

REACTOR THERMAL-HYDRAULIC ANALYSIS  
IMPROVEMENT AND APPLICATION OF THE CODE COBRA-IIIC/MIT

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and Master of Science in Nuclear Engineering

ABSTRACT

Several improvements have been made to COBRA-IIIC/MIT. All of the improvements, except for one, have been made in response to recommendations of past research. The improvements are included in a new version of the code as new modeling options. The new modeling options overcome limitations and disadvantages of old modeling options. The improvements are as follows:

1. Addition of a new fuel pin conduction model which includes temperature dependent properties and burn-up dependent gap heat transfer coefficient.
2. Addition of a new heat transfer package which covers a broad range of flow regimes and contains more consistent logic.
3. Addition of a quality dependent mixing model for two-phase flow.
4. Addition of new correlations for BWR CHF and CPR calculation.
5. Addition of new options for using transverse momentum coupling parameters for the single-pass method.

The improvements have been tested individually and during application of the improved code to PWR and BWR test cases. Testing mainly involved comparison of the predictions of different modeling options and in some instances, compari-

son of predictions with experimental measurements. MDNBR, MCPR and MCHFR predictions showed only small sensitivities to the fuel rod and heat transfer modeling options used for the PWR and BWR transient test cases analyzed. Differences in predictions of the old and new heat transfer models resulted in different clad temperature predictions. Clad temperature varies more smoothly in the axial direction when the new heat transfer model is used. The CISE-4 MCPR predictions were in agreement with experimental CHF measurements. Hench-Levy MCHFR predictions were conservative for the CHF test cases. The new transverse momentum parameters had no significant effect on steady state hot channel predictions of the single-pass method.

The improved version of COBRA-IIIC/MIT was applied to BWR bypass analysis. Two small test cases were analyzed to gain an understanding of how to analyze bypass flow in the E.I. Hatch Unit I reactor. COBRA-IIIC/MIT was limited in the power and bypass flow combinations it could analyze. Estimated full power and full flow conditions were found to be outside the zone of COBRA-IIIC/MIT applicability. Transient THERMIT analysis was performed in an attempt to obtain steady state predictions. Oscillation of predictions during this analysis indicated that THERMIT bypass analysis should be performed using a pressure drop boundary condition.

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## I. INTRODUCTION

Thermal-hydraulic analysis of light water reactor (LWR) cores is usually performed using a computer code. Thermal-hydraulic analysis calculates parameters such as temperature, density, or departure from nucleate boiling ratio (DNBR). "Subchannel codes" may be used for this analysis. "Subchannel codes" represent the geometry of a core using coolant and fuel rod nodes. There are a number of "subchannel codes," one of which is COBRA-IIIC/MIT.

COBRA-IIIC/MIT research has continued since its initial development in 1976 (Ref. 1). Past COBRA-IIIC/MIT research efforts have followed two paths. One path is concerned with the development and assessment of the bundle-wide analysis tool, MEKIN/T.H., which is based on COBRA-IIIC/MIT and is the thermal-hydraulic part of the three-dimensional core-wide kinetics code, MEKIN.

The second path was concerned with COBRA-IIIC/MIT. Early efforts along this path focussed on development of the single-pass analysis method, whereby an entire PWR core is analyzed in one stage using a fine mesh in a zone surrounding subchannels with higher radial peaking factors, and a coarser mesh outside this zone. More recent efforts along this second path compared COBRA-IIIC/MIT predictions with predictions of other codes and experimental data. Past research along the two paths has indicated several areas for COBRA-IIIC/MIT improvement.

Several improvements have been made to COBRA-IIIC/MIT. All of the improvements, except for one, have been made in response to recommendations of past research. The improvements are included in a new version of the codes as new modeling options. The new modeling options overcome limitations and disadvantages of old modeling options.

First, past research will be reviewed to provide an understanding of why COBRA-IIIC/MIT has been improved. Secondly, individual improvements will be described. Then, results of

testing individual improvements and application of the improved COBRA-IIIC/MIT version to transient test cases will be presented. Lastly, data input for the new version will be described.



## II. REVIEW OF PAST RESEARCH

### A. Overview

Since completion of the initial development of COBRA-IIIC/MIT in 1976 (Ref. 1), work on the code has continued at MIT under both EPRI and individual utility sponsorship. This work has proceeded along two paths. One path is concerned with the development and assessment of the bundle-wide analysis tool, MEKIN/T.H., which is based on COBRA-IIIC/MIT and is the thermal-hydraulic part of the three-dimensional core-wide kinetics code, MEKIN. The other path is concerned with development and improvement of the single-pass, mixed-lattice version of COBRA-IIIC/MIT. Although their goals are somewhat different, the two paths have complemented each other to some extent. Therefore, research work along both paths has been reviewed. A summary of this work is provided in Table II-1. The following discussion is separated into a discussion of work done prior to Fall 1977 (Ref. 1-11) and work done between Fall 1977 and Fall 1978 (Ref. 12).

### B. Work Completed Prior to Fall 1978

Rodack (Ref. 2) used MEKIN/T.H. to study Reactivity Insertion Accident (RIA) type transients in PWRs and related topics, including the sensitivity of thermal-hydraulic predictions to several parameters. His results indicate the importance of considering the spatial and temporal variation of the gap heat transfer coefficient,  $h_{gap}$ , in order to accurately calculate steady state fuel rod temperature distributions and transient surface heat fluxes. The effects of the temperature-dependence of fuel material properties and the quality-dependence of turbulent mixing parameter  $\beta$ , were also evaluated and shown to be significant.

The sensitivity study performed by Emami (Ref. 3) related to both MEKIN and COBRA-IIIC/MIT development but used COBRA-IIIC/MIT rather than MEKIN/T.H.. This study considered steady state conditions for both PWR and BWR systems. Overall thermal results were not significantly affected by wide ranges

TABLE II-1

Summary of Past Research

Emphasis on		System		Operation		Code Used	Information/Results	Reference
NEKIN	COBRA-IIIC/MIT	PWR	BWR	S.S.	Transient			
+	-	+	-	-	+	MEKIN/T.H.	PRR-RIA study, sensitivity study of T-H input parameters on fuel temperatures and coolant density, void fraction. List of most important parameters.	(2)
+	+	+	+	+	-	COBRA-IIIC/MIT	Sensitivity of COBRA solution to user input parameters and user selected correlations.	(3)
-	+	+	-	-	+	COBRA-IIIC/MIT	Lumped and mixed lattice approach (single pass method).	(4,7)
-	+	+	-	-	+	COBRA-IIIC/MIT	Verification of the single pass method in transients, discussion of experimental verification of COBRA.	(5,7)
-	+	+	-	+	-	COBRA-IIIC/MIT	Transport coefficients to improve results of lumped, mixed lattice approach.	(6,7)
-	+	+	-	+	-	COBRA-IIIC/MIT	Sensitivity of COBRA solution to user input parameters.	(8)
+	+	+	-	+	+	COBRA-IIIC/MIT	Development of a new solution method based on pressure field, convergence studies.	(9,10)
+	+	+	+	+	+	-	Study of different fuel pin models.	(11)
-	+	+	+	+	+	COBRA-IIIC/MIT & IV-I	Assessment and comparison/similar results except for clad temperature predictions.	(12)

Key: + yes  
- no

of values used for the transverse momentum parameters,  $s/l$  and  $K_{ij}$ , except for cases of large inlet flow upset of blockage. Variation of the turbulent mixing parameter,  $\beta$ , greatly affected flow and enthalpy predictions under two-phase conditions.

Work described in Refs. 4-8 was concerned with COBRA-IIIC/MIT development. The major portion of this reasearch was directed toward the development of a single-pass method, a method whereby an entire core is analyzed in only one stage using a fine mesh in a zone surrounding sub-channels with higher radial peaking factors, and a coarser mesh outside this zone. The parameter primarily concentrated on during this development was DNBR, since it was considered to be the most important parameter for licensing purposes. The research by Moreno (Ref. 4) and Liu (Ref. 5) provided the basis for justification of the method developed for steady-state and transient analyses as compared to the multi-pass (chain) methods used by reactor vendors. Chiu (Ref. 6) examined the applicability of two-dimensional transport coefficients to improve the lumped energy transfer models. Transverse momentum coupling parameters were investigated and found to have negligible effect for steady state conditions considered. All these research efforts are summarized in Ref. 7, which together with Refs. 5 and 8 comprised the state-of-the-art of the single-pass method and status of COBRA-IIIC/MIT development as of September 1977. The major conclusion from this work is that the simplified (single-pass) method yields accurate DNBR predictions, consistent multi-stage methods for PWRs under steady state and some transient conditions.

The research done by Masterson (Refs. 9 & 10) developed more efficient methods for solving the set of conservation equations of COBRA-IIIC/MIT. The COBRA-IIIP/MIT code was the result of his effort. COBRA-IIIP/MIT is numerically more efficient by allowing the use of iterative solution methods for sets of linear equations. COBRA-IIIP/MIT solves for the pressure distributions at individual axial levels, rather than crossflows. COBRA-IIIP/MIT generates converged crossflow distributions for decreasing axial mesh sizes, unlike COBRA-IIIC/MIT.

Finally it should be pointed out that recommendations to investigate fuel rod modeling given by Rodack (Ref. 2) have been followed to some extent by Mehrabian (Ref. 11) who compared various fuel pin models.

C. Work Completed Between Fall 1977 and Fall 1978

Between Fall 1977 and Fall 1978, research work was conducted by Kelly (Ref. 12) to evaluate the applicability of COBRA-IIIC/MIT for the thermal-hydraulic analysis of various Boiling Water Reactor (BWR) and Pressurized Water Reactor (PWR) cases of interest to utility engineers. The evaluation was made by comparing predictions of COBRA-IIIC/MIT with predictions of COBRA-IV-I and experimental data. During the investigation, COBRA-IIIC/MIT was modified to eliminate various inconsistencies and failures.

Application and testing of COBRA-IIIC/MIT during this project included the following:

- 1) BWR Bundle Analysis
  - a. Steady-State
  - b. Pressurization Transient
- 2) PWR Analysis
  - a. Severe Power Transients
  - b. Loss of Flow Transients
- 3) Comparisons with Experimental Data
  - a. Maine Yankee Exit Temperature Comparison
  - b. B&W Inter-bundle Crossflow Experiment
  - c. EIR Flow Blockage Experiment

The cases analyzed and results obtained are summarized in Table II-2. Conclusions made on the basis of these results were as follows:

- 1) Improvements are needed in both the heat transfer logic and the procedure for calculating the rod-to-coolant heat transfer coefficient.
- 2) As a result of the modifications made, it is now possible to use COBRA-IIIC/MIT to analyze a BWR core on a bundle-wide basis for transient conditions and to analyze a PWR

**Table II-2**  
**Summary of Cases Analyzed and Results from Ref. 11**

Case	Steady state or transient initial conditions				Levy sub-cooled Boiling Model Used	Results
	G (Mlb/hr-ft <sup>2</sup> )	P (psia)	T (°F)	q" (Mbtu/hr-ft <sup>2</sup> )		
BWR	Inlet Flow Sensitivity	1.25	1035	527	0.152	Inlet flow shows some sensitivity to many perimeters
		1.25	1035	527	0.152	
		1.25	1035	514	0.152	
	Steady State Comparisons	1.25	1035	527	0.152	COBRA-IIIC/MIT and COBRA-IV-I clad temperature predictions different
		1.25	1035	527	0.152	
		1.25	1035	514	0.152	
	Pressurization Transient	1.25	1035	527	0.152	Using Levy model improves predictions. Code modifications made are described in Ref. 11, pages 54-68
		1.25	1035	527	0.152	
PWR	Severe Power Transient	2.48	2100	635	10.	COBRA-IIIC/MIT and COBRA-IV-I predictions nearly the same except for clad temperature. Levy model fixed to prevent oscillations as described in Ref. 11, pages 95-103
		2.48	2100	635	10.	
		0.25	2100	635	1000.	
	Loss of Flow Transient	2.48	2100	541	0.1695	COBRA-IIIC/MIT and COBRA-IV-I predictions nearly the same except for clad temperature
		2.48	2100	541	0.30	
		2.48	2100	570	0.30	
	Maine Yankee Exit Temp. Comparison	2.48	2100	532	0.173	Exit temperature predictions of COBRA-IIIC/MIT and COBRA-IV-I in good agreement with data
	BAR Crossflow Experiment	---	near atmospheric	ambient	0.0	COBRA-IIIC/MIT and COBRA-IV-I crossflow predictions sensitive to axial nodalization; aside from this, predictions in good agreement with data
	EIR Flow Blockage Experiment	---	near atmospheric	ambient	0.0	Predictions in fairly good agreement with data. COBRA-IV-I gave better predictions by modeling variation of flow area directly

transient using small time steps. However, the crossflow solution is sensitive to axial mesh size.

- 3) Despite the difficulties with the heat transfer calculation, COBRA-IIIC/MIT appears to provide adequate PWR DNBR predictions. However, the code does not contain the logic or correlations needed to calculate BWR Critical Power Ratio (CPR).

These conclusions are each discussed in the following paragraphs and some examples of underlying calculational results are provided.

The need for improvement of the COBRA-IIIC/MIT rod-to-coolant heat transfer model became apparent from the comparison of COBRA-IIIC/MIT predictions with those of COBRA-IV-I. The inconsistency of the two code predictions is clearly shown in Figure II-1, which is a graph of steady state temperature vs. axial position for a BWR bundle analysis case. As shown in the figure, the clad temperature predicted by COBRA-IIIC/MIT varies discontinuously in the axial direction near the inlet and is significantly different from the COBRA-IV-I predictions. As discussed in Ref. 12, this difference is caused by differences in the heat transfer logic and energy equations used in the two codes in the subcooled boiling regime, with COBRA-IIIC/MIT being the least accurate.

Application of COBRA-IIIC/MIT to BWR and PWR transient analysis cases indicated that the code had not previously been adequately tested for such cases. One problem encountered was an oscillatory behavior of mass flow rate predictions during iteration, as shown in Figure II-2. This figure shows the variation of mass flow rate with iteration number at the point where boiling starts during analysis of PWR power transient. The oscillatory behavior was eliminated by a correction which prevented the quality from oscillating unrealistically between positive and negative values once it becomes positive in a particular node. Elimination of this and similar problems subsequently allowed COBRA-IIIC/MIT to analyze and make reasonable predictions for several PWR and BWR transients, as mentioned in Table II-2.

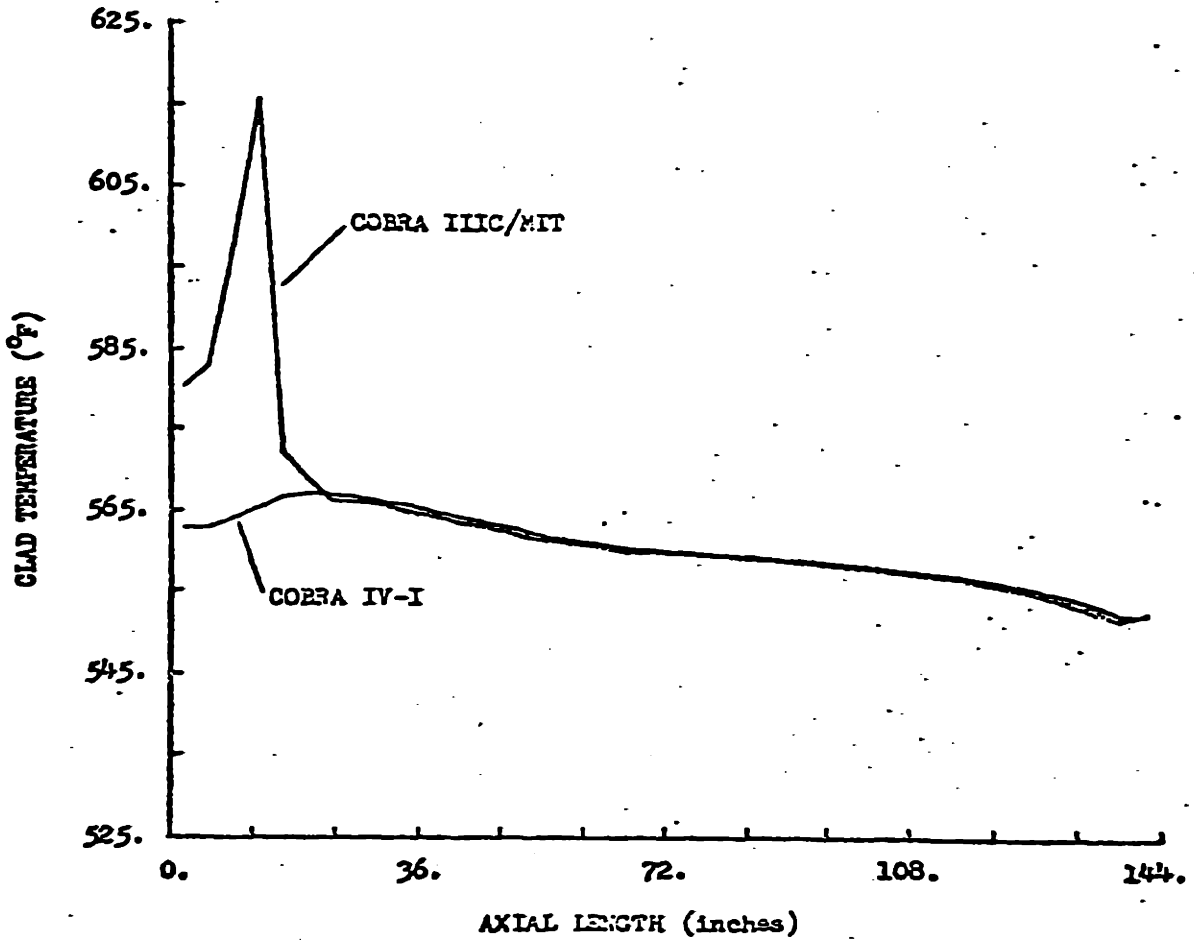


Figure II-1 (FIGURE 2.7 of Ref. 12)  
 CLAD TEMPERATURE VERSUS AXIAL LENGTH  
 RESULTS FOR STEADY STATE BWR CASE

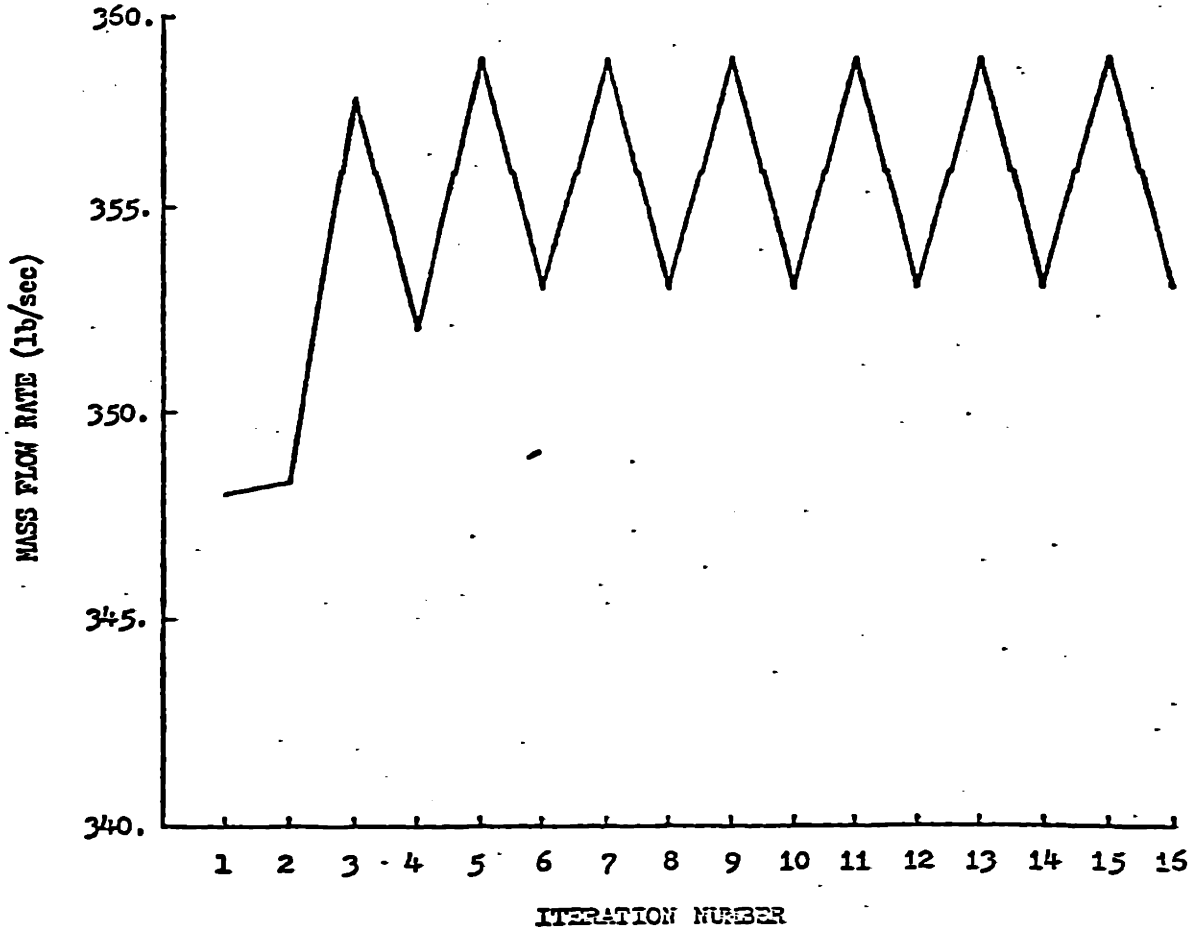


Figure II-2 (Fig. 3.4 of Ref. 12)

Mass Flow Rate Versus Iteration Number at  
Initiation of Boiling for PWR Severe Power Transient Case



Sensitivity of the crossflow solutions of both COBRA-IIIC/MIT and COBRA-IV-I to axial nodalization was encountered during analysis of the B&W crossflow experiment. The B&W isothermal test apparatus is shown in Figure II-3. The apparatus contains two bundles, separated above and below a common mixing length by divider plates. The flow control valves were adjusted to give the two bundles different flow rates; thus, inlet flow upset conditions were simulated. Sensitivity of crossflow predictions can be seen in Figs. II-4 and II-5. Figure II-4 shows COBRA-IIIC/MIT crossflow predictions of experimental results inferred from pressure measurements. Both COBRA-IIIC/MIT predictions use six channels to represent the experiment. One set of COBRA predictions uses 20 axial nodes and the other uses 36. The predictions show significant differences. Figure II-5 contains a pair of COBRA-IV-I crossflow predictions similar to those of COBRA-IIIC/MIT in Figure II-4. The differences between the predictions of COBRA-IV-I when the number of axial nodes change from 20 to 36 is dramatic. The crossflow solutions of both COBRA-IIIC/MIT and COBRA-IV-I failed to converge when 72 axial nodes were used. Figure II-6 shows the consistent set of results obtained for 20, 36, and 72 axial nodes when THERMIT, (Ref. 13) a code with greater capabilities than either COBRA-IIIC/MIT or COBRA-IV-I, was used. THERMIT contains the complete Navier-Stokes equations for momentum transport in all three directions, thereby avoiding any of the simplifications in the transverse momentum equations which are common for COBRA-IIIC/MIT and COBRA-IV-I.

Finally, despite the need for improvements in the COBRA-IIIC/MIT heat transfer logic and the procedure for calculating the rod-to-coolant heat transfer coefficient, the code appears to provide adequate PWR DNBR predictions. As discussed in Ref. 12, with the exception of clad temperature predictions, COBRA-IIIC/MIT and COBRA-IV-I predictions were in good agreement with each other and experimental measurements; and the DNBR was not affected by the clad temperature discrepancies. However, COBRA-IIIC/MIT does not contain the logic or correlations needed to calculate Critical Power Ratio (CPR), a figure-of-merit for BWR thermal margin.

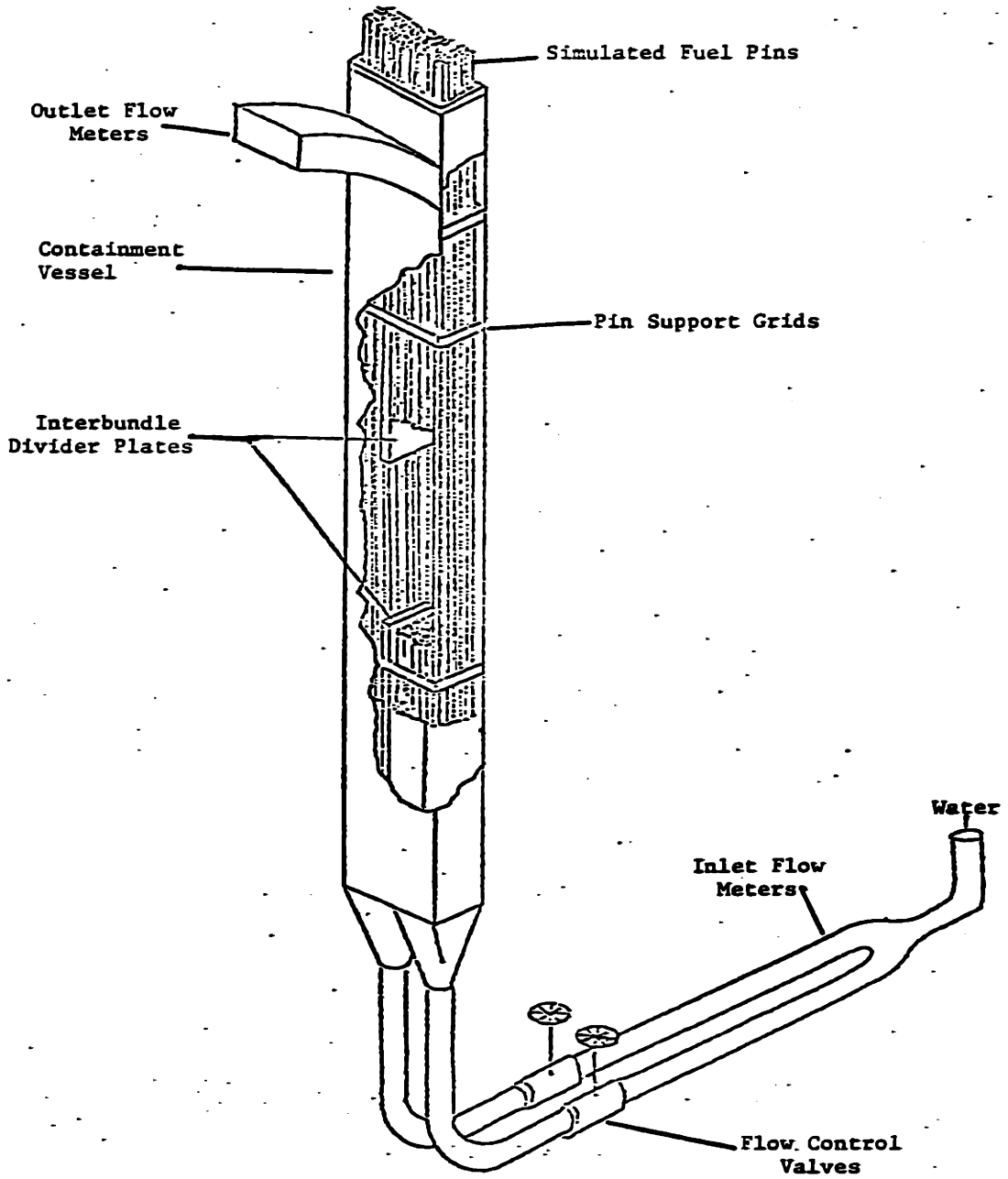


Figure II-3 (Fig. 5.2 of Ref. 12)

Schematic of B&W Apparatus

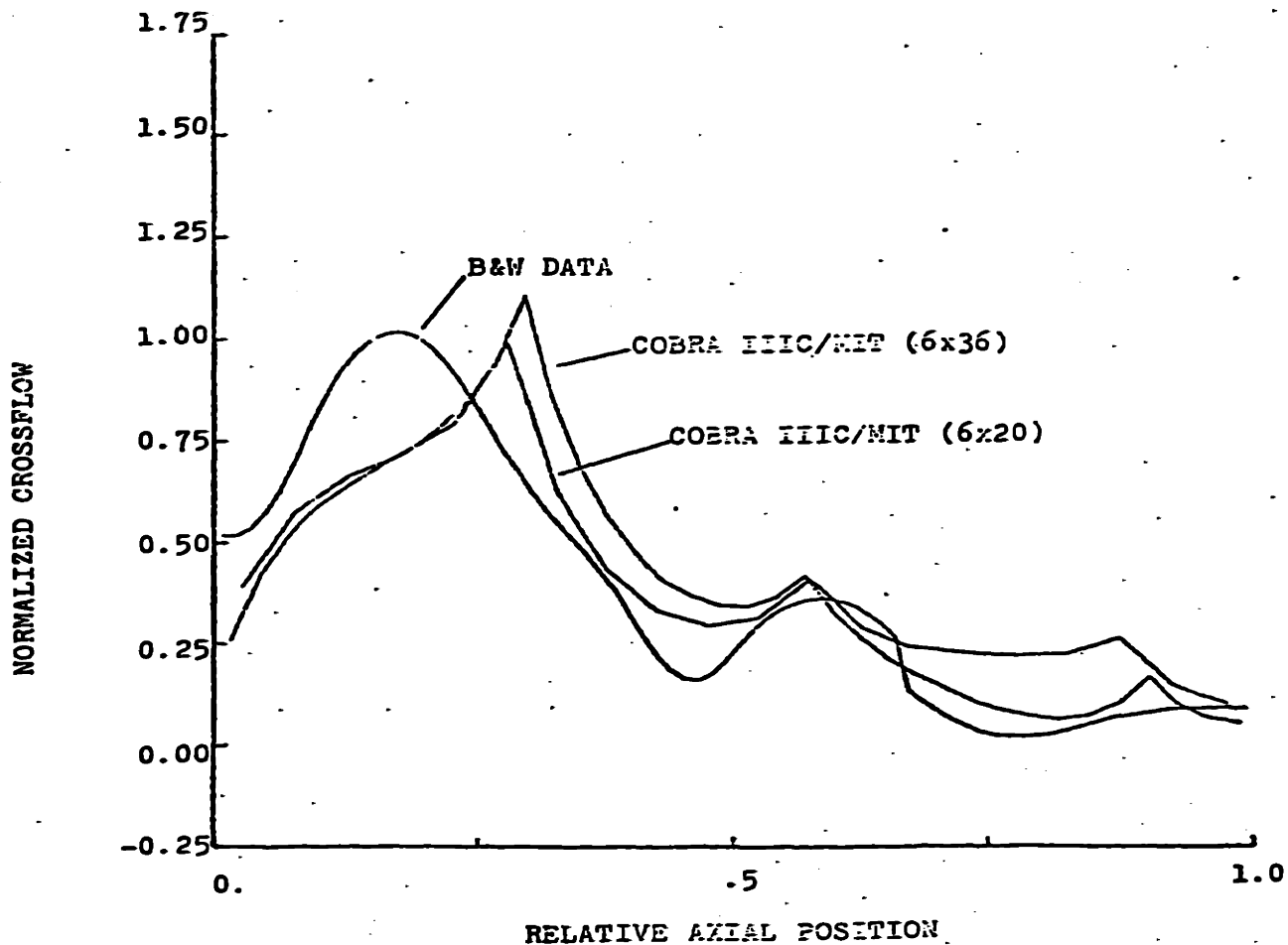


Figure II-4 (FIGURE 3.18 of Ref. 11)  
 NORMALIZED CROSSFLOW VERSUS AXIAL POSITION  
 COBRA-IIIC/MIT RESULTS FOR B&W CROSSFLOW EXPERIMENT

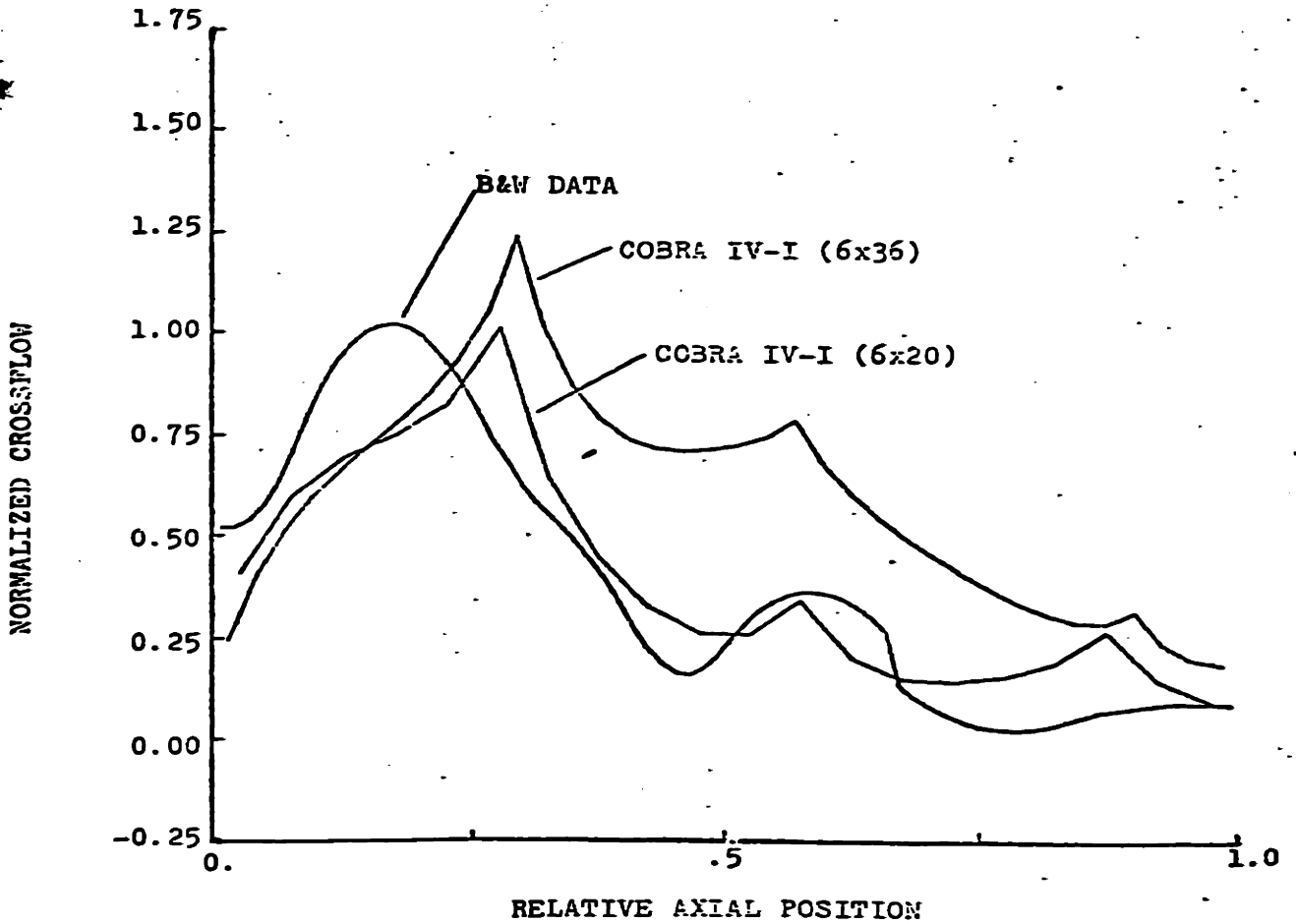


Figure II-5 (FIGURE 3.19 of Ref. 12)  
 NORMALIZED CROSSFLOW VERSUS AXIAL POSITION  
 COBRA-IV-I RESULTS FOR B&W CROSSFLOW EXPERIMENT

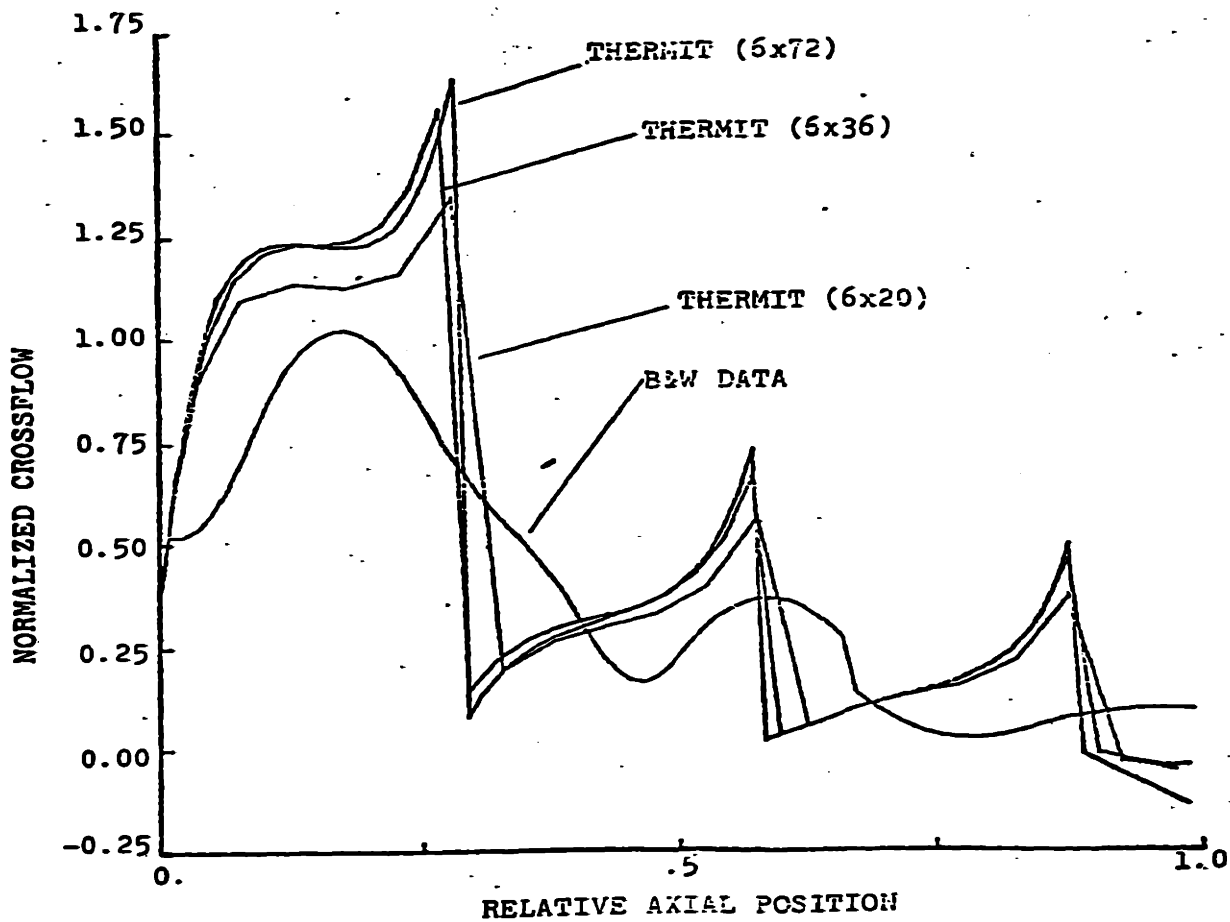


Figure II-6 (FIGURE 3.20 of Ref. 12)  
 NORMALIZED CROSSFLOW VERSUS AXIAL POSITION  
 THERMIT RESULTS FOR B&W CROSSFLOW EXPERIMENT (Ref. App. H)

**D. Summary of Major Conclusions Leading to Present Research**

The major conclusions from past research that have led to the research described in this report can be summarized as follows:

- 1) A new fuel rod model containing temperature-dependent properties and considering spatial and temporal variation of the gap heat transfer coefficient should be added.
- 2) The heat transfer model of COBRA-IIIC/MIT has poor logic which causes unrealistic discontinuities in its predictions.
- 3) Flow and enthalpy predictions are sensitive to the turbulent mixing parameter,  $\beta$ , which varies greatly with respect to quality in the two-phase region.
- 4) The simplified (single-pass) method yields accurate DNBR predictions, consistent with multi-stage methods for PWRs under steady-state and some transient conditions. However, COBRA-IIIC/MIT does not have the logic or correlations needed to calculate BWR CPR.
- 5a) Transverse momentum coupling parameters are of negligible importance for steady state (near-normal) conditions.
- 5b) Overall thermal results are not dependent on the cross-flow parameters  $s/l$  and  $K_{ij}$ , except for cases of large inlet flow upset or blockage.
- 6) Crossflow predictions are sensitive to axial nodalization.

### III. CODE IMPROVEMENTS

#### A. Introduction

Several improvements have been made to COBRA-IIIC/MIT. The improvements are briefly described in Table III-1. The need for improvements a through d was seen during past research. Improvements a through d correspond to conclusions 1 through 4 given in Section II.D. Improvement e is the result of a suggestion by Prof. J. Weisman (Ref. 14). Conclusions 5a and 5b of Section II.D are related to the technical issue behind improvement e. The improvements are options of the improved version of COBRA-IIIC/MIT. Code changes made during implementation of improvements are described in Appendix A. Improvements will be individually described in the following sections.

#### B. New Fuel Rod Model

A new fuel rod model has been added to COBRA-IIIC/MIT. This model is based on the MATPRO model developed at INEL (Ref. 15) and eliminates the following disadvantages of the old COBRA-IIIC/MIT fuel rod model:

- 1) Fuel and cladding properties were assumed to be independent of temperature.
- 2) A single value of the fuel-clad gap heat transfer coefficient,  $h_{\text{gap}}$ , was used for the entire reactor core.
- 3) Gap and clad conductivity were lumped into single node.
- 4) Gap thickness was assumed to be zero.

The need for considering the temperature dependence of fuel rod properties is indicated by results of past research, as discussed in Section II. These results (Ref. 2) showed that transient thermal-hydraulic predictions are especially sensitive to fuel thermal conductivity and heat capacity and fuel-to-clad gap heat transfer coefficient. The temperature variation of fuel conductivity and heat capacity is shown in Figures III-1 and III-2, respectively.

TABLE III-1

COBRA-IIIC/MIT IMPROVEMENTS

IMPROVEMENT	PREVIOUS STATUS	DESCRIPTION OF IMPROVEMENT	ADVANTAGES/DISADVANTAGES
a) New Fuel Rod Model	No temperature dependence, constant value of fuel-clad gap heat transfer coefficient used for entire core	New model with temperature dependent properties, higher numerical accuracy and burnup-dependent fuel-clad gap heat transfer coefficient	Better modeling, higher accuracy improved heat transfer prediction/ slightly increased computation time
b) New Rod-to-Coolant Heat Transfer Model	Inconsistent logic causes poor clad temperature predictions.	New heat transfer model with greater capabilities in the high quality regime and more consistent logic	Improved heat transfer predictions/ Increased computation time
c) New Mixing Model	Cannot account for quality dependence	Added option enabling use of quality dependent mixing model	Improved capability for BWR subchannel analysis/ Slightly increased computation time
d) Critical Power Ratio (CPR) and Critical Heat Flux Ratio (CHFR) Calculation Options	No CPR or CHFR calculation options available for BWR analysis	Added options calculate CPR and CHFR	CPR and CHFR can be calculated/ No disadvantage expected
e) New Transverse Momentum Coupling Parameters for the Single-Pass Method	Only one value of $s/t$ and $K$ used for all gap interconnections of a case	$s/t$ and $K$ are, in effect, varied from one gap interconnection to another by coupling parameters	May slightly improve accuracy of predictions under some conditions/ Extra input preparation effort, no axial variation of coupling parameters



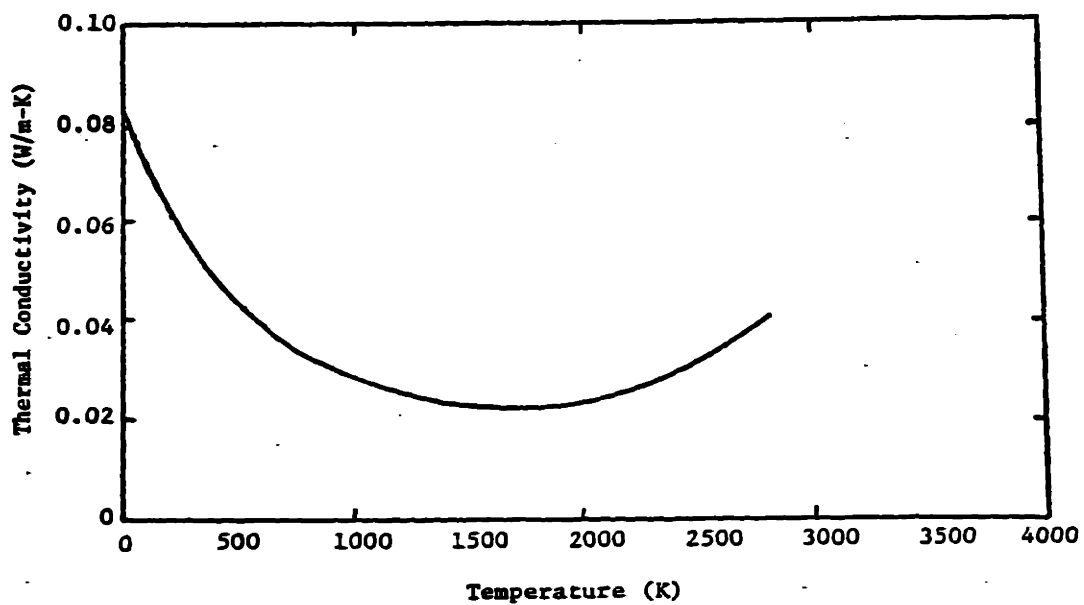


Figure III-1 (Based on Fig. A-2.1 of Ref. 15)  
Thermal Conductivity of  $UO_2$

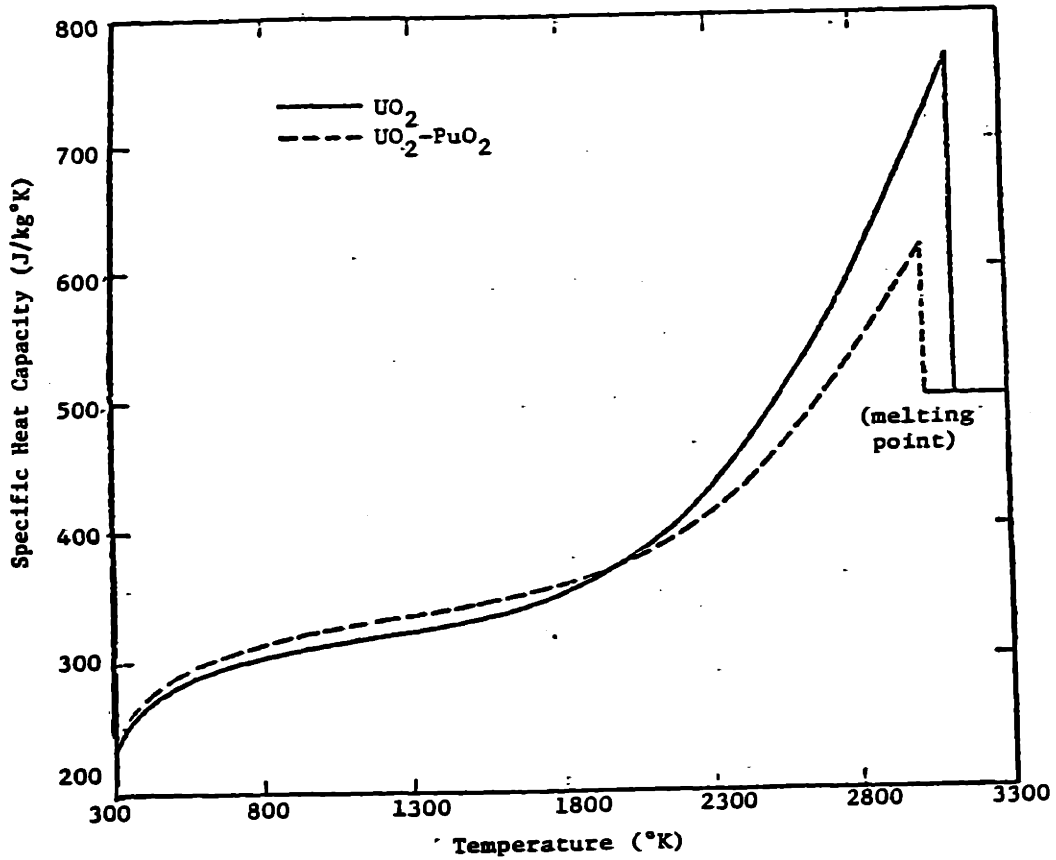


Figure III-2 (Fig. A-1.1 of Ref. 15)  
 Specific Heat Capacity of  $UO_2$  and  $UO_2-PuO_2$

The methods used by the new fuel rod model to represent temperature dependent properties and  $h_{gap}$  are described in Appendix B. The old and new fuel rod model is called by subroutine HEAT and used in the calculation of fuel rod temperatures and surface heat fluxes as described in Appendix C.

