$$

Notches become more dense with increasing distance and this is due to the wide spread of time delays caused by multiple paths of the truss. The low coherence below $5kHz$ might be spurious since the frequency resolution in this measurement is $\Delta f = 8Hz$ and this can cause contamination in low frequency bands. However, some of low coherence bands might be not spurious but caused by resonance frequencies of the struts.

Due to their location, the spatial coherence at the base struts of length $700mm$, $ac$, might clearly show the contribution of these resonance frequencies rather than the coherence at the joints. Fig. 2-4 shows the spatial coherence at the base struts with the same frequency resolution of $\Delta f = 8Hz$. This figure shows more wide low coherence bands than the coherence of the joints, since the resonance is sensed more clearly on the struts than at the joints. The low coherence bands become deeper with increasing distance. At the farthest position shown, $b12$, three apparent low coherence bands are well developed, except below $5kHz$, approximately at $5.5kHz$, $10.5kHz$, and $14.5kHz$. 
Another measurement with the frequency resolution of $\Delta f = 2Hz$ was conducted to examine whether this low coherence is spurious or caused by the stop bands of the truss. Fig. 2-5 shows the spatial coherence at the base struts with the frequency resolution of $\Delta f = 2Hz$. Coherence of unity is observed between the input and outputs at all locations and suggests that the low coherence below 5kHz is spurious. Low coherence occurs below 0.5kHz and this supports the frequency resolution error again. Thus the low coherence in the low frequency band is spurious and caused by resolution error. However, wide range notches at 3kHz and 5.5kHz are still observed regardless of the enhanced frequency resolution and these might not be spurious low coherence levels. These might be due to stop bands caused by resonance of the struts of the truss.

Stop band analysis was conducted with the assumption of struts as Euler-Bernoulli beams for convenience. Fig. 2-6 and Fig. 2-7 give the multiple half-wave resonance frequencies of the different elastic waves and struts respectively. Fig. 2-8 shows the total resonance frequencies in the truss. These resonance frequencies present a very large impedance to waves at the strut inputs, and thus cause stop bands in wave propagation, and therefore low spatial coherence. The major stop bands shown in Fig. 2-4 and Fig. 2-5, near 3kHz, 5kHz, 10kHz, and 14kHz, are well developed among the other resonance frequencies. It seems that these stop bands coincide with the frequencies where the strut lengths equal the integral multiple of half the elastic wavelengths such as $\lambda/2, \lambda, 3\lambda/2, \cdots$.

The flexural wavelength equals twice the strut length below 1kHz at the frequencies equivalent to the first few integral multiple of half the elastic wave lengths, and this frequency band is in the resolution error region; hence coherence becomes small regardless of strut resonance. Apart from these stop band frequency regions, coherence rises to unity again. The elastic wavelengths as a function of frequency are plotted in Fig. 2-9. The stop band near 3kHz coincides with the frequencies where the strut lengths equal half of the torsional and compressional wavelengths. The next two major stop bands near 5kHz and 10kHz coincide with the frequencies where the
strut lengths equal these elastic wavelengths. The stop band near 14kHz can also be explained in this manner.

The magnitude and phase of the cross-spectra at the joints are shown in Fig. 2-10 and Fig. 2-11 respectively. The resonance of struts and time delays through multipaths in a stationary process cause dips and small oscillating peaks in the magnitude of the cross-spectra. If the elastic waves propagate with roughly constant group delay, the phase of the cross-spectra at the joints in Fig. 2-11 should be periodic. However, there is no such periodicity in the phase of the cross-spectra; thus the phase of the cross-spectra looks like a random process.

However, I unwrapped the phase to investigate more thoroughly the tendency of wave propagation to have constant group delay. The unwrapped phase of the cross-spectra at the joints is shown in Fig. 2-12. The unwrapped phase angle plot roughly shows a linear phase relationship with increasing frequency. These slopes are not constant and vary with frequency due to dispersion. Table 2.1 shows the slopes averaged over all measured frequencies and within octave bands.

Table 2.1: Phase slope of cross-spectra at the joints of the MIT truss.

<table>
<thead>
<tr>
<th>Joint</th>
<th>0-25.6kHz</th>
<th>16kHz octave band</th>
<th>8kHz octave band</th>
<th>4kHz octave band</th>
</tr>
</thead>
<tbody>
<tr>
<td>c04</td>
<td>-0.69 x 10^3</td>
<td>-0.43 x 10^3</td>
<td>-0.88 x 10^3</td>
<td>-0.65 x 10^3</td>
</tr>
<tr>
<td>c06</td>
<td>-1.72 x 10^3</td>
<td>-1.06 x 10^3</td>
<td>-2.14 x 10^3</td>
<td>-2.15 x 10^3</td>
</tr>
<tr>
<td>c08</td>
<td>-2.63 x 10^3</td>
<td>-1.88 x 10^3</td>
<td>-3.06 x 10^3</td>
<td>-3.49 x 10^3</td>
</tr>
<tr>
<td>c10</td>
<td>-2.99 x 10^3</td>
<td>-2.37 x 10^3</td>
<td>-2.99 x 10^3</td>
<td>-3.31 x 10^3</td>
</tr>
<tr>
<td>c12</td>
<td>-3.26 x 10^3</td>
<td>-2.63 x 10^3</td>
<td>-3.34 x 10^3</td>
<td>-3.50 x 10^3</td>
</tr>
</tbody>
</table>

Both Fig. 2-12 and Table 2.1 show, with a few exceptions, that the magnitude of the phase slope increases with increasing distance from the input. From the octave band phase slopes, it seems that the magnitude of the phase slope decreases with increasing frequency. As mentioned above, this phase slope variation is due to dispersive propagation.
The exponential part of a harmonic traveling wave can be expressed as

\[ e^{i(kx - \omega t)}, \quad (2.18) \]

and its phase is \( \varphi \),

\[ \varphi = kx - \omega t. \quad (2.19) \]

Wavenumber, phase speed, time delay, group speed and group speed can be extracted from two adjacent joints, which are regarded as measuring points at the virtual beam, and become as follows

\[
\begin{align*}
  k &= \frac{\Delta \varphi}{\Delta x}, \quad (2.20) \\
  c_{\text{phase}} &= \frac{\omega}{k}, \quad (2.21) \\
  \tau &= \frac{\Delta x}{c_p}, \quad (2.22) \\
  c_{\text{group}} &= \frac{\partial \omega}{\partial k}, \quad (2.23) \\
  g_r &= \frac{\Delta x}{c_g}, \quad (2.24)
\end{align*}
\]

where \( \Delta \varphi \) is the phase difference between two adjacent joints, \( \Delta x \) is the axial interval between two adjacent joints, 0.783m (taken as positive), \( \tau \) is time delay, and \( g_r \) is group delay.

Fig. 2-13 shows the extracted wavenumber by the above method. It shows that negative wavenumbers are developed over all pairs of the adjacent joints, (c04,c06), (c06,c08), (c08,c10), and (c10,c12), except for the frequencies approximately below 1kHz at (c04,c06) and (c10,c12). This is contrary to the positive wave propagation direction (\( k \) is positive for waves propagating in the direction away from the source), and needs to be interpreted (see page 31).

Fig. 2-14 shows the extracted phase speed of propagating waves via this virtual beam. Two different trends of phase speed magnitude are observed. For the first two pair of the two adjacent joints, (c04,c06) and (c06,c08), the fluctuation of phase
speed is large at frequencies below 1kHz, and the phase speeds become approximately 300m/sec at the highest frequency, 25.6kHz. For the next two pair of the adjacent joints, (c08,c10) and (c10,c12), the phase speeds increase with increasing frequency and become approximately 800m/sec and 1050m/sec respectively at the highest frequency. These two phase speeds also show that the wave propagates with different propagation signs below approximately 11kHz.

Fig. 2-15 shows the extracted time delay of wave propagation. The time delays also are partly negative (mostly at low frequencies). However, the results above the order of 10kHz (for some above 3kHz) are positive.

Fig. 2-16 shows the group speed of propagating waves via this virtual beam. All of the group speeds extracted show high fluctuation of propagation direction above approximately 1kHz. This high fluctuation of propagation direction in the group speed is different from the relatively simple change of propagation signs in the phase speed as shown in Fig. 2-14 and this shows that the wave energy travels with more complications than the phase of waves, as measured on the joints at the truss.

Fig. 2-17 shows the extracted group delay. The group delay also shows high fluctuation in sign above 1kHz due to the extracted group speed results. It must be emphasized, however, that the group speed and group delay plots shown in Fig. 2-16 and Fig. 2-17 are not accurate, since the frequency resolution of these results ($\Delta f = 8Hz$) and this might cause under sampling which is crucial in differentiating angular frequency with wavenumber.

The reversals in propagation direction of waves in the truss are due to the phase data of the cross-spectra measured at the joints. Difficulties often arise in interpreting the phase data when obstructions or discontinuities lie in the path between the input and output locations. The obstructions or discontinuities cause scattering of the propagation energy; concerning scattering, obstructions or a lack of homogeneity along a propagation path can cause wide-angle reflections or an effective bending of the propagating energy flow[16]. Since the fully energized truss for a stationary input creates a highly reverberant environment, the phase between two points tends
to indicate the direction of power flow out of the reverberant region, independent of the direction from the source to the sensor position[18].

Moreover, absorption in a restricted location causes flow of power to that location; hence it can influence the phase data if the location of the absorption is to the side of the outputs rather than aft of them[16]. But Dyer[19] believes that the absorption is uniformly distributed in the truss, except at the source location. This uniformly distributed absorption causes more complicated phase data and hence both wave propagation directions in the truss. For the truss, reverberation cannot be avoided due to its characteristic configuration. Thus the phase data incorporate the complicated reverberation and scattering effects.

Bondaryk[20] showed that in some frequency bands the total power of the compressional wave is approximately $10dB$ higher than the power of the other elastic waves; those bands can be found from Fig. 2-18 and they are $0.2\sim0.4kHz, 1\sim1.1kHz$, etc. There is a possibility of extracting more readily interpretable wavenumber, phase speed, time delay, group speed and group delay from the phase data, if these extraction methods are applied to these bands with a higher frequency resolution.

Fig. 2-19 shows the phase of the cross-spectra at the joints with the enhanced frequency resolution $\Delta f = 0.5Hz$. Simple phase relationships are found near the frequency bands $0.3\sim0.4kHz$ and $0.8\sim1.1kHz$ as expected. I chose the $0.3\sim0.4kHz$ frequency band to extract wavenumber, phase speed, time delay, group speed and group delay from the phase data and the results are shown in Fig. 2-20, Fig. 2-21, Fig. 2-22, Fig. 2-23 and Fig. 2-24 respectively. The results still show negative values and this is due to the reverberation and scattering of the truss as mentioned previously regardless of the enhanced frequency resolution. Moreover, the relatively wide interval between two adjacent joints and the inherent limitation of the phase unwrap algorithm also hinder extracting accurate results from the phase data.

These results show that the wave propagates in both positive and negative axial directions via these virtual attachment points. Fig. 2-25 shows a frequency-wavenumber spectrum of the inertance of the truss from the cross-spectra with the frequency res-
olution of \( \Delta f = 2\text{Hz} \). If there is any elastic wave propagation with the same wave speed of the elastic wave propagation in the struts within the array, aliasing occurs at high frequencies for flexural, torsional, and compressional waves approximately above \( 52\text{Hz} \), \( 1,986\text{Hz} \), and \( 3,259\text{Hz} \) respectively, because of the sparse spatial sampling. Except for stop bands of the inertance spectrum and some other spots, it is hard to see any trend in the dispersion relationship.

