Sizing the X-ray Spectral Resolution Limits of the REgolith X-ray Imaging Spectrometer (REXIS) Instrument at Asteroid 1999RQ36

By

Suzanna Megyery

Submitted to the Department of Earth, Atmospheric and Planetary Science in Partial Fulfillment of the Requirements for the Degree of

> BACHELOR OF **SCIENCE** in Earth, Atmospheric and Planetary Sciences at the

MASSACHUSETTS INSTITUTE OF **TECHNOLOGY**

January 17, 2013 **E** February 2013]

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OSIRIS-REx Asteroid Sample Return Mission

NASA Goddard Space Flight Center Lockheed Martin University of Arizona Massachusetts Institute of Technology Harvard University

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Massachusetts Institute of Technology

Department of Earth, Atmospheric **&** Planetary Sciences Kavli Center for Astrophysics Department of Aeronautics **&** Astronautics Space Systems Laboratory

In Conjunction with: Harvard Center for Astrophysics

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Dated: **2013.01.17**

Author:

Suzanna Megyery Massachusetts Institute of Technology *zsuzsa@mit.edu* Department of Earth, Atmospheric **&** Planetary Sciences

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Abstract

The REgolith X-ray Imaging Spectrometer (REXIS), a Charge-Coupled Device (CCD)-based coded aperture soft X-ray **(0.3-7.5** keV) telescope for remote geochemical X-ray Fluorescence (XRF) spectrometry, will be flying on board the Origins Spectral Interpretations Resource Identification Security Regolith Explorer (OSIRIS-REx) asteroid sample return mission that will be visiting the asteroid **1999 RQ36** and sending a sample back to Earth. REXIS will detect elemental XRF lines and produce a histogram of results as the spacecraft orbits the asteroid as well as produce a global map of elemental abundance ratios. The accuracy requirement for measuring the global ratios of elements and the spectral resolution requirement for discriminating unique XRF lines from each other have been set in place.

The correct interpretation of X-ray measurements from the surface of **1999 RQ36** is limited **by** properties that are intrinsic to the **CCD** detector, CCID-41, that has been chosen for REXIS. This thesis study outlines student experimentation and results that were conducted on the CCID-41 detector to gauge the intrinsic detector noise as a function of detector temperature. Further, the widening of spectral lines on the resultant histogram was also equated as a function of detector temperature.

Members of the REXIS Team built a spectral resolution model to investigate both the widening of spectral lines as a function of detector temperature and the accuracy of the measurement of elemental abundance line ratios as a function of detector temperature. Data from the student laboratory experimentation suggested that the detector temperature remain at or below **-75'C** to minimize intrinsic noise properties. Data from the computational analyses of the spectral resolution model suggest consistent results that the detector temperature remains at or below **-550C** to remain within the established REXIS requirements.

The combination of these three results leads to the author's recommendation that a detector temperature requirement be set that the temperature of the **CCD** detector onboard REXIS shall not exceed **-55'C** and that a detector temperature goal be set that the temperature of the **CCD** detector shall not exceed -75°C.

Acknowledgments

For the completion of my undergraduate journey through MIT and this thesis **I** am grateful to my advisors and supervisor Dr. Sara Seager and Dr. Richard Binzel.

Dr. Seager has been an incredible advisor throughout the last two years. She is a source of inspiration and motivation, always challenging me to learn new things **-** professionally and personally **-** and to apply the information **I** already know. She has given me the confidence to continue the degree and even shared personal examples of innovation, insight and grit to help my studies.

Dr. Binzel has been an asteroid to rely on; **I** am grateful for his advice and patience, his direction regarding the work on the REXIS project and his capacity for an objective view of every problem always providing guidance and believing in me.

^Iwould like to acknowledge with gratitude the assistance and support of Niraj Inamdar, Matthew Adendorff, Jane Connor, the MIT/Harvard REXIS Team, members of the OSIRIS-REx mission, and members of the MIT Earth, Atmospheric **&** Planetary Science Department.

For the unconditional belief and support I received from Dr. Susan Mausshardt, Dr.'s Carolyn & Adam Conklin, Dean David Randall, Dr. **Silk** and Ms. Barbara Lechner **I will** forever be grateful.

I thank MIT, the Linda Gronlund Fund and Mr. **&** Mrs. Dan and Marian Martin for their financial support.

Biographical Sketch

The choices of my life have given me a unique set of opportunities and experiences but have brought me a most interesting set of challenges, all of which have affected the person **I** am and am capable of becoming.

In 2004 I was a busy, complicated and misaligned young girl who wrote a profound personal statement and was admitted to MIT. **I** will safely say that the unbridled optimism of youth combined with the allure of a degree from MIT is a dangerous combination. In **2005** I came to MIT after seventeen years of reinforcement of bad behaviors. The challenges that ensued were not just about trying to separate the past from the present but about severing whatever fictitious tie was keeping me anchored to a chaotic family environment, continually reminding me that no matter what **I** attempted **I** couldn't measure up **-** handicapped **by** a y-chromosome. Confronting the demands of the scholastic environment at MIT, I did the only thing I knew how to do at that point... **I** ran.

My childhood raised me with mechanical skills **-** machining, welding, auto mechanics **-** languages, martial arts and combative survival as well as a mental and emotional strength that can only be conditioned growing up with an ex-KGB agent as a father and a subservient mother. Despite our strained relationship, in part, my mission is a result of those two people who created me and built me: Mr. Sándor Alex Megyery and Mrs. Erzsébet Megyery.

In **2008** I left MIT to discover and erect my own definitions of peace, clarity, integrity and purpose. After several months of a kinesthetically-focused education, with everything from kayaking, surfing and mountaineering, **I** reevaluated my situation.

With the help of older, and what I am sure are wiser eyes, **I** established a set of goals that are both worthy and achievable in improving the future, not only for myself, but, also for the world we live in. In furtherance of that goal, **I** relocated and attended the University of Washington and targeted employment, to challenge and improve myself and to apply the discipline that **I** achieved to teach and train others. The planning and execution of the steps required to return to MIT would not have been possible without the support of dear friends Heidi Sitton and Alan Heddings.

In 2011 **1** returned to MIT with focus, equilibrium, anticipation and integrity. The two-year journey upon my return has become a locust of vectors all charged with the energy and destination of fulfillment. The alignment that ensued within myself has been rapid and extraordinary. **My** adventures since my return to MIT have continued to be varied, and those journeys have helped me to find center, purpose and home. Not long after my return, the MIT Earth Resources Laboratory took a team of students on a geophysics groundwater project to St Lucia and for the first time in my life **I** felt that **I** had come home.

The challenges of how **I** grew up have given me a deep connection to the people and culture of countries that are less technologically developed than the United States. **My** connection to St Lucia and the West Indies is profound and intimate. For the summer of 2012 I was awarded an MIT Public Service Center Project Fellowship Award as well as Undergraduate Research Opportunities Program funding to support my project in water security, public health and safety, working alongside the Caribbean Environmental Health Institute **(CEHI)** in St Lucia. For four months **I** lived and

worked on-island and found family in community through my coworkers at **CEHI** and a local cycling charity group called Project Breakaway. Every member of these teams was an inspiration, a friend, and a source of encouragement for my return to MIT to complete my studies and to return to St Lucia someday soon.

^Iwould like to exclusively thank MIT Professor Frank Dale Morgan for taking a chance on me and supporting my ideas when most everything inside of him suggested otherwise. From my work with **CEHI I** would like to extend my appreciation to Dr. Chris Cox and Ms. Shermaine Clauzel. These two worked closely with me for the entirety of my project and believed in me even when **I** was battling Dengue Fever; they also saw first-hand just how St Lucian **I** had become. This wonderful family reminded me that we work to live rather than live to work and cautioned me to avoid the trap of spending fifteen years building a great career, only to throw it in one day upon realizing that **I** had juggled life and work to make everyone happy **-** but myself.

The Project Breakaway group rides with a purpose and taught me that having family makes your life richer; in that, family can be found anywhere, and because of them **I** shared a fulfilled life in St Lucia. **I** would like to thank all the members of Project Breakaway for reminding me that there is a different way to live and the pursuit of such happiness is the chase of a lifetime. **I** would herein like to include: Mr. Ernest Ottley and Mrs. Brenda Ottley, Mr. Ross Stevenson, Mr. Raymond Volney, Kurt and Kirk Miraj, Richard Monplaisir, Jason Tameh, Roderick Cherry; Andrew, Bev and Gillan Moses, Chandoy Moses, Fidel Mangal, Oliver Cadet, Mr. L. Ryan Joseph and additional members of Project Breakaway.

Upon my return to MIT from St Lucia it was time to complete my thesis work and not simply put a check in a box. **I** have been a part of the MIT REXIS Instrument team as a student scientist since September of 2011. This project has given me the perfect opportunity to combine my varied undergraduate studies that span earth sciences, space sciences, engineering, and physics into a concrete and valuable contribution.

^Iam grateful for the challenge **I** have been privileged to be a part of: the MIT undergraduate experience.

^Ihave always told myself **I** can pass any test a human being can pass; **I** plan to prove it.

Dedication

Words cannot express the gratitude **I** have for the following individuals who have been and continue to be my best friends and my family.

I am grateful for the never-ending love and encouragement of **my** brother and best friend, **my** comrade, with whom I survived the war and torture of our childhood, Mr. Sindor (Alex) Megyery, and his incredibly supportive wife, Mrs. Alethea L. Megyery. **My** brother Alex is my rock and their combined force has reminded me that fulfillment **will** not be found in all the materials, school and high expectations others have put out there for you, but rather that contentment is found within one's self. They have helped me to figure out just what that means.

Parallel in direction and magnitude are the contributions of the couple that have filled the shoes in my soul and spirit as parents: Mr. Walt Giuffrida and Ms. Jacqueline Wheatley. Without Walt's attention and dedication to the old Suzy, that little girl so full of life, **I** would have never known of MIT at that young age. Upon my admission to MIT they gave me a personalized copy of Emily *Post's Etiquette,* a little black dress, and charged me to sever myself from my past and to go out into the world and become the sweetest person I can be.

^Achildhood without the feeling of family taught me to simplify and that your selection of friends indeed *are* your family. With that, I adore my brother Joel Ramine and sister Cypriana George in St Lucia, my best friends from my MIT experience Brandon Reese, Sebastian Figari, Dhaval Adjodah and Melina Flores, and the unparalleled support of David Ellis.

^Iwant you to know that I am very, very fortunate. **I** have gone through many different experiences to come to this new journey. **I** fully believe with all my heart and soul that life finds me even more equipped to take this next leap. **I** believe I will glean so much more out of it because there's a more complete Suz. **I** know **I** have some amazing adventures ahead of me... some of these I could have missed if **I** were my younger self.

My wish for all who read this is that you are content with you. Many people spend their whole lives looking for that contentment only to have missed it inside themselves. The world is at your fingertips, it's all in how you see it, and **I** hope you see it **-** especially Ms. Ashlynn Jones.

I lovingly dedicate this thesis to Alex & Alethea Megyery, Walt Giuffrida & Jacqui Wheatley, Joel Ramine, Cypriana George, Brandon Reese, Sebastian Figari, Dhaval Adjodah, Melina Flores, and David Ellis.

Ms. Ashlynn Jones, I believe in you.

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Acronyms

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Chapter 1: Introduction

1.1. Project Scope

This thesis was written in partial fulfillment of the Bachelor of Science degree in the Department of Earth, Atmospheric **&** Planetary Science **(EAPS)** at the Massachusetts Institute of Technology (MIT). The content of the thesis falls under the topic of geochemical remote X-Ray Fluorescence (XRF) spectrometry in the context of REgolith X-ray Imaging Spectrometer (REXIS), a Charge-Coupled Device (CCD)-based coded aperture soft X-ray **(0.3-7.5** keV) telescope for remote geochemical XRF spectrometry.

REXIS has been selected as an education and public outreach element of the OSIRIS-REx mission investigation after a competitive selection process. As a student collaboration experiment, REXIS will be incorporated into the OSIRIS-REx spacecraft on a non-impact basis and **will** remain clearly separable from the rest of the OSIRIS-REx investigation. Data analysis will be handled as a student project. The REXIS project is a collaboration between students and faculty at MIT and Harvard University. At MIT, faculty leadership is provided **by** Professor David Miller, Professor Richard Binzel, and Professor Sara Seager. At Harvard, the faculty leadership is provided **by** Professor Josh Grindlay. For other REXIS-related work that **I** conducted prior to this document, supplemental information is available in Appendix **A** of this thesis.

REXIS **will fly** onboard the Origins Spectral Interpretations Resource Identification Security Regolith Explorer (OSIRIS-REx) asteroid sample return mission that will be visiting the asteroid **1999 RQ36** and sending a sample back to Earth. The spacecraft will conduct mapping, which will allow comparison with Earth-based observation, and the mapping of surface organics, volatiles and potential resources. Finally, the mission will map surface features to show the correlation of soil pools with local gravity and asteroid rotation.

OSIRIS-REx will accomplish these through a variety of instruments on board, among them, REXIS. **All** aspects of the mission, including REXIS, lead to identifying the best sampling site on the asteroid. REXIS will map the surface elemental abundance ratios of **1999 RQ36** globally and spatially, contextualizing the physical sample return, using CCD-based, fluoresced X-ray detection.

In the general case, XRF is a well-established technique in the laboratory for performing both qualitative and quantitative elemental analysis of samples. CCD-based, fluoresced X-ray detection of a continuous source, with a variety of geochemical elements present, will produce a spectrum or histogram. The spectrum will span a range of characteristic elemental energies with specific intensities. To discriminate fluorescent X-rays of major elements, a specific energy resolution, here known as spectral resolution, is required. Spectral resolution is also dependent on an effective detection area to accumulate statistically significant counts of X-ray photons in a given integration time for quantitative elemental analysis. Spectral resolution degrades with detector temperature and detector degradation due to radiation damage. Quantifying the limits for accurate spectral resolution is vital to XRF science.

This thesis outlines laboratory experimentation that was conducted on the REXIS CCDs and analyzes derivative work that was completed from the results of that experimentation to define inputs into a spectral resolution model. The results of that spectral resolution model then dictate the photon count detection accuracy and the temperature required of the **CCD** to meet the REXIS requirements. It is mandatory to meet these requirements in the **CCD** detector for efficient spectral resolution and useful spatial resolution of the asteroid surface to complete the mission science goals.

1.2. Audience

I have written this thesis with the express purpose of accessing several different audiences, each of who may find interest in specific sections of the thesis, but not others. It is the intention that some of the results of this report be used to improve the scientific community's understanding of the limits of remote geochemical CCD-based XRF sensing as detector conditions degrade.

The groups that **I** anticipate being immediately interested in this thesis include the REXIS project partners, to include the MIT Space Systems Laboratory **(SSL),** the MIT **EAPS** Department, the MIT Kavli Institute for Astrophysics, the Harvard College Observatory, the Harvard Department of Astronomy, the **NASA** Goddard Space Flight Center **(GSFC).** It is my hope that this thesis contributes to the scientific and engineering development of the REXIS Instrument and future Instruments of this nature. Also, current and future MIT students from the MIT **EAPS** and Physics Departments who may express interest in learning more about XRF and the limits of the **CCD**based detection technique for their own projects; this thesis can serve as an introduction for them.

1.3. REXIS Mission Objectives

The REXIS Instrument **will fly** onboard the OSIRIS-REx spacecraft as one of five Instruments. REXIS **will** provide an X-ray global map of **1999 RQ36,** complementing core OSIRIS-REx mission science.

1.3.1. OSIRIS-REx Mission

NASA will launch the OSIRIS-REx spacecraft in **2016** to the primitive, near-Earth asteroid **1999 RQ36 (101955).** The spacecraft will use a robotic arm to pluck samples from the asteroid and send them back to the Earth; this will be the first **U.S.** mission to carry samples from an asteroid back to Earth. OSIRIS-REx **will** approach the asteroid in **2019.** The next six months of the mission will be spent conducting surface mapping with a variety of different instruments. The science team shall then choose the sample site and the spacecraft will move closer to take the sample. The sample will then be sent back to Earth for analysis.