### C. New Rod-to-Coolant Heat Transfer Model

A new heat transfer model based on the BEEST (Ref. 16) package has been added to COBRA-IIIC/MIT. The new model has greater capabilities and better heat transfer logic than the model previously used. The old heat transfer model was limited to pre-CHF conditions and used questionable logic to switch from forced convection to nucleate boiling heat transfer. Void fraction rather than wall temperature determines when the switch is made.

The new model can construct a complete boiling curve, such as the one shown in Figure III-3, for each space and time step. The boiling curve shown has positive slope up to point A, where critical heat flux occurs. Between point A and B is a transition boiling region. Point B is at the metastable film boiling temperature. The curve continues to the right from B in the film boiling region. The new model constructs portions of the curve only as they are needed in order to avoid unnecessary computation.

The new heat transfer model has two options. The first option is to consider only pre-CHF conditions. This option bypasses calculations which are made to check if CHF has been exceeded. The second option is to consider pre- and post-CHF conditions. If the first option is used for a case which includes post-CHF conditions, pre-CHF correlations will be mistakenly used for post-CHF conditions. One may be able to detect this error by noticing a CHFR, CPR, or DNBR prediction which is less than unity.

The new heat transfer model is further described in Appendix D. The equations and data bases of the pre-CHF correlations used in the old and new heat transfer models are given in Appendix E. The COBRA-IV-I heat transfer model is similar to the new heat transfer model because it can also construct a complete boiling curve. The COBRA-IV heat transfer model is described in Appendix F.

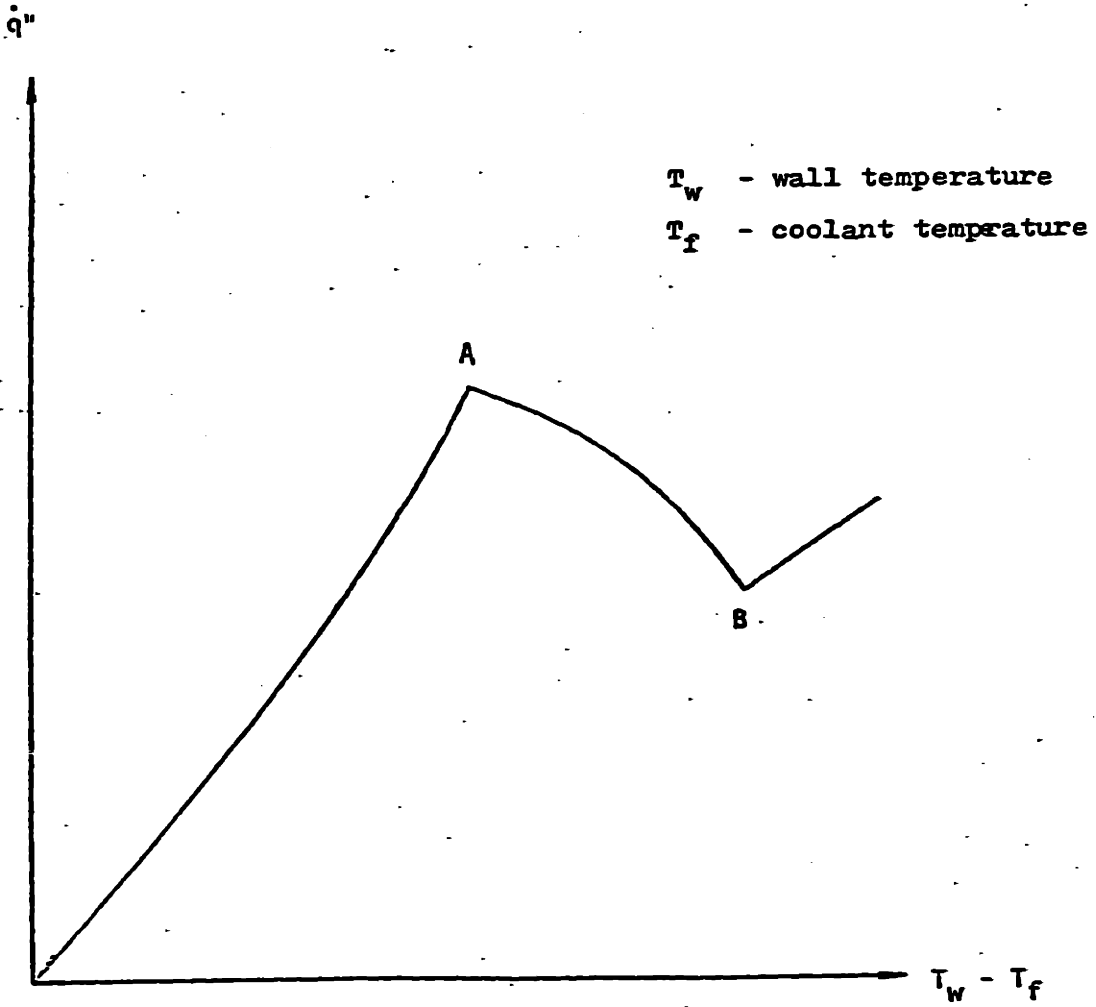


Figure III-3

A Typical Boiling Curve of New Heat Transfer Model

## D. New Mixing Model

### 1. Description of Model

The Beus quality dependent mixing model (Ref. 17) has been added to COBRA-IIIC/MIT to improve the prediction of turbulent mixing for two-phase flow in rod bundles. The model assumes existence of two mixing regions corresponding to the bubbly-slug and annular flow regimes. The region is determined by  $x$ ,  $G$ ,  $P$ , and geometry ( $s/D_h$ ). Figures III-4 and III-5 represent typical curves showing the variation of mixing with quality and pressure in these two mixing regions. The equations describing the model are contained in Appendix G.

The model has been constructed from the data which were taken within the following limitations:

System Pressure	$50 \leq P \leq 775$	psia
Mass Velocity	$7.3 \times 10^4 \leq G \leq 3 \times 10^6$	lb/hr-ft <sup>2</sup>
Quality	$0 \leq x \leq .80$	
Gap Width	$0.2 \leq s \leq .10$	in.

## E. New Correlations for Critical Power Ratio (CPR) and Critical Heat Flux Ratio (CHFR) Calculation

Correlations have been added to the code to enable it to calculate CPR and CHFR. The new correlations and associated logic are described in the following subsections and Appendix H.

### 1. Critical Power Ratio (CPR) Correlation

#### a. Introduction

A common measure for thermal margin is the Critical Heat Flux Ratio (CHFR) which is defined as the ratio of CHF given by a correlation for a given set of local conditions to the local heat flux. Under BWR conditions this "local condition hypothesis" is not generally applicable. Thus, GE has adopted Critical Power Ratio (CPR) to replace CHFR as the figure of merit for evaluating BWR thermal margin as part of the GE Thermal Analysis Basis (GETAB). CPR is defined as the ratio of critical bundle power to operating bundle power. The GETAB design procedure uses the BEXL correlation (Ref. 18) as part of a statistical treatment of the required thermal margin. The GEXL correlation is a critical quality - boiling

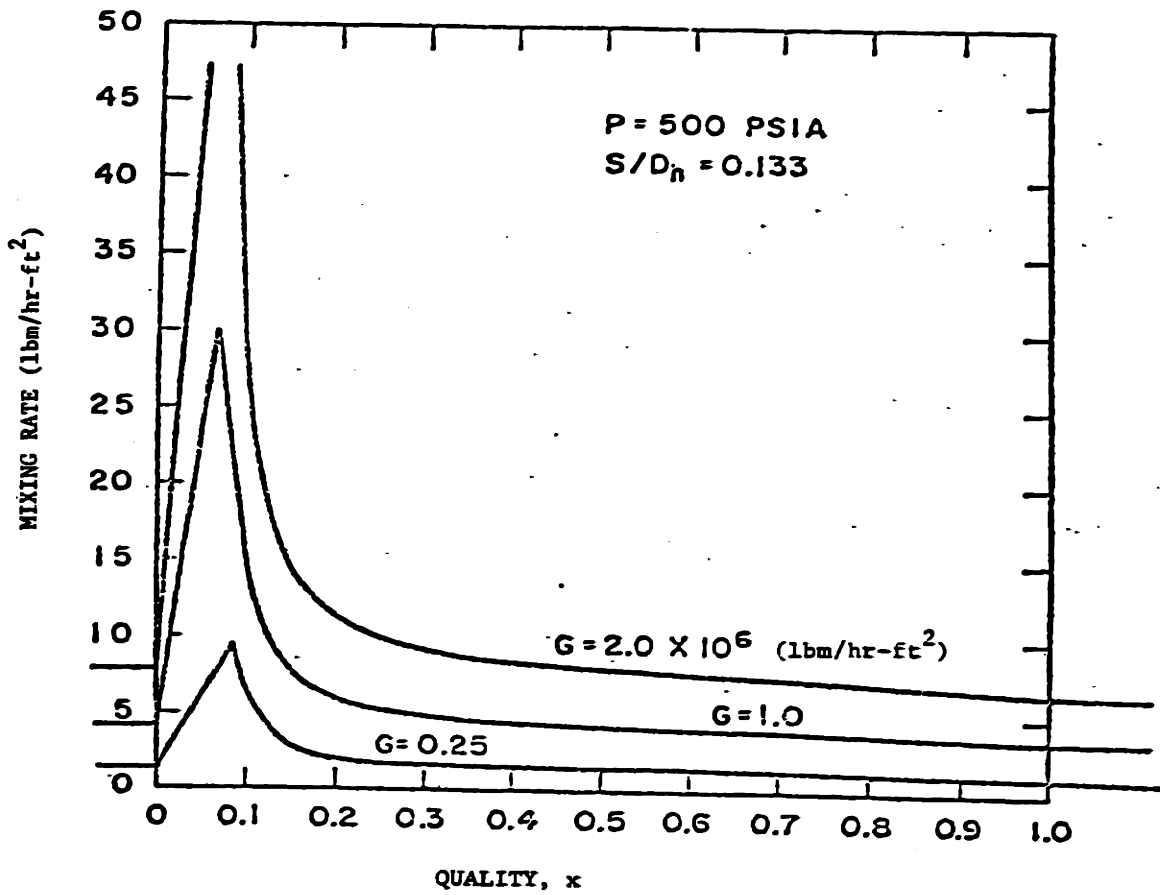


Figure III-4 (Fig. 8 of Ref. 16)

Mixing Rate Variation with Quality

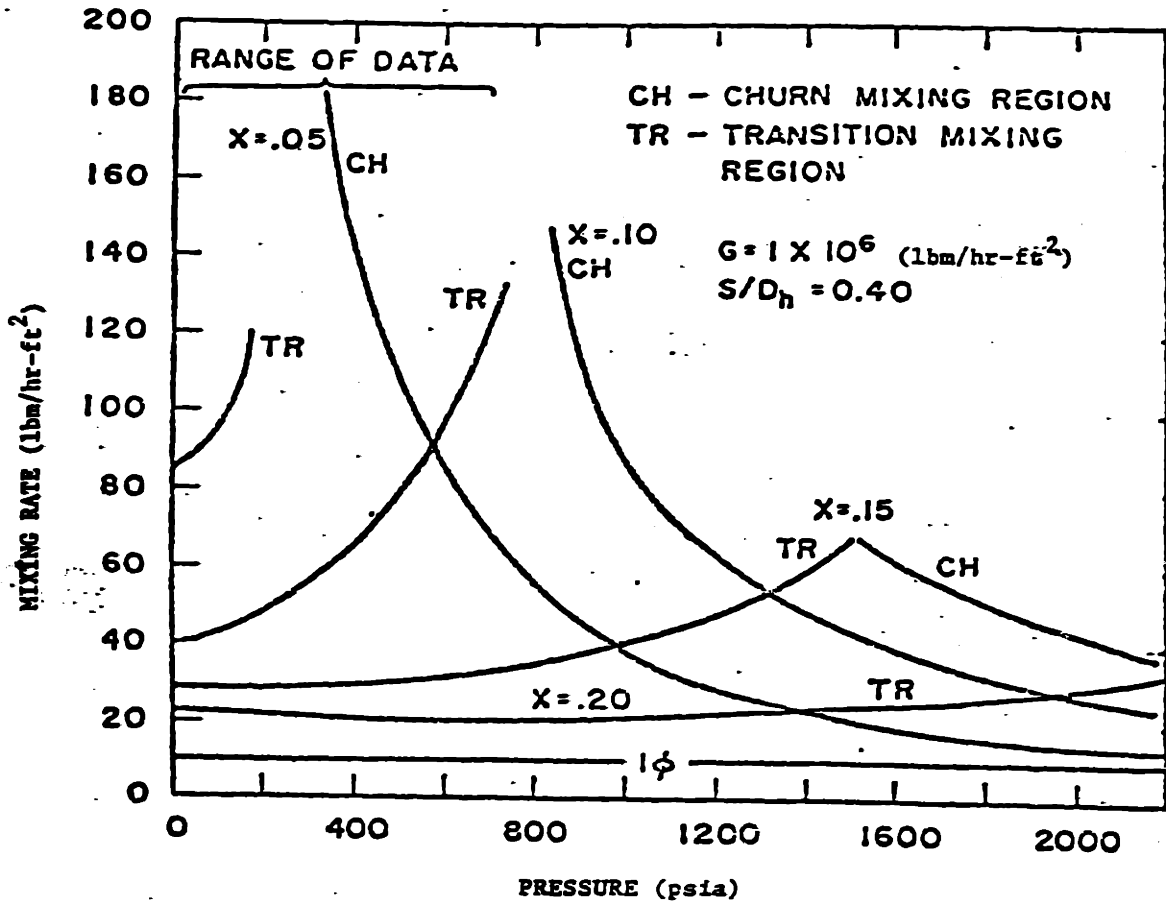


Figure III-5 (Fig. 9 of Ref. 16)

Mixing Rate Variation with Pressure

length approach. This approach lends itself automatically to the CPR concept as a figure-of-merit for evaluating thermal margin.

The correlation, expressed in its most general form, is:

$$X_C = X_C(L_B, D_Q, G, L, P, R)$$

where:

$X_C$  = bundle average critical quality;

$L_B$  = boiling length;

$D_Q$  = thermal diameter (i.e., four times the ratio of total flow area to total rod perimeter);

$L$  = heated length;

$P$  = system pressure;

$R$  = a parameter which characterizes the local peaking pattern with respect to the most limiting rod; and

$G$  = mass flux.

The parameter  $R$ , in addition to being a function of the local peaking pattern, is also dependent on lattice dimensions and on the grid spacer configuration. In effect,  $R$ , takes into account the details of the flow and enthalpy distribution which are ordinarily only accounted for by a detailed subchannel analysis.

The range of conditions over which the GEXL correlation is considered to be valid:

Pressure:	800 to 1400 psia
Mass Flux:	$0.10 \times 10^6$ to $1.25 \times 10^6$ lb/hr-ft <sup>2</sup>
Inlet Subcooling:	0 to 100 BTU/lb.

As shown in Figure II-6, the heat balance curve which touches the GEXL correlation determines the critical power. The calculation of critical power involves an iterative procedure. The critical power curve is associated with a minimum critical power ratio (MCPR) of one which reduces the critical quality difference,  $\langle \Delta X_e \rangle_c$ , as shown in the figure to zero.

In order to comply as much as possible with the new BWR design procedure, the CISE-4 critical quality-boiling length correlation has been added to COBRA-IIIC/MIT. The CISE-4 correlation,



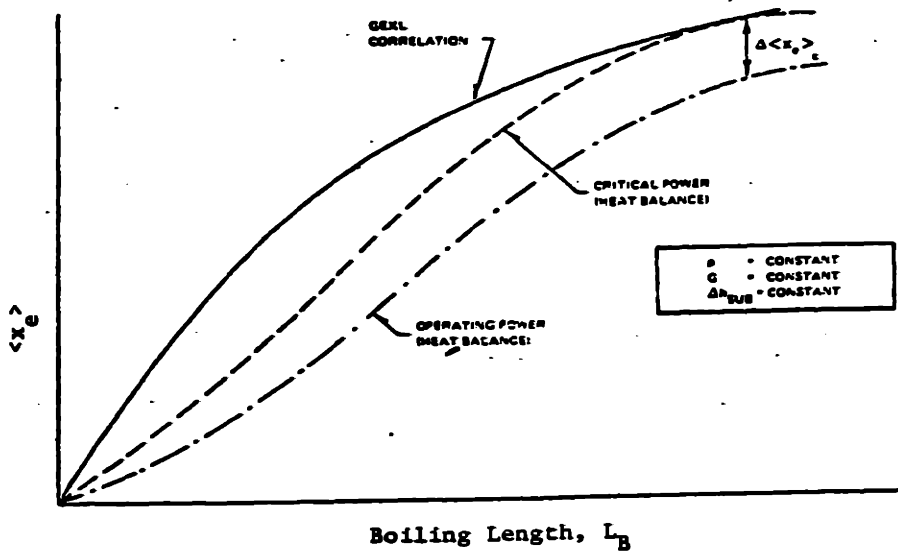


Figure III-6 (Fig. 4-40 of Ref. 19)

Graphic Display of GEXL Correlation and BWR Heat Balance Curves

(Refs. 20 and 21), the starting point of GE's own development, was introduced by Bertoletti, et. al. The CISE-type approach uses critical quality versus boiling length. Boiling length is the length over which bulk boiling occurs. Figure III-7 shows the boiling boundary,  $\lambda$ , the critical boiling length,  $L_{Bc}$ , and critical quality,  $\langle x_e \rangle_c$ . Data from experiments with uniform and non-uniform axial heat flux profiles are collapsed onto curves of  $\langle x_e \rangle_c$  vs.  $L_{Bc}$  as shown in Figure III-8.

b. CISE-4 Correlation

CISE-4 is a modified version of the earlier CISE-3 correlation (Ref. 20 & 21). The modification extends the range of the correlation's applicability to lower flow rates. The CISE-4 correlation is intended for analysis using rod-centered subchannels rather than coolant-centered subchannels, such as COBRA uses. The use of CISE-4 correlations for coolant-centered subchannels is thought to be permissible, however, for analysis of central bundle subchannels.

The general functional form of the correlation is:

$$\langle x_e \rangle_c = \frac{D_h}{D_e} \frac{a(P,G)L_{Bc}}{[L_{Bc} + b(P,G,D_e)]} \quad (\text{Eqn. III-1})$$

In COBRA-IIIC/MIT, the critical power ratio (CPR) prediction is based on a heat balance, which yields the following equation:

$$\text{CPR} = 1 + \frac{\langle x_e(L_{Bc}) \rangle_c = \langle x_e(L_{Bc}) \rangle}{\langle x_e(L_{Bc}) \rangle + \frac{h_f - h_{in}}{h_{fg}}} \quad (\text{Eqn. III-2})$$

Equation (III-2) is approximate in that it assumes that the distribution of coolant flow among channels does not change with power level. This assumption is fairly accurate in the general vicinity of critical power. The accuracy is sufficient for iteration on power until CPR=1.

\* Ref. page H-9 of Appendix H for definition of nomenclature used in this section.

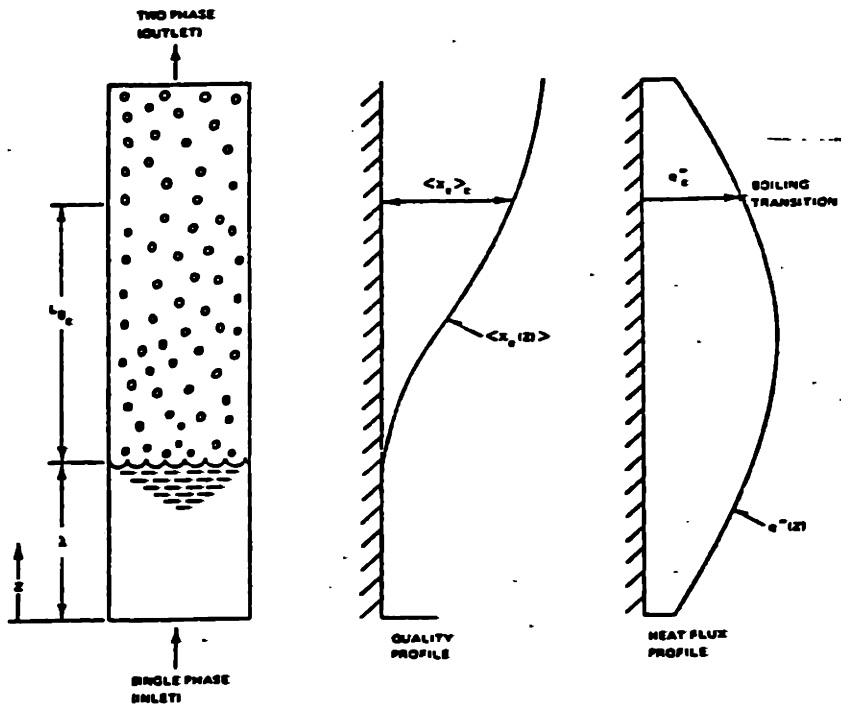


Figure III-7 (Fig. 4-27 of Ref.19 )

Schematic Showing Relationship

Between  $L_{BC}$ ,  $\langle x_e \rangle_c$  and the Boiling Transition

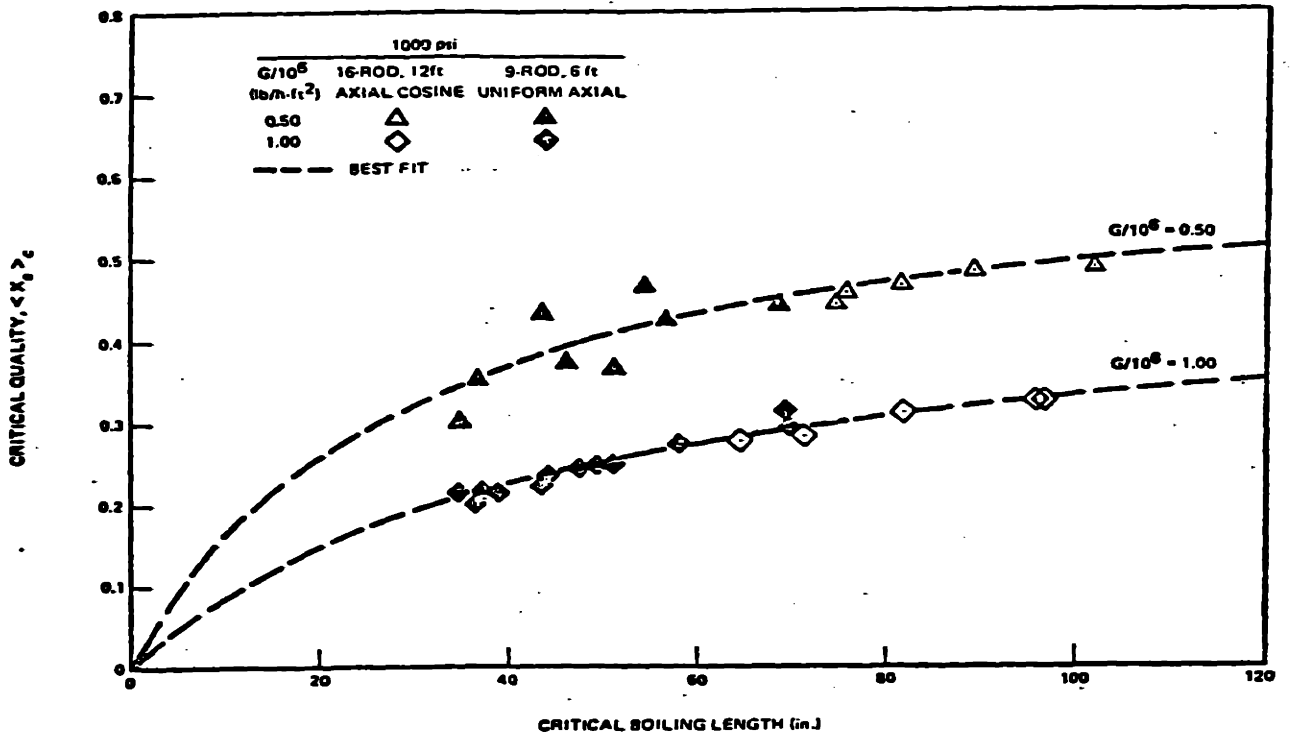


Figure III-8 (Fig. 4-31 of Ref. 19.)

GE Nine-rod and Sixteen-rod Critical Quality  
Versus Boiling Length Curves

2. Hench-Levy CHF Correlation

The Hench-Levy correlation (Ref. 22) uses limit lines to define a lower envelope to the CHF data. Hench-Levy limit lines are shown in Figure III-9. The limit line approach is conservative in that it predicts CHF at a power level below the power level at which the experimental data indicates it would actually occur. Because it does not account for non-uniform axial heat flux effects, however, it does not accurately predict the axial CHF location. Also, under some conditions, it can conservatively predict the power levels at which CHF occurs while non-conservatively predicting the local CHF at the critical power. An example of this paradox is given in Figure III-10.

3. Biasi/Void-CHF Correlation

The Biasi/Void-CHF correlation was initially provided in the new heat transfer model for the CHF calculation required in order to construct a boiling curve. This calculation is also an additional option for CHF calculation.

The Biasi/Void-CHF correlation is actually a combination of the Biasi (Ref. 23) and Void-CHF (Ref. 24) correlations. The combination was developed for calculation of local CHF during transients. Simplicity and applicability to a wide range of coolant conditions were high priorities. CHF prediction accuracy was a lesser priority.

The form of the Biasi/Void-CHF correlation is:

$$(q''_{CHF})_{Biasi} = f(D_e, G, P, x) \quad (\text{Eqn. III-3})$$

$$(q''_{CHF})_{Void-CHF} = f(\alpha, \sigma, \rho_f, \rho_g, H_{fg}) \quad (\text{Eqn. III-4})$$

where,

Eqn. (III-3) is used for  $G \geq G_1$ ;

A linear interpolation between Eqn. (III-3) and (III-4) is used for  $G_0 < G < G_1$ ;

Eqn. (III-4) is used for  $G \leq G_0$ .

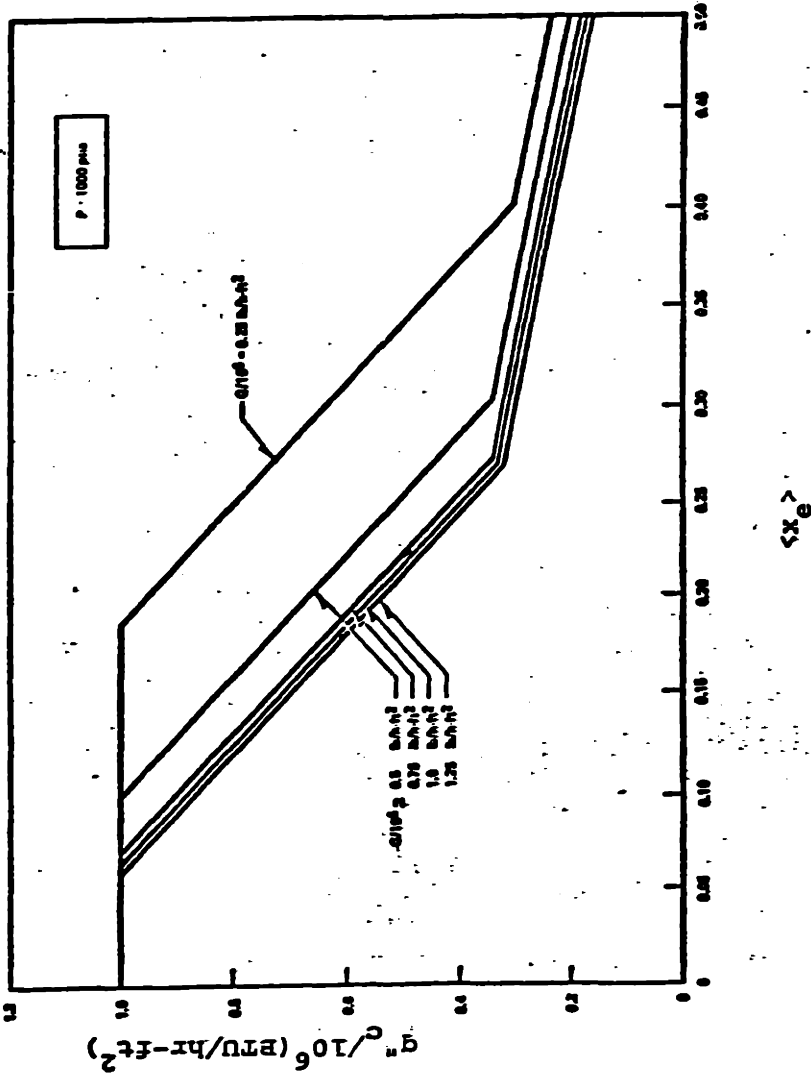


Figure III-9

Hench-Levy Limit Lines  
(Ref. 18)

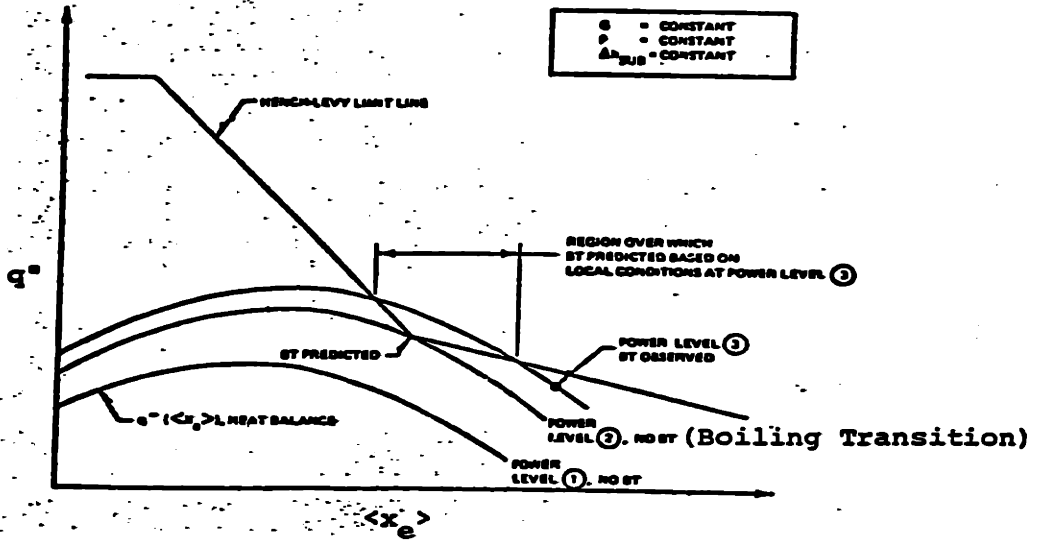


Figure III-10  
Experimentally Observed Trend  
in CHF Data Compared to the  
Hench-Levy Limit Line  
 (Ref. 19)

Ref. Appendix H for a more detailed description of the correlation, including information concerning its range of applicability.

#### 4. Summary of the Correlations Provided in the Improved Version of COBRA-IIIC/MIT

Appendix H provides a summary of the correlation provided in COBRA-IIIC/MIT for calculation of CHF and CPR. Also included are the W-3 and B&W-2 DNBR correlations. This summary provides references, equations and range of data base for each correlation.

#### F. New Transverse Momentum Coupling Options for the Single-Pass Method

##### 1. Background

Weisman (Ref. 14) has suggested that the transverse momentum parameters used in COBRA-IIIC/MIT,  $s/l$  and  $K$ , should be modified when the code is used for analysis cases involving interconnected regions of different size.\* This suggestion has also been made by Chiu (Ref. 11). The old COBRA-IIIC/MIT approach is compared with the modified approaches suggested by Weisman and Chiu in Appendix I. COBRA-IIIC/MIT has been modified to provide the option of using the Weisman and Chiu approaches in addition to the old COBRA approach for transverse momentum modeling.

##### 2. Description of Code Modification

The old equations for transverse momentum [Eqns. (I-1) and (I-2) of Appendix I] are changed to the following:

$$\frac{\partial}{\partial t}[W_{ij}] + \frac{\partial(u^*W_{ij})}{\partial x} = (f_{sl})_{ij} \frac{s}{l}(P_i - P_j) - F_{ij} \quad (\text{Eqn. III-5})$$

$$F_{ij} = \frac{K|W_{ij}|W_{ij}}{2(S_{ij})^2 \rho^*} \frac{s}{l}(f_{slk})_{ij} \quad (\text{Eqn. III-6})$$

---

\*Such cases are encountered when using the single-pass method for core thermal-hydraulic analysis (Ref. 12).



For the Weisman approach,

$$(f_{sl})_{ij} = \left(\frac{N_g}{N_r}\right)_{ij} \quad (\text{Eqn. III-7})$$

$$(f_{slk})_{ij} = (N_g)_{ij} \quad (\text{Eqn. III-8})$$

For the Chiu approach,

$$(f_{sl})_{ij} = \frac{(N_g)_{ij}}{(N_o)_{ij}} \quad (\text{Eqn. III-9})$$

$$(f_{slk})_{ij} = (N_g)_{ij} \quad (\text{Eqn. III-10})$$

When a user does not select the new transverse momentum option, the  $f_{sl}$  and  $f_{slk}$  factors are set to unity.

## IV. TESTING AND APPLICATION

### A. Introduction

Most of the new COBRA-IIIC/MIT options have been tested either individually or by application of the improved version of the code to transient test cases. The new fuel rod, heat transfer, and mixing models have been individually tested. The new correlations for CPR and CHF and the transverse momentum parameters have also been individually tested. The improved version has been applied to PWR and BWR transient test cases. New options which have not been tested are post-CHF rod-to-coolant heat transfer and Biasi/Void-CHF CHF predictions. Section IV.B will cover individual testing of new COBRA-IIIC/MIT options. Section IV.C will cover application of the improved version to transient test cases.

### B. Individual Testing of New Models

#### 1. Testing of New Fuel Rod Model

The new fuel rod model has been tested using steady state and transient test cases. Some test cases were run to test the solution method for numerical stability and energy conservation. Further tests compare predictions of new fuel rod model options with predictions of the old fuel rod model. The following subsections describe the tests and the results obtained.

##### a. Steady State Predictions

Predictions of the old and new fuel rod models are compared for the case of constant fuel and clad properties and gap conductance. The results are shown in Figure IV-1 which gives the radial fuel rod temperature distributions for the two cases. Data used by the fuel rod models is also given in the figure. The difference between the predictions is in the clad and gap regions. The old fuel rod model lumps the clad and gap regions together while the new fuel rod model considers them as separate regions.

The new fuel rod model was also individually tested by calculating a steady state temperature distribution for one axial node of a fuel rod. The heat generation rate, coolant tempera-

IV-2

	<u>Data</u>	
Fluid temperature		532°F
Rod-to-coolant heat transfer coefficient		4751 BTU/hrft <sup>2</sup> °F
Rod surface heat flux		0.2076 $\frac{\text{MBTU}}{\text{hr-ft}^2}$
Gap heat transfer coefficient		600 BTU/hr ft <sup>2</sup> °F

<u>Properties</u>	<u>Fuel</u>	<u>Clad</u>
conductivity	1.4	8.8 BTU/hr-ft°F
density	650	410 lb/ft <sup>3</sup>
specific heat	.08	.078 BTU/lb°F

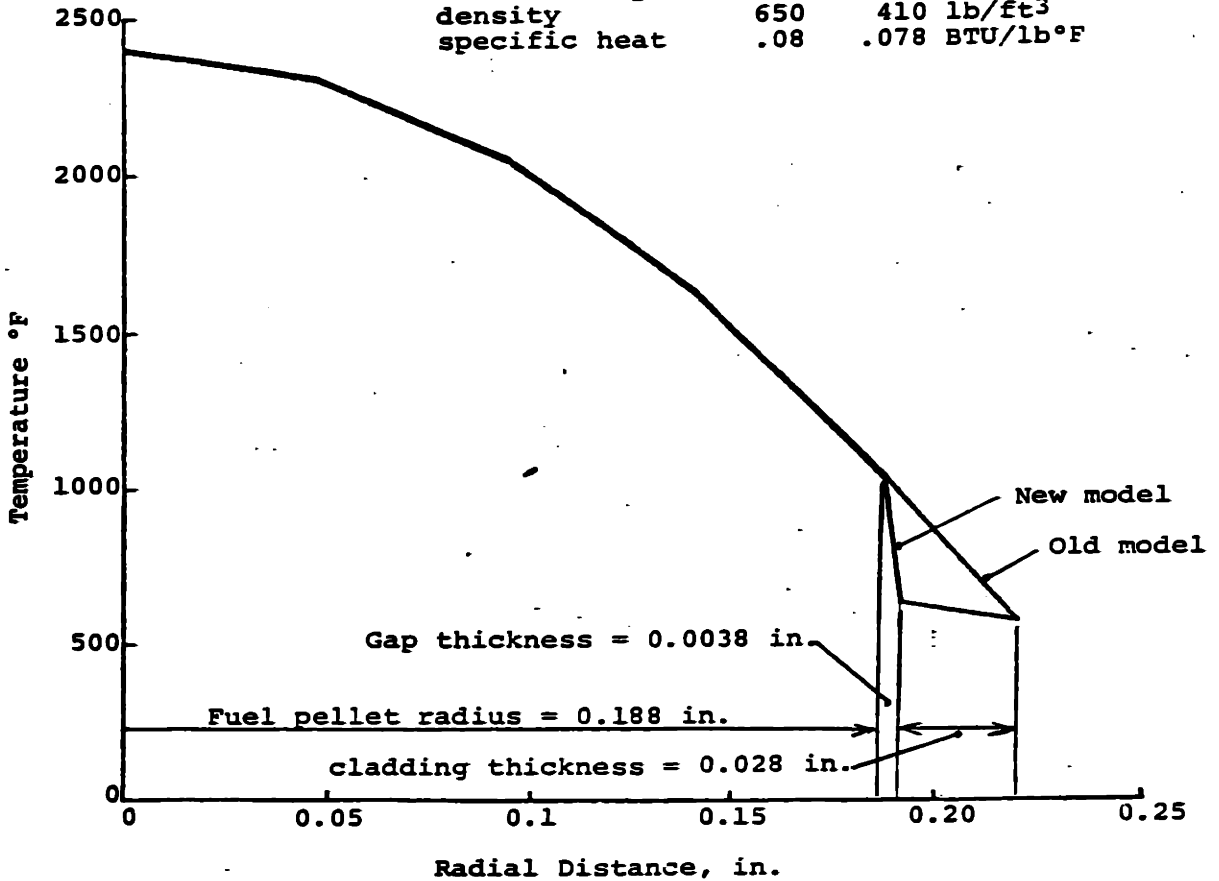


Figure IV-1

Predictions of Old and New Fuel Rod Models  
Using Constant Properties and  $h_{\text{gap}}$  Option

ture, and rod-to-coolant heat transfer coefficient were held constant. The nodalization scheme used was four radial fuel nodes, one gap node, and one clad node, for a total of six radial fuel rod nodes. Steady state temperature predictions were obtained using the three options of the new model. These are:

- 1) Constant properties, user input values of fuel and clad properties and gap conductance,  $h_{\text{gap}}$ .
- 2) Fuel and clad properties calculated, user input value of  $h_{\text{gap}}$ .
- 3) All properties calculated.

The results of these predictions are shown in Figure IV-2. The three temperature profiles shown have similar shapes. The radial position of the gap region is marked by a sharp temperature drop near  $r=0.15$  in. One effect of temperature dependent properties can be seen in the difference between the temperature profile predicted using the constant properties option and the other two profiles predicted using calculated fuel and clad properties. In the fuel region, which extends from  $r=0$  to  $r=0.15$  in., the negative slope magnitude of the profile predicted by the constant properties option is exceeded by the slopes of profiles predicted by the other two options as radius goes from 0 to 0.15 in. This observed difference is due to decreasing calculated thermal conductivity of the fuel with decreasing temperature (increasing radius).

#### b. Transient Predictions

The new fuel rod model was further tested by calculating transient temperature distributions for one axial node of a fuel rod. The coolant temperature and rod-to-coolant heat transfer coefficient were held constant. The nodalization schemes used thirteen radial fuel nodes, one gap node, and three clad nodes, or seventeen radial fuel rod nodes in all. Temperature distributions were obtained using two options, all properties calculated and all properties constant. At time zero, with temperatures at steady state as shown in Figure IV-3, the heat generation rate was assumed to undergo a step increase by a factor of

IV-4

Fuel Rod Radius	.22 in.
Fluid temperature	640 °F
Rod-to-coolant heat transfer coefficient	6000 Btu/hr-ft <sup>2</sup> -°F
Linear heat generation rate	5.8 Kw/ft

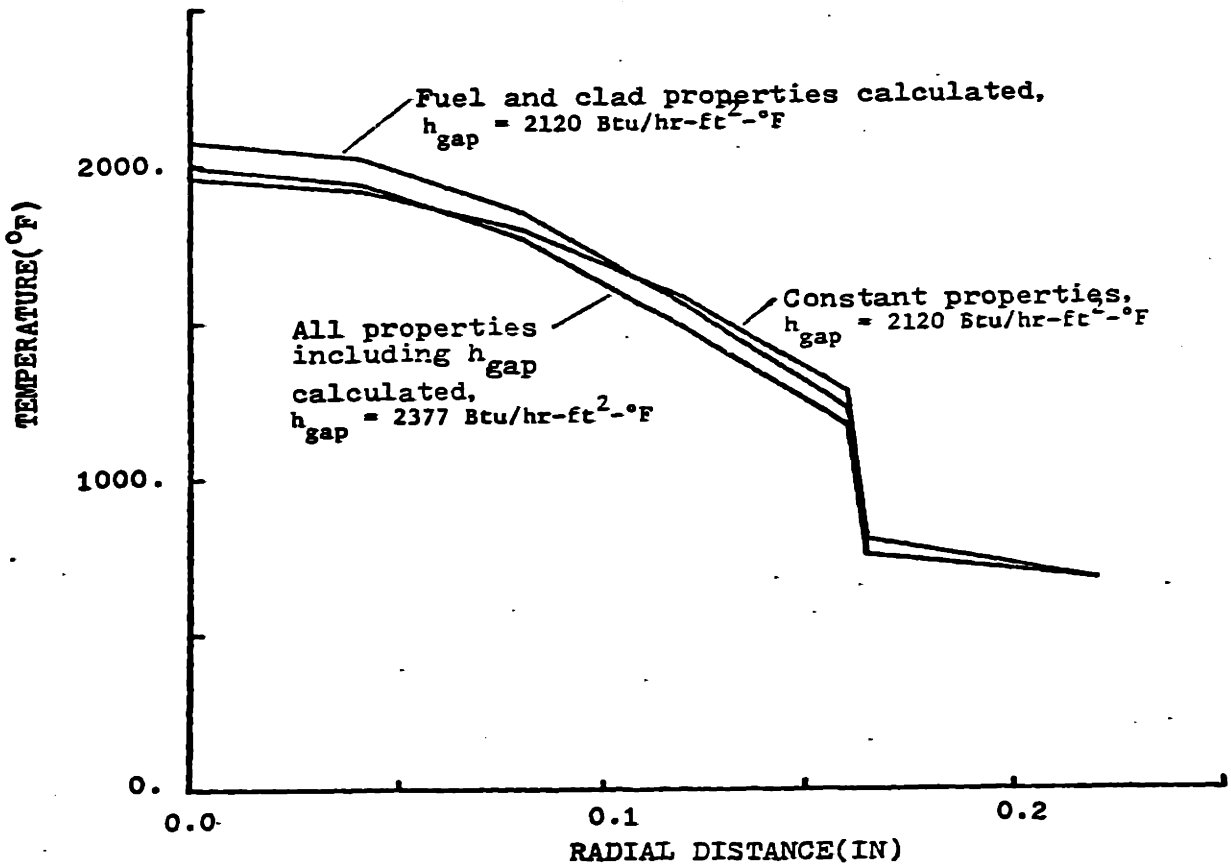


Figure IV-2

Predictions of the Three Options  
of New Fuel Rod Model

IV-5

Constant  $h_{gap} = 8000. \text{ W/m}^2\text{-k}$

Calculated  $h_{gap} = 3020. \text{ at } t=0$

Calculated  $h_{gap} = 4110. \text{ at } t=\infty$

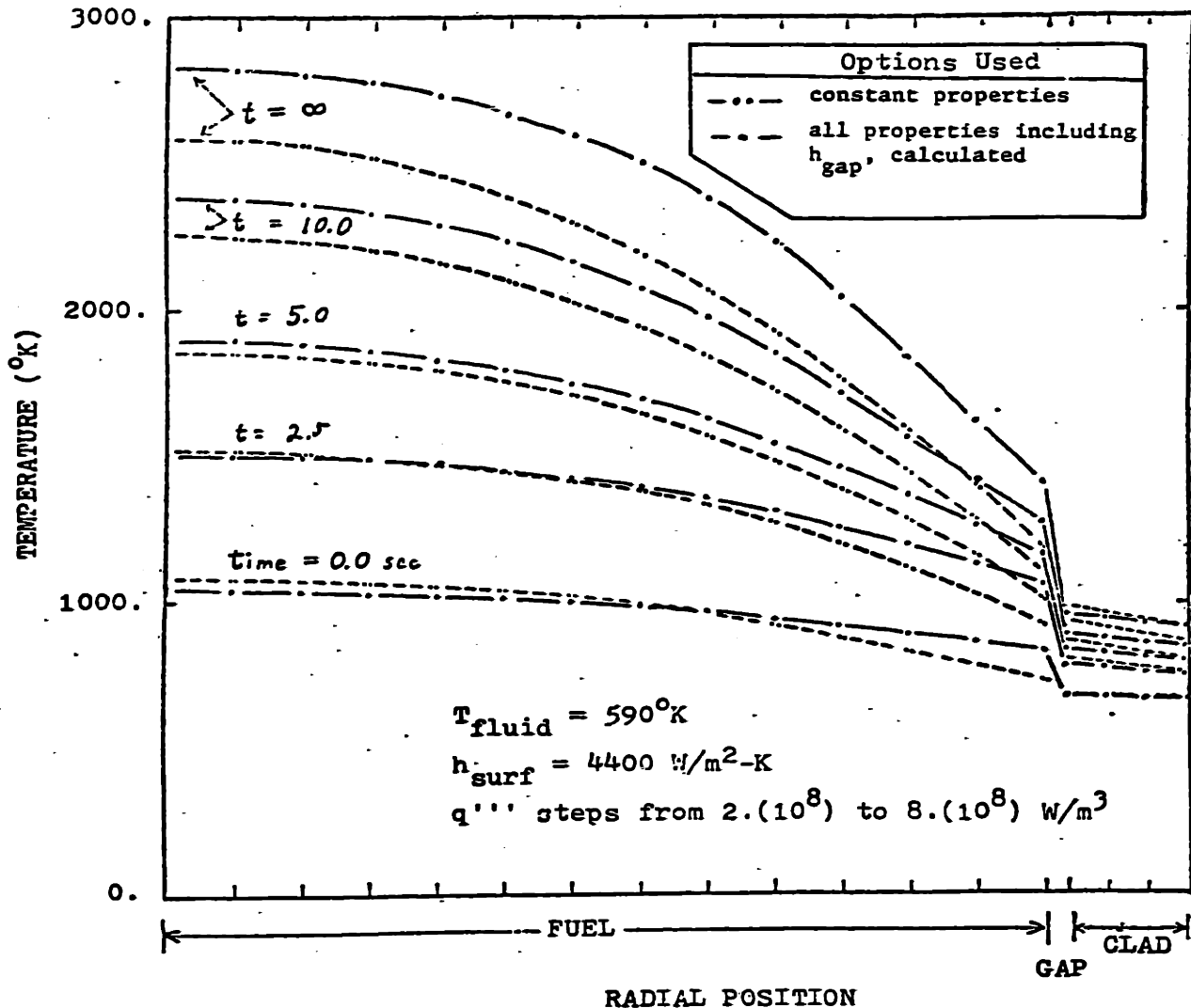


Figure IV-3

Transient Predictions for Two Options  
of the New Fuel Rod Model

four. Eventually the temperature profiles reached a new steady state. One of the differences between the two sets of steady state predictions is seen in the change of centerline temperature predictions. At time zero the centerline temperature obtained using calculated properties is less than the value obtained using constant properties. At the new steady state ( $t=\infty$ ) the centerline temperature prediction based on constant properties is less than the value predicted using calculated properties and  $h_{gap}$ .

