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Sustainable Drag Reduction in Turbulent Taylor-Couette Flows by Depositing Sprayable Superhydrophobic Surfaces

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We demonstrate a reduction in the measured inner wall shear stress in moderately turbulent Taylor-Couette flows by depositing sprayable superhydrophobic microstructures on the inner rotor surface. The magnitude of reduction becomes progressively larger as the Reynolds number increases up to a value of 22% at Re = 8.0 × 10^4. We show that the mean skin friction coefficient C_f in the presence of the superhydrophobic coating can be fitted to a modified Prandtl–von Kármán–type relationship of the form (C_f/2)^1/2 = M ln (Re(C_f/2)^1/2) + N + (b/Δx)Re(C_f/2)^1/2 from which we extract an effective slip length of b ≈ 19 μm. The dimensionless effective slip length b^+ = b/δ_v, where δ_v is the viscous length scale, is the key parameter that governs the drag reduction and is shown to scale as b^+ ≈ Re^1/2 in the limit of high Re.

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It is well known that superhydrophobic (SH) surfaces can reduce drag in laminar flows by presenting an effective wall slip boundary condition due to stable pockets of vapor within the asperities of the textured substrate [1,2]. The trapped vapor layer adjacent to the solid wall lubricates the fluid flow by introducing an effective slip boundary condition along portions of the wall, and consequently reduces the overall drag [3]. The magnitude of the effective slip length b in viscous laminar flows is governed by the surface feature length scale and the wetted solid fraction [4–6]. Measurements in various microchannel flows [7–11] yield values of b that are typically in the range of 10–30 μm.

There is less consensus on the extent to which microscopic effective slip can influence macroscopic skin friction in turbulent flows [12,13]. Numerical simulations of turbulent channel flows indicate that the shear-free liquid-vapor interface can reduce skin friction by introducing an effective slip velocity in the viscous sublayer [14,15], and by the suppression of turbulent flow structures in the near-wall region [16]. While recent experimental studies report varying amounts of drag reduction in turbulent flows using SH surfaces [17–19], there are inconsistencies in the magnitude of observed drag reduction across studies, and its dependence on the slip length, surface characteristics, and Reynolds number in turbulent flow remain unclear.

In this Letter, we demonstrate sustained reduction in frictional drag in turbulent Taylor-Couette (TC) flows by applying a polymeric SH coating to the inner rotor. The extent of drag reduction DR = 100 × (T_{flat} – T_{SH})/T_{flat} based on the inner rotor torque T, steadily increases with Re up to 22% at Re = 80000. The reduction in friction arises from finite slip effects at the moving rotor. The two key results we describe in our Letter are (i) the magnitude of drag reduction is directly related to a dimensionless slip length b^+ ≡ b/δ_v, which couples the effective slip length b to the viscous length scale δ_v = ν/τ_i, where τ_i is the wall shear stress measured at the inner rotor and u_* = √τ_i/ρ is the friction velocity), and (ii) the proposal of a modified skin friction law from which a single value of b is obtained that accounts for the entire range of measured drag reduction for various Re.

The large gap TC apparatus [see Fig. 1(a)] has a rotating recessed end inner cylinder of radius r_i = 14 ± 0.01 mm and stationary outer cylinder of radius r_o = 34.3 ± 0.1 mm (so that η = r_i/r_o = 0.41; Δr = r_o – r_i = 20.3 mm). The recessed end traps a pocket of air inside the cavity [see Fig. 1(a)] to avoid end effects in torque measurements. The
height of the inner rotor was \( L = 42.0 \pm 0.1 \) mm, having an aspect ratio of \( \zeta = L/(r_o - r_i) = 2.1 \). The gap between the cylinders is filled with deionized water at \( 25^\circ C \). The maximum rotation rate of the inner rotor was \( \Omega_{max} = 250 \) rad/s, which represents a maximum attainable Reynolds number of \( Re_{max} = r_i \Omega_{max} \Delta r/\nu = 8 \times 10^4 \). This allows us to generate shear driven turbulent TC flows (past the Taylor vortex regime) \([20,21]\), the onset of which occurs at a transition Reynolds number of \( Re_c \approx 1 \times 10^4 \) for our system (for details, see the Supplemental Material \([22]\)). We fabricated our SH surface on the inner rotor by spray coating a 50/50 (by weight) solution of poly(methyl methacrylate) (PMMA)/fluorodecyl polyhedral oligomeric silsesquioxane (FPOSS) at a concentration of \( c = 50 \) mg/ml, as described previously \([27,28]\). The coating was separately applied to produce either an unconnected configuration \([\text{Fig. } 1(b)]\) where the trapped air (or “plastron”) is isolated, or a connected configuration \([\text{Fig. } 1(c)]\) in which the air layer is in continuous contact with an external air reservoir.