L3.1.. OSIRIS-REx Project Motivation

Telescopic observations have helped define the orbit of asteroid **1999 RQ36** that brings it close to the Earth every six years. It is critical to know the orbit of **1999 RQ36** because recent calculations produced a **1** in **1800** chance of impact with Earth in the year **2182** (The University of Arizona, **NASA** Goddard Space Flight Center, **2013).** Part of the OSIRIS-REx mission is to refine understanding of effects on this orbit and its intersection with Earth's.

The primary mission objective is to return the first pristine sample of carbon-rich material from the surface of a primitive asteroid. The target asteroid, near-Earth object **1999 RQ36,** is an accessible, volatile- and organic-rich remnant from the early solar system (The University of Arizona, **NASA** Goddard Space Flight Center, **2013).**

Carbonaceous asteroids, such as **1999 RQ36,** are the direct remnants of the building blocks of terrestrial planets (Jagoutz, 2011). Their chemical and physical nature, distribution, formation, and evolution are fundamental to understanding planet formation and the origin of life. In contrast to stony and iron asteroids that formed within **2-3 AU** of the Sun, the carbonaceous asteroids contain a large fraction of organics and are volatile rich. Although **1999 RQ36** is currently located at a close distance to the Sun, it originally formed beyond **2-3 AU** and has recently, within the past few million years, been moved into its current orbit **by** the gravity of the major planets (The University of Arizona, **NASA** Goddard Space Flight Center, **2013; A.** Milani, **2009).**

1.3.1.2. Mission Science Objectives

Telescopic observations have revealed some basic properties of **1999 RQ36.** They indicate that **1999 RQ36** is very dark and has been classified in a rare subgroup of the dark, carbonaceous asteroids that are considered "primitive" having undergone little processing from their time of formation (The University of Arizona, **NASA** Goddard Space Flight Center, **2013).**

Through sample, mapping and measuring the asteroid, the science objectives define the questions the mission wants to ask about the asteroid and the information the mission hopes to learn.

From the Mission's Science Objectives webpage (The University of Arizona, **NASA** Goddard Space Flight Center, **2013),** these Objectives are:

- Return and analyze a sample of pristine carbonaceous asteroid regolith in an amount sufficient to study the nature, history, and distribution of its constituent minerals and organic material.
- **"** Map the global properties, chemistry, and mineralogy of a primitive carbonaceous asteroid to characterize its geologic and dynamic history and provide context for the returned samples.
- Document the texture, morphology, geochemistry, and spectral properties of the regolith at the sampling site in situ at scales down to millimeters.
- Measure the Yarkovsky effect, a thermal force on the object, on a potentially hazardous object and constrain the asteroid properties that contribute to this effect.
- **"** Characterize the integrated global properties of a primitive carbonaceous asteroid to allow for direct comparison with ground-based telescopic data of the entire asteroid population.

1.3.2. The REXIS Instrument

Sample return is the principal objective of the OSIRIS-REx mission. Maximizing the science yield from the sample analysis requires the best possible asteroid context for the returned sample. The REXIS science objective is to complement onboard mineral mapping **by** adding spatially resolved elemental abundance ratio mapping achieved through X-ray spectrometry.

13.2.1. REXIS Instrument Overview

REXIS is a coded aperture soft X-ray **(0.3-7.5** keV) telescope that images X-ray fluorescence line emission produced **by** the interaction of solar X-rays and the solar wind with the regolith of **1999 RQ36.** Images are formed with 21-minute arcminute resolution (4.3 m spatial resolution at a distance of **700 m).** Imaging is achieved **by** correlating the detected X-ray image with a 64 x 64 element random mask **(1.536** mm pixels). REXIS will store each X-ray event in order to maximize the data storage usage and to minimize the risk. The pixels **will** be addressed in 64 x 64 bins and the **0.3-7.5** keV range will be covered **by 5** broad bands and **11** narrow line bands. **A** 24-sec resolution time tag **will** be interleaved with the event data to account for **1999 RQ36** rotation. Images **will** be reconstructed on the ground after downlink of the event list. Images are formed simultaneously in **16** energy bands centered on the dominant lines of abundant surface elements from O-K **(0.5** keV) to Fe-KB **(7** keV) as well the representative continuum. During orbital phase 5B, a 21-day orbit **700** m from the surface of **1999 RQ36,** a total of at least **133** events/asteroid pixel/energy band are expected under 2 keV; enough to obtain significant constraints on element abundances at scales larger than **10** m (The University of Arizona, **NASA** Goddard Space Flight Center, **2013).**

L3.2.2. REXIS Requirements

As **it** is the REXIS design goal to complement OSIRIS-REx Science Objectives, the REXIS Instrument Requirements are a direct flow-down from the following OSIRIS-REx Science Requirements (The University of Arizona, **NASA** Goddard Space Flight Center, **2013)** that REXIS anticipates contribution to:

- **"** Surface Chemistry Functional Requirement: OSIRIS-REx shall document the texture to submm resolution, morphology, geochemistry, and spectral properties of the sampling site.
- * Surface Chemistry Performance Requirement: OSIRIS-REx shall determine mineral, organic, and phase abundances that have spectral absorptions **>5%** on the surface of **1999 RQ36,** at a global spatial resolution of **50** m or better.
- **"** Volatile Outgassing Requirement: OSIRIS-REx shall search for and characterize regions of active volatile outgassing from the surface of **1999 RQ36.**

In order to complement the OSIRIS-REx geochemical requirements, the following are the REXIS Instrument requirements:

REXIS has the following Level **1** requirements to **(1)** measure global elemental abundances with sufficient sensitivity to distinguish the mean global composition of the asteroid for classification and (2) perform an elemental abundance analysis of the surface of **1999 RQ36** with sufficient spatial resolution to contribute to sample site selection.

Table 1: REXIS Level 1 Requirements

(MIT/Harvard REXIS Team, **2013)**

REXIS has the following Level 2 requirements describing **(1)** the accuracy requirement for measuring global ratios of elements of the major meteorite groups, (2) the abundances of key elements, **0,** Fe, and **Mg,** with a specific SNR and spatial resolution in order to aid in sample site selection, **(3)** the abundance of **Si** with a greater enhancement than those of (2) because all elemental ratios are in reference to Silicon, and (4) multiple element detection through spectroscopic identification with a specific SNR in order to understand the elemental composition of **1999 RQ36.**

Table 2: REXIS **Level 2 Requirements**

(MIT/Harvard REXIS Team, **2013)**

Meeting Level 2 requirement REX-2.1.1 is a goal of this thesis. **A** thorough explanation of the model developed to meet Level 2 Requirement REX-2.1.1 **will** be discussed in Section **3.2** of this thesis.

REXIS has the following Level **3** requirements describing spectral resolution requirements to discriminate unique XRF lines from each other and the Field of View (FOV) requirement in order to collect a quantity of photons statistically large enough to resolve higher energy lines.

Table 3: REXIS Level 3 Requirements

(MIT/Harvard REXIS Team, 2013)

Meeting Level **3** requirement REX-3.1.1 is also the subject matter of this thesis. **A** thorough explanation of the model developed to meet Level **3** Requirement REX-3.1.1 will be discussed in Section **3.2** of this thesis.

In order to meet these requirements, and hopefully REXIS aims, discussed in Section 1.3.2.4, the REXIS instrument preliminary design is introduced in the next section.

1.3.2.3. REXIS Design Components

To fulfill the above REXIS Science and design requirements, the REXIS instrument has the following preliminary design:

Figure **1** depicts the key components of the REXIS preliminary design. Relative to this thesis is the coded aperture mask and the detector plane.

Figure 1: The REXIS preliminary design depicting key components to contextualize the CCDs to be used in the REXIS **instrument (Image Credit: MIT/Harvard** REXIS **Structures Team, 2011).**

Coded aperture imaging is typical for point sources and continuous sources, and so, the coded aperture mask **will** consist of a mask sufficient to perform the best possible deconvolution and correlation methods for image reconstruction with minimal noise as REXIS observes a continuous source, **1999 RQ36.**

The REXIS detector plane will feature a 2 x 2 grid of CCID-41 **CCD** detectors that will be built **by** MIT Lincoln Laboratories. Optimizing the spectral resolution for these CCDs, defining the spectral resolution limits and the **CCD** requirements to meet these requirements is the focus of this thesis.

The REXIS design, including the CCID-41, is built on flight heritage to minimize associated risk factors. The Japanese Hayabusa spacecraft, formerly known as **MUSES-C** (launched **2003)** was the first time a **CCD** had been used for the purpose of XRF spectroscopy on a planetary mission (T. Okada M. K., 2000). The Japanese Suzaku spacecraft (launched **2005)** was Japan's **5th** X-ray astronomy satellite; it is an Earth-orbiting observatory imaging various X-ray sources. Suzaku is equipped with four independent CCD-based X-ray Imaging Spectrometer **(XIS)** sensors that were built **by** MIT Lincoln Laboratory (K. Koyama, **2006). A** thorough discussion on the flight heritage of REXIS can be found in Appendix B.1.

1.3.2.4 REXIS Science Aims

As a student collaboration experiment, REXIS will be incorporated into the OSIRIS-REx spacecraft on a non-impact basis and **will** remain clearly separable from the rest of the OSIRIS-REx investigation. Data analysis **will** be handled as a student project.

It is the hope of the REXIS team that the Instrument will successfully achieve its science objective to map the surface elemental abundance ratios of **1999 RQ36** globally and spatially. Achieving this objective, REXIS may be able to assist in contextualizing the physical sample return from the OSIRIS-REx spacecraft. Further, it is possible that with good conditions, successful completion of our objectives, and real-time data analysis is conducted, that REXIS science shall contribute to the sample site selection of the OSIRIS-REx Science Team.

1.4. REXIS **Technological Baseline and Introduction**

Meeting the REXIS Level 2 requirement REX-2.1.1, the accuracy requirement for measuring global ratios of elements, and Level **3** requirement REX-3.1.1, describing spectral resolution requirements to discriminate unique XRF lines from each other, is the subject matter of this thesis. **A** thorough explanation of the model developed to meet these requirements will be discussed in Section **3.2** of this thesis.

An X-ray detector such as the Lincoln-Laboratory **CCID-41,** which will be used in the REXIS Instrument, is grown from a single silicon crystal. Silicon is used because of the material's favorable charge transfer properties. The detector operates **by** collecting charge imparted **by** incoming photons on pixels etched into the crystal surface. The pixels hold the charge until the detector's electronics measure the charge at each pixel and then ground of 'refresh' the pixel array to begin the next imaging cycle.

The quality of X-ray spectrometry is dependent on **(1)** the efficiency of charge transfer and (2) the accuracy of charge measurement. High-energy particles from the Sun and deep space degrade the detector's ability to efficiently transfer charge, thereby reducing spectral resolution. These properties affecting the CCD's spectral resolution are all intrinsic to the detector. Specifically, the SNR of the **CCD** detector has a carry-over effect on the spatial resolution that can be attained.

The correct interpretation of X-ray measurements from the surface of **1999 RQ36 will** require a detailed understanding of XRF science and the limitations of a CCD-based X-ray detector. **CCD-** based X-ray detection produces a histogram of photon intensities, each with a characteristic energy and wavelength. Ideally, without intrinsic noise interference, the read-out from the detector would be a set of very sharp peaks for each line as delta functions. The widening of spectral lines can depend on a number of factors to include effects from the detector hardware, say, due to detector temperature.

To effectively discuss these factors that contribute to the degradation of a CCD's spectral resolution, Chapter 2 discusses the baseline **by** which these scientific processes occur and the detailed technology of these operations in order to accurately frame the scientific questions that this thesis proposes to answer. Chapter **3** introduces the investigations that have been conducted in order to test and model the limits that will 'break' or surpass the REXIS Instrument requirements REX-2. **1.1** and REX-3.1.1. Finally, Chapter **3** clearly defines the scientific results of this analysis of the MIT/Harvard REXIS Team's work in the hope of expanding the REXIS Level 2 and Level **³** requirements **by** recommending additional instrument requirements to acquire useful scientific results from the instrument.

Chapter 2: Scientific Baseline & Physical Foundations of Operations

Meeting the REXIS Level 2 requirement REX-2.1.1, the accuracy requirement for measuring global ratios of elements, and Level **3** requirement REX-3.1.1, describing spectral resolution requirements to discriminate unique XRF lines from each other, is the driver of this thesis analysis. Chapter 2 discusses XRF science technology as it is relevant to REXIS at **1999 RQ36** (Section 2.1), the solar X-ray environment as it is relevant to the detection of elements **by** REXIS at **1999 RQ36** (Section 2.2), CCD-based coded aperture imaging (Section **2.3),** and CCD-based XRF detection as it applies to the REXIS instrument at **1999 RQ36** (Section 2.4) to outline the background necessary to understand the science questions this thesis hopes to address.

2.1. Physical Foundations of Operation: XRF for Geochemical Remote Sensing

X-ray fluorescence (XRF) is a well-established technique in the laboratory for performing both qualitative and quantitative elemental analysis of samples. X-ray spectroscopy from an orbiting spacecraft can be used to identify elemental composition at the surface of an airless body such as **1999 RQ36.** The crux of geochemical analyses of planetary bodies in the solar system is that only ratios, and not necessarily absolute concentrations, are similar between the different reservoirs. In the case of REXIS, these ratios are normalized to Silicon. The sun is the origin of the X-ray photons that fluoresce from the asteroid surface; however, the Sun is a continuous and time-variable source of X-rays. Solar X-ray emissions across all parts of the spectrum vary temporally, in terms of origin on the solar surface and in terms of the direction they are emitted. The effects of the solar X-ray environment are discussed in Section 2.2.

2.1.1. Introduction to X-ray Fluorescence

X-ray fluorescence (XRF) is a well-established laboratory technique for performing elemental analysis of samples both qualitatively and quantitatively. Usually an X-ray source is used to eject inner-shell electrons from the atoms in the sample **by** way of photoelectric absorption. The subsequent rearrangement of the atomic electrons gives rise to the emission of monoenergetic Xrays **-** the fluorescent X-rays **-** whose energies, in the 0.1-to 100-keV range, are characteristic of the specific chemical elements in the sample. Spectral analysis of these fluorescent X-rays can therefore provide us with information concerning the elemental composition of the sample.

As was proven during the Apollo **15** and **16** missions, major elemental composition can be determined through the remote XRF method for atmosphere-free planetary surfaces (T. Okada T. **A., 2005; I. J.** Adler, **1977).** Solar X-rays irradiate a regolith surface to excite each atom of the uppermost surface materials. Immediately, those atoms settle to the ground state and X-rays characteristic of major elements are illuminated off the surface. However, intensity and spectral profiles of solar X-rays vary through time, which affects those XRF rays off the surface as well. The variability of the solar X-ray environment is discussed in Section 2.2. Therefore, major elemental composition can be mapped from the orbiting altitude with remote XRF spectrometry, together with concurrent monitoring of solar X-rays (T. Okada T. **A., 2005).** XRF telescopes have mission flight heritage that is included in Appendix B.1.

X-ray Fluorescence for Geochemical Remote Sensing 2.1.2.