## 2. Testing of New Rod-to-Coolant Heat Transfer Model

The new rod-to-coolant heat transfer model was tested by running two test cases using the old and new heat transfer models. Steady state and transient pre-CHF conditions were considered.

### a. Steady State Predictions

Steady state predictions were obtained for a case which consisted of three BWR channels with different radial peaking factors. Predictions for coolant parameters such as enthalpy and density were nearly the same for both the old and new models. Wall temperature predictions showed differences as great as  $40^{\circ}\text{F}$  in the hot channel as can be seen in Figure IV-4. The wall temperature predictions of the new model vary more smoothly than those of the old. The coolant temperature reaches a plateau near the inlet, indicating the axial position where voiding occurs. The old heat transfer model uses voiding to switch from a forced convection to a nucleate boiling heat transfer correlation. This switch causes the sharp discontinuity in clad temperature predictions based on the old model, as shown in Figure IV-4 and also earlier in Figure II-1. In spite of the large differences in wall temperature predictions of the old and new heat transfer models shown in Figure IV-4, the MDNBR predictions are nearly identical.

### b. Transient Predictions

A transient case was analyzed which considered adjacent PWR channels. These channels were assumed to be initially at nearly zero power and then subjected to a short burst of power sufficient to cause some voiding. This case was run previously as part of the comparison of COBRA-IIIC/MIT and COBRA-IV-I described in Ref. 12.

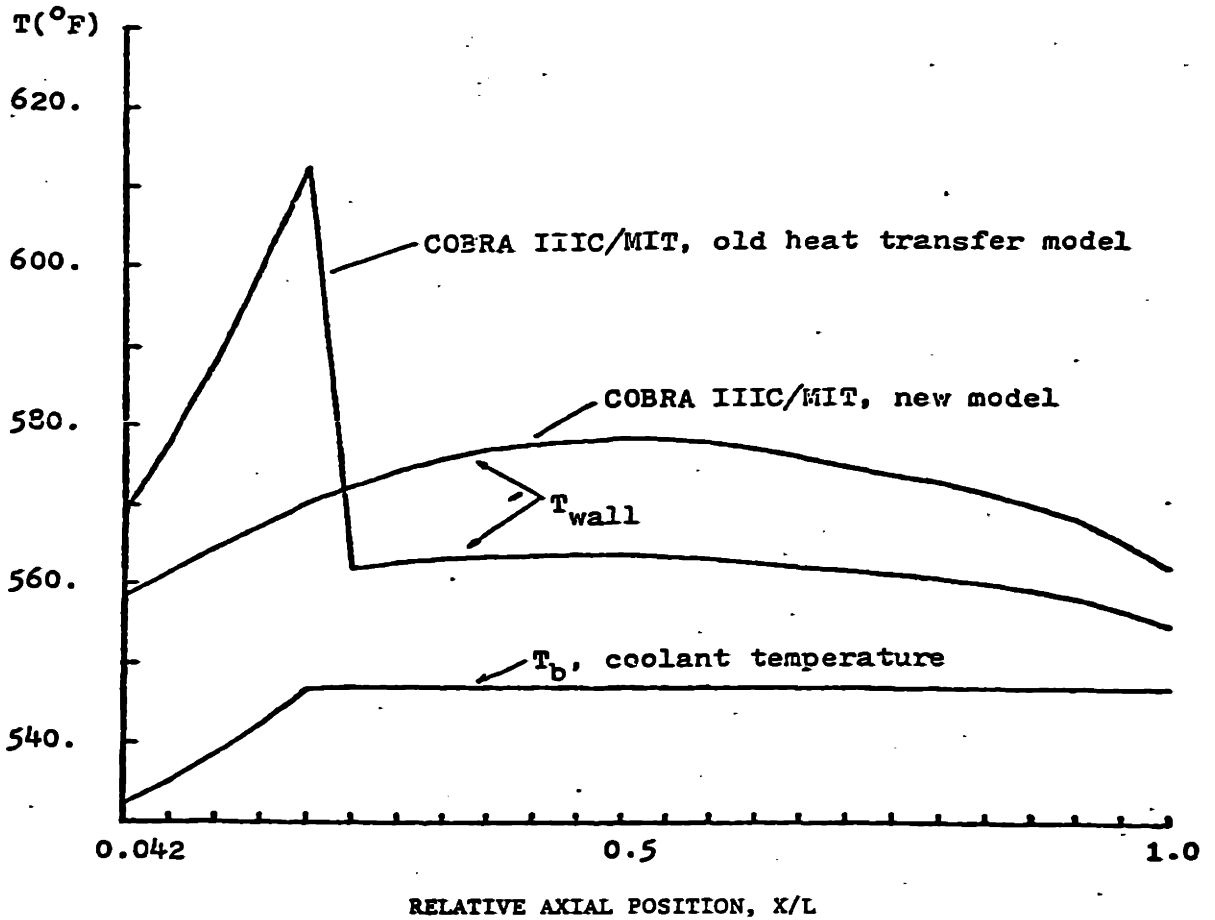


Figure IV-4

Axial Temperature Profiles for Steady State  
BWR Hot Channel Calculation



The case was rerun using the new heat transfer model of COBRA-IIIC/MIT. The channel pressure drop predictions are shown as a function of time in Figure IV-5. The COBRA-IIIC/MIT results using the new heat transfer model show a much lower pressure drop spike before  $t=0.2$  sec. than the results using the old heat transfer model. The difference in behavior between the old and new heat transfer models is due to the discontinuity of the old heat transfer model predictions when voiding occurs. The pressure drop predictions of the new COBRA-IIIC/MIT heat transfer model are similar to those of COBRA-IV-I, which also uses an advanced heat transfer package capable of constructing a complete boiling curve. The heat transfer model of COBRA-IV-I is described in Appendix F.

### c. Conclusions

Testing of the new rod-to-coolant heat transfer model led to the following conclusions:

- 1) Heat transfer predictions of the new model vary smoothly as heat transfer changes from forced convection to the nucleate boiling heat transfer regime.
- 2) Clad temperature predictions showed differences which were explainable from differences in the heat transfer correlations and logic used.
- 3) Minimum Departure from Nucleate Boiling Ratio (MDNBR) predictions were nearly the same.
- 4) Predictions of coolant parameters such as density, enthalpy, and pressure drop were the same for both models in steady state.

### 3. Testing of New Mixing Model

The new mixing model was tested by comparing COBRA-IIIC/MIT predictions with data from the GE 9-Rod Mixing Tests and the Columbia 16-Rod Mixing Tests (Ref. 26). COBRA-IIIC/MIT predictions for the test cases were obtained using the new mixing model and  $\beta=0.02$ . Predictions using the two models for mixing are compared to experimental data in the following subsections.

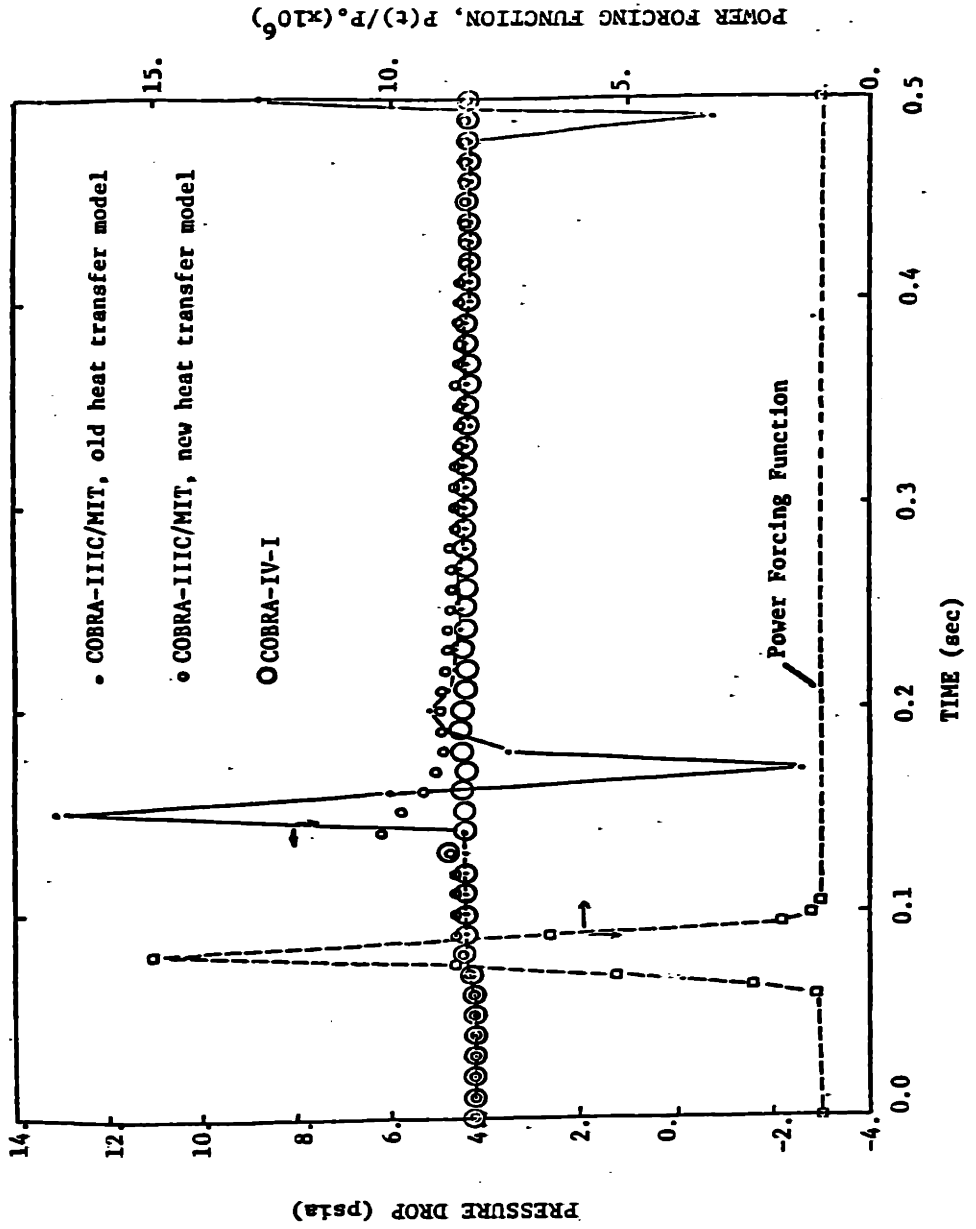


Figure IV-5  
Channel Pressure Drop vs. Time

a. Comparison with GE 9-Rod Mixing Tests

1) Description of Tests

The GE 9-rod mixing tests were carried out for a range of conditions typical of operating BWRs. The experiments were performed using water. The test section was an electrically heated 3x3 rod bundle. Pressure and enthalpy measurements were made for corner, side and interior subchannels. The geometry, test conditions and measurement locations are shown in Figure IV-6. Nine test cases were analyzed with COBRA-IIIC/MIT using the old and new mixing models. The analysis was done for one-fourth of an assembly, assuming quarter-assembly symmetry. The cases analyzed are listed in Table IV-1.

2) Comparison of COBRA-IIIC/MIT Predictions with Test Cases

Four isothermal test cases (1B, 1C, 1D and 1E) were analyzed with COBRA-IIIC/MIT. Axial friction pressures drop predictions for the isothermal test cases were made similar to experimental measurements by adjusting the single-phase friction factor correlation. The usual form for the correlation, given below, was used.

$$f = a(Re)^b$$

(Eqn. IV-1)

The "a" coefficient was set equal to 0.286. "b" was given the smooth-tube friction factor correlation value of -0.2. Comparisons of predicted and measured axial friction pressure drops are given in Table IV-2. COBRA exit mass flow distribution predictions are compared with experimental data in Figure IV-7. Each curve in the figure is based on three calculated values of data points. These are the values of normalized mass flux for the corner, side and center subchannels. The COBRA predictions for each subchannel are within 1% of one another for all four isothermal cases. The COBRA predictions are within the spread of data in the corner subchannel and near the spread of data in the side and corner subchannels.

Figure IV-6

GE 9-Rod Mixing Tests  
Geometry, Test Conditions and Measurement Locations (Ref. 25)

Number of Rods	9
Rod Diameter	.570 inch
Radius of Corner Subchannel	.420 inch
Rod Rod Clearance	.168 inch
Rod Wall Clearance	.135 inch
Hydraulic Diameter	.474 inch
Heated Length	72 inch
Pressure	1000 psia
Average Bundle Mass Flux	0.48 to 1.970 $\text{Mlbm/hr-ft}^2$
Average Heat Flux	0.225 to 0.675 $\text{MBtu/hr-ft}^2$
Inlet Subcooling	29.1 to 504.6 $\text{Btu/lbm}$

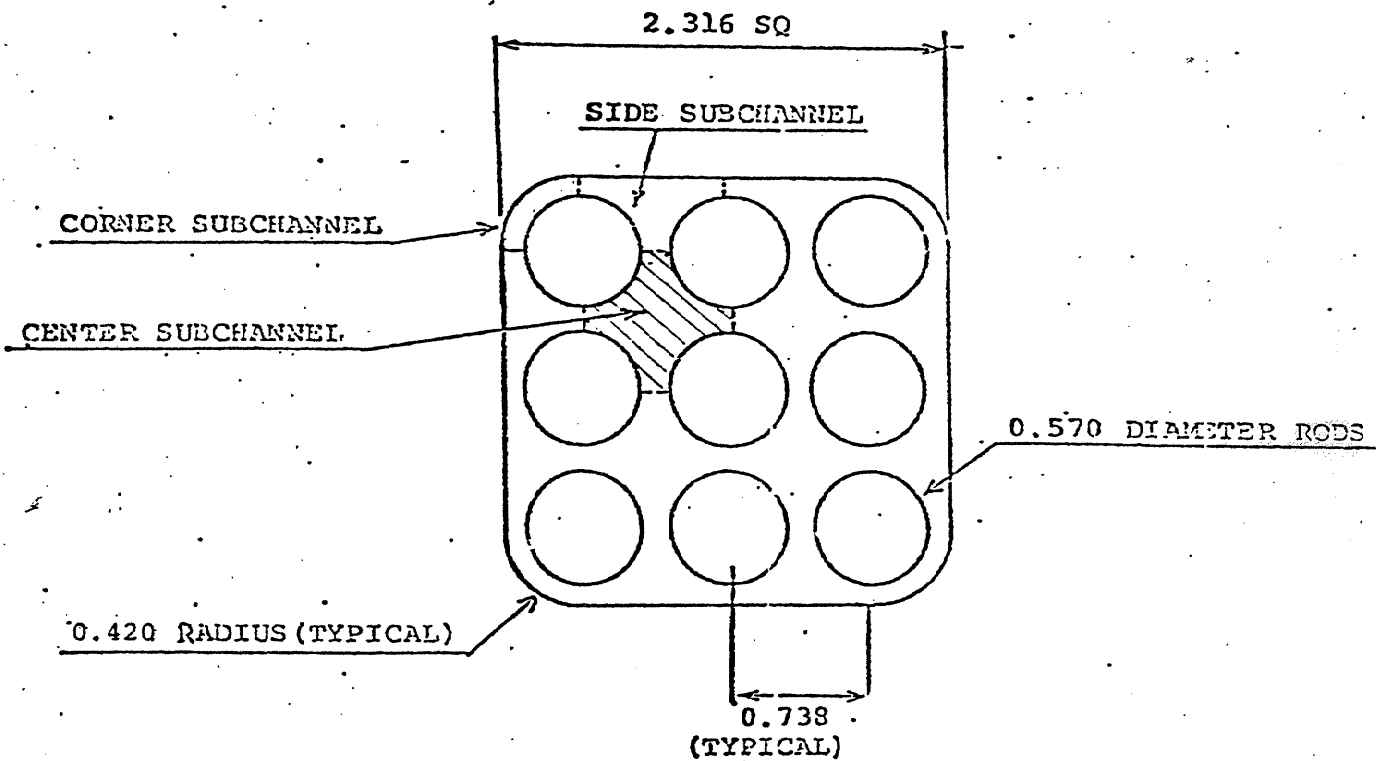


Table IV-1

GE 9-Rod Mixing Test Cases Analyzed

Test Case Number	Mass Flux (Mlb/hr ft <sup>2</sup> )	Average Heat Flux (MBTU/hr ft <sup>2</sup> )	Power Distribution	Inlet Subcooling (BTU/lb)	Average Exit Quality	Boiling Length L <sub>B</sub> /L
1B	0.48	0.0	-	504.6	0.	0.00
1C	0.99	0.0	-	504.6	0.	0.00
1D	1.51	0.0	-	504.6	0.	0.00
1E	1.97	0.0	-	504.6	0.	0.00
2G1	1.070	0.675	uniform	225.9	0.038	0.10
2G2	1.080	0.675	uniform	189.8	0.090	0.24
2G3	1.070	0.675	uniform	146.7	0.160	0.41

IV-12

Range of Data Base for Beus Correlation  
(Ref. 4)

System Pressure	50 ≤ P ≤ 775	psia
Mass Flux	.073 ≤ G ≤ 3.	Mlb/hr ft <sup>2</sup>
Quality	-0.2 ≤ X ≤ .80	
Gap Width Between Subchannels	.02 ≤ S ≤ .10	in.

Table IV-2

Measured and Predicted Axial Friction Pressure Drop

Test Case	$(\Delta P_f)$ measured (psia)	$(\Delta P_f)$ predicted* (psia)
1B	0.2128	0.21
1C	0.7130	0.75
1D	1.596	1.60
1E	2.540	2.58
1D (repeated)	1.610	1.60

\*Frictional pressure drop with COBRA-IIIC/MIT using the single-phase friction correlation  
 $f = 0.286(\text{Re})^{-0.2}$  .

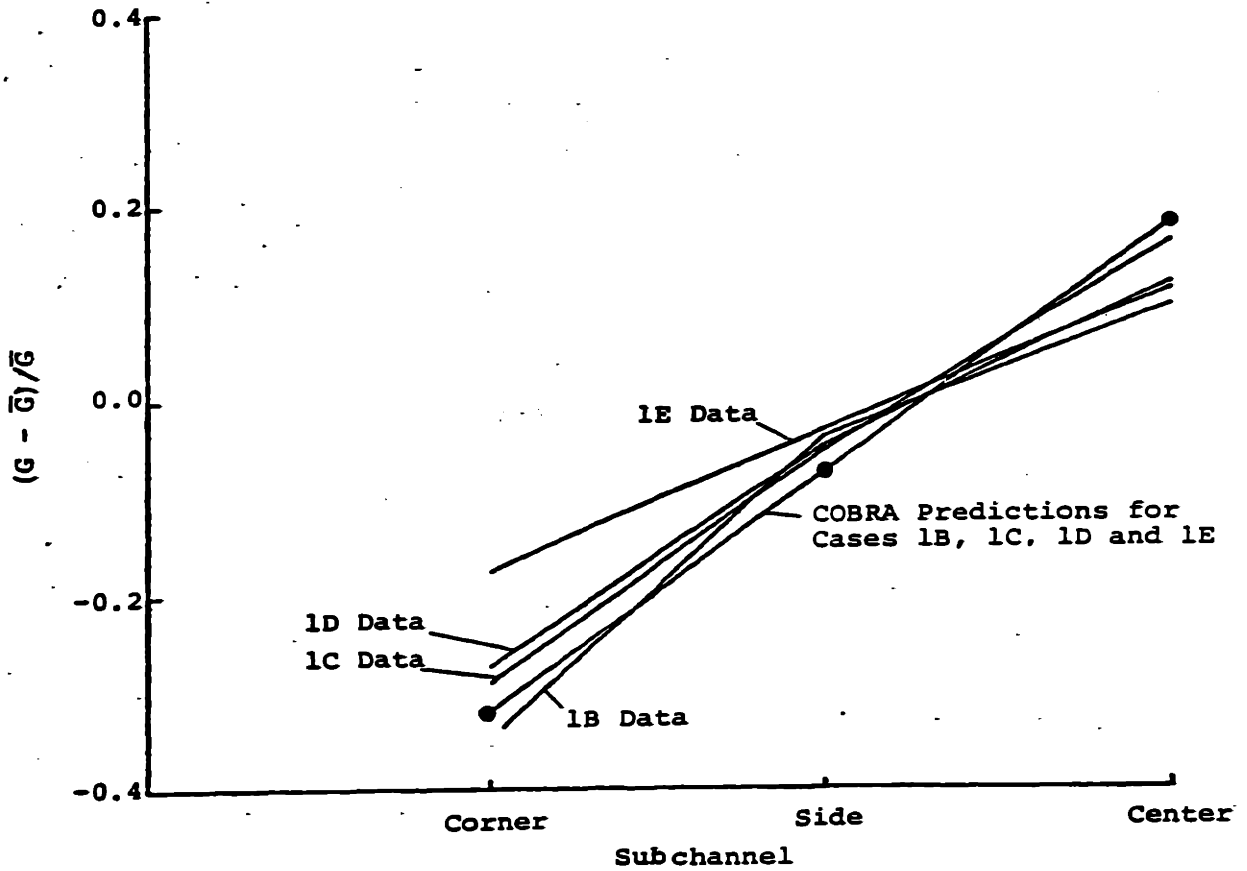


Figure IV-7

GE Mixing Test Cases 1B, 1C, 1D and 1E  
Normalized Exit Mass Flux Distributions

Three adiabatic test cases (2G1, 2G2 and 2G3) were analyzed with COBRA-IIIC/MIT using the old and new mixing models. The analyses with the old model used  $\beta=0.02$ . The new mixing model is the Beus model. COBRA exit mass flux and enthalpy predictions are compared to data in Figures IV-8 through IV-11 and IV-13 through IV-15.

Figures IV-8 through IV-10 compare predicted and measured enthalpy distributions. Enthalpy becomes increasingly over-predicted as exit quality increases, going from case 2G1 to 2G2 and on to case 2G3. The enthalpy distribution predictions using  $\beta=0.02$  are essentially the same for all three cases.

Beus and  $\beta=0.02$  enthalpy predictions differ because the Beus model predicts less mixing: thus, the Beus model is similar to using a  $\beta$  less than 0.02. However, the Beus predictions do follow the quality dependence of the model's mixing predictions. This can be seen by comparing Figure IV-11, where corner sub-channel enthalpy predictions and data are compared for cases 2G1, 2G2 and 2G3 with Figure IV-12.

Figure IV-11 includes the predictions of a temporary modification of the Beus mixing model, whereby the single-phase component of Beus mixing,  $W_L$ , is predicted using  $\beta=0.02$ . This change affects the mixing predictions from subcooled conditions up to the beginning of the transition mixing region shown in Figure IV-10. Mixing rate predictions in the transition mixing region are unaffected by the modification. The normalized enthalpy prediction of  $\beta=0.02$  shown in Figure IV-9 changes little as quality increases.

In going from an exit quality of 0.038 to 0.16, the Beus corner subchannel enthalpy prediction falls and rises. The behavior is due to the increased turbulent interchange of enthalpy from the corner subchannel for case 2G2, where exit quality is 9%. For a given geometry, mass flow rate and pressure, Beus mixing predictions are a function of quality, as shown in Figure IV-12. The mixing rate starts at a single-phase liquid value and increases to a maximum value as quality increases. Then, mixing rate decrease asymptotically to a single-phase vapor value at high qualities. For cases 2G1, 2G2 and 2G3,



$$\bar{G} = 1.07 \text{ Mlb/hr ft}^2$$

$$q'' = 0.675 \text{ MBTU/hr ft}^2$$

$$\bar{x}_{\text{exit}} = 0.038$$

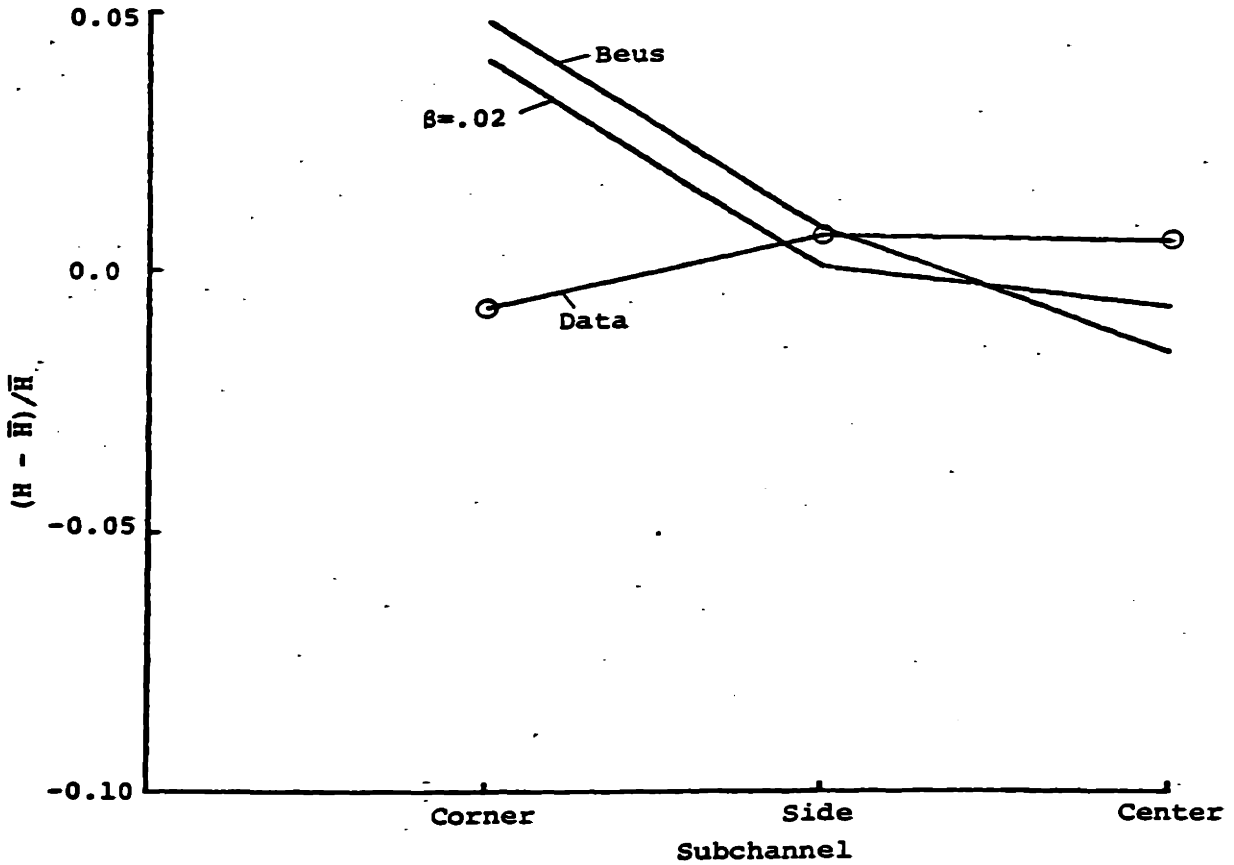


Figure IV-8

GE Mixing Test Case 2G1  
Normalized Exit Enthalpy Distribution

$$\bar{G} = 1.08 \text{ Mlb/hr ft}^2$$

$$q'' = 0.675 \text{ MBTU/hr ft}^2$$

$$\bar{x}_{\text{exit}} = 0.09$$

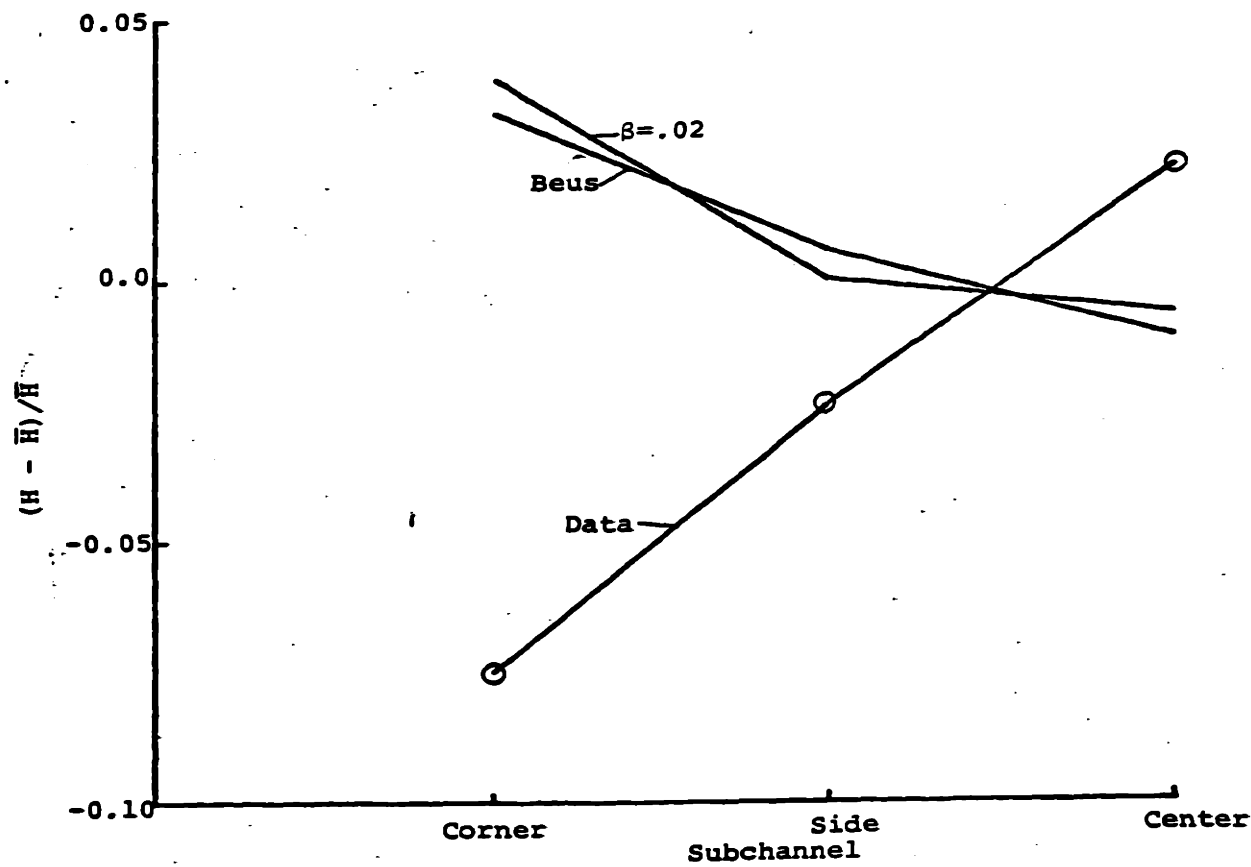


Figure IV-9  
GE Mixing Test Case 2G2  
Normalized Exit Enthalpy Distribution

IV-18

$$\begin{aligned}\bar{G} &= 1.07 \text{ Mlb/hr ft}^2 \\ q'' &= 0.675 \text{ MBTU/hr ft}^2 \\ \bar{x}_{\text{exit}} &= 0.16\end{aligned}$$

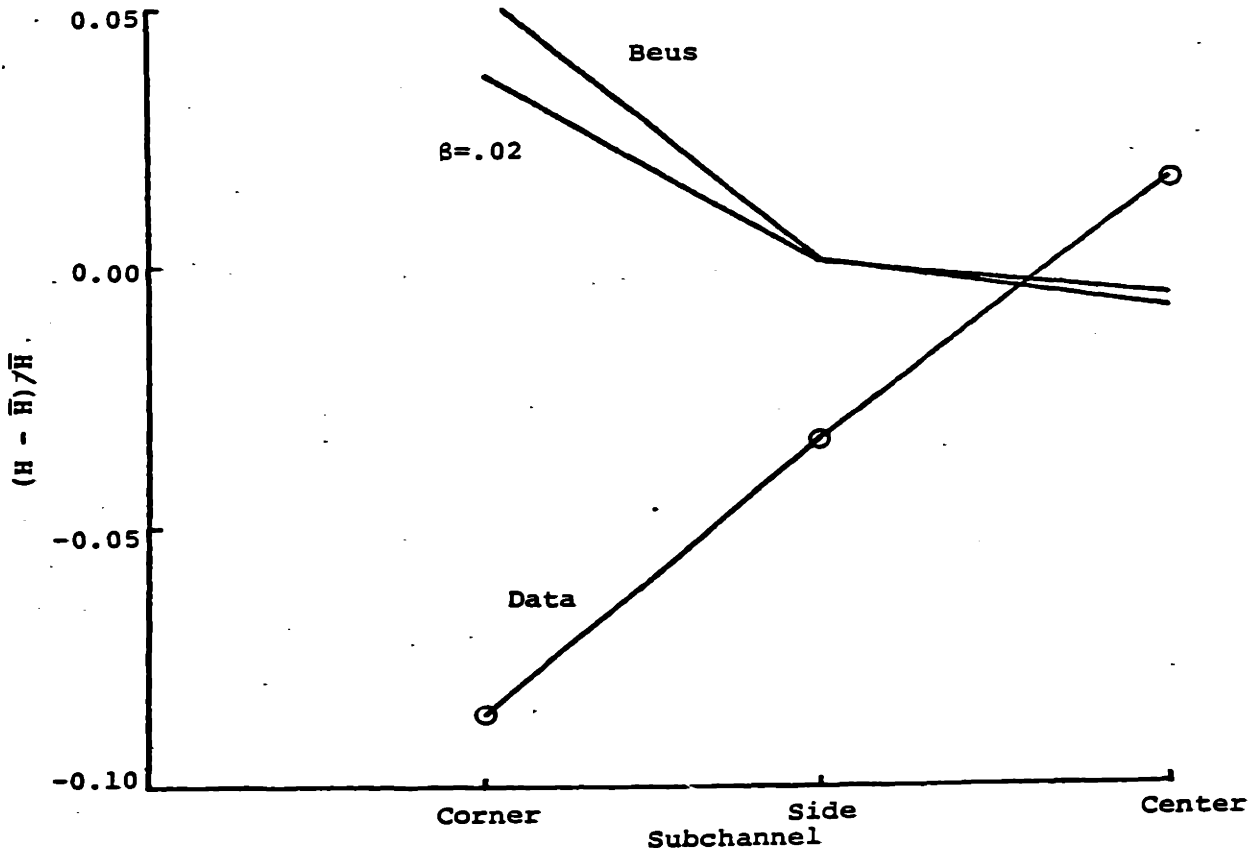


Figure IV-10  
GE Test Case 2G3  
Normalized Exit Enthalpy Distribution

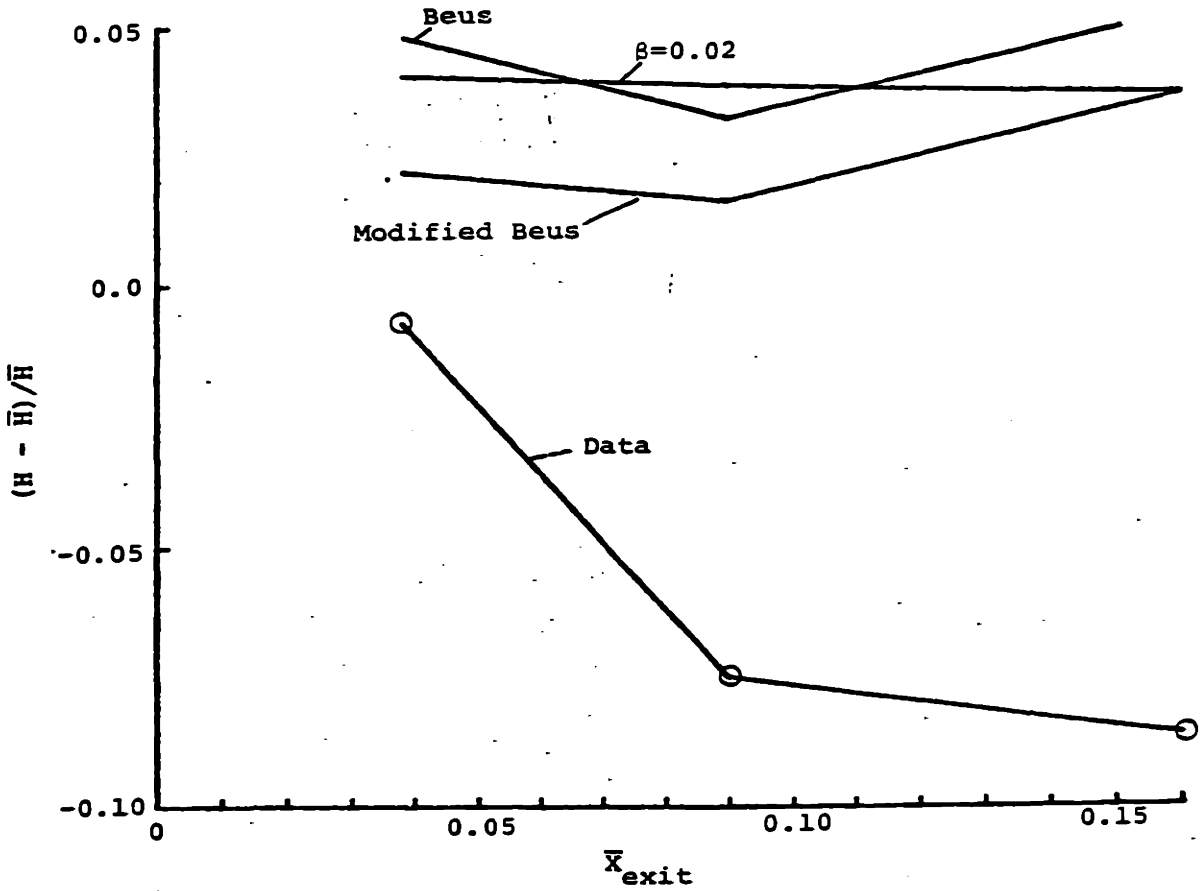


Figure IV-11  
GE Test Cases 2G1, 2G2 and 2G3  
Normalized Corner Channel Enthalpy vs. Exit Quality

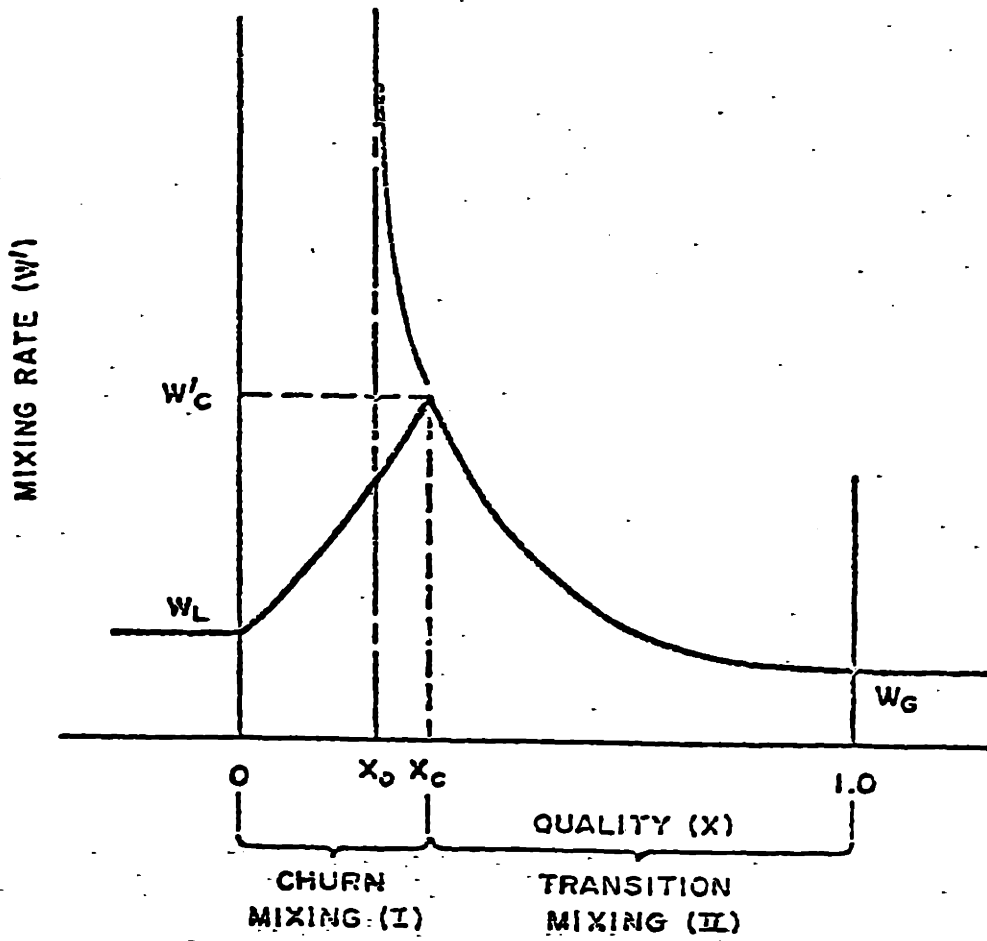


Figure IV-12

Plot of Mixing Model Showing Variation with Quality  
 (Fig. E.1 of Ref.17)

$$\bar{G} = 1.07 \text{ Mlb/hr ft}^2$$

$$q'' = 0.675 \text{ MBTU/hr ft}^2$$

$$\bar{X}_{\text{exit}} = 0.038$$

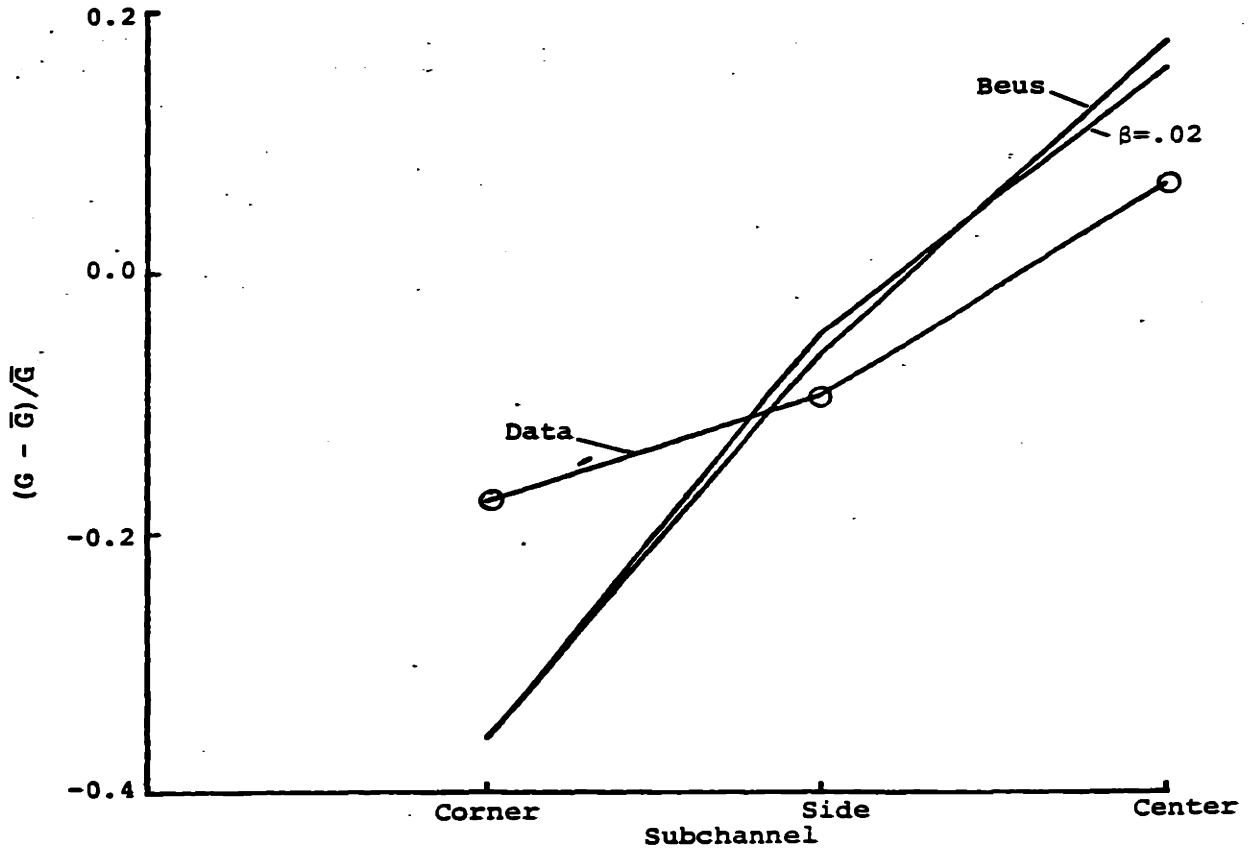


Figure IV-13  
GE Mixing Test Case 2G1  
Normalized Exit Mass Flux Distribution

IV-22

$$\begin{aligned}\bar{G} &= 1.08 \text{ Mlb/hr ft}^2 \\ q'' &= 0.675 \text{ MBTU/hr ft}^2 \\ \bar{X}_{\text{exit}} &= 0.09\end{aligned}$$