To quantify the initial wetted solid fraction, we extend the approach of Ref. \([29]\) to enable \textit{in situ} imaging of the plastron on a random textured SH surface using confocal microscopy, as detailed in the Supplemental Material \([22]\). In \textit{Fig. 2(a)}, we show the reconstructed 3D isosurface of the composite solid-liquid-air interface when a water drop (fluorescently labeled green), was deposited onto the SH coating (fluorescently labeled red). As expected, we observed that the liquid meniscus wets only on a fraction of the PMMA/FPOSS structures. The cut-away region reveals the unwetted (dry) PMMA/FPOSS structures beneath the liquid. \textit{Figure 2(b)} shows an orthoslice of the spray coated texture and the liquid meniscus resting in the nonwetting state \([30]\) along the \( AA' \) direction in \textit{Fig. 2(a)}. We determined the local height profile \( h(x, y) \) of the PMMA/FPOSS and the vertical position of the liquid meniscus \( z_0(x, y) \) from a series of these slices. Computing the averaged height-height correlation function \([31]\) \( H(r) = \langle (h(x + r) - h(x'))^2 \rangle_{x,y} \) allows us to estimate a lateral correlation length, which corresponds to the size of the sprayed features, as \( \xi = 15 \) \( \mu m \) and a root-mean-square roughness \( \omega = 8.5 \) \( \mu m \).

In order to investigate the influence of the SH coating in turbulent TC flows, we systematically varied rotation rates to measure the global torque \( T \) on coated and uncoated rotors for \( N = 17 \) values of \( Re \in [1600, 80000] \). A single run consisted of either a ramp up in discrete steps from a minimum rate of \( \Omega = 5 \) rad/s \((Re = 1600)\) or a ramp down from the maximum \( \Omega = 250 \) rad/s \((Re = 80000)\). The torque \( T \) is measured over a period of 600 s at each Re. Viscous heating effects are negligible over this short time period (see Supplemental Material \([22]\)). At least four replicates were performed for both the connected and unconnected state [see \textit{Fig. 1(b)}]; the torque was averaged across all runs for each Re. In \textit{Fig. 3}, we plot the averaged dimensionless torque \( G = T/(\rho \nu^2 L) \) for the various Re. For the uncoated inner rotor, where the relative standard deviation in the measured torque is \( \sigma_r = 0.01 \), we fit the measured torque to the predicted scaling \( G \sim g(\eta)Re^{5/3} \) for a turbulent TC flow in the wide-gap limit \([32]\), as indicated by the dotted line in \textit{Fig. 3}. While our best fit is in good agreement with the predicted exponent of \( 5/3 \), the value of \( g(\eta) = 0.3 \) that we obtained is smaller than the predicted value of \( g(\eta) = [(1+\eta)/2]/[\eta^{1/3}(1-\eta)^2] \) from Ref. \([32]\), suggesting that a very large gap width \( \eta < 0.5 \) and small aspect ratio \( \zeta = 2.1 \) influences the prefactor but not the torque scaling.

![FIG. 2 (color online). (a) Reconstructed 3D isosurface from confocal microscopy of water (green) resting on PMMA/FPOSS structures (red) in the nonwetting state. (b) Superimposed fluorescence images of the PMMA/FPOSS structures (red) and the liquid meniscus (green) along the AA' slice plane.](image)