X-ray spectroscopy from an orbiting spacecraft can be used to identify elemental composition at the surface of an airless body in the following manner:

- **"** X-rays from the Sun shining on an asteroid **will** produce significant X-ray fluorescence from the elements contained in the very surface of the asteroid (the top **1** mm)
- **"** When an element fluoresces, it emits X-rays with discrete energies that are unique or characteristic of that element (the strongest being $K \alpha$ X-rays)
- **"** The asteroid pointing portion of the X-ray spectrometer detects these characteristic X-rays of a particular energy range
- From an analysis of the measured energy spectrum, elemental composition can be inferred

The intensity of detected X-rays is a strong function of solar illumination and activity. This relation is quantified in Section **3.1.** To ensure proper quantitative analysis, X-ray spectrometers point an additional X-ray detector sunward to directly measure the incident solar flux (Clark, **1979).** The signal strength measured **by** the X-ray spectrometer **will,** in general, improve as the Sun approaches solar maximum and will be particularly strong during the random solar flares that produce brief periods of high-level, high-energy solar X-rays **(N.** Crosby, **1993).**

By observing asteroids, we hope to gain an understanding of early solar-system history. Evidence that asteroids preserve information that cannot be acquired from the study of major planets and satellites alone comes from the study of meteorites. Examining meteorites has indicated their parent bodies to have a variety of thermal histories. Some have been untouched since their original accretion from the solar nebula, whereas others are the products of global melting and differentiation (R. Michelson, **2006).** Asteroids are the best parent-body candidates for meteorites (Lim, **2005).**

The derivation of surface chemistry from spectral observations requires assumptions about the physical nature of the surface and thus remains at least somewhat speculative. High energy solar Xrays directly induce the production of X-ray lines, the energies of which are characteristic for chemical elements present on the surfaces of rocky bodies. This emission is strongly dependent on the chemical composition of the surface as well as on the solar spectrum **(I.** Adler **J.** T., **1970;** I. Adler **J. G., 1972;** P. Clark **I. A., 1978;** Clark, **1979;** L. Yin, **1993).** Generally, the strength of the given line indicates how much of a given component, a major element, is present. Soft X-rays are readily attenuated, so the atmosphere must be virtually non-existent for X-rays to be produced and detectable from orbit. Also, an X-ray source must be in close enough proximity to generate lines which have sufficient SNR to be measurable **by** a remote detector during its encounter with a body. This limitation restricts XRF experiments to the sunlit surfaces of atmosphereless bodies in the inner **(<3 AU)** solar system, when the Sun is used as the primary source **(I.** Adler J. T., **1970; I.** Adler **J. G., 1972).**

When solar X-rays interact with atoms at the very surface (to a depth of tens of microns), characteristic X-ray K α fluorescent lines for the major elements in this very surficial layer are produced. The solar flux, which is much more intense at lower energies, is sufficient to generate measureable lines not only for the major elements with lower energy lines such as **Mg, Al, Si,** but even for the higher energy lines of Ca, Ti, and Fe, although longer integration times are required.

Providing the surface is sufficiently "gardened" or mixed **by** micrometeorite impact over long periods of time, as was generally the case on the Moon, measurements made of the surficial layer are representative of underlying regolith (Taylor, **1975;** P. Clark B. R., **1990).** This may not be the case on asteroids, for example, and that result **will** have major implications for asteroid formation (P. Clark J. T., **1997).**

2.1.3. Geochemical Remote Sensing of Asteroids

In geochemical analyses of planetary bodies in the solar system, only ratios, and not necessarily absolute concentrations, are similar between the different reservoirs. Nearly all of the carbonaceous chondrite meteorite groups are composed mainly of silicates, oxides and sulfides. The nature of geochemical remote sensing and the assumed composition of **1999 RQ36,** that it is a carbonaceous chondrite, is the reasoning for this abundance-ratio approach to elemental mapping (MIT/Harvard REXIS Team, **2011).**

It is believed that carbonaceous asteroids are representative of the early solar system and thus the building blocks of planets and the origins of life. **A** major research subject in Earth science **is** to understand planetary formation and differentiation processes. It is widely believed that the Earth formed in an area of the solar nebula that was depleted of the more volatile elements. Additionally, during or after accretion, the core formed due to the segregation of a Fe-rich liquid towards the center of the Earth (Jagoutz, **2011).**

In geochemical sciences it is convention to separate the solar system, the Earth and any other planetary objects we have samples from, in different geochemical reservoirs. For example, some important reservoirs on the Earth are physical and geological entities like the core, mantle, crust and atmosphere. In more detailed investigations, the mantle itself is often separated into a multitude of smaller reservoirs or components, which do not necessarily differ in their physical or mineralogical properties.

Even using volcanic eruptions and mantle processes we can directly investigate only the upper **~150-200** km of the Earth. That is not much given the mean radius of the Earth **(6,371** km). In order to make inferences about inaccessible reservoirs we make two major assumptions:

- **(1)** We assume that the mantle is not compositionally layered. This assumption is justified **if** the mantle gets continuously stirred up and mixed **by** processes e.g., related to plate tectonics. Numerical simulations of mantle convection support this assumption. **If** that's true, then the uppermost **150-200** km of the mantle should be representative of the remaining **-3,283** km which we cannot access (Jagoutz, 2011). So, we can now claim that by direct sampling and chemical analyses of rock samples we can constrain the geochemistry of the entire silicate part of the Earth, called the Bulk Silicate Earth **(BSE),** which extends from the Earth's surface to the core-mantle boundary at **3,483** km. However, there are no samples available to constrain the composition of the core.
- (2) We assume the compositions of certain asteroids, carbonaceous chondrites, are representative of the composition of the solar nebula (with the exception of H and He, which make up the bulk of the solar nebula). This assumption is justified because we can approximate the bulk composition of the Sun, which makes up **~99.85%** of the mass of our

solar system. The composition of the Sun is indeed very similar to the relative abundance of elements in these asteroids, **if** internally renormalized. Since these asteroids fell on Earth, we have direct samples that approximate the bulk composition of our inner solar system (with the exception of the extremely volatile elements H, He) (Jagoutz, **2011).**

The crux of geochemical analyses of planetary bodies in the solar system is that only ratios, and not necessarily absolute concentrations, are similar between the different reservoirs. In the case of REXIS, these ratios are normalized to Silicon. Nearly all of the carbonaceous chondrite groups are composed mainly of silicates, oxides and sulfides (Lim, **2005).** The nature of geochemical remote sensing and the assumed composition of **1999 RQ36,** that it is a carbonaceous chondrites, is the reasoning for this abundance-ratio approach to elemental mapping.

2.1.4. **Thermal Effects on 1999 RQ36 - Mission Validation**

The near-Earth object **(NEO) (101955) 1999 RQ36** is a potentially hazardous asteroid known to have a non-negligible probability $({\sim}10^{-3})$ of impacting the Earth in the second half of the 22nd century **(A.** Milani, **2009).** The interest in **RQ36** boosted several campaigns of physical characterization: Radar delay-Doppler data allowed an accurate reconstruction of the shape of the object (H. Campins **A.** M., 2010) and showed that **RQ36** is nearly spheroidal, has a diameter of ~0.5km and a double top-shaped silhouette. Thermophysical modeling **(TPM) (A.** W. Harris, 2002) of thermal infrared observations obtained from *Spitzer* indicates a size of 0.61km and a surface thermal inertia of ~ 600 J m⁻² s^{-0.5} K⁻¹ (H. Campins A. M., 2010). Given an absolute magnitude of **20.9** (MPC) and a diameter of 0.5km, the geometric visible albedo is **0.03.** This makes **RQ36** a very dark object, consistent with a primitive organic-rich composition. Moreover, a recent study has identified the family of the low-albedo asteroid (142) Polana as the most likely main belt origin of **RQ36** (H. Campins K. **H.-A.-D.,** 2010).

Important issues related to sample return missions to primitive asteroids are **(1)** to determine the likelihood that the target candidate originates from a source region of primitive material in the solar system (e.g., the Polana family in the case of **RQ36);** and (2) to assess whether the surface (and the subsurface) of the object has been preserved from weathering processes active in space.

The implantation of ions **by** the solar wind and the bombardment of micrometeorites can alter the spectroscopic properties of asteroids (Hapke, 2001; **S.** Sasaki, 2001). However, these processes affect only the first few microns of the surface, while current sampling devices aim at collecting material at depth larger than a few millimeters

Solar heating can also alter the original composition of NEOs **-** they can typically reach temperatures **>370K,** and the heat penetration depth is of the order of some centimeters for a body with a rotation period of some hours *(J.* R. Spencer, **1989;** Harris, **1998).**

There is a probability $\leq 20\%$ that the surface temperature of RQ36 was ever above 600K (M. Mueller, **2011).** However, for temperatures below 600K, the probability rises rapidly: there is a **50%** and 80% probability that the temperature was \gtrsim 500K and \gtrsim 450K, respectively. The probability is **-1** that the surface was heated above 400K (M. Mueller, **2011).**

We expect at least part of the material of **RQ36** to be of similar composition as carbonaceous meteorites, as suggested **by** their similarity in spectral properties (M. Mueller, 2011). In these meteorites, around **25%** of the organics are present as solvent-soluble or free organic matter, but the majority is in the form of a complex solvent-insoluble macromolecular material (Sephton, 2002).

In order to measure the kinetic parameters for thermal degradation of organic matter in the carbonaceous meteorite Murchinson, Delbo et al. (M. Delbo, 2011) have performed laboratoryheating experiments. They found that at **370K** the aliphatic **C-H** bond of the insoluble organics **is** lost approximately within 200years. However, the bulk organics of Murchinson are lost in only on year at 370K (or 200 years at 300K). Lauretta et al. (D. S. Lauretta, 2001) have also studied the release of volatile elements as a function of temperature in **CM** and **CV** carbonaceous chondrites. Their results show that volatiles (in particular **Hg)** are released from most of the meteorites from about 470K.

From M. Delbo, (2011) it is noted that the surface of **RQ36** is certainly dehydrated and its organic matter degraded. Some volatiles are also likely lost. Delbo and Michel (2011) concluded that **if** interlayer H₂O was present in the minerals at the surface of RQ36, this water is gone as the body entered the near-Earth space and surface temperature rose above the dehydration threshold. On the other hand, it is unlikely that water was lost from the dehydroxylation of phyllosilicates at the surface of **RQ36,** and it is even less likely that this happened at the subsurface, which is positive result for the objectives of a sample return mission (M. Delbo, **2011).**

2.2. **Physical Foundations: The Solar X-ray Environment**

The solar output varies **by** three or more orders of magnitude across the energy range of interest and can be **highly** variable on timescales as short as minutes; thus, establishing the relationship between solar output, X-ray fluorescence line intensities, and surface concentrations is essential for quantitative results. Absolute elemental concentrations can be derived from instrumental measurements only when the quantitative nature of this relationship has been established.

2.3.1. The Solar X-ray Spectrum as a Function of Solar Activity

When REXIS will be operating, primarily phase 5B of the OSIRIS-REx mission, solar minimum **is** the expected solar state. The solar output is **highly** variable in the soft X-ray region and the rate of decrease of the incident flux is steeper than that of a higher solar states. Thus, sufficient solar flux to induce the fluorescent X-rays required to effectively detect Mg/Si and S/Si ratios is expected for the quiescent conditions that are expected during REXIS encounter with **1999 RQ36.**

The solar spectrum consists of a continuum, which is generated **by** thermal bremsstrahlung and Radiative recombination, with superimposed major lines that result from emission due to transitions in H- and He-like ions (P. Clark **J. T., 1997).** The sources for these lines are ions with appreciable abundances in active regions. In modeling the solar output in their region of interest **(1** to **10** KeV), Clark and Trombka (P. Clark **J. T., 1997)** used the best-published levels of solar output (Zombeck, **1990)** over a range of levels of solar output, with only major lines included in the spectra.

Solar output is **highly** variable in the soft X-ray region, both in overall intensity and slope (R. Kreplin, **1977;** Donnelly, **1976;** National Geophysical Data Center, **1991).**

Table 4 outlines the main solar flare states **by** classification and gives approximate energy level ranges for each. REXIS will encounter **1999 RQ36** during a minimum or 'quiet' solar state. **A** graphical representation of these states can be seen in Figure 2.

Table 4: Classification of Solar Soft X-ray Output

(P. Clark **J.** T., **1997)**

Some of the various solar states, described in Table 4 above, are graphically depicted in Figure 2 below.

Figure 2: Graphical depiction of the solar flux for various flare states ranging from very low flux (Class A, ~4 MK) to very high flux (Class X, ~21 MK). Solar X-ray emissions across all parts of the spectrum vary tempora

REXIS **will** encounter **1999 RQ36** during a minimum solar state; the solar flux for this quiet solar state is graphically depicted in Figure **3** below.

Figure **3:** Solar spectrum X-ray counts as a function of the quiet or minimum solar states in Mega Kelvin emissions temperatures from the solar corona. This figure shows that at all quiet solar states, including minimum (MK **1),** X-ray emissions vary continuously over time. Specifically, the solar intensity decreases by three to four orders of magnitude from 1 to **10** keV, the result being that fluorescent lines, as **well** as the scatter-induced background, have greater intensity at lower energies. Finally, the rate of decrease of incident solar flux on the asteroid is greater for the minimum solar state (MK1) than that of MK2 through MK4 (Image Credit: Lucy Lim, 2012).

The solar intensity decreases by three to four orders of magnitude from $1\rightarrow 10$ KeV, the result being that fluorescent lines, as well as the scatter-induced background, have greater intensity at lower energies. As the level of solar activity increases, relatively more output occurs at higher energies, the slope of the spectrum becomes less steep, and the overall magnitude of the X-ray flux increases.

Coronal emission arises from **highly** ionized elements, at temperatures **->** 1MK **(1** Mega Kelvin **1,000,000** Kelvin). The physical understanding of this **high** temperature in the solar corona is still a fundamental problem in astrophysics, because it seems to violate the second thermodynamic law, given the much cooler photospheric boundary, which has an average temperature of *T* **= 5785** K (and drops to $T \sim 4500$ K in sunspots). However, it is important to interpret this emission spectrum of the solar corona because of the optically thin emission from the corona in soft X-rays (Aschwanden, **2006).**

Counts as a function of energy

Figure 4: Solar spectrum X-ray counts as a function **of various quiet solar states in Mega Kelvin** emissions from the solar corona with the elemental X-ray emission lines of interest to REXIS marked vertically across the spectrum. This graphic clearly shows that the lower energy X-ray emission lines have greater intensity than those of higher energy lines, and the relative intensities of two adjacent lines is greater during a time of solar minimum than a state of greater solar activity (Image Source Credit: Lucy Lim, 2012; Image Source Edited: Megyery, **2013).**

The process by which solar intensity decreases by orders of magnitude from $1\rightarrow10$ KeV, and that fluorescent lines, as well as the scatter-induced background, have greater intensity at lower energies, is called hardening. As the level of solar activity increases, relatively more output occurs at higher energies, the slope of the spectrum becomes less steep, and the overall magnitude of the X-ray flux increases. In Figure 4 we can see that elements with higher atomic numbers (Ca, Ti, and Fe) require a more active Sun to induce characteristic X-ray lines at intensities which are comparable to intensities for lower atomic number elements **(Mg, Si, S)** induced under more quiescent conditions. In general, for every order of magnitude increase in the level of solar activity, there is a corresponding order of magnitude decrease in the probability that the Sun **will** achieve that level of activity (P. ClarkJ. T., **1997).**

When REXIS will be operating, primarily phase 5B of the OSIRIS-REx mission, solar minimum **is** the expected solar state. Thus, sufficient solar flux to induce the fluorescent X-rays required to effectively detect Mg/Si and **S/Si** ratios is expected.

2.3.2. Temporal Variations in Solar Activity

To expand on the data presented in Section **2.3.1,** the Sun's output must be monitored and measured simultaneously with surface measurements due to the quick variability of incident solar flux and occurrence of solar flares. Figure **5** depicts this variation over four days in July **1990,** validating the need for a Solar X-ray Monitor (SXM) to be coupled with the REXIS instrument aboard OSIRIS-REx.