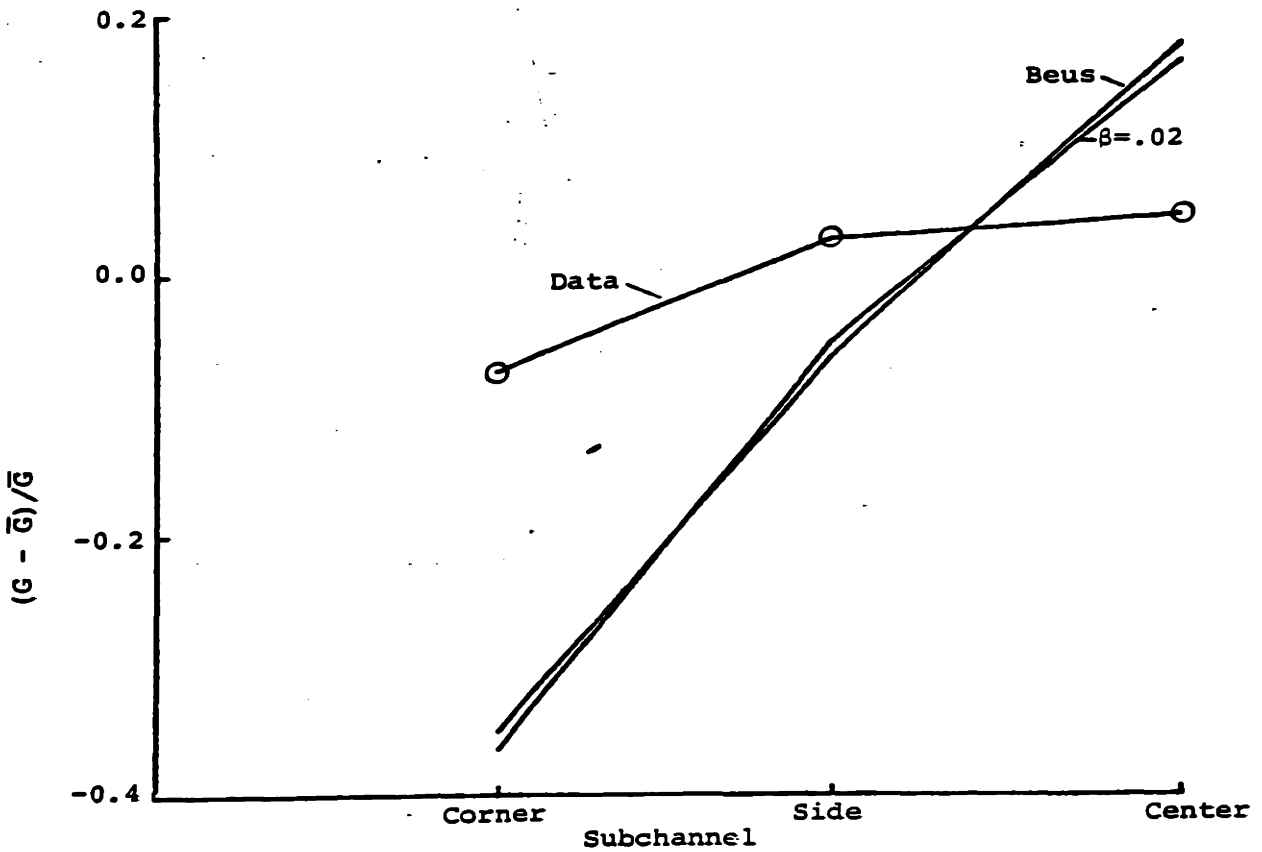


Figure IV-I4

GE Mixing Test Case 2G2  
Normalized Exit Mass Flux Distribution

IV-23

$$\bar{G} = 1.07 \text{ Mlb/hr ft}^2$$

$$q'' = 0.675 \text{ MBTU/hr ft}^2$$

$$\bar{x}_{\text{exit}} = 0.16$$

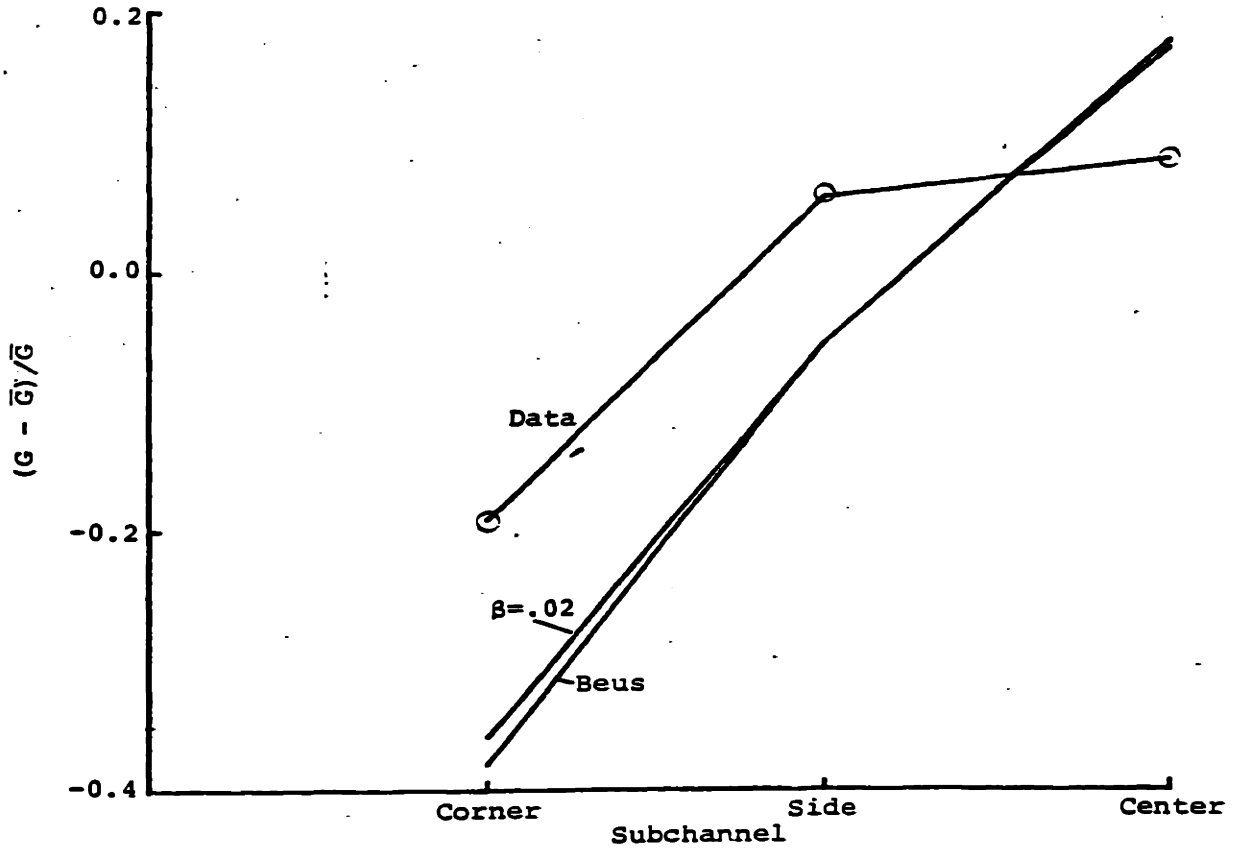


Figure IV-15

GE Mixing Test Case 2G3  
Normalized Exit Mass Flux Distribution



$X_c$ , the quality at which the peak mixing rate occurs, is about 10%. The normalized enthalpy predictions of the modified Beus model start lower, closer to the data than the other predictions and rise to meet the  $\beta=0.02$  predictions of 16% quality.

Figures IV-13 through IV-15 compare predicted and measured mass flux distributions. The effect of mixing rate on mass flow distribution is a second order effect. The general trends of predictions and data are similar. Mass flux was underpredicted in the corner subchannel and overpredicted in the center subchannel. Mass flux in the side subchannel is underpredicted for two of the three cases. The Beus and  $\beta=0.2$  mass flux distribution trends show little difference.

In conclusion, enthalpy distribution is predicted differently than data. Enthalpy is overpredicted in the corner subchannel and underpredicted in the center channel. Use of the Beus mixing model does not make much difference. A void-drift model or other similar approach is needed to account for the tendency of vapor to move toward the center of the bundle.

#### b. Comparison with Columbia 16-Rod Mixing Tests

##### 1) Description of Tests

The Columbia 16-rod mixing tests were carried out for both subcooled and boiling conditions using an electrically heated 4x4 bundle of typical PWR fuel geometry. Simultaneous measurements of water flow and enthalpy were made at the exits of two interior subchannels. The power profile was uniform in the axial direction but varied radially so as to provide a power tilt. The geometry, test conditions and measurement locations are shown in Figure IV-16.

##### 2) Comparison of COBRA-IIIC/MIT Predictions with Test Data

Nine test cases were analyzed with COBRA-IIIC/MIT using the old and new mixing models. The analyses were made for one-half of an assembly, assuming half-assembly symmetry. The cases analyzed are listed in Table IV-3. COBRA-IIIC/MIT predictions for channel 5 and 11 exit mass flux and enthalpy are compared with experimental measurements for cases 22, 25, 27, 29 and 30 in Figures IV-17 through IV-20.

Figure IV-16

Columbia 16-Rod Mixing Tests  
Geometry, Test Conditions and Measurement Locations  
 (Ref. 26)

Rod Outside Diameter	0.422 in.
Rod Pitch	0.555 in.
Rod to Wall Spacing	0.143 in.
Total Flow Area	0.02389 ft <sup>2</sup>
Radial Heat Flux	
Hot Rods (H)	100%
Cold Rods (C)	86%
Heated Length	60 in.
Pressure	500 and 1200 psia
Average Bundle Mass Flux	1 x 10 <sup>6</sup> ; 2 x 10 <sup>6</sup> ; 3 x 10 <sup>6</sup> lbm/hr ft <sup>2</sup>
Inlet Temperature	172°F to 484°F
Average Heat Flux	0.384 x 10 <sup>6</sup> ; 0.56 x 10 <sup>6</sup> ; 0.967 x 10 <sup>6</sup> BTU/hr ft <sup>2</sup>
Traverse Heating Ratio	Colder/Hotter: 0.86 Colder/Average: 1.02 Hotter/Average: 1.08

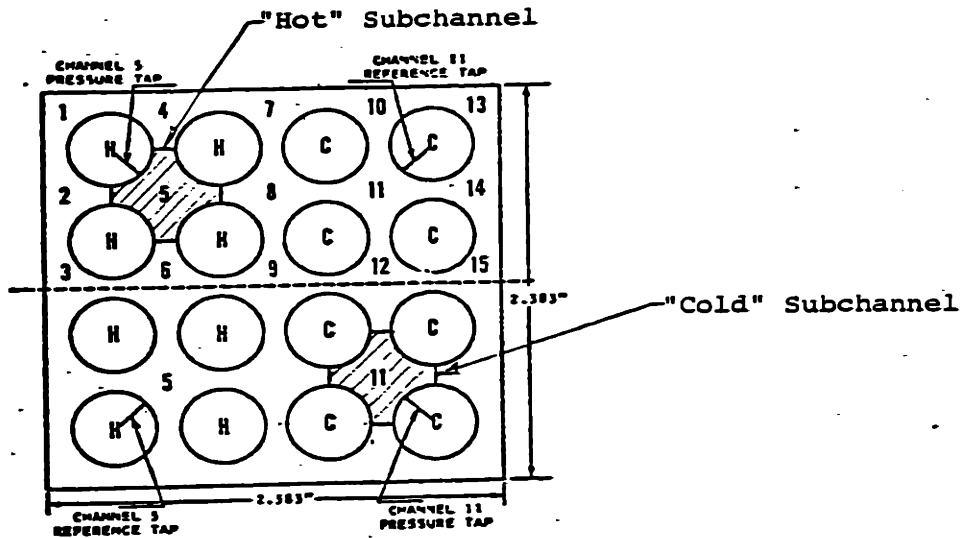


Table IV-3  
Columbia 16-Rod Mixing Test Case Analyzed

System Pressure, P = 1200 psia for all cases listed.

Test Case Number	Mass Flux (Mlb/hr ft <sup>2</sup> )	Average Heat Flux <sub>2</sub> (MBTU/hr ft <sup>2</sup> )	Power Distribution	Subcooling BTU/lb	Average Exit Quality	Boiling Length Fraction, L <sub>R</sub> /L
22	1.01	0.38	non-uniform	-400.	-0.424	0.00
25	1.01	0.38	non-uniform	-268.	-0.209	0.00
27	1.03	0.38	non-uniform	-217.	-0.132	0.00
29	1.00	0.38	non-uniform	-152.	-0.015	0.00
30	0.99	0.38	non-uniform	-124.	0.036	0.15
35	1.50	0.58	non-uniform	-301.	-0.317	0.00
39	1.50	0.58	non-uniform	-173.	-0.110	0.00
42	1.49	0.58	non-uniform	-137.	-0.051	0.00
90	1.48	0.58	non-uniform	-88.	0.028	0.16

Range of Data Base for Beus Correlation  
(Ref. 4)

System Pressure      50 ≤ P ≤ 775      psia  
 Mass Flux            .073 ≤ G ≤ 3.      Mlb/hr ft<sup>2</sup>  
 Quality                -0.2 ≤ X ≤ .80  
 Gap Width Between Subchannels      .02 ≤ S ≤ .10      in.

$$\bar{G} = 1. \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.38 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

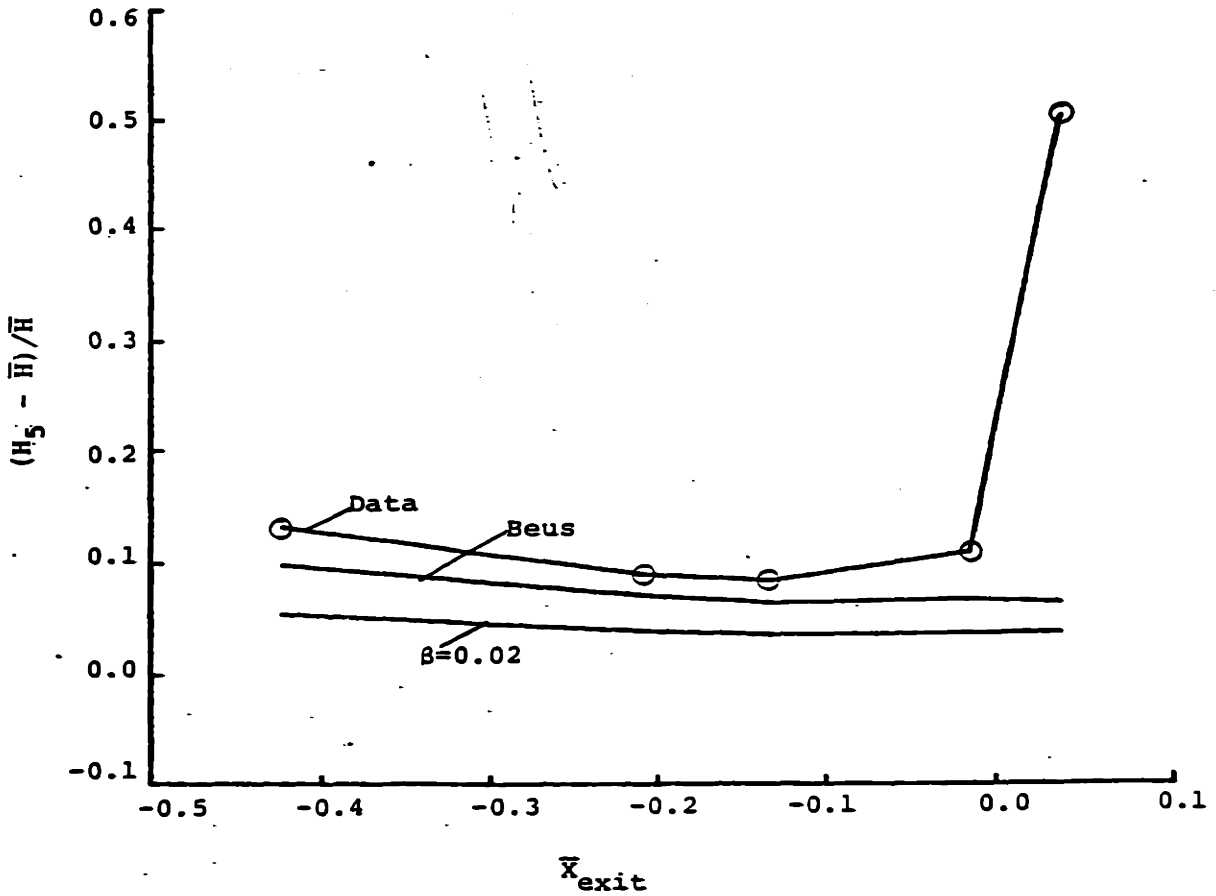


Figure IV-17

Columbia Test Cases 22, 25, 27, 29 and 30  
Normalized Channel 5 Exit Enthalpy vs. Quality

IV-28

$$G = 1. \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.38 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

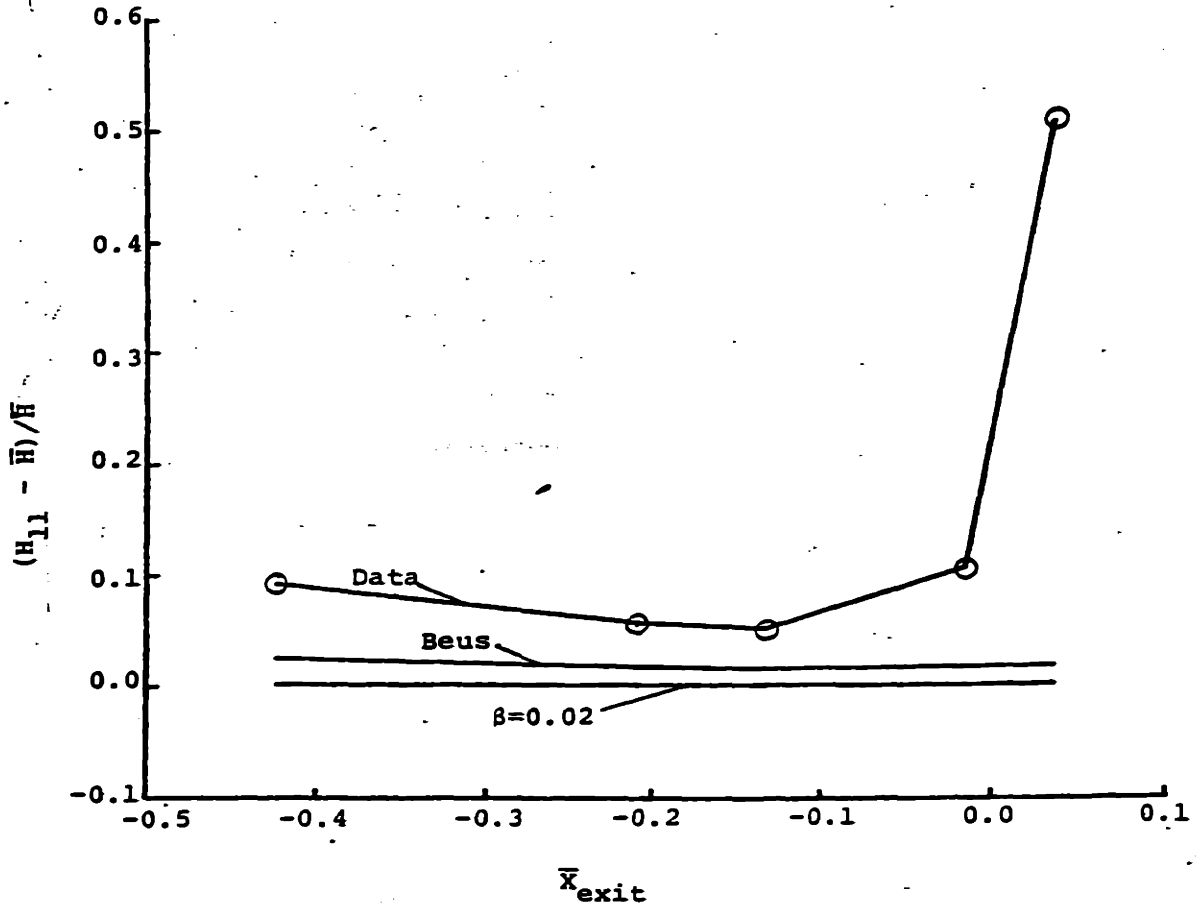


Figure IV-18

Columbia Test Cases 22, 25, 27, 29 and 30  
Normalized Channel 11 Exit Enthalpy vs. Quality

$$G = 1. \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.38 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200. \text{ psia}$$

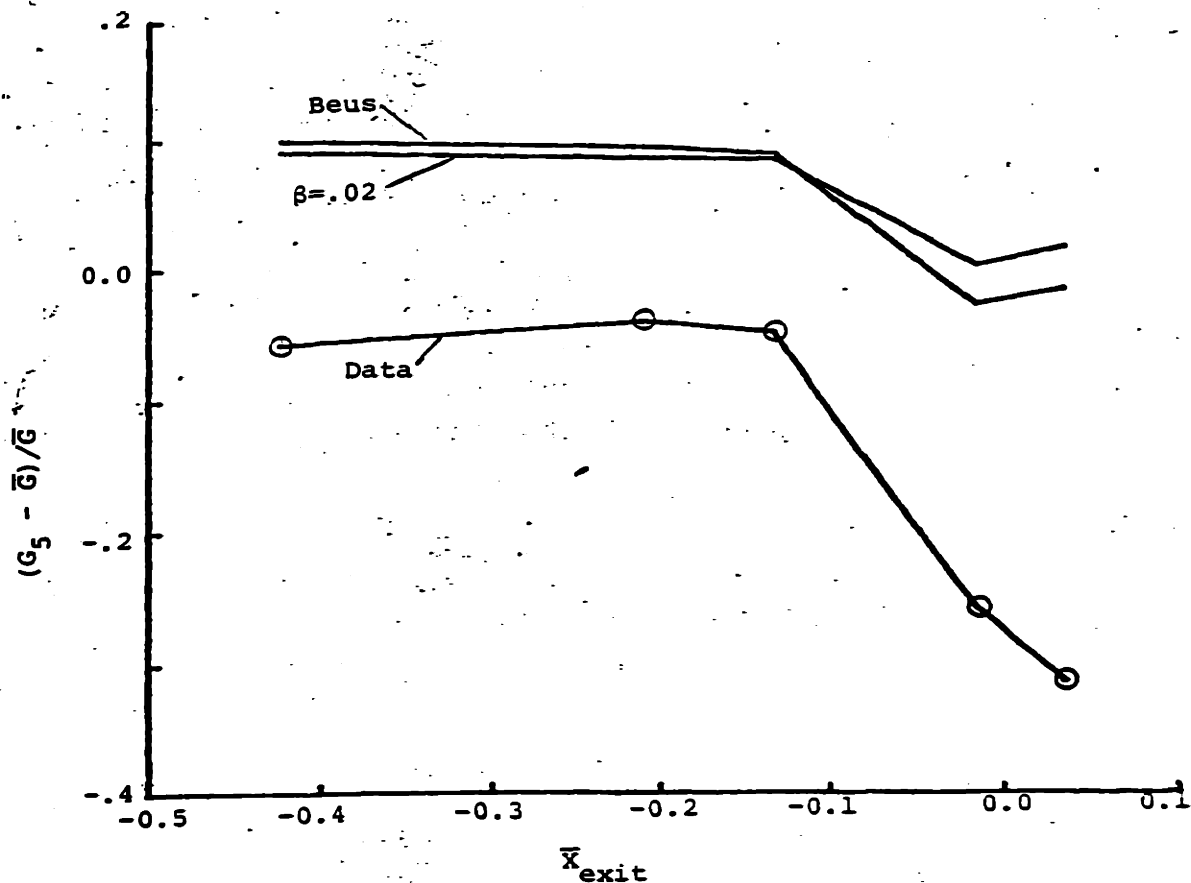
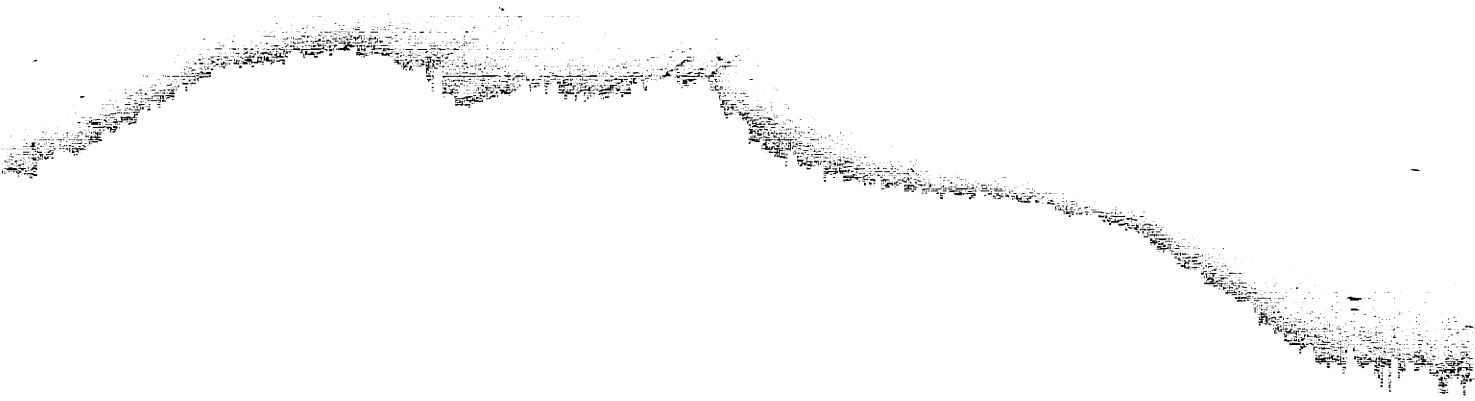


Figure IV-19

Columbia Test Cases 22, 25, 27, 29 and 30  
Normalized Channel 5 Exit Mass Flux vs. Quality



$$G = 1. \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.38 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200. \text{ psia}$$

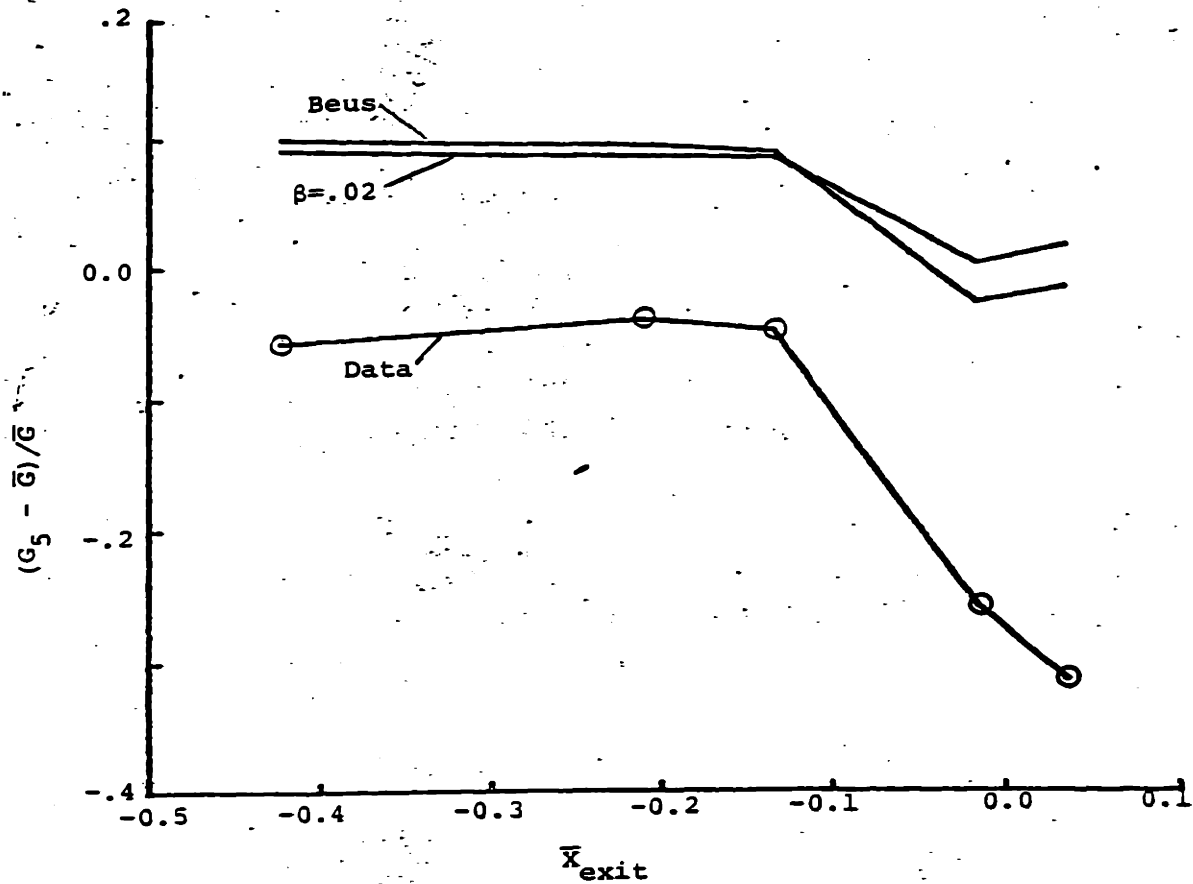


Figure IV-19

Columbia Test Cases 22, 25, 27, 29 and 30  
Normalized Channel 5 Exit Mass Flux vs. Quality



IV-30

$$\bar{G} = 1. \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.38 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$p = 1200. \text{ psia}$$

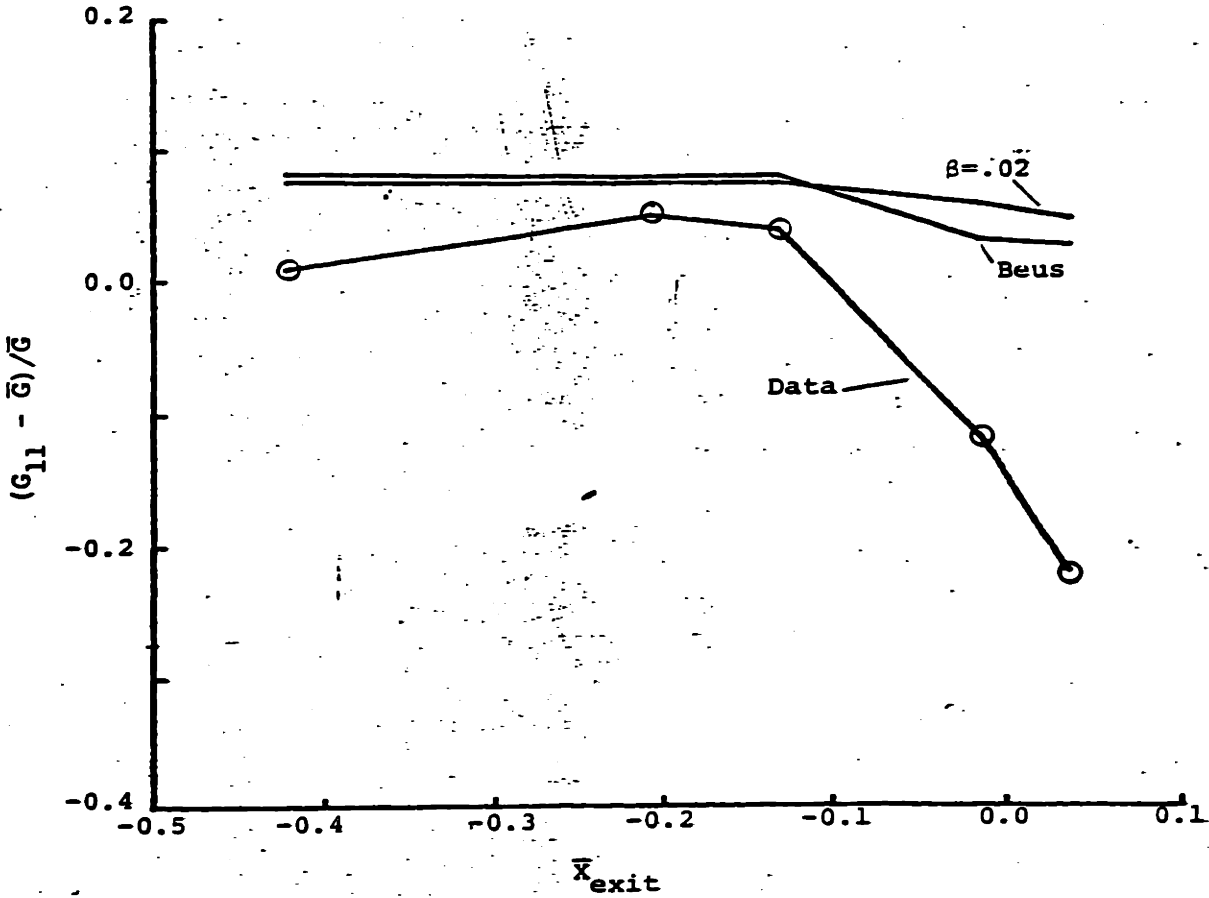


Figure IV-20

Columbia Test Cases 22, 25, 27, 29 and 30  
Normalized Channel 11 Exit Mass Flux vs. Quality

Normalized exit enthalpy as a function of average exit quality is shown for channels 5 and 11 in Figures IV-17 and IV-18, respectively. The data shows higher than average enthalpy in channels 5 and 11. Beus predicts a higher than average enthalpy but less than the data. The  $\beta=0.02$  enthalpy predictions are less than the Beus predictions because  $\beta=0.02$  predicts greater mixing than the Beus model. The sharp normalized enthalpy increase in channels 5 and 11 as exit quality increases in the vicinity of saturated liquid conditions is not reflected in the predictions.

Normalized exit mass flux is shown as a function of average exit quality for channels 5 and 11 in Figures IV-19 and IV-20, respectively. The data shows a general decline of normalized mass flux in channels 5 and 11 as exit quality increase above -0.1. The predictions are similar for each channel, as expected, since the effect on mass flux distribution is a second order effect, especially in the single-phase liquid flow regime. Mass flux was overpredicted in channels 5 and 11.

Data and predictions for higher mass and heat flux case 35, 39, 42 and 90 show behavior similar to data and predictions discussed for cases 22, 25, 27, 29 and 30. However, predictions were closer to data, especially the Beus predictions. Channel 5 and 11 exit enthalpies were closer to bundle average values. The results for cases 35, 39, 42 and 90 are shown in Figures IV-21 through IV-24.

In summary, predictions are closer to data for subcooled conditions. Data trends for boiling conditions are not well predicted. The Beus predictions are closer to data than  $\beta=0.02$  predictions. Enthalpy is predicted closer to data than mass flux. The data for high mass and heat flux cases are more closely predicted.

#### 4. Testing of New Correlations for Critical Power Ratio and Critical Heat Flux Ratio

The CISE-4 correlation for CPR and Hench-Levy correlation for CHF were tested using the GE 9-Rod CHF Tests (Ref. 27). The Biasi/Void-CHF correlation for CHF has not been tested. CPR and CHF predictions were obtained for conditions under which CHF was experimentally found to occur.

IV-32

$$\bar{G} = 1.5 \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = .53 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

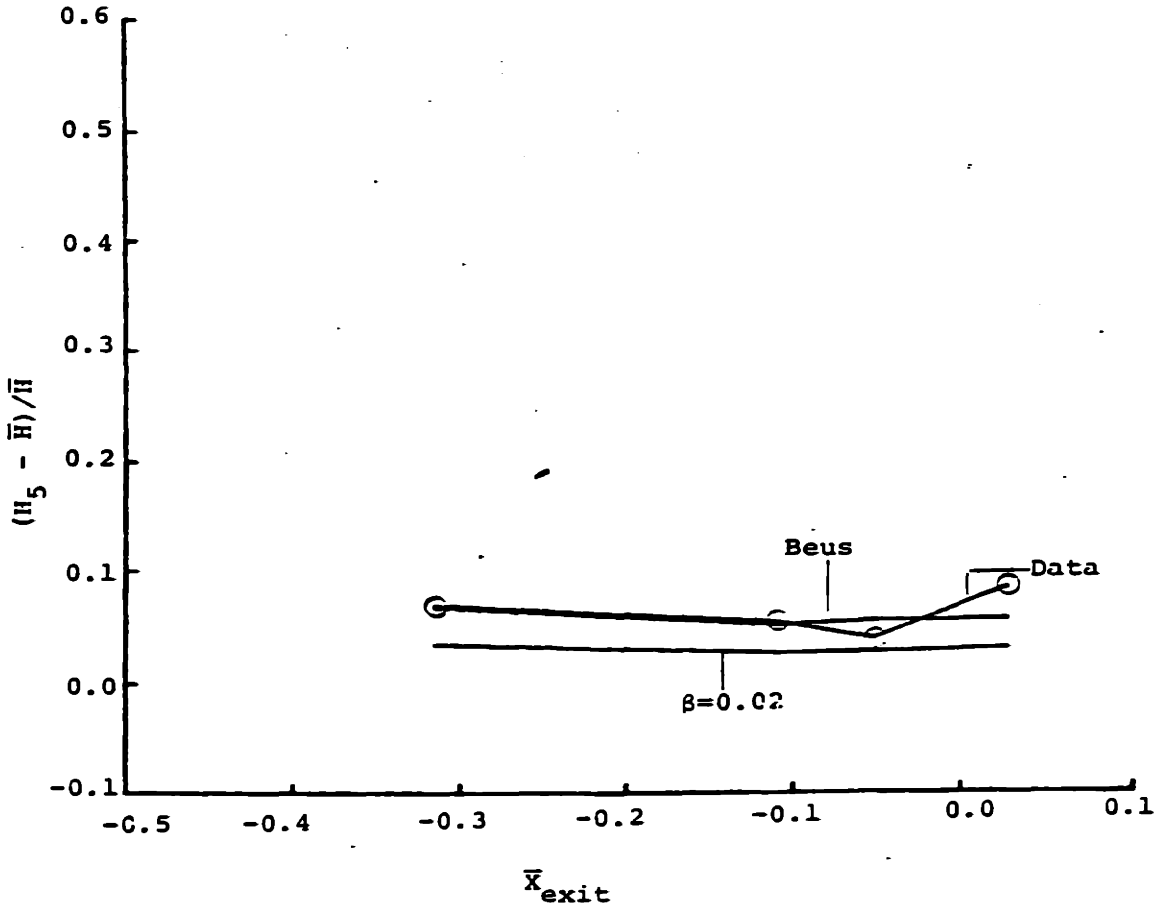


Figure IV-21  
Columbia Test Cases 35, 39, 42, and 90  
Normalized Channel 5 Exit Enthalpy vs. Quality

$$\bar{G} = 1.5 \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = .53 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

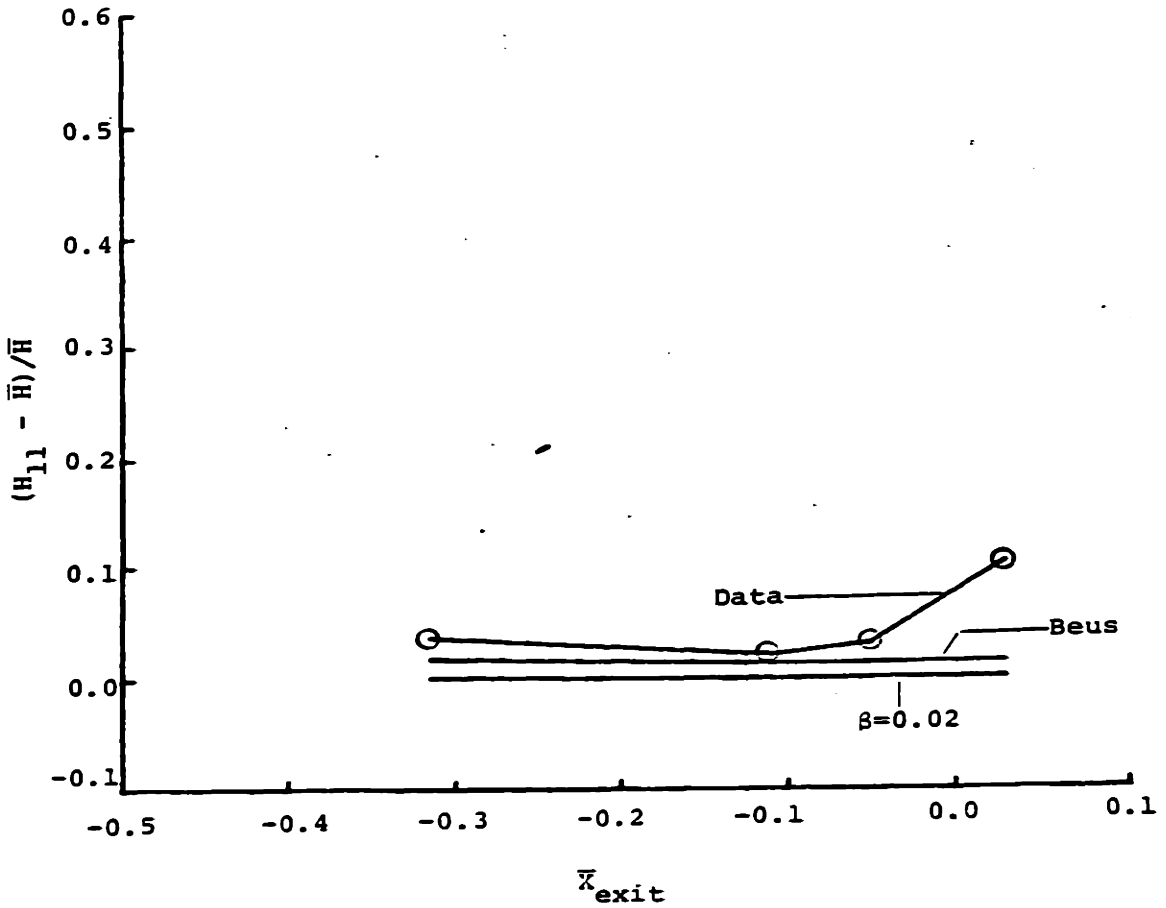


Figure IV-22.  
Columbia Test Cases 35, 39, 42, and 90  
Normalized Channel 11 Exit Enthalpy vs. Quality

IV-34

$$\bar{G} = 1.5 \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.58 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

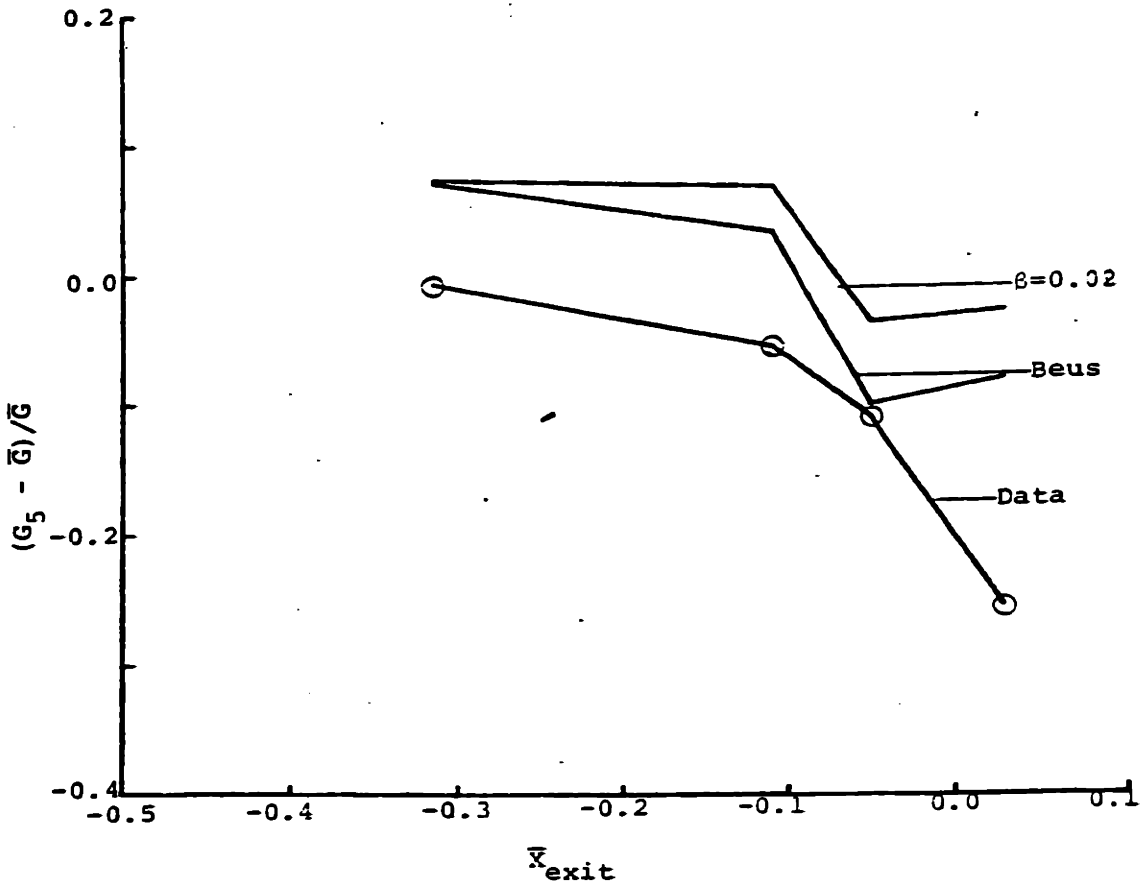


Figure IV-23

Columbia Test Cases 35, 39, 42, and 90  
Normalized Channel 5 Exit Mass Flux vs. Quality

IV-35

$$\bar{G} = 1.5 \frac{\text{Mlb}}{\text{hr ft}^2}$$

$$q'' = 0.58 \frac{\text{MBTU}}{\text{hr ft}^2}$$

$$P = 1200 \text{ psia}$$

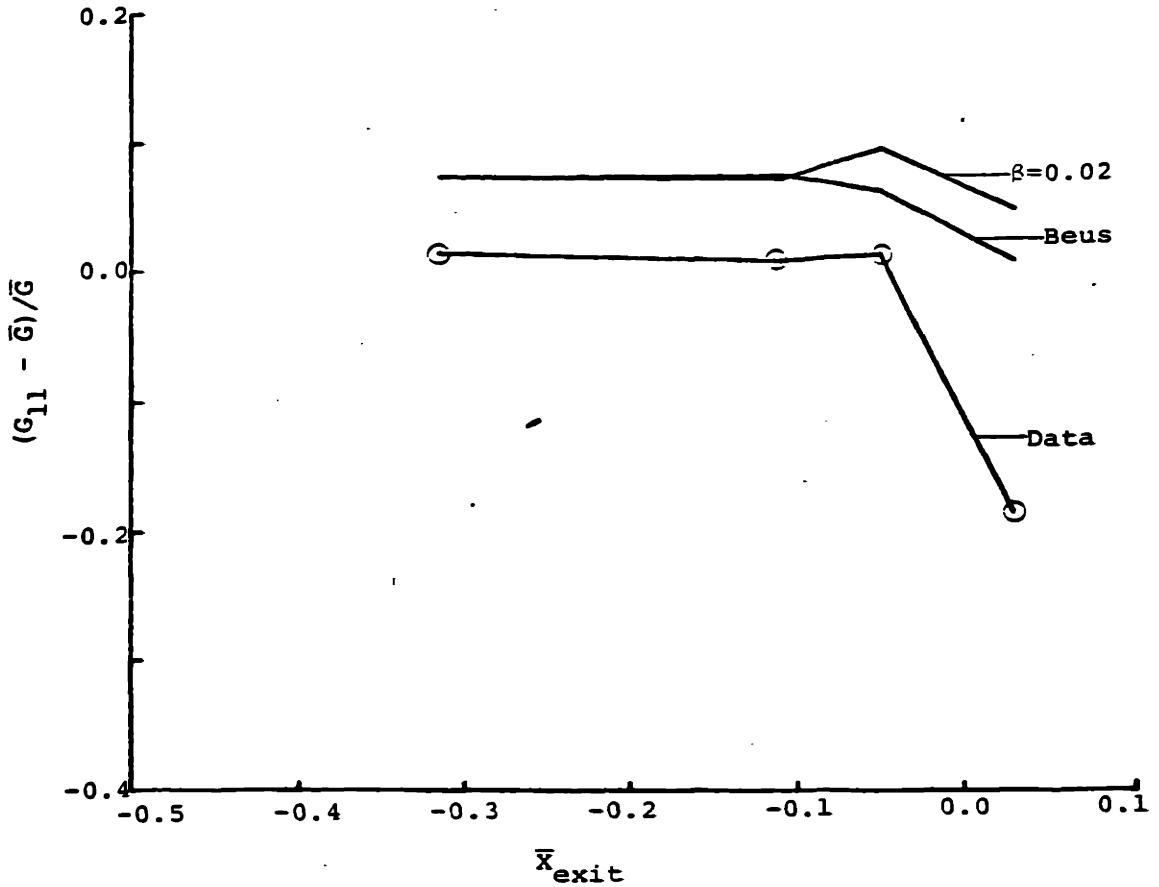


Figure IV-24

Columbia Test Cases 35, 39, 42, and 90  
Normalized Channel 11 Exit Mass Flux vs. Quality

a. Description of GE 9-Rod CHF Tests

The GE 9-Rod CHF tests were carried out using the bundle geometry and test conditions shown in Figure IV-25. The five test channels shown in Figure IV-26 were used. The test channels all had the same grid type spacers and rods. Surface heat flux was uniform axially and radially. The test channels had different heated length and spacer-locations.

b. Comparison of COBRA-IIIC/MIT Predictions with Data

Channel 3 and 4 test cases listed in Table IV-4 were analyzed using COBRA. CISE-4 critical power ratio (CPR) and Hensch-Levy CHF predictions were obtained. CISE-4 was developed for rod-centered subchannel analysis. Coolant-centered subchannel analysis, the type COBRA performs, is less suitable for CISE-4 than analyzing the 9-rod bundle as a single channel. However, COBRA subchannel analysis is appropriate for use with the Hensch-Levy CHF correlation.