Fig. 2-26 shows the cross section of this frequency-wavenumber plot with \( k = 0.02m^{-1} \). The major stop bands are developed approximately at \( 1k\text{Hz} \), \( 2.3k\text{Hz} \), \( 2.8k\text{Hz} \), \( 3.1k\text{Hz} \), \( 6k\text{Hz} \), etc. I plot in Fig. 2-27, the inertance wavenumber spectrum at some peak inertance bands at \( 1,471\text{Hz} \), \( 2,829\text{Hz} \), \( 3,774\text{Hz} \), and \( 5,374\text{Hz} \), to investigate wavenumber distribution. In the \( 1,471\text{Hz} \) inertance band, some inertance wavenumber packets are developed either in positive and negative wavenumber regions and these packets resemble each other. However, with increasing frequency, the wavenumber spectrum spreads and becomes flatter and converges to the magnitude between -10 and 10 \( dB \) re \( 1/\text{kg} \) except for the \( 3,774\text{Hz} \) inertance band which is about 20\( dB \) lower than other inertance bands. Thus it seems that for high inertance frequency bands no specific wavenumber packet is found.
Figure 2-3: Spatial coherence of total elastic wave field between the input and the acceleration at the joints of the truss with frequency resolution of $\Delta f = 8 \text{Hz}$.
Figure 2-4: Spatial coherence of total elastic wave field between the input and the acceleration at the base struts of the truss with $\Delta f = 8Hz$. 
Figure 2-5: Spatial coherence of total elastic wave field between the input and the acceleration at the base struts of the truss with $\Delta f = 2Hz$. 
Figure 2-6: Multiple half-wave resonance frequencies of elastic waves, strut length=810mm (+), 700mm (o), 484mm (x), and total (·).
Figure 2-7: Resonance frequencies of struts, flexural (+), torsional (○), compressional (×), and total (·).
Figure 2-8: Resonance frequencies of truss, flexural (+), torsional (o), compressional (x), and total (·).
Figure 2-9: Elastic wavelength as a function of frequency.
Figure 2-10: Magnitude of cross-spectra of total elastic wave field between the input and the acceleration at the joints of the truss with $\Delta f = 8 Hz$. 
Figure 2-11: Phase angle of cross-spectra of total elastic wave field between the input and the acceleration at the joints of the truss with $\Delta f = 8Hz$. 
Figure 2-12: Unwrapped phase angle of cross-spectra of total elastic wave field between the input and the acceleration at the joints of the truss with $\Delta f = 8 Hz$. 
Figure 2-13: Extracted wavenumber between two adjacent joints as a function of frequency with $\Delta f = 8 \text{Hz}$.
Figure 2-14: Extracted phase speed between two adjacent joints as a function of frequency with $\Delta f = 8\text{Hz}$.
Figure 2-15: Extracted time delay between two adjacent joints as a function of frequency with $\Delta f = 8Hz$. 
Figure 2-16: Extracted group speed between two adjacent joints as a function of frequency with $\Delta f = 8Hz$. 
Figure 2-17: Extracted group delay between two adjacent joints as a function of frequency with $\Delta f = 8Hz$. 
Figure 2-18: Third octave band average of measured power of elastic waves for the strut $a6e6$ ($Z$ excitation) with $\Delta f = 8Hz$ (Ref[20]).
Figure 2-19: Phase angle of cross-spectra of total elastic wave field between the input and the acceleration at the joints of the truss with $\Delta f = 0.5\,Hz$. 

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Figure 2-20: Extracted wavenumber between two adjacent joints as a function of frequency with $\Delta f = 0.5 Hz$. 
Figure 2-21: Extracted phase speed between two adjacent joints as a function of frequency with $\Delta f = 0.5\,Hz$. 
Figure 2-22: Extracted time delay between two adjacent joints as a function of frequency with $\Delta f = 0.5Hz$. 
Figure 2-23: Extracted group speed between two adjacent joints as a function of frequency with $\Delta f = 0.5\text{Hz}$.
Figure 2-24: Extracted group delay between two adjacent joints as a function of frequency with $\Delta f = 0.5 \text{Hz}$. 
Figure 2-25: Frequency-wavenumber (f-k) spectrum of inertance (Z excitation) at the joints with $\Delta f = 2Hz$. 
Figure 2-26: Cross section of inertance frequency-wavenumber spectrum at $k = 0.02 \text{ m}^{-1}$. 
Figure 2-27: Inertance wavenumber spectrum of high inertance bands.
Chapter 3

Early Arrival Times of Elastic Waves

3.1 Bandpass Filtering

An early attempt to measure the arrival time was conducted by Heath[1] by using flexural wave speeds from the Euler-Bernoulli beam theory. However, the truss follows the Timoshenko beam theory with increasing frequency in flexural wave propagation, as shown by Machens and Dyer[3].

I measured the major early arrival times of elastic waves and compared with the calculated arrival time using the Timoshenko beam theory for the flexural wave. The application of the Timoshenko beam theory to the MIT truss case is discussed in Appendix A. A transient input signal is used instead of the stationary random signal to suppress unfavorable measurement of noise. I used impulse response functions rather than correlation to detect wave arrival since, in a multi-path problem, the resolution of the former is better than that of the latter. I deconvolved the source signal from the output signals to find the impulse response functions. Since the dispersive flexural wave propagation is included, a band-limited analysis is required.
However, two conflicts arise\cite{16}:

- the need for a narrow-bandwidth analysis to confine dispersion within useful bounds.

- the need for a wide-bandwidth analysis to suppress overlapping of adjacent correlation peaks.

I used a Gaussian octave bandpass filter to reconcile these conflicts. A Gaussian bandpass filter in the signal amplitude can be expressed as

$$G(f) = \exp \left( \frac{-(f_o - |f|)^2}{2\sigma_f^2} \right)$$ \hspace{1cm} (3.1)

where $f_o$ is the center frequency at which $G(f) = 1$, and $\sigma_f$ is a bandwidth parameter to make the magnitude of filter $6dB$ down at the band edges\cite{21}. The advantage of a Gaussian bandpass filter over a usual finite impulse response bandpass filter is the Gaussian filter’s effective side lobe suppression due to the smallest possible duration-bandwidth product of a Gaussian pulse\cite{22}. A Gaussian bandpass filter with the center frequency at $16kHz$ and its impulse response are shown in Fig. 3-1.

I calculate the complex envelope of the impulse response function, since this is very convenient for looking at the arrival of energy at joints compared to a simple impulse response function\cite{21}.

### 3.2 Early Arrival Times

The calculated first arrival times with an assumption of no change of wave type are shown in Table 3.1. The group speed of the flexural wave of the Timoshenko beam theory is used since it is faster than the phase speed and the elastic wave field propagates with this group speed. At the $c04$ joint, there are three possible paths from the source ($SRC$): $SRC \rightarrow c04$, $SRC \rightarrow a04 \rightarrow c04$, and $SRC \rightarrow c04 \rightarrow a04 \rightarrow c04$. However, in reality the measured arrival time is longer than these calculated values.
since the joints act as wave transformers, scattering from one strut to others, and from one wave type to the other two.

Table 3.1: The first arrival time of elastic waves with single wave type propagation.

<table>
<thead>
<tr>
<th>Joint</th>
<th>Distance (m)</th>
<th>Arrival time (msec)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td></td>
<td>T_{comp}</td>
</tr>
<tr>
<td>c04</td>
<td>0.545</td>
<td>0.11</td>
</tr>
<tr>
<td>c06</td>
<td>1.125</td>
<td>0.22</td>
</tr>
<tr>
<td>c08</td>
<td>2.090</td>
<td>0.41</td>
</tr>
<tr>
<td>c10</td>
<td>3.061</td>
<td>0.60</td>
</tr>
<tr>
<td>c12</td>
<td>4.031</td>
<td>0.79</td>
</tr>
</tbody>
</table>

The input signal used in the experiment is not an ideal impulse but generated from two signal generators. Thus its behavior has some disadvantages for direct application of conventional finite impulse response (FIR) filters. Fig. 3-2 shows the characteristics of the input signal and it clearly shows limits of realization of an ideal impulse in my experiment. Its disadvantages are difficulties in matched filter calculations due to the long tail of the input signal, and wide side-lobes in the frequency domain.

I lowpass filtered the impulse response functions with a cut-off frequency of \( f_c = 25.6kH\lowercase{z} \) to avoid aliasing, since the sampling frequency of the data acquisition system is 65,536\( H\lowercase{z} \). A FIR filter is applied due to phase distortion concerns. The frequency response function of this FIR filter is shown in Fig. 3-3, and the lowpass filtered impulse response functions with full time length, and up to 5\( m\lowercase{sec} \), are shown in Fig. 3-4 and Fig. 3-5 respectively. These two figures show the time delay of the energy arrival at each joint.

Fig. 3-5 clearly shows the long tail of the input signal (at \( S\lowercase{R}C \)), and this is due to the anomaly of the applied input compared to an ideal impulse. Thus it is difficult to get bandpass filtered impulse response functions with low side lobes by using FIR bandpass filters. The lowpass filtered impulse response functions at the joints are difficult to interpret since all dispersive waves with different group speeds are mixed in these plots.
To mitigate the foregoing difficulties, Gaussian octave bandpass filtered impulse response functions with complex envelop representation are used to plot the impulse response functions. A Gaussian octave bandpass filter has advantages such as low side-lobe levels, and a complex envelope representation makes easy observation of the arrival trend, since there is no oscillating temporal pattern.

Gaussian octave bandpass filtered impulse response functions with the center frequency of 16kHz, 8kHz, 4kHz, 2kHz, 1kHz, 0.5kHz, 0.25kHz, and 0.125kHz are shown in Fig. 3-6, Fig. 3-7, Fig. 3-8, Fig. 3-9, Fig. 3-10, Fig. 3-11, Fig. 3-12, and Fig. 3-13 respectively. For the 16kHz octave band impulse response function, it is easy to look at the energy arrival and decay at each joint, compared to the lowpass filtered impulse response functions. One sees an initial rise of the energy, a peak reached later in time with increasing distance of the joint from the source, and finally a slow or gradual decay. This pattern is observed from the 16kHz to the 2kHz octave band. However, below the 2kHz octave band it is hard to find these trends, and this is due to the bandwidth conflict of the dispersive flexural wave propagation and the global behavior of the truss in the low frequency region.

The Gaussian octave bandpass filtered impulse response functions with center frequency of 16kHz, 8kHz, and 4kHz are shown as a function of time, up to 5msec, in Fig.3-14, Fig.3-15, and Fig.3-16 respectively. The 16kHz and 8kHz octave band impulse response functions show the major early arrivals of each wave packet more clearly compared to that of the 4kHz octave band impulse response function. It is harder to find the wave packet arrivals in the latter because the Gaussian resolution, as seen at the source, is too wide to resolve individual paths.

Table 3.2, Table 3.3, and Table 3.4 summarize all possible candidates of the major early arrival times calculated for the shortest path, including all combinations of the possible elastic waves based on Fig.3-14, Fig.3-15, and Fig.3-16, and they are plotted in these figures with ‘o’ symbols. The 16kHz and 8kHz octave band impulse response functions show good agreement with the predicted values. See, for example, the second arrival at joint c04 in the 8kHz Gaussian octave band (Fig.3-15), and
the predicted value in Table 3.3 for the same joint. This agreement comes from the assumption of a combination of the torsional and the flexural wave. But, it is hard to find agreements in the 4kHz octave band impulse response except for the first arrival at each joint. It seems that the compressional waves are dominant first arrivals, and the flexural waves are dominant at the later arrivals. However, there is no specific preference in elastic wave type distribution during the early arrival times except for the above properties.

The major early arrival times for the three highest octave bands are shown in Fig. 3-17 with the shortest path as the X axis. The dotted lines are linear regressions of the first arrival time at each joint. It shows that there is a linear relationship of the early arrival times with increasing octave band. At the 16kHz octave band the arrival times at each joint have approximately constant group delay. This means the dispersive flexural wave is not the dominant first arrival of energy. However, this trend disappears with decreasing frequency and at the 4kHz octave band the arrival trend does not have constant group delay. From these plots the group speeds of the first arrivals are approximately 3450m/sec, 3140m/sec, and 2980m/sec for the 16kHz, 8kHz, and 4kHz Gaussian octave band respectively. Apparently these speeds are slower than the compressional wave speed in the aluminum struts (5092m/sec) and this is because a pure compressional wave cannot be propagated from the source to the observation point (see Table 3.2).

This figure is re-plotted with the axial distance as the X axis in Fig. 3-18. Again the dotted lines are linear regressions of the first arrival times. The offsets at the distance of zero are due to the time delays of energy arrival from the source location since the first joint has the same axial location as the source. The wave speeds of the first arrivals in this plot are 3120m/sec, 2950m/sec, and 2830m/sec for the 16kHz, 8kHz, and 4kHz Gaussian octave band respectively.

Theoretically, if pure compressional wave propagation exists through the shortest path, the observed axial wave speed via these joints becomes $C_{comp}/\sqrt{2} = 3600m/sec$, since the bottom of each cell of the truss is a square and the length between the two
adjacent joints is \( \sqrt{2} \) times longer than a side of the bottom square. However, the extracted wave speeds of the first arrivals are slower than this wave speed through the virtual beam. This is due to the wave bearing characteristics of the truss.

The group delay in Fig. 3-18 does not agree with that in Fig. 2-17 and Fig. 2-24. This can be easily shown by comparing the arrival times at the farthest joint in Fig. 3-18. All of them are less than 3\( \text{msec} \), which are less than the results in Chapter 2. This is so because a time frame length of 0.125\( \text{sec} \) was used in time domain experiment to mitigate noise, overlap, and reverberation.

The decay rates of the elastic waves in Fig. 3-6, Fig. 3-7, and Fig. 3-8 look constant and invariant with position. From these figures it is clear that the decay is not single, but double: one with large decay rate and the other with small decay rate. To investigate this, the decay rate of the total elastic wave field is extracted from the Gaussian bandpass filtered impulse response functions, assuming exponential decay of the energy, \( E(t) \),

\[
E(t) = E_0 e^{-\eta_0(\omega)t} + E_1 e^{-\eta_1(\omega)t},
\]

(3.2)

where \( \omega \) is angular frequency and \( \eta_0(\omega) \) and \( \eta_1(\omega) \) are the two temporal decay rates, which are functions of frequency. I define \( \eta_0 \) as the steeper temporal decay rate and \( \eta_1 \) as the smaller temporal decay rate. Since the energy in the truss is coupled, it is natural that the number of decay rates can be more than one.

