



MIT  
International Center for  
Air Transportation

# **BLOCK 1 PROCEDURE RECOMMENDATIONS FOR LOGAN AIRPORT COMMUNITY NOISE REDUCTION**

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# Executive Summary

Recent developments in navigation and surveillance technology have enabled new high-precision approach and departure operational procedures using GPS and Required Navigation Performance (RNP) standards. These procedures have proven effective for reducing fuel consumption and streamlining some aspects of air traffic control. However, flight tracks that were previously dispersed over wide areas due to less precise navigation or ATC vectoring are more concentrated on specific published tracks with effects on underlying communities.

This study is an initial investigation to identify potential modifications to approach and departure procedures at Boston Logan International Airport (BOS) which would reduce community noise impact in areas which experience flight track concentration. Potential procedure modifications were separated into two sequential “Blocks”. Block 1 procedures were characterized by clear predicted noise benefits, limited operational/technical barriers and a lack of equity issues. Block 2 procedures exhibit greater complexity due to potential operational and technical barriers as well as equity issues (defined as noise redistribution between communities for the purposes of this study). This report presents recommendations for an initial set of Block 1 procedures. Continued analysis and community outreach will inform the identification and development of Block 2 procedures.

RNAV procedures were implemented at BOS between 2012 and 2013. Candidate approach and departure modifications were first identified based on an analysis of historical flight track densities over the communities surrounding BOS before and after the implementation of new RNAV procedures coupled with noise complaint records and US Census population data. Potential procedure modifications were considered for each identified arrival and departure runway including: lateral flight track adjustment to avoid noise-sensitive areas, vertical trajectory modifications including speed, thrust or configuration management as well as techniques to reintroduce dispersion into flight trajectories.

The technical recommendations presented in this report are not developed to an implementation-ready stage. Rather, the work completed to date represents a preliminary feasibility analysis for each recommended procedure. Prior to implementation of any of these recommendations, the FAA will need to execute internal verification and validation processes. Modifications to the recommended procedures may be required. The noise-reduction objectives for each procedure should be retained in any necessary procedure refinements.

Procedure modification options were assigned to Block 1 or Block 2 based on a preliminary evaluation of noise reduction potential, operational/technical feasibility and potential equity issues. Some candidate procedures were rejected due to safety concerns or lack of noise benefits. The noise analysis compared the proposed modification with current procedures on a single-event basis. Noise contours and corresponding population exposures were calculated for the maximum noise level ( $L_{MAX}$ ) and Sound Exposure Level (SEL) metrics.

The technical feasibility analysis included an examination of flight safety, aircraft performance, navigation and flight management system (FMS) limitations, pilot workload, ATC workload, and procedure design criteria. The process of procedure identification and refinement was informed by outreach to impacted stakeholders including community representatives, FAA

regional and national offices, air traffic control (ATC) managers and specialists, airline technical pilots, and public officials.

As a result of this process the procedures which were identified for Block 1 and their primary noise benefits are listed below.

### Block 1 Procedure Recommendations

Proc. ID D = Dep. A = Arr.	Procedure	Primary Benefits
1-D1	Restrict target climb speed for jet departures from Runways 33L and 27 to 220 knots or minimum safe airspeed in clean configuration, whichever is higher.	Reduced airframe and total noise during climb below 10,000 ft (beyond immediate airport vicinity)
1-D2	Modify RNAV SID from Runway 15R to move tracks further to the north away from populated areas.	Departure flight paths moved north away from Hull
1-D3	Modify RNAV SID from Runway 22L and 22R to initiate turns sooner after takeoff and move tracks further to the north away from populated areas.	Departure flight paths moved north away from Hull and South Boston
1-D3a	<i>Option A:</i> Climb to intercept course (VI-CF) procedure	
1-D3b	<i>Option B:</i> Climb to altitude, then direct (VA-DF) procedure	
1-D3c	<i>Option C:</i> Heading-based procedure	
1-A1	Implement an overwater RNAV approach procedure with RNP overlay to Runway 33L that follows the ground track of the jetBlue RNAV Visual procedure as closely as possible.	Arrival flight paths moved overwater instead of over the Hull peninsula and points further south
1-A1a	<i>Option A:</i> Published instrument approach procedure	
1-A1b	<i>Option B:</i> Public distribution of RNAV Visual procedure	

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## Acronyms and Abbreviations

<b>Term</b>	<b>Definition</b>
<b>A4A</b>	Airlines for America
<b>AEDT</b>	Aviation Environmental Design Tool
<b>ASDE-X</b>	Airport Surface Detection Equipment Model X
<b>ATC</b>	Air Traffic Control
<b>BADA-4</b>	Base of Aircraft Data Version 4
<b>BOS</b>	Boston Logan International Airport
<b>DNL</b>	Day-Night Average Level
<b>FAA</b>	Federal Aviation Administration
<b>HMMH</b>	Harris Miller Miller and Hanson, Inc.
<b>IAP</b>	Instrument Approach Procedure
<b>ILS</b>	Instrument Landing System
<b>L<sub>MAX</sub></b>	Maximum Sound Pressure Level
<b>Massport</b>	Massachusetts Port Authority
<b>MCAC</b>	Massport Community Advisory Committee
<b>MIT</b>	Massachusetts Institute of Technology
<b>MOU</b>	Memorandum of Understanding
<b>MTOW</b>	Maximum Takeoff Weight
<b>N<sub>ABOVE</sub></b>	Number of Events Above Set Level
<b>NAS</b>	National Airspace System
<b>NASA</b>	National Aeronautics and Space Administration
<b>NATCA</b>	National Air Traffic Controllers Association
<b>NAVAID</b>	Navigation Aid
<b>NPD</b>	Noise Power Distance
<b>PBN</b>	Performance Based Navigation
<b>RNAV</b>	Area Navigation
<b>RNP</b>	Required Navigation Performance
<b>RVFP</b>	RNAV Visual Flight Procedure
<b>SEL</b>	Sound Exposure Level
<b>SID</b>	Standard Instrument Departures
<b>SPL</b>	Sound Pressure Level
<b>STAR</b>	Standard Terminal Arrival Route
<b>TARGETS</b>	Terminal Area Route Generation, Evaluation, and Traffic Simulation
<b>TASOPT</b>	Transport Aircraft System Optimization

# I. Introduction

Aircraft noise is a growing concern for communities near airports around the United States. While modern aircraft are quieter on a flight-by-flight basis than their predecessors<sup>1</sup>, aircraft overfly some communities with increasing frequency due to traffic growth and flight track concentration. The precision of aircraft navigation has improved over the past few decades due to the introduction of GPS and other advanced navigation systems. This has led to the introduction of advanced Performance Based Navigation procedures<sup>2</sup>, including Area Navigation (RNAV) and Required Navigation Performance (RNP) procedures as illustrated in Figure 1.

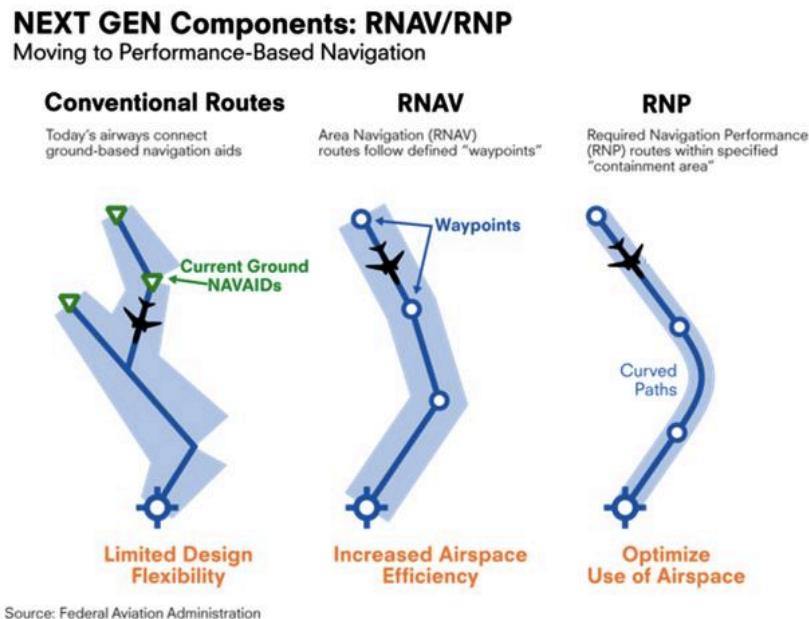


Figure 1. Comparison between conventional, RNAV, and RNP navigation (Figure source: FAA)

Historically, routes were defined by radio navigation aids (NAVAIDS) located at various locations on the ground. Approach and departure procedures consisted of tracks connecting existing NAVAIDS or compass headings issued by air traffic controllers either through published procedures or by radar vectoring. A combination of natural variation in navigational precision and controller instruction timing resulted in a natural dispersion of flight trajectories. This can be seen in the left side of Figure 2 which shows flight tracks of 2010 Runway 33L departures from Boston Logan Airport (BOS) prior to the implementation of RNAV departures.

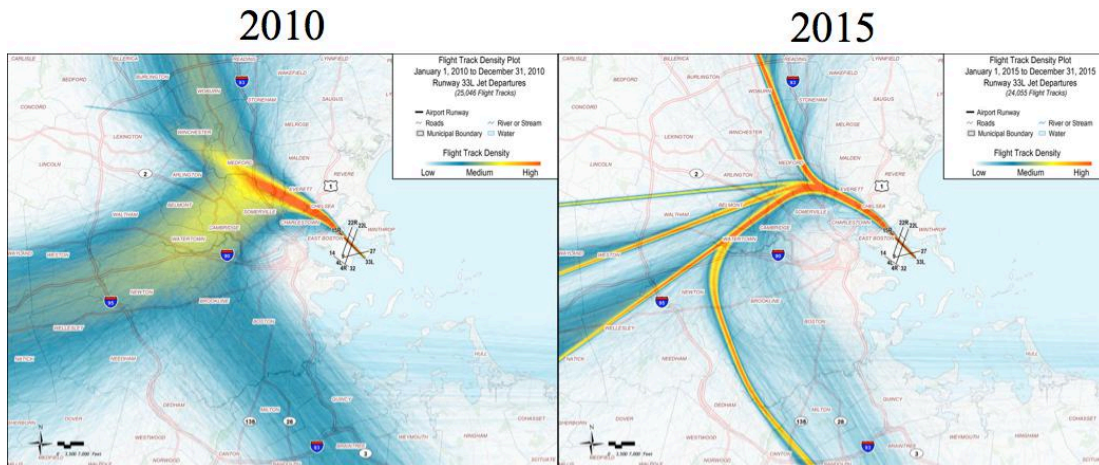


Figure 2. Flight track concentration from Runway 33L departures in 2010 and 2015 (before and after RNAV implementation)

Area Navigation (RNAV) provides the ability for aircraft to navigate between waypoints which can be defined at any location. This improves the precision, safety and flexibility in flight procedures. RNAV procedures are generally comprised of an ordered sequence of waypoints with altitude and/or speed constraints at some or all of the waypoints. Required Navigation Performance (RNP) procedures are even more precise and allow curved flight segments and more precise vertical guidance. RNP procedures can be designed with tighter tolerances than conventional routes or RNAV procedures due to the onboard monitoring and alerting capability of participating aircraft.<sup>3</sup>

In recent years, it has become evident that some PBN procedures have potential unintended consequences in terms of community noise impact.<sup>4</sup> The increased use of Area Navigation (RNAV) and Required Navigation Performance (RNP) procedures has resulted in a concentration of lateral tracks near airports due to the increased precision of these procedures. While this increased precision has allowed operational benefits such as improved safety, reduced ATC workload, higher runway throughput, reduced fuel burn, better terrain avoidance, and lower approach minimums<sup>3</sup>, it has also resulted in noise concentration and community opposition as aircraft fly consistent and repetitive tracks over the same communities. The right panel of Figure 2 shows an example of flight track concentration at Boston Logan Airport (BOS) arising from RNAV departure procedure implementation between 2010 and 2015.

Jet departures at BOS are normally assigned to one of nine RNAV departure procedures. These procedures are typically flown by an airplane's autopilot system, although they can also be flown manually with guidance from the aircraft's onboard navigation systems. Each of the procedures ends at a waypoint that serves as a transition into the high-altitude airway system for a particular direction of flight. The purpose of the published procedures is to provide a safe, systematic, and efficient transition for departing aircraft from liftoff through the cruise phase of flight. However, the precision of the new procedures has removed much of the dispersion in flight tracks that existed prior to RNAV implementation.

Arrivals at BOS also use RNAV Standard Terminal Arrival Routes (STARs) for the transition from the high-altitude airway structure to the airport terminal environment. The final approach and landing may also occur with PBN guidance at some runways, although most flights

use the conventional radio-based Instrument Landing System (ILS) or visual guidance for the final approach to landing. The observed lateral navigation precision of aircraft flying the ILS is similar to RNAV.

Communities around the US have expressed frustration with flight track concentration and noise arising from PBN implementation.<sup>5</sup> At the same time, operational and safety benefits of PBN and the worldwide implementation of new procedures make it difficult to revert to non-PBN procedures. Ideally, PBN technology and procedures could be used to reduce overflight noise while retaining operational benefits.<sup>6</sup> This study is part of an effort to identify PBN approach and departure procedures that could reduce overflight noise and address concerns raised by RNAV noise concentration.

## **II. Study Approach**

### **A. Overview of Study Approach**

The objective of this study is to identify potential procedure modifications at BOS to reduce overflight noise arising from PBN track concentration. The process to reach this objective included a review of flight procedures and radar records from before and after RNAV implementation, identification of problematic runways and procedures in terms of complaints and population impact, identification and noise analysis of candidate procedure modifications for each area of concern, and evaluation of potential barriers to implementation for the proposed modifications. The results of this study are intended to inform procedure design and implementation efforts at the FAA intended to mitigate overflight noise arising from PBN track concentration.

### **B. Identification of Key Problem Areas**

This study used a data-driven approach to identify those runways where approach and departure procedure modifications would have a significant community noise reduction impact. In order to evaluate the drivers of community annoyance from aircraft noise, a review of historical radar tracks and community complaints was undertaken.

#### **1. Flight Track Density Evaluation**

This process included review and visualization of published arrival and departure procedures from the time period before and after implementation of RNAV at Boston Logan Airport. Historical radar data was used to evaluate changes in flight track density for arrivals and departures from each runway used by jet aircraft. For each arrival and departure procedure, areas of flight track concentration were identified for further evaluation. Figure 2 shows an example flight track density plot generated for Runway 33L jet departures before and after RNAV implementation, clearly illustrating the communities which are impacted by increased track concentration. Visualizations for flight track density for the key runways at BOS were generated by Harris, Miller, Miller and Hanson Inc. (HMMH) and are provided for reference in Appendix D.

#### **2. Complaint Analysis**

In addition to raw radar data, complaint data from the Massport noise office were used to identify regions of widespread annoyance arising from specific arrival or departure procedures. These complaints are logged with Massport via phone, voicemail, internet, or mail. The exact time of each complaint was not included in the analysis because complaints are not always filed at the time of the motivating event. Figure 3 shows complaint data from August 2015 to July 2016, after the implementation of RNAV arrivals and departures at BOS. Each address where at least one complaint was filed is shown with a red dot. The left side of the figure shows departure radar tracks and the right side shows arrivals, including both jet and propeller aircraft.

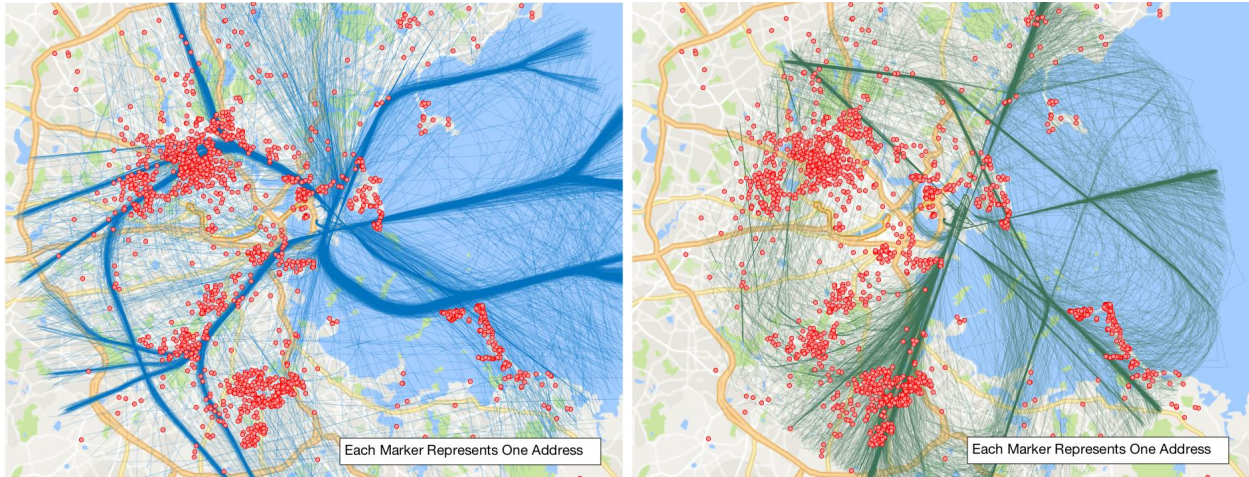


Figure 3. Complaints at BOS between August 2015 and July 2016 (one dot per address) with departure (left) and arrival (right) tracks from 12 days in the same time period

Qualitative assessment of the complaint map shows several areas where complaint clusters were associated with particular arrival or departure corridors. Departures from Runway 33L drive a broad set of complaints in the vicinity of Medford, Somerville, Cambridge, Arlington, and beyond. Departures from Runway 27 are associated with a region of complaints ranging from the South End of Boston to Roxbury, Jamaica Plain, and points beyond. Departures from Runway 22L and 22R drive complaints in South Boston and the Hull peninsula. In terms of arrivals, approaches to runways 4R and 4L drive a region of complaints along the approach path including Braintree, Milton, Dorchester, and South Boston. Approaches to Runway 33L appear to drive additional complaints in the vicinity of Hull. Approaches to runway 22L and 22R appear to drive complaints from Revere, Lynn, Peabody, and other North Shore communities. Complaints outside of these primary clusters (including those outside the geographic bounds of the maps shown in Figure 3) were also evaluated to determine potential annoyance drivers and mitigation strategies further from the airport.

Noise concerns arising from both arrivals and departures in close-in communities surrounding the airport are also evident in the complaint map. However, RNAV technology has a minimal impact on typical flight tracks immediately after takeoff or before landing. RNAV procedure modifications, such as those under investigation in this study, are unlikely to have significant impacts on noise in the immediate vicinity of the airport.

Complaint data is important for identifying high-level annoyance trends, but can also be influenced by outside factors such as unequal access to complaint mechanisms. Therefore, direct community engagement and outreach was also a key component of the procedure evaluation process to identify and understand problem areas for overflight noise.

### 3. Procedure Identification

For each departure and arrival corridor, a set of candidate procedure modifications were identified with input from communities, airline technical pilots, air traffic controllers, and the project technical team. These procedures were evaluated in terms of noise reduction potential, flight safety, community equity, operational implications including airport capacity and throughput, fuel burn/flight time impact, and regulatory requirement.



The following flight procedure concepts were considered in the preliminary phase<sup>7</sup>:

- Modified lateral routing for arrivals and departures to avoid high population density areas, with an emphasis on implementing overwater flight tracks
- Thrust cutbacks on departure
- High-thrust steep climbs
- Reduced speed climbs
- Steeper descent angles on approach
- Multi-segment approaches with a steep segment transitioning to a standard final approach
- Noise-masking approach procedures that overfly regions of high ambient noise (major freeways and industrial areas)

### **C. Phased Approach: Block 1 and Block 2**

In order to provide noise relief to communities in a timely manner, this study involved the development and recommendation of procedures in two phases. The initial set of procedures (Block 1) is characterized by noise benefits in terms of absolute population exposure, no significant equity issues, and manageable operational or technical barriers. These procedures are intended to be “win-win” concepts with strong potential for implementation, pending verification and environmental review.

A follow-on set of procedures (Block 2) will be recommended after further analysis. The Block 2 procedures are expected to exhibit greater complexity due to potential operational and technical barriers as well as equity issues. Altering flight procedures may benefit one community at the expense of another. While such changes may have merit in terms of noise redistribution or environmental equity, negotiation and governance strategies between impacted communities will be needed to reach consensus.

In addition to community equity considerations, Block 2 procedures may involve additional complexity due to operational or technical barriers. Some PBN procedures require specialized pilot training and/or cockpit avionics that may reduce the initial utilization rate in day-to-day operations. Other procedures in Block 2 may be easily flown using standard operating procedures and avionics but require airspace or procedure design waivers.

### **D. Community and Stakeholder Feedback**

Stakeholder feedback was solicited throughout the procedure evaluation process. Communities provided feedback on preliminary concepts through open-forum public meetings as well as briefings to the Massport Community Advisory Committee (MCAC) Aviation Subcommittee. Through these meetings, several concepts were suggested, tested, and/or revised in order to consider specific areas of concern for highly-impacted communities. Due to procedural complexity and potential equity concerns, some of these suggestions were incorporated into analysis plans for Block 2 of this study. Community input also motivated several specific modifications to the Block 1 procedures, including waypoint relocation to maximize potential noise benefits for communities near proposed flight tracks. Community concerns were also communicated through meetings with public officials and political representatives at the state and federal level.



Operators were engaged in this project through several meetings with airline technical pilots and the trade association Airlines for America (A4A). These pilots represented air carriers with significant operational footprints at BOS. The meetings provided feedback on potential operational constraints from the airline perspective including safety concerns arising from specific procedure proposals (including steep approaches, two-segment steep approaches, and speed management on departure). Preliminary versions of certain Block 1 candidate procedures were also test-flown in a full-motion Boeing 767 simulator by technical pilots from a major US airline. This test was intended to provide insight on basic feasibility and flyability of the proposed procedures. No flyability concerns were found based on these informal simulator trials of the Block 1 procedures, although official and detailed procedure design and evaluation is still required to confirm the qualitative preliminary findings.

Regulators and air traffic controllers were also engaged throughout the process. Representatives from the FAA Air Traffic Organization (ATO) were consulted to gain insight and understanding of air traffic control procedures, airspace layouts, standard operating procedures, and potential ATC-related constraints to procedure modification. Meetings with ATC included representatives from the Boston Tower, Boston Terminal Radar Approach Control, Boston Air Route Traffic Control Center, FAA New England Regional Office, the National Air Traffic Controllers Association (NATCA), and FAA headquarters. In addition to ATC, additional FAA engagement included meetings with the following offices: Environment and Energy, ATO Mission Support Services, Flight Standards, Airport Planning and Programming, and NextGen.

## **E. Noise Modeling and Analysis**

Candidate procedures were evaluated using two noise models. The NASA Aircraft Noise Prediction Program (ANOPP) was used for procedures where aircraft speed and/or configuration played a key role in projected noise benefits. The FAA Aviation Environmental Design Tool (AEDT) was used for procedures where the primary noise benefit arises from modified track definitions. This is because AEDT does not fully account for airframe noise changes arising from speed and configuration changes. Noise levels were computed on a 0.1 nautical mile square grid for all desired noise metrics. For calculating population exposure, block-level data from the 2010 US Census was re-gridded onto a 60 nautical mile square grid centered at BOS.

Analysis was performed for three aircraft types representative of the fleet mix at BOS: the Boeing 737-800 (single-aisle, medium range), Boeing 777-300 (twin-aisle, long range), and Embraer 170 (regional jet, short range). Results for all three types are presented in this report for procedures where aircraft-specific configuration and performance plays a key role. For all other procedures, results for the 737-800 alone are shown. This aircraft is representative of narrowbody twin-engine aircraft types that comprise the majority of operations at BOS. In terms of flight profile definitions, each departure procedure was modeled at 90% of maximum takeoff weight (MTOW) and each arrival procedure was modeled at 75% MTOW. For departures, the baseline vertical profile and thrust levels were derived from the median of historical radar tracks. For arrivals, the baseline vertical profile was a 3° glideslope. In both cases, the thrust profile was derived from historical radar tracks and a force-balance kinematics model. This thrust calculation method used aerodynamic data (lift and drag coefficients) calculated using the Eurocontrol Base

of Aircraft Data 4 (BADA-4). The noise analysis tools and methods used in this study are described in greater detail in Appendix A.

## **F. Metrics Used for Procedure Evaluation**

All noise analyses for the Block 1 procedure concepts were performed on a single-event basis. The objective was to evaluate the noise reduction potential for each individual operation rather than integrated impacts. Noise contours and corresponding population exposures were calculated for the maximum noise level ( $L_{MAX}$ ) and Sound Exposure Level (SEL) metrics.  $L_{MAX}$  describes the loudest absolute sound level generated during an overflight, regardless of the duration of the noise event. SEL accounts for the duration of an event.<sup>8</sup> Both  $L_{MAX}$  and SEL showed noise benefits for each Block 1 recommendation presented in this report. For simplicity, only  $L_{MAX}$  results are presented in the main body of this report. Additional details about the single-event noise metrics used for this study are provided in Appendix B. SEL contours and population exposure values are provided in Appendix C for completeness.