For comparison purposes, CISE-4 CPR and Hensch-Levy CHF predictions were obtained using the single channel and sub-channel analysis methods for test cases 266 and 268. Predictions using the two analysis methods are compared in Table IV-5. The CISE-4 and Hensch-Levy predictions are less conservative using single channel analysis. In order to show how the least conservative method compares with the experimental data, single channel analysis was used to analyze the rest of the test cases analyzed.

Single channel analysis MCPR and MCHFR predictions for test channels 3 and 4 are given in Table IV-4. All the Hensch-Levy MCHFR predictions are conservative. The CISE-4 MCPR predictions are not nearly as conservative as the Hensch-Levy MCHFR predictions. The MCPR predictions are slightly non-conservative for one of six channel 3 cases and three of four channel 4 cases. MCPR is overpredicted by less than 3% and underpredicted by less than 20%. Hensch-Levy underpredicts MCHFR by 13 to 55%.

Figure IV-27 compares critical power data and prediction versus inlet coolant subcooling. All the CHF predictions fall below the data. The CISE-4 predictions are within 7% of the critical power data for the cases shown. Figure IV-28 compares

Figure IV-25  
GE 9-Rod CHF Tests  
Geometry and Test Conditions (Ref. 27)

Number of Rods	9
Rod Diameter	.570 inch
Radius of Corner Subchannel	.420 inch
Rod Rod Clearance	.168 inch
Rod Wall Clearance	.135 inch
Hydraulic Diameter	.474 inch
Heated Length	72 inch
Pressure	800 to 1000 psia
Average Bundle Mass Flow	0.5 to 1.25 $\text{mlb}_m/\text{hr ft}^2$
Inlet Subcooling	35 to 200 BTU/lb

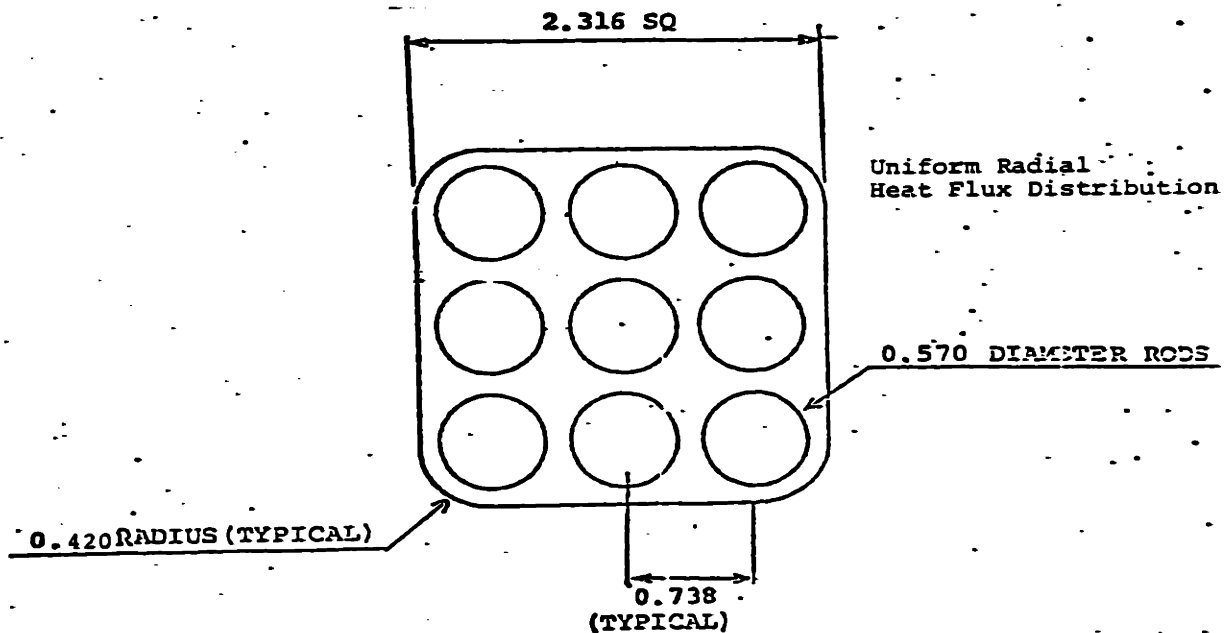






Table IV-4  
 9-Rod GE-CHF Experiments Analyzed and Single Channel Analysis Predictions

Test Channel Case	Test Case No.	p (psia)	Moss Flux (Mlb/hr ft <sup>2</sup> )	Inlet Subcooling (BTU/lb)	q" (MBTU/hr ft <sup>2</sup> )	COBRA Single Channel Analysis Predictions	
						CISE-4 MCPR	Hench-Levy MCHFR
3	266	1005.	1.008	7.1	0.510	0.9320	0.6017
	268	1015.	1.004	96.5	0.633	0.9950	0.6665
	270	1000.	1.000	191.8	0.785	0.9936	0.7662
	279	1000.	0.500	70.7	0.474	0.8634	0.4622
	286	997.	0.249	42.0	0.289	0.8028	0.4528
	296	1000.	1.248	12.8	0.522	1.0198	0.8130
4	301	1019.	1.051	29.4	.518	1.0013	0.6849
	302	1007.	1.075	54.6	0.560	1.0074	0.7685
	303	1018.	1.134	110.2	0.665	1.0289	0.8659
	320	1027.	0.306	197.4	0.410	0.9170	0.5098

Note: Ranges of data base for CISE-4 and Hench-Levy correlations are given in Table II-2 of Appendix H.

**Table IV-5**  
**Comparison of MCPR and MCHFR Predictions**  
**Using Single Channel and Subchannel Analysis**

Test Case No.	Analysis Method	CISE-4 MCPR	Hench-Levy MCHFR
266	Single channel	0.9320	0.6017
	Subchannel	0.7657	0.5955
268	Single channel	0.9950	0.6665
	Subchannel	0.9126	0.6241

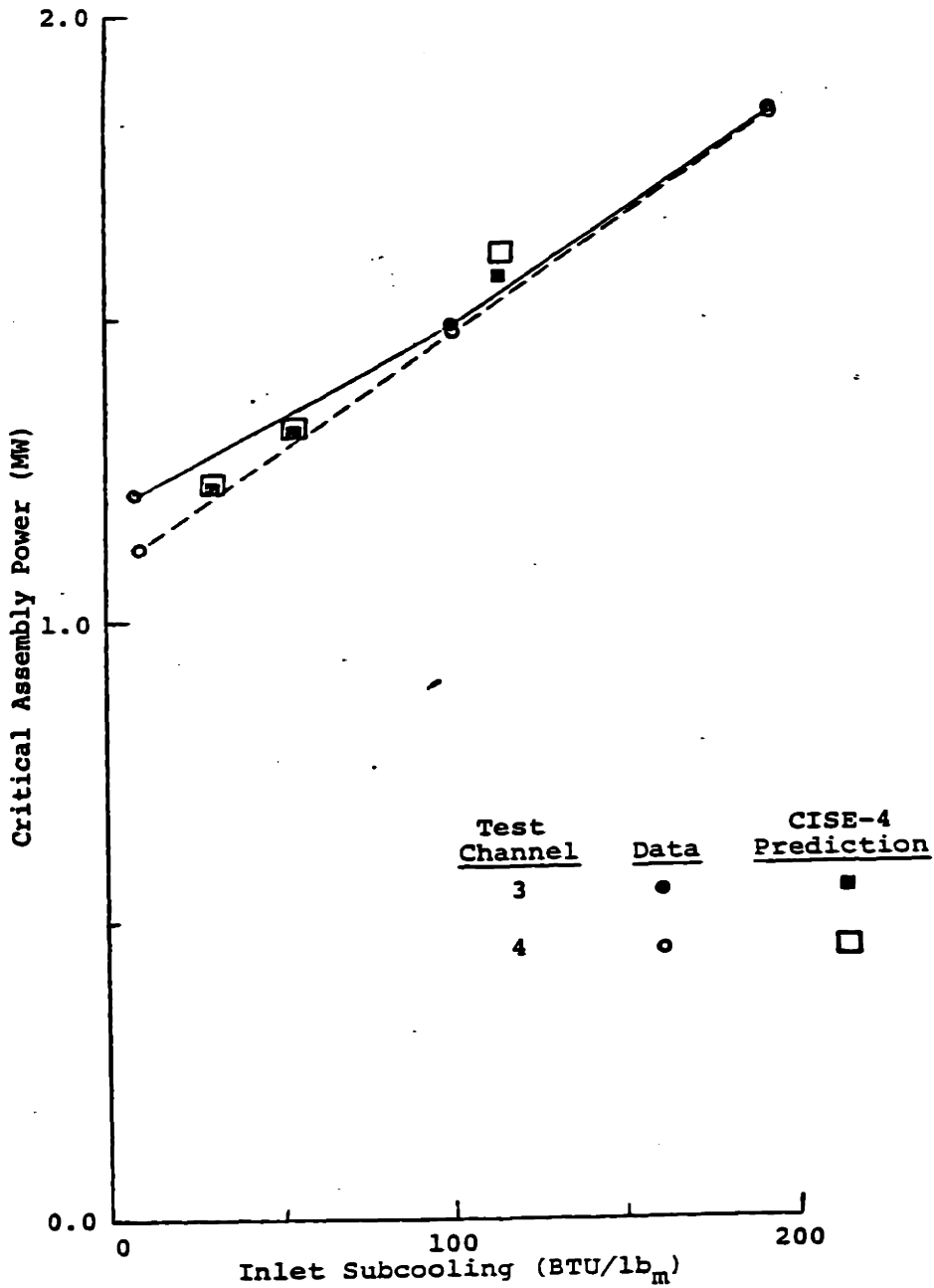


Figure IV-27

GE 9-Rod CHF Tests

Critical Assembly Power vs. Inlet Subcooling

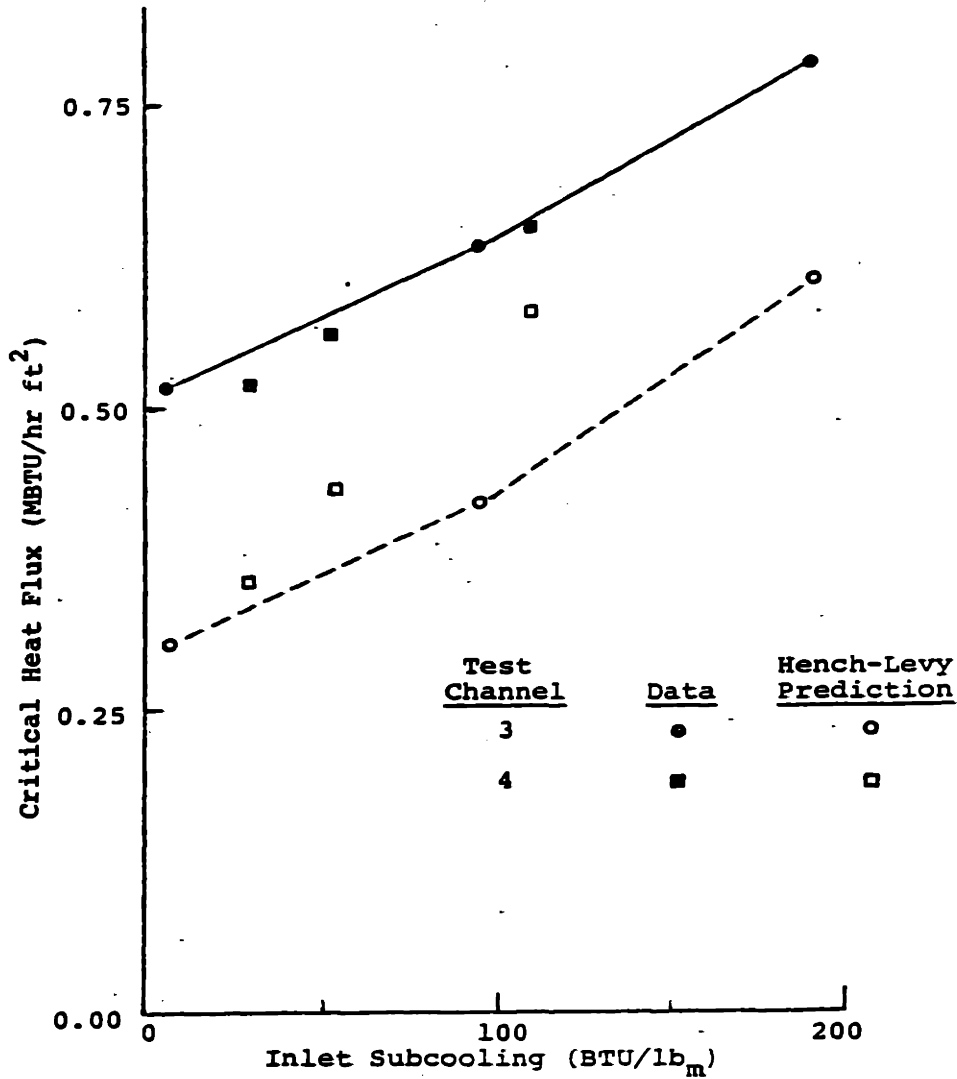


Figure IV-28  
GE 9-Rod CHF Tests  
Critical Heat Flux vs. Inlet Subcooling

CHF data and Hensch-Levy predictions. The Hensch-Levy predictions are within 40% of the CHF data for the cases shown.

CISE 4 closely predicts critical power ratio. The less conservatism of the CISE-4 predictions in comparison to those of Hensch-Levy can be understood in terms of the intended purpose of each. CISE-4 was developed to predict critical heat flux in accordance to experimental data. Hensch-Levy was developed for design purposes rather than accurate CHF prediction; thus, it tends to underpredict critical heat flux.

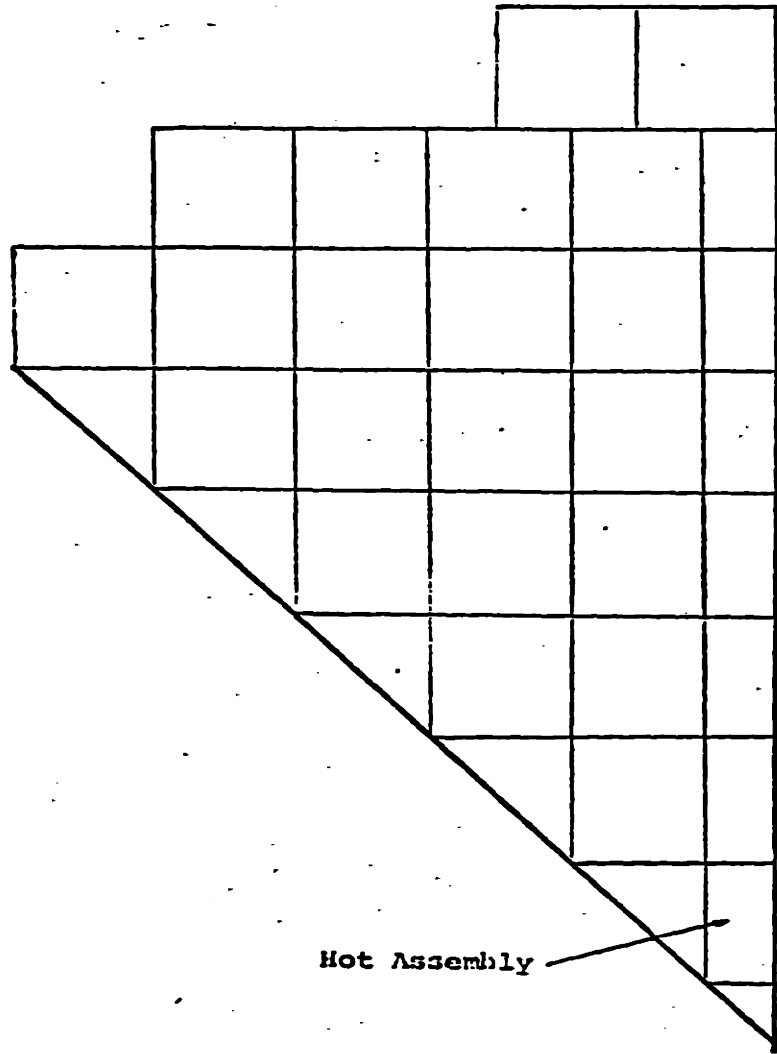
##### 5. Testing of One of the Two New Transverse Momentum Options for Single-Pass Method

One of the two new transverse momentum options was tested by comparing predictions obtained using this option with predictions obtained using the "standard" option. The new option tested was the "Weisman" option. The test case used was a single-pass analysis of a PWR core.

The 1/8 section of the PWR core shown in Figure IV-29 was modeled using the layout shown in Figure IV-30. Geometric and thermal-hydraulic data used is given in Appendix J. Rod 12 was the hot rod and channel 9 was the "hot" subchannel where MDNBR for each axial level occurs. Figure IV-31 shows the top-peaked axial heat flux profile used to make predictions. Analysis results are shown in Figures IV-32 through IV-36.

Predictions for the hot subchannel (channel 9) were nearly the same for the two analysis approaches. Figure IV-32 shows enthalpy as a function of axial position in the hot subchannel. The predictions of the two approaches are essentially the same. Predictions of net crossflow out of the hot subchannel are also nearly the same, as shown in Figure IV-33. MDNBR predictions of the two approaches lie on top of one another, as shown in Figure IV-34.

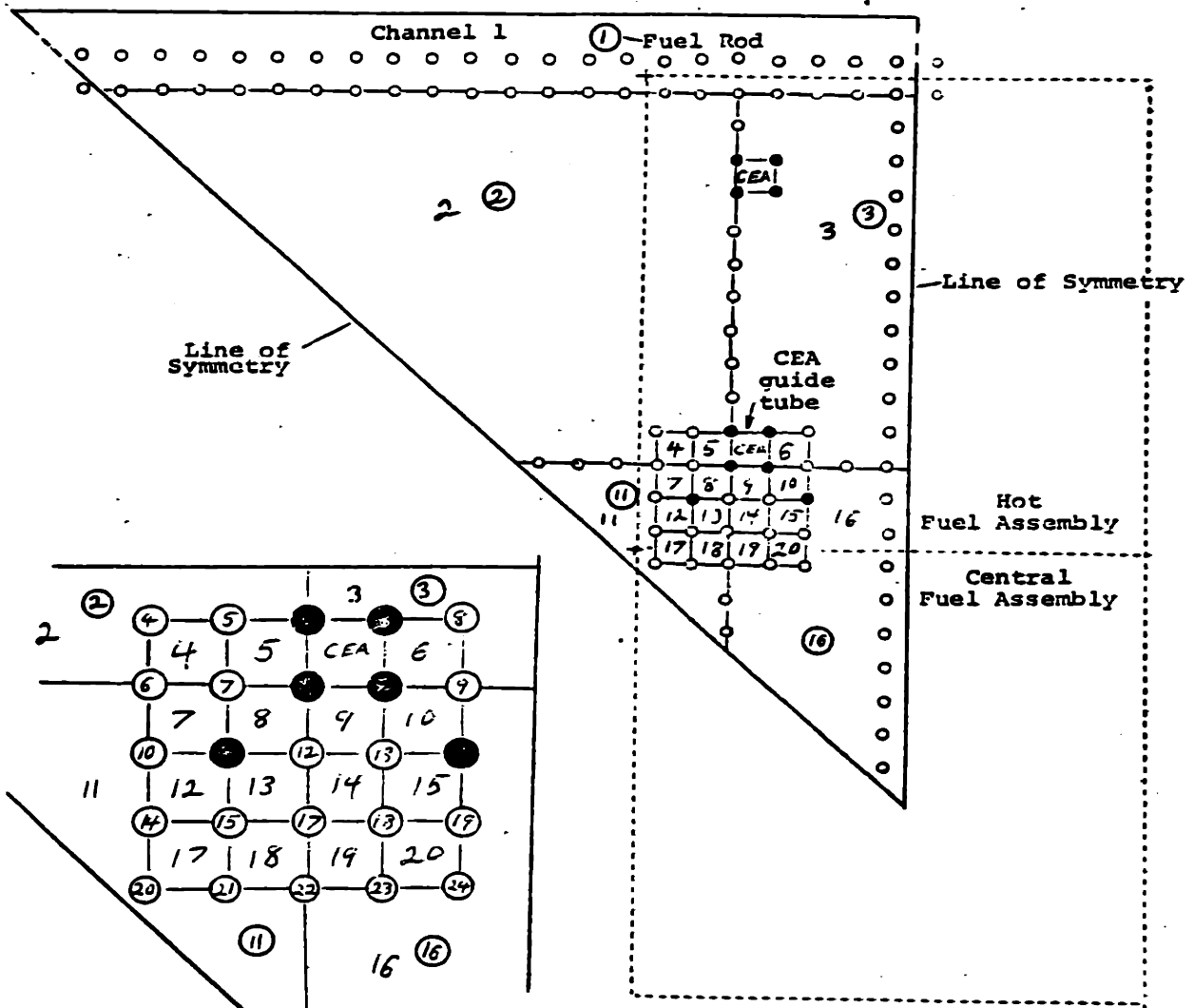
The axial crossflow distributions showed some change for gaps connecting the fine mesh region to the coarse mesh region. Figure IV-35 shows the axial crossflow predictions of the two analysis methods for the gap connecting channels 7 and 11. The profile shapes are different; however, the net crossflow, represented by the area under each curve, appears to be similar.



Hot Assembly

Figure IV-29

1/8 Section of PWR Core Used for Test Case



Note: Rod 12 is the hot rod and channel 9 is the hot subchannel

Figure IV-30

Layout Used for 1/8 Core Single-Pass Case



Note: Ref. Appendix J for other thermal hydraulic data used.

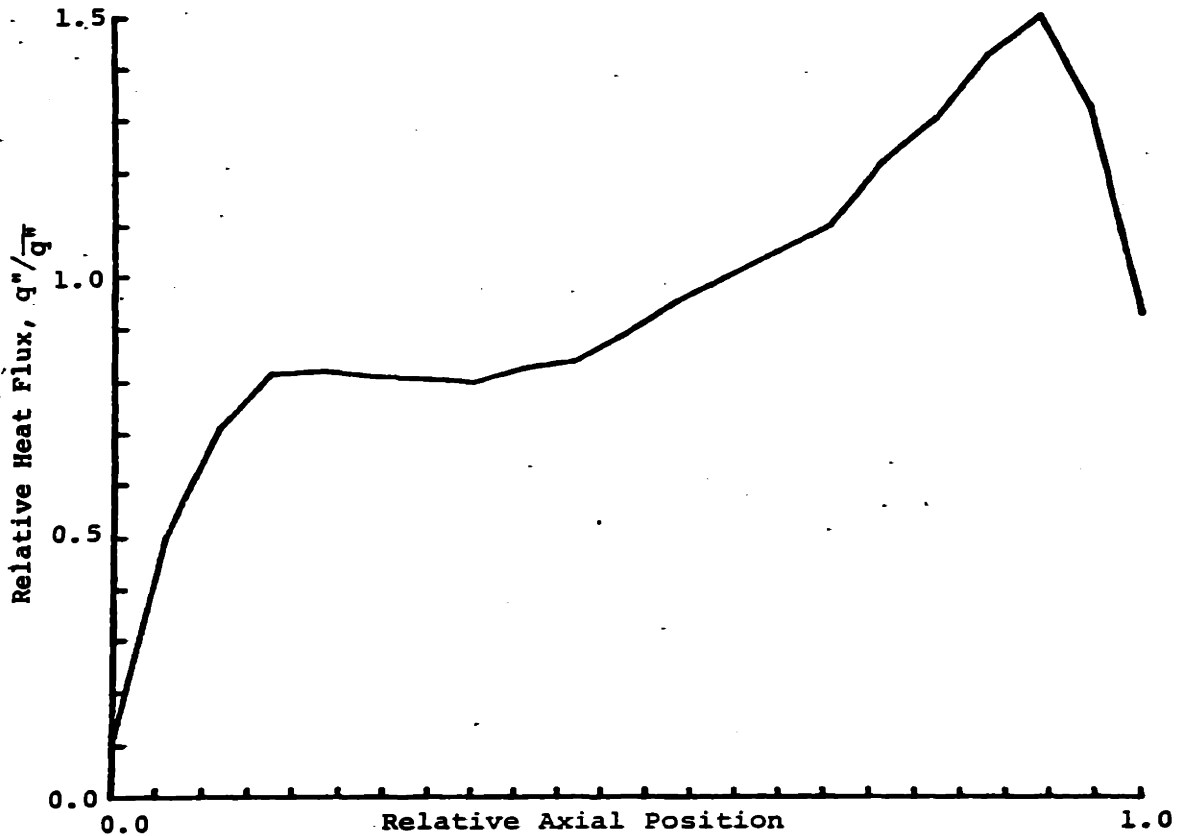


Figure IV-31

Top-Peaked Axial Heat Flux Profile  
Used for 1/8 Core Single-Pass Analysis

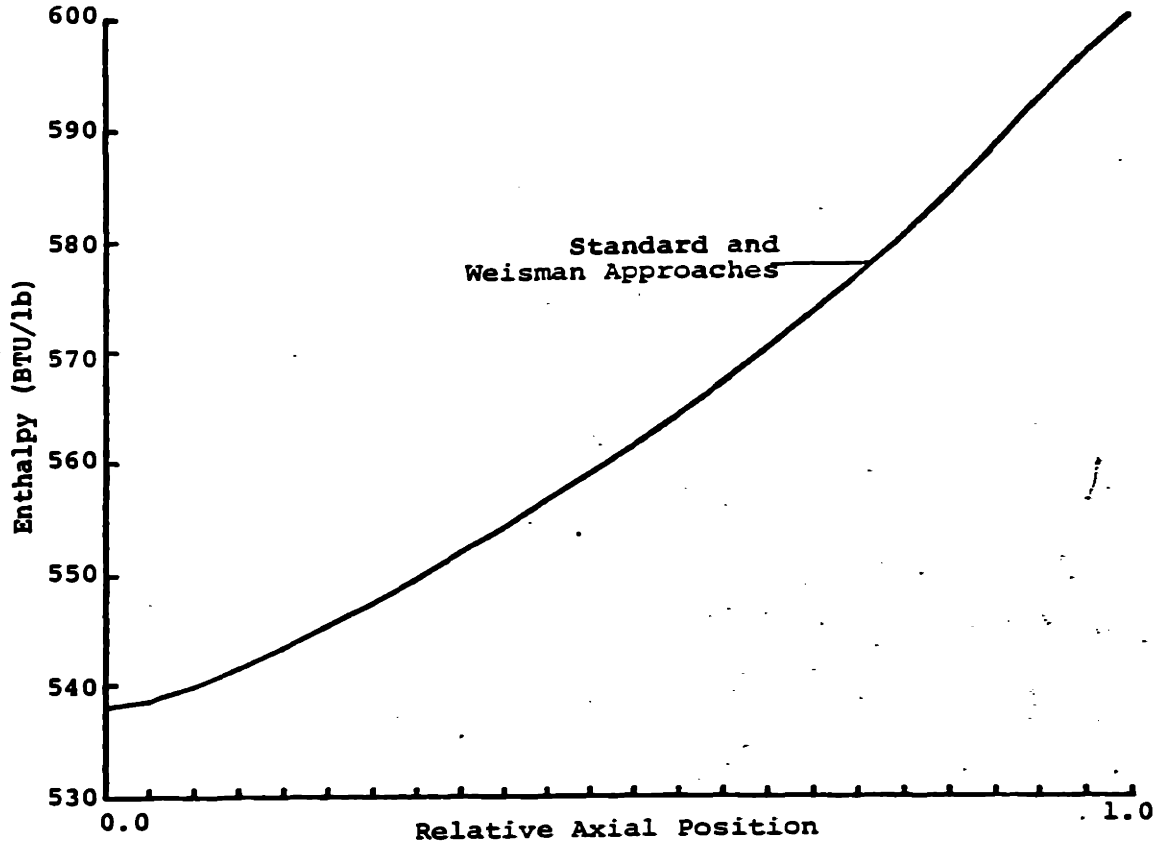


Figure IV-32  
1/8 Core Single-Pass Analysis Case  
Enthalpy in Channel 9 (Hot Subchannel)  
vs. Relative Axial Position

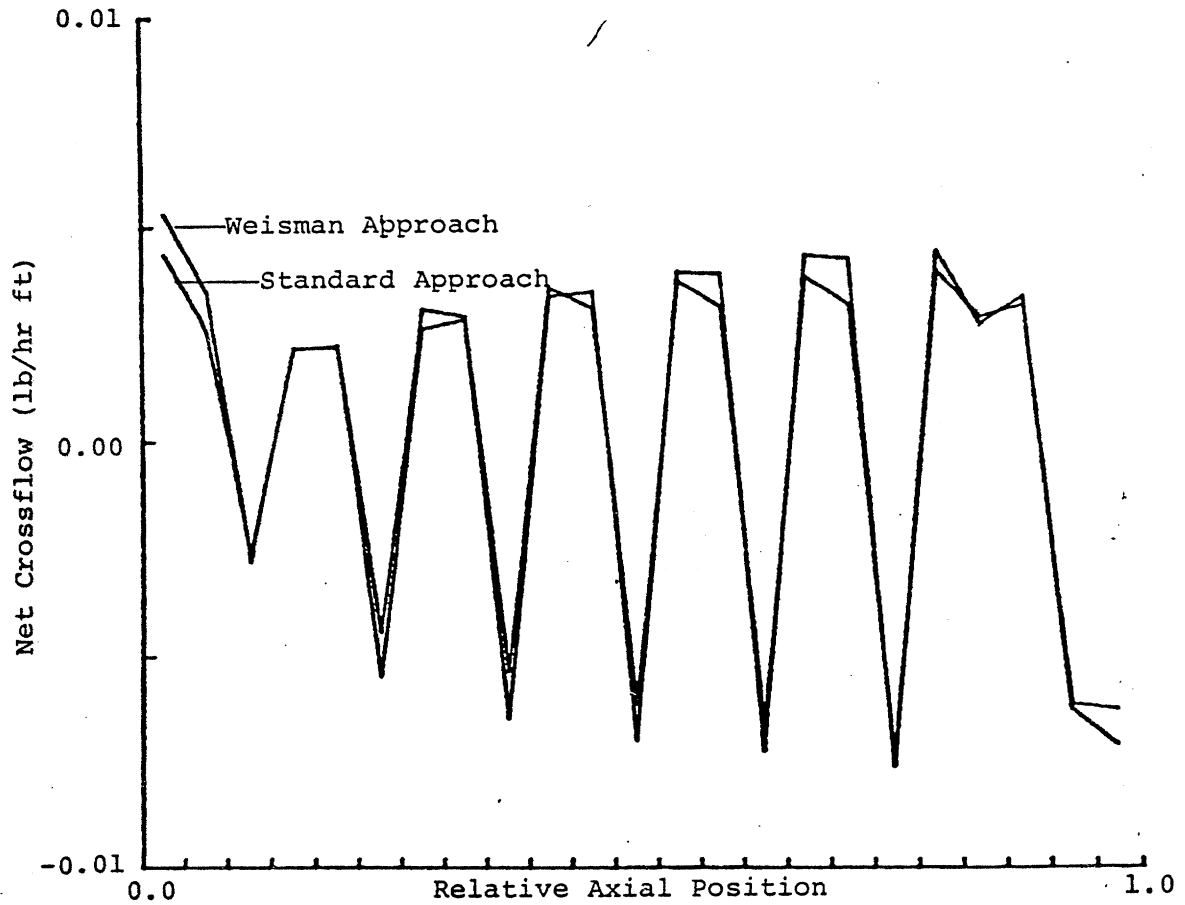


Figure IV-33  
1/8 Core Single-Pass Analysis Case  
Net Crossflow Out of Channel 9 (Hot Subchannel)  
vs. Relative Axial Position

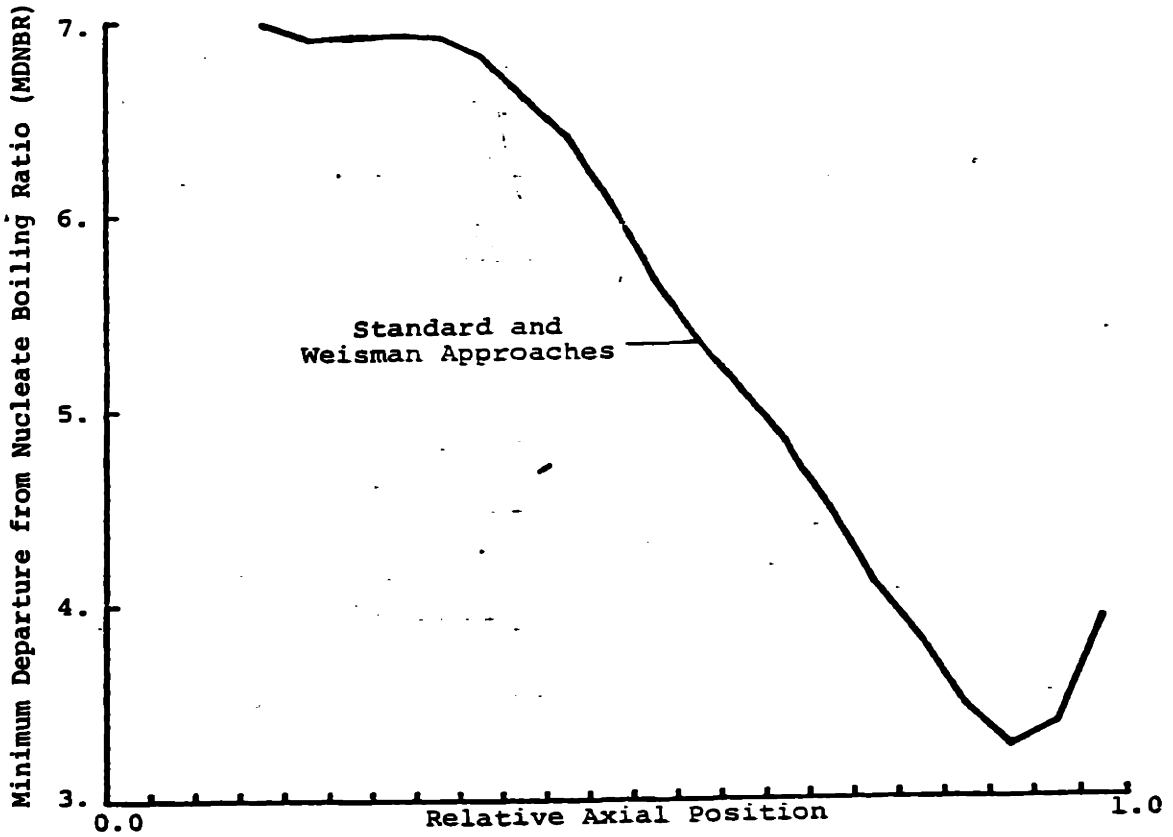


Figure IV-34  
1/8 Core Single-Pass Analysis Case  
MDNBR vs. Axial Position

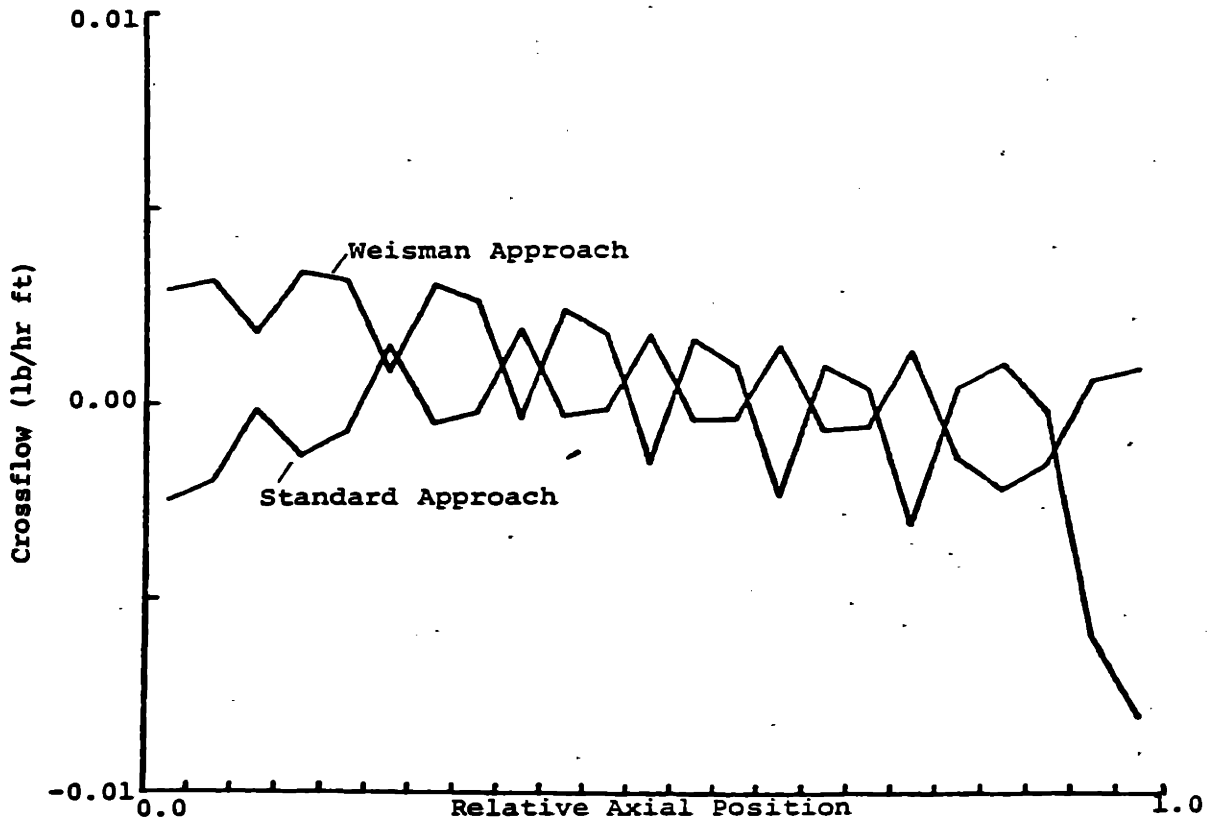


Figure IV-35

1/8 Core Single-Pass Analysis Case

Crossflow from Channel 7 to 11 vs. Relative Axial Position

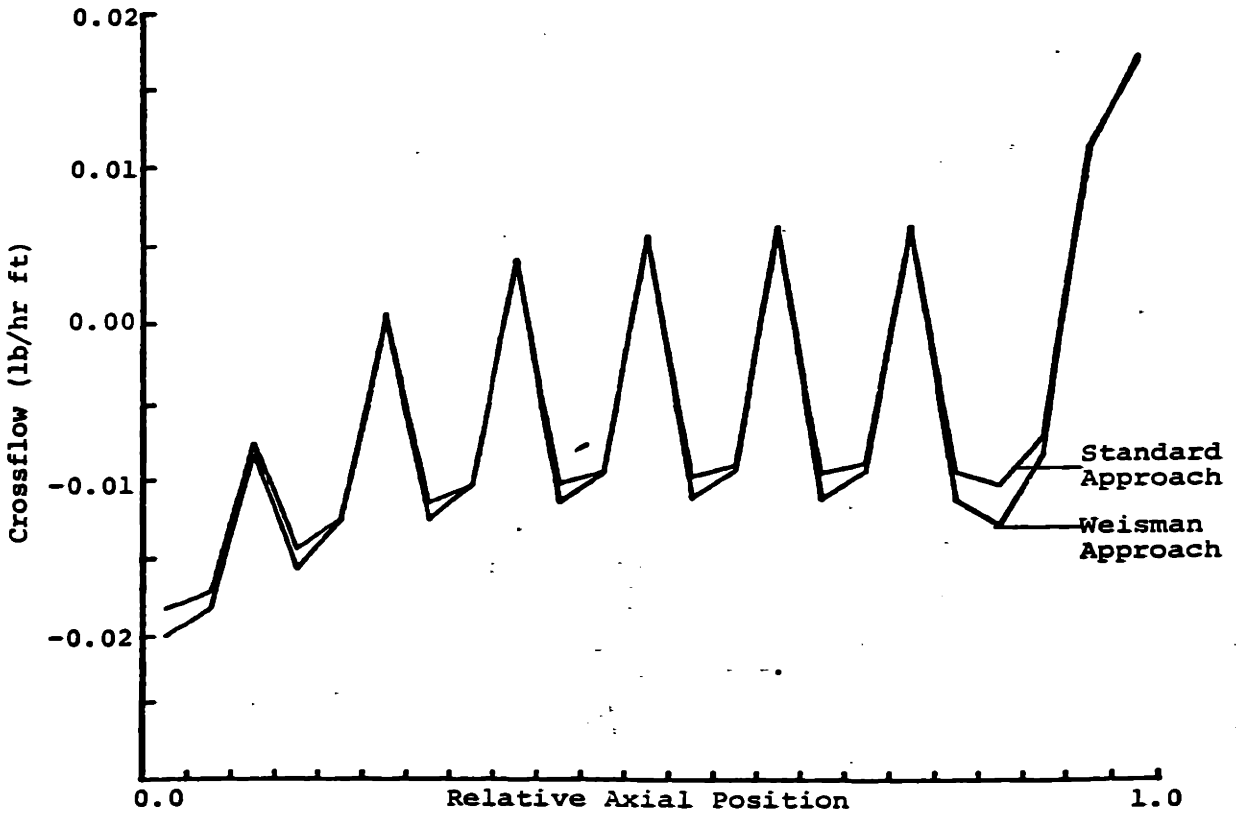


Figure IV-36

1/8 Core Single-Pass Analysis Case

Crossflow from Channel 7 to 8 vs. Relative Axial Position

The difference in axial crossflow profiles is much less for the gap connecting fine mesh regions 7 and 8, as Figure IV-36 shows.

Thus, for the case analyzed, the standard and Weisman approaches\* give nearly the same results for crossflow and enthalpy distribution; and MDNBR predictions are the same. It is possible that the two approaches might not give the same results for more off normal conditions such as a case involving flow blockage, for example. In this case, however, it may be questionable whether COBRA-IIIC/MIT should be used for the analysis.

### C. Application to Transient Test Cases

COBRA-IIIC/MIT has been tested by application to transient test cases. A PWR loss of flow transient and BWR turbine trip transient were analyzed using both new and old modeling options.

#### 1. PWR Transient Test Case - Loss of Flow Transient

##### a. Description of Loss of Flow Transient

The PWR transient test case is a postulated loss of coolant accident for the Maine Yankee reactor (Ref. 28 & 29). In this accident, all three primary coolant pumps lose electrical power during full power operation. Flow coasts down, causing a low flow reactor trip signal. Control element assemblies (CEAs) are assumed to fall into the core three seconds after initiation of flow coastdown. The minimum value of DNBR occurs between three and four seconds after initiation of flow coastdown.

##### b. Description of Modeling

The loss of flow transient was analyzed using single-pass COBRA-IIIC/MIT analysis. A 1/8 section of the Maine Yankee core was modeled using the layout shown schematically in Figure IV-29. Rods 5 and 15 are the hot rods. MDNBR is predicted to occur on either rod 5 or 15 during the transient.

