

## 2.081J/16.230J Plates and Shells

Quiz

Monday, March 24

**Problem 1** [Must be completed in the class.]

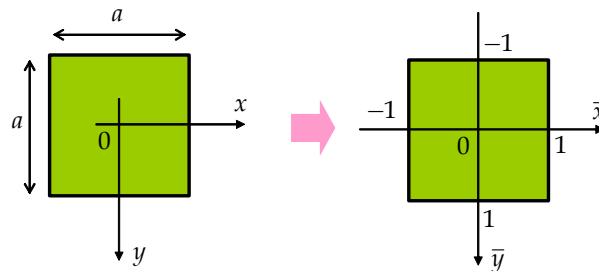
A simply supported square elastic plate ( $a \times a$ ) is loaded by a transverse load  $q(x, y)$ . The origin of the coordinate system is at the center of the plate. The shape of the deflected plate was precisely measured and then fit by the following function:

$$w(\bar{x}, \bar{y}) = w_0 (1 - \bar{x}^2) (1 - \bar{y}^2)$$

where

$$\bar{x} = \frac{x}{a/2} \quad ; \quad \bar{y} = \frac{y}{a/2}$$

and  $w_0$  is the central deflection of the plate  $w_0 = w(0, 0)$ .



- Check if the kinematic boundary conditions are satisfied.
- Calculate the components of the bending moment tensor  $M_{\alpha\beta}$  and effective shear force vector  $V_\alpha$ .
- Check if the static boundary conditions are satisfied (both moments and effective shears).

- (d) Determine the magnitude and the distribution of the transverse loading  $q(\bar{x}, \bar{y})$  that equilibrates the plate.
- (e) Is the plate resistance derived from the direct bending moments  $M_{xx}$  and  $M_{yy}$  or from the twisting moment  $M_{xy}$ ?
- (f) Calculate the relationship between  $w_0$  and  $q$  and compare it with the exact solution to be found in the note (see Section 4.1.5). Discuss the reason for a large error.

**Problem 2** [Take-home problem due to Monday 27, March 1:00pm.]

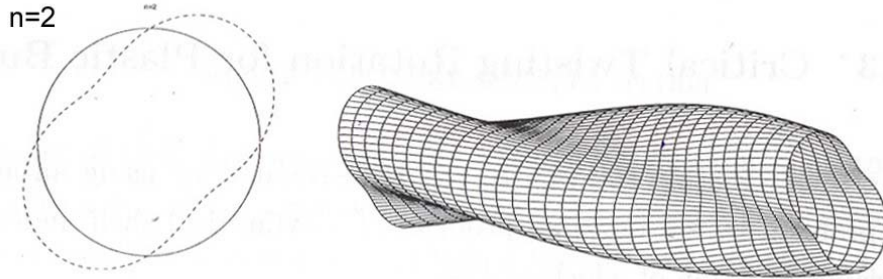
A long elastic cylindrical shell made of mild steel ( $E = 210GPa$ ,  $\nu = 0.3$ ) is subjected to a twisting moment  $M$  producing a uniform field of in-plane shear membrane force  $N_{x\theta}^\circ$ :

$$N_{\alpha\beta}^\circ = \begin{vmatrix} 0 & N_{x\theta}^\circ \\ N_{x\theta}^\circ & 0 \end{vmatrix}$$

The buckling mode consists of a number of circumferential waves that spiral around the cylinder from one end to the other according to:

$$w = \sin\left(\frac{\bar{m}x}{a} - n\theta\right)$$

where, as in the notes<sup>1</sup>,  $\bar{m} = m\pi a/l$ . It was shown in the notes that the smallest buckling force is obtained from  $n = 2$ . The resulting buckling mode is shown in the figure below.



- (a) Derive an expression for the critical twisting moment  $M_{cr}$  causing buckling.

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<sup>1</sup>Here, the following notations are used:

$a$  : radius,  $h$  : thickness,  $l$  : length

- (b) Keeping the radius  $a$  constant, how much should the thickness of the shell increase to double the magnitude of the twisting moment?
- (c) Keeping the thickness of the shell constant, how much should the radius of the shell increase to double the magnitude of the twisting moment?
- (d) Assuming  $a/h = 100$ , what should be the length-to-radius ratio  $l/a$  so that half of the spiral is formed ( $m = 1$ ), as shown in the figure above?

**Problem 3 [EXTRA CREDIT]** [*Take-home problem due to Monday 27, March 1:00pm.*]

Derive an expression for the load-carrying capacity of a clamped, circular, rigid-plastic plate under uniform transverse load using Tresca yield condition.