Fig. 3-19, Fig. 3-20, and Fig. 3-21 show Gaussian bandpass filtered impulse response functions and their least square fitted curves, following equation Eq. (3.2), at each joint of the truss with the center frequency at 16\( kHz \), 8\( kHz \) and 4\( kHz \) respectively. Fig. 3-22 shows two decay rates as a function of the shortest path with three Gaussian octave bands. These figures show that the steeper decay rate, \( \eta_0 \), increases with increasing frequency and becomes flatter with decreasing frequency. The second decay rate, \( \eta_1 \), is smaller than the major decay rate, and decreases with increasing frequency. This decay rate, \( \eta_1 \), is, with a few exceptions, approximately invariant with position. The two decay rates in different octave bands at the joints are shown in Table 3.5. From this table it is clear that there is no constant exponential energy
The temporal patterns of Fig. 3-19 to Fig. 3-22 have many significant properties. First, it takes time for the truss to reach its early multipath build up, from 0 to about 10msec, roughly linear with increasing axial distance of the measurement from the location of the source. This is so for each of the 16, 8, 4kHz Gaussian octave bands. (The build up delay is not in the model of Eq. (3.2).) Note that this delay corresponds to some sort of average speed of about 400m/sec for the shortest paths in the truss. The average arrival time at joint c12 with all the possible early arrivals is calculated from Table 3.2, Table 3.3, and Table 3.4 and it is about 1.6msec. This may be interpreted such that the maximum energy build up is reached after the waves reverberate about 6 times the total shortest path length in the truss; thus the mean-free-path of energy build up in the truss is 6 times the total shortest path length in the truss[19].

Second, the transition in Eq. (3.2) from the $\eta_0$ to the $\eta_1$ decay rate is at about 45~50msec in the 16kHz band, 60~70msec in the 8kHz band, and 60~85msec in the 4kHz band. The mean time to transition in all bands, and for measurements at all joint locations, is about 60msec. This may be interpreted such that the mean-free-path of energy decay is also about 6 times the total shortest path length in the truss; that is, the waves reverberate 6 times the truss length to reach equilibrium with the absorption processes in the truss, before reaching the final state of the slow decay rate $\eta_1$. However, this is not as robust as the first result, since decay rates depend on frequency. The energy build up and decay processes in the truss are schematically shown Fig. 3-23. Energy build up is an elastic process, whereas energy decay is a loss process which depends on frequency.
Figure 3-1: Frequency response (top) and impulse response (bottom) of Gaussian bandpass filter.
Figure 3-2: Characteristics of the input signal.
Figure 3-3: Frequency response of FIR lowpass filter with cut-off frequency $f_c = 25.6\text{kHz}$ (solid) and rectangular lowpass filter (dashed).
Figure 3-4: Magnitude of lowpass filtered impulse response function from 0 to 0.125 sec with $f_c = 25.6 kHz$. 
Figure 3-5: Magnitude of lowpass filtered impulse response function from 0 to 5\textit{msec} with $f_c = 25.6\text{kHz}$. 
Magnitude of Gaussian Octave Bandpass Filtered Impulse Response Function with $f_o=16kHz$, dB

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Figure 3-7: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125sec with $f_o = 8kHz$. 
Figure 3-8: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125 sec with $f_o = 4kHz$. 
Figure 3-9: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125 sec with \( f_o = 2kHz \).
Figure 3-10: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125 sec with $f_o = 1 kHz$. 
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Figure 3-12: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125 sec with \( f_o = 0.25 \text{kHz} \).
Figure 3-13: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 0.125sec with $f_o = 0.125kHz$. 
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Figure 3-15: Magnitude of Gaussian octave bandpass filtered impulse response function from 0 to 5msec with $f_o = 8kHz$ and predicted major early arrival times ($\circ$).
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Figure 3-23: Schematic of energy build up and decay processes in the truss
Table 3.2: Early arrival times of Gaussian bandpass filtered elastic waves with $f_o = 16 kHz$.

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Table 3.3: Early arrival times of Gaussian bandpass filtered elastic waves with $f_o = 8kHz$.

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Table 3.4: Early arrival times of Gaussian bandpass filtered elastic waves with $f_o = 4kHz$.

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Chapter 4

Conclusions

4.1 Summary and Conclusions

A study of elastic waves in a three-dimensional truss with use of spatial spectral analysis and temporal impulse analysis, was conducted within a single-input/multiple-output context. The frequency and time domain experiments were performed to investigate both the magnitude and phase of the transfer function at the joints of the truss.

The results of the frequency domain study can be summarized:

- Spatial coherence of the total elastic wave field is high throughout the truss except at stopbands caused by the periodicity of the truss and in bands at which the strut lengths equal the integral multiples of half the elastic wavelengths. The strongest stopbands are $3kHz, 5kHz, 10kHz, 14kHz$, etc.

- The extracted wavenumber, phase speed, time delay, group speed and group delay from the phase of the cross-spectra show that the elastic waves propagate both in positive and negative axial directions through the joints of the truss due to reverberation and scattering.
The frequency-wavenumber spectrum shows that high inertance spectrum bands lie between approximately -10 dB and 10 dB re 1/\text{kg} and the wavenumber spectrum in these bands becomes flatter with increasing frequency. The wave propagates through the joints both in positive and negative axial direction and this is caused by phase randomness.

The results of the time domain study can be summarized:

- The transient elastic wave energy travels with approximately constant group delay in the high frequency bands, and the measured arrival times agree with the predicted times in the two highest octave bands.

- Decay rates of the elastic wave field are not constant, but functions of frequency and location; The major decay rate increases with increasing frequency up to 8.4 dB/10 msec in the 16 kHz octave band, and decreases with increasing distance down to near 3.0 dB/10 msec in the 4 kHz octave band. The minor decay rate is approximately invariant with position and decreases with increasing frequency from 2.0 to 0.0 dB/10 msec.

- The mean-free-paths of energy build up and energy decay are about 6 times the total axial length of the truss; that is, the waves reverberate 6 times the truss length to reach the maximum energy build up, and also the equilibrium with the absorption processes in the truss. However, the latter is not as robust as the former since decay rates depend on frequency.

From the above results, I conclude that a broad wavenumber distribution at the joints of the truss is developed over all frequency bands. Thus the wave propagates in both positive and negative axial directions and this is due to the multi-paths of the truss structure. The multiple paths cause phase randomness even though the phase of the cross-spectra at the joints shows an approximately linear relationship. The results of time domain experiment shows roughly constant group delay of wave propagation with a transient input source. However, this tendency will disappear after the truss becomes fully energized and the phase at the joints becomes random.
4.2 Future Work

I investigated the spatial spectra of the MIT truss to understand the response at the joints with a simple single-input/multiple-output system.

However, in a real application the truss will carry many equipments which may or may not be correlated with each other. Thus it would be desirable to study the truss as a multiple-input/multiple-output system with correlated and uncorrelated sources.

If the sources are completely uncorrelated, superposition of single-input/multiple-output systems can characterize this multiple-input/multiple-output system. For the partially correlated source case, problems in designing reasonable sources should be overcome to investigate a multiple-input/multiple-output system.

I used as input only forcing at a base strut. However, to fully understand the spatial spectra of truss structures, experiments with different input locations, such as on plates and joints as well as with different truss configurations, are also desirable.
Appendix A

Timoshenko Beam Theory

The Timoshenko beam theory includes effects of rotary inertia and shear deformation which are neglected in the Euler-Bernoulli beam theory[23]. With a free vibration, the Timoshenko beam equation can be expressed as

\[
\frac{EI}{\rho A} \frac{\partial^4 z}{\partial x^4} + \frac{\partial^2 z}{\partial t^2} - \frac{I}{A} \left(1 + \frac{E}{KG}\right) \frac{\partial^4 z}{\partial x^2 \partial t^2} + \frac{\rho I}{KA} \frac{\partial^4 z}{\partial t^4} = 0
\]  

(A.1)

where

\[E = \text{Young’s modulus},\]
\[G = \text{shear modulus},\]
\[I = \text{area moment of inertia},\]
\[A = \text{area of beam},\]
\[\rho = \text{density of beam},\]
\[K = \text{Timoshenko’s shear coefficient, and}\]
\[z = \text{transverse displacement of beam}.\]

The dispersion relation of the Timoshenko beam equation Eq. (A.1) becomes

\[k = \frac{1}{2} \sqrt{2A\omega^2 + 2\sqrt{(A^2 - 4C)\omega^4 + 4B\omega^2}} \]

(A.2)

where

\[A = \frac{\rho}{E} + \rho KG, \quad B = \frac{\rho A}{EI}, \quad \text{and} \quad C = \frac{\rho^2}{EG}.\]
Eq. (A.2) is different from the dispersion relation of an Euler-Bernoulli beam which is

\[ k = \frac{4}{E I} \sqrt{\frac{\rho A}{E I}} \sqrt{\omega}. \quad (A.3) \]

The phase speed and group speed of waves on a Euler-Bernoulli beam are

\[ C_{\text{phase}} = \frac{\omega}{k} = \sqrt{\omega \sqrt{\frac{E I}{\rho A}}} = \sqrt{\omega \kappa \sqrt{\frac{E}{\rho}}}, \quad (A.4) \]

\[ C_{\text{group}} = \frac{\partial \omega}{\partial k} = 2 \sqrt{\omega \kappa \sqrt{\frac{E}{\rho}}} \quad (A.5) \]

where \( \kappa \) is a radius of gyration. For the MIT truss the radius of gyration becomes

\[ \kappa = \frac{1}{2} \sqrt{r_{\text{outer}}^2 + r_{\text{inner}}^2} \quad (A.6) \]

thus \( \kappa = 0.0039m \).

The phase speed and group speed of flexural wave on a Timoshenko beam are

\[ C_{\text{phase}} = \frac{\omega}{k} = \frac{\omega}{\frac{1}{2} \sqrt{2A \omega^2 + 2 \sqrt{(A^2 - 4C)\omega^4 + 4B\omega^2}}} \quad (A.7) \]

\[ C_{\text{group}} = \frac{\partial \omega}{\partial k} = \frac{\omega}{\frac{4}{4A \omega + \sqrt{(4A^2 - 16C)\omega^4 + 8B\omega}}} \quad (A.8) \]
For more accurate calculation I choose the Timoshenko’s shear coefficient of a thin-walled round tube as

\[ K = \frac{2(1 + \nu)}{4 + 3\nu}, \]  

(A.9)

and set \( \nu \) be 1/2 to satisfy the Timoshenko beam equation in the limit of zero wavelength in the waves on high frequency region; thus, \( K \) becomes 0.545 instead of 0.5[24][25]. The dispersion relation and flexural wave speed of the MIT truss with the Timoshenko and Euler-Bernoulli beam theory are shown in Fig. A-1 and Fig. A-2 respectively. The wave speeds of the truss are shown in Table A.1.

![Dispersion Relation of Timoshenko and Euler-Bernoulli Beam](image)

Figure A-1: Dispersion relation of Timoshenko beam (solid) and Euler-Bernoulli beam (dashed).
Figure A-2: Flexural wave speed of Timoshenko and Euler-Bernoulli beams. Solid and dashed lines are phase speed of Timoshenko and Euler-Bernoulli beam respectively, and dash-dot and dotted lines are group speed of Timoshenko and Euler-Bernoulli beam respectively.
Table A.1: Elastic wave speeds in the MIT truss.

<table>
<thead>
<tr>
<th></th>
<th>Young's Modulus of Aluminum (GPa)</th>
<th>70</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Shear Modulus of Aluminum (GPa)</td>
<td>26</td>
</tr>
<tr>
<td></td>
<td>Density of Aluminum (kg/m³)</td>
<td>2700</td>
</tr>
<tr>
<td></td>
<td>Compressional wave Speed (m/sec)</td>
<td>5092</td>
</tr>
<tr>
<td></td>
<td>Torsional wave Speed (m/sec)</td>
<td>3103</td>
</tr>
<tr>
<td></td>
<td>Flexural wave Speed (m/sec)</td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Octave Band $f_o$ (kHz)</th>
<th>Euler-Bernoulli Beam $C_{phase}$</th>
<th>$C_{group}$</th>
<th>Timoshenko Beam $C_{phase}$</th>
<th>$C_{group}$</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.125</td>
<td>126</td>
<td>251</td>
<td>126</td>
<td>251</td>
</tr>
<tr>
<td>0.25</td>
<td>178</td>
<td>355</td>
<td>177</td>
<td>354</td>
</tr>
<tr>
<td>0.5</td>
<td>251</td>
<td>503</td>
<td>250</td>
<td>497</td>
</tr>
<tr>
<td>1</td>
<td>355</td>
<td>711</td>
<td>353</td>
<td>696</td>
</tr>
<tr>
<td>2</td>
<td>503</td>
<td>1005</td>
<td>496</td>
<td>964</td>
</tr>
<tr>
<td>4</td>
<td>711</td>
<td>1422</td>
<td>691</td>
<td>1309</td>
</tr>
<tr>
<td>8</td>
<td>1005</td>
<td>2011</td>
<td>951</td>
<td>1715</td>
</tr>
<tr>
<td>16</td>
<td>1422</td>
<td>2844</td>
<td>1275</td>
<td>2116</td>
</tr>
</tbody>
</table>
Appendix B

MATLAB Codes

Chapter 2

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
%%%%%
Coherence of Elastic Waves at Joints of Truss
%%%%%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

