

General Planetary Motion: Cont.

$$\mu \frac{d^2 \vec{r}}{dt^2} = F_{12}$$

Kinetic Energy and Momentum

$$K = \frac{1}{2} m_1 v_1^2 + \frac{1}{2} m_2 v_2^2$$

$$v_1^2 = \left(\frac{d\vec{r}_1}{dt} \right)^2 \quad v_2^2 = \left(\frac{d\vec{r}_2}{dt} \right)^2$$

Use Eq (5) + (4) for \vec{r}_1 and \vec{r}_2

$$K = \frac{1}{2} M \left(\frac{d\vec{R}}{dt} \right) \left(\frac{d\vec{R}}{dt} \right) + \frac{1}{2} \mu \left(\frac{d\vec{r}}{dt} \right) \left(\frac{d\vec{r}}{dt} \right)$$

$$= \frac{1}{2} M v_{cm}^2 + \frac{1}{2} \mu v^2$$

\uparrow CM Vel. \uparrow Rel Vel.

= KE of CM with vel. V

KE of hypothetical particles

of mass μ and vel. v .

Total \vec{P} in frame O' :

$$\vec{P} = \vec{P}_1 + \vec{P}_2 = M \vec{V}_{cm}$$

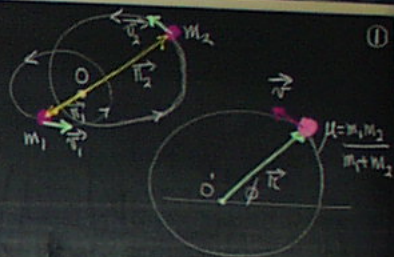
Relative linear momentum

$$\mu \vec{v} = \frac{m_1 m_2}{M} (\vec{v}_2 - \vec{v}_1)$$

$$= \frac{1}{M} (m_1 \vec{P}_2 - m_2 \vec{P}_1)$$

Isolated system $\vec{P} = \text{const.}$

Total \vec{P} in CM system $\equiv 0$.



Angular Momentum

Central Forces

No torques

$$\vec{L}_{cm} = \vec{l}_1 + \vec{l}_2 = \text{constant.}$$

$$= m_1 \vec{r}_1 \times \vec{v}_1' + m_2 \vec{r}_2 \times \vec{v}_2'$$

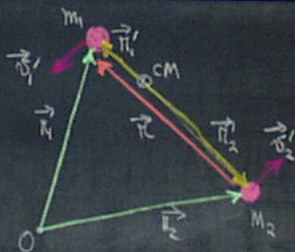
$$\vec{r} = \vec{r}_1' - \vec{r}_2'$$

$$\vec{r}_1' = \frac{m_2}{m_1 + m_2} \vec{r} \quad \vec{r}_2' = -\frac{m_1}{m_1 + m_2} \vec{r}$$

$$\vec{L}_{cm} = \frac{m_1 m_2}{m_1 + m_2} \vec{r} \times \vec{v}_1' - \frac{m_1 m_2}{m_1 + m_2} \vec{r} \times \vec{v}_2'$$

$$= \mu \vec{r} \times (\vec{v}_1' - \vec{v}_2')$$

$$= \mu \vec{r} \times \vec{v} = \mu r^2 \omega$$



Total Energy: E

Assume F : conservative

$$E = K + U$$

$$= \frac{1}{2} M V^2 + \frac{1}{2} \mu v^2 + U(r) = \text{constant}$$

For $m_1 \gg m_2$, $\mu \approx m_2$.

Motion reduces to motion of m_2 about a fixed m_1 !!!