### III. Block 1 Procedure Recommendations

The Block 1 recommendations identified and presented below are intended to:

1. Provide noise benefits in terms of absolute population exposure
2. Generate no significant equity issues in terms of noise redistribution between communities
3. Impose minimal operational, technical, or implementation barriers

In the process of evaluating flight tracks, complaints, and community feedback, several communities were identified where noise impacts were clearly evident but no procedures were identified consistent with Block 1 criteria. Arrival and departure procedures for such communities will be considered under Block 2. The specific procedures recommended under Block 1 are listed in Table 1 and are expanded upon in this section of the report.

Table 1. Block 1 Procedure Recommendations

Proc. ID D = Dep. A = Arr.	Procedure	Primary Benefits
1-D1	Restrict target climb speed for jet departures from Runways 33L and 27 to 220 knots or minimum safe airspeed in clean configuration, whichever is higher.	Reduced airframe and total noise during climb below 10,000 ft (beyond immediate airport vicinity)
1-D2	Modify RNAV SID from Runway 15R to move tracks further to the north away from populated areas.	Departure flight paths moved north away from Hull
1-D3	Modify RNAV SID from Runway 22L and 22R to initiate turns sooner after takeoff and move tracks further to the north away from populated areas.	Departure flight paths moved north away from Hull and South Boston
1-D3a	<i>Option A:</i> Climb to intercept course (VI-CF) procedure	
1-D3b	<i>Option B:</i> Climb to altitude, then direct (VA-DF) procedure	
1-D3c	<i>Option C:</i> Heading-based procedure	
1-A1	Implement an overwater RNAV approach procedure with RNP overlay to Runway 33L that follows the ground track of the jetBlue RNAV Visual procedure as closely as possible.	Arrival flight paths moved overwater instead of over the Hull peninsula and points further south
1-A1a	<i>Option A:</i> Published instrument approach procedure	
1-A1b	<i>Option B:</i> Public distribution of RNAV Visual procedure	

## **A. 1-D1: Runway 33L and 27 Reduced Speed Departures**

*Restrict target climb speed for jet departures from Runways 33L and 27 to 220 knots or minimum safe airspeed in clean configuration, whichever is higher.*

### **1. Summary**

Typical jet aircraft departures involve an acceleration to 250 knots shortly after takeoff. At this speed, the NASA ANOPP noise model indicates that, for modern aircraft, airframe noise dominates engine noise. By reducing departure climb speed to a level where airframe noise is similar to engine noise, total source noise can be minimized. ANOPP results indicate that the airframe/engine noise equivalence speed is in the vicinity of 220 knots for most jet aircraft. It is recommended that a speed constraint of 220 knots be assigned to all jet departures. For aircraft not capable of safe operation at 220 knots in a clean configuration, the minimum safe airspeed may be used.

The specific noise benefits and population exposure reduction presented in this report are based on NASA ANOPP modeled results. These results are consistent with the best publicly-available noise analysis data and methods. It may be valuable to conduct initial flight tests or operational trials to provide empirical validation of modeled results. However, the physical drivers of speed-based noise reduction are clear, so implementation of this recommendation is expected to have a beneficial impact regardless of model fidelity.

### **2. Technical Basis for Recommendation**

Aircraft noise is generated by a combination of engine and airframe sources. Improvements in materials and engine design over the past several decades have significantly reduced engine noise. In older generations of aircraft, engines were the dominant noise source during departure. As engine noise has decreased, airframe noise has become more perceptible from the ground. Airframe noise arises due to turbulence in the airflow around components such as flaps and landing gear. Airframe noise is highly dependent on aircraft speed, with higher speeds resulting in higher noise levels. Airframe noise also increases when flaps are extended, speed brakes are used, and/or the landing gear is deployed.<sup>9</sup>

In a typical jet departure, the aircraft accelerates on the runway and performs its initial climb segment at a predetermined takeoff thrust. The initial thrust level may vary based on aircraft weight, runway length, weather conditions, and other variables. During this initial segment, the aircraft climbs at an initial climb speed dependent on aircraft weight. Upon reaching a transition altitude, typically between 1,000 ft and 1,500 ft, the thrust is reduced to a climb setting and the aircraft accelerates to a target climb speed. The target climb speed is typically 250 knots, which is the maximum speed permitted below 10,000 ft in the United States. As the aircraft accelerates, the flaps are incrementally retracted until the wing is in its clean configuration.<sup>10</sup> Figure 4 shows a schematic of a typical departure profile.

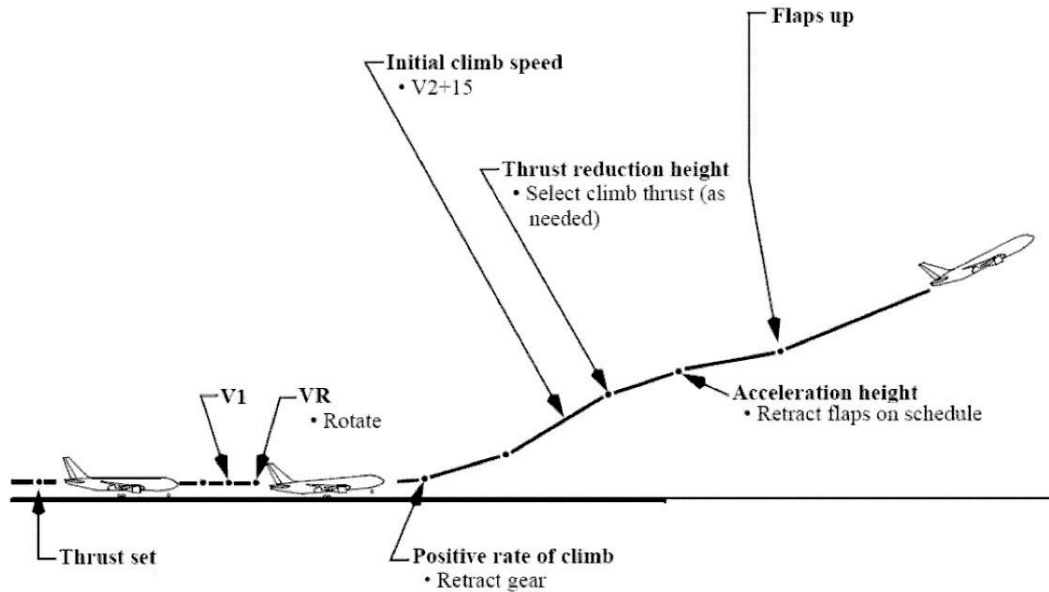


Figure 4. Standard jet departure profile (figure source unknown)

Noise model results indicate a strong interaction between aircraft speed and airframe noise. To demonstrate this effect, the departure profile shown in Figure 4 was modeled with a variable target climb speed ranging from 160 knots to 250 knots. For modeling purposes, thrust levels were held constant for each departure speed. Flaps were assumed to be configured as required for the target speed.

$L_{MAX}$  noise contours for the variable-speed departure profiles for a Boeing 737-800 are shown in Figure 5, illustrating the contribution of engine and airframe sources to the total noise contour at a range of climb speeds. At 160 knots, noise is dominated by engine sources. As the target climb speed increases, airframe noise becomes more pronounced. At 220 knots, engine and airframe noise sources are similar under the departure path. At 250 knots, airframe noise is the dominant source. The transition from engine-dominated to airframe-dominated noise occurs in the range of 210 knots to 230 knots for each of three aircraft types examined in this analysis (Boeing 737-800, Boeing 777-300, and Embraer 170).

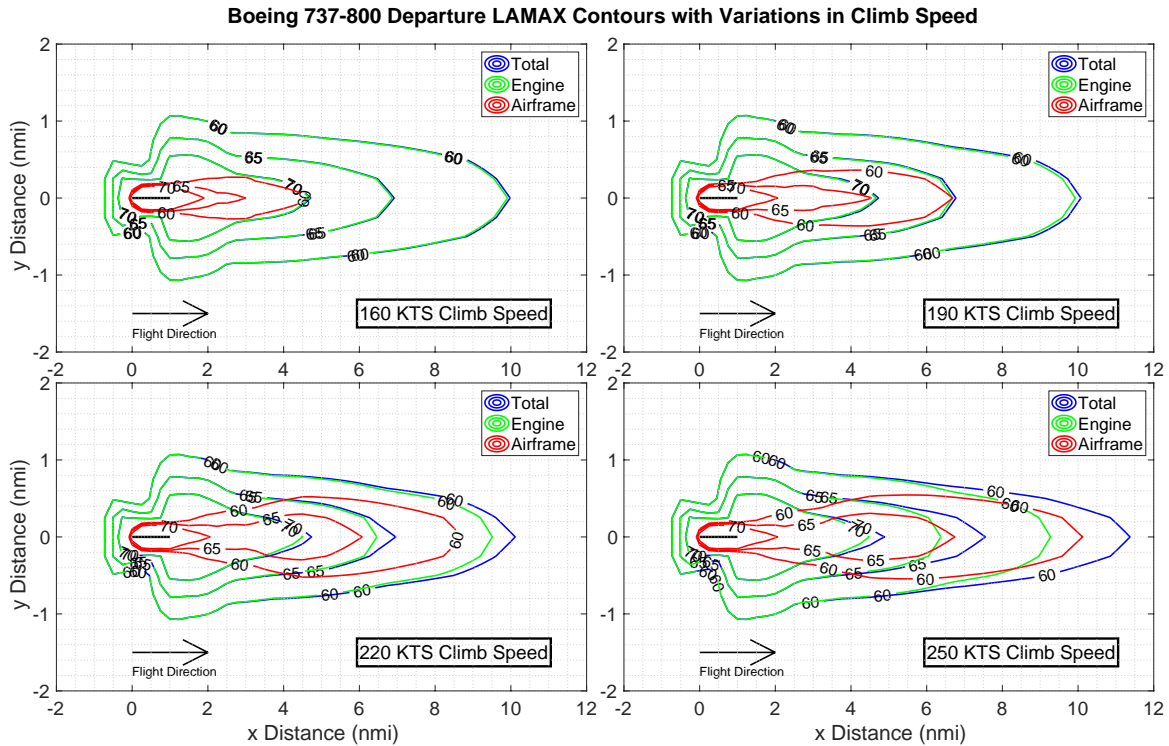


Figure 5.  $L_{MAX}$  noise contours for a 737-800 departure with target climb speeds varying from 160 knots to 250 knots.

For an aircraft operating in the airframe-dominated noise regime, speed reduction results in a reduction of total noise. This presents an opportunity to reduce total noise for departing jet aircraft by setting a target climb speed that is lower than 250 knots, ideally near the transition speed where airframe and engine noise sources are of similar magnitude. Climbing near this transition speed provides the majority of the noise reduction benefit from reduced airframe source while minimizing operational impact.

The benefits from reducing departure speed occur from the initial climb thrust cutback point approximately 5 miles from departure to the point where the aircraft reaches 10,000 ft. This noise reduction occurs primarily underneath the centerline of the departure flight track, which is where the RNAV track concentration effects are most pronounced.

### 3. Track Density Plots

Runway 33L and 27 are the two departure runways at BOS where the climb segment below 10,000 ft occurs primarily over land. Therefore, this procedure recommendation focuses on those runways. Figure 6 shows jet track concentration for departures from Runway 33L before and after implementation of RNAV procedures (2010-2015). Figure 7 shows the same data for Runway 27. In both cases, increased concentration is evident after the implementation of RNAV procedures, especially for communities more than 5 nautical miles away from the airport where tracks were historically dispersed.

Reduced speed departures would serve as an initial step to provide noise relief to those underneath the centerline of departure corridors by reducing the noise associated with each

overflight. Any further procedure modification requiring reallocation of traffic or movement of tracks over other communities does not meet the criteria for Block 1 recommendation (see Section C on page 12).

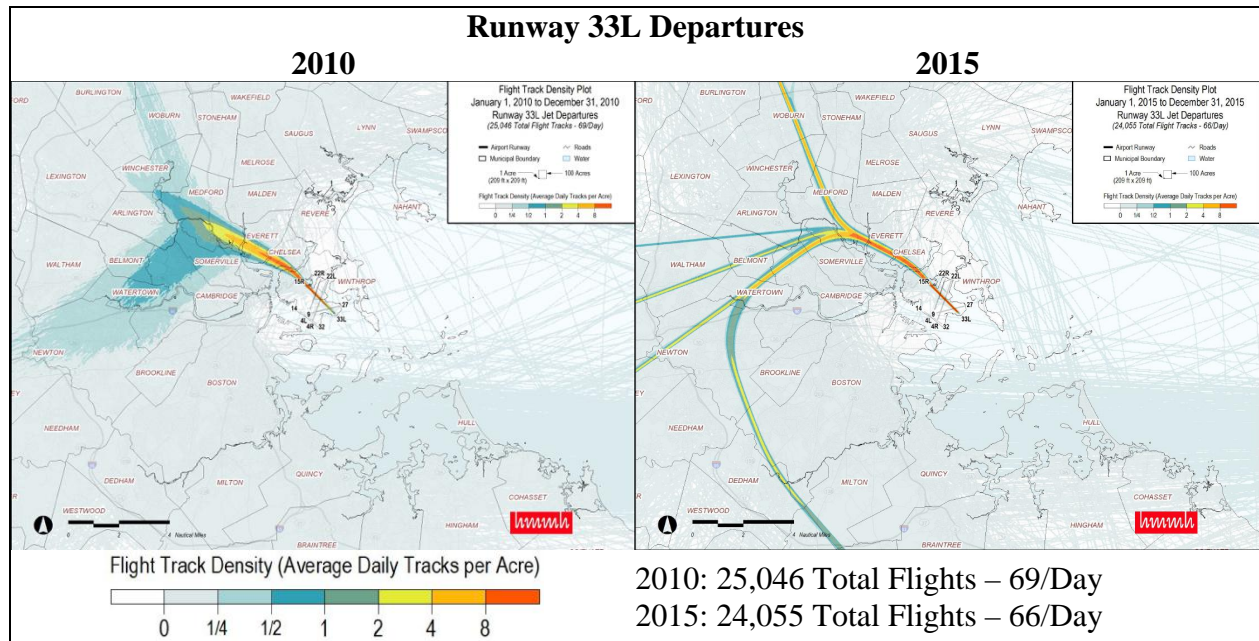


Figure 6. Comparison between flight track density from Runway 33L jet departures between 2010 and 2015

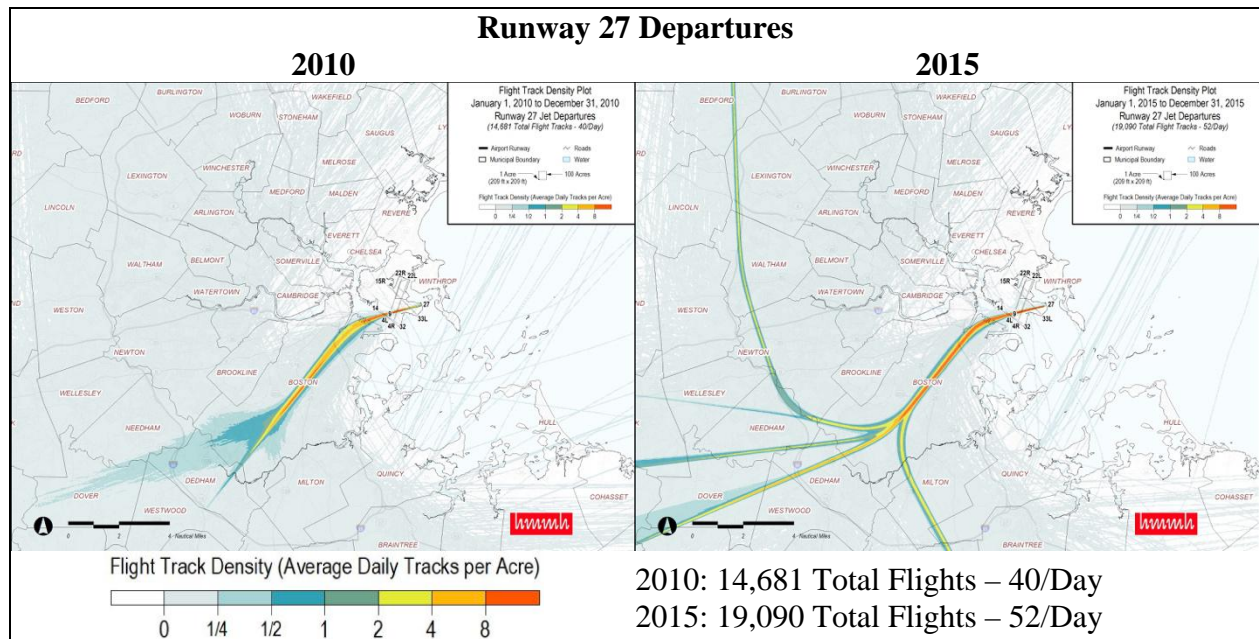


Figure 7. Comparison between flight track density from Runway 27 jet departures between 2010 and 2015

#### 4. Procedure Recommendation Details

Based on modeling results, it is recommended that speed reductions be implemented for jet departures from runways 33L and 27 at BOS. This is expected to reduce noise over populated

areas under the centerline of published departure procedures away from the immediate airport vicinity. This speed reduction could be accomplished through multiple operational strategies, including ATC clearances or modification to published procedures.

The objective of this recommendation is to reduce target climb speed to a value where airframe and engine noise are roughly equivalent in the clean configuration (flaps up). In order to simplify air traffic management and sequencing, it is recommended that the same speed constraint be applied to all departing jet traffic. Noise model results indicate that the airframe/engine noise equivalence speed is in the vicinity of 220 knots for most jet aircraft. Therefore, this procedure consists of modifying the standard departure profile illustrated in Figure 4 with a reduced target climb speed of 220 knots.

Not all aircraft types are capable of operating safely at 220 knots in a clean configuration. There is precedence for safety-based exceptions to speed constraints in the Federal Aviation Regulations under 14 C.F.R. §91.117(d), which state that an aircraft may use the minimum safe airspeed for any particular operation if that speed is greater than the prescribed legal limit. In practice, this would result in certain aircraft types exceeding the 220 knot limitation. This is driven by multiple factors including aircraft weight and wing design. Analysis of the 2015/2016 fleet mix at BOS indicates that 6.9% of departures would likely need to fly at a minimum safe climb speed higher than 220 knots. The need to fly faster than 220 knots would be determined by airline procedures based on aircraft type, weight, and flight conditions. Traffic spacing would be managed by air traffic controllers using the same techniques currently applied to aircraft operating at different speeds.

In order to observe benefits for outlying communities under the departure flight path, the reduced speed must be maintained until an altitude where noise levels are below an acceptable threshold. Based on noise modeling for the 737-800, 777-300, and E-170, an acceleration altitude of 10,000 ft. captures the noise reduction benefit for both heavy and light aircraft. An acceleration altitude of 6,000 ft. was found to retain the population exposure benefits for light aircraft but significantly reduce benefits for heavy aircraft (which typically generate more source noise and climb at a shallower gradient). Therefore, it is recommended to implement the speed restriction to 10,000 ft. to maximize population exposure benefits from the procedure.

In terms of implementation strategy, the procedure modification could be accomplished through a notation on existing SIDs or through explicit air traffic controller instructions for departing aircraft. There is precedent for published speed restrictions of 220 knots on existing SIDs elsewhere in the NAS, such as the STAAV Eight RNAV Departure from Las Vegas McCarran airport shown in Figure 8. These restrictions are typically motivated by procedure design constraints assuming worst-case wind conditions. However, similar constraints could be applied for noise mitigation reasons. For rapid implementation (or implementation on a trial basis), the speed constraint could be assigned by the tower controller as part of the takeoff clearance or the departure controller as part of the initial climb clearance.



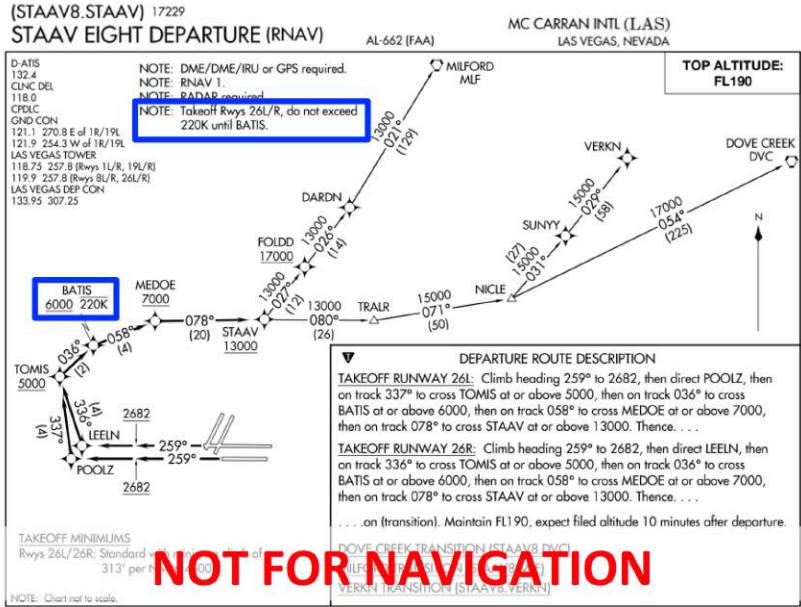


Figure 8. STAAV Eight RNAV SID at Las Vegas McCarran airport with 220 knot speed restriction before BATIS waypoint

### 5. Noise Modeling Results and Population Exposure

Noise was modeled for the proposed reduced speed departure procedures using the NASA ANOPP model described in Appendix A. In order to evaluate population impact for a single representative departure, each of these aircraft was modeled on the “BLZZR Four” RNAV standard instrument departure (SID) from Runways 33L and 27, a typical route used for departures to southwesterly destinations such as Atlanta and Dallas. For a procedure baseline, the analysis uses a standard departure profile with a 250-knot target climb speed and a vertical profile derived from median radar data for that aircraft type and runway. The thrust cutback altitude for the baseline procedure and all modified procedure was also based on this historical data.

For all aircraft types, the contour geometry is unchanged in the immediate vicinity of the airport. Contour contraction occurs approximately five to thirty miles from the departure end of the runway where unrestricted departures would have already accelerated beyond 220 knots. This corresponds to regions of concern for RNAV track concentration. Figure 9 shows single-event noise contours ( $L_{MAX}$ ) and population exposure results for the 737-800 in a clean configuration with a target climb speed of 220 knots. Figure 10 shows similar results for the 777-300, although the target climb speed was limited to 240 knots due to minimum speed constraints for that aircraft type. Figure 11 shows contours for the E-170 with a target climb speed of 220 knots. Figure 12 shows contours for 737-800 with a target climb speed of 220 knots from runway 27. According to these modeled results, all three aircraft types show noise reduction due to reduced speed departures. Large population exposure reductions are evident, particularly at the 65 dB level and below. Specific reductions depend on the underlying population density which varies by departure runway and procedure. For both runways, areas of noise reduction occur in locations under the departure procedure centerline corresponding to areas of frequent community noise complaints. No communities experience an increase in noise as a result of reduced speed departures.