---

\*The "standard" approach used the old transverse momentum option ( $f_{sl}$  and  $f_{slk}$  equal to unity). The "Weisman approach used coupling factors as defined by Eqns. (III-7) and (III-8).

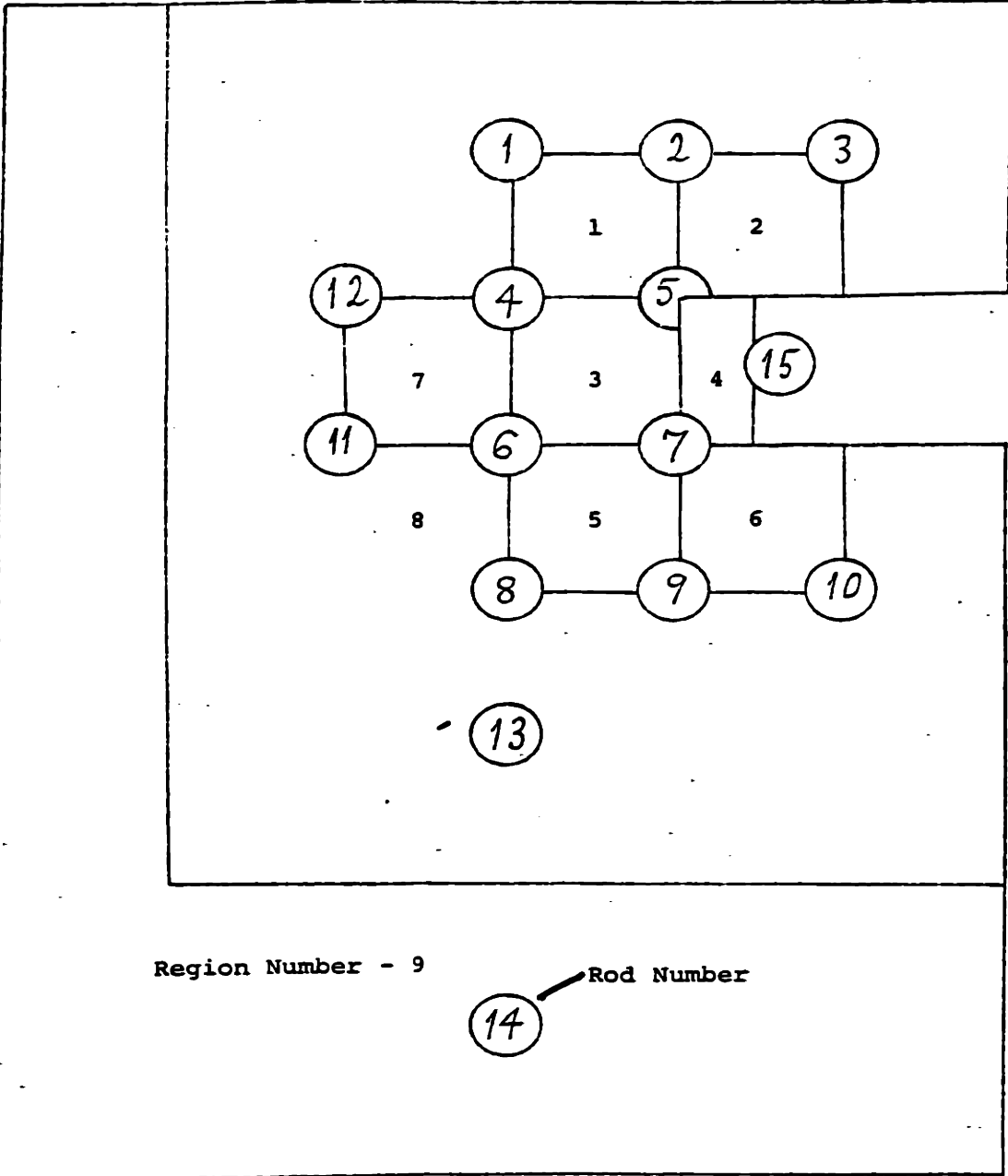


Figure IV-37  
Schematic of Layout Used for Loss of Flow Analysis



Fine radial nodalization (subchannel size coolant and nodes) is used in the vicinity of rods 5 and 15. Coarser radial nodalization is used outside the fine mesh region. Regions one to eight represent one 14x14 fuel rod assembly. Region nine represents the remaining assemblies in the 1/8 section of core.

Four COBRA-IIIC/MIT analyses were made using various modeling options as indicated in Table IV-6. Transient forcing functions used by the analyses are shown in Figure IV-30. The core inlet flow forcing function is based on plant data. The heat flux and power level forcing functions are based on predictions of the CHIC-KIN code (Ref. 28). Heat flux was used as a forcing function for analysis cases which did not use a fuel rod model. Power level was used as a forcing function for analysis cases which used a fuel rod model. The loss of flow transient was analyzed for five seconds using a time step size of 0.25 sec. for all cases. Channels were divided axially into twenty nodes. Predictions were printed once every two time steps. COBRA-IIIC/MIT input for the loss of flow transient is scribed in Appendix K.

### c. Analysis of Results

The predictions of the four analysis cases were similar. MDNBR predictions were nearly the same. The largest dissimilarities in predictions were due to differences between the old and new rod-to-coolant heat transfer models. Clad surface temperature predictions of the two heat transfer models showed differences.

Since DNBR is usually the limiting parameter for a loss of flow accident, comparison of analysis case predictions will begin with this parameter. Predicted MDNBR is shown as a function of time for the four analysis cases in Figure IV-31. The predictions are close. The maximum difference between MDNBR predictions is less than 5%. The MDNBR predictions show the same trend. MDNBR decreases as flow coasts down and power is constant in the time range from 0. to 3. seconds. Reactor shutdown initiates at three sec. while flow coastdown continues. MDNBR predictions reach their minimum values near 3.5 sec. and increase

Table IV-6Models Used for Loss of Flow Analysis Cases

<u>Analysis Case Number</u>	<u>Fuel Rod Model</u>	<u>Fuel &amp; Clad Material Properties</u>	<u>Heat Transfer Model</u>
1	none	none	none
2	old	constant	old
3	new	temperature-dependent	new
4	old	constant	new

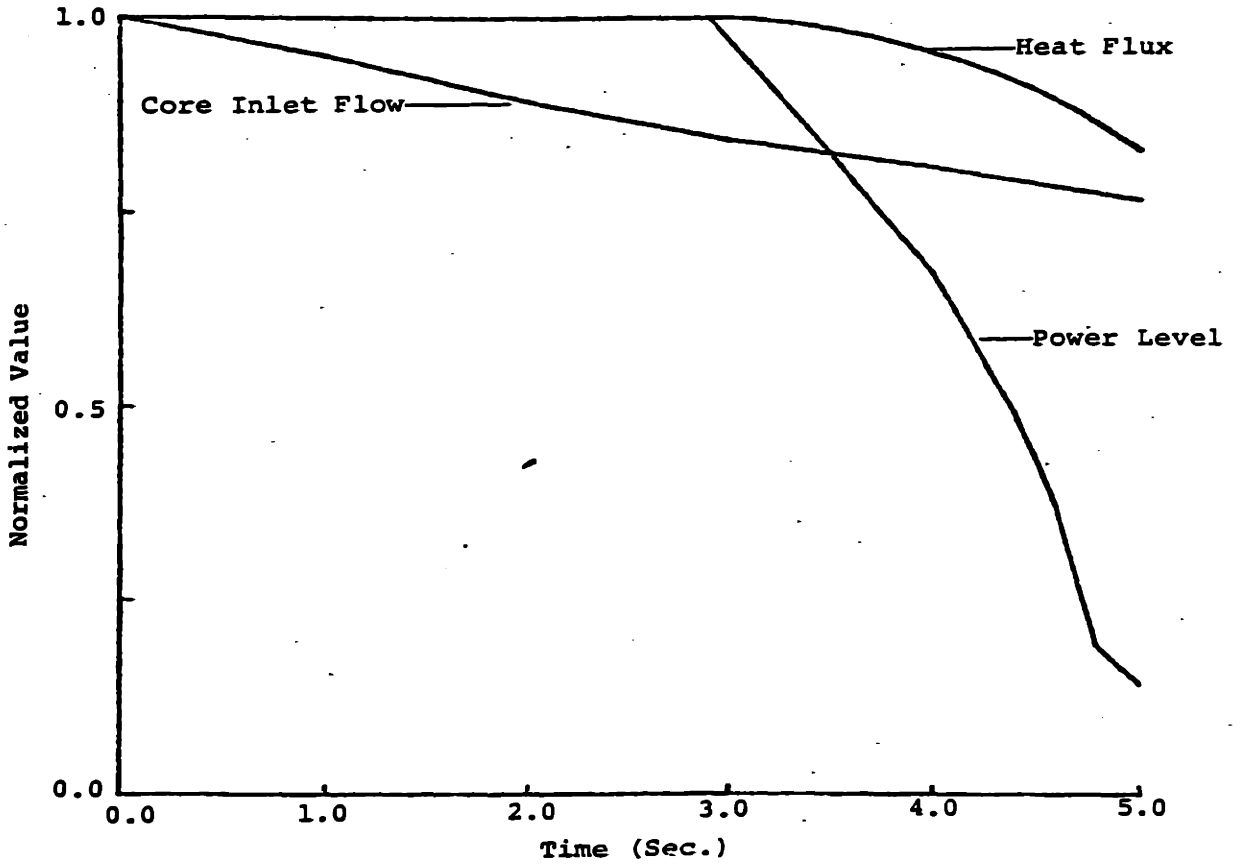


Figure IV-38  
Transient Forcing Functions  
PWR Loss of Flow Transient Test Case

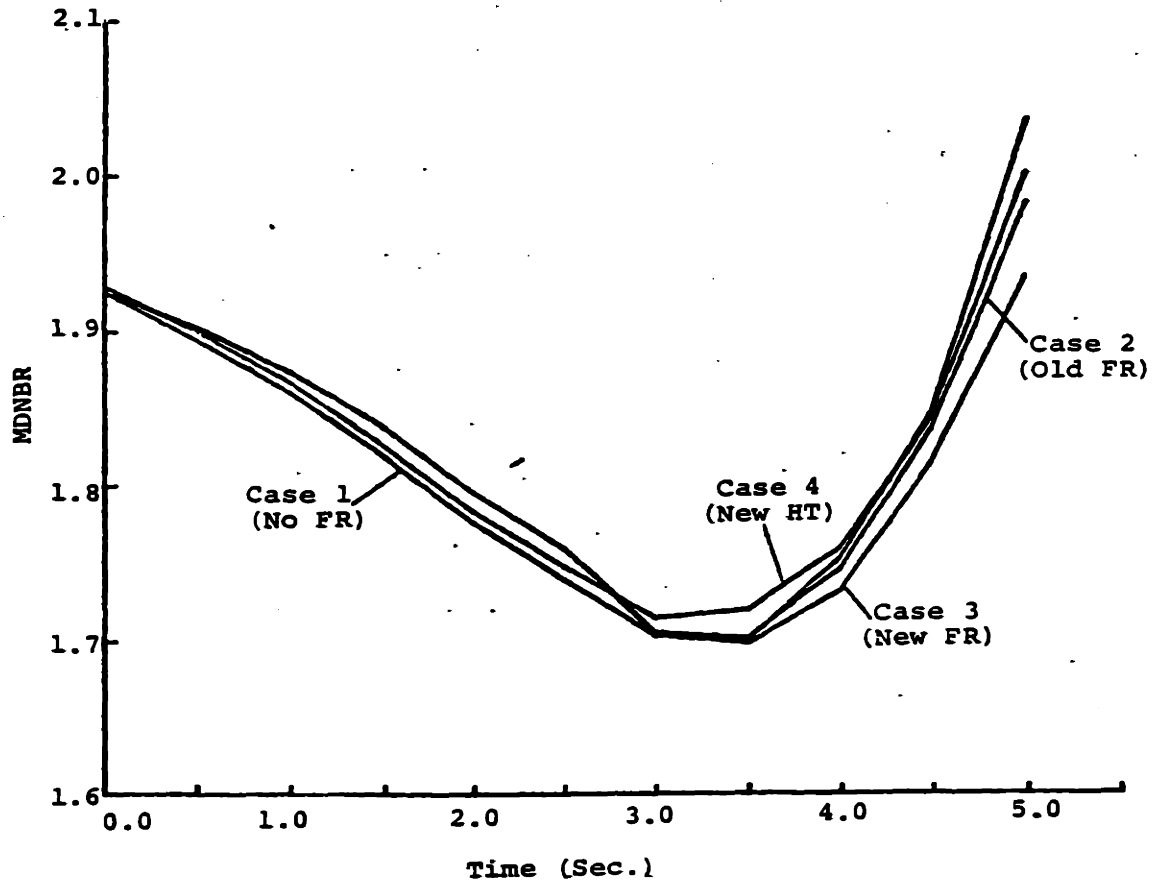


Figure IV-39  
Predicted MDNBR vs. Time  
PWR Loss of Flow Transient Test Case

as time continues to 5.0 seconds. The minimum values of MDNBR predicted during the loss of flow transient are within the 1% of each other.

DNBR predictions depend largely on heat flux predictions. The close agreement between MDNBR predictions was due to agreement between heat flux predictions of the analysis cases. Maximum predicted heat flux is shown as a function of time in Figure IV-32. The predicted maximum heat flux is nearly constant up to 3.0 seconds. Maximum heat flux falls in the time range from three to five seconds. The predicted maximum heat fluxes are within 5% of each other during the transient. The closeness of maximum heat flux predictions indicate a general similarity of heat flux predictions.

Heat flux predictions will be further compared by considering rod 51 axial heat flux problems. Rod 15 is selected for comparison because it is predicted to be the location of MDNBR for a large portion of the transient. In Analysis Cases 1, 2, and 3 predictions, the location of MDNBR shifts temporarily from rod 15 (facing channel 4) to rod 5 (facing channel 3), due to voiding in channel 3. Almost no voiding occurs in channel 4. Analysis Case 4 predicts that MDNBR is located on rod 15 through the transient. Figure IV-33 shows exit void fraction of channel 3 as a function of time. Exit void fraction peaks at 3.5 seconds. The void fraction predictions of Analysis Case 4 are less than those of other cases. This may account for the fact that the location of MDNBR remains on rod 15 throughout the transient in the predictions of Analysis Case 4.

Heat flux profiles of rod 15 are compared in Figure IV-34 and IV-35. Figure IV-34 shows axial heat flux profiles at 0.0 and 2.5 seconds. The profiles of all cases are exactly the same at 0.0 seconds and nearly the same at 2.5 seconds. Figure IV-35 shows axial heat flux profiles at 0.0 and 5.0 seconds. The profiles at 5.0 seconds are close. A comparison of Figures IV-34 and IV-35 will show a larger change in heat flux between 2.5 and 5 seconds.

Although heat flux predictions of the analysis cases were close, differences between the old and new rod-to-coolant heat transfer models caused differences in clad surfact temperature

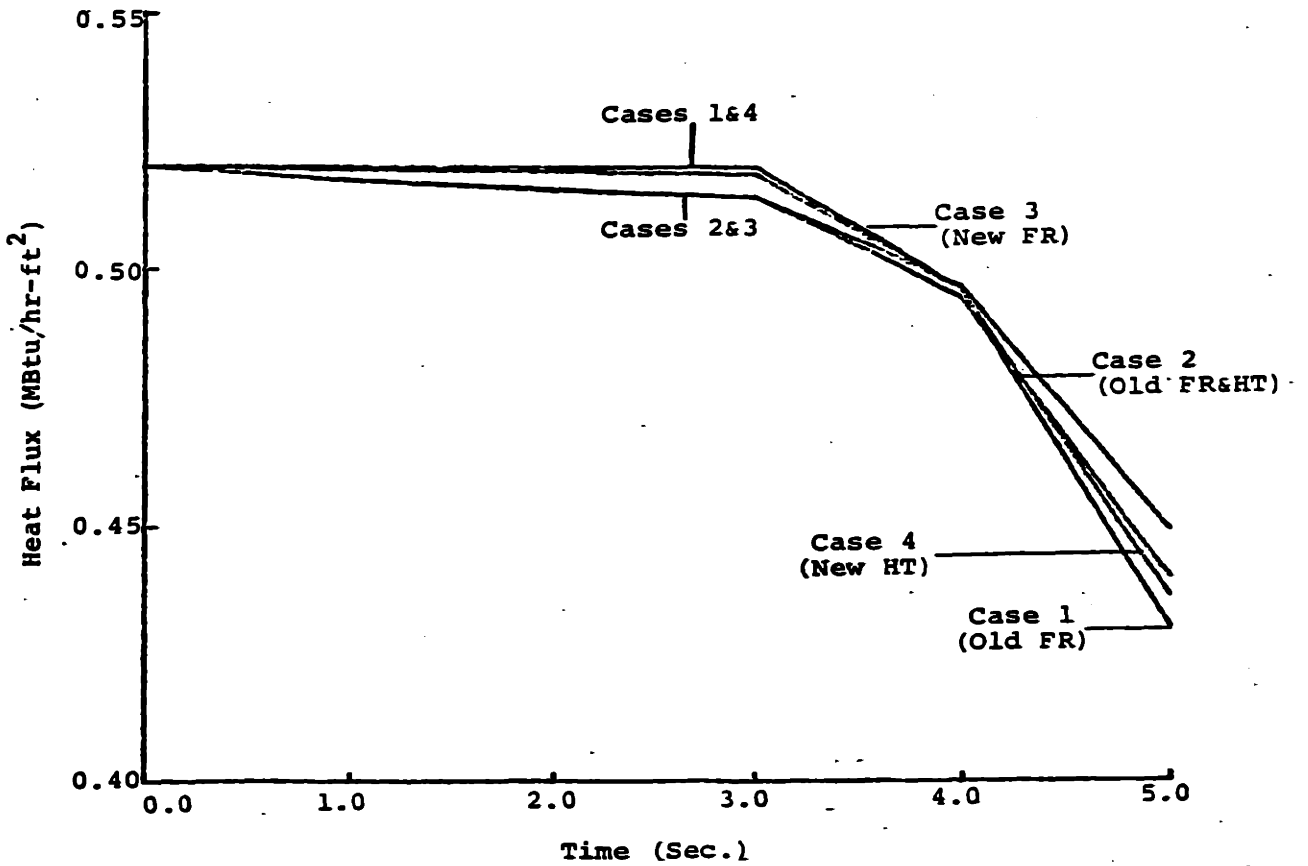


Figure IV-40

Maximum Heat Flux vs. Time  
 PWR Loss of Flow Transient Test Case

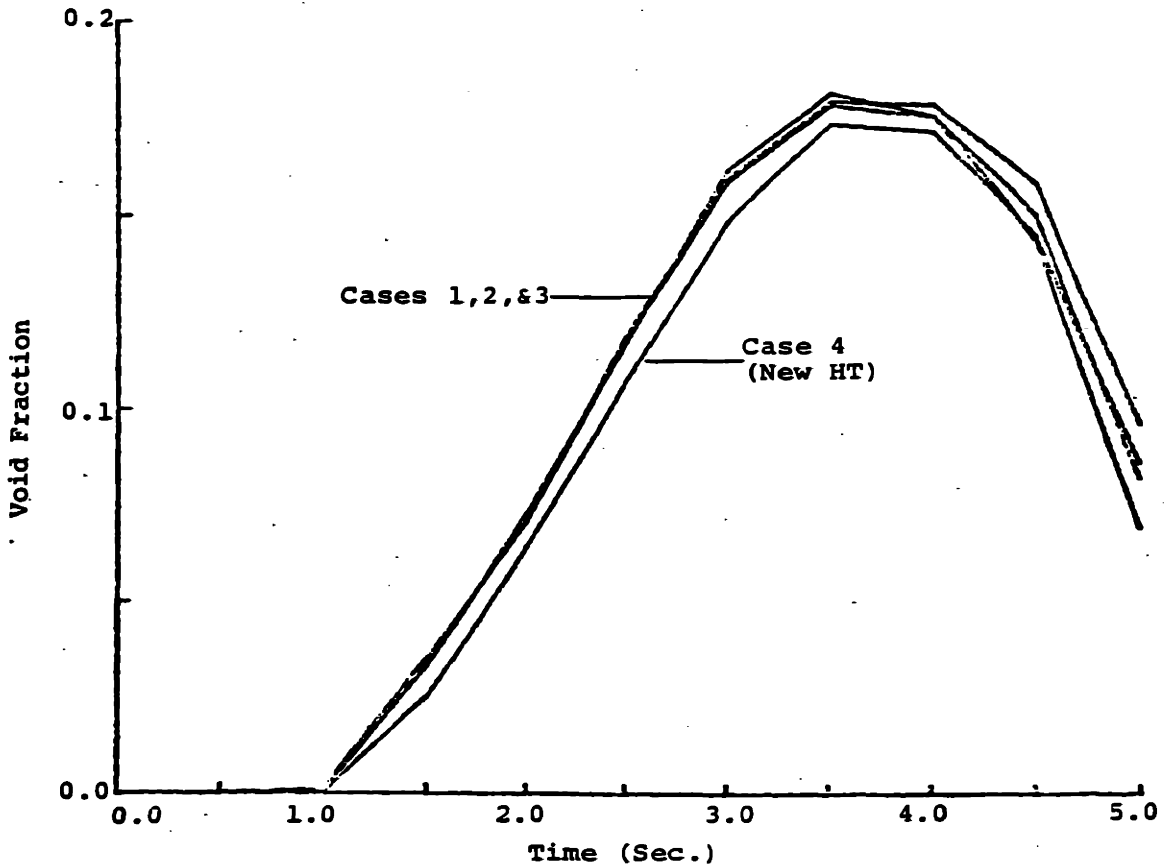


Figure IV-41

Exit Void Fraction vs. Time  
Channel 3  
PWR Loss of Flow Transient Test Case

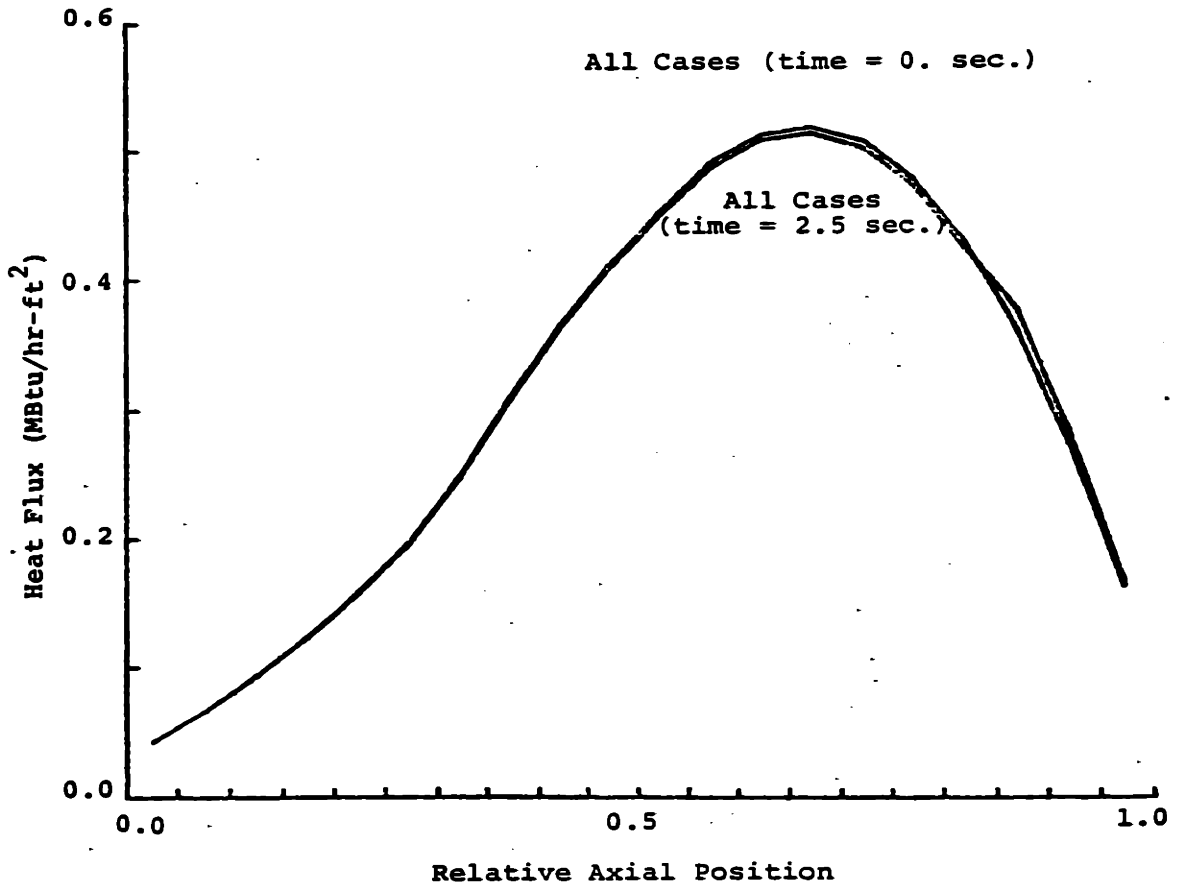


Figure IV-42

Axial Heat Flux Profile

Rod 15

PWR Loss of Flow Transient Test Case



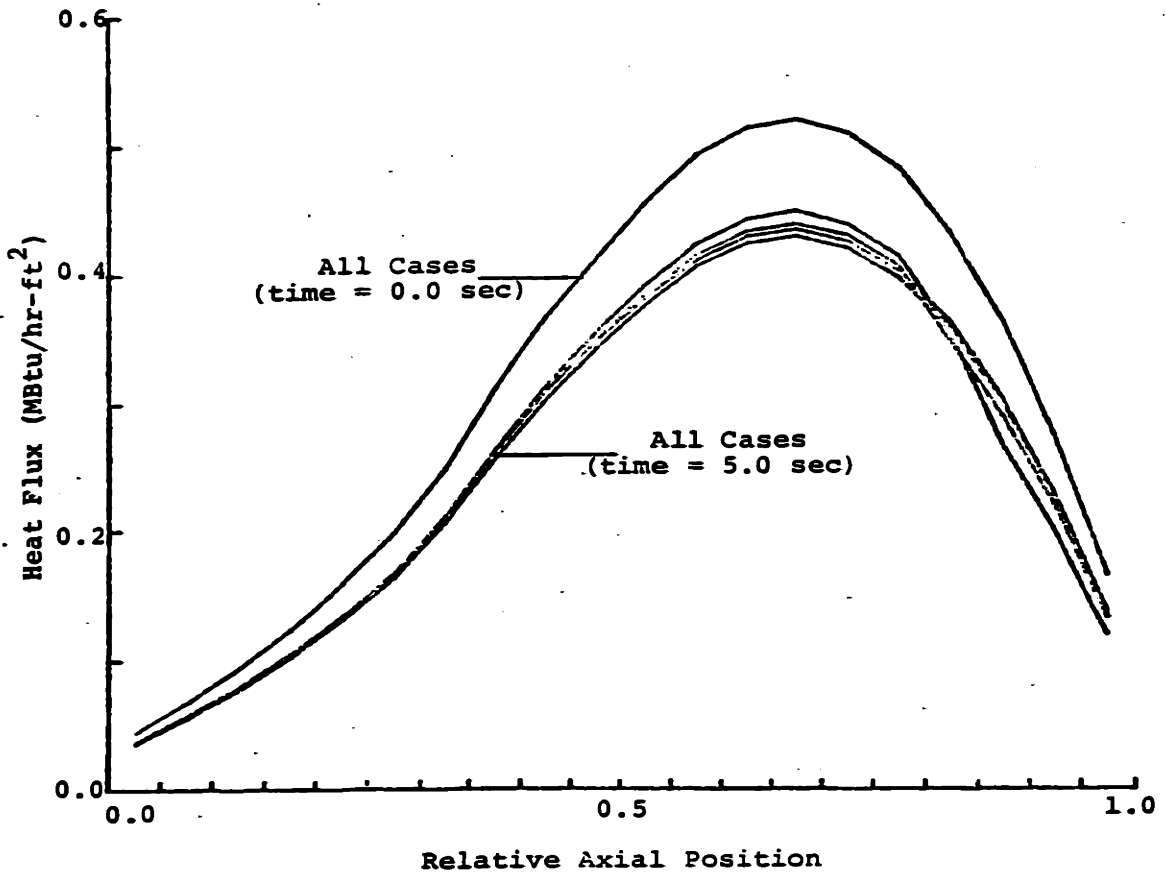


Figure IV-43

Axial Heat Flux Profile  
Rod 15  
PWR Loss of Flow Transient Test Case

predictions. Figures IV-36, IV-37 and IV-38 contain axial clad surface temperature profiles for rod 15. Clad temperature profiles at 0.0 seconds are shown in Figure IV-36. In the top half of rod 15, clad temperature predictions of Analysis Cases 2 and 3, which use the old heat transfer model rise well above the saturation temperature. Clad temperature predictions of case 4, which uses the new heat transfer model do not rise as far above the saturation temperature. Higher wall temperature represents slightly larger stored heat. Differences in the heat transfer logic contained in the two heat transfer models is the major cause of the large differences in clad temperature predictions.

The old heat transfer model switches from forced convection to nucleate boiling heat transfer when void fraction is greater than 0. The new heat transfer model switches from forced convection to nucleate boiling heat transfer when wall temperature is greater than saturation temperature. Figure IV-37 shows Analysis Case 2 clad temperature profiles at 0., 2.5, and 5 seconds. The profile has an irregular shape at 2.5 seconds. Increased void fraction when time is near 2.5 seconds causes a sudden change in rod-to-coolant heat transfer since Analysis Case 1 uses the old heat transfer model. The sudden change in heat transfer produces the irregular clad temperature profile. Similar clad temperature behavior was seen in Analysis Case 3 prediction which also used the old heat model. Figure IV-38 shows Analysis Case 4 axial clad temperature profile predictions. These predictions of the new heat transfer model show only small changes in time and none of the discontinuities apparent in the predictions of the old heat transfer model.

#### d. Summary

The loss of flow transient was analyzed by four analysis cases which all used the one-pass method. One analysis case did not use a fuel rod model. The other three cases used old and new fuel rod and heat transfer models. MDNBR and heat flux predictions of the analysis cases were close. Clad temperature predictions differed according to the rod-to-coolant heat transfer model used.

## 2. BWR Transient Test Case - Turbine Trip Without Bypass

### a. Description of Turbine Trip Transient

The BWR transient test case is a postulated turbine trip without bypass transient for the Shoreham reactor. Failure of the turbine bypass system to operate would result in an increase in system pressure and cause the power level to reach 231% of the initial steady state value. The power level increase is caused by void reactivity feedback. Increasing pressure decreases the amount of voids in the core. Power level increases due to void reactivity feedback. The transient forcing functions for power level, system reference pressure and core inlet flow are shown in Figure IV-39.

### b. Description of Modeling

The turbine trip transient was analyzed using two channels to represent the central hot and central average assemblies of the Shoreham Nuclear Power Station Unit One (SNPS-1) reactor. Data from the SNPS-1 FSAR (Ref. 30) was used in the analysis. Four COBRA-IIIC/MIT analyses were made using fuel rod and rod-to-coolant heat transfer model options as listed in the Table IV-7. Transient forcing functions used by the analyses are contained in Figure IV-39. The transient was analyzed for 2.5 seconds using 0.05 second time steps. The two channels were divided axially into twenty nodes. Predictions were printed once every five time steps. COBRA-IIIC/MIT input for the turbine trip transient is described in Appendix L.

### c. Analysis Case Predictions

Examination of analysis case predictions will begin with MCPR and MCHFR predictions. MCPR and MCHFR predictions are useful for comparison of modeling option predictions. However, the applicability of the CPR and CHFR correlations to transient conditions and assemblies represented by single channels is uncertain. The CISE-4 MCPR correlation was developed for rod-centered subchannels. Although the MCPR and MCHFR predictions may be unreliable, they are based on calculated predictions of COBRA-IIIC/MIT models and can indicate differences in these predictions.

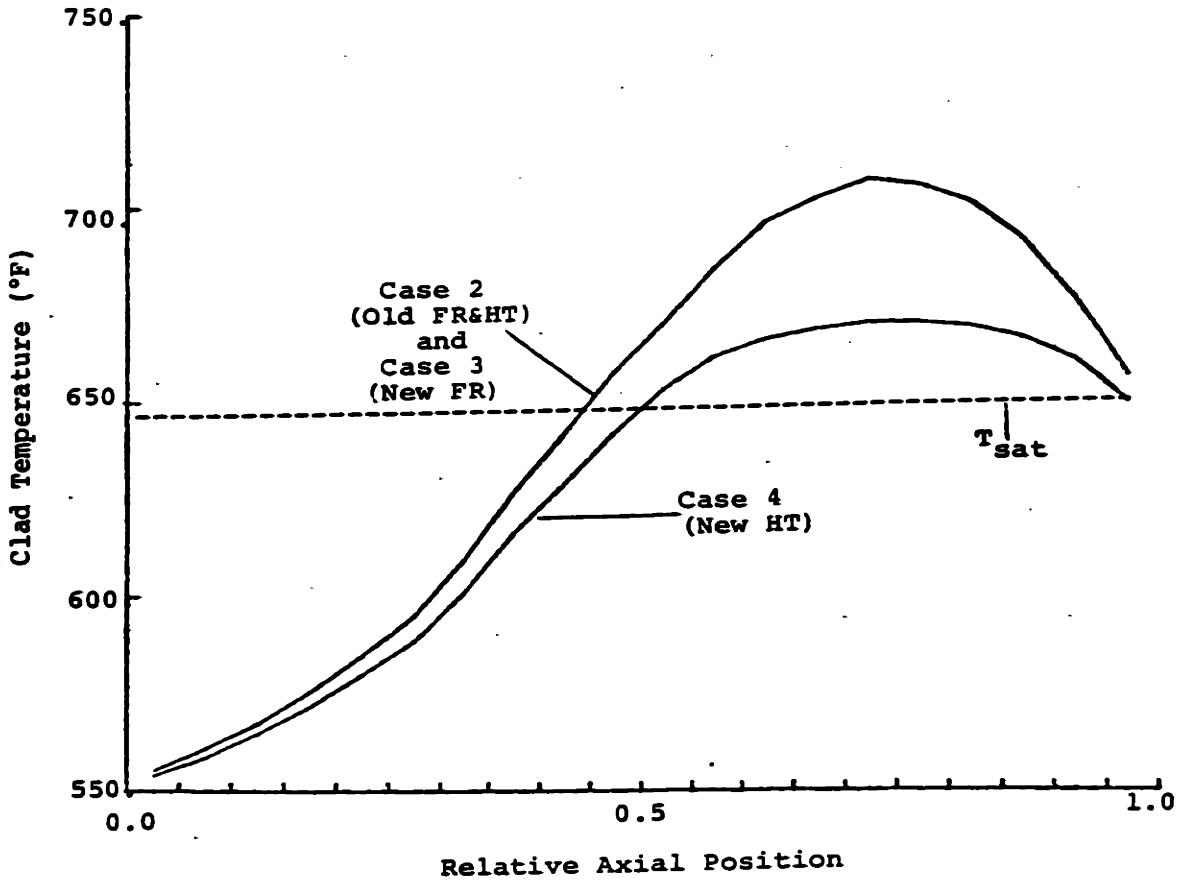


Figure IV-44

Axial Clad Temperature Profile  
 Rod 15, Time = 0  
 PWR Loss of Flow Transient Test Case

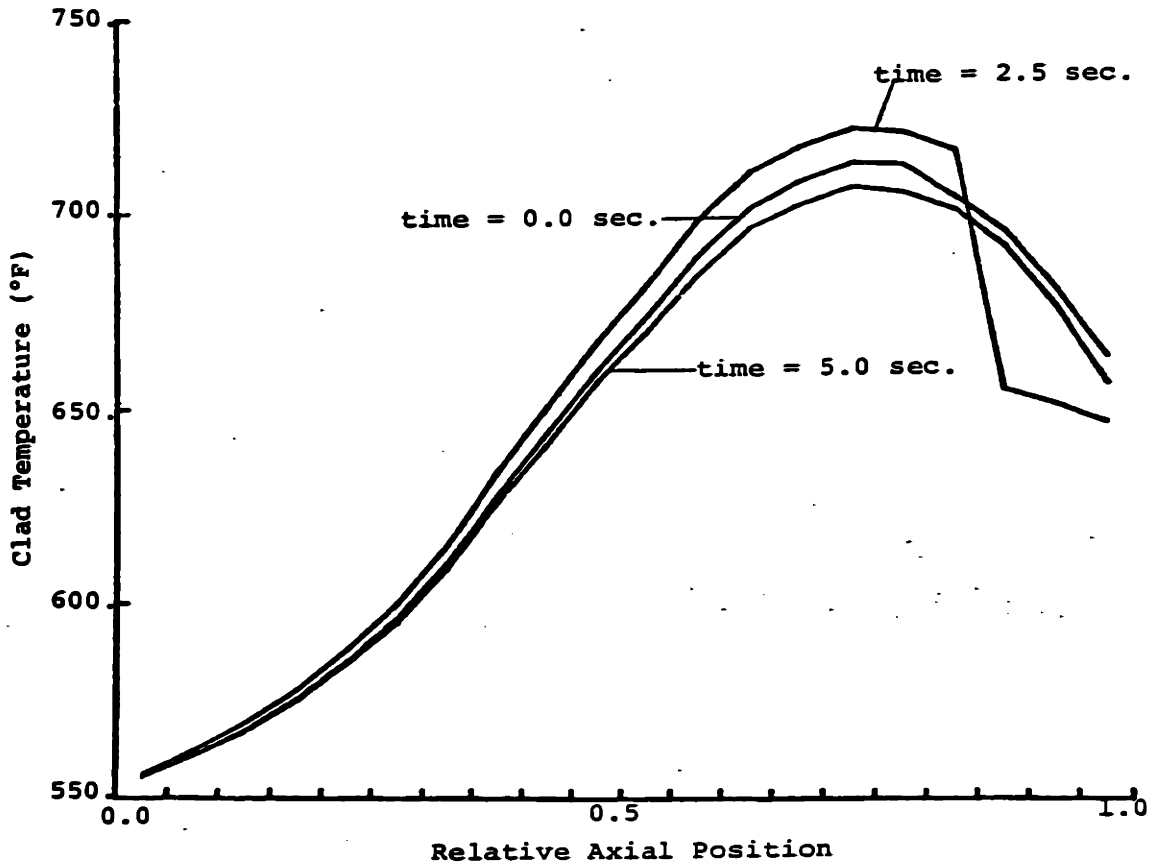


Figure IV-45

Axial Clad Temperature Profile

Rod 15

Analysis Case 2 (Old FR&HT)

PWR Loss of Flow Transient Test Case

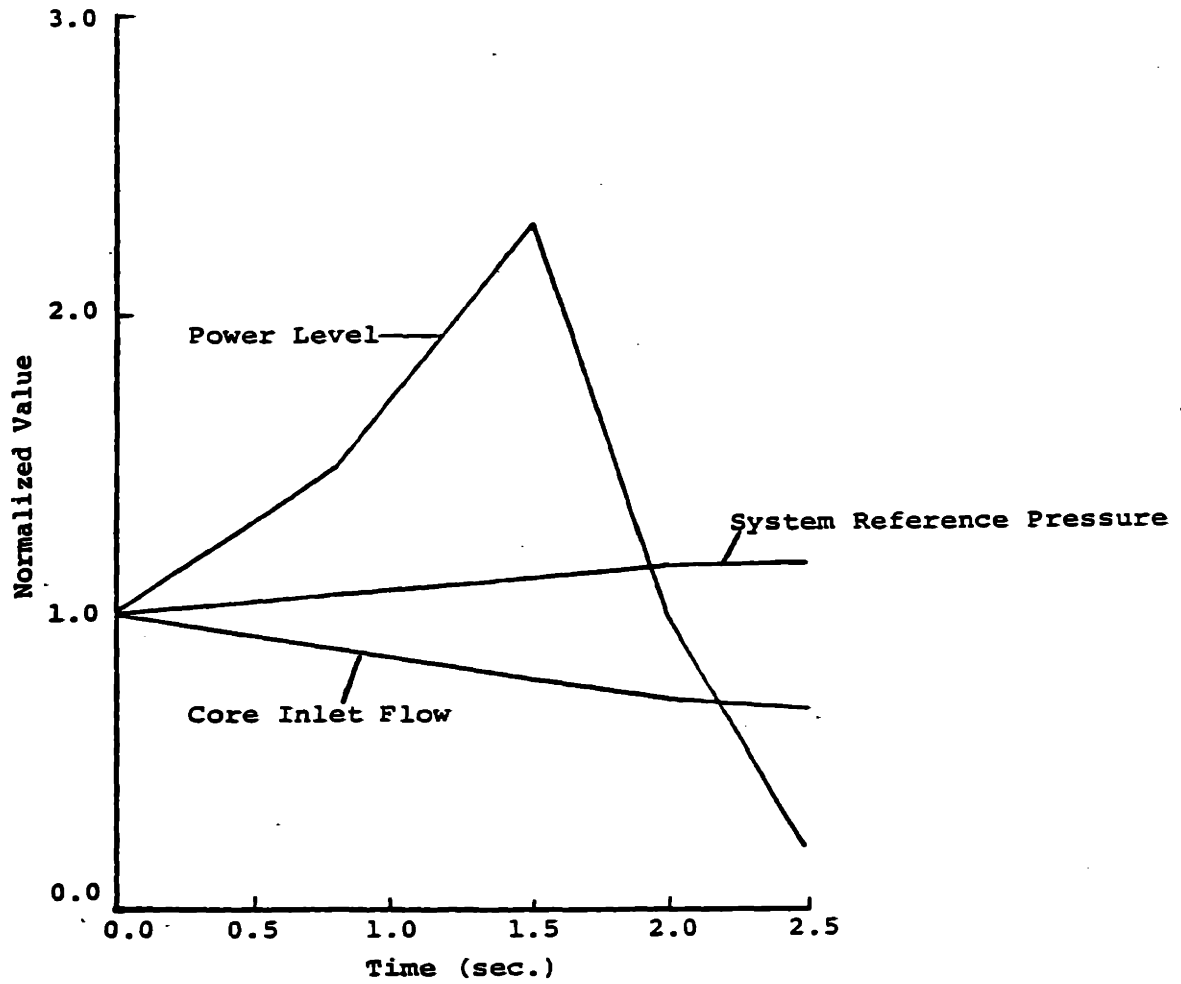


Figure IV-46

Transient Forcing Functions  
BWR Turbine Trip Transient Test Case

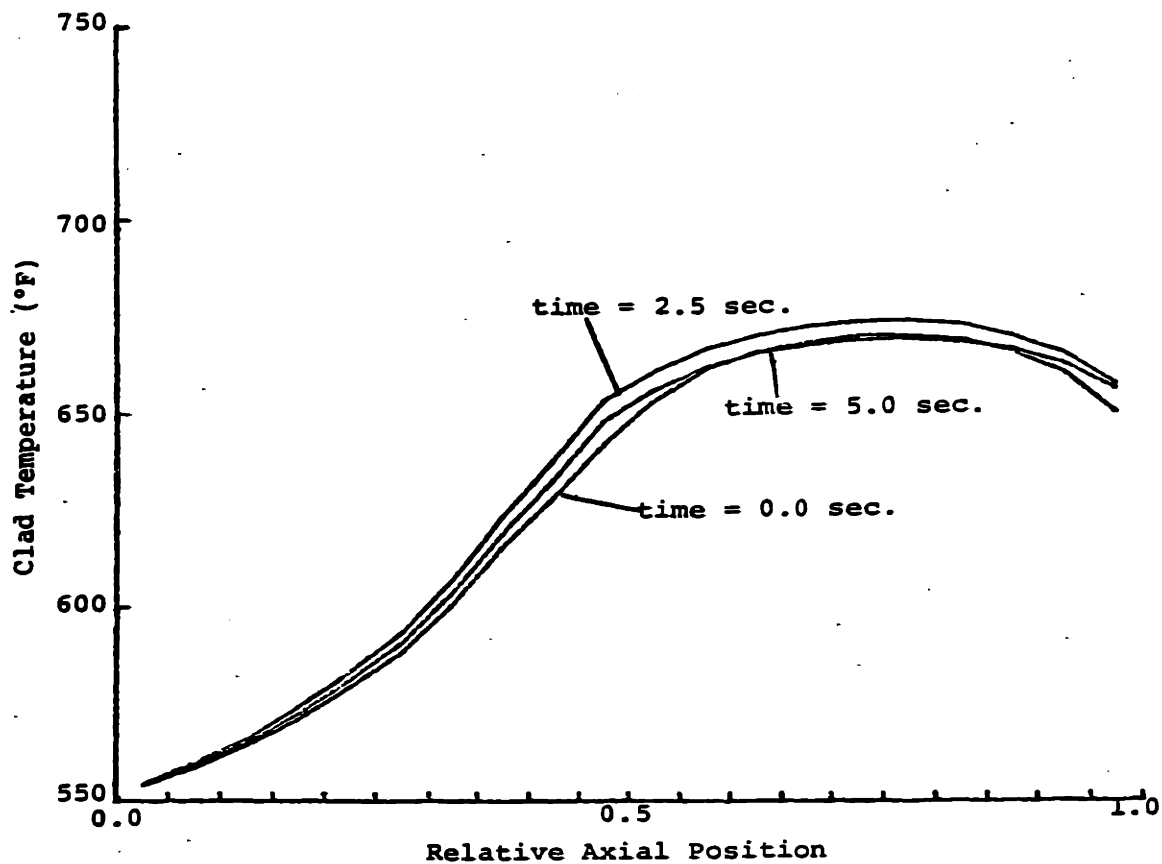


Figure IV-47

Axial Clad Temperature Profile  
Rod 15  
Analysis Case 4 (New HT)  
PWR Loss of Flow Transient Test Case

Table IV-7Models Used for Turbine Trip Analysis Cases

<u>Analysis Case Number</u>	<u>Fuel Rod Model</u>	<u>Fuel &amp; Clad Material Properties</u>	<u>Heat Transfer Model</u>
1	old	constant	old
2	new	temp.-dependent	old
3	old	constant	new
4	new	temp.-dependent	new



Analysis case predictions of MCPR version time are contained in Figure IV-40. MCPR predictions are within 3% of one another. Analysis Case 3 predictions are lowest. Analysis Case 1 MCPR predictions end at 2.0 seconds. The Case 1 flow solution failed to converge one time step after 2.0 seconds. This problem will be discussed later. The predictions shown a general downward trend which appears to level off near 2.5 seconds. The lowest MCPR value is 1.017.

MCHFR predictions are shown in Figure IV-41. MCHFR predictions are within 6% of one another. Analysis Case 3 predictions are lowest. The minimum predicted MCHFR value, 1.060 occurs at 2.25 seconds. MCHFR predictions at 2.5 seconds are larger than at 2.25 seconds.