More recent studies of the Sun's soft X-ray output have indicated that the Sun's output is **highly** variable on all timescales (Bouwer, **1983;** R. Donnelly, **1981; N.** Crosby, **1993).** During the months around solar minimum (current OSIRIS-REx mission plan to encounter **RQ36** is during solar minimum), less than one flare per day is reported on average; however, such flares, which can increase the intensity of the solar flux **by 3** to 4 orders of magnitude, generally occur in clusters, with as many as **5** to **10** a day for as long as a few days. In between these periods of enhanced activity, weeks may go **by** when little or no flare activity occurs (Bouwer, **1983;** R. Donnelly, **1981; N.** Crosby, **1993).**

Figure **5:** This figure shows minute to minute average solar flux over four days in July **1990.** Features to note are the Mclass flare and extended rise and fall time typical of large flares on July 13, and repeated occurrences of low level, C-class, flares on the remaining three days, while the more linear background is at a more quiescent su National Geophysical Data Center, **1991;** Image analysis credit: P. Clark **J.** T., **1997).**

Figure **5** (National Geophysical Data Center, **1991)** illustrates the high variability possible in solar **flux,** even close to solar maximum as well as the apparent grouping of flare events. On July **13,** measureable output is continuously at low-class flare level, and a class-M flare occurs, elevating the flux level for many hours. On the next **3** days, the measureable output is still above the typical Class B subflare level, with less than an order of magnitude variation to the level of a modest flare (Level **C).** (The background level is an order of magnitude lower at solar minimum.) For this reason, the Sun's output must be monitored and measured simultaneously with surface measurements, validating the need for a Solar X-ray Monitor (SXM) to be coupled with the REXIS instrument aboard OSIRIS-REx.

2.3. Technology of Operation: CCD-based Coded Aperture Imaging

The REXIS Instrument is being designed and developed as a CCD-based coded aperture telescope that relies on ray tracing and shadow casting to reconstruct spatial distribution of incident radiation. This section introduces the mechanism **by** which a **CCD** detector functions (Section **2.3.1)** and basic limitations of **CCD** detector performance that are relative to this thesis (Section **2.3.2).**

The ability to create a spatially accurate global map of elements at **1999 RQ36** relies on codedaperture imaging. The system consists of:

(1) A coded mask **-** i.e., a plate with areas that are transparent or opaque to photons. The transport and opaque areas, 'mask elements', all have an equal size and are distributed in a pre-determined pattern, which is placed on a regular grid. The form of a mask element may be arbitrary.

(2) **A** position-sensitive detector, such as a **CCD,** whose spatial resolution is sufficiently matched to the mask-pattern grid size and which is sensitive to photons.

Figure 6: The REXIS open design mock-up depicting a coded **aperture** mask and detector image (MIT/Harvard Aperture **&** Detector Team, **2011).**

As two point sources illuminate the detector through the mask, the detector records two projections of the mask pattern. The shift of each projection encodes the position of the corresponding point source in the sky; the 'strength' of each projection encodes the intensity of the point source.

process (MIT/Harvard Aperture & Detector Team, 2011).

The REXIS Instrument is being designed and developed as a CCD-based detector telescope relying on coded aperture imaging technology. Coded aperture imaging relies on simple ray tracing and shadow casting to reconstruct spatial distribution of incident radiation.

This thesis considers the factors in the **CCD** detector for efficient spectral resolution in order to attain useable scientific data in regard to spectral resolution. The following sections **will** discuss the foundations of **CCD** operations, **CCD** imaging inefficiencies, and related factors to imaging and how they will affect the REXIS Science product.

2.3.1. Introduction to CCDs

A CCD is a device for the movement of electrical charge, usually from within the device to an area where the charge can be manipulated, for example conversion into a digital value. This is achieved **by** "shifting" the signals between stages within the device one at a time. CCDs move charge between capacitive bins or potential wells in the device, with the shift allowing for the transfer of charge between the wells.

A CCD pixel consists of an SiO₂ layer on top of a Positive-type semiconductor.

CCD area detectors have a linear response to light and because CCDs store charge, from an electronics point of view, they are capacitors. Technically, a **CCD** is a Metal Oxide Semiconductor capacitor **(MOS** capacitor). CCDs employ "P-type" (Positive-type) semiconductors. Silicon has 4 e in its valance band, out of the total of 8 normally available valence sites. In a P-type semiconductor, "trivalent" impurities are added to Silicon to fill an additional 3 e sites. Thus, the existing 4 e + 3 e added $= 7$ e⁻ out of the 8 available valence sites. By missing one electron, the resulting Si has a positive charge.

An electrode, called a 'gate', at the top of the pixel 'bucket' applies a positive 'bias voltage.' The result is a 'potential well' to hold electrons. Each gate can transfer charge to its neighboring gate and are thus, "charge-coupled." Gates are inter-connected to transfer charge, even **if** the neighboring gate is in the next pixel. To read out a **CCD,** voltages are varied on each of the gates in a precise time sequence that transfers or 'couples' charge from one gate to another; in other words,
transferred from one potential well to another. The voltage sequencing of the gates that enables the transfer is called 'clocking the **CCD.'** Charge transfer efficiency **(CTE)** measures the efficiency of each pixel-to-pixel transfer, this transfer process is diagramed in Figure **8.**

Figure 8: Shifting potential wells in a three-phase CCD. Two pixels in the same register (either parallel or serial) are illustrated here. At the end of the shift, electrons in pixel 1 have shifted to pixel 2 (Binzel, 2011

These capacitors are biased above the threshold for inversion when image acquisition begins, allowing the conversion of incoming photons into electron charges at the semiconductor-oxide interface; the **CCD** is then used to read-out these charges.

From a report produced **by** Sternberg (2012), for X-ray CCDs, incoming X-rays strike the silicon surface of the CCDs. The impact from these energetic photon particles is capable of moving silicon valence electrons to the conduction band, where they are captured in the potential wells. Each X-ray

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wavelength has an associated energy, and the higher the energy of the photon, the more electrons will be excited into the conduction band. Consequently, CCDs, are particularly useful in determining the elemental spectra of objects, including the asteroid **1999 RQ36,** because each element fluoresces at a particular frequency and energy level. **By** allowing only one photon to hit a pixel before it is read out, i.e. the pixel charge is measured, it is possible to determine the energy of the incident photon, and there fore the element from which it was fluoresced, **by** counting the number of electrons in the pixel well (Howell, 2000). Although this process is **highly** efficient, noise is introduced at nearly every step in the detection-through-readout process.

The following section introduces **CCD** calibration requirements for effective performance of the processes outlined above.

2.3.2. CCD Detector Performance - Thermally Dependent Factors

CCDs are excellent detectors that require a great deal of calibration because they carry a variety of sources of noise. The detector's performance and resolution ability is greatly affected **by** its operating temperature and other smaller sources of noise. When imaging with CCDs a number of different types of calibration frames must be taken and applied to the science image(s) to remove these sources of noise and error to the scientific image frames.

The gain of a **CCD** is determined **by** the amplification performed **by** an on-chip amplifier that **is** built directly into the silicon circuitry (Sternberg, 2012). This amplification is defined as the amount of voltage required to produce one analog-to-digital **(ADU)** unit. The **ADU** converter converts the analog voltage reading to a digital number. **A** common method for determining the value to set as the gain is to relate the full pixel well capacity to the largest number that can be represented **by** the onboard analog/digital converter, provided that the **CCD** response is linear (Howell, 2000). **CCD** linearity means that the input charge varies linearly with the digital value of the output signal.

Readout noise is related to the gain, since it represents the number of electrons that are added per pixel into the final signal (Sternberg, 2012). These additional electrons do not originate from the incident X-ray photons, but rather from the process of the **ADU** converter and from the electronics that naturally introduce electrons throughout the imaging process (Pryor). The temperature of the CCDs and its thermal properties significantly affect the rate of non-photon generated electrons introduced. Because data processing takes place at for each pixel, read noise is added to each pixel at each readout time; consequently, each pixel has additional electrons added from readout noise each time the pixel is read (Howell, 2000).

Shot noise is related to the number of photons that are converted into electrons when they impact the **CCD** silicon, where the quantum efficiency is the primary temperature dependent factor in determining the shot noise. Since the quantum efficiency varies with temperature, it must be managed to ensure that the shot noise is low and that the spectral resolution meets REXIS requirements. Additionally, there is a shot noise specifically associated with X-ray CCDs that **is** related to the ability of the **CCD** silicon to interact with the incoming X-ray; if an incident photon does not interact with the silicon and instead passes through, then the **CCD** will be unable to read the energy of the photon and cause a loss of data (Howell, 2000). This noise is very small, however, as silicon's useful photoelectric effect ranges from **1.1** to **10** eV, which includes the soft X-ray region that REXIS measures.

Thermal effects are directly responsible for adding dark current noise. Because the CCDs are above absolute zero, a small number of electrons are able to escape from the valence band of the **CCD** silicon, becoming trapped in the potential well of the **CCD** pixels. These electrons create a Poisson noise distribution that must be filtered from the readout in order to separate the inherent dark noise from the image pixel **by** pixel (Janesick, 2001). Since the temperature of the **CCD** determines the thermal agitation and electron freeing potential, cooling the CCDs reduce the amount of dark current, making the thermal noise as low as possible **by** reducing the number of thermal electrons (Howell, 2000). It is additionally important to maintain the CCDs at uniform temperatures to prevent thermal changes across the CCDs that could artificially influence the readouts based on pixel location in the **CCD** array (Howell, 2000).

2.3.3. Spectral Degradation Due to Temperature

CCD performance depends on the detector's physical state which has a profound effect on the output scientific data. As **CCD** performance degrades our ability to measure spectral lines and distinguish neighboring lines from one another will degrade.

The temperature of CCDs is a crucial factor in the noise level seen in output data. Noise must be reduced or managed to achieve the required spectral resolution, thus temperature management **is** paramount. There are many factors of noise, the two most significant being the readout and dark current noises. Together, they form the majority of the total noise component in CCDs, so they are the two sources that are examined here (Sternberg, 2012).

The total noise *N* is defined as the following sum of the read and dark noises (Janesick, 2001):

$$
N = \sqrt{\sigma_{read}^2 + \sigma_{dark}^2}
$$

In this equation, the total noise is the square root of the sum of each noise component squared; the other forms of noise, such as shot noise, could similarly be added into this equation (Sternberg, 2012).

2.3.1.1. Readout Noise

The readout noise is proportional to the square root of the temperature and can be written as the following equation:

$$
\sigma_{read}^2 = 4kTBR
$$

In the above equation *k* is the Boltzmann constant of 1.38 x 10^{23} J/K, T is the absolute temperature in Kelvin of the **CCD,** *B* is the noise power bandwidth in Hz, and R is the resistance of the

conducting path in Ohms (Janesick, 2001). The values *T*, *B*, and *R* are unknown values and are intrinsic to each **CCD** detector.

In order to determine the read noise as a function of temperature alone, values for *B* and R must be determined using existing data from flight heritage of the CCID-41 from the Suzaku mission. According to a paper **by** Bautz (2004), the read noise was measured on the Suzaku CCID-41s to be 2.5 e - RMS at 41 kpix/sec and -90° C. Using these values for *T* and σ_{read} , the above equation can be solved for *BR*, yielding *BR* = $9.2029x10^{18}$ Hz-Ohms. Therefore,

 σ_{read}^2 = $4 \times 1.38x10^{-23} \times 9.2029x10^{18} \times T$

$$
\sigma_{read}^2 = 5.08x10^{-4} \times T
$$

This readout noise expression is shown graphically in Figure **9** below.

Figure 9: Graphically depicts readout noise versus temperature in degrees Celsius. This figure shows how the readout noise is affected by the CCD temperature. This linear relationship with temperature shows that the readout noise is best reduced at lower operating temperatures, showing the importance of cooled CCDs for improved spectral resolution (Sternberg, 2012).

The readout noise exhibits a linear relationship with temperature and is measured in e-RMS. Therefore, the readout noise contributes to the total noise expression values that vary linearly with the **CCD** operating temperature, with lower temperatures producing less readout noise. This noise must be removed with the subtraction of a dark image, i.e., am image where no X-ray sources are present, so that the readout noise is inherently part of this baseline frame (Janesick, 2001).

2.3.1.2. Dark Current Noise

The other major component of the total noise expression is the dark current noise, which is specific to each **CCD.**

The dark current noise is specific for each **CCD,** and is given for the CCID-41s as the following equation where T is temperature in Kelvin (Hamamatsu Photonics K.K.):

$$
\sigma_{dark}^2 = 0.00092 \times T^{1.5} \times e^{-6995.35963/T} \left[\frac{A}{cm^2}\right]
$$

Based on experimental data that is CCID-41 specific, conducted **by** the REXIS Science Team during the Summer of **2012,** it is representative of exactly what is expected from the REXIS CCID-41s (Megyery, REXIS Science Team **ETU-1:** CCID-41 Laboratory Experimentation, 2012). The dark current noise is the largest component of the total noise of a **CCD.** Figure **10** shows a plot of this dark current noise. This noise source varies greatly with temperature, producing increasing noise with increasing temperature (Sternberg, 2012).

This dark current noise is shown graphically in Figure **10** below.

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Figure **10:** Graphically depicts dark noise versus temperature in degrees Celsius. This figure shows how the dark current noise varies as a function of **CCD** temperature. The low temperature regime shows markedly reduced dark noise, again showing the importance of cooled CCDs for improved spectral resolution (Sternberg, 2012).

2.3.1.3. Total Noise

As stated above, the total noise is the sum of several noise sources, most notably the readout and dark current noises. The analysis presented **by** Sternberg (2012) focuses on the larger magnitude sources of readout and dark current noises.

Figure **11** below shows a plot of the total noise as a sum of the readout noise (Figure **9)** and dark current noise (Figure **10)** sources, as derived in the total noise equation above.

Figure 11: Graphically depicts total noise versus temperature in degrees Celsius. This plot shows the increase in total noise as a function of CCD temperature. The noise greatly increases with increasing temperature, especially above -75"C (Sternberg, 2012).

The total noise, here depicted as a sum of readout noise and dark current noise, is a crucial component in determining the instrument's spectral resolution.

2.4. CCD-based XRF Detection

2.4.1. CCDs for X-ray Fluorescence

To gather information from a **CCD** area detector we count the electrons that are collected in each potential well.

When an *X-ray* photon comes into the **CCD** detector and interacts with the silicon, that X-ray gets converted into electrons. For an X-ray with a particular energy you assume that all of that energy gets converted into an equivalent number of electrons.

On the smallest level, an incoming photon to the semiconductor **will** hit an atom and excite it. The atom is now in an excited state and can kick out an electron and is now a positive ion, which we call a hole, with an electron floating around. This electron-hole pair is called an exoton. This exoton **is** where the information is and it must be captured before it is lost, because all exotons end up being recombined.

In example, for an X-ray with 1keV energy it gets converted and produces an average energy of 3.65eV/electron. Thus, for a 1keV photon:

1000eV \cong 273*electrons 3.65eV/electron*

This cloud of charge gets collected and converted in the **CCD,** with a certain conversion gain, that gives the number of Analog/Digital Units (ADUs).

For example, Fe55 has two peaks in its spectrum – the k α at 5.9keV and k β at 6.4keV. From these energies we can compute the number of electrons that are excited when a $k\alpha$ or $k\beta$ X-ray photon interacts, respectively:

 $\text{Fe55 k}\alpha$ $\frac{5900eV}{3.65eV/electron} \cong 1616electrons$

$$
Fe55 \text{ k}\beta \quad \frac{6400 \text{ eV}}{3.65 \text{ eV/electron}} \cong 1753 \text{ electrons}
$$

This may seem like a large number of electrons, but in fact, the well-depth of CCDs can be anywhere from **50,000** to **150,000** electrons. The well-depth of a **CCD** dictates the number of electrons it can collect in a pixel before it starts to spill over into the next pixel. So this is actually a very narrow range of ADUs in the system.

2.4.1.1. Point Source Imaging

Some X-ray events **will** land neatly in one pixel and these are referred to as grade-zero events. In these cases, the electrons that come in all get deposited on one pixel meaning there is total charge collection. In this case, the ka and $k\beta$ peaks will show up as very sharp lines.