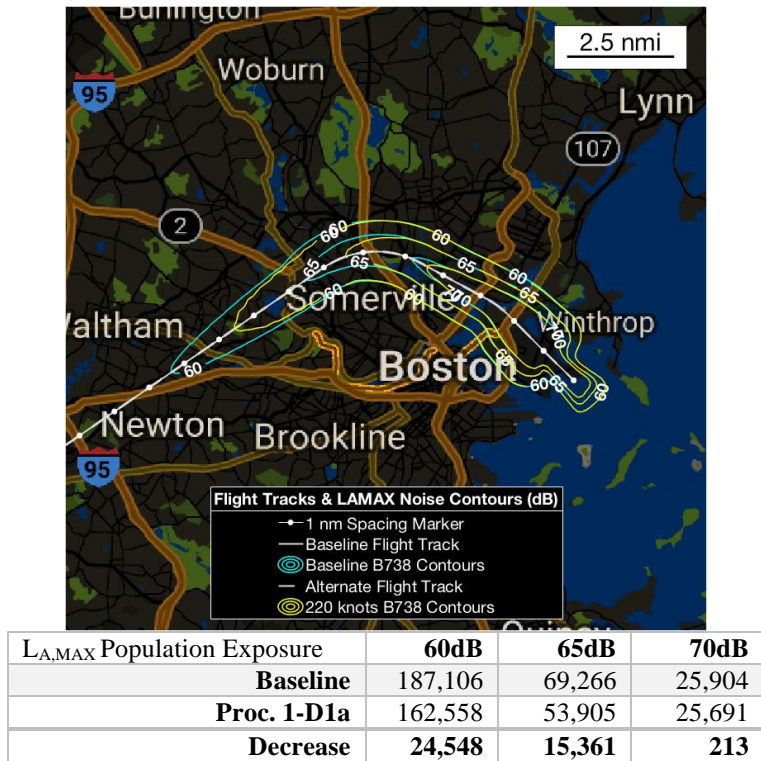


Figure 9. Noise exposure reduction for the Boeing 737-800 departing runway 33L via the BLZZR4 departure on a standard climb profile compared to a 220-knot reduced speed departure. Noise Model: NASA ANOPP

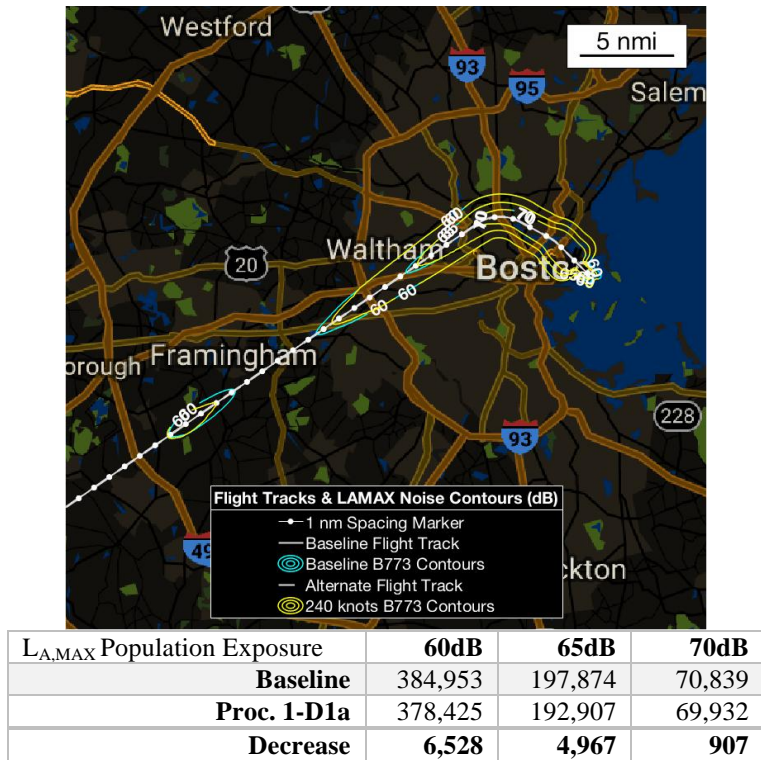


Figure 10. Noise exposure reduction for the Boeing 777-300 departing runway 33L via the BLZZR4 departure on a standard climb profile compared to a 240-knot reduced speed departure. Noise Model: NASA ANOPP

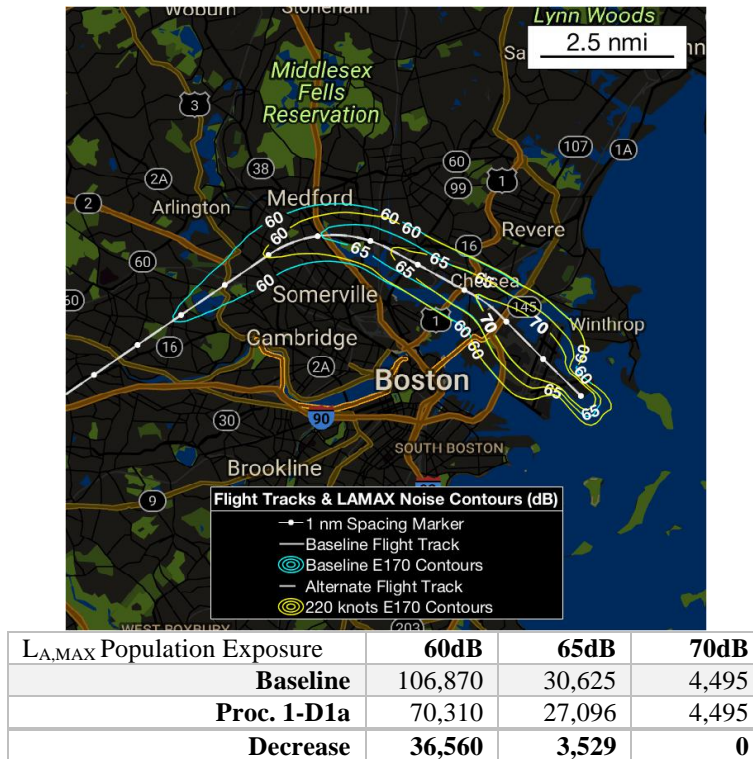


Figure 11. Noise exposure reduction for the Embraer E-170 departing runway 33L via the BLZZR4 departure on a standard climb profile compared to a 220-knot reduced speed departure. Noise Model: NASA ANOPP

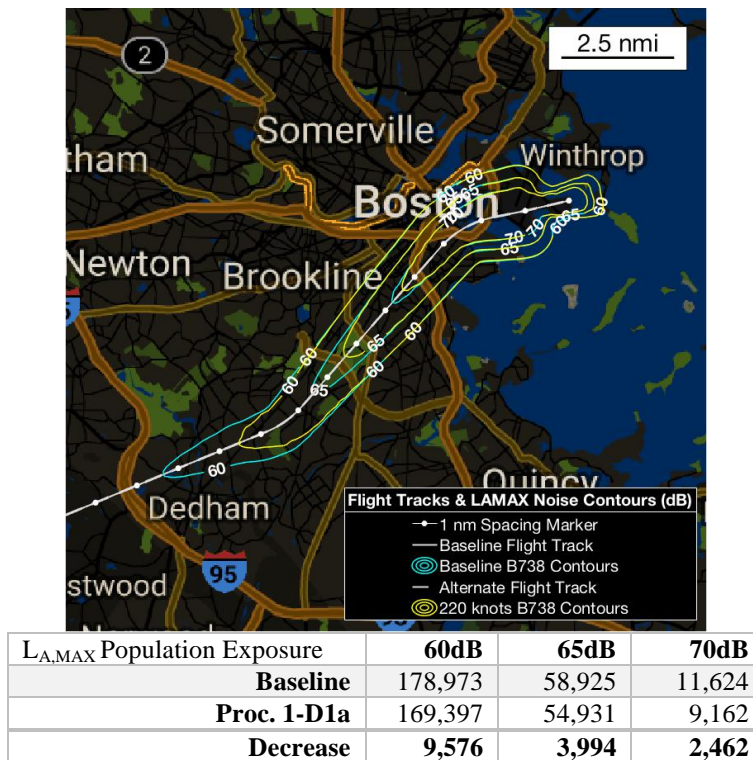


Figure 12. Noise exposure reduction for the Boeing 737-800 departing runway 27 via the BLZZR4 departure on a standard climb profile compared to a 220-knot reduced speed departure. Noise Model: NASA ANOPP

## 6. Potential Barriers to Implementation

Three potential barriers to entry were identified in consultation with operational stakeholders:

- Fuel burn and flight time increase
- Potential runway throughput reduction
- Limitations on aerodynamic maneuvering margins at 220 knots

Each of these potential barriers to entry was evaluated as part of the study and found not to pose an unmanageable issue. Details of each potential barrier are provided below.

### *a) Fuel Burn and Flight Time*

Performance modeling of reduced-speed climbs was conducted using the Eurocontrol BADA-4 model and indicates a slight fuel burn and flight time penalty from the procedure. This is because the aircraft are required to cover the baseline track distance at a slower speed. Naturally, this results in a slight time increase. Fuel burn also increases slightly for each aircraft type examined in this study, which can be attributed to the increased flight time as well as slightly lower aerodynamic efficiency at reduced speeds. Table 2 shows the fuel burn and time impact for representative reduced-speed departures with an acceleration altitude of 10,000 ft. These relatively small values (under 11 gallons of fuel and 30 seconds of flight time) are not considered significant and are smaller than penalties for other common noise abatement procedures.

Table 2. Fuel consumption and flight time implications from reduced speed climb procedures

<b>Aircraft</b>	<b>Climb Speed</b>	<b>Fuel Burn Increase vs. Baseline</b>	<b>Flight Time Increase vs. Baseline</b>
<b>737-800</b>	220 Knots	46 lbs (6.8 gallons)	30 seconds
<b>777-300</b>	240 Knots	71 lbs (10.4 gallons)	12 seconds
<b>E-170</b>	220 Knots	9 lbs (1.3 gallons)	22 seconds

### *b) Departure Sequencing and Runway Throughput*

When tower controllers release aircraft for takeoff, they commonly assume that the leading aircraft will accelerate and take this into consideration when determining the departure release time for the trailing aircraft. Airborne aircraft are subject to minimum separation requirements. In general, aircraft must be separated by 3 nautical miles horizontally and/or 1,000 ft. vertically or placed on divergent headings. Detailed separation requirements are specified in FAA Joint Order 7110.65X<sup>11</sup>. For the purpose of departure metering, air traffic controllers must provide a sufficient time interval between takeoff clearances to ensure 3 nautical mile separation between leading and trailing aircraft after the trailing aircraft becomes airborne and throughout the departure procedure. Imposing reduced speed constraints on departing aircraft has the potential to impact the required interval between takeoff clearances.

In order to evaluate potential throughput implications of reduced speed departures, historical radar tracks from the Airport Surface Detection Equipment X (ASDE-X) were analyzed. This system logs aircraft position, altitude, and speed in 1-second intervals within 10-

12 nautical miles of the airport. The analysis data set consisted of 2015 and 2016 departures from Runways 33L and 27 at BOS, for a total of 27,713 operations. Each pair of sequential departures in this set was analyzed on a second-by-second basis using the baseline (as-flown) speed profile as well as a modified speed profile limited to 220 knots or the minimum safe airspeed for the respective aircraft type, whichever was greater. In the reduced speed scenario, the start of takeoff roll time was maintained at the baseline value. Minimum horizontal separation was determined on a second-by-second basis for both the baseline and modified scenarios.

The historical radar data analysis showed minimal throughput implications for the proposed reduced speed departure procedure. 54 departure pairs that had maintained 3 nautical mile separation in the baseline case would have violated that horizontal spacing after the imposition of reduced speeds if no adjustments to release time occurred. This corresponds to 1 departure out of every 513 that would have required air traffic control action different from what occurred in the 2015-2016 timeframe. The departure release delay required to remove these conflicts was small, with a median delay of 1.1 seconds. Therefore, the potential departure sequencing and runway throughput impact of reduced speed departures is expected to be small and manageable by air traffic controllers without requiring significant changes in standard operating practices.

*c) Slow-Speed Maneuvering*

Some aircraft types cannot operate with adequate maneuvering margins at 220 knots in a clean configuration at high takeoff weights. This is addressed through a provision for minimum safe airspeed in lieu of the 220 knot restriction for aircraft with such constraints. For the majority of the fleet mix at BOS, the 220 knot recommendation is safely flyable in the clean configuration at normal weights. However, airline policy and pilot discretion can guide the use of alternative minimum safe airspeed on a case-by-case basis. This allows sufficient flexibility to pilots and air traffic controllers to implement the noise-driven departure modification without compromising safety.

The recommendation also calls for minimum safe airspeed in the clean configuration rather than with flaps or slats extended. This reduces noise from flap gaps and edges, fatigue on structural components, and potential issues with extended high-lift devices in icing conditions. It also minimizes the fuel burn penalty associated with the recommended procedure. Therefore, concerns regarding flaps-extended climbs have been minimized to the extent possible in this recommendation.



## B. 1-D2: Runway 15R RNAV Waypoint Relocation

*Modify RNAV SID from Runway 15R to move tracks further to the north away from populated areas.*

### 1. Summary

Turbojet departures from Runway 15R currently climb on runway heading before proceeding to an RNAV waypoint located 0.46 nautical miles north of Hull. This waypoint location concentrates overflights near a populated area rather than further to the north over Boston Harbor. It is recommended that the initial segment of the standard RNAV SID departure be redesigned to remain farther north, maximizing overflight of Boston Harbor rather than the Hull Peninsula.

### 2. Track Density Plots

Figure 13 shows jet track concentration for departures from Runway 15R before and after implementation of RNAV procedures (2010-2015). It is clear from the figure that the departure tracks became more concentrated after RNAV implementation and that the centroid of the departure corridor shifted south toward Hull. It is desirable to retain the benefits of RNAV technology while returning the departure flight path from Runway 15R closer to its pre-RNAV centroid over Boston Harbor.

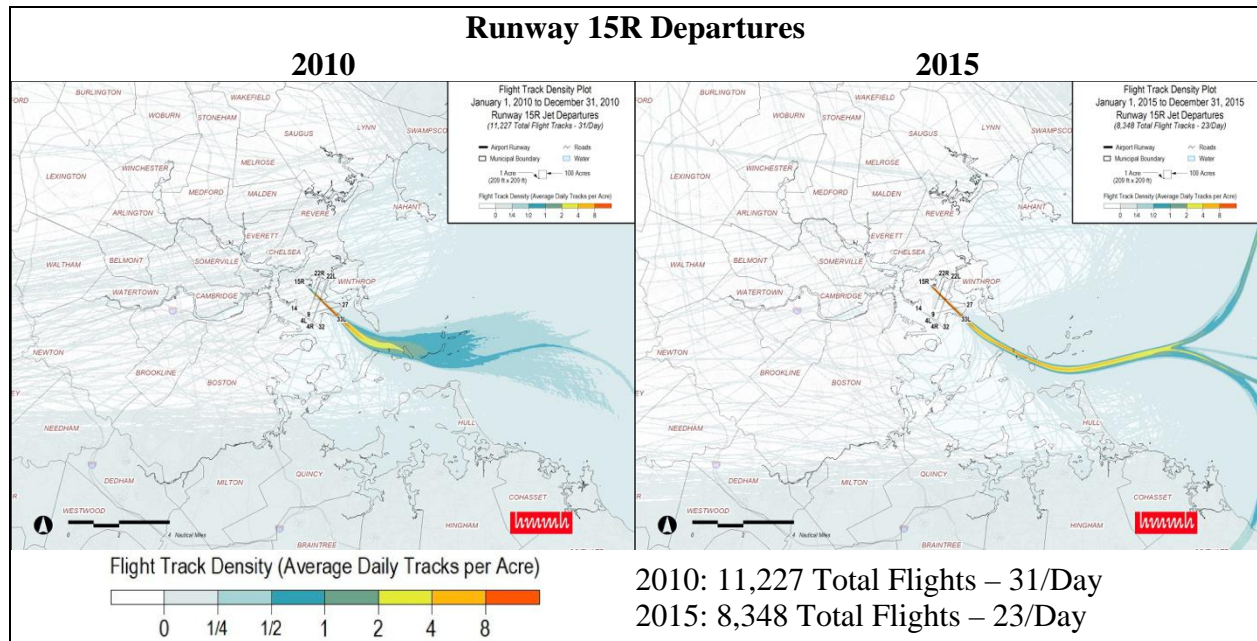


Figure 13. Comparison between flight track density from Runway 15R jet departures between 2010 and 2015

### 3. Procedure Recommendation Details

Turbojet departures from Runway 15R currently climb on runway heading to intercept course 131° to the FOXXX waypoint located 0.46 nautical miles north of the Hull Peninsula, then eastbound on course 091° to the BRRRO waypoint 4 nautical miles offshore in Massachusetts Bay before diverging onto the various departure procedure tracks. Because Hull is

impacted by departures and arrivals from multiple runways at BOS, it is desirable to move the published RNAV procedures as far overwater to the north as possible to provide relief. While a specific procedure definition is recommended in this report, any modification that shifts the centerline further north over Boston Harbor would accomplish the underlying objective of this recommendation. Figure 14 shows the recommended procedure modification, bypassing the FOXXX waypoint and proceeding directly to BRRRO as close as possible to the center of Boston Harbor.

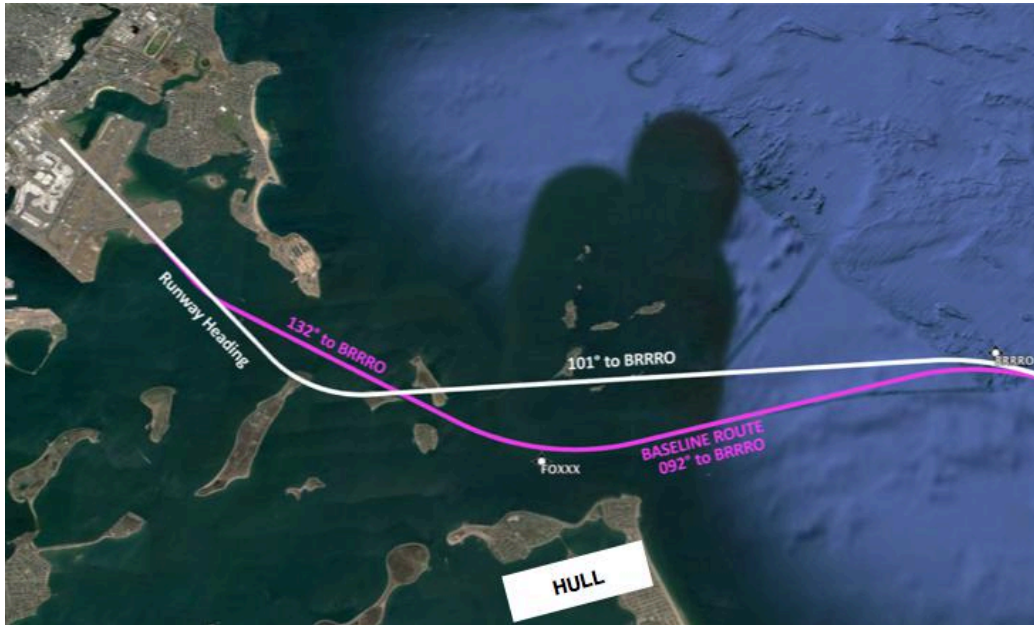


Figure 14. Procedure illustration for a 15R departure climbing via the BLZZR4 departure (baseline) compared to procedure 1-D2

This recommendation is intended to comply with existing RNAV SID design constraints. Bypassing the FOXXX waypoint reduces noise in Hull while maintaining the basic structure of the baseline procedure (a climb on runway heading to intercept a course to a specific waypoint, or “VI-CF” type procedure). Waypoint coordinates are provided in Table 3, corresponding to the white line shown in Figure 14. All waypoints are designated as flyby rather than flyover.

Table 3. Waypoint locations and leg type definitions for procedure recommendation 1-D2

Leg Number	Leg Definition	From	To	Notes
1	<i>Climb to Intercept (VI)</i> Runway Heading (approx. 150°)	RW15R 42°22'27.25"N 71°01'04.35"W	Turn Point (Approx.) 42°19'35.04"N 70°57'14.03"W	Precise intercept location may vary based on initial heading and other factors
2	<i>Course to Fix (CF)</i> Course 101° to BRRRO	Turn Point (Approx.) 42°19'35.04"N 70°57'14.03"W	BRRRO 42°20'00.78"N 70°48'05.48"W	
3+	As defined in baseline	BRRRO 42°20'00.78"N 70°48'05.48"W	As Defined	Remainder of existing RNAV SID definitions unchanged

#### 4. Noise Modeling Results and Population Exposure

Noise was modeled for the proposed waypoint relocation using the AEDT model described in Appendix A. Analysis was performed for the Boeing 737-800. The “BLZZR Four” RNAV SID from Runways 15R was modeled for the purpose of this report, although each of the published SIDs uses the same initial segment definition in the vicinity of interest near Hull. The baseline procedure was a standard departure profile with a 250-knot target climb speed and a vertical profile derived from median radar data for that aircraft type and runway.

Figure 15 shows single-event  $L_{MAX}$  contours and the population exposure reduction over the Hull Peninsula for a Boeing 737-800 following procedure 1-D2 as illustrated in Figure 14. No communities are exposed to new noise as a result of this recommended modification.

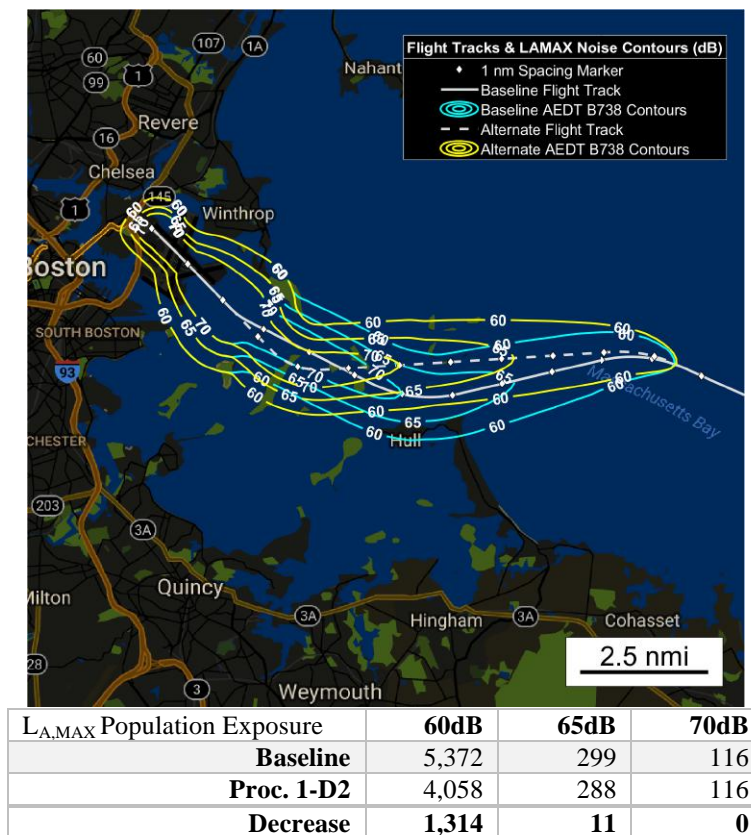


Figure 15. Noise exposure reduction for the Boeing 737-800 departing runway 15R climbing via the BLZZR4 departure on an AEDT-standard climb profile compared to procedure 1-D2

#### 5. Potential Barriers to Implementation

No significant barriers to entry are anticipated for this procedure recommendation. The procedure involves a minimal track length reduction relative to the baseline procedure. Compliance with procedure design criteria will require final verification in TARGETS. Additionally, the presence of Boston Harbor under and to the north of the existing departure corridor provides substantial flexibility to modify detailed aspects of this recommended procedure if needed while maintaining the overall objective of noise reduction in Hull.



## C. 1-D3: Runway 22L and 22R RNAV Waypoint Relocation

*Modify RNAV SID from Runway 22L and 22R to initiate turns sooner after takeoff and move tracks further to the north away from populated areas.*

*Option A: Climb to intercept course (VI-CF) procedure*

*Option B: Climb to altitude, then direct (VA-DF) procedure*

*Option C: Heading-based procedure*

### 1. Summary

Turbojet departures from Runway 22L and 22R currently climb on runway heading to a specified point before making a left turn overwater towards a waypoint three miles to the southeast. At that waypoint, the departures turn eastbound toward Massachusetts Bay along an RNAV procedure segment that is offset from Hull by less than 0.5 nautical miles. This RNAV segment concentrates overflights near a populated area rather than further to the north over Boston Harbor. It is recommended that the segment of the standard RNAV SID near Hull be redesigned to remain farther north, maximizing overflight of Boston Harbor rather than the Hull Peninsula. In addition, it is recommended that procedure definition be modified to initiate the overwater turn as early as practical after takeoff to reduce noise in South Boston.

### 2. Track Density Plots

Figure 16 shows jet track concentration for departures from Runway 22R before and after implementation of RNAV procedures (2010-2015). Similar to the Runway 15R departures, the departure tracks became more concentrated after RNAV implementation and the centroid of the departure corridor shifted south toward Hull. In addition, some aircraft appear to have initiated the initial turn after takeoff sooner prior to RNAV implementation. It is desirable to retain the benefits of RNAV technology while returning the departure flight path from Runway 22R closer to its pre-RNAV state, including an earlier turn after takeoff and centroid further north over Boston Harbor.

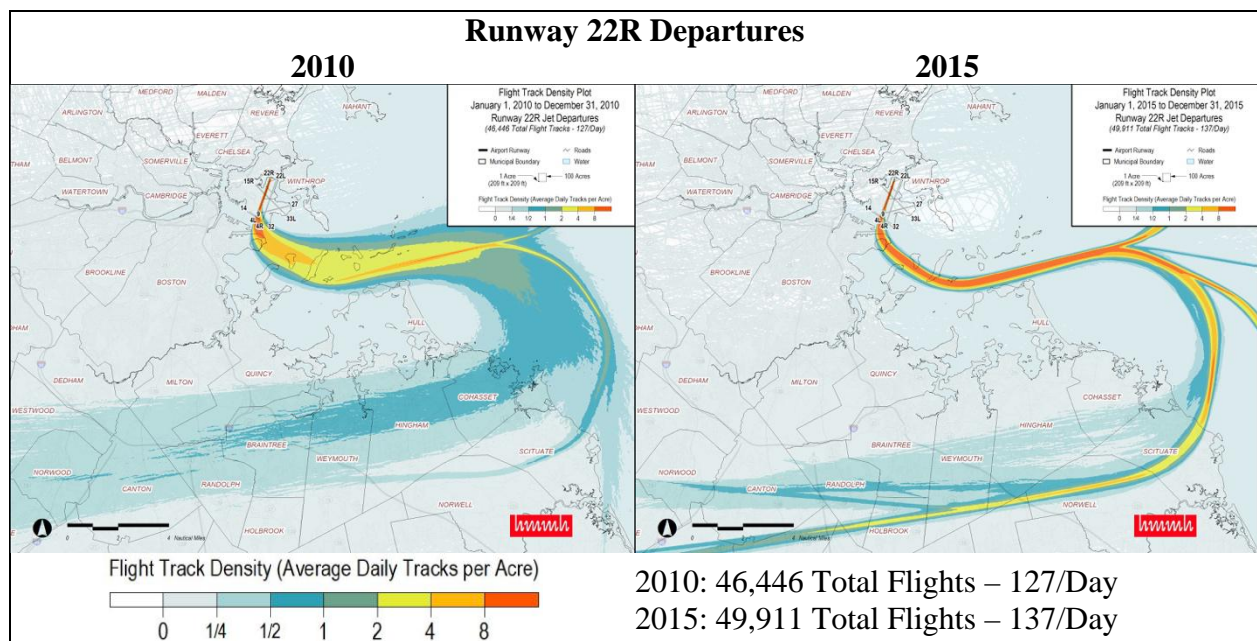


Figure 16. Comparison between flight track density from Runway 22R jet departures between 2010 and 2015

### 3. Procedure Recommendation Details

Turbojet departures from Runway 22L and 22R currently climb on runway heading to intercept course  $139^\circ$  or  $143^\circ$  (respectively) to the TJAYY waypoint located in Boston Harbor, then eastbound on course  $091^\circ$  to the BRRRO waypoint 4 nautical miles offshore in Massachusetts Bay before diverging onto the various departure procedure tracks. The preliminary turn to TJAYY is intended to reduce overflights of South Boston, while the eastbound segment is intended to keep departure trajectories overwater until aircraft reach sufficient altitude to reduce noise. Figure 17 shows the baseline RNAV SID geometry for departures from Runway 22L and 22R in magenta. The figure also shows in white the final approach corridor for traffic landing on Runway 27 which must be separated from departure flows when both procedures are simultaneously in use. In this situation, separate air traffic control sectors generally handle arrivals and departures. The boundary between the arrival and departure sectors is shown in green.