Angular Momentum

Central Forces

No torques

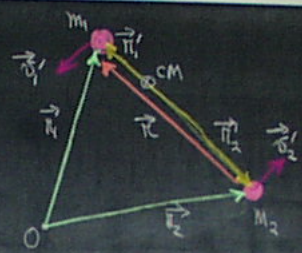
$$\vec{L}_{cm} = \vec{L}_1 + \vec{L}_2 = \text{constant.}$$

$$= m_1 \vec{r}_1' \times \vec{v}_1' + m_2 \vec{r}_2' \times \vec{v}_2'$$

$$\vec{r} = \vec{r}_1' - \vec{r}_2'$$

$$\vec{r}_1' = \frac{m_2}{m_1 + m_2} \vec{r} \quad \vec{r}_2' = -\frac{m_1}{m_1 + m_2} \vec{r}$$

$$\begin{aligned} \vec{L}_{cm} &= \frac{m_1 m_2}{m_1 + m_2} \vec{r} \times \vec{v}_1' - \frac{m_1 m_2}{m_1 m_2} \vec{r} \times \vec{v}_2' \\ &= \mu \vec{r} \times (\vec{v}_1' - \vec{v}_2') \\ &= \mu \vec{r} \times \vec{v} = \mu r^2 \omega \end{aligned}$$



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Planetary Motion - 2 Particle Systems

Sun + Planet

Planet + Moon

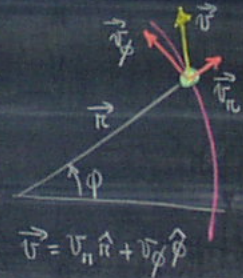
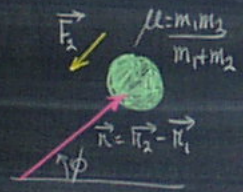
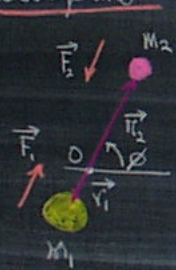
Earth + Satellites

Binary Stars

For simplicity

CM coord system.

$$\vec{R} \equiv 0; \quad \vec{V}_{cm} = 0$$



Total Energy: E

$$E = K + U = \frac{1}{2} \mu v^2 + U(r) \quad [V \equiv 0 \text{ CM System}]$$

$$\vec{v} = v_r \hat{r} + v_\phi \hat{\phi}$$

$$v_\phi = v_\phi = r\omega = r \left(\frac{d\phi}{dt} \right)$$

$$L = \mu r^2 \omega = \mu r^2 \frac{d\phi}{dt}$$

$$\frac{L}{\mu r} = r\omega = v_\phi$$

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$$E = \frac{1}{2} \mu (v_r^2 + v_\phi^2) + U(r)$$

$$= \frac{1}{2} \mu v_r^2 + \frac{L^2}{2\mu r^2} - \frac{Gm_1 m_2}{r}$$

KE due to radial motion rel. to CM.

KE due to ang. rot. about CM

Grav. PE

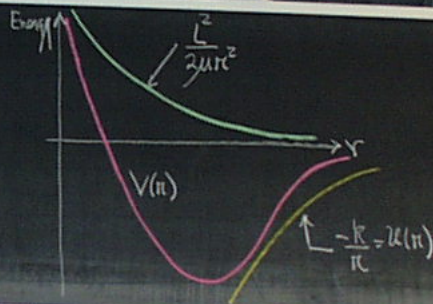
let: $V(r) = \frac{L^2}{2\mu r^2} - \frac{Gm_1 m_2}{r}$

Effective PE Function of r only.

let $k = Gm_1 m_2 = G\mu M$

$$V(r) = \frac{L^2}{2\mu r^2} - \frac{k}{r}$$

$$E = \frac{1}{2} \mu v_r^2 + V(r)$$



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Solve Equations of Motion

Want: $r = f(t)$

$\phi = g(t)$

$r = r(\phi) \Rightarrow$ trajectory.

$$v_r = \frac{dr}{dt} = \sqrt{\frac{2}{\mu} (E - V(r))}$$

$$\frac{d\phi}{dt} = \frac{L}{\mu r^2}$$

$$\frac{d\phi}{dr} = \frac{L}{\mu r^2} \frac{1}{\sqrt{\frac{2}{\mu} (E - V(r))}}$$

Solve

$$r = \frac{L^2 / \mu k}{1 - \sqrt{1 + \frac{2EL^2}{\mu k^2}} \sin(\phi - \phi_0)}$$

$\phi_0 = -\pi/2$ convention!

$$r = \frac{r_0}{1 - \epsilon \cos \phi}$$

$$\frac{r_0}{r} = 1 - \epsilon \cos \phi$$

Eq. of a Conic Section

$r_0 = \frac{L^2}{\mu k}$ = radius of circular orbit. with $\epsilon = 0$

$$\epsilon = \sqrt{1 + \frac{2EL^2}{\mu k^2}}$$

= eccentricity of ellipse.