Occasionally there will be splits, where an event lands between pixels and the energy from that X-ray event is transferred into two, or more, neighboring potential wells. Say, for example, that **3/4** of the event falls in one pixel and $\frac{1}{4}$ of the event falls in another. In this case, the ka peak will have two energies and so the net effect is that the histogram will not show one sharp line. In this case we **will** simply sum the electrons from both pixels for that event. For this reason, we must always first find the centroid of each X-ray event.

Since X-ray events are small and sparse on a **CCD** area a 3x3pixel area is margin enough to center around any particular X-ray event. Thus, summing up all the electrons in that 3x3 box of pixels **will** give the count value of that particular event.

However, XRF imaging of a planetary body is not point-source imaging but rather continuous source imaging.

2.4.1.2. Continuous Source Imaging

Continuous source imaging can be considered a massive integration **of** point sources all interacting with the **CCD** continuously upon the detector area. There **will** no longer be clean peaks and clean, sparse X-ray events in an otherwise blank field. The conditions when imaging Asteroid **1999 RQ36** will be that of many elements fluorescing X-rays at the REXIS **CCD** detector all immediately next to each other. To properly analyze this area in order to determine what elemental peaks are present in our continuous spectrum **will** take some mastery and familiarity.

In this case, a simple 3x3 pixel box size will not be sufficient. There **will** be a certain surface resolution that **will** need to be optimized against a certain spatial resolution in order to be able to distinguish individual peaks within a spectrum within an area of pixels. This will be discussed in detail in Section **3.1.**

2.4.2. Spectral Resolution **FWHM**

The spectral resolution full-width half-maximum (FWHM) can be expressed as (Janesick, 2001):

$$
\Delta E_{FWHM} = 2.345 \times 3.65 \sqrt{N^2 + \frac{FE_x}{3.65}}
$$

In this equation the FWHM is measured in eV, the 2.345 term models the Gaussian distribution of the spectral resolution, **3.65** is measured in eV/e-, *F* is the Fano Factor, or **0.1** for silicon, and both

N and E_x are measured in ϵ , since they represent the number of electrons of noise and the number of electrons produced for an X-ray with energy E_x , respectively (Sternberg, 2012).

By combining the FWHM equation with the expressions for read noise, σ_{read} , and dark current noise, σ_{dark} , one can derive an expression relating the FWHM spectral resolution as a function of temperature and incoming X-ray energy. The resulting expression **is:**

$\Delta E_{FWHM} = \sqrt{0.0372T + 2.4264 \times 10^{12} T^{1.5} e^{-6.9954 \times 10^{3} T} + 2.0074 E}$

The above expression has only one term dependent on the energy, *E,* and two terms dependent on temperature, *T.* This expression allows for the maximum noise to be found for a given X-ray energy to meet a particular FWHM spectral resolution requirement. Consequently, the REXIS requirement REX-3.1.1 can be expanded to include the maximum noise permitted in order to abide **by** the spectral resolution requirements.

Table 5: Maximum noise levels associated with REXIS spectral resolution requirements.

FWHM (eV)	X-ray Energy (keV)	Maximum Noise (e RMS)
< 80	$\rm 0.5$	8.58
< 100		10.45
< 120		11.91
< 150		14.05
200		19.53

(Sternberg, 2012)

It is clear from Table **5** that the noise levels that are required to meet and exceed the FWHM requirements are well above any of the noise levels seen through the analysis and graph shown in Figure **11.** Therefore, the noise levels expected from theoretical results are well below the levels at which the FWHM requirements would be exceeded (Sternberg, 2012).

In the spring of 2012 the REXIS Science Team (Megyery, REXIS Science Team; **ETU-1:** CCID-41 Laboratory Experimentation) conducted radiation measurements of a CCID-41 with an ⁵⁵Fe point source to allow the team to model **CCD** energy resolution performance at **6** keV as a function of temperature.

To correlate the data from the analytical model conducted **by** Sternberg (2012) with the experimental data, from the experiment designed **by** Megyery and the REXIS Team (2012) and conducted **by** the MIT/Harvard REXIS Science Team (2012), the two sets of data were plotted together.

FWHM as a function of temperature: Experimental data

Figure 12: Graphical results of the correlation of the analytical model **by** Sternberg (2012) with the experimental data of the CCID-41, from the experiment designed **by** Megyery and the REXIS Team (2012) and conducted **by** the MIT/Harvard REXIS Science Team (2012). The experimental result data was fitted with a curve that distinctly shows the FWHM of the detected elemental line significantly increases at detector temperatures above **-75"C** (Image Credit: Inamdar, **2013).**

The experimental data results agree with the graphical result for total noise that above **-75'C** the FWHM [eV] sharply increases and goes beyond the REXIS requirement thresholds. This result on the FWHM as a function of temperature is simply based on the readout noise and dark current noise, both intrinsic to the detector. In order to determine the upper bound on **CCD** operating temperature, the FWHM must be analyzed as a function of temperature. **By** varying the temperature in the FWHM equation above until the REXIS requirement is exceeded, it is possible to determine this upper temperature bound. However, an additional parameter that increases the FWHM of spectral lines is the cross-contamination of neighboring lines which alter the specific energy intensity that is read out **by** the **CCD** and directly affect the accuracy requirement set forth for REXIS in REX-2.1.1. Both of these requirements must be met and the method for this analysis and results is discussed in Chapter **3.**

Chapter 3: Investigation of Spectral Resolution Limits

3.1. Theoretical Global Model for X-ray Line Production

Fluorescent X-rays characteristic of major elements are produced when solar X-rays with energies greater than the energy of the line itself bombard a surface. One group of lines produced in this way, known as K α lines, are those that result from processes that occur in the innermost K-shell. The K-shell absorption edge of an atom is the binding energy of its innermost electron shell. The energies of K-shell absorption edges for major elements on interests (Fe, Ti, Ca, Si, **Al,** and **Mg)** are in the soft X-ray region of interest (P. Clark J. T., **1997).** In order to eject a K-shell electron from an atom through photo-absorption, thereby causing the emission of a fluorescent $K \alpha$ X-ray, incident X-rays must have energies greater than K-shell binding energy. Thus, only X-rays with energies greater than the energy of the K-edge, and greater than the energy of the fluorescent line as well, are capable of exciting that element. Increasing the intensity, or level of the source, meaning increasing the level of solar activity, translate into the achievement of better statistics, shorter integration times, and hence higher resolution maps. As a consequence of the inverse relationship between solar energy and output, production of the higher energy XRF lines for the heavier elements is significantly reduced relative to the production of lines for lighter elements. Thus, longer integration times are required to achieve the same levels of sensitivity for the heavier elements. Longer integration for orbital data results in inherently lower spatial resolution, as well.

3.1.1. X-ray Fluorescence Equation

The method for calculating the X-ray fluorescence for the K α line for each element, as well as the coherent and incoherent scatter in the vicinity of the line (determined **by** integrating the output flux across the line' s full width at half maximum as determined **by** its resolution) is presented in Clark and Trombka (P. Clark **J.** T., **1997).**

The following equation gives the relationship used to model fluorescent line production, I_F , for each element of interest (R. Jenkins, **1967;** Clark, **1979).**

$$
I_F = \omega_j g_j \frac{r_j^{-1}}{r_j} \frac{d\Omega}{4\pi} \frac{C_j}{D_j} \int_{\lambda_{min}}^{\lambda_{edg}} J(\lambda) \cos \alpha \frac{\mu_j(\lambda)}{\sum_{i=1}^N \left[\mu_i(\lambda) + \mu_i(\lambda_j) \frac{\cos \alpha}{\cos \beta}\right]} d\lambda
$$

Table **6:** Definition of model parameters used to model fluorescent line production

(Clark, **1979)**

The fluorescent flux is calculated by integrating over a range of wavelengths, λ_l , from a minimum wavelength above which fluorescence is empirically determined to become negligible to the absorption edge. The Sun's soft X-ray output, variable **J,** is calculated as a function of energy with a polynomial least squares fit to published estimates of solar output as shown in Table 2 **(P.** Clark **J. T., 1997).** The strength of the source, J, depends on the complement of the incident flux angle of the source, α ; thus, J is multiplied the cosine of α . Power laws, as shown in ST2, are used to describe an element's X-ray photoabsorption capability, **i,** as a function of energy (Clark, **1979;** Hubbell, **1969).** The series of power laws that predict photoabsorption for each element each apply within the wavelength limits indicated, the limits representing absorption edges for that element. The mass absorption, μ_i , of the fluorescing element, j , is directly related to its fluorescence intensity. C_i is the concentration of the fluorescing element. The mass absorption, μ_i , by other elements, *i*, which are major components of the surface with concentrations C_i , is inversely related to mass absorption at each wavelength, μ_i , and to fluorescence at the wavelength of the chosen line, μ_i . The ratio between the complements of the arriving flux angle, α , and the departing flux angle, β , is also part of the mass absorption term. When REXIS is nadir-pointing, the complement of the departing flux angle, β , is always 0° .

Table **7** shows the value of X-ray production parameters for the elements considered across REXIS' range. Fluorescence yield, jump ratio (K-shell absorption probability), and K-shell hole transfer factor are all probability factors associated with given elements. Fluorescence yield systematically improves as the atomic weight increases. The energy angular distribution term depends on the relationship between the detector area and footprint area, which itself depends on the altitude and angular field of view of the detector.

--

Table **7:** X-ray production parameters for elements considered in calculations

Element	Atomic	Atomic	Absorpt	Κα		Fluore Jump	Mass Absorption $(\lambda in A)$	
	Number	Weight	ion Edge λ_{edg}	Line	Ratio r_i	scence Yield ω_i	Power Law	Power Law
Ni	28	58.7	1.49				113.0($\lambda^{2.75}$) @ 0 < λ < 1.49	$15.3(\lambda^{2.75}) \otimes 1.49 < \lambda < 12.3$
Co	27	58.9	1.61				$100.0(\lambda^{2.78}) \ @ \ 0 < \lambda < 1.61$	$13.6(\lambda^{2.63})$ @ $1.61 < \lambda < 13.4$
Fe	26	55.9	1.74	1.94	8.1	0.342	95.3($\lambda^{2.66}$) @ 0 < λ < 1.74	13.2($\lambda^{2.59}$) @ 1.74 < λ < 14.7
Mn	25	54.9	1.90				$86.0(\lambda^{1.90}) \ @ \ 0 < \lambda < 1.90$	$11.4(\lambda^{2.63}) \otimes 1.90 < \lambda < 16.1$
Cr	24	52.0	2.07				76.0($\lambda^{2.69}$) @ 0 < λ < 2.07	$10.6(\lambda^{2.58})$ @ 2.07 < λ < 17.9
Ti	22	47.9	2.50	2.75	9.1	0.221	$61.3(\lambda^{2.34})\otimes 0 < \lambda < 2.50$	$7.93(\lambda^{2.63})$ @ 2.50 < λ < 22.0
Ca	20	40.1	3.07	3.34	8.6	0.163	52.3($\lambda^{2.69}$) @ 0 < λ < 3.07	7.79($\lambda^{2.53}$) @ 3.07 < λ < 28.3
K	19	39.1	3.44				$43.9(\lambda^{2.75}) \otimes 0 < \lambda < 3.44$	$5.50(\lambda^{2.61}) \otimes 3.44 < \lambda < 21.5$
S	16	32.1	5.03	5.37	10.3	0.082	$27.5(\lambda^{2.75}) \otimes 0 < \lambda < 5.03$	$2.85(\lambda^{2.72}) \otimes 5.03 < \lambda < 18.5$
P	15	31.0	5.79				22.2($\lambda^{2.75}$) @ 0 < λ < 5.79	2.23($\lambda^{2.75}$) @ 5.79 < λ < 15.5
Si	14	28.1	6.75	7.12	11.2	0.044	$19.9(\lambda^{2.84}) \otimes 0 < \lambda < 6.75$	$2.78(\lambda^{2.55})$ @ 6.75 < λ < 45.0
Al	13	27.0	7.96	8.34	11.9	0.038	$14.3(\lambda^{2.83})\otimes 0 < \lambda < 7.96$	$1.90(\lambda^{2.67})$ @ 7.96 < λ < 45.0
Mg	12	24.3	9.51	9.89	12.8	0.030	11.4($\lambda^{2.80}$) @ 0 < λ < 9.51	$1.60(\lambda^{2.56})$ @ 9.51 < λ < 45.0
Na	11	23.0	11.6				$8.00(\lambda^{2.95}) \ @ \ 0 < \lambda < 11.6$	$1.24(\lambda^{2.50}) \otimes 11.6 < \lambda < 45.0$
Ω	8	16.0	23.3				$2.95(\lambda^{3.00})\ @\ 0 < \lambda < 9.6$	$10.9(\lambda^{2.48})$ @ 9.60 < λ < 23.3
C	6	12.0	43.7				$1.24(\lambda^{3.06})\ @\ 0 < \lambda < 18.0$	$8.00(\lambda^{2.36}) \ @ \ 18.0 < \lambda < 45.0$

(Clark, **1979)**

In other terms, if *E* is the X-ray energy and the incident intensity of parallel solar X-rays on the surface of an atmosphereless body is $I_0(E)$ (photons $\,$ cm⁻² $\,$ s⁻¹) then the intensity of fluorescence from a given X-ray line (photons cm⁻² s⁻¹ steradian⁻¹) can be determined in terms of fundamenta physical parameters **by** the relationship shown in the following equation, adapted from Clark and Trombka (P. ClarkJ. T., **1997; G.** W. Fraser, 2010).

$$
I_{line} = \frac{1}{4\pi} \omega_i g C_i \left(\frac{r-1}{r}\right) \int_{E_{abs}}^{\infty} I_0(E) \rho \cos \theta \frac{\left(\mu(E)_i\right)}{\mu(E) + \mu(E_{line})(\cos \theta) \left(\cos \theta\right)} dE
$$

Table **8:** Defines each of **the model parameters in the X-ray Fluorescence Line Intensity equation above**

(P. Clark **J.** T., **1997)**

50

The equation above also assumes a perfectly smooth and homogeneous planar surface; no allowance is made for effects due to the real properties of the regolith, including surface roughness and packing density. Also unaccounted for are shadowing effects due to large-scale topography. According to Fraser, et al. **(G.** W. Fraser, 2010), energy independent effects of the regolith and topography (e.g. shadowing) may be removed **by** the use of elemental abundance ratios (e.g. **Mg/Si,** Al/Si) instead of absolute abundances, however the use of ratios cannot remove any effects for which there is an energy dependency, and it has been shown in preliminary measurements that for large values of θ and ϕ the measured ratios of elemental line intensities varies as a function of surface roughness **by** 10-20% **(J.** Naranen H. P., **2007; J.** Naranen H. P., **2008).** These effects are mentioned in Appendix B.2. Work is continuing at the University of Leicester, the University of Helsinki and Birkbeck College to quantify the effects of regolith properties on line intensities, in order to provide a more complete interpretation of X-ray fluorescence data from planetary surfaces **(G.** W. Fraser, 2010).

3.2. Developing Inputs for Spectral Resolution Modeling

Spectral resolution modeling determined and verified the detector requirements for temperature, spectral resolution, quantum efficiency, **by** their relation with accuracy of line ratio measurements (accuracy requirement of 10%). The spectral modeling also determined the spectral contamination of each X-ray fluorescent line. The results of the spectral modeling will be used to provide an estimate for photon counts of each X-ray fluorescent line, which can be used for input of imaging and collimation simulation and analysis.