Figure 17. Baseline procedure definitions for RNAV SIDs from Runway 22L and 22R (magenta) shown with ILS Localizer to Runway 27 (white) and the air traffic control sector boundary (green)

Two communities could benefit from implementation of modified departure procedures from Runway 22L and 22R. Earlier turns after takeoff could reduce overflight noise in South Boston, while waypoint relocation in the vicinity of Hull could move tracks further overwater and reduce impacts in that heavily-impacted community. Three specific procedure definition options are recommended in this report. Each of these options has unique benefits mechanisms as well as potential implementation barriers that may require evaluation and mitigation prior to implementation. Should revision of these recommendations be required prior to implementation, any modification that shifts the centerline of departures further north over Boston Harbor would accomplish the underlying objective of this recommendation. It is also desirable to enable earlier turns on departures through RNAV design or alternative methods.

a) 1-D3a: Runway 22L/R RNAV waypoint relocation (climb to intercept course)

Figure 18 shows the procedure recommendation 1-D3a (white line). This procedure variant uses the same leg types and geometry used in the current published departures, shifting the segment between TJAYY and FOXXX on the baseline procedure to the north. It retains the current turn location after takeoff.

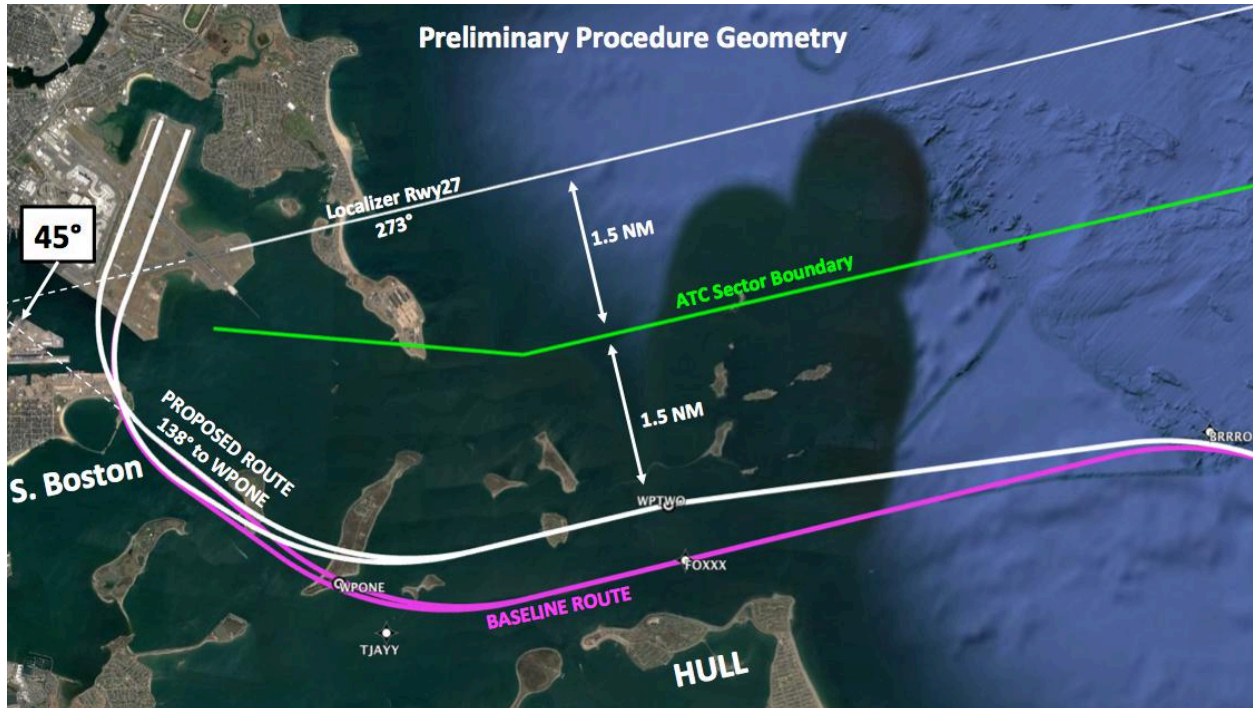


Figure 18. Procedure illustration for a 22L/R departure climbing via the BLZZR4 departure (baseline) compared to procedure 1-D3a

Procedure 1-D1a maintains the existing leg type definition with modified waypoint location (a climb on runway heading to intercept a course to a specific waypoint, or “VI-CF” type procedure). The procedure is intended to maintain a 45° divergence angle between the initial departure flow from Runway 22R and the extended centerline of Runway 27 until 3 nautical miles of separation is achieved<sup>11</sup>. The waypoint locations were selected to maintain minimum separation with the ATC sector boundary (1.5 nautical miles) as well as to provide procedural separation of 3 nautical miles with the localizer to Runway 27. Waypoint coordinates are provided Table 4, corresponding to the white line shown in Figure 18. All waypoints are designated as flyby rather than flyover.

Table 4. Waypoint locations and leg type definitions for procedure recommendation 1-D3a (listed for Runway 22R departures only)

Leg Number	Leg Definition	From	To	Notes
1	<i>Climb to Intercept (VI)</i> Runway Heading (approx. 215°)	RW22R 42°22'41.76"N 71°00'16.30"W	Intercept Point 42°20'32.28"N 71°01'19.05"W (Approx.)	Precise intercept location may vary based on initial heading and other factors
2	<i>Course to Fix (CF)</i> Course 143° to WPONE	Intercept Point 42°20'32.28"N 71°01'19.05"W (Approx.)	WPONE 42°18'43.78"N 70°58'10.07"W	
3	<i>Direct to Fix (DF)</i>	WPONE 42°18'43.78"N 70°58'10.07"W	WPTWO 42°19'24.04"N 70°54'21.64"W	
4	<i>Direct to Fix (DF)</i>	WPTWO 42°19'24.04"N 70°54'21.64"W	BRRRO 42°20'00.78"N 70°48'05.48"W	
5+	As defined in baseline	BRRRO 42°20'00.78"N 70°48'05.48"W	As Defined	Remainder of existing RNAV SID definitions unchanged

*b) 1-D3b: Runway 22L/R RNAV waypoint relocation (climb to altitude then direct)*

Figure 19 shows the procedure recommendation 1-D3b (white lines). This procedure variant uses a modified procedure definition that allows for earlier turns after takeoff for certain steep climbing aircraft. The figure shows three possible ground tracks: the earliest turn represents a steep-climbing aircraft, the next turn represents a typical narrow-body departure, and the final track represents the latest permitted turn location based on minimum climb gradient (which would occur very rarely in actual operations). The majority of departures on this procedure would follow a ground track close to the middle trajectory (also used for noise modeling purposes in this study).





Figure 19. Procedure illustration for a 22R departure climbing via the BLZZR4 departure (baseline) compared to procedure 1-D3b

Procedure 1-D1b modifies the waypoint location in the vicinity of Hull and also changes the initial leg type to a climb on runway heading to an altitude threshold then direct to a specific waypoint (a “VA-DF” type procedure). Waypoint coordinates are provided in Table 5, corresponding to the set of white paths shown in Figure 19. No coordinates are provided for the expected turn point because of variability in climb rate between aircraft types.

Table 5. Waypoint locations and leg type definitions for procedure recommendation 1-D3b

Leg Number	Leg Definition	From	To	Notes
1	<i>Climb to Intercept (VA)</i> Runway Heading (approx. 215°) to 500' AGL	RW22R 42°22'41.76"N 71°00'16.30"W	Point Reaching 500' AGL	Turn location may vary based on aircraft climb gradient, autopilot engagement delay, or other factors
2	<i>Direct to Fix (DF)</i> Direct WPTWO	Point Reaching 500' AGL	WPTWO 42°19'24.04"N 70°54'21.64"W	
3	<i>Direct to Fix (DF)</i>	WPTWO 42°19'24.04"N 70°54'21.64"W	BRRRO 42°20'00.78"N 70°48'05.48"W	
4+	As defined in baseline	BRRRO 42°20'00.78"N 70°48'05.48"W	As Defined	Remainder of existing RNAV SID definitions unchanged

c) *1-D3c: Runway 22L/R heading-based departure when Runway 27 arrivals not in use*

Figure 20 shows the procedure recommendation 1-D3c (white line). In this procedure, the local tower controller would issue a heading of 100° at the time of takeoff clearance. Aircraft would have the flexibility to commence the turn based on pilot discretion and company policy, likely allowing earlier turns than the current RNAV engagement altitude between 400 and 500 ft above ground level. Once clear of population-sensitive areas, the aircraft may continue on ATC vectors or be cleared to a downstream fix on a published RNAV SID. This procedure is only possible when Runway 27 is not in use for arrivals.

Recommendation 1-D3c is not mutually exclusive from 1-D3a and 1-D3b. When Runway 27 arrivals are in use, one of the other options would be required to provide separation. This heading-only procedure places departures over the center of Boston Harbor and has the largest noise benefit of the three recommendation options. It would have a positive impact on surrounding communities if implemented when traffic conditions allow.

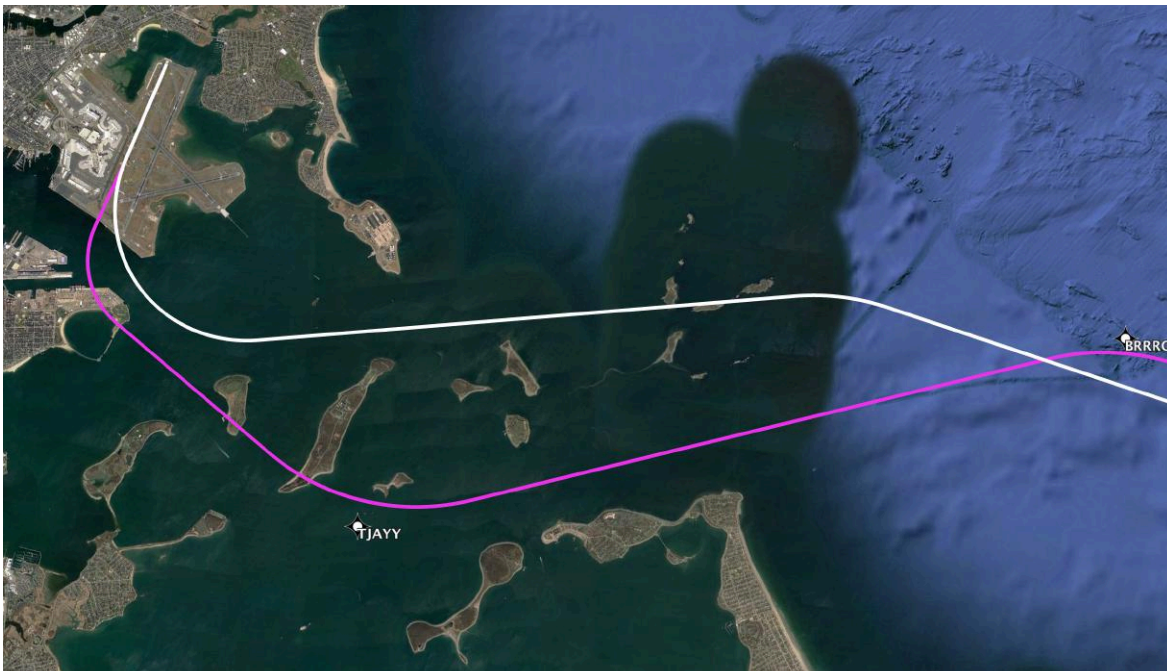


Figure 20. Procedure illustration for a 22R departure climbing via the BLZZR4 departure (baseline) compared to procedure 1-D3c

Procedure 1-D3c involves ATC vectors after takeoff to join existing departure streams, so no additional RNAV waypoint definitions or leg types must be specified.

#### **4. Noise Modeling Results and Population Exposure**

Noise was modeled for the three procedure recommendation options using the AEDT model described in Appendix A. Analysis was performed for the Boeing 737-800. The “BLZZR Four” RNAV SID from Runways 22R was modeled for the purpose of this report. Similar to the Runway 15R recommendation, each of the published SIDs from runway 22L and 22R uses the same initial segment definition in the vicinity of interest near Hull, so the same noise results are applicable for all jet departures regardless of SID assignment. The baseline procedure was a

standard departure profile with a 250-knot target climb speed and a vertical profile derived from median radar data for that aircraft type and runway.

Figure 21 shows single-event  $L_{MAX}$  contours and population exposure reduction results for a Boeing 737-800 following procedure 1-D3a as illustrated in Figure 18. The primary noise benefit occurs at the 60 dB level in the vicinity of Hull. No communities are exposed to new noise as a result of this recommended modification. However, this procedure does not change the turn altitude after takeoff, so no noise benefits are realized in South Boston under this version of the recommendation.

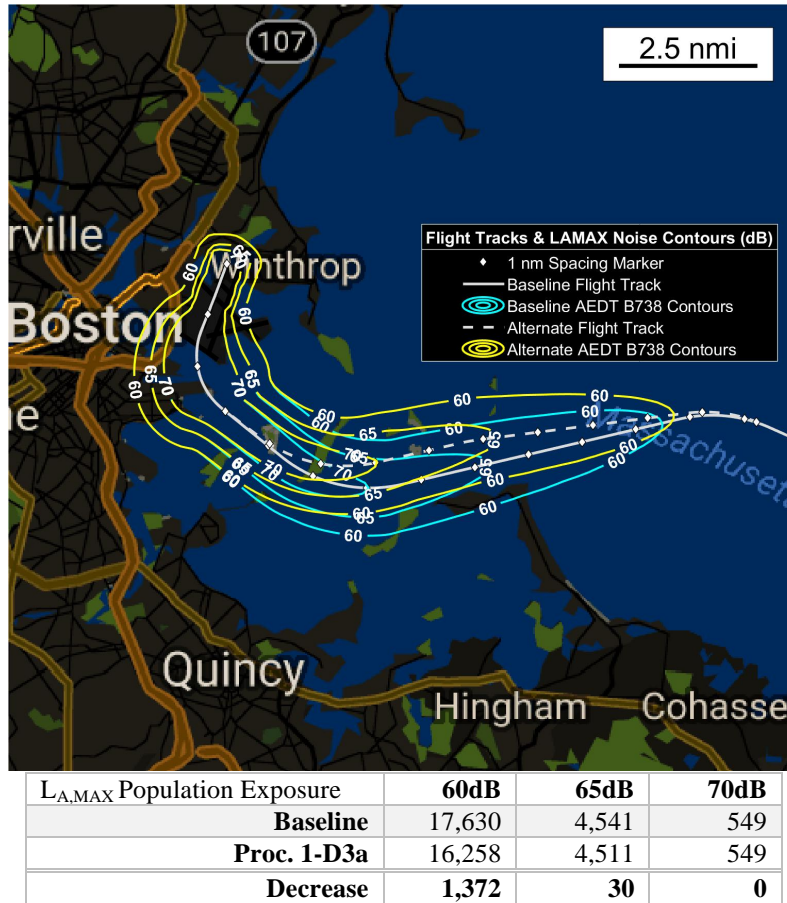
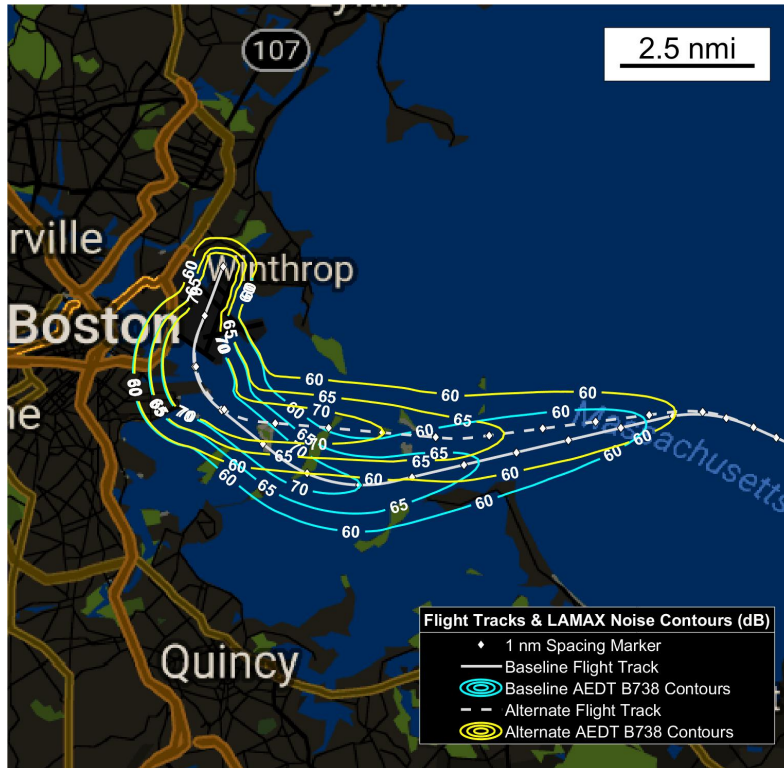


Figure 21. Noise exposure reduction for the Boeing 737-800 departing runway 22R climbing via the BLZZR4 departure on an AEDT-standard climb profile compared to procedure 1-D3a

Figure 22 shows single-event  $L_{MAX}$  contours and population exposure reduction results for a Boeing 737-800 following procedure 1-D3b as illustrated in Figure 19. As in procedure 1-D3a, no new communities are exposed to noise in this procedure. Noise benefits at the 60dB level in the vicinity of Hull are retained. Additionally, this procedure recommendation results in an earlier turn after takeoff relative to the baseline for most aircraft types. This provides additional noise benefits to South Boston under this version of the recommendation compared to 1-D3a.



$L_{A,MAX}$ Population Exposure	60dB	65dB	70dB
<b>Baseline</b>	17,630	4,541	549
<b>Proc. 1-D3b</b>	16,250	4,511	537
<b>Decrease</b>	<b>1,380</b>	<b>30</b>	<b>12</b>

Figure 22. Noise exposure reduction for the Boeing 737-800 departing runway 22R climbing via the BLZZR4 departure on an AEDT-standard climb profile compared to procedure 1-D3b

Figure 23 shows single-event  $L_{MAX}$  contours and population exposure reduction results for a Boeing 737-800 following procedure 1-D3c as illustrated in Figure 20. This procedure has the largest noise benefit to South Boston of the three recommended options for Runway 22R departures due to a combination of an early turn on departure with a ground track near the center of Boston Harbor. These noise benefits could be realized if a procedure similar to 1-D3c was implemented during periods when Runway 27 was not in use for arrivals.





Figure 23. Noise exposure reduction for the Boeing 737-800 departing runway 22R climbing via the BLZZR4 departure on an AEDT-standard climb profile compared to procedure 1-D3c

## 5. Potential Barriers to Implementation

There are two potential barriers to implementation for the RNAV SID recommendations. The first regards procedure design criteria in terms of RNAV leg length, minimum turn arcs, and airspace/procedure separation standards. Due to constrained geography and airspace, the recommended procedures push against the limits of some existing criteria and may require waivers during procedure development. The second potential barrier involves flight path length variability for procedures 1-D3b and 1-D3c. This could introduce increased monitoring requirements for departures from Runway 22L and 22R.

Both of these potential barriers should be addressable during the procedure refinement process. Should operational or criteria constraints dictate modifications to the proposed procedures, the procedure objectives should be retained if possible to initiate turns as soon as possible after takeoff and move tracks farther north over Boston Harbor in the vicinity of the Hull peninsula.

### a) Procedure Design Criteria

Procedure recommendation 1-D3a uses leg lengths that are as short as possible to provide the maximum feasible noise benefit by turning aircraft overwater as quickly as possible. Some of these leg lengths may require waivers against standard design criteria, which are generally based

upon worst-case winds and high airspeeds. Upon further evaluation against procedure design criteria, waivers may be required to enable implementation of a noise-minimizing VI-CF procedure similar to that presented in recommendation 1-D3a. Alternatively, speed constraints could be applied in the initial phase of the RNAV SID to enable the shorter than typical leg lengths. Regardless of criteria evaluation and waiver processes, flight checks and operational evaluation will expose any potential safety concerns. No flyability or passenger comfort issues are anticipated due to the proposal's similarity to existing procedure geometry.

Recommendation 1-D3b may have a similar speed-based turn arc criteria violation on the initial direct-to-fix leg off the runway. This could potentially be addressed through a waiver, speed restriction, and/or waypoint relocation that preserves the flight track offset from Hull.

For both RNAV SID recommendations (1-D3a and 1-D3b), waivers for the 1.5 nautical mile sector boundary separation requirement, 3 nautical mile procedural separation standard with the Runway 27 arrival corridor, and/or 45° divergence angle requirement from the arrival flow could provide additional flexibility in this design. The RNAV SID recommendations presented here move the waypoints as far to the north over Boston Harbor as possible when those standards are used as hard constraints. However, the vertical separation between arrival and departure flows (well more than the minimum required separation of 1000 ft in most cases, thus complying with standards) suggests a potential opportunity for procedure-specific waivers to move departures further north over the harbor as was standard procedure prior to RNAV implementation.

*b) Path Variability and ATC Procedure*

Procedure recommendation 1-D3b and 1-D3c involve greater path length variability than the baseline VI-CF procedure. In certain high-rate departure scenarios, this could require additional monitoring by departure sector controllers to ensure that an early-turning aircraft does not overtake a later-turning aircraft. High departure rate situations may require additional ATC monitoring. Similar monitoring requirements would be required for the heading-based departure recommendation 1-D3c, which would involve active heading vectors and ATC monitoring when the procedure is active.

## D. 1-A1: Runway 33L Low-Noise Overwater Approach Procedures

*Implement an overwater RNAV approach procedure with RNP overlay to Runway 33L that follows the ground track of the jetBlue RNAV Visual procedure as closely as possible.*

*Option A: Published instrument approach procedure*

*Option B: Public distribution of RNAV Visual procedure*

### 1. Summary

Current approaches to Runway 33L overfly the Hull Peninsula from the southeast to the northwest as part of the final approach segment or during vectors to final. This results in noise exposure to underlying communities that are also impacted by departures from Runway 22R, 22L, and 15R. There is an opportunity to reduce noise for the communities underlying this final approach course by designing an overwater RNAV procedure with RNP overlay that avoids the Hull Peninsula to the extent possible given procedure design criteria.

### 2. Track Density Plots

Figure 24 shows jet track concentration for arrivals to Runway 33L before and after implementation of RNAV procedures (2010-2015). Noise concentration along the final approach corridor is evident in both images, spanning several populated land masses to the southeast of the airport. Utilization of the “Light Visual” approach with its overwater dog-leg segment appears to have been more prevalent in 2010 than in 2015.