Force at B4 in Z-direction & REF at C04 accelerometer

clear;

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
%%%%%
X Force at B4 in Z-direction & REF at C04 accelerometer
%%%%%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

%data,q1,q2,q3,q4,q5,q6,q7]=ReadUniversal('s5b4b4z_25K.unv');

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
%%%%%
X frequency band 25kHz
load /arcticl/jhpark/truss/Febl3/b4c4z_old
mf=25600
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
%%%%%
X frequency band 5kHz WhiteNoise 25kHz
%load /arcticl/jhpark/truss/Febl3/b4b4zS5_25K
%mf= 6400
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
%%%%%
X frequency band 1kHz WhiteNoise 25kHz
%load /arcticl/jhpark/truss/Febl3/b4b4zS1_25K
%mf= 1600

%%%%%
%%%%%
mf % maxmun frequency in Hz
nf=3200; % number of frequency data
f=[0:mf/nf:mf];
fkHz=f/le3; % frequency in kHz

%%%%%
%%%%%
% impedance head correction curve
x=[1000,2000,4000,10000,20000,30000];
y=[0,0,0,-.8,-5.5,-10];
yi=spline(x,y,f);
yi=(10.^(yi/10)).';

ff=real(data(:,1)); % force auto-specturm
ff=ff./(yi.^2); % force correction

aa=real(data(:,24)); % accel auto-specturm

%%%%%
%%%%%
% This is C joint arrangement
ct=real(data(:,4)); % c02 4
c04=real(data(:,2)); % c04 2
c06=real(data(:,11)); % c06 11
cc8=real(data(:,15)); % c08 15
c10=real(data(:,19)); % c10 19
c12=real(data(:,23)); % c12 23

%%%%%
%%%%%
% cross-specturm w.r.t. ref as force
cf02=data(:,27)./yi; % cfr02 27
%! cross-spectrum w.r.t. ref acceleration at \(c_{04}\)
\[\text{Kf0402} = \left(\frac{|\text{abs(cf02)}|^2}{f_f \cdot c_{02}}\right);\]
\[\text{Kf0404} = \left(\frac{|\text{abs(cf04)}|^2}{f_f \cdot c_{04}}\right);\]
\[\text{Kf0406} = \left(\frac{|\text{abs(cf06)}|^2}{f_f \cdot c_{06}}\right);\]
\[\text{Kf0408} = \left(\frac{|\text{abs(cf08)}|^2}{f_f \cdot c_{08}}\right);\]
\[\text{Kf0410} = \left(\frac{|\text{abs(cf10)}|^2}{f_f \cdot c_{10}}\right);\]
\[\text{Kf0412} = \left(\frac{|\text{abs(cf12)}|^2}{f_f \cdot c_{12}}\right);\]

%! Coherence Factor ref: Force at \(C_{04}\)
\[|Saf|^{-2}\]
\[\text{Kf} = \left(\frac{S_{af}^2}{S_{af}^2}\right);\]

%! Coherence Factor ref: Acceleration at \(B_{04}\)
\[\left|a_{a2a1}\right|^{-2}\]
\[\text{K_a} = \left(\frac{|\text{abs(conj(a2a1)*}|^2}{c_{04} \cdot c_{02}}\right);\]

%! unwrap phase of cross-spectrum
\[\text{dPHASE} = \text{unwrap(angle(cf04))};\]
\[\text{dPHASE2} = \text{unwrap(angle(cf06))};\]
\[\text{dPHASE3} = \text{unwrap(angle(cf08))};\]
\[\text{dPHASE4} = \text{unwrap(angle(cf10))};\]
\[\text{dPHASE5} = \text{unwrap(angle(cf12))};\]

\[\text{cp} = \frac{\text{v}}{k}, \text{Phase Speed}\]
\[\text{c1} = (f \cdot 2\pi) / \text{k1};\]
\[\text{c2} = (f \cdot 2\pi) / \text{k2};\]
\[\text{c3} = (f \cdot 2\pi) / \text{k3};\]
\[\text{c4} = (f \cdot 2\pi) / \text{k4};\]

\[\text{tau} = \text{dx} / \text{cp}, \text{Time Delay}\]
\[\text{Tc1} = (\text{ones(size(c1))}) \cdot (4.7/6);\]
\[\text{Tc2} = (\text{ones(size(c1))}) \cdot (4.7/6);\]
\[\text{Tc3} = (\text{ones(size(c1))}) \cdot (4.7/6);\]
Tc4=(ones(size(c1)).*(4.7/6))./c4;

% % % % Cg=dv/dk, Group Speed
omega=f.*(2*pi);
dv=diff(omega);
dx=diff(x);
dk2=diff(k2);
dk3=diff(k3);
dk4=diff(k4);
Cg1=dv./dx;
Cg2=dv./dx';
Cg3=dv./dx';
Cg4=dv./dx';

%%%%% to find phase slope in degrees by finite difference
up1=diff((phase1.*(180/pi)).');
up2=diff((phase2.*(180/pi)).');
up3=diff((phase3.*(180/pi)).');
up4=diff((phase4.*(180/pi)).');
up5=diff((phase5.*(180/pi)).');
dfk=diff(fk);

%%%%%%%% slope
slp1=up1./dfk;
slp2=up2./dfk;
slp3=up3./dfk;
slp4=up4./dfk;
slp5=up5./dfk;

%%%%% total slope
asl1=mean(slp1);
asl2=mean(slp2);
asl3=mean(slp3);
asl4=mean(slp4);
asl5=mean(slp5);

slopetot=[asl1 asl2 asl3 asl4 asl5];

%%%%% f16 octave band slope
af16sll=mean(slp1(1416:2830));
af16sll2=mean(slp2(1416:2830));
af16sll3=mean(slp3(1416:2830));
af16sll4=mean(slp4(1416:2830));
af16sll5=mean(slp5(1416:2830));
slopef16=[af16sll af16sll2 af16sll3 af16sll4 af16sll5];

%%%%% f8
af8sll=mean(slp1(708:1416));
af8sll2=mean(slp2(708:1416));
af8sll3=mean(slp3(708:1416));
af8sll4=mean(slp4(708:1416));
af8sll5=mean(slp5(708:1416));
slopeg8=[af8sll af8sll2 af8sll3 af8sll4 af8sll5];

%%%%% f4
af4sll=mean(slp1(355:708));
af4sll2=mean(slp2(355:708));
af4sll3=mean(slp3(355:708));
af4sll4=mean(slp4(355:708));
af4sll5=mean(slp5(355:708));
slopeg4=[af4sll af4sll2 af4sll3 af4sll4 af4sll5];

S='TOT f16 f8 f4';
SLOPE=[slopetot' slopeg16' slopeg8' slopeg4'];
slope= ( round ( SLOPE.*100 ) )./100

%%%%% change mf & mfk for plot axis
% delta f
df=mf/nf
mfk=floor(mf/1e3);
if mfk=1
   mfk=1.5;
end

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axis tailoring

mf;
mf=mfk*le3;
figure(1)
orient tall
subplot(511)
plot(fk,Kf0404)
axis([0 mfk 0 1])
ylabel('Coh(F,c04)')
title('Coherence Function between Input Force and Acceleration at Joints')
subplot(512)
plot(fk,Kf0406)
axis([0 mfk 0 1])
ylabel('Coh(F,c06)')
subplot(513)
plot(fk,Kf0408)
axis([0 mfk 0 1])
ylabel('Coh(F,c08)')
subplot(514)
plot(fk,Kf0410)
axis([0 mfk 0 1])
ylabel('Coh(F,c10)')
subplot(515)
plot(fk,Kf0412)
axis([0 mfk 0 1])
ylabel('Coh(F,c12)')
xlabel('Frequency in kHz')
figure(2)
orient tall
subplot(511)
semilogy(fk,(abs(cf04)))
axis([0 mfk 1e-5 1e0])
ylabel('C(F,c04)')
title('Magnitude of Cross-spectra between Input Force and Acceleration at Joints in Nm/s^2')
subplot(512)
semilogy(fk,(abs(cf06)))
axis([0 mfk 1e-6 1e-13])
ylabel('C(F,c06)')
subplot(513)
semilogy(fk,(abs(cf08)))
axis([0 mfk 1e-6 1e-1])
ylabel('C(F,c08)')
subplot(514)
semilogy(fk,(abs(cf10)))
axis([0 mfk 1e-7 1e-2])
ylabel('C(F,c10)')
subplot(515)
semilogy(fk,(abs(cf12)))
axis([0 mfk 1e-7 1e-2])
ylabel('C(F,c12)')
xlabel('Frequency in kHz')
figure(3)
orient tall
subplot(511)
plot(fk,angle(cf04))
axis([0 mfk -pi pi])
ylabel('C(F,c04)')
title('Phase Angle of Cross-spectra between Input Force and Acceleration at Joints, radians')
subplot(512)
plot(fk,angle(cf06))
axis([0 mfk -pi pi])
ylabel('C(F,c06)')
subplot(513)
plot(fk,angle(cf08))
axis([0 mfk -pi pi])
ylabel('C(F,c08)')
subplot(514)
plot(fk,angle(cf10))
axis([0 mfk -pi pi])
ylabel('C(F,c10)')
subplot(515)
plot(fk,angle(cf12))
axis([0 mfk -pi pi])
ylabel('C(F,c12)')
xlabel('Frequency in kHz')

figure(4)
orient tall
subplot(511)
plot(fk,phasel)
axis([0 mfk -1.5e3 1e2])
ylabel('C(F,c04)')
title('Phase Angle of Cross-spectra between Input Force and Acceleration at Joints, radians')
subplot(512)
plot(fk,phase2)
axis([0 mfk -1.5e3 1e2])
ylabel('C(F,c06)')
subplot(513)
plot(fk,phase3)
axis([0 mfk -1.5e3 1e2])
ylabel('C(F,c08)')
subplot(514)
plot(fk,phase4)
axis([0 mfk -1.5e3 1e2])
ylabel('C(F,c10)')
subplot(515)
plot(fk,phase5)
axis([0 mfk -1.5e3 1e2])
ylabel('C(F,c12)')
xlabel('Frequency in kHz')

figure(5)
orient tall
subplot(411)
loglog(f,kl,'-',f,abs(kl),':')
axis([0 mf le-3 le3])
set(gca,'YTick',[],le-3 le-2 le-1 1 lel le2 le3)
ylabel('k(c04,c06)')
title('Wavenumber, k = dPHASE/dX, in 1/m, dx=0.783m (1cell)')
legend('Positive','Negative')
subplot(412)
loglog(f,k2,'-',f,abs(k2),':')
axis([0 mf le-3 le3])
set(gca,'YTick',[],le-3 le-2 le-1 1 lel le2 le3)
ylabel('k(c06,c08)')
subplot(413)
loglog(f,k3,'-',f,abs(k3),':')
axis([0 mf le-3 le3])
set(gca,'YTick',[],le-3 le-2 le-1 1 lel le2 le3)
ylabel('k(c08,c10)')
subplot(414)
loglog(f,k4,'-',f,abs(k4),':')
axis([0 mf le-3 le3])
set(gca,'YTick',[],le-3 le-2 le-1 1 lel le2 le3)
ylabel('k(c10,c12)')
xlabel('Frequency in Hz')

figure(6)
orient tall
subplot(411)
loglog(f,cl,'-',f,abs(cl),':')
axis([0 mf leO leb])
ylabel('Cp(c04,c06)')

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title('Phase Speed, \(C_p = \frac{w}{k}\), in m/sec')
legend('Positive', 'Negative')

subplot(412)
loglog(f,c2,'-',f,abs(c2),':')
axis([0 1e6 1e5])
ylabel('\(C_p(06,08)\)')
subplot(413)
loglog(f,c3,'-',f,abs(c3),':')
ageis([0 mf 1e6 1e5])
%grid
ylabel('\(C_p(08,10)\)')

subplot(414)
loglog(f,c4,'-',f,abs(c4),':')
axis([0 mf 1e6 1e5])
ylabel('\(C_p(10,12)\)')

xlabel('Frequency in Hz')