As introduced in Chapter 2, **CCD** performance depends on the detector's physical state which has a profound effect on the output scientific data. As **CCD** performance degrades our ability to measure spectral lines and distinguish neighboring lines from one another will degrade. **CCD** noise depends primarily on **(1)** the Energy of the incident radiation and (2) the Temperature of the **CCD.** In order to meet REXIS requirements REX-2.1.1 and REX-3.1.1 these values need to be quantitatively analyzed and the **CCD** temperature limit needs to be set in order to meet the **10%** accuracy of the incident energy measurement requirement. Experimentation results for the CCID-41 yielded FWHM as a function of detector temperature (Figure 12). Chapter 2 also introduced that noise is dependent on incident X-ray energy and **CCD** temperature.

3.2.1. Model Input Development

Using the model solar spectrum minimum/quiet solar state (Figure **3)** (Lucy Lim, 2012) as a guide, the ideal Class-A X-ray solar spectrum was derived **by** Allen (2012) and is shown in Figure **13.**

Counts as a function of energy

Figure **13:** Ideal solar spectrum for Class-A solar flare state used as input into the XRF Equation. REXIS elemental lines of interest are vertically marked on the graph. This ideal solar spectrum was used as input into the XRF equation to generate the ideal chondrite X-ray spectrum for a Class-A solar state (Image Source Credit: Allen, 2012; Image Source Edit: Megyery, **2013).**

Figure **13** shows the ideal solar spectrum for Class-A solar *flare* state and was used as the input solar spectrum into the XRF Equation to yield the ideal chondrite spectrum as shown in Figure 14.

Figure 14: Ideal chondrite spectrum for Class-A solar flare state as **output from** the XRF **Equation.** REXIS elemental lines of interest are vertically marked on the graph. This spectrum does not include any noise effects that would affect the FWHM of lines or the contamination of counts from neighboring lines (Image Source Credit: Allen, 2012; Image Source Edit: Megyery, **2013).**

Figure 14 shows the ideal solar spectrum for Class-A solar flare state and was used as the input solar spectrum into the XRF Equation to yield the ideal chondrite spectrum. This ideal chondrite spectrum is used as input into the spectral resolution model to be discussed in Section **3.3.**

3.3. Spectral Resolution Modeling

The full spectral resolution model results define the necessary **CCD** conditions to meet the REXIS accuracy requirement REX-2.1.1 and the spectral resolution requirement REX-3.1.1; the model has two main inputs. The first input is the results of the CCID-41 laboratory experiment (Figure *12),* accounting for the intrinsic detector properties contributing noise to the spectrum. The second input is the ideal chondrite spectrum (Figure 14), developed from the Class-A solar flare state which is the expected state when REXIS encounters **1999 RQ36.** These two inputs are then put into the spectral resolution model to generate a more accurate looking spectrum with Gaussian-like peaks, unlike the ideal chondrite spectrum (Figure 14). These peaks are then broadened in the model until the model 'breaks' and neighboring peaks are no longer discretely discernible.

3.3.1. Curve Contamination and Measurement Accuracy Requirement

As has been discussed earlier, the FWHM of spectral lines is not only dependent on **CCD** detector noise. To mitigate this, REXIS has line measurement accuracy requirement of **10%.** This section graphically and quantitatively defines the derivation of this **10%** accuracy requirement.

Unfortunately, due to intrinsic detector noise, the X-ray detection results of REXIS **will** not be sharp, distinct peaks as shown in the ideal chondrite spectrum of Figure 14. Rather, we will see Gaussian shaped lines centered about a specific Energy value with a peak Intensity. As many of these lines are very near each other, the counts from one line will contaminate the counts of neighboring lines as outlined in the following model developed **by** Inamdar **(2013).**

The model begins **by** choosing a width or region of interest about a peak; this width can be chosen arbitrarily, but is typically at FWHM. Let this total width be given by $2\varepsilon_i$ for line i. We are in principle calculating how many counts are contaminated between adjacent lines. For two lines, ideally, heights of $I_{0,1}$ and $I_{0,2}$, the actual distribution will be, due to line broadening from natural effects and detector effects, $I_1(E)$ and $I_2(E)$. The associated peak energies are given by $E_{0,1}$ and $E_{0.2}$

The contamination from one line *i* into the line *j*, C'_{ij} into the other is given by:

$$
C'_{1,2} = \int_{E_2 - \varepsilon_2}^{E_2 + \varepsilon_2} I_1 dE
$$

$$
C'_{2,1} = \int_{E_1 - \varepsilon_1}^{E_1 + \varepsilon_1} I_2 dE
$$

The contribution from line i to itself is given by:

$$
C_i = \int_{E_i - \varepsilon_i}^{E_i + \varepsilon_i} I_i dE
$$

^Agraphical representation of this contamination of counts of neighboring lines is given in Figure **¹⁵** below.

Figure 15: Graphical depiction of curve contamination of counts of individual curves within the FWHM of neighboring curves. In this graphic the curve's FWHM is used as the region of interest about the peak and the differen regions in each show the contamination from Curve 1 in Curve 2 and Curve 2 in Curve 1. The spectral resolution model
described in this section provides the quantitative solution to accounting for this contamination in orde accuracy requirement set for REXIS (Image Credit: Megyerv, 2013).

To determine the acceptable spectral resolution, we pick two lines and form widths (usually the FWHM) around these peaks. We know the ideal line heights, and, in the perfect case, will want to measure the ratio of the heights $I_{0,2}/I_{0,1}$. What, in reality, we will have are measurements of the contamination as above:

$$
C_{1, total} = C'_{2,1} + C_1
$$

$$
C_{2,total} = C'_{1,2} + C_2
$$

The accuracy, η , is then set by comparing the actually measured counts (with contamination) with the ideal line ratios:

$$
\left| \frac{I_2/I_1}{(C'_{1,2} + C_2)/(C'_{2,1} + C_1)} \right| = 1 + \eta
$$

We require, then, that:

$$
\left|\frac{I_2/I_1}{(C'_{1,2}+C_2)/(C'_{2,1}+C_1)}\right|-1<\eta^*
$$

Where η^* is the accuracy as determined by the REXIS requirements, according to requirement REX-2.1.1 that accuracy is **10%.**

For reference, the *Ka* Line energy value in keV and the FWHM for the major elements of REXIS interest (based on requirements) are given in Table **9** below.

(Lawrence Berkeley National Laboratory Center for X-ray Optics and Advanced Light Source, **2009)**

3.3.2. Spectral Resolution Limits

The spectral resolution model, developed jointly **by** Hong (2012) (in IDL) and Inamdar **(2013)** (in MATLAB) (Model code files produced in MATLAB (Inamdar, **2013)** available in Appendix **C.),** generates a more accurate looking spectrum with Gaussian-like peaks, unlike the ideal chondrite spectrum (Figure 14). These peaks are then broadened in the model and these broadened cases are overlaid on the image to show where the model 'breaks' and neighboring peaks are no longer discretely discernible.

Figure 16: Graphical results of the spectral resolution model in which individual peaks are broadened until the model
'breaks' and elemental lines of interest are no longer discernible. Marked on the graphic are the XRF li substantially **by** approximately **64%.** These results do not include contributions from background radiation (the degrading orange line) and is a subject of ongoing work (Hong, 2012; Inamdar, **2013).**

Figure **16** shows the graphical results of the spectral resolution model **by** broadening the peaks (as a function of increased CCD detector temperature) until the model result 'breaks' and the accuracy
requirement is no longer met. Degradation of the lines was commensurate to the detector state per the analytical model for CCD response. At a cool -70°C detector temperature Aluminum can be discerned. The plot shows incremental increase in detector temperature up to **-50'C by** which point the peaks' FWHM have all broadened **by** approximately 64%. Being that the detection of Aluminum is not a science requirement of REXIS but rather a science aim, we can see that temperature up to **55'C** would be sufficient to meet REXIS requirement REX-3.1.1.

3.3.3. Line Measurement Accuracy Limit

By combining the results of the student experiment on the CCID-41's line detection accuracy (Megyery, REXIS Science Team, 2012) with the equations developed for FWHM as a function of detector temperature and line energy (Sternberg, 2012), we are able to plot the line measurement accuracy of elemental abundance ratios as a function of detector temperature. The results of this combination from the spectral resolution model are shown in Figure **17** below.

Figure **17:** Graphical representation of the line measurement accuracy of elemental abundance ratios as a function of **CCD** detector temperature. Elemental abundance ratios (Mg/Si, S/Si, Fe/Si) are plotted for their accuracy of the line ratio measurement as a function of detector temperature in **"C.** The pink horizontal line represents the **10%** accuracy requirement set forth **by** REXIS requirement REX-2.1.1. The intersection of the **10%** accuracy requirement with the Mg/Si ratio is the point at which that detection limit of that ratio would surpass the requirement and no longer **be** discernible. This corresponds to the **red** vertical line at **just** above -540C. Thus, a detector temperature requirement of **-55"C** would suffice to fulfill REXIS requirement REX-2.1.1 (Image Credit: Inamdar, **2013).**

Figure **17** graphically depicts the line measurement accuracy of elemental abundance ratios **(Mg/Si,** S/Si, Fe/Si) as a function of **CCD** detector temperature. The REXIS line measurement accuracy requirement of 10% is marked with the horizontal magenta line at $\sim 10^{-1}$ on the vertical axis. The intersection of that line with the Mg/Si ratio is marked **by** the vertical red line just above -54*C on the horizontal axis. At temperatures warmer than this intersection not only is the accuracy requirement REX-2.1.1 surpassed but these elemental abundance ratios would be discernible due to the FWHM line broadening effects as depicted in Figure **16.**

3.4. Results and Recommendations

The results of the spectral resolution model are two-fold and both results indicate a. nearly exact maximum threshold for **CCD** detector temperature in order to remain within REXIS requirements.

In Figure **16** we graphically see the broadening of several elemental line emissions peaks until the model 'breaks' and the elemental lines of interest are no longer discernible. At **-55"C** the peak for Aluminum is entirely buried within the broadening of neighboring lines **Mg** and Si. At temperatures beyond **-55'C** all four elemental lines have broadened substantially **by** approximately 64% and these results do not include contributions from background radiation. The line-broadening spectral model indicates that the CCD detector should remain at a constant temperature below -55°C in order to remain within the FWHM spectral resolution requirement for REXIS, REX-3.1.1, and be able to obtain useful scientific spectral data.

In Figure **17** we graphically see the representation of the line measurement accuracy of elemental abundance ratios as a function of **CCD** detector temperature. Elemental abundance ratios **(Mg/Si,** S/Si, Fe/Si) are plotted for the accuracy of their measurement as a function of detector temperature in **'C.** The main result of this graphic indicates that the **10%** line measurement accuracy requirement is surpassed **if** the **CCD** detector temperature exceeds **-55"C.** At temperatures warmer than this that ratio would surpass the REXIS science requirement REX-2.1.1 and no longer be discernible.

The results of the student experiment (Megyery, REXIS Science Team **ETU-1:** CCID-41 Laboratory Experimentation, 2012) when combined with the equations for total intrinsic detector noise (Sternberg, 2012) graphically indicate in Figure **11** that total noise versus temperature remains steady until approximately **-75'C.** Above **-75"C** the total noise greatly, although linearly and steadily, increases with detector temperature. This graphic result indicates that it is recommended, in order to minimize intrinsic detector noise, that the CCDs onboard the REXIS instrument be held at or below \sim -75 $\rm{^oC}$.

The combination of these three results leads to the author's recommendation that a detector temperature requirement be set that the temperature of the **CCD** detector onboard REXIS shall not exceed **-55'C** in order to remain within the constraints of the established REXIS Science requirements REX-2.1.1 and REX-3.1.1.

Further, it is the author's recommendation that a detector temperature goal be set that the temperature of the **CCD** detector onboard REXIS shall not exceed **-75'C** in order to minimize noise contamination that is intrinsic to the CCID-41 detector that has been chosen for the REXIS instrument.

3.5. **Research Needs and Future Work**

Spectral resolution modeling determined and verified the detector temperature requirement and recommendation for temperature as dictated **by** the REXIS spectral resolution requirement REX-**3.1.1** and the accuracy of line ratio measurements requirement (accuracy requirement of **10%)** REX-2.1.1. The spectral modeling also quantified the spectral contamination of each X-ray fluorescent line into its neighboring lines. The results of the spectral modeling **will** be used to provide an estimate for photon counts of each X-ray fluorescent line, which can be used for input of imaging and collimation simulation and analysis.

Further, as mentioned with Figure **16** above, those results do not include contributions from background radiation (the degrading orange line) and is a subject of ongoing work. The basis for developing those models can be found in the following subsections regarding instrument background and minimum detectable flux, Section **3.5.1,** and instrument background scattering, Section **3.5.2.**

3.5.1. Instrument Background Levels and Minimum Detectable Flux

The collecting power of an X-ray Spectrometer can be described in terms of the product of effective area and FOV – which in astronomy is known as Grasp, $G(E)$ (units of cm^2sr) – derived below, for a telescope. Where X-rays in the energy interval $E, E + dE$ are emitted from an extended source (i.e. the surface of **1999 RQ36)** and are incident on the instrument aperture with a flux *I(E)* (in *phot* \Box *ns cm*⁻²*s*⁻¹*sr*⁻¹*keV*⁻¹) then the number of detected photons *C* detected in time *t* is given **by:**

$C = I(E)G(E)t$ *dE*

The sensitivity of a photon-detecting instrument can be quantified **by** the minimum detectable flux I_{min} , which is the minimum signal flux from a source that can be distinguished from the background B. Fraser et al. [18] assume than an isotropic flux *I* (in units of *photons cm*⁻²*s*⁻¹*sr*¹) of fluorescent X-rays at a discrete energy (in keV) from an extended source is incident on the front of an instrument with a grasp **G.** The instrument has a focal plane detector with an area for photon detection *Adet* and energy resolution *SE.* The detected fluorescent X-ray signal is accompanied **by** a background of solar X-rays which are scattered from the surface under examination, denoted **by** I_{scat} (in units of *photons cm*⁻²*s*⁻¹*sr*⁻¹*keV*⁻¹).

Although features in the scattered continuum may contain information on surface properties and absorption features in the continuum may yield compositional information, in terms of measure fluorescence spectra the continuum constitutes a background. The detector also has a non-X-ray background component, caused by charged particle events B_p (units of *events* $cm^{-2}s^{-1}keV^{-1}$). For a cooled, photon-counting detector, intrinsic thermal noise is negligible and the scattered X-ray component and the particle-induced background **will** dominate. **If** a signal-to-noise ratio *R* is required for the detection of an X-ray line, then I_{min} may be defined as the flux that, in a given

integration time, produced a count R standard deviations of B above its mean. I_{min} for a given value of *R* is given **by:**

$$
I_{min} = \frac{R\sqrt{\delta E(B_p A_{det} + I_{scat}G)}}{G\sqrt{t}}
$$

The particle-induced X-ray background B_p results from a combination of both high-energy cosmic rays and lower energy (10-100MeV) protons ejected from the Sun. To reduce *Bp* REXIS's detectors will be shielded.

3.5.2. Instrument Background Scattering

The following equations show the relationship used to model coherent, I_R , and incoherent, I_C , scatter **(J.** Hubbell, **1975; D.** Cromer, **1965)** which constitutes instrument background, or noise. Coherent scatter refers to the condition when the wavelength of the radiated photon energy is equal to that of the incident photon energy. Incoherent scatter refers to the condition when the wavelength of the radiated or scattered photon energy is not equal to that of the incident photon.