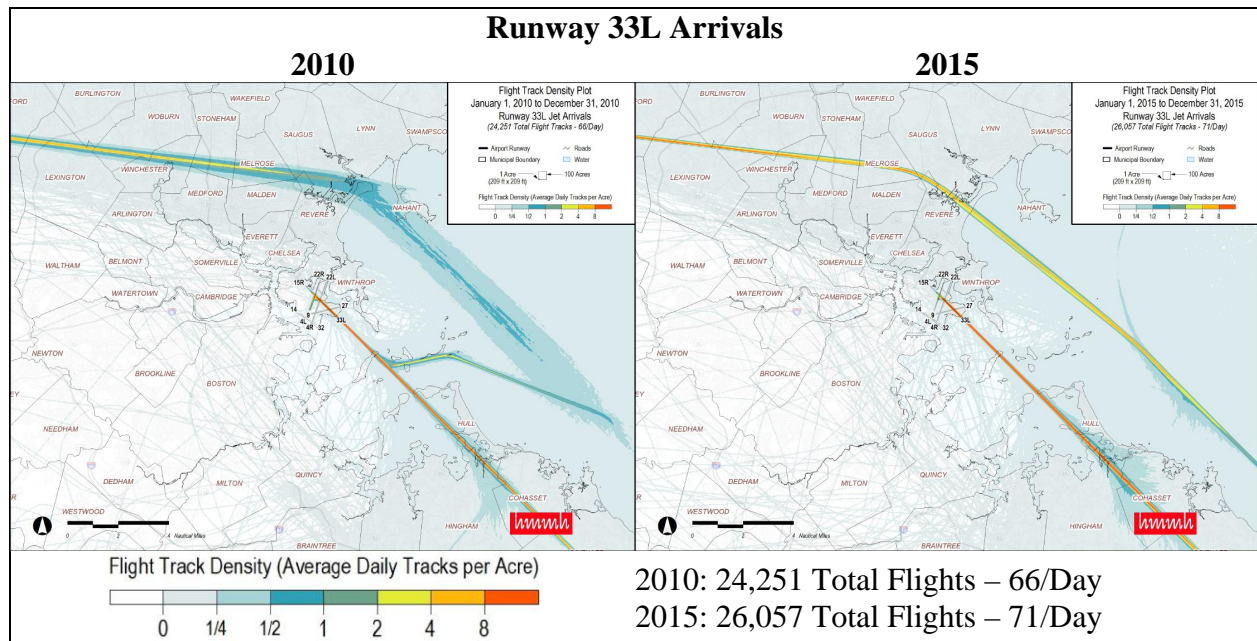


Figure 24. Comparison between flight track density from Runway 33L jet arrivals between 2010 and 2015

### 3. Procedure Recommendation Details

A visual approach procedure to Runway 33L which moves arrival tracks away from Hull has been available for several years for use in good weather conditions (minimum of 3,000 ft. cloud ceilings and 5 miles of visibility). The procedure, shown in Figure 25, includes a dogleg over Boston Harbor with a 55° turn to intercept the final approach path at a point 2.95 nautical miles from the runway threshold. The “Light Visual” procedure was intended for use during low-

demand periods, particularly during late night operations. The procedure is operationally challenging as a visual approach due to the lack of lighted features on the water at night.

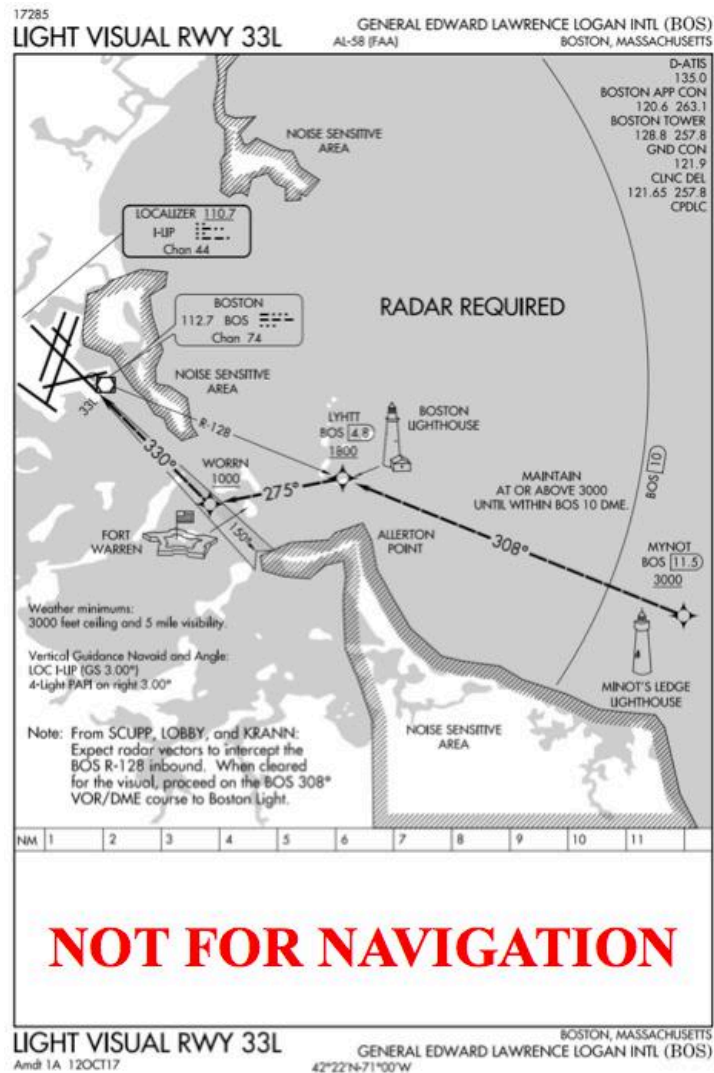


Figure 25. “Light Visual” approach procedure for Runway 33L at BOS

In an effort to increase utilization of the overwater approach procedure concept, jetBlue Airways developed a company-specific RNAV Visual Flight Procedure (RVFP) approach to Runway 33L that closely mirrored the original Light Visual from the southeast with the addition of an additional feeder route from the northwest. RVFP approaches are a hybrid between visual approaches (where navigation is ultimately the responsibility of the pilots) and formal instrument approach procedures (IAPs) which are developed, checked, and published by the FAA for use in poor weather conditions. This provides greater flexibility for procedures designed as RVFP approaches.<sup>12</sup> These approaches are not restricted in final turn angle or minimum final leg length because pilots are able to visually monitor and avoid terrain. The jetBlue “RNAV Visual” approach chart is shown in Figure 26. The RVFP allows jetBlue pilots and aircraft to fly the visual procedure with improved guidance from the aircraft flight management system, improving safety and helping improve conformance to the desired overwater flight tracks.

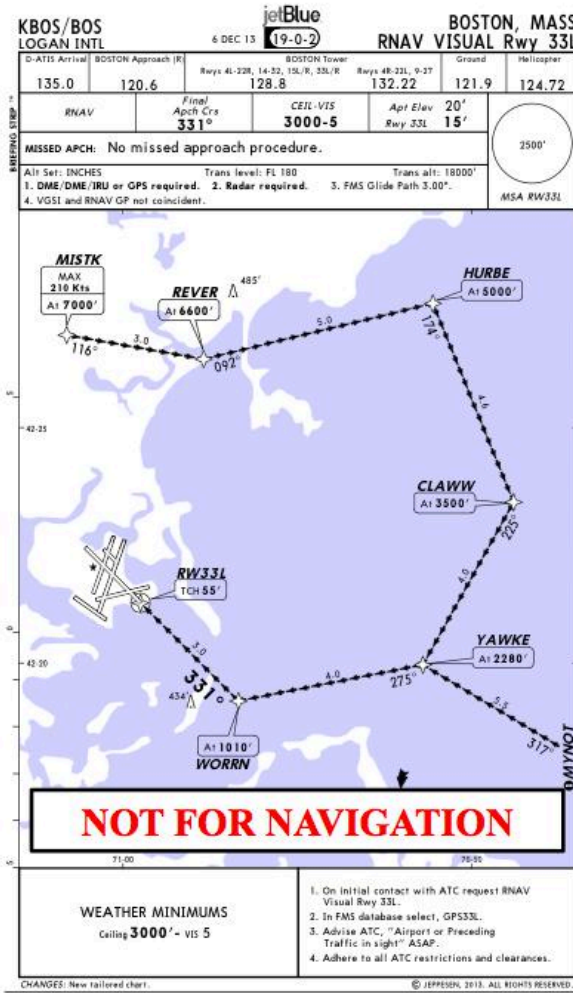


Figure 26. “RNAV Visual” approach procedure for Runway 33L at BOS developed by jetBlue Airways

The primary benefit of RVFPs compared to published RNAV IAPs is a relaxation of procedure design criteria. RNAV IAPs with vertical guidance have a maximum final approach intercept angle of 15° and a final approach stage length of 3.1 nautical miles for typical 3° glideslope procedures. RNAV IAPs without vertical guidance allow final approach intercept angles up to 30°<sup>13</sup>. RVFPs are not subject to these criteria, allowing noise-minimizing designs such as the jetBlue example which has a final approach intercept angle of 56°.

In order to extend the noise benefits of the Light Visual and jetBlue RVFP, two recommended modifications are discussed below:

- 1-A1a: Develop an overwater RNAV instrument approach procedure with RNP overlay which as closely as possible follows the existing jetBlue “RNAV Visual” track while complying with more stringent IAP design criteria
- 1-A1b: Develop a public distribution mechanism for RVFP procedures for use by a broader subset of operators at BOS



Figure 27 shows a comparison of the ground track for the jetBlue RVFP (blue track) with an example RNAV instrument approach procedure concept that complies with nonprecision (no altitude guidance) approach design criteria (green track). The approach design constraints on IAPs prevented an exact overlay of the jetBlue approach, although the required waypoint changes are not substantial. This ground track is recommended as an example implementation of a nonprecision RNAV IAP that can be overlaid with an RNP equivalent for appropriately-equipped aircraft.

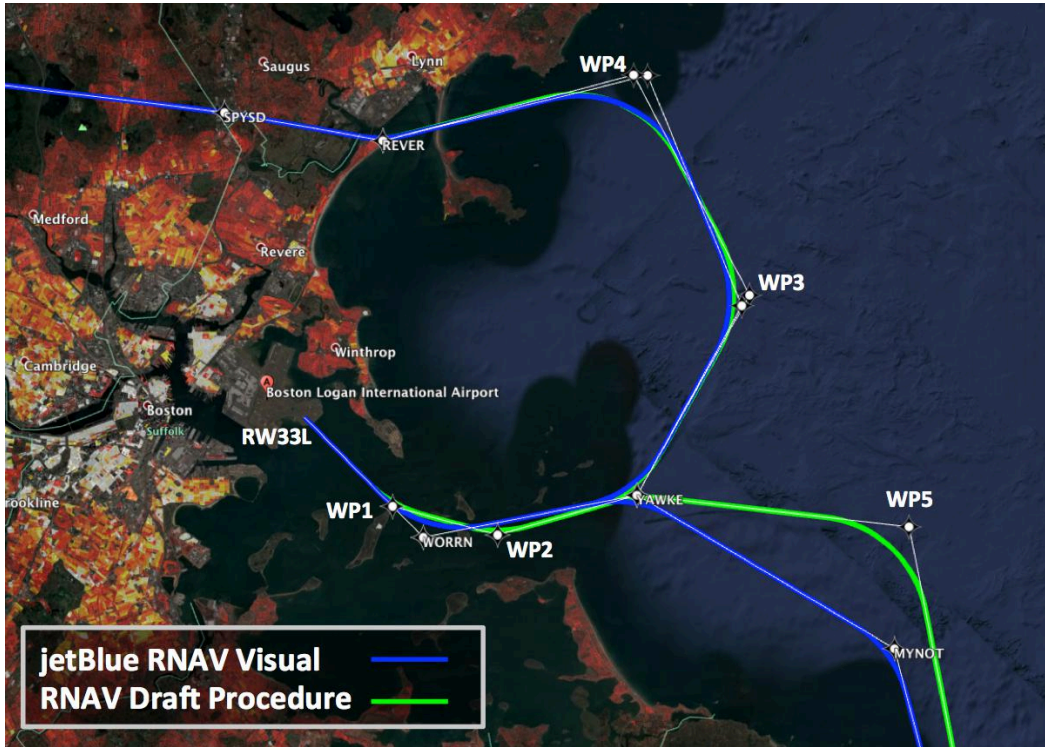


Figure 27. jetBlue RNAV Visual approach procedure to Runway 33L (blue) compared with an example RNAV draft nonprecision instrument approach procedure

This recommendation is intended to comply with existing RNAV nonprecision approach procedure design constraints. Waypoint coordinates are provided in Table 6 for northerly arrivals and Table 7 for southerly arrivals, corresponding to the green tracks shown in Figure 27. All waypoints are designated as flyby rather than flyover.

Table 6. Waypoint locations and leg type definitions for the northern component of procedure recommendation 1-A1a

<b>Leg Number</b>	<b>Leg Definition</b>	<b>From</b>	<b>To</b>
1	<i>Direct to Fix (DF)</i>	<b>SPYSD (7,000')</b> 42°26'58.450" N 71°01'37.250" W	<b>REVER (6,600')</b> 42°26'27.480" N 70°57'41.310" W
2	<i>Direct to Fix (DF)</i>	<b>REVER (6,600')</b> 42°26'27.480" N 70°57'41.310" W	<b>WP4 (5,000')</b> 42°27'39.207" N 70°51'27.753" W
3	<i>Direct to Fix (DF)</i>	<b>WP4 (5,000')</b> 42°27'39.207" N 70°51'27.753" W	<b>WP3 (3,500')</b> 42°23'36.905" N 70°48'36.024" W
4	<i>Direct to Fix (DF)</i>	<b>WP3 (3,500')</b> 42°23'36.905" N 70°48'36.024" W	<b>YAWKE (2,200')</b> 42°19'57.400" N 70°51'24.050" W
5	<i>Direct to Fix (DF)</i>	<b>YAWKE (2,200')</b> 42°19'57.400" N 70°51'24.050" W	<b>WP2 (1,400')</b> 42°19'13.850" N 70°54'51.180" W
6	<i>Direct to Fix (DF)</i>	<b>WP2 (1,400')</b> 42°19'13.850" N 70°54'51.180" W	<b>WP1 (800')</b> 42°19'45.338" N 70°57'27.285" W
7	<i>Direct to Fix (DF)</i>	<b>WP1 (800')</b> 42°19'45.338" N 70°57'27.285" W	<b>RW33L (landing)</b> 42°21'16.743" N 70°59'29.710" W

Table 7. Waypoint locations and leg type definitions for the southern component of procedure recommendation 1-A1a

<b>Leg Number</b>	<b>Leg Definition</b>	<b>From</b>	<b>To</b>
1	<i>Direct to Fix (DF)</i>	<b>MYNOT</b> 42°17'07.810" N 70°45'01.990" W	<b>WP5 (3,800')</b> 42°19'21.690" N 70°44'39.720" W
2	<i>Direct to Fix (DF)</i>	<b>WP5 (3,800')</b> 42°19'21.690" N 70°44'39.720" W	<b>YAWKE (2,200')</b> 42°19'57.400" N 70°51'24.050" W
3	<i>Direct to Fix (DF)</i>	<b>YAWKE (2,200')</b> 42°19'57.400" N 70°51'24.050" W	<b>WP2 (1,400')</b> 42°19'13.850" N 70°54'51.180" W
4	<i>Direct to Fix (DF)</i>	<b>WP2 (1,400')</b> 42°19'13.850" N 70°54'51.180" W	<b>WP1 (800')</b> 42°19'45.338" N 70°57'27.285" W
5	<i>Direct to Fix (DF)</i>	<b>WP1 (800')</b> 42°19'45.338" N 70°57'27.285" W	<b>RW33L (landing)</b> 42°21'16.743" N 70°59'29.710" W

It is also recommended that an RNP overlay be developed following the RNAV ground track as closely as practical to enable seamless ATC integration between flights using the two different approaches. This would enable RNP-equipped aircraft to fly the procedure with higher precision including vertical guidance<sup>13</sup>. The overlay would use radius-to-fix turns in lieu of flyby

waypoints. The safety and efficiency benefits from the overlay approach would increase as RNP equipage levels increase.

#### 4. Noise Modeling Results and Population Exposure

Noise was modeled for the proposed waypoint relocation using the AEDT model described in Appendix A. Analysis was performed for the Boeing 737-800. The baseline procedure was a straight-in ILS to runway 33L at 75% of maximum takeoff weight and a 3° glideslope. The modified procedure used the same weight assumption and glideslope, varying only procedure track. The thrust profile was derived from a force-balance kinematics model.

Noise impacts from procedure recommendations 1-A1a and 1-A1b are nearly identical due to the similarity between the recommended nonprecision RNAV to the jetBlue RVFP. Figure 28 shows single-event  $L_{MAX}$  contours and population exposure reduction results for a Boeing 737-800 following procedure 1-A1a as illustrated in Figure 27. All populated landmasses fall outside of the 60 dB  $L_{MAX}$  contour for the proposed overwater procedure, with Hull being the primary noise reduction beneficiary. No communities experience an increase in noise as a result of the recommended procedure modifications.

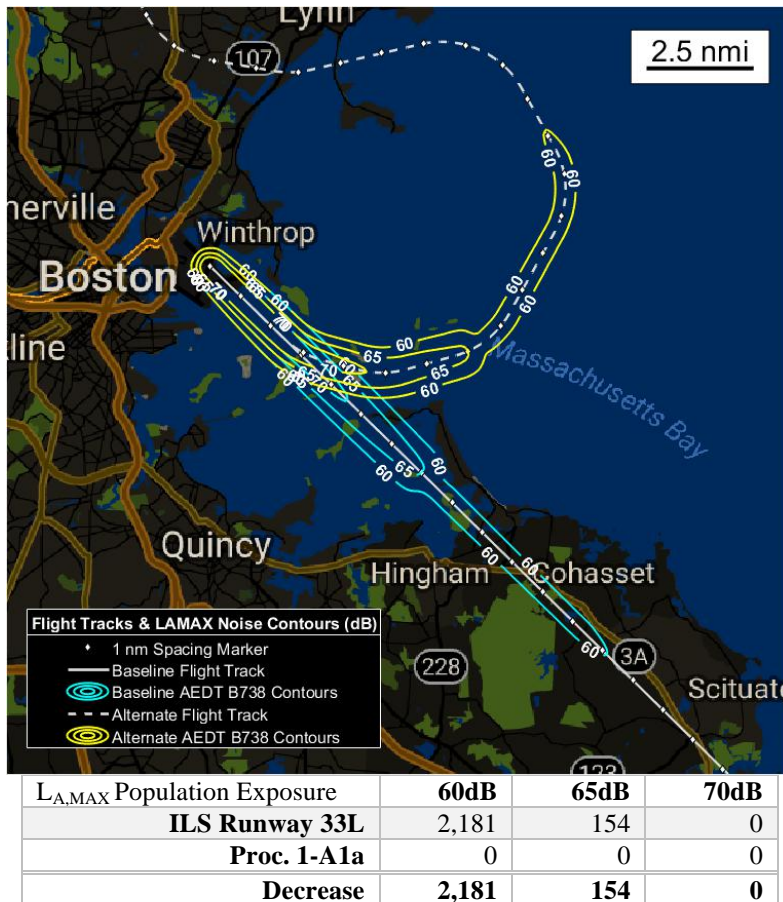


Figure 28. Noise exposure reduction for the Boeing 737-800 arriving Runway 33L descending via procedure recommendation 1-A1a on 3° descent profile compared to a straight-in ILS approach



## **5. Potential Barriers to Implementation**

### *a) Sequencing, Merging, and Spacing*

A preliminary implementation of a low-noise overwater approach procedure would likely have lower throughput than a straight-in procedure due to reduced ATC flexibility to sequence, merge, and space arrivals onto final approach. Therefore, the procedure would likely be limited to low-traffic time periods. Utilization would be focused initially on late-night periods when noise relief is most needed. Over time, improved controller experience and decision support tools may allow expanded utilization of this and similar procedures during high-traffic periods.

### *b) Vertical Guidance*

As discussed above, RNAV IAPs with vertical guidance are restricted to final approach intercept angles of 15°. RNAV IAPs without vertical guidance allow final approach intercept angles up to 30°. The 56° final approach intercept angle on the jetBlue RVFP is outside the criteria limits for both types of procedures. In order to follow the ground track of the jetBlue RVFP as closely as possible, it was necessary to design an RNAV approach without vertical guidance. A procedure designed under the criteria for RNAV with vertical guidance would not be sufficiently flexible to avoid overflight of Hull, significantly reducing potential noise benefits. Alternatively, waivers to the procedure design criteria could be considered due to the lack of obstacles on the final approach course and the operational history of the jetBlue RVFP approach.

Some aircraft are not equipped to fly RNAV approaches without vertical guidance. In addition, operators may prefer approaches with vertical guidance for operational consistency. These factors prevent universal adoption of any nonprecision RNAV procedure without vertical guidance. In order to maximize the number of aircraft following the recommended ground track to maximize noise benefits in the vicinity of Hull, an RNP overlay (including vertical guidance) should be designed for use by appropriately equipped aircraft. Operators could elect to use the nonprecision RNAV procedure or the RNP alternative depending on equipment.

## **IV. Conclusion**

The procedures identified for Block 1 include reducing climb speed for jet departures from Runways 33L and 27, modifying RNAV SID definitions for Runways 15R, 22L, and 22R, and introducing an overwater approach procedure for Runway 33L. For each of these procedures, high level objectives were provided alongside specific recommendations for implementation. Implementation is subject to FAA review and standard processes. If modifications to specific recommendations are required, the original high-level objectives should be retained to the extend possible.

The next phase of this project will involve evaluation of Block 2 procedure opportunities. These procedures will exhibit greater complexity due to potential operational and technical barriers as well as potential equity issues (noise redistribution between communities). Continued analysis and community outreach will inform the identification and development of Block 2 procedures.

# Appendix A Noise Analysis Method

## A. Noise Analysis Tools

The analysis framework used to evaluate the noise impact of current and modified arrival and departure procedures is shown in Figure 29.

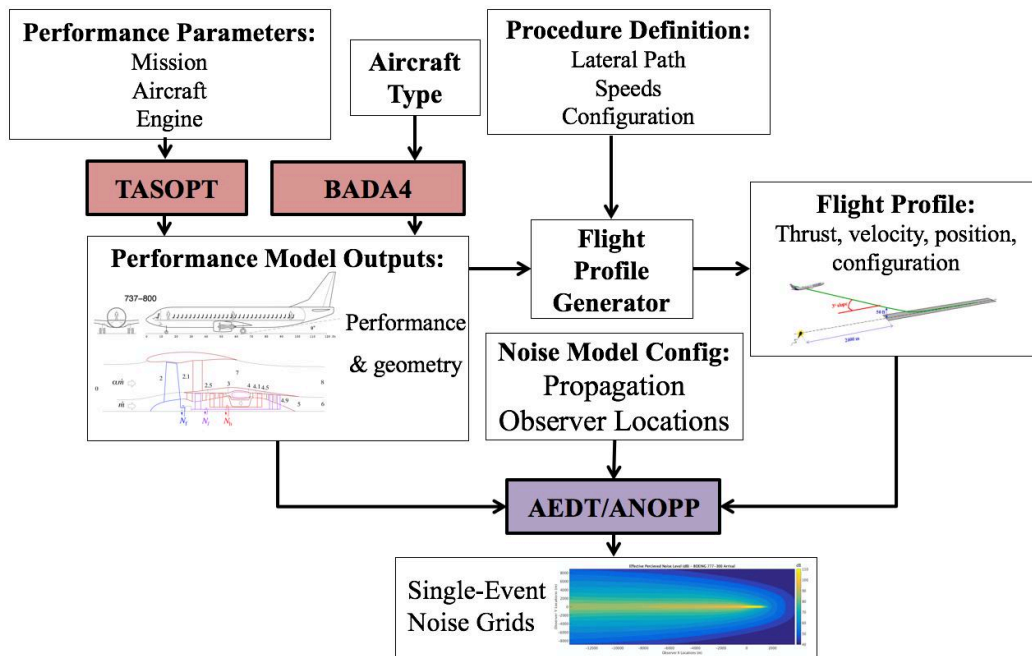


Figure 29. Integrated TASOPT and ANOPP analysis process to generate high fidelity approach and departure noise estimates

For procedures which involved only track modifications, the FAA Aviation Environmental Design Tool (AEDT) was used. AEDT uses Noise-Power-Distance (NPD) lookup tables derived from flight test and certification data and computes noise propagation through the atmosphere for a standard day. AEDT models noise referenced to a fixed airspeed (160 knot) and does not fully capture aerodynamic noise changes away from that speed.<sup>14</sup> For procedures which involved speed or configuration modifications, the NASA Aircraft Noise Prediction Program (ANOPP) was used. ANOPP was originally developed in the 1970s to provide predictive capabilities for individual aircraft studies and parametric multivariable environmental evaluations. It computes noise levels from multiple sources, both airframe and engine (fan, core, and jet), for a three-dimensional observer grid based on user-defined arrival and departure procedures.<sup>15</sup> ANOPP's source noise computations are semi-empirical, incorporating both historical noise data and physics-based acoustics models. Propagation based on a standard day atmosphere was used to obtain the ANOPP results included in this report. A series of modules take aircraft and engine parameter inputs to generate cumulative noise projections.

Both the AEDT and ANOPP noise models require aircraft performance models that provide thrust levels that are used for the noise computations. Aerodynamic drag data for each aircraft type in this study were obtained from the Eurocontrol Base of Aircraft Data (BADA-4), a

database of aircraft performance parameters obtained from aircraft manufacturers.<sup>16</sup> ANOPP also requires aircraft geometry inputs and internal engine parameters (such as internal engine stage pressures, temperatures, and mass flow rates) which were derived using the MIT Transport Aircraft System OPTimization (TASOPT) model.<sup>17</sup> TASOPT was used in this report to design aircraft matching the performance of the aircraft types presented in order to provide the detailed component performance parameters required for the ANOPP noise analysis.