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(7)
orient tall
subplot(411)
loglog(f,Tc1,'-',f,abs(Tc1),':')
ylabel('\(\tau(04,06)\)')
axis([0 1e-5 1e0])
legend('Positive', 'Negative')
title('Time Delay, \(\tau = \frac{\Delta X}{C_p}\), in sec')
subplot(412)
loglog(f,Tc2,'-',f,abs(Tc2),':')
ylabel('\(\tau(06,08)\)')
axis([0 1e-5 1e0])
subplot(413)
loglog(f,Tc3,'-',f,abs(Tc3),':')
ylabel('\(\tau(08,10)\)')
axis([0 1e-5 1e0])
subplot(414)
loglog(f,Tc4,'-',f,abs(Tc4),':')
ylabel('\(\tau(10,12)\)')
axis([0 1e-5 1e0])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(8)
orient tall
subplot(411)
loglog(f(1:length(f)-1),Cg1,'-',f(1:length(f)-1),abs(Cg1),':')
ylabel('\(C_g(04,06)\)')
axis([0 1e0 1e5])
legend('Positive', 'Negative')
title('Group Speed, \(C_g = \frac{\Delta V}{\Delta x}\), in m/sec')
subplot(412)
loglog(f(1:length(f)-1),Cg2,'-',f(1:length(f)-1),abs(Cg2),':')
ylabel('\(C_g(06,08)\)')
axis([0 1e0 1e5])
subplot(413)
loglog(f(1:length(f)-1),Cg3,'-',f(1:length(f)-1),abs(Cg3),':')
ylabel('\(C_g(08,10)\)')
axis([0 1e0 1e5])
subplot(414)
loglog(f(1:length(f)-1),Cg4,'-',f(1:length(f)-1),abs(Cg4),':')
ylabel('\(C_g(10,12)\)')
axis([0 1e0 1e5])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(9)
orient tall
subplot(411)
loglog(f(1:length(f)-1),Tg1,'-',f(1:length(f)-1),abs(Tg1),':')
ylabel('\(G_{\tau}(04,06)\)')
axis([0 1e-5 1e0])
legend('Positive', 'Negative')
title('Group Delay, \(G_{\tau} = \frac{\Delta X}{C_g}\), in sec')
subplot(412)
loglog(f(1:length(f)-1),Tg2,'-',f(1:length(f)-1),abs(Tg2),':')
ylabel('Gtau(c06,c08)')
axis([0 mf le-5 le0)
subplot(413)
loglog(f(1:length(f)-1),Tg3,'-',f(1:length(f)-1),abs(Tg3),':')
ylabel('Gtau(c08,c10)')
axis([0 mf le-5 le0)
subplot(414)
loglog(f(1:length(f)-1),Tg4,'-',f(1:length(f)-1),abs(Tg4),':')
ylabel('Gtau(c10,c12)')
axis([0 mf le-5 le0])

return

Coherence of Elastic Waves at Base Struts of Truss

Force at B4 in Z-direction & REF at B04 acclerometer
clear;
%[dataqi,q2,q3,q4,q4, q6,q7]=ReadUniversal('s5b4b4z_25K.unv');
%frequency band 25kHz
load /arcticl/jhpark/truss/Feb13/b4b4z-old
mf=25600
%frequency band 5kHz WhiteNoise 25kHz
%load /arcticl/jhpark/truss/Feb13/b4b4zS5.25K
Yanf= 6400
%frequency band 1kHz WhiteNoise 25kHz
%load /arcticl/jhpark/truss/Feb13/b4b4zS1_25K
nf= 1600
mf
% number of frequency data
f=[0:mf/nf:mf];
fk=f./le3;
% frequency in kHz
%impedance head correction curve
x=[1000,2000,4000,10000,20000,30000];
y=[0,0,0,0,0,0];
yi=spline(x,y,f);
yi=(10.^(yi/10)).';
ff=real(data(:,1));
% force auto-specturm
rf=ff./yi; % force correction
aa=real(data(:,24)); % accel auto-specturm
% This is B strut arrangement
b04=real(data(:,1));
b06=real(data(:,10));
b08=real(data(:,14));
b10=real(data(:,18));
b12=real(data(:,22));
% cross-spectrum w.r.t. ref as force
bf02=bf2./yi;
bf04=bf4./yi;
bf06=bf6./yi;
bf08=bf8./yi;
bf10=bf10./yi;
bf12=bf12./yi;
% cross-spectrum w.r.t. ref as acceleration at b04
ba02=ba2;
bai0=data(:,63);
bai2=data(:,67);
clear data q1 q2 q3 q4 q5 q6 q7

% Coherence Factor ref: Force at B04

Kf = ------------
     SaaSff

Kf0402 =(abs(bf02)).^2./(ff.*b02);
Kf0404 =(abs(bf04)).^2./(ff.*b04);
Kf0406 =(abs(bf06)).^2./(ff.*b06);
Kf0408 =(abs(bf08)).^2./(ff.*b08);
Kf0410 =(abs(bf10)).^2./(ff.*b10);
Kf0412 =(abs(bf12)).^2./(ff.*b12);

% Coherence Factor ref: Accel at B04

Ka = ------------ , |ai1a2e|^2 = (abs(conj(a2a1e)))^2
     aial*a2a2*

Ka0402 =(abs(ba02)).^2./(b04.*b02);
Ka0404 =(abs(ba04)).^2./(b04.*b04);
Ka0406 =(abs(ba06)).^2./(b04.*b06);
Ka0408 =(abs(ba08)).^2./(b04.*b08);
Ka0410 =(abs(ba10)).^2./(b04.*b10);
Ka0412 =(abs(ba12)).^2./(b04.*b12);

% change mf & mfk for plot axis
% delta f
df=mf/nf
mfk=floor(mf/1e3);
if mfk==1
    mfk=1.5;
end

figure(1)
orient tall
subplot(511)
plot(fk,Kf0404)
axis([0 mfk 0 1])
ylabel('Coh(F,b04)')
title('Coherence Function between Input Force and Acceleration at Base Struts')

subplot(512)
plot(fk,Kf0406)
axis([0 mfk 0 1])
ylabel('Coh(F,b06)')

subplot(513)
plot(fk,Kf0408)
axis([0 mfk 0 1])
ylabel('Coh(F,b08)')

subplot(514)
plot(fk,Kf0410)
axis([0 mfk 0 1])
ylabel('Coh(F,b10)')

subplot(515)
plot(fk,Kf0412)
axis([0 mfk 0 1])
ylabel('Coh(F,b12)')
xlabel('Frequency in kHz')
return

Natural Frequencies of Truss Struts
E=70e9;
G=26e9;
rho=2700;
kappa=3.9e-3;
Cl=sqrt(E/rho);
Ct=sqrt(G/rho);

lengths of struts
L1=0.484;
L2=0.700;
L3=0.810;

for L=0.484

i=1; %dummy control var
while flong < 25e3,
   flong=i*Cl*L1;
   flongLi(i)=flong;
   i=i+1;
end

i=1; %dummy control var
while ftors < 25e3,
   ftors=i*Ct*L1;
   ftorsLI(i)=ftors;
   i=i+1;
end

fn for bending wave of an E-B beam
fn=(bL)-2*Cl*kappa/(2*pi*L^2)
bL=[4.730041 7.853205 10.995608 14.137165];
fflexL1 = bL.^2.*Cl*kappa/(2*pi*L1^2);
i=4; %dummy control var
while fflexL1(i) < 25e3,
   bl(i+1)=bl(i)+pi;
   fflexLi(i+1)=bl(i+1)-2*Cl*kappa/(2*pi*L1^2);
   i=i+1;
end

for L=0.700

i=1; %dummy control var
while flong < 25e3,
   flong=i*Cl*L2;
   flongL2(i)=flong;
   i=i+1;
end

i=1; %dummy control var
while ftors < 25e3,
   ftors=i*Ct*L2;
   ftorsL2(i)=ftors;
   i=i+1;
end

fn for bending wave of an E-B beam
fn=(b_nL)^2*Cl*kappa/(2*pi*L^2)

bL=[4.730041 7.853205 10.995608 14.137165];
fflexL2 = bL.^2.*Cl*kappa/(2*pi*L2^2);
i=4; % dummy control var
while fflexL2(i) < 25e3,
bL(i+1)=bL(i)+pi;
fflexL2(i+1)=bL(i+1)^2*Cl*kappa/(2*pi*L2^2);
i=i+1;
end
SignlongL2=ones(1,length(flongL2));
SigntorsL2=ones(1,length(ftorsL2));
SignflexL2=ones(1,length(fflexL2));

for L=0.810...

flong=0;
ftors=0;
i=1; % dummy control var
while flong < 25e3,
flong=i*Cl*L3;
ftorsL3(i)=ftors;
i=i+1;
end

i=1; % dummy control var
while ftors < 25e3,
flong=0;
ftors=i*Cl*L3;
ftorsL3(i)=ftors;
i=i+1;
end

for L=0.810...

fn for bending wave of an E-B beam

figure(i)
orient tall
subplot(311)
set(gca,'YTickLabels','')
ylabel('')
axis([0 25 0 1.2])
subplot(312)
plot(ftorsL1/le3,SigntorsL1.*.6,'x',ftorsL2/le3,SigntorsL2.*.8,'o',... ftorsL3/le3,SigntorsL3,'+',ftorsL1/le3,SigntorsL1.*.4,'.',... ftorsL2/le3,SigntorsL2.*.4,'.',ftorsL3/le3,SigntorsL3.*.4,'.') title('Torsional Wave')
set(gca,'YTickLabels','')
ylabel('')
axis([0 25 0 1.2])
subplot(313)
plot(fflexL1/le3,SignflexL1.*.6,'x',fflexL2/le3,SignflexL2.*.8,'o',... fflexL3/le3,SignflexL3,'+',fflexL1/le3,SignflexL1.*.4,'.',... fflexL2/le3,SignflexL2.*.4,'.',fflexL3/le3,SignflexL3.*.4,'.') title('Flexural Wave (Euler-Bernoulli Beam)')
set(gca,'YTickLabels','')
xlabel('Frequency in kHz')
Global to Local transition of truss dynamics

rho=2700; % density of A16061
E=70e9; % Young's modulus
G=26e9; % Shear modulus
K=0.950e-3; % radius of gyration
f=[0:10:25e3];
w=f.*2*pi;

Cl=sqrt(E/rho); % longitudinal wave speed
Ct=sqrt(G/rho); % torsional wave speed
\[ A = \rho/(E + \rho/(tc^2 G)) \]
\[ B = \rho/E/K^2 \]
\[ C = \rho^2/(E tc G) \]

\[ kT = \sqrt{w_2^2 \cdot 2A + \sqrt{w_4^2 (A^2 - 4C + w_2 B^4).2}} \cdot 0.5 \]
\[ CT_p = \sqrt{w_2^2 \cdot 2A + \sqrt{w_4^2 (A^2 - 4C + w_2 B^4)}.2} \]

%%%%%

Euler Beam

\[ k_B = \sqrt{w_2 \cdot (\rho/E/K^2)^{1/4}} \]
\[ CB_p = \sqrt{w_2 \cdot (E/K^2/\rho)^{1/4}} \]
\[ CB_g = \sqrt{w_2 \cdot 2 \cdot (E/K^2/\rho)^{1/4}} \]

%%%%%

Array Processing with Cross-Spectra at Joints

%-------------------

load /arcticl/jhpark/truss/Febl3/b4b4zS25
Xmf=25600
load /arcticl/jhpark/truss/Febl3/b4b4zS5_25K
mf=6400
load /arcticl/jhpark/truss/Febl3/b4b4zS1_25K
Xmf=1600

% put proper stuff in A
A(:,1:)=data(:,27)./data(:,1,:);
A(2,:)=(data(:,30)./data(:,1)).';
A(3,:)=(data(:,34)./data(:,1)).';
A(4,:)=(data(:,38)./data(:,1)).';
A(5,:)=(data(:,42)./data(:,1)).';
A(6,:)=(data(:,46)./data(:,1)).';

% Frequency wavenumber spectrum
i = sqrt(-1);
N=6
nf=3201
df=32
dx=0.783
mk=400
nk=400
mk=pi/(dx)
dx=2*mk/nk

j = [0:N-1]';
z([1:N,1]) = (j*dx)';
z([1:N,2]) = (0+j)';
for f=0:df:nf,
    fmf/(1000*nf)
    fi=f/df+1;

    % calculate sx
    sx = zeros(N,N);
    for n=1:N,
        xa(n) = A(n,f+1);
    end

    sxa = xa.' * conj(xa);

    % calculate p(v,k)
    for k = -mk:dk:mk
        c = 1+ (kvmk)/dk;
        kt = [k;0];
        et = exp(-i*etak);
        etn = et.*hamming(N);
        norm = sum(hamming(N));
        pa(f,c) = (etn*sxa*et)/(N*N);
        end
    end

    pa = 20*log10(abs(pa));
    f=[0:nf/df]fmf/((nf/df)*1000);
    k=[-mk:dk:mk];

    %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%% figure(1)
    imagesc(k,f,pa)
    axis('xy')
    colormap(gray),colorbar
    shading('flat')
    xlabel('Wavenumber in 1/m');
    ylabel('Frequency in kHz')
    title('Frequency Wavenumber (f-k) Spectrum dB re i/kg')
    axis([-mk*a,mk*a,O,mf/le3])

    %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%% figure(2)
    plot(f,pa(:,202))
    axis([O mf/ie3 -140 40])
    title('Inertance Frequency Spectrum at k=0.02 m^-1')
    xlabel('Frequency in kHz')
    ylabel('Inertance in dB re 1/kg')

    %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%% figure(3)
    plot(k,pa(85,:),'--',k,pa(60,:),'--',k,pa(46,:),'--',k,pa(24,:),'--')
    11=sprintf('%4.0fHz',floor(f(85).*le3));
    12=sprintf('%4.0fHz',floor(f(60).*le3));
    13=sprintf('%4.0fHz',floor(f(46).*le3));
    14=sprintf('%4.0fHz',floor(f(24).*le3));
    legend([11,12,13,14])
    axis([-mk*a mk*a -40 40])
    title('Inertance Wavenumber Spectrum')
xlabel('Wavenumber in 1/m')
ylabel('Magnitude dB re 1/kg')

find aliasing occurring frequency
with an assumption of the same elastic wave propagation
of aluminum struts

clear
rho=2700; % density of Al6061
E=70e9; % Young's modulus
G=26e9; % Shear modulus
K=3.950e-3; % radius of gyration
f=linspace(0,4e3,2000);
w=f.*(2*pi);
Cl=sqrt(E/rho); % longitudinal wave speed
Ct=sqrt(G/rho); % torsional wave speed
Kl=(f./Cl).*2*pi; % longitudinal wavenumber
Kt=(f./Ct).*2*pi; % torsional wavenumber