![FIG. 3 (color online). Variation of the average dimensionless torque G against Re. The dotted black line represents the fit to the scaling G \~ Re^{5/3} \) from Ref. \([32]\) for the uncoated rotor. Inset: Measurements of the dimensional inner wall shear stress \( \tau \) against time for a fixed \( Re = 80000 \) over 10 minutes.](image)
Upon applying our PMMA/FPOS coating onto the inner rotor, we consistently obtained a significant reduction in the measured torque for $Re > Re_c$. Remarkably, we found that the averaged drag reduction steadily increases with $Re$ (see Fig. 3), from 6% at $Re = 19\,000$ to a maximum of 15% at $Re_{\text{max}}$ for the unconnected configuration with an isolated plastron, and up to 22% at $Re_{\text{max}}$ in the case of the connected plastron. In the inset of Fig. 3, we plot the average inner wall shear stress $\tau_i = T/(2\pi r^2 L)$ measured over a period of 600 s for a single measurement at $Re_{\text{max}}$. Initially, the rotor is at rest and is then accelerated to $\Omega = 250 \, \text{rad/s}$ at $t = 0$. The measured torque reductions due to the superhydrophobic coating are 2 orders of magnitude larger than the $\sim 0.02 \, \text{Pa}$ resolution limit of the sensor. The unconnected superhydrophobic coating shows a steady reduction of $\Delta \tau_i = 3.2 \, \text{Pa}$ over the entire duration of the measurement. The connected configuration exhibits an additional decay in the measured torque with time, caused by the augmentation of air to the connected plastron at high rotor velocities, resulting in a final reduction in torque of $\Delta \tau_i = 8.0 \, \text{Pa}$ in this configuration. Upon time averaging across multiple runs, we found larger relative deviations in the measured torque ($\sigma_r = 0.08$ at $Re_{\text{max}}$) for the SH coated rotor (see error bars in Fig. 3).

In contrast, the PMMA-only coating does not support a plastron and drops of water deposited on the surface wet in the Wenzel state. Consequently, this surface shows a drag increase ($\Delta \tau_i = -0.7 \, \text{Pa}$) as a result of the increased wetted surface area that interacts with the turbulent flow. For a roughness amplitude of $k = 35 \, \mu m$ [see Fig. 2(b)], the drag increase in the Wenzel state at the largest $Re$ corresponds to a surface with dimensionless roughness $k^+ = k/\delta_v = 6.6$. In order to relate our measured drag reduction to effective slip phenomena, we apply a boundary layer theory to relate the global shear-stress measurements to $Re$. The dimensionless wall shear stress is typically expressed using a skin friction coefficient $C_f = \tau_i/(1/2 \rho V_i^2) = 2(\tau_i/V_i)^2$. Here, $u_\tau = (\tau_i/\rho)^{1/2}$ is the friction velocity and $V_i = r_i \Omega$ is the inner rotor velocity for our TC system. In most shear driven turbulent flows, $C_f$ is described either by a power law [32] or a semiempirical logarithmic law [20,21] expressed in Prandtl–von Kármán coordinates as

$$\sqrt{\frac{2}{C_f}} = M \ln \left( \text{Re} \sqrt{\frac{C_f}{2}} \right) + N,$$  \hspace{1cm} (1)

where $M$ and $N$ are geometry-dependent constants that are analogous to the “universal” von Kármán constant $\kappa$ and the additive constant $B$ obtained for a perfect flat plate boundary layer [21]. The product $\text{Re}(C_f/2)^{1/2}$ is identical to the friction Reynolds number $Re_\tau \equiv u_\tau \Delta r/\nu$ which varies from $Re_\tau \sim 480–3810$ in our system. Although the relative validity of the power-law theory compared to the logarithmic theory is still under dispute, empirical fits to the two forms are often indistinguishable [33]. In Fig. 4, we plot the measured skin friction coefficients for the flat, uncoated inner rotor as filled circles in Prandtl–von Kármán coordinates [34]. The solid line represents a linear regression of Eq. (1) applied to data points in the shear driven turbulent flow regime, (i.e., $Re > Re_c$), yielding values of $M = 3.9$ and $N = -11.7$. Deviations to the logarithmic skin friction law given by Eq. (1) are observable only when $Re < Re_c$ and the flow is still dominated by Taylor vortices [20,21]. To determine how Navier wall slip alters the skin-friction law given by Eq. (1) in our wall-bounded turbulent TC flow with a SH surface, we apply the angular momentum defect theory of Panton [35] and incorporate finite wall slip at the inner surface. Direct measurements of angular momentum profiles [36,37] as well as direct numerical simulations of turbulent TC flows [38] verify the existence of a core region with a weakly varying angular momentum dependence, as well as thin wall layers near the inner and outer cylinders that are characterized by a sharp decay in the angular momentum. Panton [35,39] asymptotically matches the expressions for the mean azimuthal momentum in the core region and the inner wall region to derive a logarithmic skin friction law of the form expressed in Eq. (1).