The following equation is used to determine coherent scatter at a particular wavelength (Hubbell, **1969).** Total coherent scatter associated with a particular energy, *E',* is determined **by** doing a double integration across a range of energies which represent the full width at half maximum for that energy, as determined **by** the energy resolution shown in Table **6.** The Gaussian detector response function (R. Hoover, **1972)** *P(E, E'),* of the instrument is considered in that integration. Then integration is done again to sum the contributions made at each energy, *E.*

$$
\sigma_R(\lambda) = 0.0239(cos^2\theta + 1)\frac{d\theta}{4\pi} \frac{1}{D} J(\lambda) \cos(\alpha) \frac{\sum_{i}^{N} \left[\left(\frac{C_i}{W_i} \right) f_i^2 \right]}{\sum_{i}^{N} \left[\mu_i(\lambda) (1 + \frac{\cos \alpha}{\cos \beta}) \right]}
$$

$$
I_R = \int_{E_{min}}^{E_{max}} \int_{E_{min}}^{E_{max}} \sigma_R(E, E', \theta) P(E, E') dE dE'
$$

The following equations are used to determine incoherent scatter at a particular wavelength (Hubbell, **1969)** and to determine the integrated incoherent scatter, analogous to the above equation for coherent scatter.

$$
\sigma_C(\lambda) = 0.0239 \frac{d\theta}{4\pi} A(\lambda_C) J(\lambda_C) \cos(\alpha) \frac{\sum_i^N \left[\frac{C_i}{W_i} S_i\right]}{\sum_i^N \left[\mu_i(\lambda) + \mu_i(\lambda_C) \frac{\cos \alpha}{\cos \beta}\right]}
$$

$$
I_C = \int_{E_{min}}^{E_{max}} \int_{E_{min}}^{E_{max}} \sigma_C(E, E', \theta) P(E, E') dE dE'
$$

Finally, A , k , and λ_C are defined below:

$$
A(\lambda) = (1 + k(1 - \cos \theta))^{-2} \frac{1 + \cos^2 \theta + k^2 (1 - \cos \theta)^2}{1 + k(\cos \theta)}
$$

$$
k = \left[511 \left(\frac{\lambda}{12.4} \right) - (1 + \cos \theta) \right]^{-1}
$$

$$
\lambda_c = \lambda - \left(\frac{12.4}{511}\right)(1 - \cos\theta)
$$

The following equation gives the Gaussian response of the detector (R. Hoover, **1972).**

 $\ddot{}$

$$
P(E, E') = \frac{exp - \left[\frac{(E - E')}{2\sigma_m^2 E'}\right]}{\sigma_m (2\pi E')^{0.5}}
$$

Table **10** defines additional parameters used in calculating scatter (P. Clark **J.** T., **1997;** Clark, **1979;** Hubbell, **1969).**

Table **10:** Additional terms used in the above equations to model scatter from the surface

(Hubbell, **1969)**

Source flux and energy absorption need to be considered, just as in fluorescence line production calculations. In addition, the backscatter angle, θ , is significant because only X-rays which are deflected back from the surface at the proper angle **will** be able to enter the detector. At the subsolar point (the Sun is directly overhead), that angle is 180° (relative to the incident solar flux angle). The constant **0.0239** is the product of the multiplication of Avogadro's number **by** half of the square of the electron radius. This constant puts the scatter intensity on a molar rather than an atomic basis. **A** spline fit was done to published tables of coherent and incoherent scattering factors, f_i and S_i respectively, which vary as a function of the parameter q (J. Hubbell, 1975; D. Cromer, **1965).** Additional parameters need to be considered in the case of incoherent scatter: the wavelength of the incident flux, λ_c , called the Compton wavelength, is shorter than the wavelength of the resulting scattered flux, λ_i .

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Appendix A: Context of Student Work

This work is a continuation of research begun during the 2011-2012 academic year. The MIT Aeronautics **&** Astronautics Department's year-long capstone project course(s) known as **(16.831/12.431)** Space Systems Engineering and **(16.832/12.432)** Space Systems Development was a group of approximately 20 senior-level students introduced to the REXIS Instrument project.

During the first semester, Fall **2011,** a great deal of background research and study was conducted **by** all students to understand the level of science and engineering needed to make a real contribution to the design and development of the instrument. In the Spring of 2012 a goal was set to build the first Engineering Test Unit **(ETU-1)** of the REXIS Instrument with the goal of the REXIS **ETU-1** to detect X-rays from two point sources and a continuous source and read out that data to produce the respective elemental lines on a histogram.

The author was the only student registered for the course from the MIT **EAPS** Department and thus took on the role as the Science Team Lead. She even served as student representative at the **All-**Instruments Team Meeting in Tucson, Arizona in November 2011 to introduce the class's progress on the Instrument at the OSIRIS-REx headquarters at the University of Arizona.

As of the Fall of 2012 the REXIS Instrument design and development had become too advanced to introduce to a new group of students and the project has remained in the hands of the faculty, researchers, staff and graduate students still involved. Fall 2012 served as the author's final semester of her MIT undergraduate career and she opted to remain on the team as a contributor to the development and analysis of REXIS science. Thus, the logical progression of the science work on REXIS was to complete her undergraduate thesis requirement for partial fulfillment of the Degree of Bachelor of Science in the Department of Earth, Atmospheric and Planetary Sciences at MIT.

Appendix B: Supplemental Reading

B.1. **REXIS Heritage of CCDs for XRF Detection**

Remote sensing of X-ray emissions is only possible for bodies with little or no atmosphere to absorb these emissions (Goldsten, **1998).** Orbital, rather than **flyby,** missions are preferred for such measurements because of the typically long observation times required. The REXIS CCD-based spectrometer is built on flight heritage that needs to be examined in detail. This examination **will** discuss what XRF is in the context of past missions, how they were developed, the more detailed theory behind the mechanism **by** which each worked, and to outline, with each case study example, the various applications of XRF.

Particularly, each case study has not been without its own set of challenges, some of which the respective science teams were capable of overcoming. Thoroughly understanding the scientific flight heritage of the REXIS Instrument design and science trades can lead to a comprehensive understanding of the X-ray spectroscopy of asteroids and aid in the understanding of the X-ray spectroscopy of REXIS at **1999 RQ36.**

B.1.1. Hayabusa

X-ray fluorescence spectrometry of asteroid 25143 Itokawa was performed **by** the X-ray spectrometer onboard Hayabusa during the first touchdown on **19** November **2005.** The mission's intent was to determine major elemental composition of the asteroid surface and its regional variation **by** observing the X-rays excited **by** solar X-ray irradiation **by** the X-Ray Spectrometer (XRS) on board. Further, they hoped that information on surface roughness would be derived from observations at various phase angles.

B.LL. XRS Instrumentation and Observation

The XRS was an advanced XRF spectrometer with a light-weight **(1.5 kg)** sensor unit based on a **CCD** X-ray detector; this was the first time a **CCD** had been use for such a purpose on a planetary mission (T. Okada e. a., **2006). A CCD** was chosen for use as an X-ray detector to improve energy resolution and to extend detection energy range (T. Okada M. K., 2000). Four **CCD** chips were arrayed to detect X-rays from the asteroid. Another single chip was obliquely mounted to detect Xrays from the standard sample plate that solar X-rays would excite (T. Okada M. K., 2002).

The Standard Sample Plate **(SSP)** was for concurrently calibrating the XRF when it was excited **by** the Sun. The **SSP** was a glassy plate whose composition was intermediate between those of chondrites and basalts. **By** comparing X-ray spectra from the asteroid and from the **SSP,** quantitative elemental analysis could be achieved, although the intensities and spectral profiles of solar X-rays change over time (T. Okada e. a., **2006).**

The XRS instrument was required to have **(1)** energy resolution sufficient to discriminate fluorescent X-rays of major element and reject scattered solar X-rays as backgrounds, (2) energy detection range from **0.7** to **8** keV, **(3)** effective detection area to accumulate statistically significant counts of X-ray photons in a given integration time and (4) concurrent solar X-ray monitoring for quantitative elemental analysis (T. Okada M. K., 2002). There were also several technical constraints on small mass budget, mechanical and thermal conditions that the orbiter undergo during the mission, and mounted position in the Hayabusa orbiter.

The CCD-based XRS consisted of a detection system (XRS-S) and an electronic circuit (XRS-E). The specifications of the instrument are shown in Figure **1** below. To achieve the smaller size and larger active area, a full-frame transfer method was adopted as the **CCD** readout, where no inactive transfer regions or data storage areas were demanded. The active area of each chip was **1** inch square with 1024 x 1024 pixels. Each pixel size was 24 micro-m square. To attain high-energy resolution, the **CCD** chips were operated below 230K to reduce the thermal noise level. The hood was effectively designed for the Radiative cooling. To extend the lower limit of the detection energy range, the windows with **5** micro-m thick beryllium films supported **by** meshed grids and latticed frames of **90%** transparency were used to shield the visible light as well as effectively pass the X-ray photons below **1** keV. The signal readout took place every **8** micro-sec in turn **by** multiplexing the five chips. **16** pixels in a line were binned as a single data to readout in *0.5* sec per chip.

Table **11:** Specifications of the XRS Instrument onboard Hayabusa

(T. Okada M. K., 2000)

The electronic circuit was composed of the **CCD** driver, multiplexer, data handling circuit, thermoelectric cooler driver, **CUP** and interface for command and telemetry, and power supply. The clock patterns from the driver were multiplexed and delivered to five **CCD** chips in turn. The readout signals from each chip were received into the data handling circuit and, after digitized **by** a 14-bit **ADC,** the upper 12 bits were stored in the FIFO. **If** the signals were beyond a threshold level, they were regarded as X-ray events. After a given integration time of **60** sec, they were formatted as **CCSDS** packets and transmitted to the data-handling system of the spacecraft (T. Okada M. K., 2000). The data handling was controlled **by** using the specified 32-bit Hitachi **SH-3 CPU.**

In-flight performance of the XRS was confirmed **by** observations of X-rays from the **SSP,** X-rayemitting bodies such as Kepler's supernova remnants, and X-rays from the far side of the Moon.

In-flight calibration was performed **by** observation of cosmic X-rays along the trajectory of the orbiter. In situ calibration of X-ray fluorescence **by** solar X-ray irradiation into the onboard standard sample was used to improve the accuracy of the quantitative elemental analysis *(T.* Okada M. **K.,** 2000).

During the descent, the solar phase angle **-** the angle between the Sun, Itokawa, and Hayabusa **-** was between **50** and **100.** This small phase angle is favorable for performing XRF spectrometry (T. Okada e. a., **2006).** At larger phase angles, shadows on the rough surface terrain can interfere with the elemental analysis (T. Okada e. a., **2006).** During descent, the XRS continued X-ray spectrometry of Itokawa, and the **SSP** simultaneously excited **by** the Sun, so that the entire asteroid surface was observed in a longitudinal direction due to its rotation.

The effective field of view of the XRS was **3.5' by 3.50,** corresponding to a "footprint" size of **87** m **by 87** m at an altitude of 1.4 km, which became proportionally smaller as Hayabusa descended. Consequently, the XRS observed local areas of Itokawa **550** m **by 298** m **by** 244 m in size. XRS observations were sometimes affected **by** background radiation from space (X-rays and energetic particles) because the footprint was not centered on the asteroid, due to the rotation of the irregularly shaped asteroid and instability of the spacecraft's attitude control that resulted from the failure of two of the three reaction wheels.

B.1.1.2. X-Ray Spectrometer Data and Analysis

The XRS produced a set of X-ray energy spectra of Itokawa and the **SSP,** after onboard processing of data from each **CCD,** every **5** min. Much of the data could not be analyzed due to unfavorable solar activity, but six data sets with good signal-to-background ratios **-** observed during periods of relatively enhanced solar activity **-** were selected.

The descent phase data was the most desirable condition because of relatively high solar activity for X-ray excitation and low altitude to **fill** the **3.5** x **3.5** degrees field of view of the XRS **by** the asteroid surface (T. Okada K. **S., 2007).**

Quantitative elemental analysis of **Mg, Al,** and **Si,** and qualitative analysis of Ca and Fe indicated that the surface of Itokawa has chondritic composition and that LL- or L-chondrites are most likely, but H-chondrites or primitive achondrites cannot be rejected (T. Okada e. a., **2006;** T. Okada K. **S., 2007;** T. Okada K. **S., 2006).**

B.1.2. Suzaku

Suzaku was Japan's 5th X-ray astronomy satellite, an Earth-orbiting observatory imaging various Xray sources launched in **2005.** Suzaku was equipped with an X-ray Imaging Spectrometer **(XIS)** system, consisting of state-of-the-art CCDs optimized for X-ray detection, camera bodies and control electronics (K. Koyama, **2006).**
B.L2.1. XIS Instrumentation and Observation

From Koyama et al. **(2006),** the XIS consisted of four CCDs, each placed at the focus of a dedicated X-ray telescope. The XIS camera body consisted of a hood, bonnet and base. The hood had five baffles to block stray light. The bonnet included a vacuum valve, an optical blocking filter (OBF), calibration sources (55 Fe), a pressure sensor, a paraffin actuator and a door. The base contained an X-ray **CCD** mounted on an alumni **(A120 ³)** substrate and was attached to a heat sink assembly made of copper. **A** Peltier cooler incorporated in the heat sink could cool the detector to its on-orbit operating temperature of **-90'C.** Over the frame-store region of the **CCD,** an aluminum shield with a gold-plated nickel surface treatment was placed to block X-ray irradiation. The surfaces of the substrate, the heat sink assembly and the shield were also plated with gold. The bonnet and the hood were made of **Al** with black surface finish. The inside of the base was Ni-plated, except for the feedthrough plate which was gold electro-plated. Multi-layer insulators to reduce heat-flow from the satellite body covered the XIS sensor.

Each XIS sensor had three 55Fe calibration sources. One was attached to the door, and illuminated the whole imaging area. This calibration source was used on the ground and for initial on-orbit calibration before the door was opened. For normal observation, after the door was opened, this calibration source was out of the field of view of the **CCD.** The other two calibration sources were located on a side wall of the bonnet and illuminated the two far-end corners of the imaging area.

The XIS had four sensors and each sensor had one **CCD** chip, which was a MOS-type three-phase **CCD** operated in the frame transfer mode. The imagine area of the **CCD** had 1024 x 1024 pixels. The pixel size was 24 μ m x 24 μ m, giving a size of 25 mm x 25 mm for the imaging area. These XIS **CCD** detectors were fabricated in the MIT Lincoln Laboratory and are very similar to the ones flying on the REXIS instrument.

The XIS had a high X-ray spectra range in 0.2 to 12 keV energy band, where the bulk of K-shell lines of O-Ni exist.

B.2. Challenges to XRF: Regolith Effects

The Nirinen, et al. **(J.** Naranen H. P., **2008)** study has relevance for the OSIRIS-REx and REXIS mission. REXIS will produce data that shall have well-defined observation angles and also has a large effective area. It can be expected that the regolith effects, **if** not accounted for, will play a significant role as a source for error in the interpretation of the data from the REXIS mission.

Nirinen, et al. **(J.** Naranen H. P., **2008)** presented new measurements and results from their experimental study on how the surface roughness of the regolith affects the interpretation of X-ray spectra obtained **by** orbiting spacecraft. The surface roughness tends to harden the measured spectrum as the illumination angle increases in a way that the traditionally used FPE of X-ray fluorescence examples are cannot cannot be exampled by the predict

Fluorescent soft X-rays from the surfaces of atmosphereless planetary bodies in the inner solar system have been measured **by** instruments on several spacecraft, e.g. Apollo **15** and **16 (I.** Adler **J. G., 1972), NEAR** Shoemaker (L. R. Nittler, 2001), and Hayabusa *(T.* Okada K. **S., 2006).** An X-ray

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spectrometer is currently travelling to Mercury onboard the **MESSENGER** spacecraft **(C. E.** Schlemm, **2007).** The soft X-ray energy region is considered to lie between **0.5** and **10** KeV, where fluorescent X-ray emission follows photoionization **by** incident X-rays and charged particles.

The source of exciting X-rays, in XFS of atmosphere-less bodies, is the solar corona. The Sun is a relatively weak source of X-rays and the fluorescence process is not very efficient and thus studies are limited to the inner solar system, including NEOs such as **RQ36.**

Traditionally, analysis of X-ray spectra has assumed that surfaces can be considered to be homogeneous and plane-parallel media (P. Clark **J.** T., **1997).** This has been a reasonable approximation considering the relatively poor spectral and spatial resolution of the detectors used for XFS studies to date **(J.** Naranen H. P., **2008).** It has been shown, however, that elemental line intensities and their ratios are affected **by** the physical properties of the target surface and the viewing and illumination geometry of observations. An increase in line intensities has been measured as particle size in the sample is decreased and a spectral hardening (enhancement of higher-energy end over the lower-energy end) is observed as a function of increasing phase angle (i.e., the angle between the X-ray source and the detector as seen from the sample).