Outputs from both the AEDT and ANOPP noise models are single-event noise grids. These models calculate both the maximum A-weighted sound pressure level ( $L_{A,MAX}$ ) and the Sound Exposure Level (SEL) metrics shown in this report and described in Appendix B. The grids used for both ANOPP and AEDT results shown in this report were 20nm square grids with 0.25nm spacing. Results were then re-interpolated to 0.1nm spaced grids.

## **B. Flight Trajectory Inputs**

The noise computed in both AEDT and ANOPP is dependent on the assumed flight profile, including position, altitude, and thrust. ANOPP runs also require airspeed, flap and landing gear configuration as well as engine state (a function of thrust, Mach number, and altitude). In order to obtain the flight profile data used for this study, a kinematic force-balance calculation method was used. The method was used to calculate thrust and acceleration estimates using aircraft weight, drag data from BADA-4, and detailed trajectory definitions derived from historical radar data. Fuel burn results were also calculated using BADA-4<sup>18</sup>.

Departure profiles (altitude, speed, and thrust) were generated using two methods. The first method used historical radar data from the Airport Surface Detection Equipment Model X (ASDE-X) system to identify mean altitude profiles for each aircraft type. Standard acceleration profiles were assumed from liftoff to a baseline target speed of 250 kts. Thrust levels were calculated assuming a weight of 90% MTOW using the kinematic force-balance method described above. Flap settings were configured according to aircraft-specific speed ranges provided by BADA-4. This method was used to calculate profiles for recommendations 1-D2, 1-D3, and the baseline profile for 1-D1.

Figure 30 shows results of this process for the Boeing 737-800. Figure 30(a) shows the distribution of ASDE-X altitude profiles for 20 days of Boeing 737-800 departures at BOS between January 1, 2016 and March 30, 2016. Figure 30(b) shows the velocity profile and resulting thrust profile associated with the median altitude profile from ASDE-X.

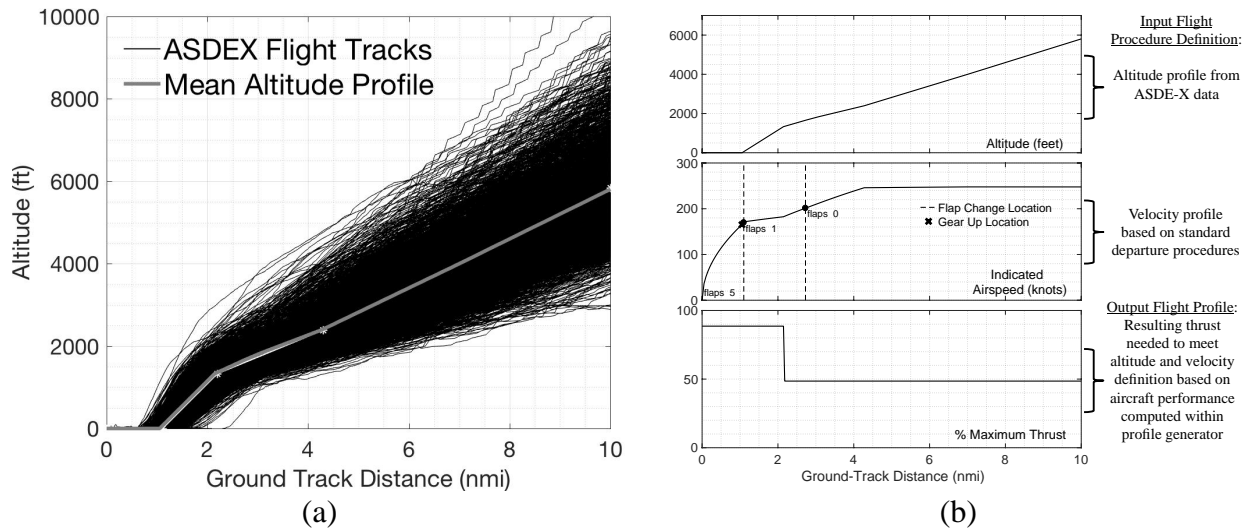


Figure 30. (a) ASDEX Boeing 737-800 altitude profiles over 20 days in 2015-2016 from all runways at BOS  
 (b) Final profile generator output matching the mean altitude profile

The second method used to derive flight profiles was by defining desired thrust, configuration, and velocity and calculating the resulting altitude profile using the force-balance kinematic method described above. Desired thrust levels can be derived from historical data, maintained at a consistent baseline profile, or modified based on noise abatement objectives. This method was used to calculate modified speed profiles for recommendations 1-D1 and all profiles for 1-A1. Figure 31 shows an example output from this method when used for evaluating recommendation 1-D1 for the Boeing 737-800.

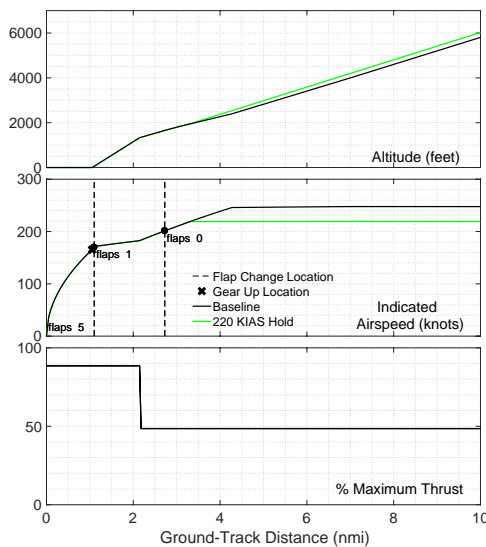


Figure 31. Flight profile generator output for a user-defined 220 kts reduced speed Boeing 737-800 departure profile compared to the standard departure profile derived from ASDEX data

### C. Population Exposure Calculations

In order to calculate population exposure at various noise levels, both noise results and demographic variables from the 2010 census data were re-gridded and compiled on a consistent 0.1nm square grid. Noise grids and population data were indexed and overlaid such that noise

impact metrics can be calculated efficiently. Figure 32 shows an example of a re-gridded population map for the Boston area, allowing for computationally efficient noise impact evaluation in that area.

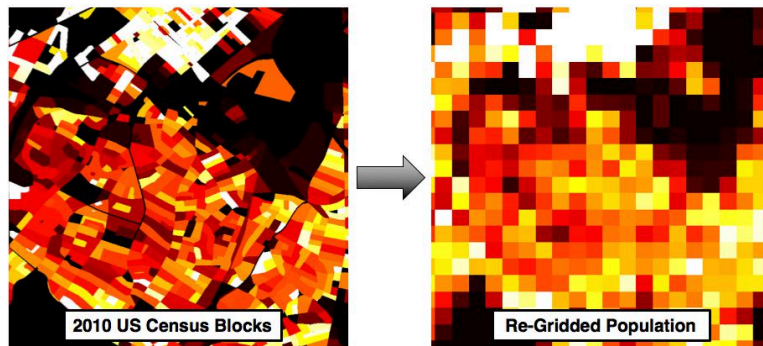


Figure 32. Re-gridded 2010 US Census data provide population data for noise impact calculations

The analysis region was a 60 nmi square grid centered on Boston Logan and is shown in Figure 33.

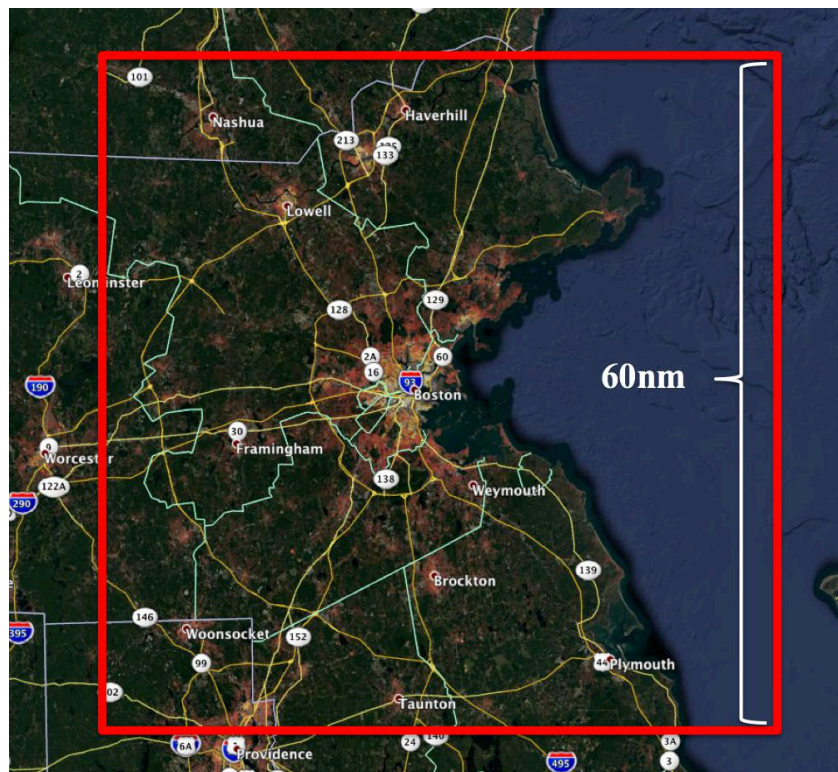


Figure 33. Geographic extent of US Census population data used for population exposure analysis

## Appendix B Noise Metrics

This study used A-weighted sound pressure level (SPL) for all analysis, an industry-standard method that emphasizes sound pressure levels in the frequency spectrum most audible to humans. Aircraft flyover events produce a characteristic rise and fall in SPL as the aircraft nears the observer, passes the point of closest approach, and recedes out of audible range. To the first order, the aircraft is only audible when the SPL rises above the background (or threshold) noise level. Figure 34 shows a typical SPL time history for a single aircraft overflight event.

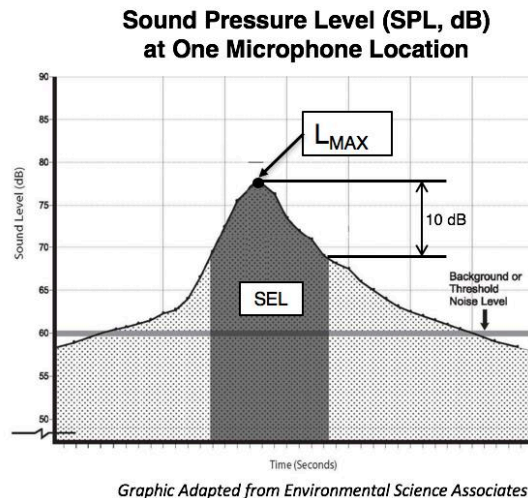


Figure 34. Sound pressure level time history at a single observer location illustrating  $L_{MAX}$  and SEL metrics

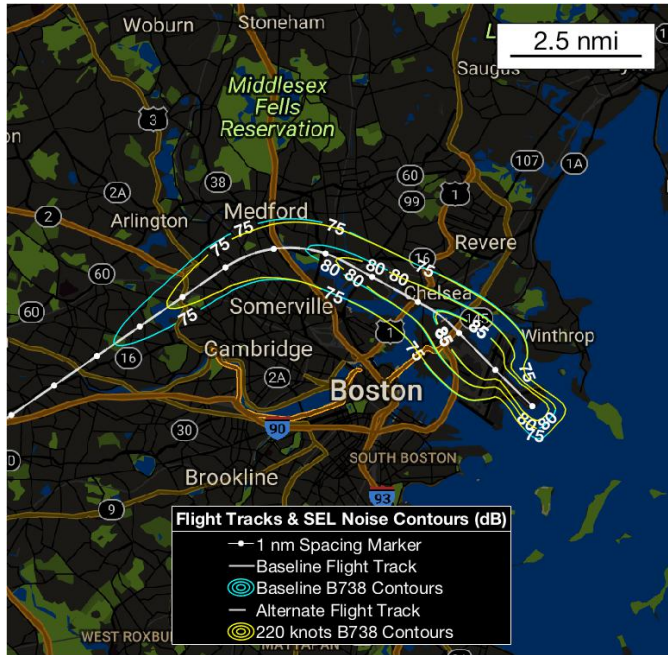
Two metrics were used in this study to evaluate single-event noise, both are illustrated in Figure 34. The primary metric used for analysis in this study was the maximum A-weighted sound pressure level ( $L_{A,MAX}$ , or simply  $L_{MAX}$  with implied A-weighting). This is an instantaneous metric that corresponds to the loudest sound level generated by an overflight without accounting for duration. In addition, the Sound Exposure level (SEL) was evaluated. SEL accounts for the duration of a noise event by integrating the total sound energy for the time during which the sound level is within 10dB of its peak. The  $L_{A,MAX}$  and SEL results were generally found to correlate for the procedures analyzed in this report. Therefore only  $L_{A,MAX}$  results are presented in the body of the report, however, SEL results are included in Appendix C for completeness.

Single-event metrics could also be used as the building block for integrated noise impact analysis. For example, the Day-Night Average Level (DNL) metric could be calculated by combining the constituent SEL measurements for a set location over an average annual day of operations with a 10 dB penalty factor for night-time operations. The number of flights above a set level ( $N_{ABOVE}$ , a standard metric for frequency of audible events) is calculated by summing the number of operations above a desired threshold level, typically 70dB during the day and 60dB at night. Detailed flight schedule and procedure utilization and aircraft type allocation assumptions are required to calculate integrated metrics such as  $N_{ABOVE}$  and DNL, which can mask flight-level noise reduction efforts. Therefore, this report focuses on single event metrics but presents results that are appropriate constituent elements of broader integrated analysis.



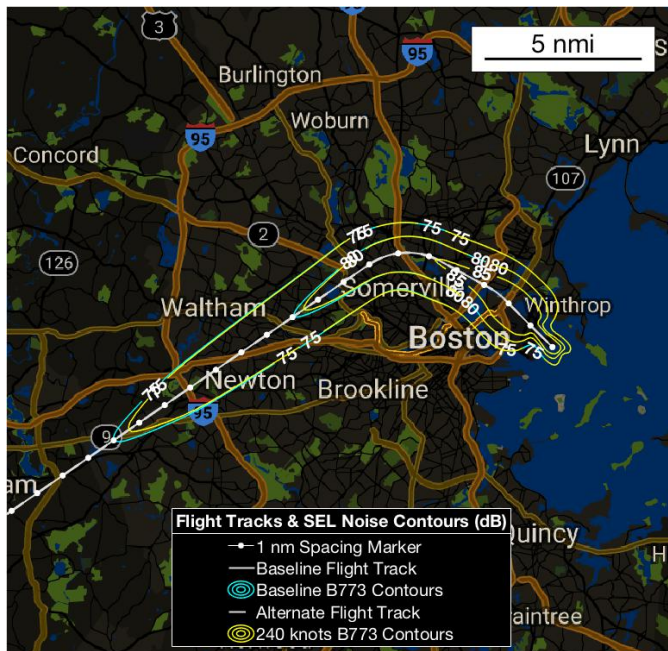
## Appendix C SEL Noise Results

The noise results presented elsewhere in this report are presented in the  $L_{MAX}$  metric because  $L_{MAX}$  and SEL results were consistent. For completeness, SEL results for each of the recommended procedures are provided below. SEL accounts for the duration of a noise event by integrating the total sound energy for the time during which the sound level is within 10dB of its peak. This metric is useful as a building block for calculating integrated metrics such as DNL.



**Procedure Recommendation:** 1-D1  
**Noise Model:** ANOPP  
**Aircraft:** Boeing 737-800

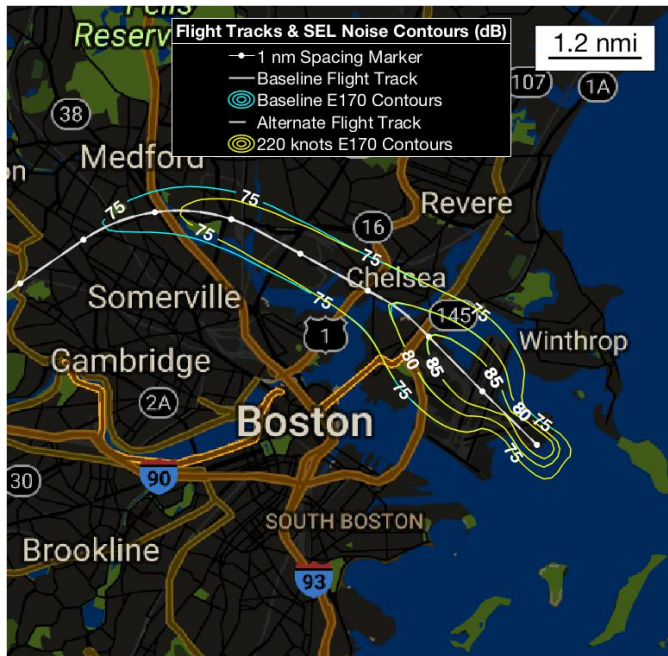
SEL Exposure	75dB	80dB	85dB
Baseline	119,289	31,960	4,169
Proc. 1-D1	102,764	27,953	3,860
Decrease	16,525	4,007	309



**Procedure Recommendation:** 1-D1  
**Noise Model:** ANOPP  
**Aircraft:** Boeing 777-300

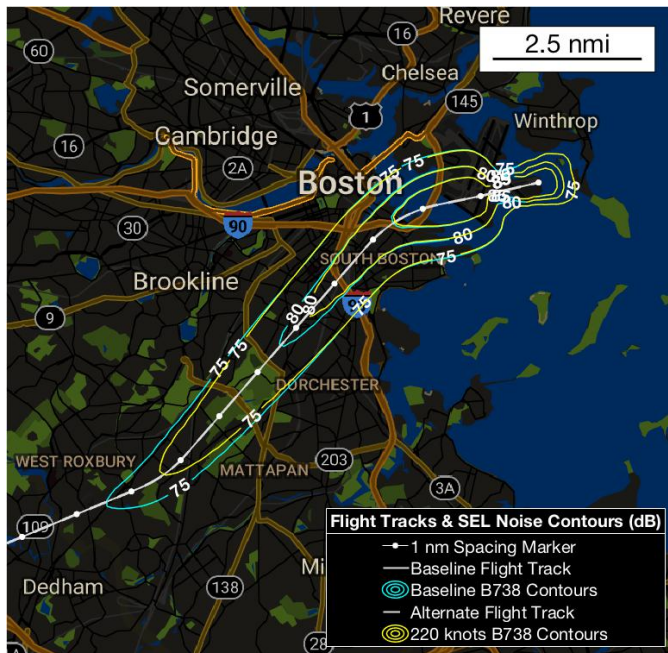
SEL Exposure	75dB	80dB	85dB
Baseline	350,349	126,925	38,314
Proc. 1-D1	346,061	122,713	38,314
Decrease	4,288	4,212	0





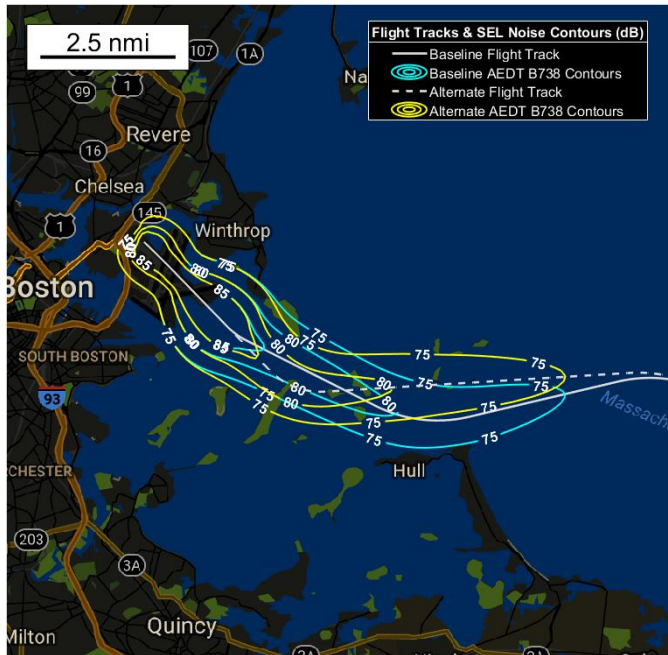
**Procedure Recommendation:** 1-D1  
**Noise Model:** ANOPP  
**Aircraft:** Embraer 170

SEL Exposure	75dB	80dB	85dB
Baseline	50,360	5,757	0
Proc. 1-D1	42,119	5,757	0
Decrease	8,241	0	0



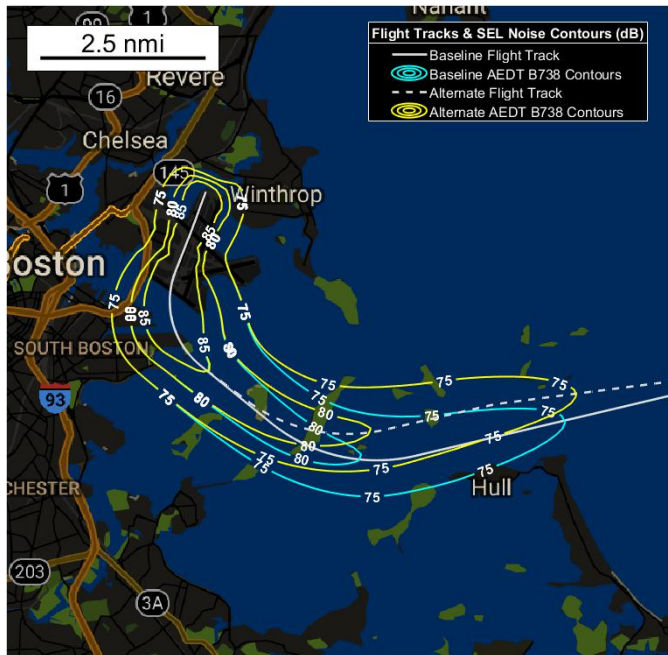
**Procedure Recommendation:** 1-D1  
**Noise Model:** ANOPP  
**Aircraft:** Boeing 737-800

SEL Exposure	75dB	80dB	85dB
Baseline	118,503	18,018	540
Proc. 1-D1	100,990	11,799	540
Decrease	17,513	6,219	0



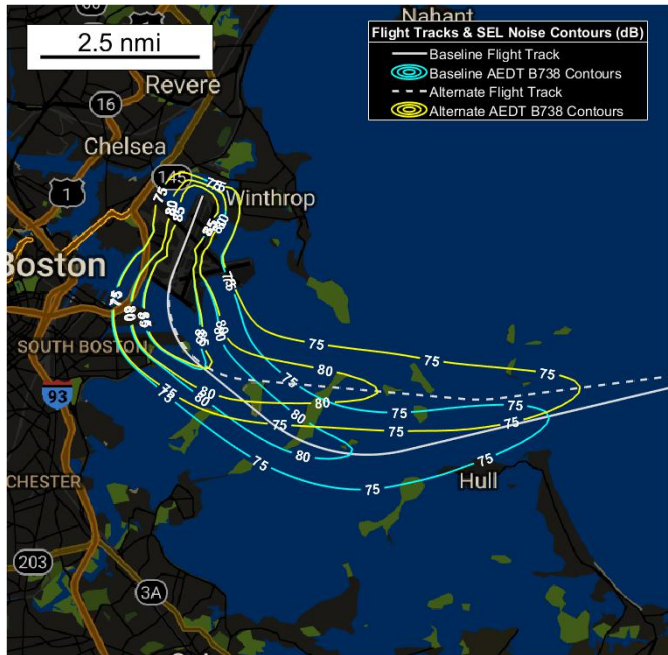
**Procedure Recommendation: 1-D2**  
**Noise Model: AEDT**  
**Aircraft: Boeing 737-800**

SEL Exposure	75dB	80dB	85dB
Baseline	6,081	223	0
Proc. 1-D2	5,721	223	0
<b>Decrease</b>	<b>360</b>	<b>0</b>	<b>0</b>



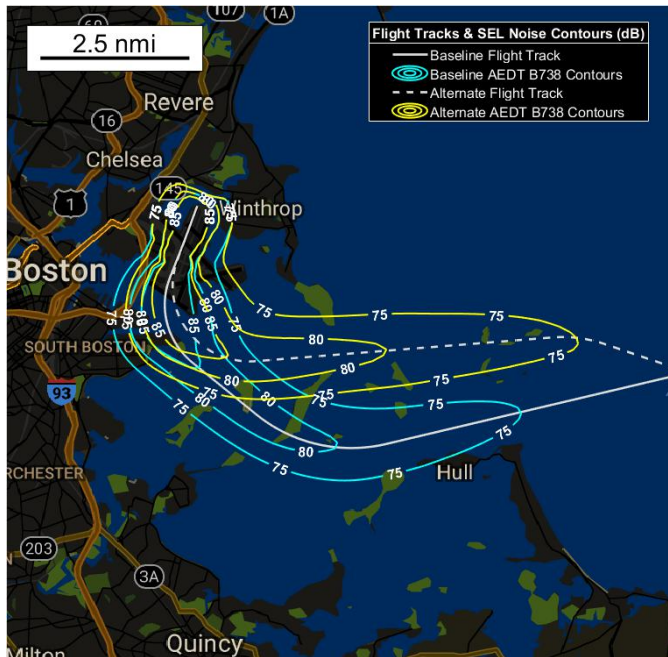
**Procedure Recommendation: 1-D3a**  
**Noise Model: AEDT**  
**Aircraft: Boeing 737-800**

SEL Exposure	75dB	80dB	85dB
Baseline	1,249	2	0
Proc. 1-D3a	1,204	2	0
<b>Decrease</b>	<b>45</b>	<b>0</b>	<b>0</b>



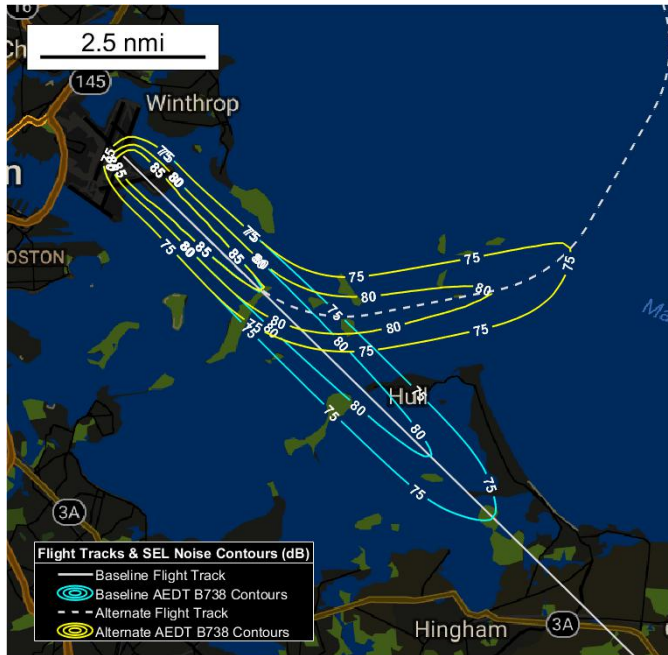
**Procedure Recommendation:** 1-D3b  
**Noise Model:** AEDT  
**Aircraft:** Boeing 737-800

SEL Exposure	75dB	80dB	85dB
Baseline	6,081	223	0
Proc. 1-D3b	5,031	211	0
<b>Decrease</b>	<b>1,050</b>	<b>12</b>	<b>0</b>



**Procedure Recommendation:** 1-D3c  
**Noise Model:** AEDT  
**Aircraft:** Boeing 737-800

SEL Exposure	75dB	80dB	85dB
Baseline	6,081	223	0
Proc. 1-D3c	2,841	14	0
<b>Decrease</b>	<b>3,240</b>	<b>209</b>	<b>0</b>



**Procedure Recommendation:** 1-A1a  
**Noise Model:** AEDT  
**Aircraft:** Boeing 737-800

SEL Exposure	75dB	80dB	85dB
ILS	815	45	0
Proc. 1-A1a	0	0	0
<b>Decrease</b>	<b>815</b>	<b>45</b>	<b>0</b>



# Appendix D Flight Track Density Plots

Flight track concentration plots are provided below for jet arrivals and departures at each major runway at BOS. For each runway, track density is shown for 2010 and 2015 to provide a comparison of before and after RNAV implementation.