% Timoshenko Beam
tc=0.545; % Timoshenko shear coeff. for thin-walled pipe 0.5~0.545
A=(rho/E+rho/(tc*G));
B=(rho/E/K^2);
C=(rho^2/(E*tc*G));
kT=sqrt(w.^2*2*A+sqrt(w.^4*(A^2-4*C)+w.^2*B.*4).*2).*0.5;
CTp=w./kT;
CTg=((sqrt(w.^2*A.*4+sqrt(w.^2*(A^2-4*C)+w.^2*B.*4).*2)) ./ ... 
    (w.*A.*4+(w.^3*(A^2-4*C)+w.*B*8)./sqrt(w.^4.*(A^2-4*C)+w.^2*B.*4))).*4;

% wave length
% f/1e3,2*pi*ones(size(kB))/kB,...
% wavenumber
% k=2*pi/wavelength, 2*pi*freq/c
% Nyquist pi/dz
figure(4)
plot(f/1e3,kT,'-',...
f/1e3,Kt,'--',...
f/1e3,K1,'-.',...)
title('Wavenumber vs. Frequency')
legend('flex','tors','comp')
xlabel('Frequency in kHz')
ylabel('Wavenumber')
axis([0 4 0 pi/0.783])

return
Wave speed matrix to calculate arrival time

```matlab
clear
rho=2700; % density of Al6061
E=70e9; % Young's modulus
G=26e9; % Shear modulus
K=3.950e-3; % radius of gyration
f=[16e3 8e3 4e3 2e3 1e3 .5e3 .25e3 .125e3];
w=f.*2*pi;

%% Timoshenko Beam
tc=0.545; % Timoshenko shear coeff. for thin-walled pipe 0.5~0.545
A=(rho/E+rho/(tc*G));
B=(rho/E/K^2);
C=(rho^2/(E*tc*G));

kT= sqrt(w.^2*2*A + sqrt(w.^4*(A^2-4*C)+w.^2*B*4).*2).*0.5;
CTp=w./kT;
CTg=(sqrt(w.^2*A*2 + sqrt(w.^4*A^2+w.^2*B*4).*2))./ (w.*A*4+(w.^3*A^2*4+w.*B*8)./sqrt(w.^4.*A-2+v.-2*B*4))).*4;
kTa= sqrt(w.^2*2*A + sqrt(wv.4*A^2+w.^2*B*4).*2).*0.5;
CTpa=w./kTa;

%% Euler
CBp=sqrt(w).*(E*K^2/rho)((1/4);
CBg=2*CBp;
Cl=sqrt(E/rho);
Ct=sqrt(G/rho);
Clt=[Cl Ct]';
Gg=[w' CTg' CTga' CBg'];
Cp=[w' CTp' CTpa' CBp'];

%% Timoshenko
% SRC at a4c4
Ttmp(1,:)= Ttmp(:,1);
for i=1:8,
    Ttmp(i+3,:)=Ttmp(:,1)./CTg(i)*1e3;
end
Ttmp;

%% Travel time matrix W/O WAVE-FORM CHANGE AFTER JOINTS
% dtime(1,1:5) = C04 C06 C08 C10 C12
% dtime(1:11,1) = Dist Tc Tt Tti6 8 4 2 1 0.5 0.25 0.125
```

---

Chapter 3
Btmp(1:3,1:10) = Ttmp(1:3,1:10);
for i=1:8,
    Btmp(i+3,:) = Btmp(1,:)./CBg(i)*1e3;
end

%%%%%
Travel time matrix W/O WAVE-FORM CHANGE AFTER JOINTS

dtime(1,1:5) : C04 C06 C08 C10 C12

dtime(1:11,1) : Dist Tc Tt Tf16 8 4 2 1 .5 .25 0.125

%%%%%
C04
Btime(:,1) = Btmp(:,1);

%%%%%
C06
Btime(:,2) = Btmp(:,2) + Btmp(:,3) + Btmp(:,4);

%%%%%
C08
Btime(:,3) = Btime(:,2) + Btmp(:,5) + Btmp(:,6);

%%%%%
C10
Btime(:,4) = Btime(:,3) + Btmp(:,7) + Btmp(:,8);

%%%%%
C12
Btime(:,5) = Btime(:,4) + Btmp(:,9) + Btmp(:,10);

Btime;

%%%%%
Calculate all travel time w/all wave possibilities

%%%%%
Possibility of traveling waves

%%%%%
C06 C06 (6 x 2)
C06 = [1 1; 1 2; 1 3; 2 2; 2 3; 3 3];

%%%%%
C06 C07 C08 (15 x 4)
C08 = [1 1 1 1; 1 1 1 2; 1 1 1 3; 1 1 2 2; 1 1 2 3; 1 1 3 3; 1 2 2 2; 1 2 2 3; 1 2 3 3; 1 3 3 3; 2 2 2 2; 2 2 2 3; 2 2 3 3; 2 3 3 3; 3 3 3 3];

%%%%%
C06 C06 C07 C08 (28 x 6)
C10 = [1 1 1 1 1 1; 1 1 1 1 1 2; 1 1 1 1 1 3; 1 1 1 1 2 2; 1 1 1 1 2 3; 1 1 1 1 3 3; 1 1 1 2 2 2; 1 1 1 2 2 3; 1 1 1 2 3 3; 1 1 1 3 3 3; 1 1 2 2 2 2; 1 1 2 2 2 3; 1 1 2 2 3 3; 1 1 2 3 3 3; 1 1 2 3 3 3; 1 1 2 3 3 3; 1 1 2 3 3 3];
calculate

set Travel-time matrix [comp, tors, flex]

freq 16 8 4 2 1 .5 .25 .125

Flexf 4 5 6 7 8 9 10

Traveltime=[Tmean(2:3) Tmean(Flexf)];

position=C12;
[M,N]=size(position);
time=zeros(M,1);
for i=1:M,
    for j=i:N,
        if position(i,j) == 1
            time(i) = time(i) + Traveltime(1);
        elseif position(i,j) == 2
            time(i) = time(i) + Traveltime(2);
        else
            time(i) = time(i) + Traveltime(3);
        end
    end
end

%%%%% Final time includind Comp, Tors, Flex at A4
Ftime(:,1)=time + ones(M,1).*Ttmp(2,2);
Ftime(:,2)=time + ones(M,1).*Ttmp(3,2);

%%%%% Flex time location 16 8 4 ... -> Ttmp(4:11,2)
Ftime(:,3)=time + ones(M,1).*Ttmp(Flexf,2);

%%%%% C04
Ftimei=zeros(6,1);
for i=1:6,
    if C04_1(i,i) == 1
        Ftimei(i)=Ttmp(2,1);
    elseif C04_1(i,1) == 2
        Ftimei(i)=Ttmp(3,1);
    else
        Ftimei(i)=Ttmp(Flexf,1);
    end
end

for i=1:6,
    if C04_1(i,2) == 1
        Ftimei(i)=Ftimei(i) + a4c4(2,1);
    elseif C04_1(i,2) == 2
        Ftimei(i)=Ftimei(i) + a4c4(3,1);
    end
end

Ftime2=zeros(10,1);
for i = 1:10,
if C04_2(i,1) == 1
  Ftime2(i)=Ttmp(2,2);
elseif C04_2(i,1) == 2
  Ftime2(i)=Ttmp(3,2);
else
  Ftime2(i)=Ttmp(Flexf,2);
end
end

for i=1:10,
  for j=2:3,
    if C04_2(i,j) == 1
      Ftime2(i)= Ftime2(i)+a4c4(2,1);
    elseif C04_2(i,j) == 2
      Ftime2(i)= Ftime2(i)+a4c4(3,1);
    else
      Ftime2(i)= Ftime2(i)+a4c4(Flexf,1);
    end
  end
end

%%%%%
C04 Arrival
C04dir=[Ttime(2,1); Ttime(3,1); Ttime(Flexf,1)];
round(C04dir.*100)./100
C04refAO4=[C04_1 round(Ftimel.*100) ./100]
C04refCO4AO4=CC04_2 round(Ftime2.*100)./100

%sampling frequency
Fs=65536;
% data length
Fl=8192;=f=[0:16383]/(Fl*2)*Fs;
t=[0:16383]/Fs;
f0=16e3;
for i=1:8,
   fc(i)=f0/2^i-1);
   f1(i)=fc(i)/sqrt(2);
   fbw(i)=sqrt((fc(i)-f1(i))^2/(2*6.906e-1));
   index for fc
   idx=floor(Fl*fc(i)/f(Fl))+1;
   %
end

for i=1:8,
  for j=1:F1,
    gaussfilter(i,j)=exp(-(fc(i)-f(j))^2/(2*6.906e-1));
    gaussfilter(i,F1+2-(j-1))=gaussfilter(i,j);
  end
end

fbw(1) % frequency bandwidth
t=linspace(-.125,.125,16384);
F=ifft(gaussfilter(i,:));
F=fftshift(F);
figure(1)
orient tall

subplot(211)
plot((f/1e3,gaussfilter(i,:))
axis([0 Fs/2e3 0 1.2])
grid
xlabel('Frequency in kHz')
ylabel('Magnitude')
title('Frequency Response Function of Gaussian Bandpass Filter with fo=16kHz')

subplot(212)
plot(t.*1e3,F)
axis([-1 1 -.4 .4])
xlabel('Time in msec')
ylabel('Amplitude in relative unit')
title('Impulse Response Function of Gaussian Bandpass Filter with fo=16kHz')

%%%%% Input Signal Characteristics, FIR Lowpass Filter Design and Filtered Signal
%%%%% INPUT Trigger freq: 1Hz Trigger: every frame
% Freq span: 22kHz Level: 10%
% spectral line: 3201 Slope: positive
% Max freq: 25600Hz
% Frame size: 8192 Pretrigger
% delta t: 1.52588e-5 samples: 100
% Frame length: 0.125sec time: 0.00152588
% Sampling freq: 65536.0Hz percentage: 1.2207
% [data,q1,q2,q3,q4,q5,q6,q7]=ReadUniversal('/arctic1/jhpark/truss/Oct30/Bpulse.unv');
load /arctic1/jhpark/truss/Oct30/Bpulse
Fs=65536;
F1=8192;
src = real(data(:,24).');
c04 = real(data(:,7).');
c06 = real(data(:,11).');
c08 = real(data(:,15).');
c10 = real(data(:,19).');
c12 = real(data(:,23).');
src = src - mean(src);
c04 = c04 - mean(c04);
c06 = c06 - mean(c06);
c08 = c08 - mean(c08);
c10 = c10 - mean(c10);
c12 = c12 - mean(c12);
t=[0:16383]/Fs;
f=[0:16383]/(F1*2)+Fs;
% XXXX FFT with double length
fsrc=fft(src,F1*2);
f04=fft(c04,F1*2);
f06=fft(c06,F1*2);
f08=fft(c08,F1*2);
f10=fft(c10,F1*2);
f12=fft(c12,F1*2);
% XXXX Deconvolution w/ SRC in Freq domain
cvsrc=fsrc./fsrc;
cvc04=f04./fsrc;
cvc06=f06./fsrc;
cvc08=f08./fsrc;
cvc10=f10./fsrc;
cvc12=f12./fsrc;
% XXXX inverse FFT
isrc=ifft(cvsrc);
ic04=ifft(cvc04);
ic06=ifft(cvc06);
ic08=ifft(cvc08);
ic10=ifft(cvcl0);
ic12=ifft(cvcl2);

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
\%b=remez(n,f,m);
\%y=filtfilt(b,a,x); \%a=l for FIR
%%%%%%25.6kHz LOWPASS
nf= \[0 0.8084 0.9094 1]\; m = \[1 1 0 0]\; n=40
b=remez(n,nf,m)*.9906;

%%%%%%%%%'/.%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
\FREQUENCY RESPONSE
[H,ff]=freqz(b,1,Fs/2,Fs);
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
\FILTER
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
ftsrc=filtfilt(b,1,isrc);
ftc04=filtfilt(b,1,ic04);
ftc06=filtfilt(b,1,ic06);
ftc08=filtfilt(b,1,ic08);
ftc10=filtfilt(b,1,ic10);
ftc12=filtfilt(b,1,ic12);
ftsrc=ftsrc-mean(ftsrc);
ftc04=ftc04-mean(ftc04);
ftc06=ftc06-mean(ftc06);
ftc08=ftc08-mean(ftc08);
ftc10=ftc10-mean(ftc10);
ftc12=ftc12-mean(ftc12);