The Analysis Case 1 flow solution failed to converge at 2.05 seconds, one time step after 2.0 seconds, as mentioned earlier. None of the other analysis cases had this problem. Instability of the solution is caused by coupling between the heat transfer and hydraulic calculations. Symptoms of a stability problem appear in Analysis Case 1 predictions near 2.0 seconds. Flow rate predictions in channel 2 at 1.75 and 2.0 seconds are shown in Figures IV-42 and IV-43, respectively. Flow rate predictions are close and follow the same smooth trend at 1.75 seconds. Flow rate predictions of Analysis Case 1 and 4 are not as smooth at 2.0 seconds. Analysis Case 1 shows much larger variations in flow rate than the other cases at 2.0 seconds.

Rod 2 axial heat flux profiles at 0.0 and 2.0 seconds are shown in Figure IV-44. Rod 2 is located in channel 2. All analysis cases start with the same heat flux profiles. Heat flux profiles are close to each other at 2.0 seconds except for the sharp dip of Analysis Case 1 predictions.

The sharp dip is caused by large changes in rod-to-coolant heat transfer predictions of the old heat transfer model which accompany diminishing void fractions. Axial void fraction profiles in channel 2 are shown in Figures IV-45 and IV-46. Void fraction profiles at 0.0 and 2.0 seconds are shown in Figure IV-45. All analysis cases start with the same void fraction

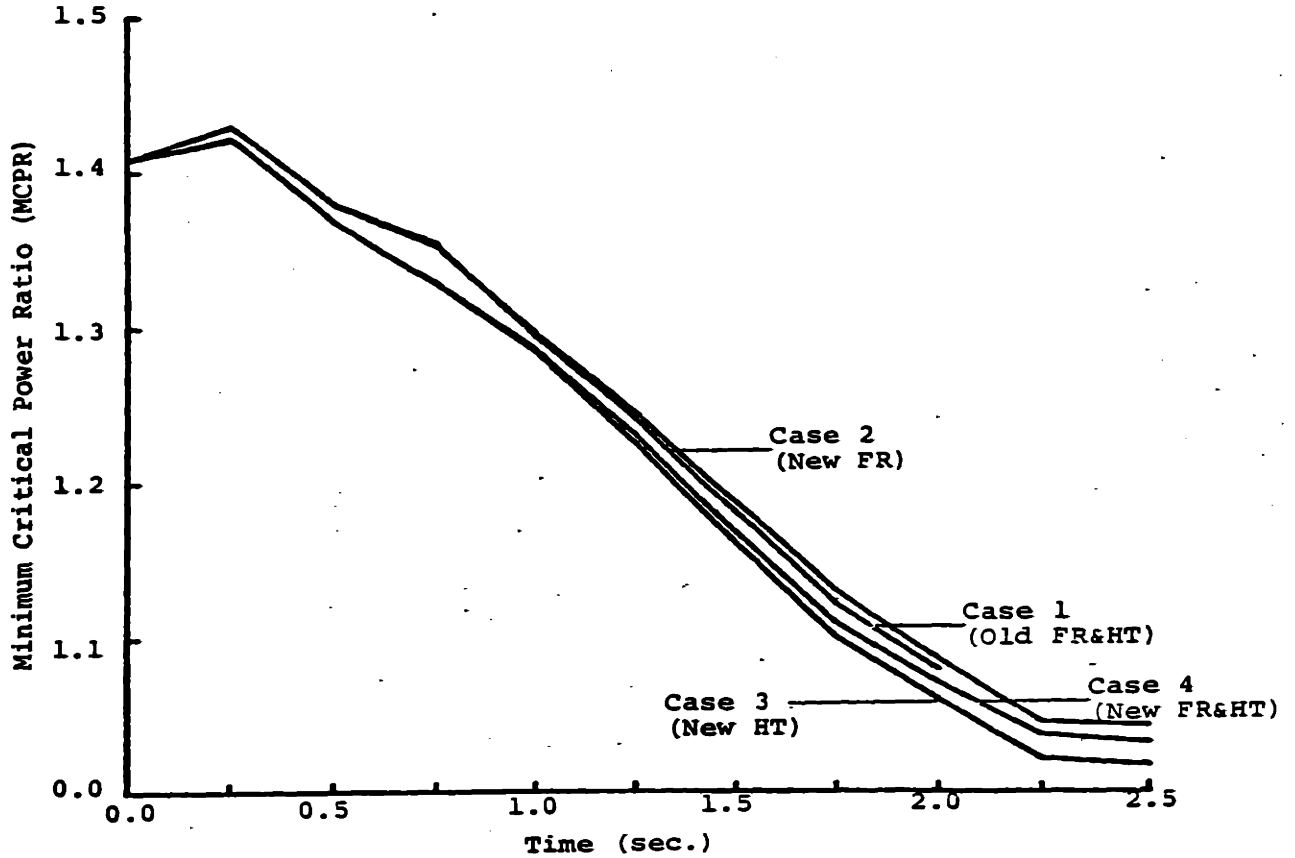


Figure IV-48

CISE-4 MCPR vs. Time  
BWR Turbine Trip Transient Test Case

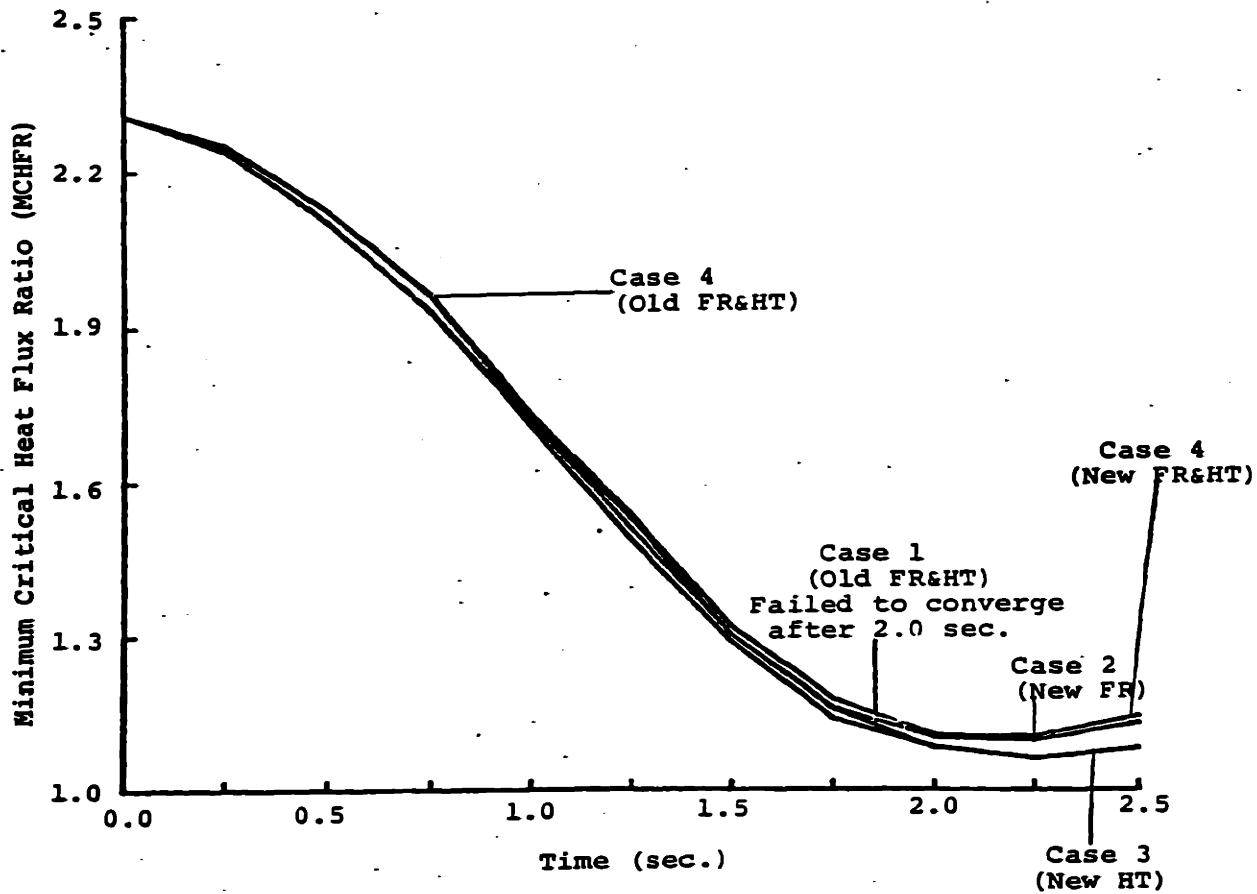


Figure IV-49

Hench-Levy MCHFR vs. Time  
 BWR Turbine Trip Transient Test Case

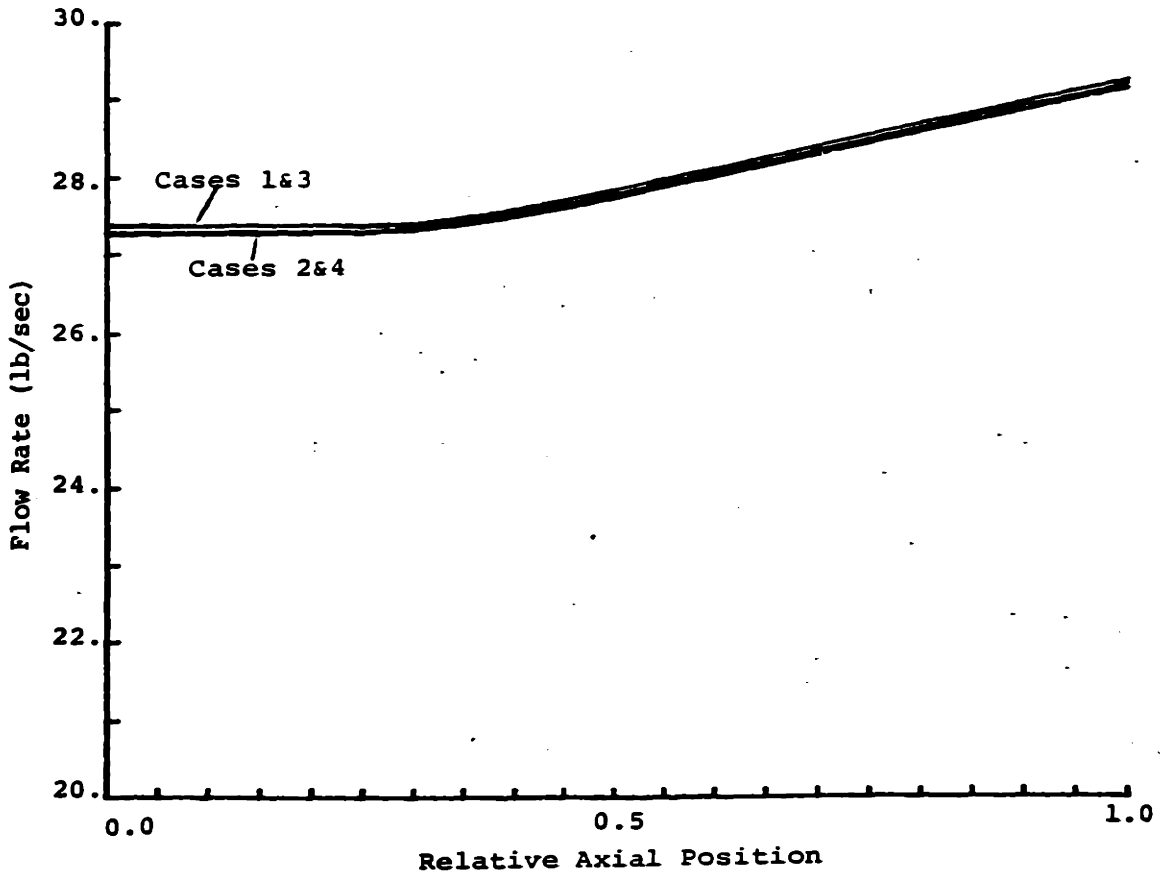


Figure IV-50

Flow Rate vs. Axial Position  
Channel 2, time = 1.75  
BWR Turbine Trip Transient Test Case

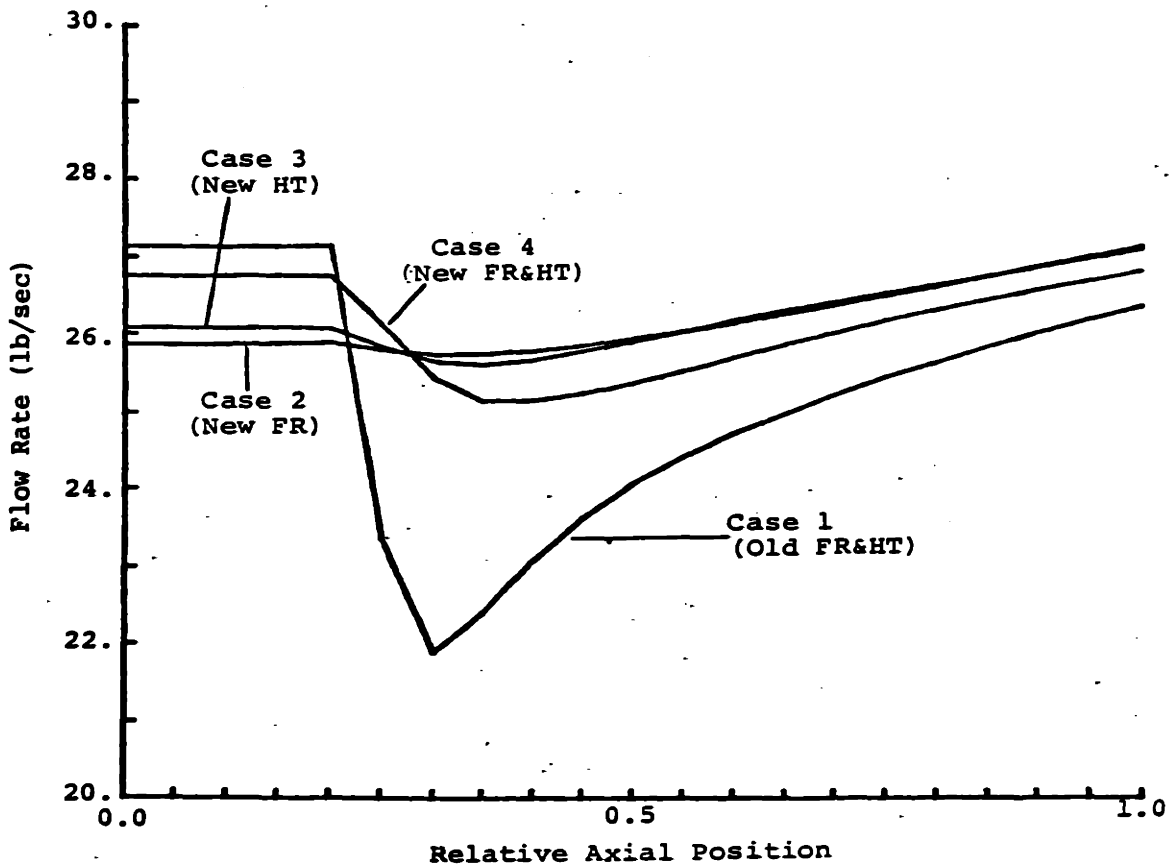


Figure IV-51

Flow Rate vs. Axial Position  
 Channel 2, time = 2.0 sec.  
 BWR Turbine Trip Transient Test Case

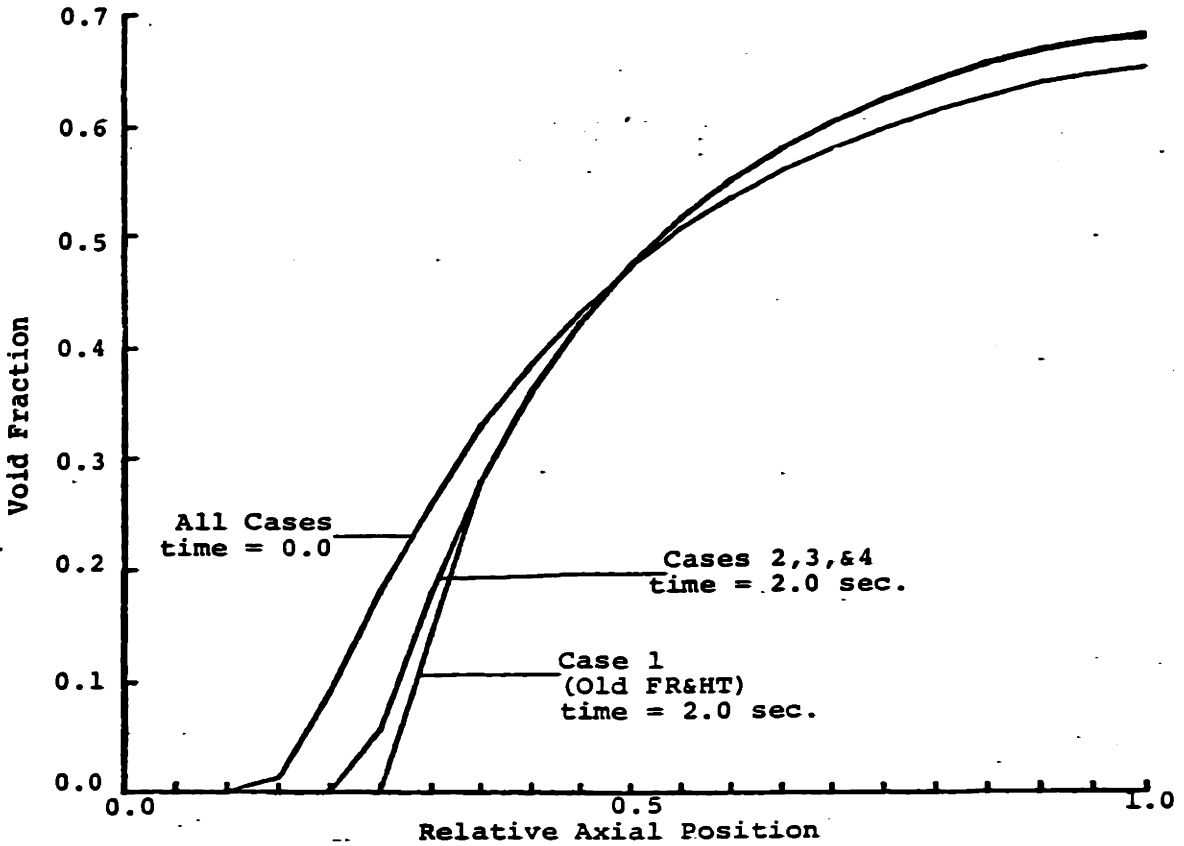


Figure IV-52

Axial Void Fraction Profile  
Channel 2  
BWR Turbine Trip Transient Test Case

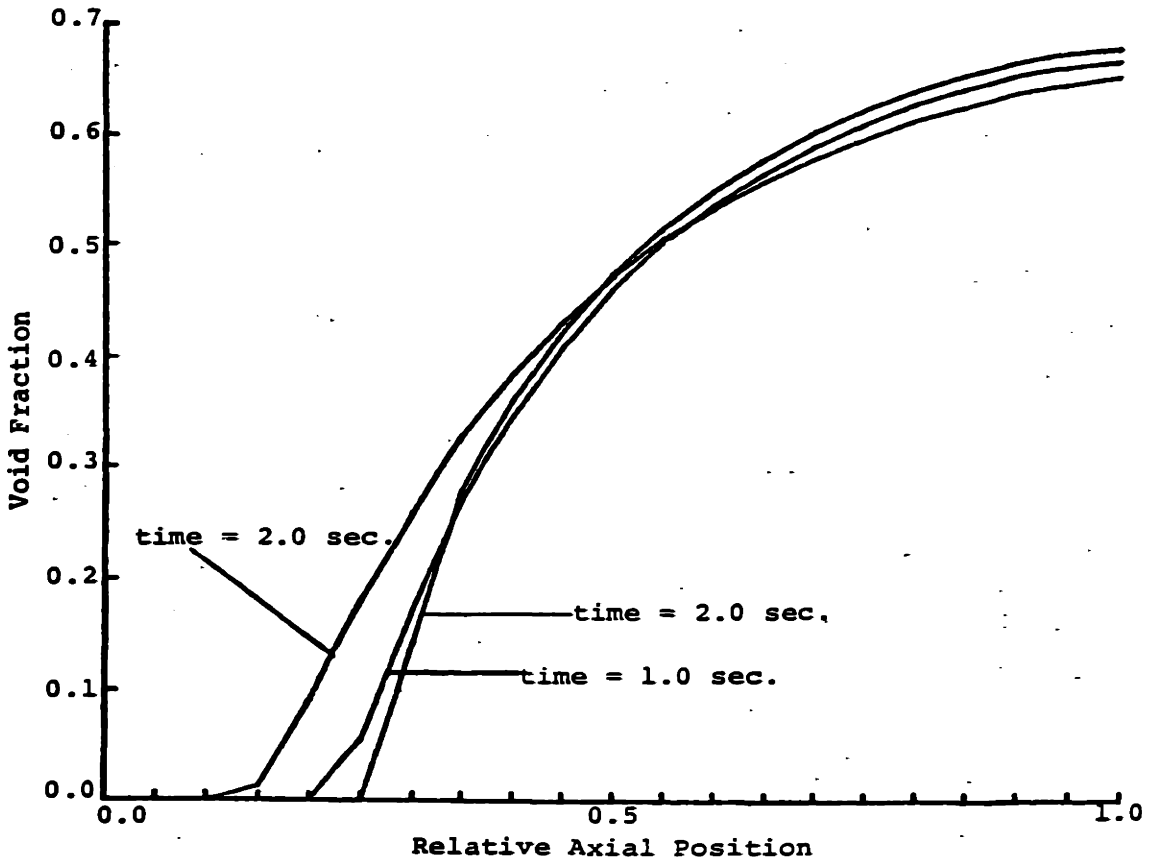


Figure IV-53

Axial Void Fraction Profile  
Channel 2  
Analysis-Case 1 (Old FR&HT)  
BWR Turbine Trip Transient Test Case

profile. Void fractions go to zero at lower axial positions due to pressure increases. Analysis Case 1 predictions at 2.0 seconds indicate that void fractions have become zero at three axial nodes. (Each tic represents one axial node.) Analysis Cases 2, 3, and 4 predictions at 2.0 seconds indicate that void fractions have become zero at two axial nodes. Decrease in void fractions at lower axial levels of channel 2 as time passes can be seen in Analysis Case 1 predictions shown in Figure IV-46. Axial clad surface temperature profiles show the effects of rod-to-coolant heat transfer models. Rod 2 axial clad temperature predictions are shown in Figures IV-47, IV-48 and IV-49. Clad temperature profiles at 0.0 seconds are shown in Figure IV-47. Analysis Cases 1 and 2 predict one profile using the new heat transfer model. Saturation temperature at 0.0 seconds is also shown in the figure. Clad temperature profiles are similar in shape to their initial profiles. The clad temperatures are higher than they were initially. Analysis Case 1 temperature profiles at 1.75 and 2.0 seconds are shown in Figure IV-49. The profile shows a change in shape due to rapid changes in rod-to-coolant heat transfer predictions of the old heat transfer model which occur when void fraction becomes zero at any axial node.

Fuel pellet temperature predictions of the old and new fuel rod models showed differences due mainly to differences in fuel pellet conductivity. The old model uses a constant value for fuel pellet conductivity. The new fuel rod model calculates fuel pellet conductivity as a function of temperature. The constant value for fuel pellet conductivity given to the old fuel rod model is too high for locations where fuel temperatures were highest. Fuel temperature predictions of the old and new fuel rod models are in better agreement at locations where fuel rod temperatures are not the highest.

Radial fuel pellet temperature distributions predicted by the old and new fuel rod models at 0.0 seconds are shown in Figure IV-50 for two fuel nodes of rod 1, the hot rod. Predictions for axial fuel nodes 5 and 10 are shown. Axial fuel node 5 is between the core inlet and midplane. The fuel pellet is



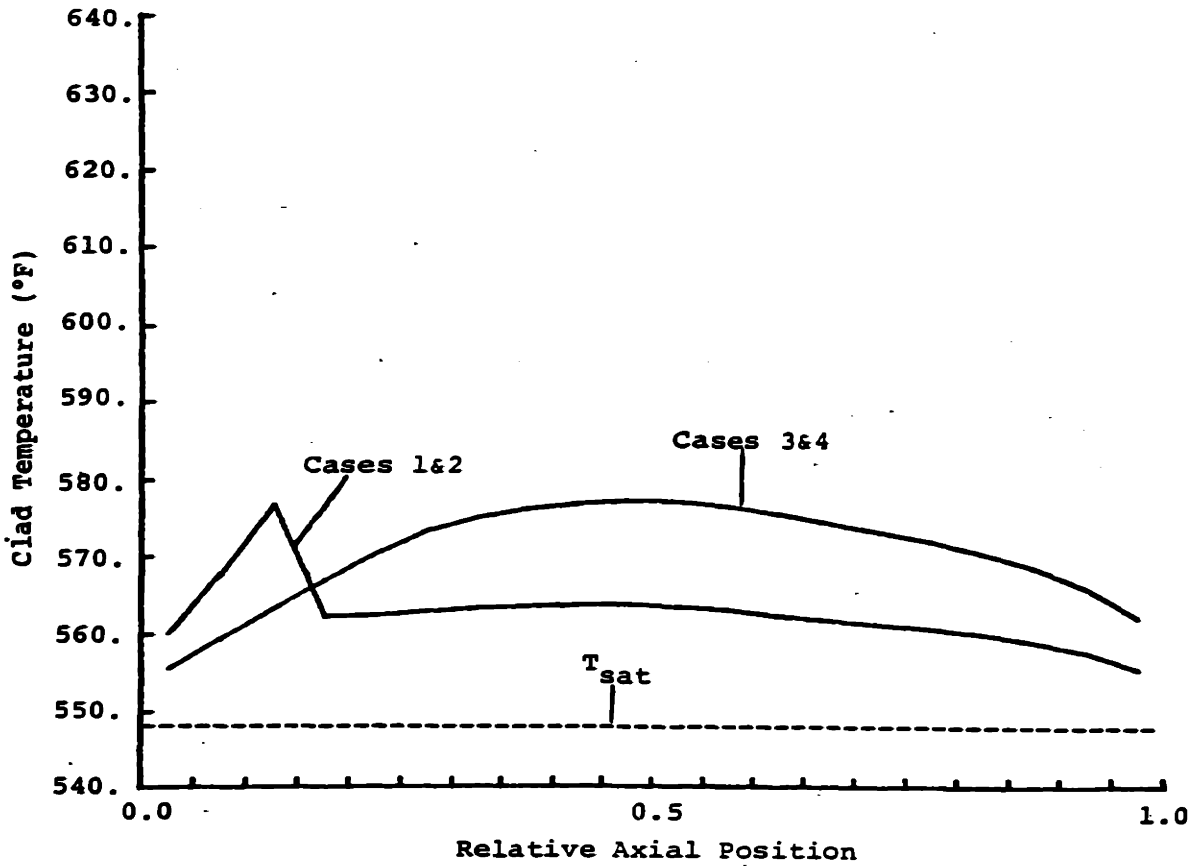


Figure IV-54

Axial Clad Temperature Profile  
Rod 2, time = 0.0 sec.  
BWR Turbine Trip Transient Test Case

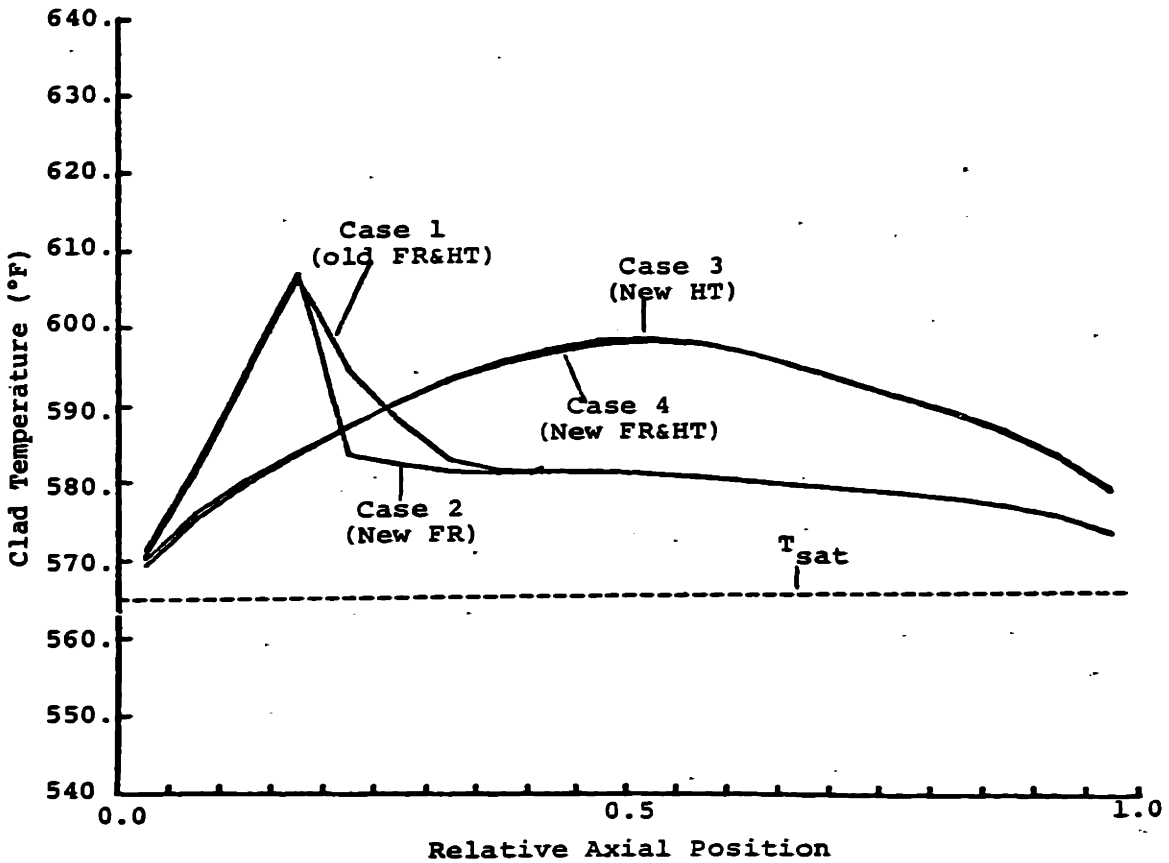


Figure IV-55

Axial Clad Temperature Profile  
 Rod 2, time = 2.0 sec.  
 BWR Turbine Trip Transient Test Case

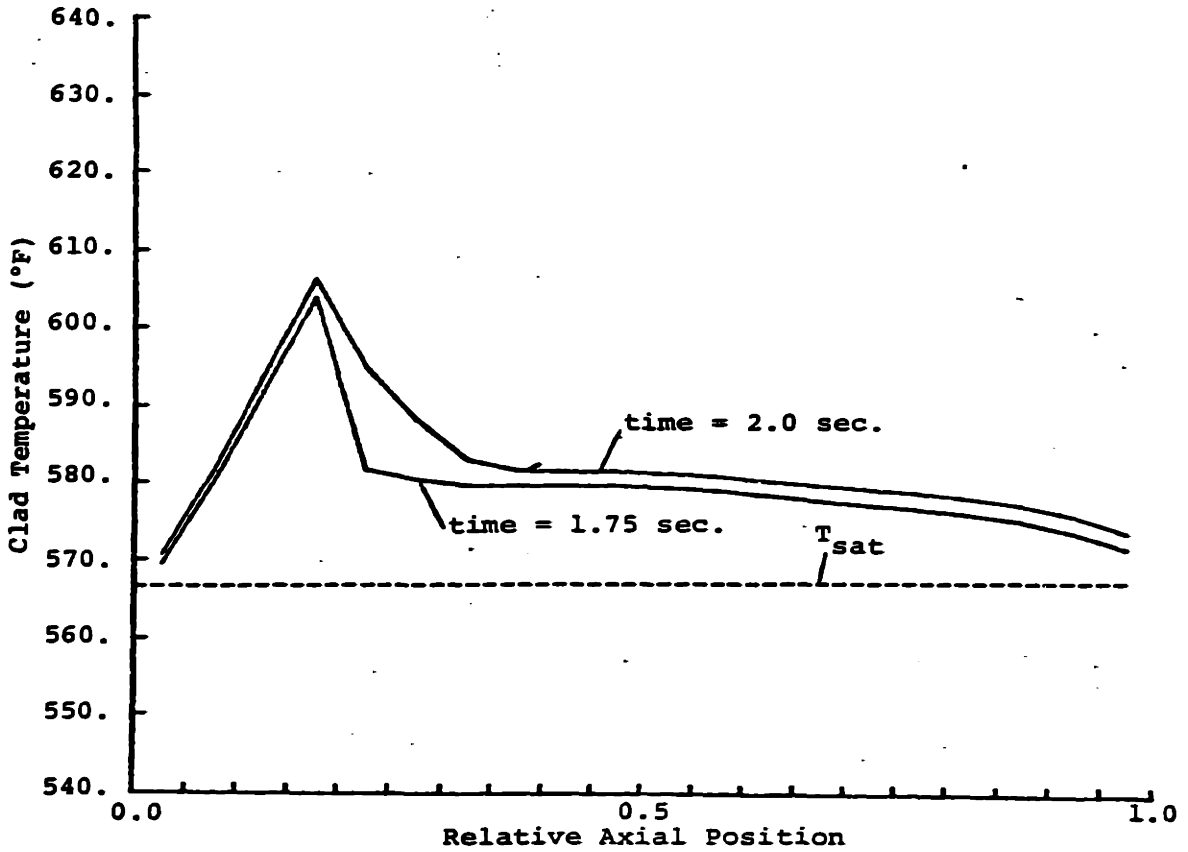


Figure IV-56

Axial Clad Temperature Profile  
Rod 2  
Analysis Case 1 (Old FR&HT)  
BWR Turbine Trip Transient Test Case

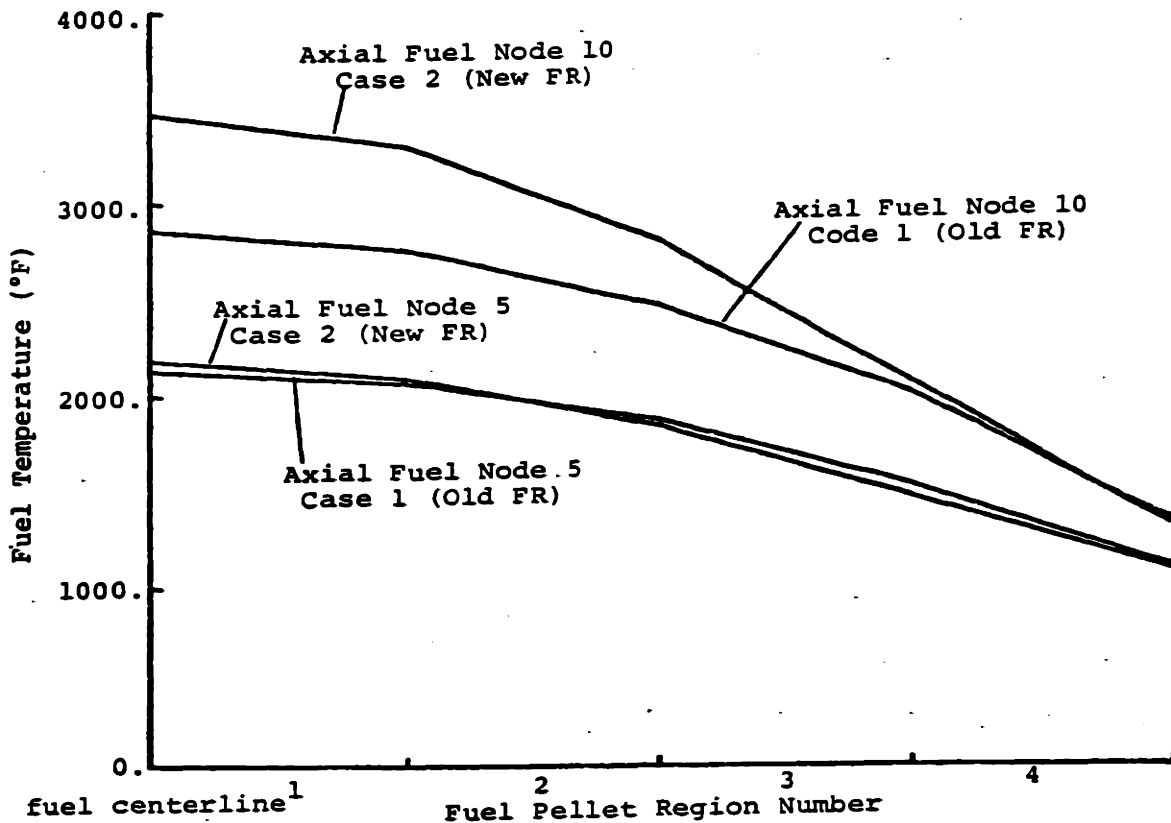


Figure IV-57

Radial Fuel Pellet Temperature Distribution  
 Rod 1, Time = 0.0 sec.  
 BWR Turbine Trip Transient Test Case

divided radially into four regions. Fuel centerline temperature predictions are at the left edge of the Figure IV-50 graph. Fuel pellet surface temperature predictions at the right edge are nearly the same for both axial fuel nodes. The old and new fuel rod model predictions are close for axial fuel node 5. The predictions are much farther apart for axial node 10, where fuel temperatures are higher than node 5. Higher temperatures are predicted by the new fuel rod model because fuel conductivity is calculated to be lower than the constant value used by the old fuel rod model.

Fuel centerline temperature predictions indicated that the constant fuel conductivity value used by the old fuel rod model was better for fuel at lower temperatures. Figures IV-51 and IV-52 show centerline temperature predictions of the old and new fuel rod models for rods 1 and 2 at 0.0 seconds. Centerline temperature predictions for rod 1 are shown in Figure IV-51. Predictions are farther apart in the vicinity of the core mid-plane. Centerline temperature predictions for rod 2 are shown in Figure IV-52. Rod 2 has a lower radial power factor than rod 1. Predictions of the old and new fuel rod models are closer together for this rod because fuel temperatures are lower.

The differences in predictions indicate a general shortcoming of the old fuel rod model. It can only use constant fuel rod properties. This limits the old fuel rod model to one value for a parameter such as fuel pellet conductivity, which is actually a function of space and time.

#### d. Summary

The turbine trip without bypass transient was analyzed using four combinations of old and new COBRA-IIIC/MIT rod-to-coolant heat transfers and fuel rod models. Predictions for MCPFR and MCHFR were close. Analysis Case 1, which used the old heat transfer and fuel rod models, had a convergence failure at 2.05 seconds. Coupling of the heat transfer and hydraulic calculations allowed sudden changes in heat transfer to cause instability in the flow solution. Analysis Cases 2, 3, and 4, which used the new heat transfer and/or new fuel rod model, did not have flow convergence problems. Differences

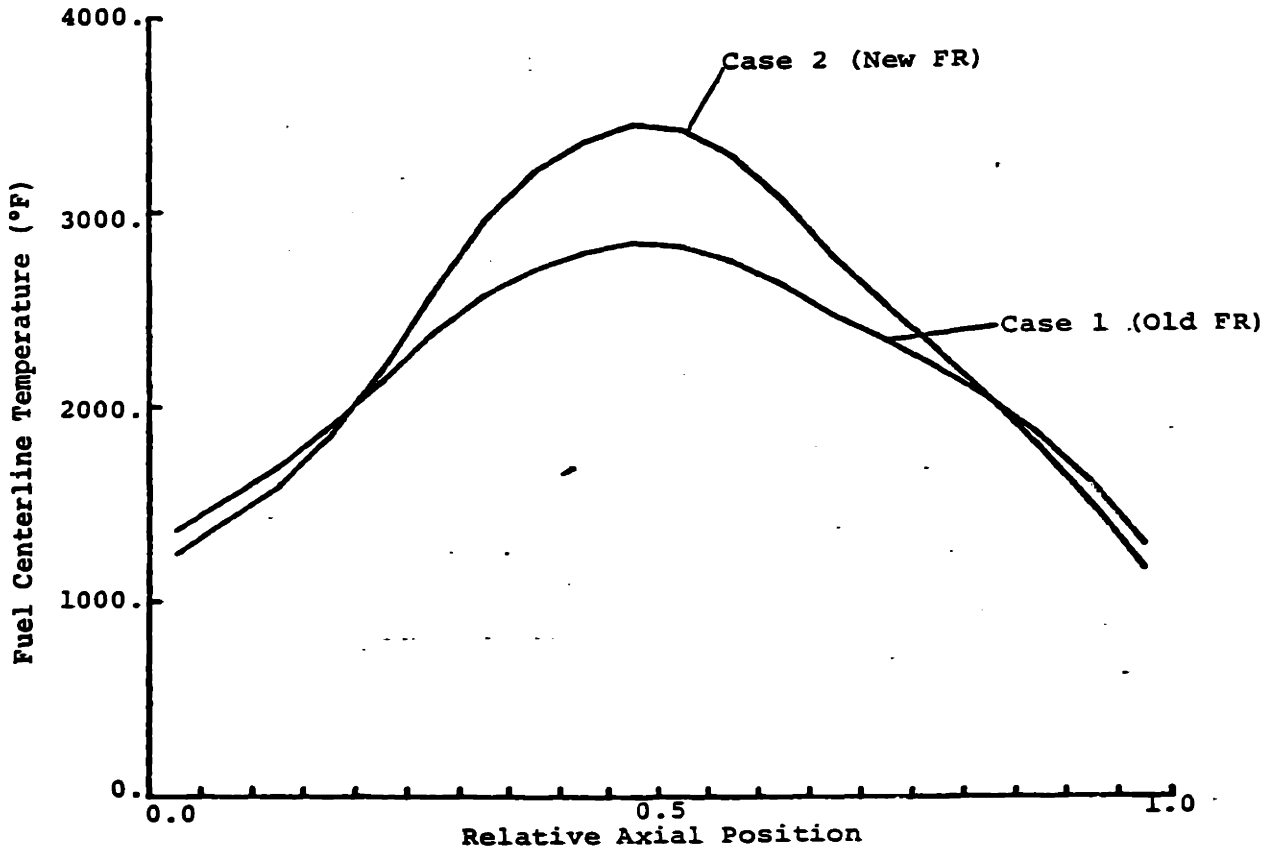


Figure IV-58

Centerline Temperature vs. Axial Position  
Rod 1, Time = 0.0 sec.  
BWR Turbine Trip Transient Test Case

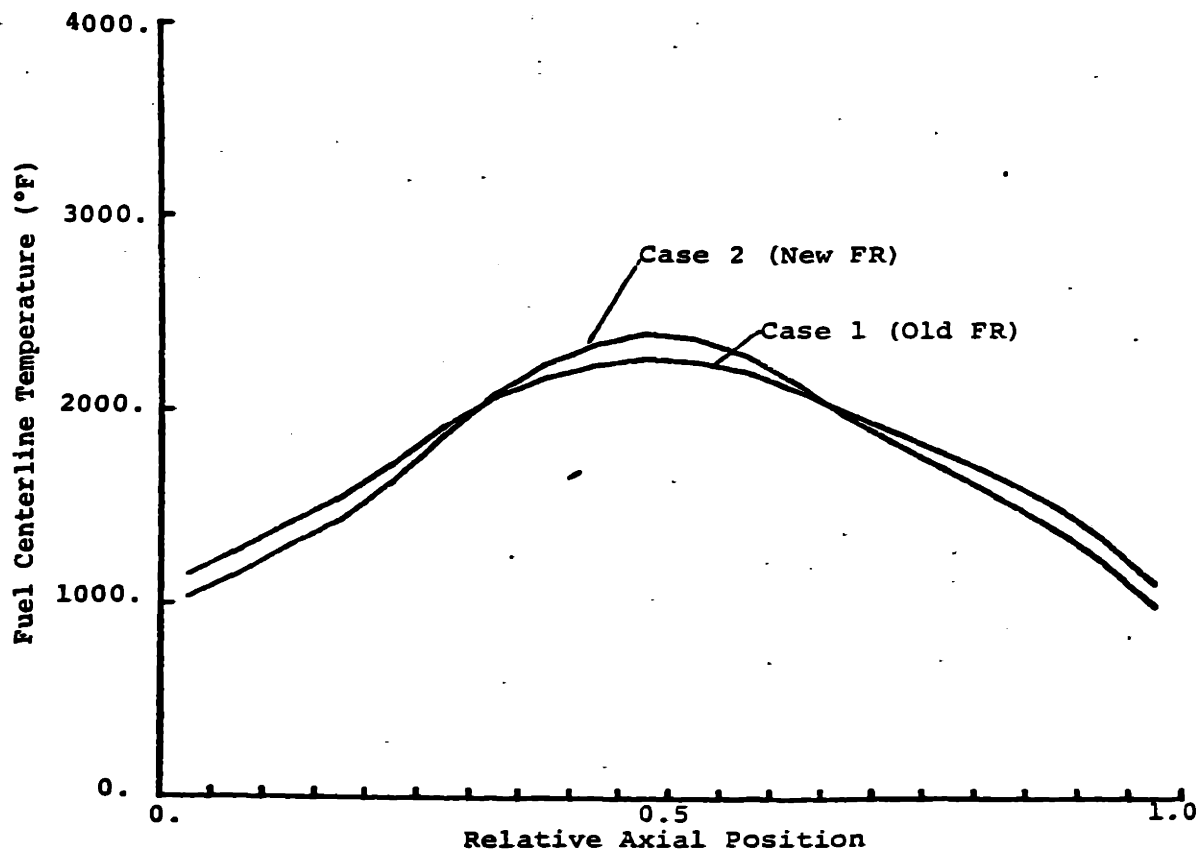


Figure IV-59

Centerline Temperature vs. Axial Position  
Rod 2, Time - 0.0 sec.  
BWR Turbine Trip Transient Test Case

exist between the predictions of the old fuel rod model, using constant fuel and clad properties, and the new fuel rod model, using temperature-dependent fuel and clad properties.

#### D. Application to BWR Bypass Analysis

##### 1. Introduction

Conditions in the bypass channels surrounding enclosed fuel assembly channels affect the operation of a BWR. Boiling in the bypass channels, if it occurs, would affect power distribution and instrument response in the reactor core. As preliminary work for bypass analysis of the E.I. Hatch Unit 1 reactor, two small test cases were analyzed to gain an understanding of how bypass flow analysis can be performed for an entire reactor. It is specifically desired to know the range of COBRA-IIIC/MIT applicability and whether recirculation is present under operating conditions. The range of COBRA-IIIC/MIT applicability has been investigated to a greater extent than the question of recirculation. The first test case consists of nineteen analysis channels representing the bypass region surrounding ten enclosed fuel assemblies. The nineteen channel test case geometry was analyzed by Carlson and Gott (Ref. 45). This geometry contains some of the complexities of a full scale analysis, such as multiple flow paths and circular flow paths between analysis channels. The second test case consists of four channels representing a smaller bypass region. The first small test case was analyzed using COBRA-IIIC/MIT for varying power and flow conditions. The second small test case was analyzed using COBRA-IIIC/MIT and THERMIT. THERMIT is a three-dimensional two-fluid code. THERMIT, unlike COBRA-IIIC/MIT, has the capability to consider recirculation. The small size of the second small test case made THERMIT analysis economical. Heat transfer to bypass coolant was approximated by fixed heat fluxes from the channel walls for all analyses. The results of small test case analyses indicate general behavior



of bypass coolant flow and provide an understanding of approaches to BWR bypass analysis.