## A. Runway 4R Arrivals

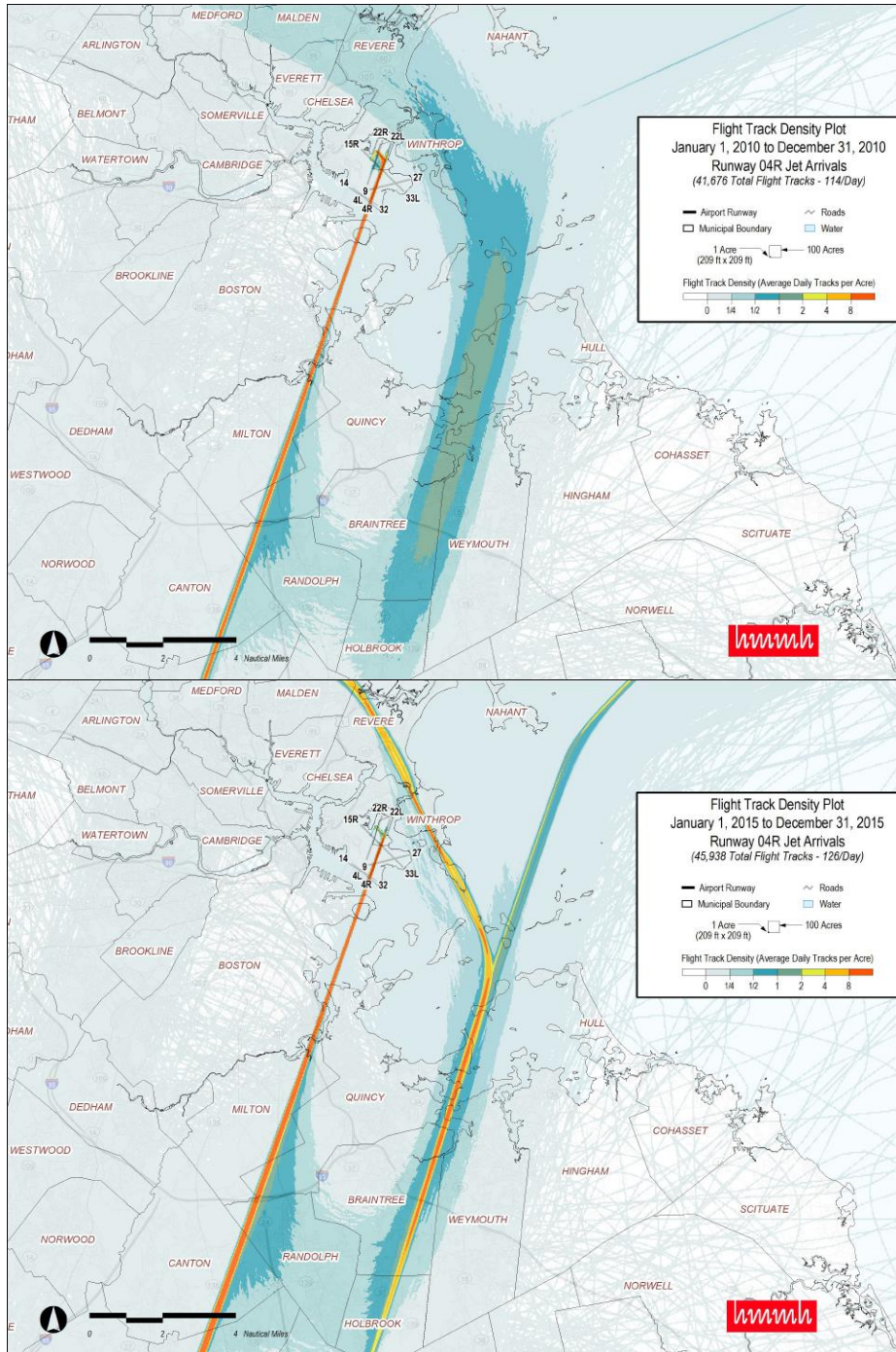


Figure 35. Runway 4R arrival flight track densities from 2010 (top) and 2015 (bottom)





### C. Runway 4L Arrivals

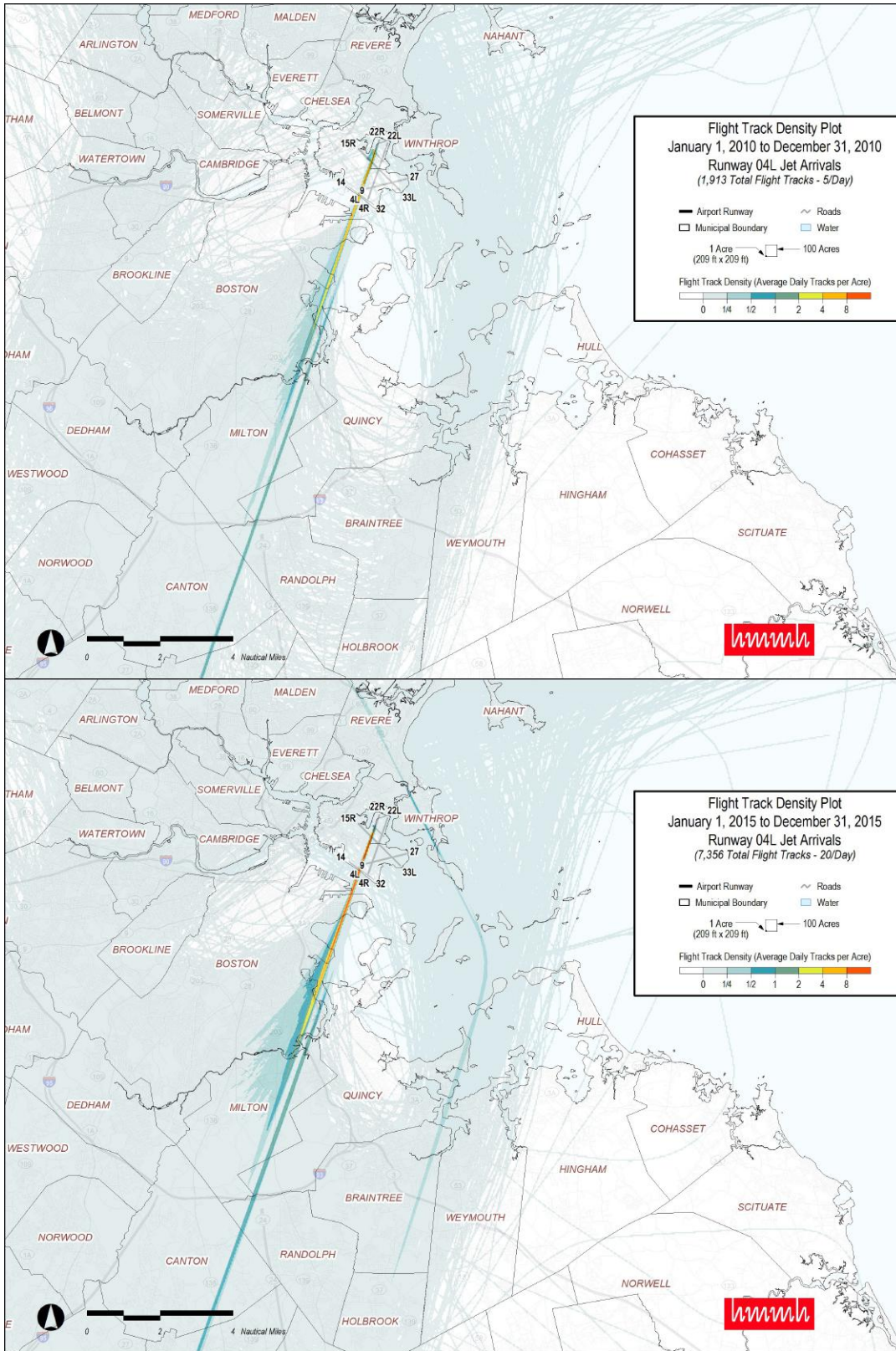


Figure 37. Runway 4L arrival flight track densities from 2010 (top) and 2015 (bottom)



## D. Runway 9 Departures

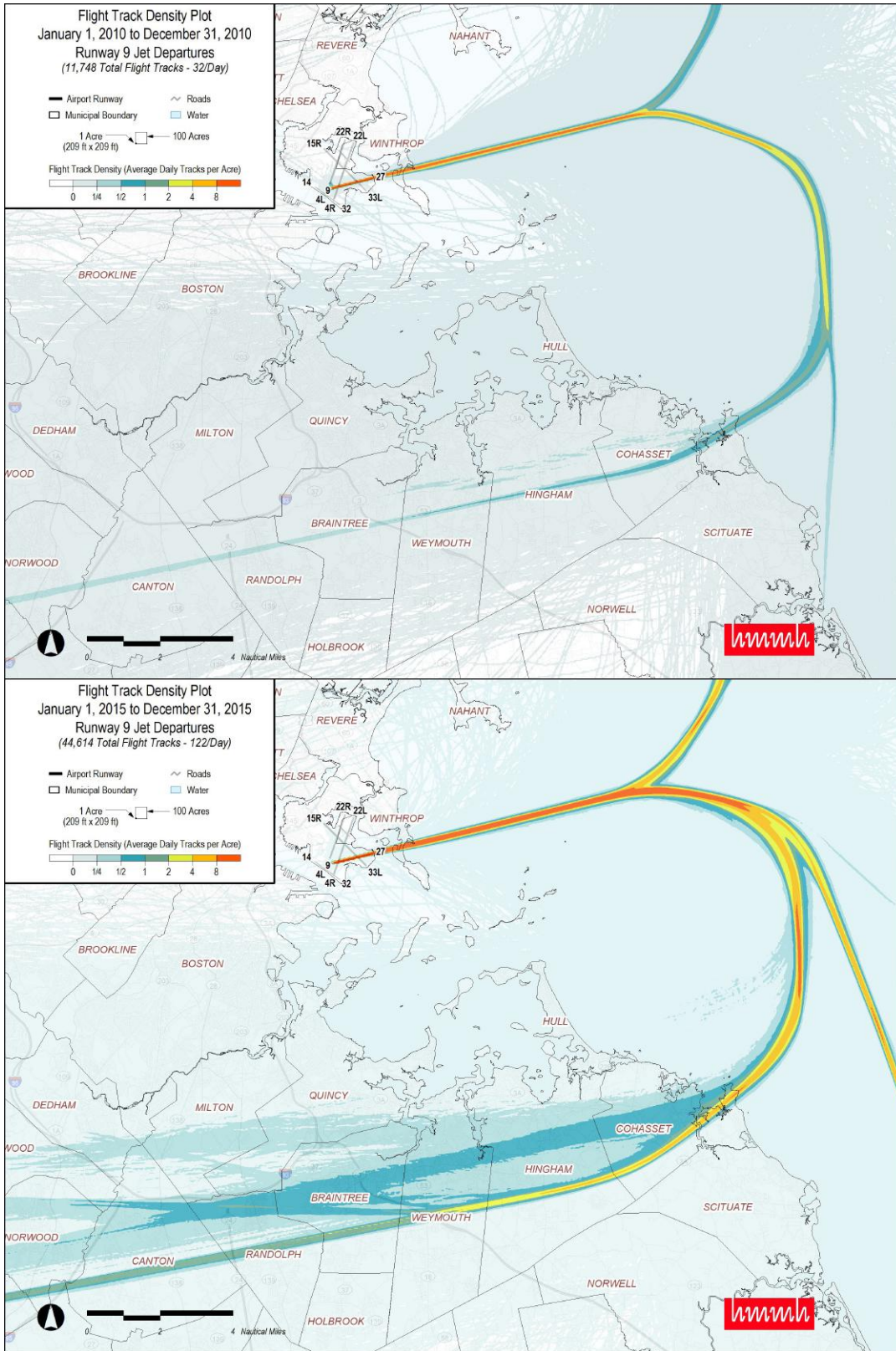


Figure 38. Runway 9 departure flight track densities from 2010 (top) and 2015 (bottom)

## E. Runway 15R Arrivals

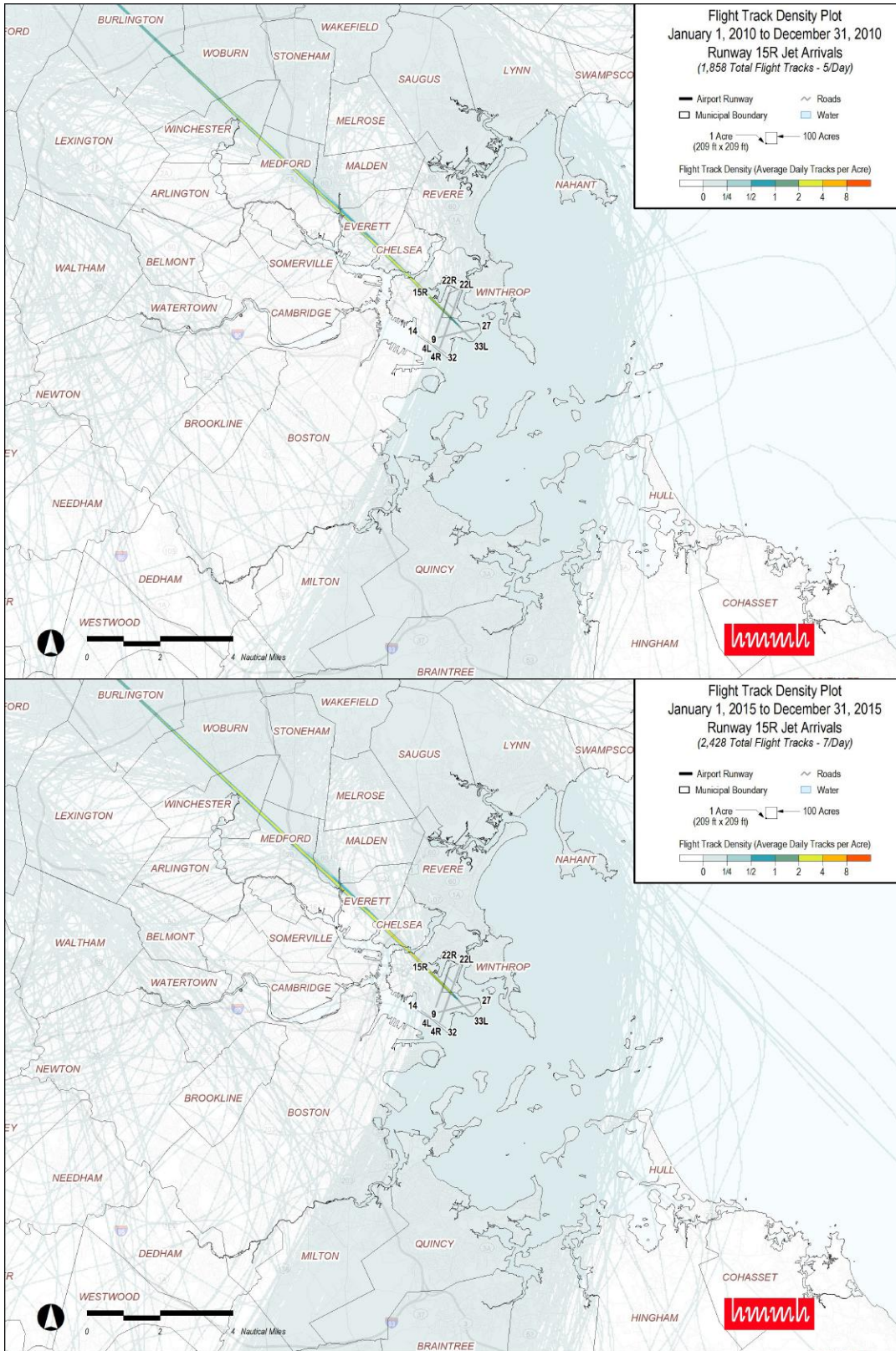


Figure 39. Runway 15R arrival flight track densities from 2010 (top) and 2015 (bottom)



## F. Runway 15R Departures

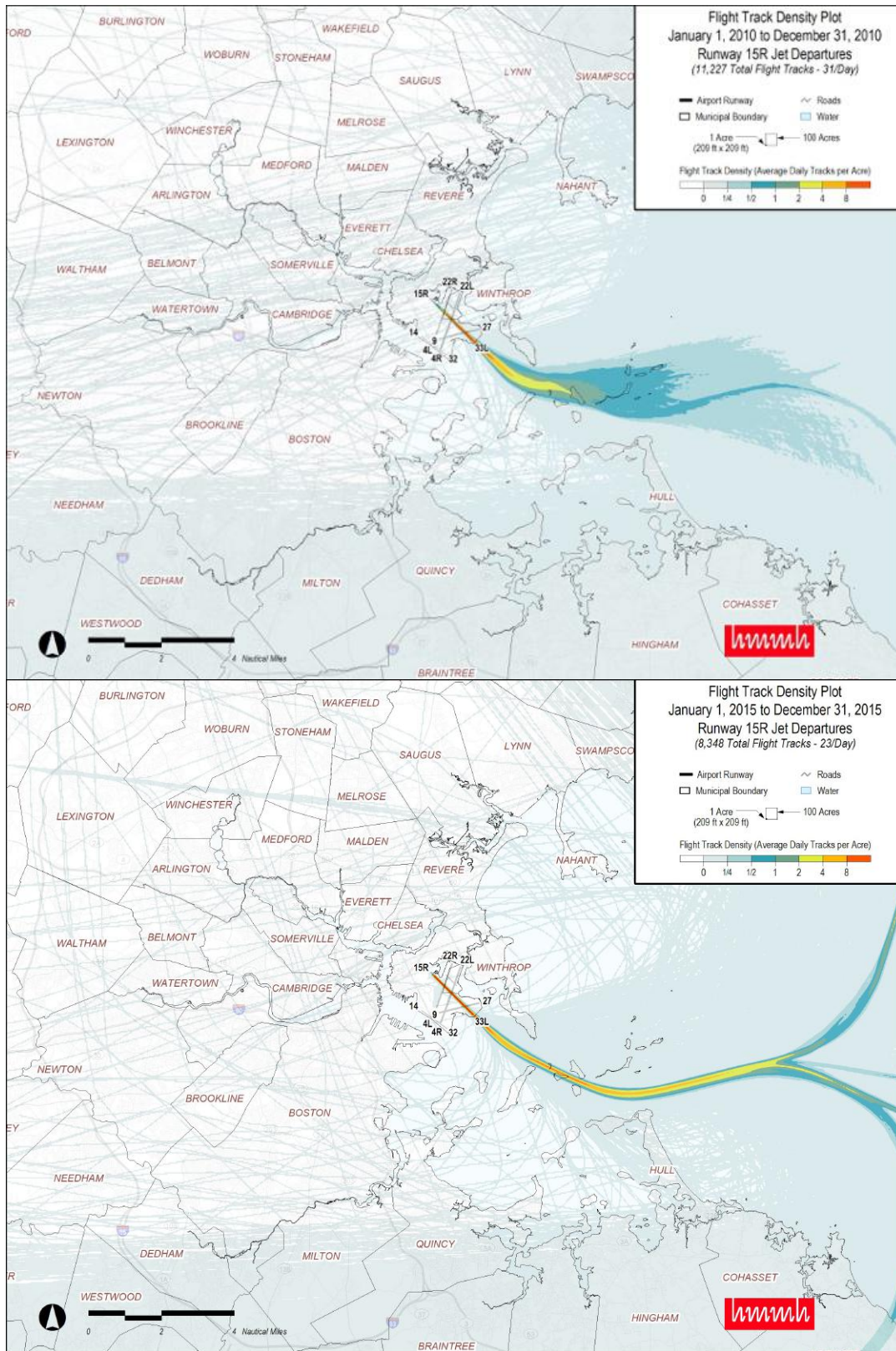


Figure 40. Runway 15R departure flight track densities from 2010 (top) and 2015 (bottom)

## G. Runway 22L Arrivals

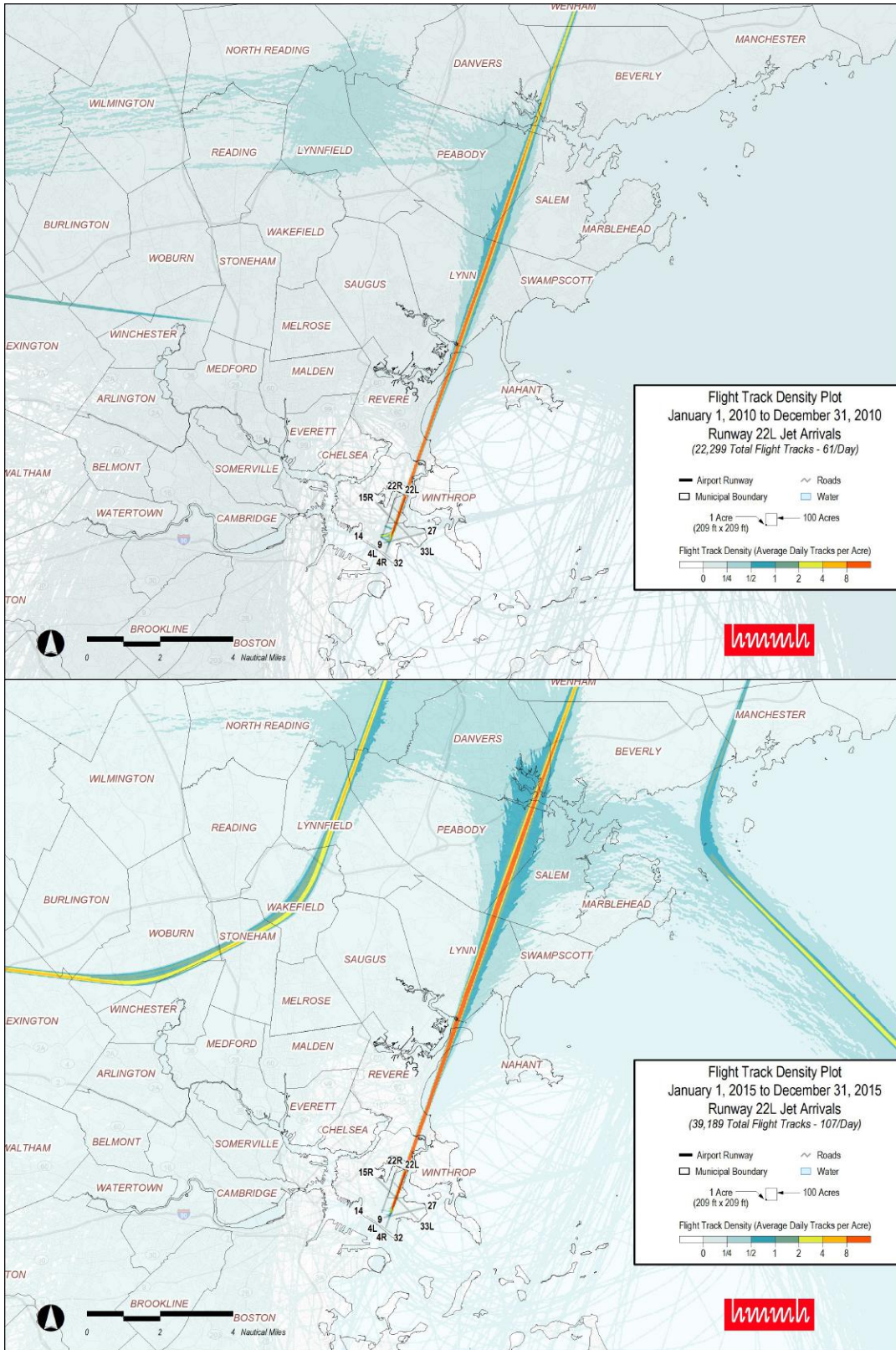


Figure 41. Runway 22L arrival flight track densities from 2010 (top) and 2015 (bottom)