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%
\'msec' index of time vector
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
t1=length(t)/256+6;
t2=length(t)/128+5;
t4=length(t)/64+8; \% 4.0131 msec
t5=length(t)/64+73; \% 5.0049 msec
t8=length(t)/32+14; \% 8.0109
t10=length(t)/32+146; \%10.0097
t12=length(t)/32+length(t)/64+20; \%12.0087
t15=length(t)/16-39; \%15.0146
t16=length(t)/16+26; \%16.0085
t20=length(t)/16+288; \%20.0042
t24=length(t)/16+660; \%24.0020
t32=length(t)/8+61; \%32.0129
t40=length(t)/8+675; \%40.0085

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(1)
orient tall
subplot(211)
plot(t(1:F1).*1e3,src)
title('Input Signal in Time Domain')
ylabel('acceleration (m/s^2)')
xlabel('Time in msec')
axis([0,t(tO).*1e3,-300,300])
subplot(212)
plot(f/1000,20*log10(fsrc))
title('Input Signal in Frequency Domain')
ylabel('accleration dB re m/s^2/Hz')
xlabel('Frequency in kHz')
axis([0,Fs/2000,0,80])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
orient portrait
plot(ff/ie3, 20*log10(abs(H)), nf/le3*(Fs/2),[0 0 -50 -50, '--'])
axis([0 Fs/2/1e3 -45 50])
title('Frequency Response of FIR Filter with fc=25.6kHz')
ylabel('20log(H)')
xlabel('Frequency in kHz')
axis([0 Fs/2/1e3 -45 50])
title('Frequency Response of FIR Filter with fc=25.6kHz')
ylabel('20log(H)')
xlabel('Frequency in kHz')

figure(3)
orient tall
subplot(611)
plot(t,20*log10(abs(ftsrc)))
title('Magnitude of Lowpass Filtered Impulse Response Function with fc=25.6kHz, dB')
ylabel('SRC')
axis([0,125e-3,-40,0])
subplot(612)
plot(t,20*log10(abs(ftc04)))
ylabel('c04')
axis([0,125e-3,-65,-25])
subplot(613)
plot(t,20*log10(abs(ftc06)))
ylabel('c06')
axis([0,125e-3,-80,-40])
subplot(614)
plot(t,20*log10(abs(ftc08)))
ylabel('c08')
axis([0,125e-3,-80,-40])
subplot(615)
plot(t,20*log10(abs(ftcl0)))
ylabel('c10')
axis([0,125e-3,-80,-40])
subplot(616)
plot(t,20*log10(abs(ftc12)))
ylabel('c12')
axis([0,125e-3,-85,-45])
xlabel('Time in sec')

figure(4)
orient tall
subplot(611)
plot(t.*1e3,20*log10(abs(ftsrc)))
title('Magnitude of Lowpass Filtered Impulse Response Function with fc=25.6kHz, dB')
ylabel('SRC')
axis([0,5,-40,0])
subplot(612)
plot(t.*le3,20*log10(abs(ftc04)))
ylabel('c04')
axis([0,5,-65,-25])
subplot(613)
plot(t.*le3,20*log10(abs(ftc06)))
ylabel('c06')
axis([0,5,-80,-40])
subplot(614)
plot(t.*le3,20*log10(abs(ftc08)))
ylabel('c08')
axis([0,5,-80,-40])
subplot(615)
plot(t.*le3,20*log10(abs(ftc10)))
ylabel('c10')
axis([0,5,-80,-40])
subplot(616)
plot(t.*le3,20*log10(abs(ftc12)))
ylabel('c12')
axis([0,5,-90,-50])
xlabel('Time in msec')

return

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Gaussian Bandpass Filtered Output Signal

INPUT Trigger freq: 1Hz Trigger: every frame
% Freq span: 22kHz Level: 10%
% spectral line: 3201 Slope: positive
% Max freq: 25600Hz
% Frame size: 8192 Pretrigger
% delta t: 1.52588e-5 samples: 100
% Frame length: 0.125sec time: 0.00152588
% Sampling freq: 65536.0Hz percentage: 1.2207
Fs=65536;
F1=8192;
t=[0:16383]/Fs;
f=[0:16383]/(F1+2)*Fs; 
%[data,q1,q2,q3,q4,q5,q6,q7]=Readuniversal('Bpulse.unv');
load /arcticl/jhpark/truss/Oct3O/Bpulse
src=real(data(:,24).');
c04=real(data(:,7).');
c06=real(data(:,11).');
c08=real(data(:,15).');
c10=real(data(:,19).');
c12=real(data(:,23).');
src=src - mean(src);
c04=c04 - mean(c04);
c06=c06 - mean(c06);
c08=c08 - mean(c08);
c10=c10 - mean(c10);
c12=c12 - mean(c12);

FFT
fsrc=fft(src,F1*2);
f04=fft(c04,F1*2);
f06=fft(c06,F1*2);
f08=fft(c08,F1*2);
f010=fft(c10,F1*2);
f012=fft(c12,F1*2);
%clear src c04 c06 c08 c10 c12

transfer function
H(1,:)=fsrc./fsrc; % src
H(2,:)=f04./fsrc; % c04
H(3,:)=f06./fsrc; % c06
H(4,:)=f08./fsrc; % c08
H(5,:)=f010./fsrc; % c10
H(6,:)=f012./fsrc; % c12
%clear fsrc f04 f06 f08 f010 f012

for i=1:6,
H(i,1:8193:16384)=zeros(1,8192);
end

Gaussian Filter G(w)=exp(-(f0-f)^2/(2*fbw^2));
The mean and std parameters used to create the filter
such that the 6dB down points are located at the band edges.
Bandwidth is approximately equivalent to that of an octave band

fwb=sqrt((f0-f1)^2/(2*6.906e-1))
f0=16e3;
for i=1:8,
f0(i)=f0/2^(i-1);
f1(i)=fo(i)/sqrt(2);
fwb(i)=sqrt((f0(i)-f1(i))^2/(2*6.906e-1));
end

index for fo
%idx=floor(F1*f0(i)/f(F1))+1;
% end
for i=1:8,
for j=1:F1,
gaussfilter(i,j)=exp(-(fo(i)-f(j))^2/(2*fbw(i)^2));
gaussfilter(i,F1*2-(j-1))=gaussfilter(i,j);
end
end

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%
TIME vs LOG Impulse response with a specific Gaussian BP Filter
%%%%%%

gaussfilter(1:8,:)=[1 6 4 2 1 0.5 0.25 0.125] kHz

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%
fo=16kHz
for i=1:6,
    h(i,:)=ifft(H(i,:).*gaussfilter(1,:));
end
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(1)
orient tall
subplot(611)
plot(t,20*log10(h(1,:)))
title('Magnitude of Gaussian Octave Bandpass Filtered Impulse Response Function with fo=16kHz, dB')
ylabel('SRC')
axis([0 125e-3 -60 -10])
subplot(612)
plot(t,20*log10(h(2,:)))
ylabel('c04')
axis([0 125e-3 -83 -43])
subplot(613)
plot(t,20*log10(h(3,:)))
ylabel('c06')
axis([0 125e-3 -97 -57])
subplot(614)
plot(t,20*log10(h(4,:)))
ylabel('c08')
axis([0 125e-3 -97 -57])
subplot(615)
plot(t,20*log10(h(5,:)))
ylabel('c10')
axis([0 125e-3 -110 -70])
subplot(616)
plot(t,20*log10(h(6,:)))
ylabel('c12')
xlabel('Time in sec')
axis([0 125e-3 -105 -65])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%
figure(2)
orient tall
subplot(611)
plot(t.*1e3,20*log10(h(1,:)))
title('Magnitude of Gaussian Octave Bandpass Filtered Impulse Response Function with fo=16kHz, dB')
ylabel('SRC')
axis([0 5 -60 -10])
subplot(612)
plot(t.*1e3,20*log10(h(2,:)))
ylabel('c04')
axis([0 5 -83 -43])
subplot(613)
plot(t.*1e3,20*log10(h(3,:)))
ylabel('c06')
axis([0 5 -97 -57])
subplot(614)
plot(t.*1e3,20*log10(h(4,:)))
ylabel('c08')
axis([0 5 -97 -57])
subplot(615)
plot(t.*1e3,20*log10(h(5,:)))
ylabel('c10')
axis([0 5 -110 -70])
subplot(616)
plot(t.*le3,20*log10(h(6,:)))
ylabel('c12')
xlabel('Time in msec')
axis([0 5 105 -65])

Plot the Estimated Arrival Time of Elastic Waves
with Center Frequency of fo=16kHz

dT=.25/16384
arvT4=[.18 .26 .44 .52 .72 .80].*1e-3
idx4=ceil(arvT4./dT+ones(length(arvT4),1))
T4=t(idx4).*le3
H4=20*log10(h(2,idx4))

arvT6=[.39].*1e-3
idx6=ceil(arvT6./dT+ones(length(arvT6),1))
T6=t(idx6).*le3
H6=20*log10(h(3,idx6))

arvT8=[.68 .79 .92].*1e-3
idx8=floor(arvT8./dT+ones(length(arvT8),1))
T8=t(idx8).*le3
H8=20*log10(h(4,idx8))

arvT10=[.95 1.01 1.08 1.33 1.44].*1e-3
idx10=floor(arvT10./dT+ones(length(arvT10),1))
T10=t(idx10).*le3
H10=20*log10(h(5,idx10))

arvT12=[1.16 1.36 1.47 1.54 1.76 1.86].*1e-3
idx12=floor(arvT12./dT+ones(length(arvT12),1))
T12=t(idx12).*le3
H12=20*log10(h(6,idx12))

figure(2)
subplot(612)
hold
plot(T4,H4,'o')
hold
subplot(613)
hold
plot(T6,H6,'o')
hold
subplot(614)
hold
plot(T8,H8,'o')
hold
subplot(615)
hold
plot(T10,H10,'o')
hold
subplot(616)
hold
plot(T12,H12,'o')
hold

Find Index for Decay Constant Calculation
run after gaus_arr.m

gaussfilter(1:8,:)=[16 8 4 2 1 0.5 0.25 0.125] kHz

for i=2:6,
    h(i,:)=ifft(H(i,:).*gaussfilter(1,:));
end
hC04=20*log10(abs(h(2,:)));
hCO6 = 20*log10(abs(h(3,:)));  
hCO8 = 20*log10(abs(h(4,:)));  
hC10 = 20*log10(abs(h(5,:)));  
hC12 = 20*log10(abs(h(6,:)));  
maxCO4 = max(hCO4);  
I04 = find(hCO4 > maxCO4 - 1);  
hCO4(I04)  
t(I04)  
maxCO6 = max(hCO6);  
I06 = find(hCO6 > maxCO6 - 1);  
hCO6(I06)  
t(I06)  
maxCO8 = max(hCO8);  
I08 = find(hCO8 > maxCO8 - 1);  
hCO8(I08)  
t(I08)  
maxC10 = max(hC10);  
I10 = find(hC10 > maxC10 - 1);  
hC10(I10)  
t(I10)  
maxC12 = max(hC12);  
I12 = find(hC12 > maxC12 - 1);  
hC12(I12)  
t(I12)  
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
return

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

%%%%  
%%%%  
Find & Plot Double Decay Rate

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

XXXX Polynomial Curve Fitting (LINEAR SCHEME)

run after gaus_arr.m

gaussfilter(1:8,:)=[16 8 4 2 1 0.5 0.25 0.125] kHz

Energy Increase Rate is difficult (impossible) to find least square curve  
% since the peak of energy arrival is rounded.  
% Can't pick the highest point for linear scheme

Fs=65536;  
F1=8192;  
fc=16kHz

for i=2:6,  
h(i,:)=ifft(H(i,:).*gaussfilter(1,:));  
end

XXXXX FROM DIRECT OBSERVATION

.5493 3.3 4.4 7.6 6.7 45 50 45 50 45 msec

last=7865 XXXXX 120msec

[37 219 289 497 441 2949 2949 3276 2949 3276 2949 last];

XXXXX DECAY 1

[p04d,So4d]=polyfit(t(idx(1):idx(6)),20*log10(abs(h(2,idx(1):idx(6)))),1);  
[y04d,delta04d]=polyval(p04d,t(idx(1):idx(6)),So4d);  
% C06  
p06d,So6d]=polyfit(t(idx(2):idx(7)),20*log10(abs(h(3,idx(2):idx(7)))),1);  
[y06d,delta06d]=polyval(p06d,t(idx(2):idx(7)),So6d);  
% C08  
p08d,So8d]=polyfit(t(idx(3):idx(8)),20*log10(abs(h(4,idx(3):idx(8)))),1);  
[y08d,delta08d]=polyval(p08d,t(idx(3):idx(8)),So8d);  
% C10  
p10d,S10d]=polyfit(t(idx(4):idx(9)),20*log10(abs(h(5,idx(4):idx(9)))),1);  

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[y10d, deltal0d] = polyval(p10d, t(idx(4):idx(9)), S10d);
% C12
[y12d, deltal2d] = polyval(p12d, t(idx(5):idx(10)), S12d);
pd = [p04d; p06d; p08d; p10d; p12d];