We modify this theory by introducing a finite averaged slip velocity $\langle V_{\text{slip}} \rangle$ that we relate to the local viscous stress at the inner rotor by the Navier slip hypothesis as $\langle V_{\text{slip}} \rangle = b(d(V_i)/dy)_{y=0}$, where $b$ is the effective slip length due to the superhydrophobic coating and $d(V_i)/dy_{y=0}$ is the time-averaged velocity gradient at the wall. In the immediate vicinity of the inner rotor, the slip velocities and angular momentum can be rescaled using inner variables defined in terms of the friction velocity $u_\tau$ and viscous length scale $\delta_v$. The distance away from the
inner rotor expressed in wall units is $y^+ = (r - r_i)/\delta_w$. If the velocity in the viscous sublayer close to the inner wall is shifted by a constant value according to Ref. [14], so that $\langle V_\theta \rangle = y^+ + b^+$ and $d\langle V_\phi \rangle/dy^+ = 1$, the Navier-slip hypothesis upon scaling reduces to $V_{\text{slip}}^+ = b^+$, where $V_{\text{slip}} = V_{\text{slip}}/u_t$ and $b^+ = b/\delta_w$. Following Ref. [35], we derive a skin-friction law that incorporates wall slip:

$$\frac{V_i}{u_t} = M \ln \left( Re + N + b^+ \right), \quad (2)$$

where $M$ and $N$ are the same geometry-dependent constants as in Eq. (1). Details of the derivation are provided in the Supplemental Material [22]. An additional simplification is introduced by assuming that the dimensional effective slip length $b$ of the textured superhydrophobic surface is independent of the Reynolds number, so that, upon scaling

$$b^+ = \left( \frac{b}{\Delta r} \right) Re \sqrt{\frac{C_f}{2}}, \quad (3)$$

With this assumption of a constant physical slip length $b$ in Eq. (2), and introducing $C_f$ and $Re$ in place of $u_t$ and $Re_r$, we obtain a modified skin-friction law for turbulent flow that accounts for wall slip at a superhydrophobic surface and is of the form

$$\sqrt{\frac{2}{C_f}} = M \ln \left( Re \sqrt{\frac{C_f}{2}} + N + \left( \frac{b}{\Delta r} \right) Re \sqrt{\frac{C_f}{2}} \right). \quad (4)$$

In Fig. 4, the measured values of $C_f$ for the unconnected and connected SH surfaces are plotted in the Prandtl–von Kármán coordinates. The deviation of the SH data from the nonlinear regression for the SH data using Eq. (5) and with $\delta_w$ determined from measurements of the Torque $T$ at various $Re$. The dashed line is a fit to the scaling $b^+ \sim \sqrt{Re}$ for the last six data points.

![FIG. 5](color online). Scaling of the dimensionless slip length $b^+ = b/\delta_w$ that is obtained using a fixed value of $b$ from the nonlinear regression of Eq. (5) and with $\delta_w$ determined from measurements of the Torque $T$ at various $Re$. The dashed line is a fit to the scaling $b^+ \sim \sqrt{Re}$ for the last six data points.

In summary, we have demonstrated a reduction in the measured inner wall skin friction in turbulent TC flows by applying a sprayable SH surface onto the inner rotor. We observed that the extent of drag reduction increases with Reynolds number from 6% at $Re \sim 2 \times 10^4$ to a maximum of 15% and 22% for the isolated and connected plastrons, respectively, at $Re \sim 8 \times 10^4$. By applying a boundary layer theory for turbulent TC flow, we developed a modified form of the Prandtl–von Kármán skin friction law from which we extracted an effective slip length $b$ that characterizes the connected and unconnected SH coatings and is consistent with earlier friction reduction measurements in viscous laminar flows. Our measurements demonstrate that the amount of drag reduction attainable in turbulent near-wall flows is directly related to the dimensionless effective slip length $b^+ = b/\delta_w$. As $b^+$ couples the effective slip length $b$ (a material parameter) to the viscous length scale $\delta_w$ of the turbulent flow, our results establish that superhydrophobic surfaces, despite typically exhibiting effective slip lengths of only tens of micrometers, can indeed reduce the skin friction in turbulent flows when the viscous sublayer is of comparable thickness, provided that the air plastron trapped in the SH surface can be stably maintained.

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See Supplemental Material at http://link.aps.org/supplemental/10.1103/PhysRevLett.114.014501 for estimating the onset of shear driven turbulent TC flow, details of the confocal microscopy technique, the absence of viscous heating effects and the derivation of the skin friction law in the presence of slip [i.e., Eq. (2)], which includes Refs. [23–26].