X-ray fluorescence can be analytically calculated **by** using what is traditionally called the fundamental parameters equation (FPE) of X-ray fluorescence. FPE normally assumes a smooth, flat, and homogeneous surface. As can be seen in the treatment below, the output of the FPE is **highly** dependent on the spectrum of the incident radiation and the elemental composition of the sample. The FPE derivation here follows that of Grieken and Markowicz (Grieken, 2002). Detailed derivation can also be found in, e.g., Criss and Birks **(J.** W. Criss, **1968).**

For a single element *I*, the linear photoionization coefficient $\mu_{photo,i}(E)$ and the total extinction coefficient $\mu_{total,i}(E)$ in ($1/m$) are given by:

$$
\mu_{photo,i}(E)=\sigma_{photo,i}(E)f_i\rho
$$

$$
\mu_{total,i}(E) = \sigma_{total,i}(E) f_i \rho
$$

In the equations above $\sigma_{photo,i}(E)$ is the photoionization cross section, $\sigma_{total,i}(E)$ the total extinction cross section, both in $[m^2/atom]$, ρ the total number density of the material in [atom/m³], and f_i , the number density fraction of element i .

Now, the total photoionization and extinction coefficients, $(\mu_{\text{photo},i}(E)$ and $\mu_{\text{total},i}(E)$, respectively) of the material are:

$$
\mu_{photo,i}(E) = \sum_{i} \mu_{photo,i}(E)
$$

$$
\mu_{total,i}(E) = \sum_{i} \mu_{total,i}(E)
$$

For each photon of energy *E* absorbed **by** the element *i,* the probability of *K1* line fluorescence $(l \in {\{\alpha, \beta\}})$ is:

$$
Q_{i,l} = \omega_K f_{i, Kl} \frac{J_{i,K} - l}{J_{i,K}}
$$

In the equation above $J_{i,K}$ is the jump ratio of the K shell of the element *i*, ω_K is the fluorescence yield of the K shell, and $f_{i,kl}$ is the transition probability.

With the given parameters, we obtain an analytic expression for the first-order K/ fluorescence $F_{i,l,E}(E)$ from element *i* for incident intensity I_o of a single energy *E*:

$$
F_{i,l,E}(E) = I_o(E)Q_{i,l} \frac{\mu_{photo,i}(E)}{\mu_{total}(E)} \frac{1}{\cos \iota + \mu_{total}(E_l) / \mu_{total}(E)} \cos \epsilon
$$

and, finally, for a continuous spectrum we have the total fluorescence intensity $F_{i,l}$ as:

$$
F_{i,l}=\int_0^\infty I_0(E)F_{i,l,E}(E)dE
$$

Realistic regolith-like surfaces are yet to be analytically modeled for FPE **(C. E.** Schlemm, **2007).** Also, secondary fluorescence (fluorescence excited **by** fluorescent X-rays) and contributions from elastic and inelastic scattering need to be included in the future, although their contribution to the overall outcome of the calculations is not very significant **(C. E.** Schlemm, **2007).**

Change of spectral shape (and thus relative intensity ratios) is predicted **by** the FPE to occur as the illumination angle increases. The FPE also describes the apparent discrepancy in the overall behavior of the relative elemental line ratios measured **by** a downward trend setup versus an upward trend setup as the overall behavior is a function of, e.g., the X-ray source spectrum. However, the FPE does not take into account the physical properties of the sample surface, such as surface roughness, porosity, and particle-size distribution.

For the measurement geometry in the experimental setup conducted by Näränen et al (J. Naranen

H. P., **2008),** the most notable effect caused **by** the physical properties of the regolith is shadowing of the incident X-rays (Y. Maruyama, 2008; **J.** Naranen H. P., **2008).** Qualitatively this means that, as the incident angle increases (large ι), the exciting radiation has an increasing chance of entering the sample medium through the side of a particle, a pore, or a roughness feature. The fluorescent X-rays excited **by** this radiation have to travel further inside the medium to be observed **by** a detector looking in the direction of the surface normal than X-rays excited **by** radiation that entered the medium at a point closer to the detector (i.e., closer to the interface between the regolith and free space). The mean-free path of soft X-rays in a regolith-like medium increase as a function of X-ray energy. Thus fluorescent X-rays of higher energies can escape the medium more freely than those with lower energies. This introduces a hardening of the spectrum as a function of the incidence angle. This effect is described **by** the equation used in the derivation of the FPE.

Näränen et al. (2008) report that regolith effects are apparent in each line ratio they studied, except for Ti-K α /Ca-K α where the energy separation between the two lines is small (i.e., only small effects are to be expected). The qualitative conclusion drawn **by** Nirinen, et al. **(2008)** from the measurements in their experiment: as the surface roughness increases, the regolith effects increase and cause hardening of the spectrum

Appendix C: Spectral Resolution Model Computational Methods

C.I. *MATLAB* Source Code: **1** ated -SpectraLRes for_ **Loop**

```
function accuracy = Updated_Spectral_Res for Loop(T)
* Updated spectral resolution code
      Niraj Inamdar
15 JanuarLy 201
Set baseline line against which you want to measure the ratios
baseline_element = 'Si-K';
% Set the allowable accuracy eta
eta = 0.1;
\frac{1}{2} Set input temperature in Kelvin
T = T;
i Bring in information regarding spectral lines of interest;
\frac{1}{2} spectIDs is a structure:
spectIDs.peakID(i) gives Lhe ih peaklines
i spectIDs.energy(i) gives the energy of the ith peak
% spectIDs.elName(i) gives the identity of the ith peak
spectIDs = spectID_info;
# Import chondrite spectrum; this could be substituted for with the solar
% spectrum; see solarSpectRead for more on how to pull that info out.
Cl_data_Ens = importdata('Cl_chondrite_spectrum_E.csv');
eV vals = C1 data Ens(:,2); in electronvolts
eV vals = eV vals (3:end) ;
C_1 data Flux = importdata('C1_chondrite_spectrum_F.csv');
C1 data Flux = xlsread('C1 chondrite spectrum F.csv');
lines = C1 data Flux(:,2);
bkg = C1 data Flux(:,3);totalintensity = lines + bkg;
& Define lines of interest; these must be strings with length between the
'a of 4
lines of interest = ['Mg-K ';'Al-K ';'Si-K ';'S-K ';'K-Ka '];
lines of interest = cellstr(lines of interest);
% Construct a structure with just the lines we want base on the definitions
above
lines to probe = struct('peakID', [l:length(lines_of_interest)],...
      'energy', zeros(1, length(lines_of_interest)), ...
      'intensity',zeros(1,length(lines_of_interest)),'elName',{{}});
for i = 1: length (lines of interest);
      for j = 1:length(spectIDs.elName)
             if strcmp(lines of interest(i),spectIDs.elName(j)) == 1
                linestoprobe.elName(i) = spectIDs.elName(j);
                lines toprobe.energy(i) = spectIDs.energy(j);
```

```
energy index = indexFind(spectIDs.energy(j),eV_vals);
                 lines_to_probe.intensity(i)= total intensity(energy index);
             end
      end
end
sfiqure
semilogy(eV vals, total intensity, 'b')
hold all
lines to probe now has all the information we need
Input model for FWHM as a function of energy; this is from Suzaku. The
I linear regression has an input energy in keV; the output of the model is
is in terms of eV
FWHM en = 11.922*(lines to probe.energy/1000) + 52.989;
Input model for FWHM as a function of temperature; this is from the data
I linefit and the same for each line; we will then take the root square sum
tof these two
a 311.286; b = 31539.04; c = -0.277426;
FWHM T =a*sqrt((2.4264e12*(T.Al.5).*exp(-6995.35963./T) + c.*T + 2.0074*5.19e3)) 
b;
Then the RSS:
FWHM line = sqrt(FWHM en.^2 + FWHM_T.^2);IAlso useful to calculate from Lhe FWHM the sigma of 6he Gaussian; the
formula is FWHM line = 2*sqrt (2 *log (2)) sigma, so thar
sigma line= FWHM line./(2*sqrt(2*log(2)));
* Now generate the Gaussians over the range of interest;
E_range = linspace(min(spectIDs.energy),max(spectIDs.energy),5000);
detector_count = zeros(1,length(E_range));
for i = \overline{1}: length(lines_of_interest)
      detector count = detector_count + ...
              lines_to_probe.intensity(i). *exp(-(E range -
       lines to probe.energy(i)).^2./(2*sigma_line(i).^2));
end
semilogy(E_range,detector_count,'r','LineWidth',2)
ylim([min(total intensity) 10*max(total intensity)])
% We now desire to determine the contamination between the lines of
& interest and the baseline line defined at the top; first, calculate the
& baseline line ratios for the perfect casesi
baseline ratios = zeros(1,length(lines_of_interest));
& Determine intensity, energy, and sigma of baseline element in ideal case
baseline intensity = 0;
E_baseline = 0;
sigma baseline = 0;
for i = 1:length(lines of interest)
       if strcmp(baseline element, lines_of interest(i)) == 1baseline_intensity = lines_to_probe.intensity(i);
              E baseline = lines to probe.energy(i);
              \overline{\text{sigma}} baseline = \text{sigma}line(i);
       end
end
% Determine baseline ratios against which to compare later
for i = 1:length(lines of interest)
       if strcmp(baseline-element,lines-to-probe.elName(i)) == 0
```

```
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```
baseline ratios(i) = lines to probe.intensity(i)./baseline_intensity;

end

end

```
\frac{1}{2} This array gives the limits of integration for counts centered around the
energy center. The first column is the min, the second column is the max,
* the third column is the min index, and the fourth column is the max index
integration limits = zeros(4,length(lines of interest));
for i= 1:length(lines of interest)
       integration_limits(i,1) = lines_to_probe.energy(i) - FWHM_line(i)/2;
       integration limits(i,2) = lines toprobe.energy(i) + FWHM_line(i)/2;
      integration limits(i,3) = indexFind(lines to probe-energy(i) - FWHM line)(i)/2, E range);
      integrationlimits(i,4) = indexFind(lines-to-probe.energy(i) + FWHM line
(i)/2, E_range);
end
Resize due to peculiarities in the indexFind function
integration limits = integration limits(1:length(lines of interest),1:4);
Baseline range: get the integration limits for the baseline
baseline range = zeros(4, 1);
for i = 1:length(lines of interest)
      if strcmp(baseline element,lines-to-probe.elName(i)) == 1
             baseline range = integration limits(i,:);
      end
end
We also define the baseline line distribution
i baseline = baseline intensity*exp(-(E range - E baseline).^2./
(2 * sigma baseline^2));
% And those of all lines
intensities = zeros(length(lines of interest), length(E range));
for i = 1:length(lines of interest)
      intensities(i,:) = \ldotslines to probe.intensity(i). *exp(-(E range -
      lines_to_probe.energy(i)).^2./(2*sigma_line(i).^2));
end
% Calculate ratios of interest for each line of interest against the
baseline (excluding the baseline, of course)
The first column in this array is the ratio against the baseline of
% counts with contamination; the second column is the counts of the line in
its region; the third column is the counts of the baseline in the line
region; the fourth column is the counts of the line in the baseline
* region; and the fifth column is the coutns of the baseline in the
baseline region. We have
C(1) = [C(2) + C(3)] / [C(4) + C(5)]The sixth column eta is the accuracy: baseline ratios(i)/Cl - 1;
% The sixth column eta is the accuracy: (Cl - baseline ratios(i) )/baseline ratios(i)
ratios of interest = zeros(length(lines of interest), 6);
Begin loop to calculate lines with contamination
for i = 1: length (lines of interest)
If this line isn't the baseline
      if strcmp(baseline element,linesto probe.elName(i)) == 0
      Elmin index = integration limits(i,3);
      Elmax index = integration_limits(i,4);
```

```
E2min_index = baseline_range(3);
E2max index = \text{baseline range}(4);
C2 = sum(intensities(i,Elmin index:Elmaxindex));
C3 = sum(i_baseline(Elmin_index:Elmax_index));
C4 = sum(intensities(i,E2min_index:E2max_index));
C5 = sum(i_baseline(E2min_index:E2max_index));
Cl = (C2 + C3) / (C4 + C5);<br>Reta = (C1 - \text{baseline\_ratios(i)}) / \text{baseline\_ratios(i)};
eta = baseline ratios(i)/Cl - 1;
ratios of interest(i,:) = [CI C2 C3 C4 C5 eta];end
```

```
end
```
ratios of interest;

Summarize the accuracies

```
accuracy_summary = struct('elNamePair', {{}}, 'accuracy', zeros(1, length
(lines of interest)));
for i = 1: length (lines of interest)
      if strcmp(baseline element, lines to probe.elName(i)) == 0accuracy_summary.elNamePair(i) = strcat(lines_to_probe.elName
(i),'/',baseline-element); \{ [lines_to_probe.elName(i),'/',baseline_element] };
             accuracy summary.accuracy(i) = ratios of interest(i,6);
      end
end
```

```
accuracy = accuracy-summary;
```
C.2. MATLAB Source Code: run Spectral Res for Loop

```
Uopda -ed Spec, rIi I.,J D
# Updated spectral resolution loop
\frac{1}{2}Niraj Inamdar
15 January O1
temperature = 273 + linspace(-70,-50,300);
eta = .1;
\frac{1}{2} accs = zeros (1, length (temperature));
MgSi = zeros(1,length(temperature));
AlSi = zeros(l,length(temperature));
SSi = zeros(1,length(temperature));
for i = 1:length(temperature)
accs = Updated Spectral Res for Loop(temperature(i));
MgSi(i) = accs.accuracy(l);
AlSi(i) = accs.accuracy(2);
SSi(i) = accs.accuracy(4);
end
spectIDs = spectID info;
accs = Updated Spectral Res for Loop(temperature(i));
figure
semilogy(temperature - 273, abs (MgSi), 'r*', 'LineWidth', 1)
hold on
semilogy(temperature - 273, abs(AlSi), 'g*', 'LineWidth', 1)
semilogy(temperature - 273, abs(SSi), 'b<sup>+</sup>', 'LineWidth', 1)
hline (eta, 'm')
1hline (-. 1,'W')
grid on
xlab = xlabel ('Detector Leiperarure
[C]', 'FontName', 'TimesNewRoman', 'FontSize',18, 'Interpreter', 'Latex');
ylab = ylabel('Line measurement
accuracy', 'FontName', 'TimesNewRoman', 'FontSize', 18, 'Interpreter', 'Latex');
leg = legend('Mg/Si','Al/Si','S/Si');
head = title('Line measurement accuracy as a function of
temperature', 'FontName', 'TimesNewRoman', 'FontSize', 18, 'Interpreter', 'Latex');
set (xlab, 'Interpreter', 'Latex');
set(ylab,'Interpreter','Latex');
set(head, 'Interpreter', 'Latex');
set(leg, 'Interpreter', 'Latex');
%ti_thing = sprintf ('Equilibrium time as a function of force coefficient \gamma for
various $b$ and $k$');
set(gca, 'FontName', 'Times New Roman', 'FontSize', 18);
%MgSiIndex = indexFind(0,MgSi-eta);
v1ine (temperature (M1Si Index) -273);
vline(zeroSearch_Simp(temperature, MgSi-eta) - 273, 'r')
```
C.3. MATLAB Source Code: spectIDs

```
$<br>$ Updated spectral resolution code
\hat{\mathcal{E}}Niraj Inamdar
15 January 2013
energy rang we're interested in probing
Emin
      .65; %In keV<br>= 2.5; %In keV
E_max
```
spectralLines **=** spectIDs;