## H. Runway 22R Departures

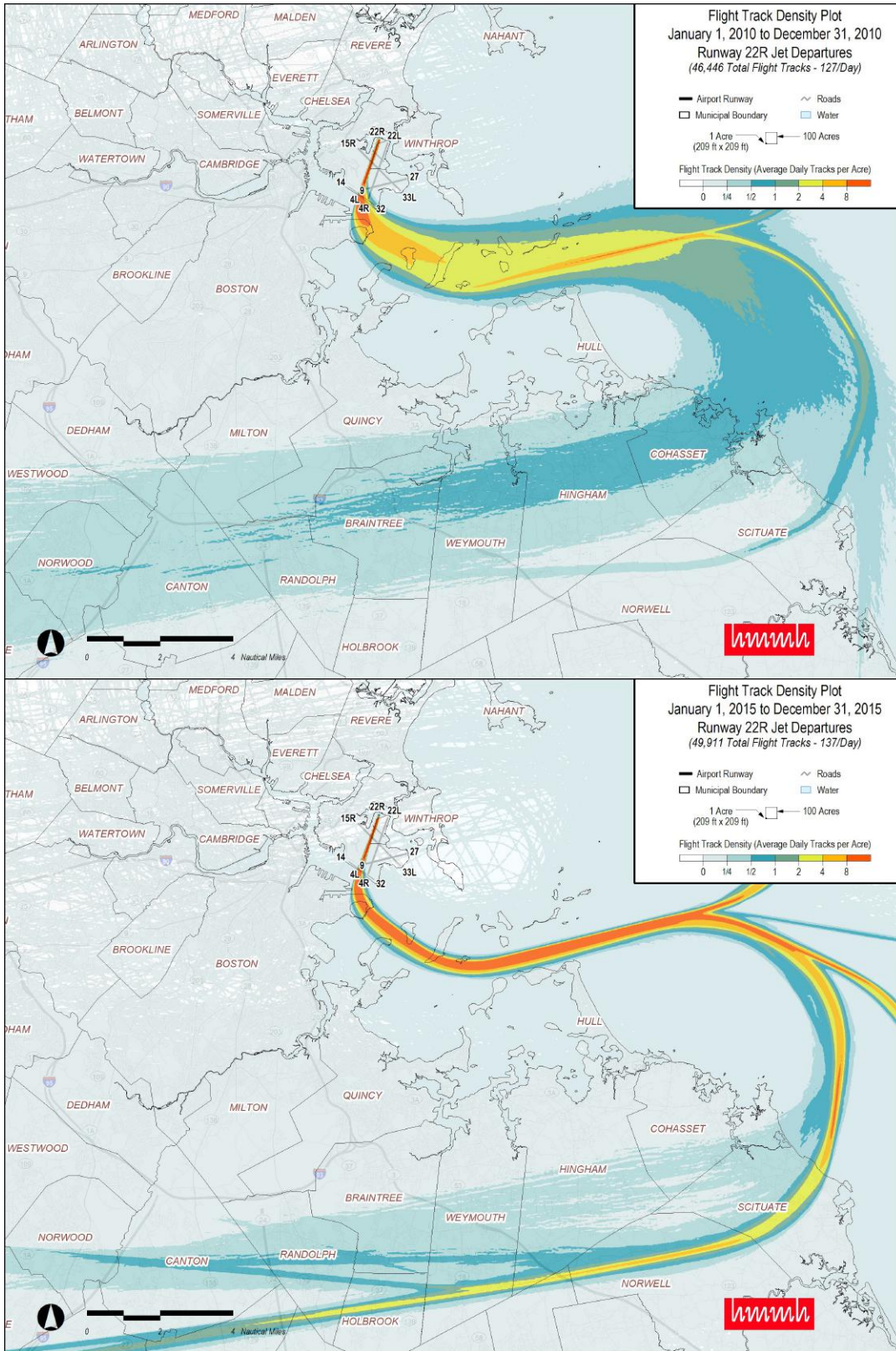


Figure 42. Runway 22R departure flight track densities from 2010 (top) and 2015 (bottom)

# I. Runway 27 Arrivals

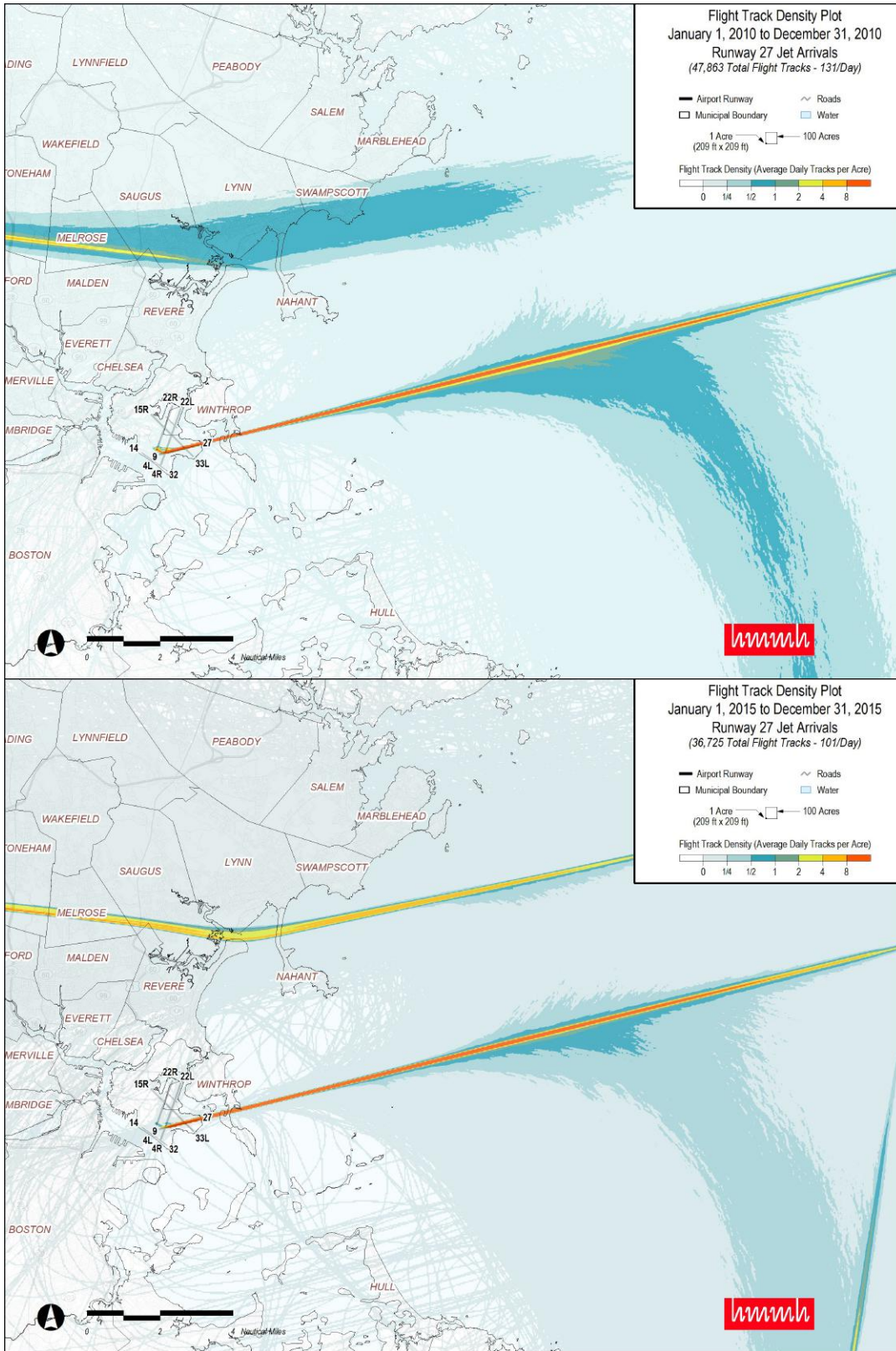


Figure 43. Runway 27 arrival flight track densities from 2010 (top) and 2015 (bottom)



## J. Runway 27 Departures

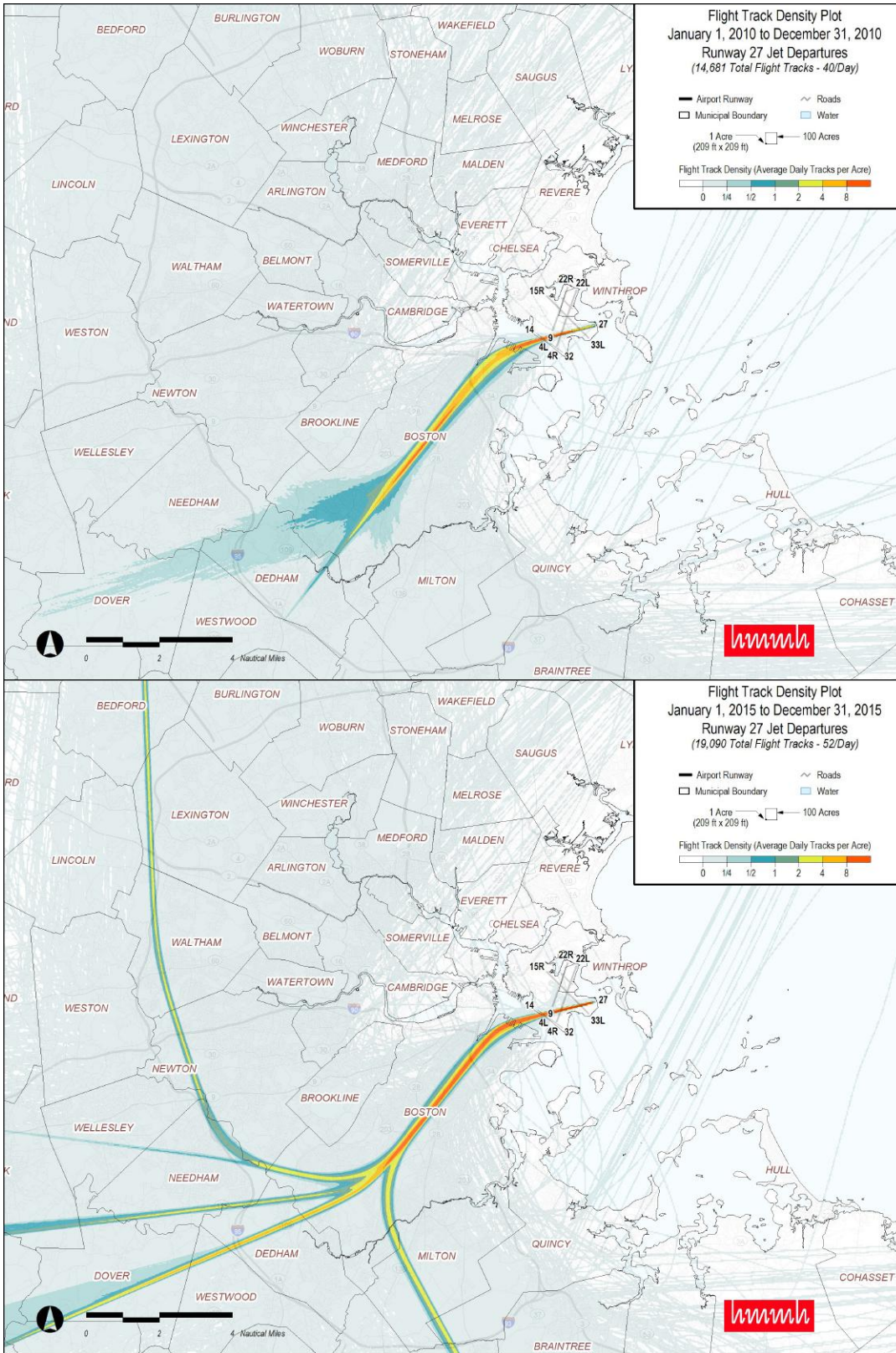


Figure 44. Runway 27 departure flight track densities from 2010 (top) and 2015 (bottom)



## K. Runway 32 Arrivals

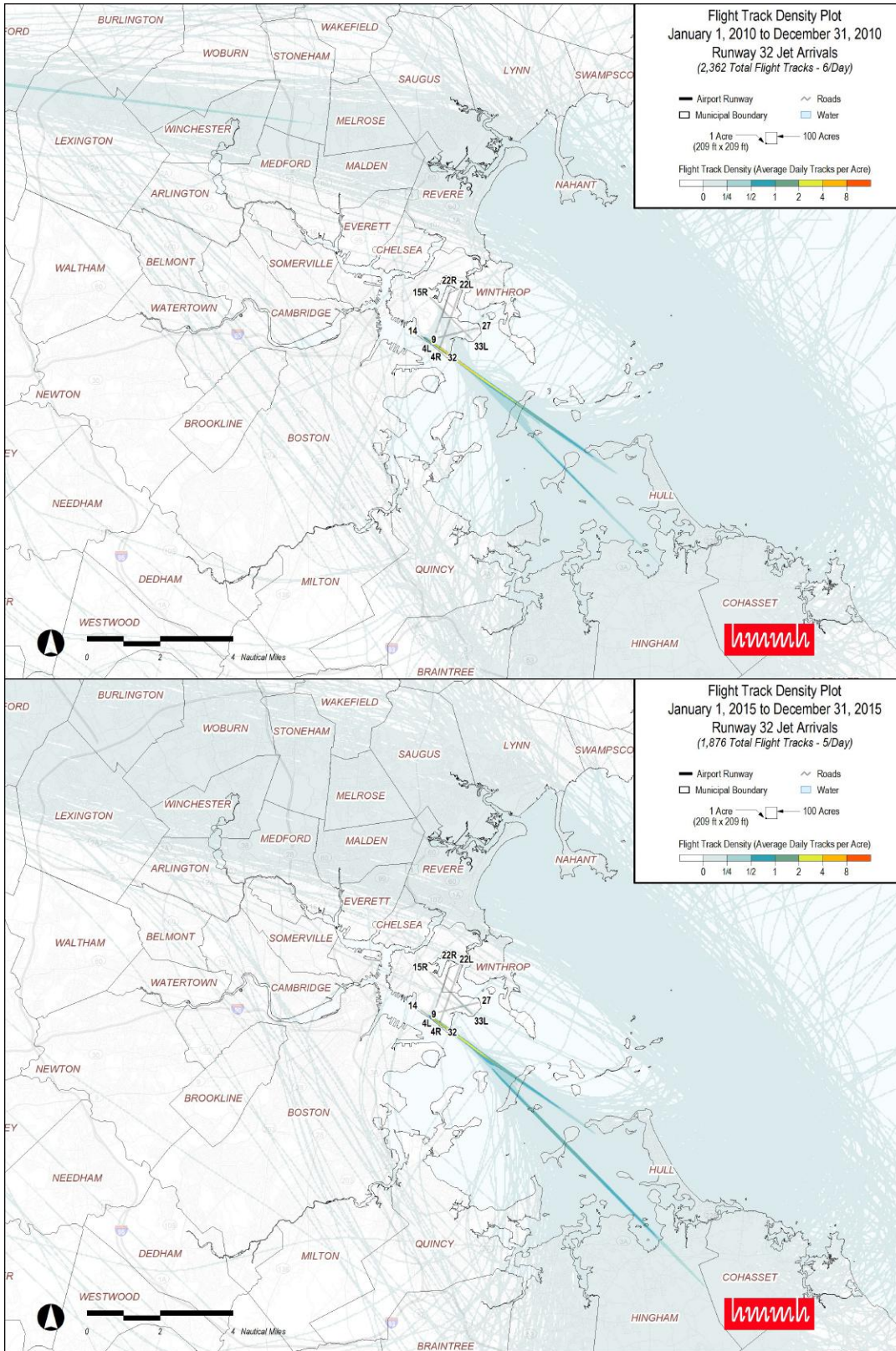


Figure 45. Runway 32 arrival flight track densities from 2010 (top) and 2015 (bottom)



## L. Runway 33L Arrivals

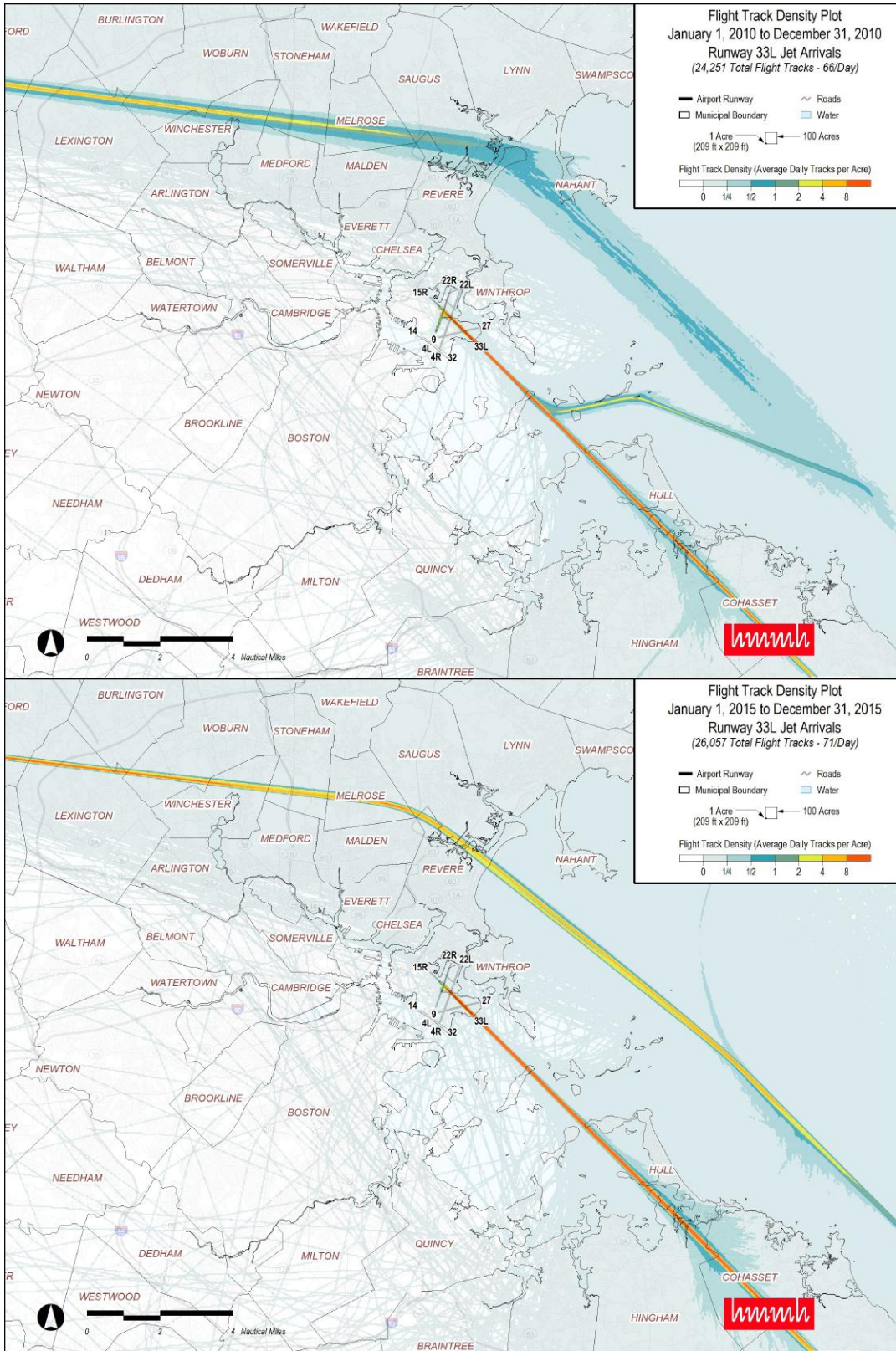


Figure 46. Runway 33L arrival flight track densities from 2010 (top) and 2015 (bottom)



## M. Runway 33L Departures

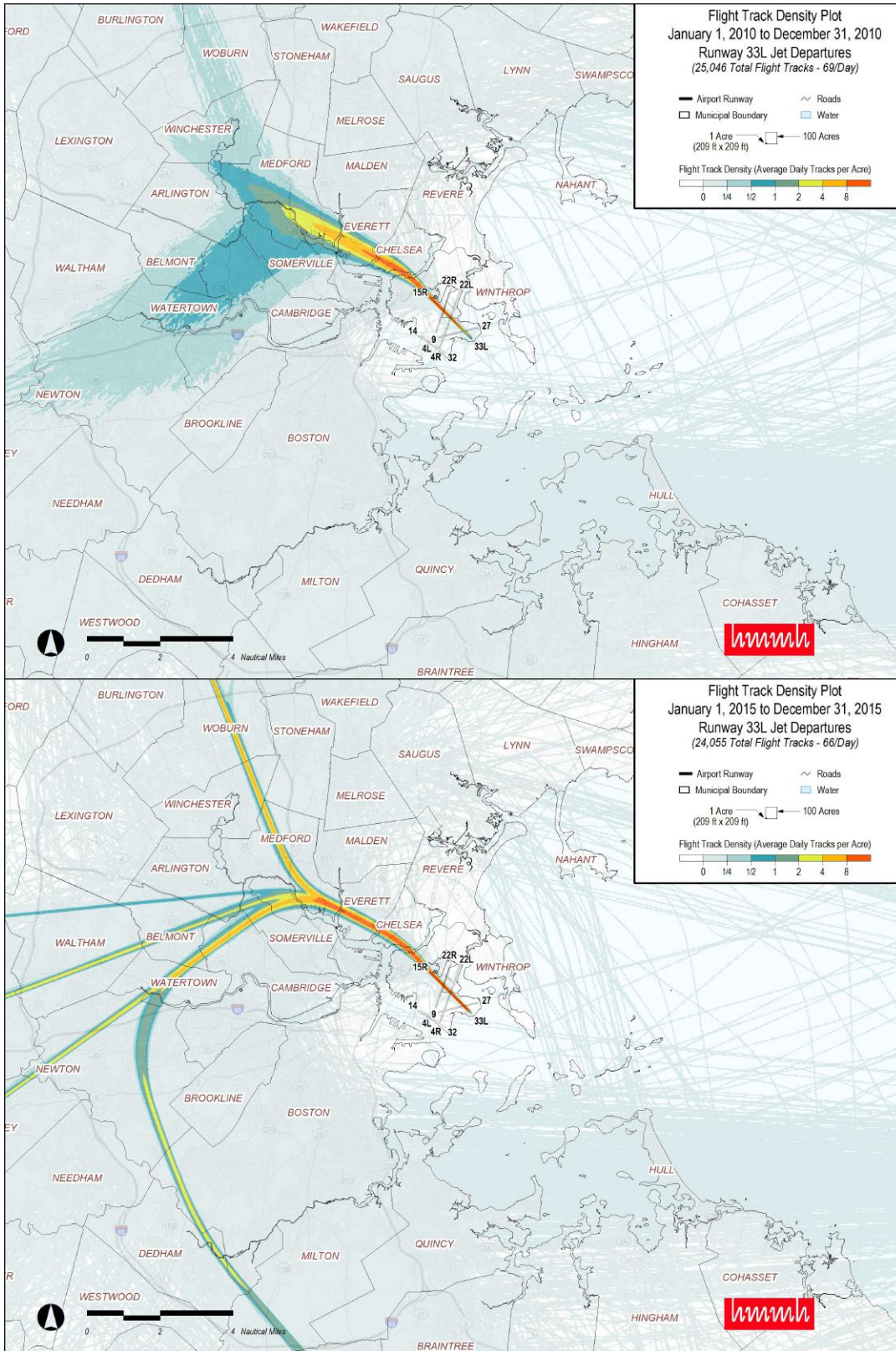


Figure 47. Runway 33L departure flight track densities from 2010 (top) and 2015 (bottom)

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