%%%%%% DECAY 2
% C04
[p04d2, S04d2] = polyfit(t(idx(6):idx(11)), 20*log10(abs(h(2,idx(6):idx(11)))),1);
y04d2, deltalo4d2 = polyval(p04d2, t(idx(6):last), S04d2);
% C06
[p06d2, S06d2] = polyfit(t(idx(7):idx(11)), 20*log10(abs(h(3,idx(7):idx(11)))),1);
y06d2, deltalo6d2 = polyval(p06d2, t(idx(7):last), S06d2);
% C08
[p08d2, S08d2] = polyfit(t(idx(8):idx(11)), 20*log10(abs(h(4,idx(8):idx(11)))),1);
y08d2, deltalo8d2 = polyval(p08d2, t(idx(8):last), S08d2);
% C10
[p10d2, S10d2] = polyfit(t(idx(9):idx(11)), 20*log10(abs(h(5,idx(9):idx(11)))),1);
y10d2, deltalo10d2 = polyval(p10d2, t(idx(9):last), S10d2);
% C12
[p12d2, S12d2] = polyfit(t(idx(10):idx(11)), 20*log10(abs(h(6,idx(10):idx(11)))),1);
y12d2, deltalo12d2 = polyval(p12d2, t(idx(10):last), S12d2);
pd2 = [p04d2; p06d2; p08d2; p10d2; p12d2];

%%%% slope (dB/10msec)
lgd='fc=16kHz'
DecayRate1 = round(pd(:,1)/10)/10
DecayRate2 = round(pd2(:,1)/10)/10

% XXXX Let’s make smooth transition from decay 1 to decay 2
y04c = y04d(length(y04d)) - y04d2(1);
y04d2=y04d+y04c;
y06c = y06d(length(y06d)) - y06d2(1);
y06d2=y06d+y06c;
y08c = y08d(length(y08d)) - y08d2(1);
y08d2=y08d+y08c;
y10c = y10d(length(y10d)) - y10d2(1);
y10d2=y10d+y10c;
y12c = y12d(length(y12d)) - y12d2(1);
y12d2=y12d+y12c;

% XXXX Plot regression data
figure(1)
orient tall
subplot(521)
plot(t,20*log10(h(2,:)))
xx = fc=16kHz
ylabel('h(c04) dB')
subplot(523)
plot(t,20*log10(h(3,:)))
ylabel('h(c06) dB')
subplot(525)
plot(t,20*log10(h(4,:)))
ylabel('h(c08) dB')
subplot(527)
plot(t,20*log10(h(5,:)))
ylabel('h(c10) dB')
subplot(529)
plot(t,20*log10(h(6,:)))
ylabel('h(c12) dB')
xlabel('Time in sec')

% XXXX Plot regression data
subplot(522)
plot(t(idx(1):idx(6)), y04d, t(idx(6):idx(11)), y04d2)
subplot(524)
plot(t(idx(2):idx(7)), y06d, t(idx(7):idx(11)), y06d2)
subplot(526)
plot(t(idx(3):idx(8)), y08d, t(idx(8):idx(11)), y08d2)
subplot(528)
plot(t(idx(4):idx(9)), y10d, t(idx(9):idx(11)), y10d2)
subplot(5,2,10)
plot(t(idx(5):idx(10)),y12d,t(idx(10):idx(11)), y12d2)
xlabel('Time in sec')

AXIS SCALING
figure(1)
for i=0:1
subplot(5,2,1+i)
axis([0 120e-3 -120 -30])
subplot(5,2,3+i)
axis([0 120e-3 -120 -30])
subplot(5,2,5+i)
axis([0 120e-3 -120 -30])
subplot(5,2,7+i)
axis([0 120e-3 -120 -30])
subplot(5,2,9+i)
axis([0 120e-3 -120 -30])
end

%%% Figure Ends %%%
%

---

% Early Arrival Trend of Total Elastic Wave Field
%
% Corresponds to joints
%
X = c04 c06 c08 c10 c12
1 = [.545 1.125 2.090 3.061 4.031];
x1 = [0 4.7/6 4.7/6+2 4.7/6+3 4.7/6+4];

f16 = [.18 .39 .68 .945 1.16];
.02 NaN .79 1.013 1.363; .34 NaN .92 1.08 1.47; .44 NaN NaN 1.22 1.543; .52 NaN NaN 1.33 1.64; .72 NaN NaN 1.44 1.76; .80 NaN NaN NaN 1.86;
f8 = [.18 .43 .68 .981 .81];
.51 NaN .917 1.237 1.248; .93 NaN 1.16 1.43 1.538; NaN NaN NaN 1.60 1.72; NaN NaN NaN NaN 1.83; NaN NaN NaN NaN 2.16;
f4 = [.34 .60 .51 1.70 1.125; .42 NaN 1.04 1.76 1.685; .44 NaN 1.18 2.06 2.01; .72 NaN 1.51 NaN 2.79; .84 NaN NaN NaN NaN];
fx16 = f16;
fx8 = f8;
fx4 = f4;
X = linspace(0,4.5);
[p,S]=polyfit([0 1],[0 f16(1,:)],[1]);
[Y16,delta]=polyval([p(1) 0],X,S);

f16_speed=1000/p(1)
[p,S]=polyfit([0 1:4],[0 f8(1,:)],[1]);
[Y8,delta]=polyval([p(1) 0],X,S);

f8_speed=1000/p(1)
[p,S]=polyfit([0 1(1:4)],[0 f4(1,:)],[1]);
[Y4,delta]=polyval([p(1) 0],X,S);

f4_speed=1000/p(1)
AX = linspace(0,3.5);
[p,S]=polyfit(lx,fx16(1,:),1);
[Y16,delta]=polyval([p(1) fx16(1,:)],AX,S);

f16_Aspeed=1000/p(1)
[p,S]=polyfit(lx,fx8(1,:),1);
[Y8,delta]=polyval([p(1) fx8(1,:)],AX,S);

f8_Aspeed=1000/p(1)
[p,S]=polyfit(lx,fx4(1,:),1);

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f8_Aspeed=1000/p(1)
[p,S]=polyfit([lx(1) lx(2) lx(3) lx(5)],fx4(1,1) fx4(1,2) fx4(1,3) fx4(1,5)],1);
[p,S]=polyfit(lx,fx4(1,:),1);
[AY4,delta]=polyval([p(1) fx4(1,1)],AX,S);
f4_Aspeed=1000/p(1)

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(1)
clg
orient portrait
axes('position',[.1 .1 .8/3 .8])
plot(1,f16(1,:),o',1,f16(2,:),'+',1,f16(3,:),',x',1,f16(4,:),'*', ...
 1,f16(5,:),'+',1,f16(6,:),',x',1,f16(7,:),',*'
legend('1st arriv','2nd','3rd','4th',...
'rem'. '5th', '6th', '7th')
title('fo=16kHz')
xlabel('Shortest Path in m')
ylabel('Time in msec')
axis([0 4.5 0 3])

cgl
axes('position',[.1+.8/3*1 .1 .8/3 .8])
plot(1,f8(1,:),',',1,f8(2,:),'+',1,f8(3,:),',x',1,f8(4,:),',*' ...
 1,f8(5,:),',+',1,f8(6,:),',x',1,f8(5,:),',o',1,f8(6,:),',o',X,Y8,:)')
set(gca,'YTickLabels','')
title('fo=8kHz')
xlabel('Shortest Path in m')
ylabel('Time in msec')
axis([0 4.5 0 3])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(2)
clg
orient portrait
axes('position',[.1 .1 .8/3 .8])
plot(1,fx16(1,:),',',1,fx16(2,:),'+',1,fx16(3,:),',x',1,fx16(4,:),',*' ...
 1,fx16(5,:),',+',1,fx16(6,:),',x',1,fx16(7,:),',*'
legend('1st arriv','2nd','3rd','4th',...
'rem'. '5th', '6th', '7th')
title('fo=16kHz')
xlabel('Axial Distance in m')
ylabel('Time in msec')
axis([-5.5 3.5 0 3])

cgl
axes('position',[.1+.8/3*1 .1 .8/3 .8])
plot(1,fx8(1,:),',',1,fx8(2,:),'+',1,fx8(3,:),',x',1,fx8(4,:),',*' ...
 1,fx8(5,:),',+',1,fx8(6,:),',x',1,fx8(5,:),',o',1,fx8(6,:),',o',AX,AY8,:)')
set(gca,'YTickLabels','')
title('fo=8kHz')
xlabel('Axial Distance in m')
ylabel('Time in msec')
axis([-5.5 3.5 0 3])

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
figure(3)
df16=[8.4 6.5 5.9 4.6 4.0];
df8=[6.8 5.1 5.0 3.7 5.2];
df4=[4.4 3.8 3.2 3.1 3.4];
df2f16=[0.8 0.2 0.0 0.4 0.0];
d2f8 = [0.6 0.4 0.1 0.9 0.1];
d2f4 = [2.0 0.8 1.9 0.7 0.7];
orient tall
subplot(211)
plot(l,d2f16,'--',1,d2f8,'--',1,d2f4,'-.',1,d2f16,'o',1,d2f8,'o',1,d2f4,'o')
title('Eta_0 Decay Rate')
xlabel('Shortest Path in m')
ylabel('dB/10msec')
axis([0 4.5 0 10])
set(gca,'YTick',[0:1:10])
legend('fo=16kHz','fo=8kHz','fo=4kHz')
grid
subplot(212)
plot(l,d2f16,'o',l,d2f8,'o',l,d2f4,'o')
title('Eta_1 Decay Rate')
xlabel('Shortest Path in m')
ylabel('dB/10msec')
axis([0 4.5 -5 5])
set(gca,'YTick',[-5:1:5])
grid
return

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Appendix A

Dispersion Relation of Timoshenko Beam

\[ \rho = 2700; \text{ % density of A16061} \]
\[ E = 70e9; \text{ % Young's modulus} \]
\[ G = 26e9; \text{ % Shear modulus} \]
\[ K = 3.950e-3; \text{ % radius of gyration} \]
\[ f = [0:10:25e3]; \]
\[ w = f .* (2 * \pi); \]
\[ C_l = \sqrt{E / \rho}; \text{ % longitudinal wave speed} \]
\[ C_T = \sqrt{G / \rho}; \text{ % torsional wave speed} \]

Timoshenko Beam
\[ tc = 0.545; \text{ % Timoshenko shear coeff. for thin-walled pipe} \]
\[ A = (\rho / E + \rho / (tc * G)); \]
\[ B = (\rho / (E * K^2)); \]
\[ k_T = \sqrt{w^2 * 2 * A + \sqrt{w^4 * (A^2 - 4 * C) + w^2 * B * 4} * 2}; \]
\[ \text{C phase(Timoshenko Beam)} \]
\[ \text{C group(Timoshenko Beam)} \]

Euler Bernoulli Beam
\[ k_B = \sqrt{w} * ((E * K^2 - 2 / \rho)^{1/4}); \]
\[ C B_p = \sqrt{w} * (E * K^2 - 2 / \rho)^{1/4}; \]
\[ C B_g = \sqrt{w} * 2 * (E * K^2 - 2 / \rho)^{1/4}; \]

figure(1)
\[ \text{plot}(f / 1e3, C_{T_p} , '-', f / 1e3, C_{B_p} , '-', f / 1e3, C_{T_g} , '--', f / 1e3, C_{B_g} , ':') \]
\[ \text{title('Wave Speed of Timoshenko and Euler-Bernoulli Beam')} \]
\[ \text{xlabel('Frequency in kHz')} \]
\[ \text{ylabel('Wave Speed (m/s)')} \]

figure(2)
\[ \text{plot}(f / 1e3, k_T , '-', f / 1e3, k_B , '--') \]
\[ \text{title('Dispersion Relation of Timoshenko and Euler-Bernoulli Beam')} \]
\[ \text{xlabel('Frequency in kHz')} \]
\[ \text{ylabel('Wave Number (1/m)')} \]

f = [125, 250, 500, 1e3, 2e3, 4e3, 8e3, 16e3];
\[ w_1 = f .* 2 * \pi; \]
\[ C_{B_p_1} = \sqrt{w_1 .* (E * K^2 - 2 / \rho)}^{(1/4)}; \]
\[ C_{B_g_1} = \sqrt{w_1 .* 2 .* (E * K^2 - 2 / \rho)}^{(1/4)}; \]
\[ k_{T_1} = \sqrt{w_1 .* 2 * A + \sqrt{w_1 .* 4 * (A^2 - 4 * C) + w_1 .* 2 * B} .* 2}; \]
\[ C_{T_1} = w_1 .* \pi; \]
\[ C_{T_g_1} = (\sqrt{w_1 .* 2 * A + \sqrt{w_1 .* 4 * (A^2 - 4 * C) + w_1 .* 2 * B} .* 2}) / \ldots \]
\[ \sqrt{w_1 .* 4 * (A^2 - 4 * C) + w_1 .* 2 * B} .* 2); \]
\[ C_1 \]
\[ C_T \]
\[ \text{idr}=['C_{B_p_1}', 'C_{B_g_1}', 'C_{T_p_1}', 'C_{T_g_1}'] \]
\[ \text{result}=[f'; C_{B_p_1}' C_{B_g_1}' C_{T_p_1}' C_{T_g_1}'] \]
\[ \text{return} \]

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Bibliography


[19] Private communication with Professor Ira Dyer at MIT, 1996.