(5)

Solve Equations of Motion

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$$\frac{dr}{dt} = \frac{L}{\mu r^2}$$

$$\therefore \frac{dr}{dt} = \frac{L}{\mu r^2} \frac{1}{\sqrt{\frac{2}{\mu} (E - V(r))}}$$

Solve

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Eq. of a Conic Section

$$r_0 = \frac{L^2}{\mu k} = \text{radius of circular orbit.} \quad (5)$$

with $\epsilon = 0$

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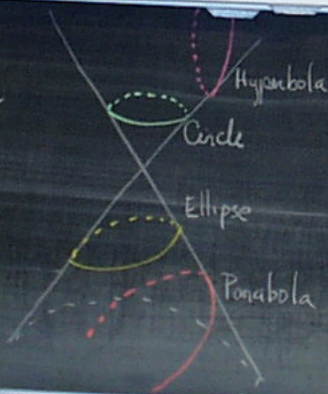
Orbit Characteristics

$E = 0$ Circular Orbits $E = E_{\min}$

$E > 1$ Hyperbolic $E > 0$

$E = 1$ Parabolic $E = 0$

$0 < E < 1$ Elliptical $E < 0$



$$E = \frac{1}{2} \mu v_r^2 + V(r)$$

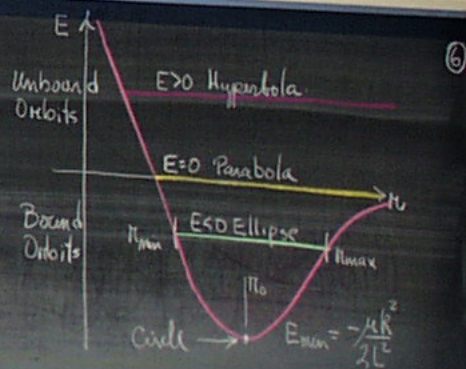
$$V(r) = \frac{L^2}{2\mu r^2} - \frac{Gm_1 m_2}{r}$$

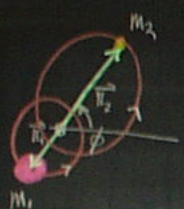
$V \rightarrow 0$ as $r \rightarrow \infty$

$$K = \frac{1}{2} \mu v_r^2 = E - V(r)$$

↑ KE for radial motion only!!

$$E = K + V(r)$$





Circular Orbits

$E = E_{min}$ when $v_a = 0$
and $\frac{dV}{dr} = 0$

$$V(r) = \frac{L^2}{2\mu r^2} - \frac{k}{r}$$

$$\frac{dV(r)}{dr} = -\frac{L^2}{\mu r^3} + \frac{k}{r^2} = 0$$

$$r_0 = \frac{L^2}{\mu k} \text{ (radius)}$$

$$E_{min} = V(r_0) = -\frac{\mu k^2}{2L^2}$$

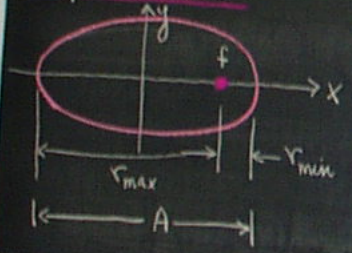
$$U(r_0) = -\frac{\mu k}{L^2}$$

$$E_{min} = \frac{1}{2} U(r_0)$$



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Elliptical Orbits



$E < 0$
 $0 < \epsilon < 1$
 $r = \frac{r_0}{1 - \epsilon \cos \phi}$
 $r_{max} = \frac{r_0}{1 - \epsilon}$ $\phi = 0$ } Turning Points
 $r_{min} = \frac{r_0}{1 + \epsilon}$ $\phi = \pi$ }
 $A = r_{min} + r_{max}$

$$A = r_0 \left[\frac{1}{1 - \epsilon} + \frac{1}{1 + \epsilon} \right] = \frac{2r_0}{1 - \epsilon^2}$$

$$= \frac{2L^2 / \mu k}{1 - \left(1 + \frac{2EL^2}{\mu k^2}\right)} = \frac{k}{(-E)} \text{ Ind. of } L$$

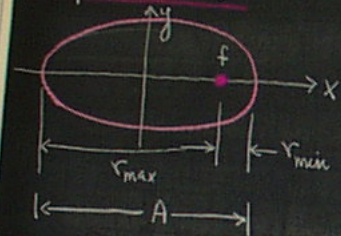
$$\frac{r_{max}}{r_{min}} = \frac{1 + \epsilon}{1 - \epsilon}$$

Same A \Rightarrow Same total E

$$E = \frac{r_{max} - r_{min}}{r_{max} + r_{min}} = \sqrt{\frac{1 + 2EL^2}{\mu k^2}} \quad (8)$$

$r_{min} = r_p$ perihelion perigee
 $r_{max} = r_a$ aphelion apogee
Planets Earth

Elliptical Orbits



$$E < 0$$

$$0 < E < 1$$

$$r = \frac{r_0}{1 - \epsilon \cos \varphi}$$

$$r_{max} = \frac{r_0}{1 - \epsilon} \quad \varphi = 0 \quad \left. \begin{array}{l} \text{Turning} \\ \text{Points} \end{array} \right\}$$

$$r_{min} = \frac{r_0}{1 + \epsilon} \quad \varphi = \pi$$

$$A = r_{min} + r_{max}$$

$$A = r_0 \left[\frac{1}{1 - \epsilon} + \frac{1}{1 + \epsilon} \right] = \frac{2r_0}{1 - \epsilon^2}$$

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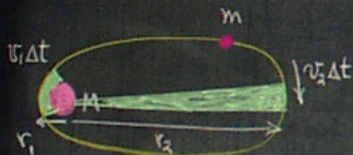
$$\frac{r_{max}}{r_{min}} = \frac{1 + \epsilon}{1 - \epsilon}$$

Same \$A \Rightarrow\$ Same total \$E\$

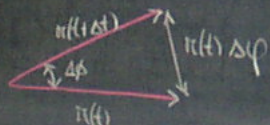
$$\epsilon = \frac{r_{max} - r_{min}}{r_{max} + r_{min}} = \sqrt{\frac{1 + 2EL^2}{\mu k^2}} \quad (8)$$

$r_{min} = r_p$ perihelion perigee
 $r_{max} = r_a$ aphelion apogee
 Planets Earth

Kepler's 2nd Law



Position vector for a planet sweeps out equal areas in equal time intervals.
 $\therefore \frac{dA}{dt} = \text{constant}$.



$$\Delta A = \frac{1}{2} r^2 \Delta \varphi$$

$$\lim_{\Delta t \rightarrow 0} \frac{\Delta A}{\Delta t} = \frac{1}{2} r^2 \lim_{\Delta t \rightarrow 0} \frac{\Delta \varphi}{\Delta t}$$

$$\frac{dA}{dt} = \frac{1}{2} r^2 \omega = \frac{L}{2\mu}$$

= Constant !!

Kepler's 3rd Law

$$\frac{dA}{dt} = \frac{2\mu}{L} \frac{dA}{dt}$$

$$\int dt = \frac{2\mu}{L} \int dA$$

$$T = \frac{2\mu}{L} A = \frac{2\mu}{L} \pi ab$$

$$b = a \sqrt{1 - \epsilon^2}$$

$$a = A/2$$

$$a(1 - \epsilon^2) = \frac{L^2}{\mu k} \quad (9)$$

Algebra

$$T^2 = \frac{4\pi^2 a^3}{G(M+m)}$$