## 2. Nineteen Channel Test Case

### 2.a. Description of Nineteen Channel Test Case Geometry and Operating Conditions

The nineteen channel test case geometry represents the bypass regions associated with ten assemblies as shown in Fig. IV-60. The bypass flow analysis channels are numbered from one to nineteen in the figure. The assemblies are numbered from one to ten. The assemblies are separated by wide and narrow gaps. Control blades may be inserted at the intersection of wide gaps. Two configurations of the nineteen channel geometry will be considered. First, without any control rods inserted and secondly, with the control rod associated with bypass channel ten partially inserted. The fuel assembly and flow channel dimensions are included in Table IV-8 along with system pressure and inlet enthalpy. Other operating conditions which overlap and differ from those analyzed by Carlson and Gott are specified in Section 2.b. The original core design had holes drilled in the core support to augment bypass flow. Later, these holes were plugged, reducing bypass flow. Carlson and Gott did not take reduction of bypass flow into account in their bypass analyses.

### 2.b. Description of Modeling

The nineteen channel geometry was analyzed using COBRA-IIIC/MIT. Heat transfer to the bypass coolant was represented by fixed heat fluxes from the channel walls. Each assembly wall used the axial heat flux profile shown in Fig. IV-61. The boundary conditions used for the flow solution scheme were as follows:

- Channel outlet pressures all equal to a single specified value.

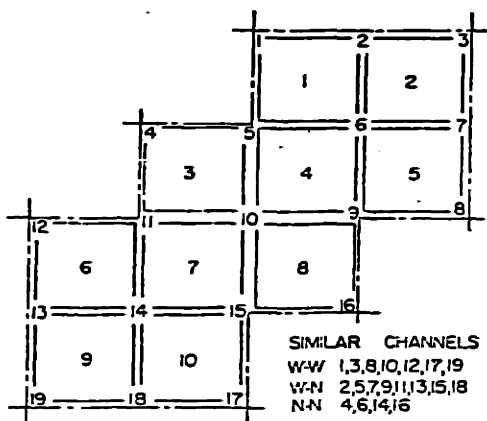


Figure IV - 60 (Fig. 3 of Ref. 45)

Nineteen Channel Test Case Geometry

Table IV-8 (Table I of Ref. 45)  
 Nineteen Channel Geometry and Operating Conditions

Subchannel Data, Control Rods Withdrawn					
Number	Area (mm <sup>2</sup> )	Wetted Perimeter (mm)	Ileated Perimeter (mm)	Adjacent Channels and Spacing (mm)	
1	1406.1	138.13	138.13	2, 9.53	5, 9.53
2	2061.4	276.35	276.35	1, 9.53	3, 9.53
3	1406.1	138.13	138.13	2, 9.53	7, 9.53
4	678.5	138.13	138.13	5, 4.75	11, 4.75
5	3092.8	414.27	414.27	1, 9.53	4, 4.75
6	2714.2	552.45	552.45	2, 9.53	5, 9.53
7	2061.4	276.35	276.35	3, 9.53	6, 9.53
8	1406.1	138.13	138.13	7, 9.53	9, 9.53
9	3092.8	414.27	414.27	6, 9.53	8, 9.53
10	524.4	552.45	552.45	5, 19.1	9, 19.1
11	3092.8	414.27	414.27	4, 4.75	10, 19.1
12	1406.1	138.13	138.13	11, 9.53	13, 9.53
13	2061.4	276.35	276.35	12, 9.53	14, 9.53
14	2714.2	552.45	552.45	11, 9.53	13, 9.53
15	3092.8	414.27	414.27	10, 19.1	14, 9.53
16	678.5	138.13	138.13	9, 4.75	15, 4.75
17	1406.1	138.13	138.13	15, 9.53	18, 9.53
18	2061.4	276.35	276.35	14, 9.53	17, 9.53
19	1406.1	123.13	138.13	13, 9.53	16, 9.53
Subchannel Data, Central Control Rod Inserted					
5	2778.0	414.27	414.27	1, 9.53	4, 4.75
9	2778.0	414.27	414.27	6, 9.53	8, 9.53
10	3612.0	552.45	552.45	5, 12.4	9, 12.4
11	2778.0	414.27	414.27	4, 4.75	10, 12.4
15	2778.0	414.27	414.27	10, 12.4	14, 9.53
				16, 4.75	17, 9.53

Table IV-8 (cont'd.)

<u>Fuel Assembly Data</u>	
Perimeter	0.5525 m
Channel Length	4.262 m
<u>Operating Conditions</u>	
System pressure	7.135 MPa
Inlet enthalpy	1.226 MJ/kg

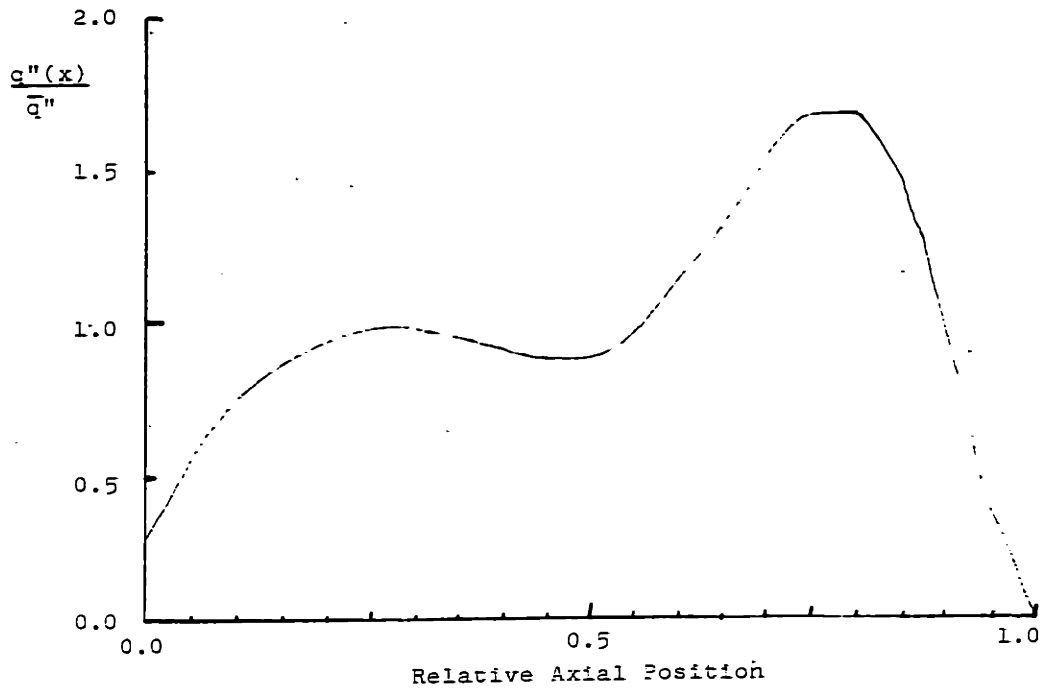


Figure IV-61  
Axial Heat Flux Profile  
Bypass Analysis

- A specified total inlet flow rate is split using an iterative scheme to give equal pressure drops across the first axial node of all channels. The flow split scheme assumes there is no crossflow and no density change as coolant flows in the axial direction.

The four nineteen channel bypass cases listed in Table IV-9 were each analyzed for varying average inlet flow rates and average heat fluxes. The cases differ in the heat apportionment method used and whether all control rods are out or one control rod is partially inserted. Two heat apportionment methods were used. Both methods rely on fixed heat fluxes, hence, no actual heat transfer calculations are made. The two heat apportionment methods will be referred to as the old and new heat apportionment methods. The old heat apportionment method evenly divides the heat transferred from a fuel assembly to surrounding bypass channels according to the fraction of fuel assembly wall perimeter in contact with each channel. The old heat apportionment method approximates the apportionment expected if the Carlson and Gott analysis methodology were applied. The new heat apportionment method approximates the apportionment expected if the heat transfer model which is developing from new understandings (Ref. 46) were applied. Nineteen channel bypass cases one to three use the radial power factors shown in Fig. IV-62. These factors are for the highest power region of a core. Case four has a flat radial power distribution. It represents the situation of radial peaking factors all equal unity.

Various phenomena were modeled using COBRA-IIIC/MIT options as indicated in Table IV-10. Options used for PWR analysis are included in the table for comparison. COBRA-IIIC/MIT input for a nineteen channel bypass case is given in Appendix P.

Table IV-9Nineteen Channel Bypass Cases

<u>Case</u> <u>Number</u>	<u>Heat</u> <u>Apportionment*</u>	<u>Control</u> <u>Rod</u>
1	New	Out
2	Old**	Out
3	Old**	
4	New - Flat radial distribution	Out

\* See accompanying text for explanation of heat apportionment methods.

\*\* The old heat apportionment method divides the heat transfer from each bundle to associated bypass channels in a manner which closely approximates the apportionment used in Carlson and Gott analyses.

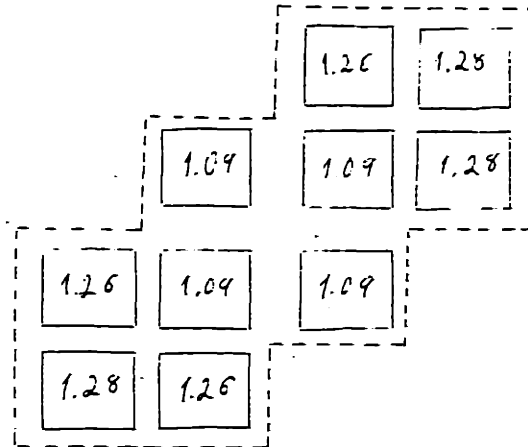


Figure IV-62

Radial Peaking Factors

Nineteen Channel Bypass Cases One, Two and Three



Table IV-10COBRA-IIIC/MIT Options Used for Bypass Analysis

<u>Phenomenon</u>	<u>"Standard" PWR Analysis</u>	<u>BWR Bypass Analysis</u>
Subcooled voids	Levy model	No subcooled void model
1-phase friction	$f = 0.184 Re^{-.2}$	same
2-phase friction	no multiplier	same
wall viscosity correction	yes	no
$K_{ij}$ (transverse friction coefficient)	0.5	same
$s/l$	0.5 (or 0.24)	0.094*
turbulent enthalpy mixing	$\beta = 0.02$	same
thermal conduction boundary conditions	none equal $\Delta P$ over first node and outlet pressures equal for all channels	same

\*  $l$  is based on centroid-to-centroid distance.  $s$  is the average of gap widths.

## 2.c Analysis Case Results

### 2.c.i COBRA-IIIC/MIT Applicability Limits

COBRA-IIIC/MIT applicability limits were obtained by analyzing the four cases listed in Table IV-9 for various combinations of power and flow. COBRA-IIIC/MIT provides predictions for some power and flow combinations and fails to provide predictions for other combinations. Failure occurs by either a failure of the flow solution to converge within a specified maximum number of iterations or, when the power-to-flow ratio is made larger, by extreme code failure. The extreme code failures occurred during property lookups under conditions approaching stagnant flow and, ultimately, flow reversal. Some channels had near-zero flow rates and unreasonably high enthalpy when these extreme failures occurred.

The most severe conditions for which COBRA-IIIC/MIT predictions can be obtained are shown in Fig. IV-63 as limits of applicability for each of the four nineteen channel bypass cases. COBRA-IIIC/MIT failed in the zone below the lines. The limit line for case three, which has a control rod partially inserted, is higher than the lines for cases one, two and four which have no control rods inserted. The case four limit of applicability is shown only for  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup>. The flat power distribution of case four gives it the largest applicability zone. The point labeled "full power and full flow conditions" is below all the lines of applicability in Fig. IV-63. This point was obtained using two assumptions. The first assumption is that 2% of assembly power will be transferred to bypass coolant. The second assumption is that the core bypass flow rate is that shown in Fig. IV-64 for 100% core flow after core support plate hole plugging. In addition, no control rods are inserted. From the location of this point relative to the lines of applicability, it appears that COBRA-IIIC/MIT will

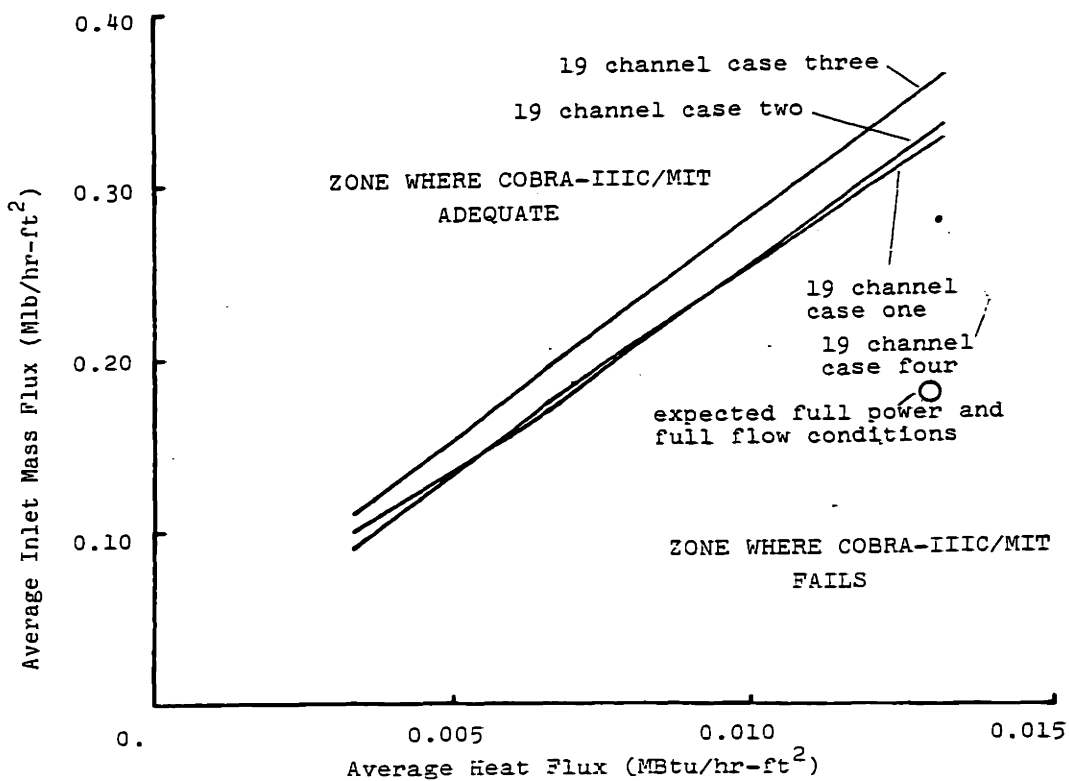
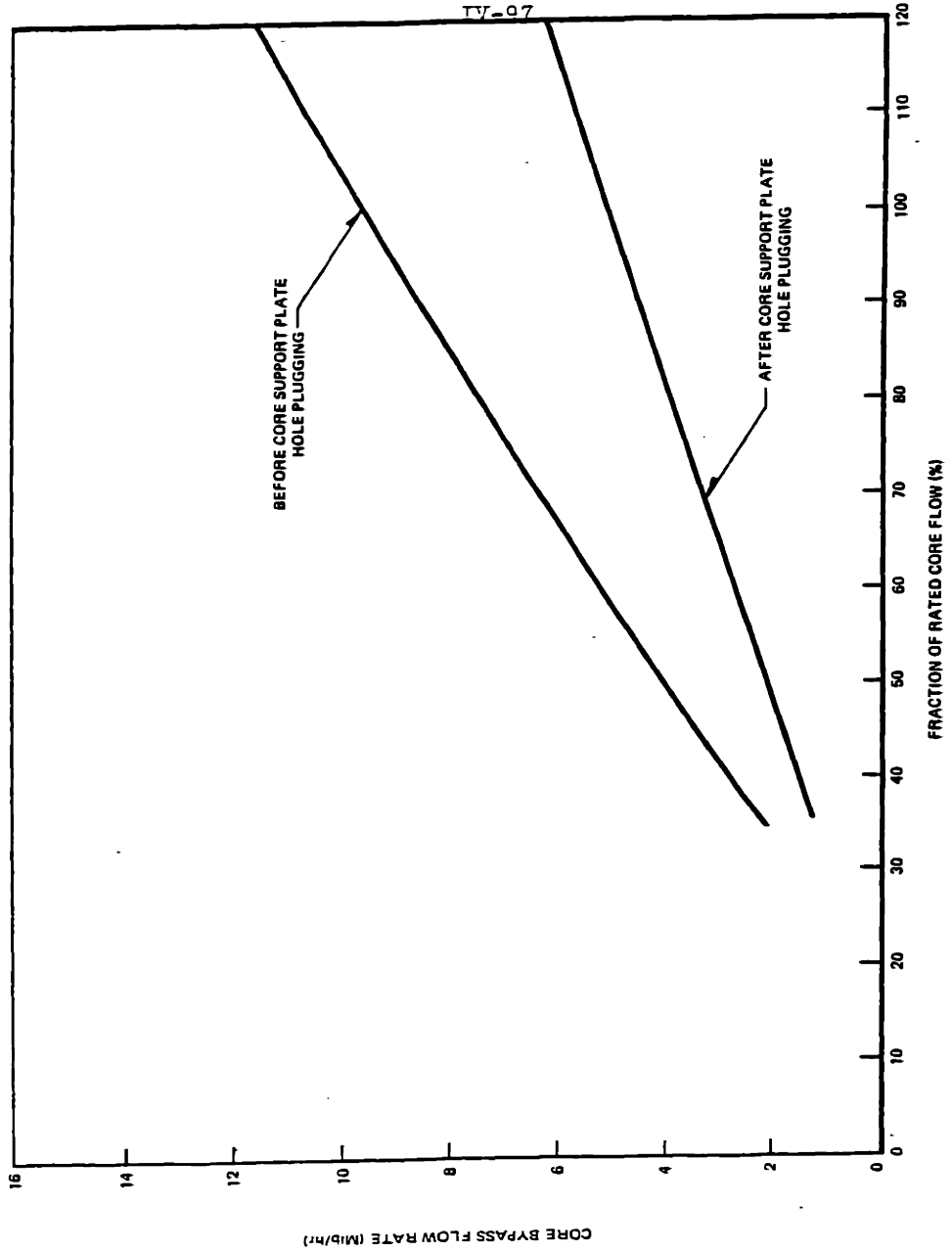


Figure IV-63

Mass Flux vs. Heat Flux

COBRA-IIIC/MIT Applicability Limits for Nineteen Channel Cases



Core Bypass Flow for Cycle 1  
 Figure IV-64 (Fig. 23 of Ref. 47)

not be able to provide predictions for bypass cases of interest at full power and probably lower power levels, as well.

Analyses made during the process of determining COBRA-IIIC/MIT applicability limits for each of the four nineteen channel cases is given in Tables IV-11 through IV-14. The particular analyses used in Fig. IV-63 are listed in Table IV-15.

#### 2.c.ii Nineteen Channel Bypass Case One Predictions

Case one has no control rods inserted and uses the new heat apportionment method. COBRA-IIIC/MIT predictions are shown in Figs. IV-65 through IV-72 for  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup>. Figures IV-65 through IV-68 are for  $\bar{G} = 0.3290$  Mlb/hr-ft<sup>2</sup>. Axial mass flux profiles are shown in Fig. IV-65. Up to a relative axial position,  $x/L$  equal to 0.75, flow is redistributing to give channels with larger hydraulic diameters, the wide-wide (W-W) channels, a larger mass flux than the narrow-narrow (N-N) channels. The flow split scheme to give equal pressure drop across the first node worked poorly since the frictional pressure drop is a small fraction of the total pressure drop. Otherwise, the flow split scheme would function better and less redistribution would occur within the axial distance from the inlet to  $x/L = 0.75$ . Near  $x/L = 0.75$ , vapor starts to form in the channels, as Fig. IV-65 shows in a plot of  $\alpha$  vs  $x$ . At  $x/L = 0.9$ , flow has redistributed to give the N-N channels greater mass flux than the W-W channels. Production of vapor in the N-N channels draws in flow from other channels because the total pressure drop is heavily dominated by gravitational pressure drop. The rapid change in density which occurs when voids are produced is seen in Fig. IV-67 which shows axial density profiles for each channel. Channels 6 and 14 are both N-N channels and are adjacent to assemblies with high radial peaking factors, as Fig. IV-62

Table IV-11

Nineteen Channel Bypass Case One  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{q}$ (Mlb/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{max}$	$\alpha_{max}$
0.0132	0.3150	no			
"	0.3200	no			
"	0.3208	no			
"	0.3290	yes	15	0.005	0.089
"	0.3368	yes	13	0.004	0.080
0.0066	0.1600	no			
"	0.1650	no			
"	0.1700	no			
"	0.1750	yes	15	0.002	0.037
"	0.1800	yes			
0.0033	0.0800	no			
"	0.0900	yes	12	0.000	0.007
"	0.1000	yes			

Table IV-12

Nineteen Channel Bypass Case Two  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{q}$ (MLb/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{max}$	$\alpha_{max}$
0.0132	0.3208	no			
"	0.3369	yes	16	0.006	0.110
0.0066	0.1700	no			
"	0.1750	yes	17	0.003	0.059
0.0033	0.0950	no			
"	0.1000	yes	17	0.000	0.000

Table IV-13  
 Nineteen Channel Bypass Case Three  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{q}$ (Mlb/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{max}$	$\alpha_{max}$
0.0132	0.3368	no			
"	0.3600	no			
"	0.3650	yes	18	0.007	0.127
"	0.3700	yes	17		
"	0.3800	yes	5		
0.0066	0.1900	no			
"	0.1950	yes	17	0.000	0.009
"	0.2000	yes	17		
"	0.2100	yes	16		
0.0033	0.1050	no			
"	0.1100	yes	19	0.000	0.000



Table IV-14  
 Nineteen Channel Bypass Case Four  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{q}$ (Mib/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{\max}$	$\alpha_{\max}$
0.0132	0.2000	no			
"	0.2200	no			
"	0.2500	no			
"	0.2700	no			
"	0.2750	no			
"	0.2800	yes	20	0.010	0.163
"	0.2900	yes	15		
"	0.3290	yes	14	0.005	0.091
"	0.3368	yes	11	0.004	0.081

Table IV-15  
 Nineteen Channel Bypass Cases  
 Limits of COBRA-IIIC/MIT Applicability

Case	$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{G}$ (Mlb/hr-ft <sup>2</sup> )	$X_{\max}$	$\alpha_{\max}$
1	0.0132	0.3290	0.005	0.084
2	0.0132	0.3369	0.006	0.110
3	0.0132	0.3650	0.007	0.127
4	0.0132	0.2800	0.010	0.163
1	0.0066	0.1750	0.002	0.037
2	0.0066	0.1750	0.003	0.059
3	0.0066	0.1950	0.000	0.009
1	0.0033	0.0900	0.000	0.007
2	0.0033	0.1000	0.000	0.000
3	0.0033	0.1100	0.000	0.000

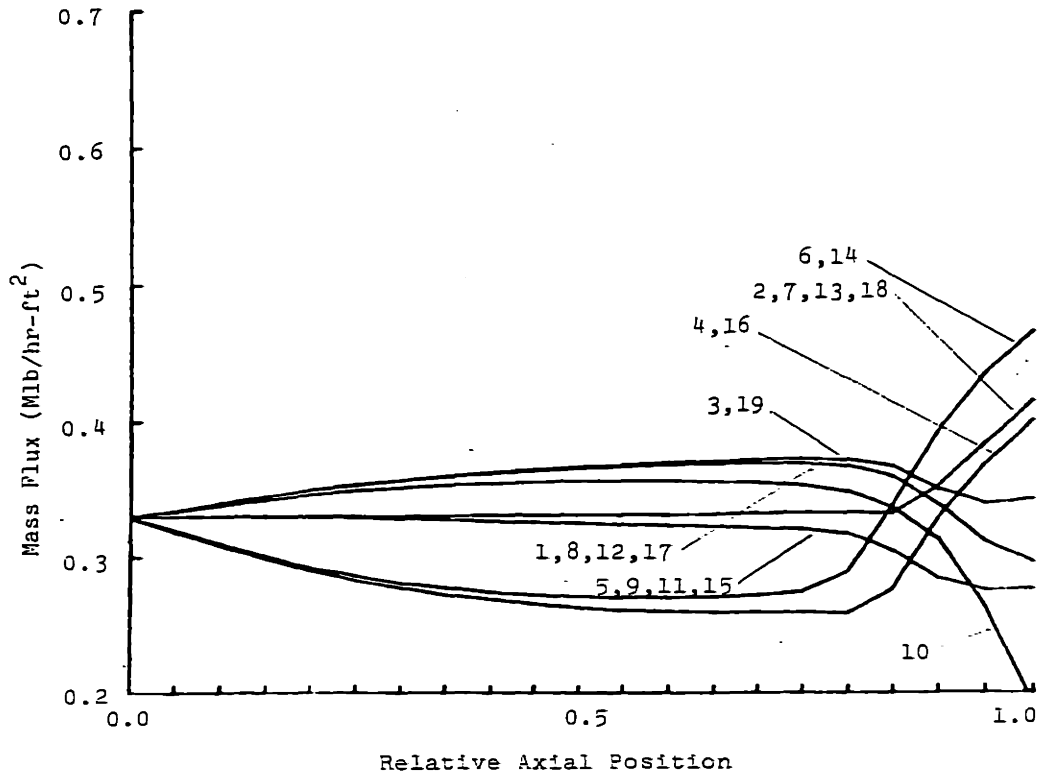


Figure IV-65

Axial Mass Flux Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{F} = 0.3290 \text{ Mlb/hr-ft}^2$$

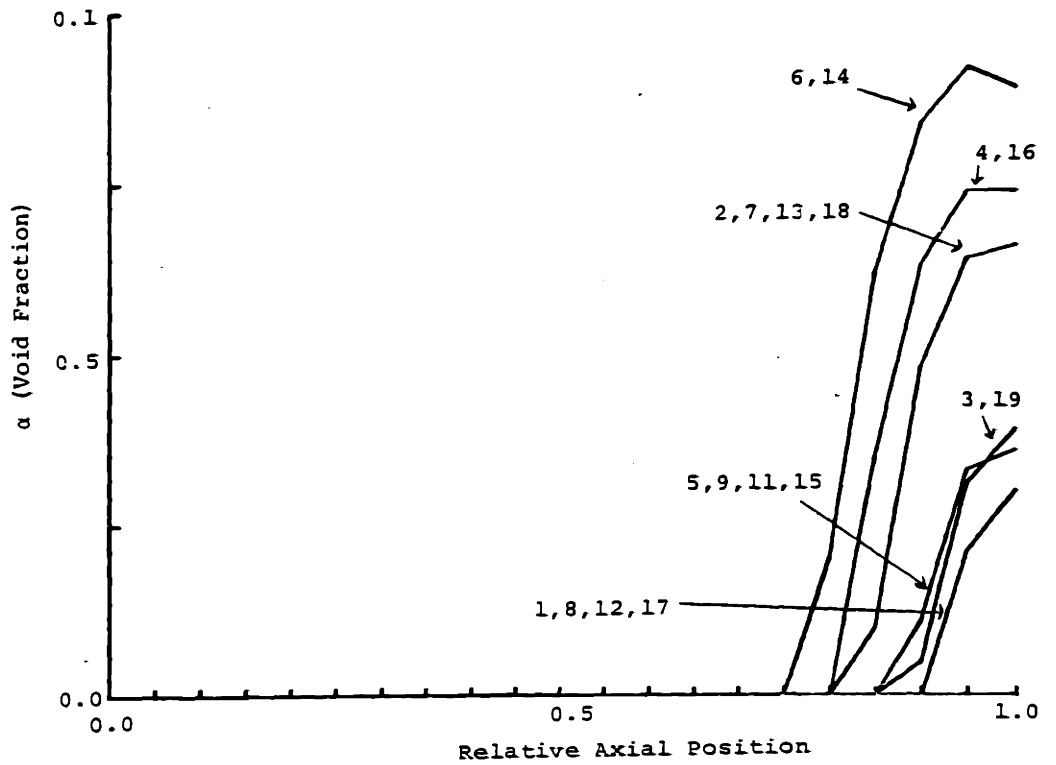


Figure IV-66

Axial Void Fraction Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3290 \text{ Mlb/hr-ft}^2$$

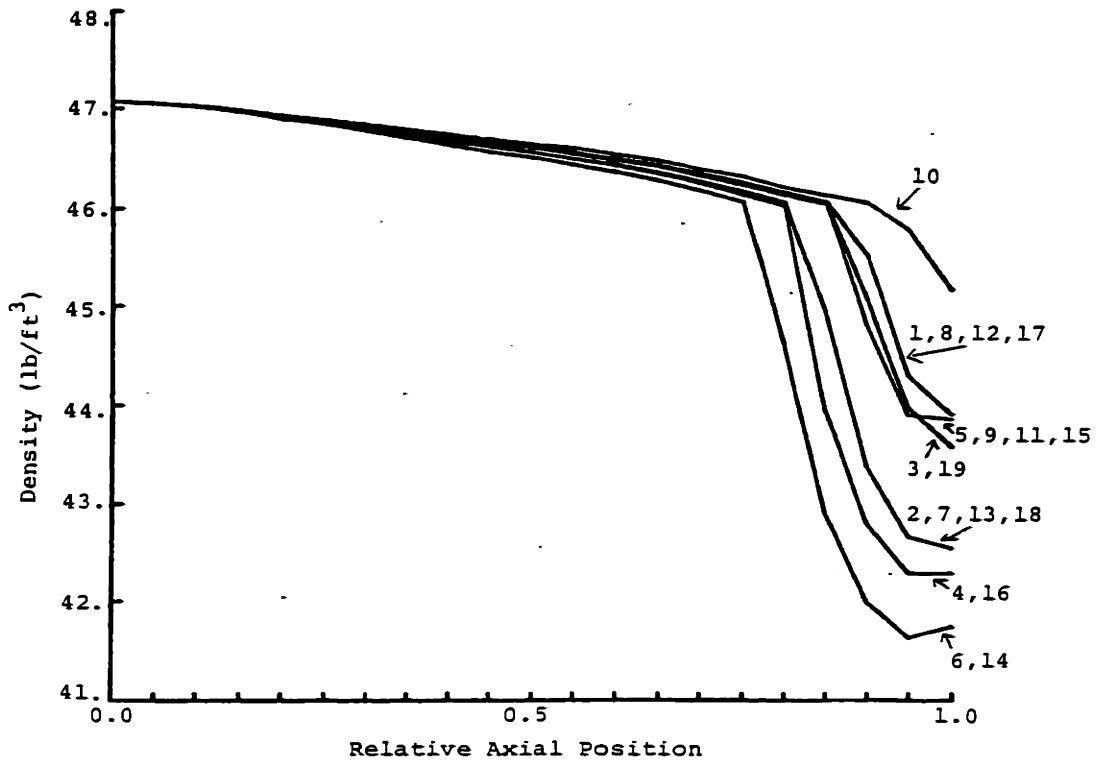


Figure IV-67

Axial Density Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3290 \text{ Mlb/hr-ft}^2$$

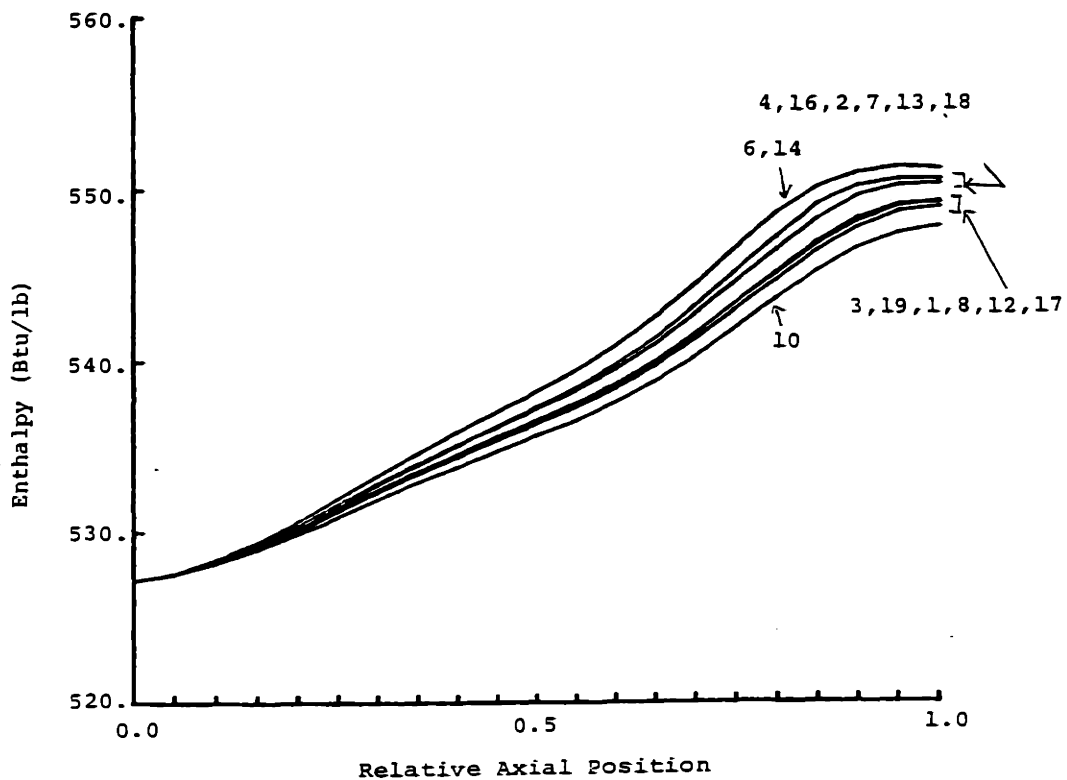


Figure IV-68

Axial Enthalpy Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3290 \text{ Mlb/hr-ft}^2$$

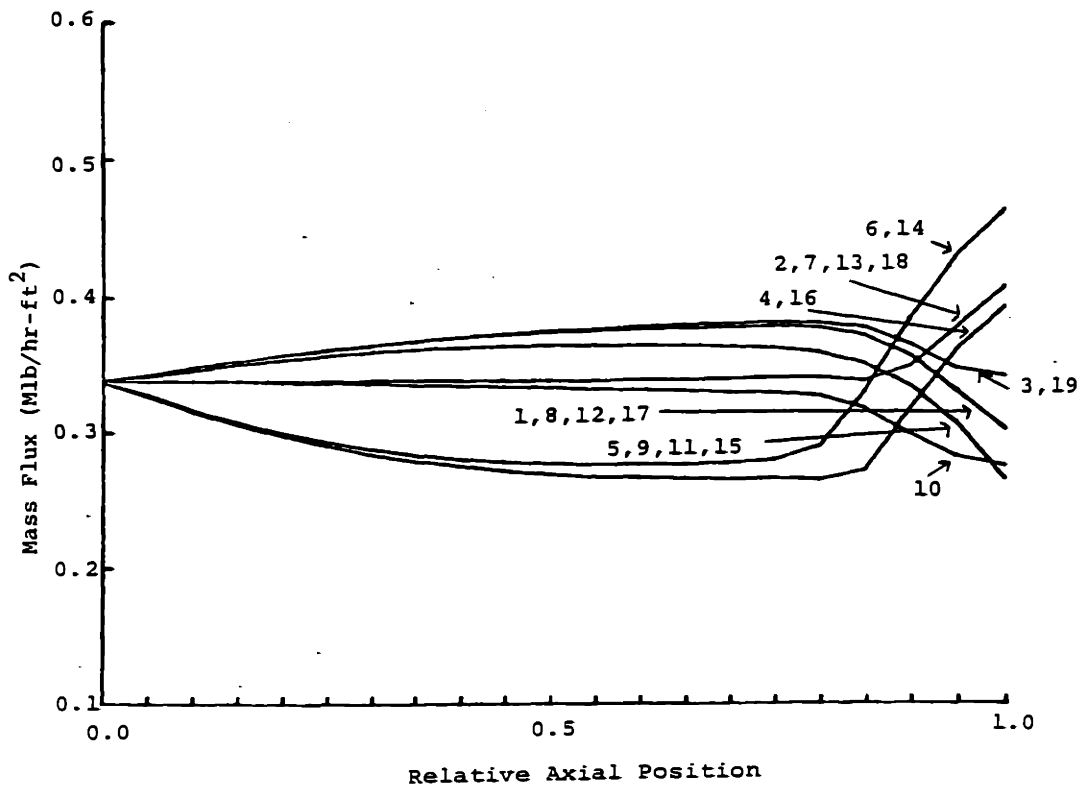


Figure IV-69

Axial Mass Flux Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$$

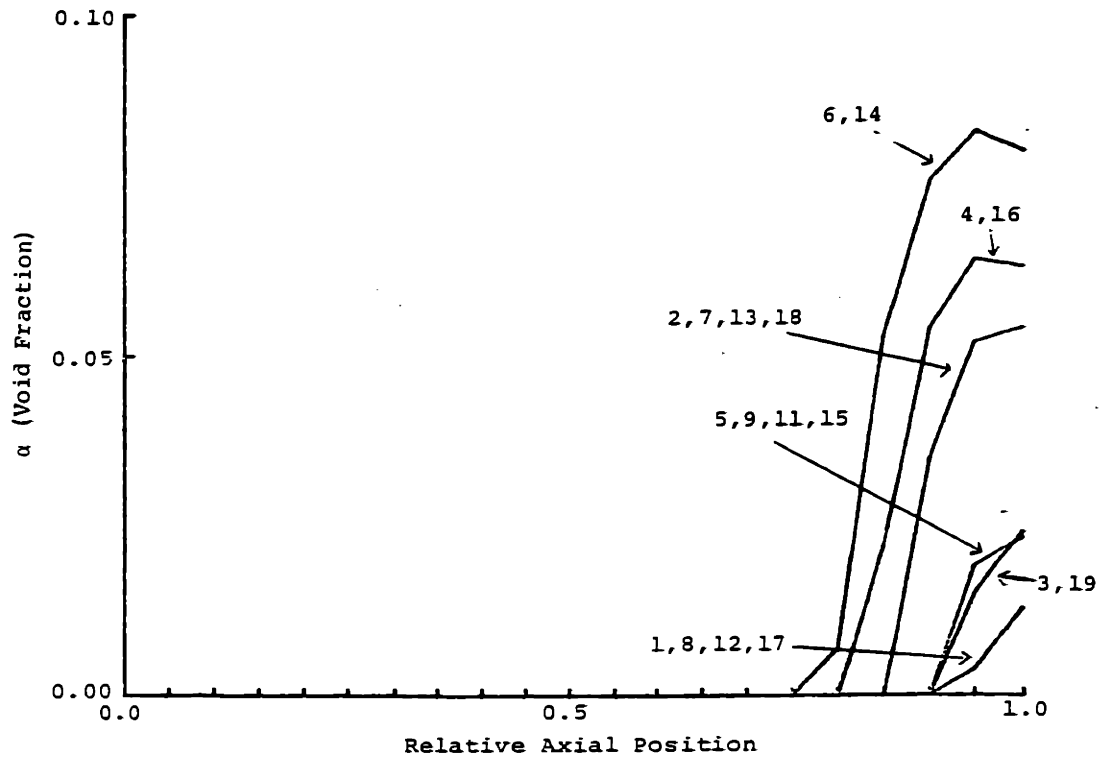


Figure IV-70

Axial Void Fraction Profiles  
 Nineteen Channel Bypass Case One  
 $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$   
 $\bar{G} = 0.3368$



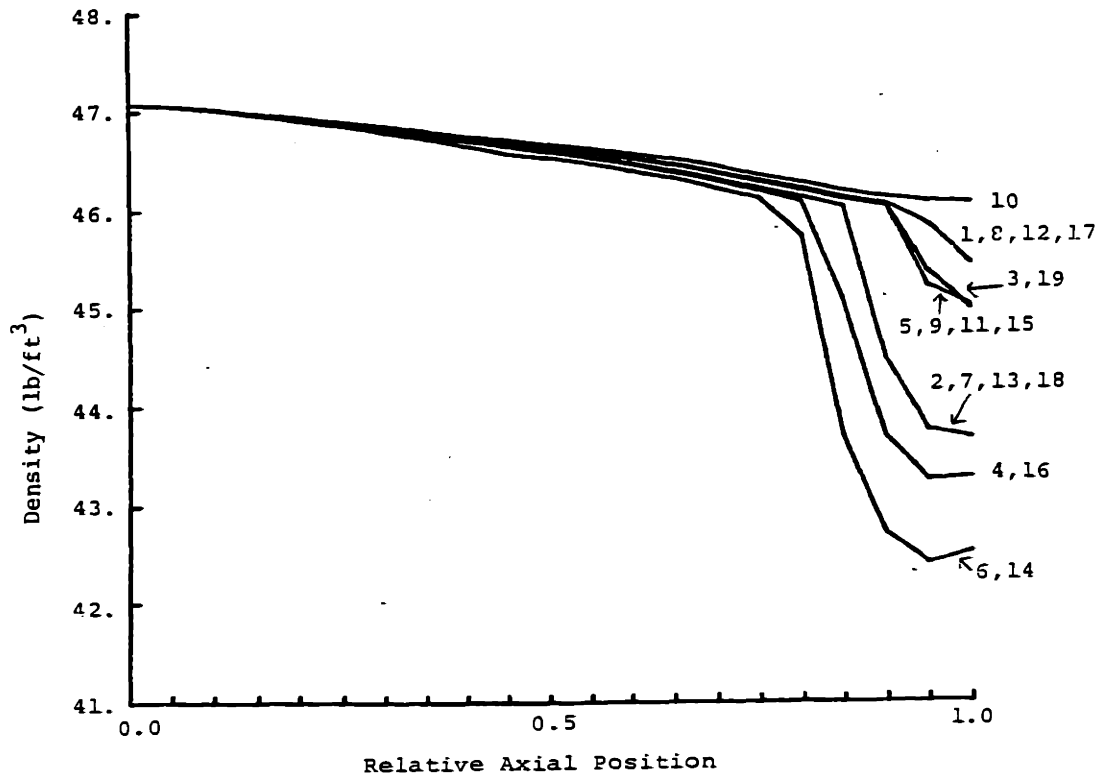


Figure IV-71

Axial Density Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3368$$

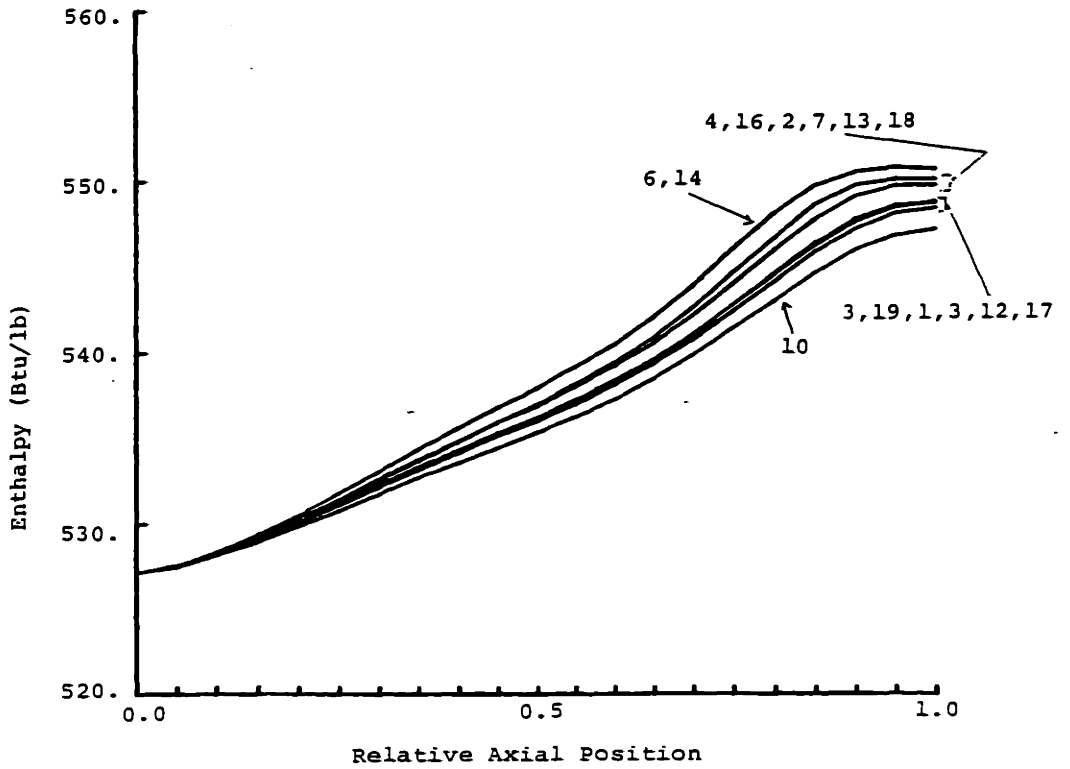


Figure IV-72

Axial Enthalpy Profiles

Nineteen Channel Bypass Case One

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$$

shows. Density in channels 6 and 14 drops very low and then partially recovers as coolant is drawn in from neighboring channels. Axial enthalpy profiles are shown in Fig. IV-68. Channels 6 and 14 have the highest enthalpy while channel 10, a W-W channel has the lowest enthalpy because it is adjacent to assemblies with low radial peaking factors and it has a large hydraulic diameter.

Figures IV-69 through IV-72 correspond to Fig. IV-65 through IV-68 for average mass flux equal to 0.3368 Mlb/hr-ft<sup>2</sup>. At this higher flow rate, the start of boiling occurs at a higher axial position, as seen in Fig. IV-70 which shows axial void fraction profiles. The flow redistribution induced by boiling is less for  $\bar{G} = 0.3368$  than for  $\bar{G} = 0.3290$  since less boiling is present. At  $\bar{G} = 0.3208$  Mlb/hr-ft<sup>2</sup> (and  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup>) COBRA-IIIC/MIT failed to converge. When failure occurred, the gravitational component of pressure drop in channel 10 was 99% of the total pressure drop. In a situation where the stabilizing influence of frictional pressure drop is outweighed by gravitational pressure drop, calculational stability is low. Also, cross-flow is significant in comparison to axial flow, contrary to a basic assumption of the COBRA-IIIC/MIT code. Thus, two factors are present which can cause failure of the COBRA-IIIC/MIT code, even in the absence of reverse flow. Reverse flow is outside the capability of the code's marching solution technique. Failures at lower average heat fluxes (0.0066 and 0.0033 MBtu/hr-ft<sup>2</sup>) also occurred when the gravitational component of pressure drop was at least 99% of the total pressure drop.

#### 2.c.iii Nineteen Channel Bypass Case Two Predictions

Case two has no control rods inserted and uses the old (similar to Carlson and Gott) heat apportionment method. Case two predictions for  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup> and  $\bar{G} = 0.3369$  Mlb/hr-ft<sup>2</sup> are shown in Figs. IV-73 and IV-74. Figure IV-73 shows axial mass flux profiles. Axial mass flux

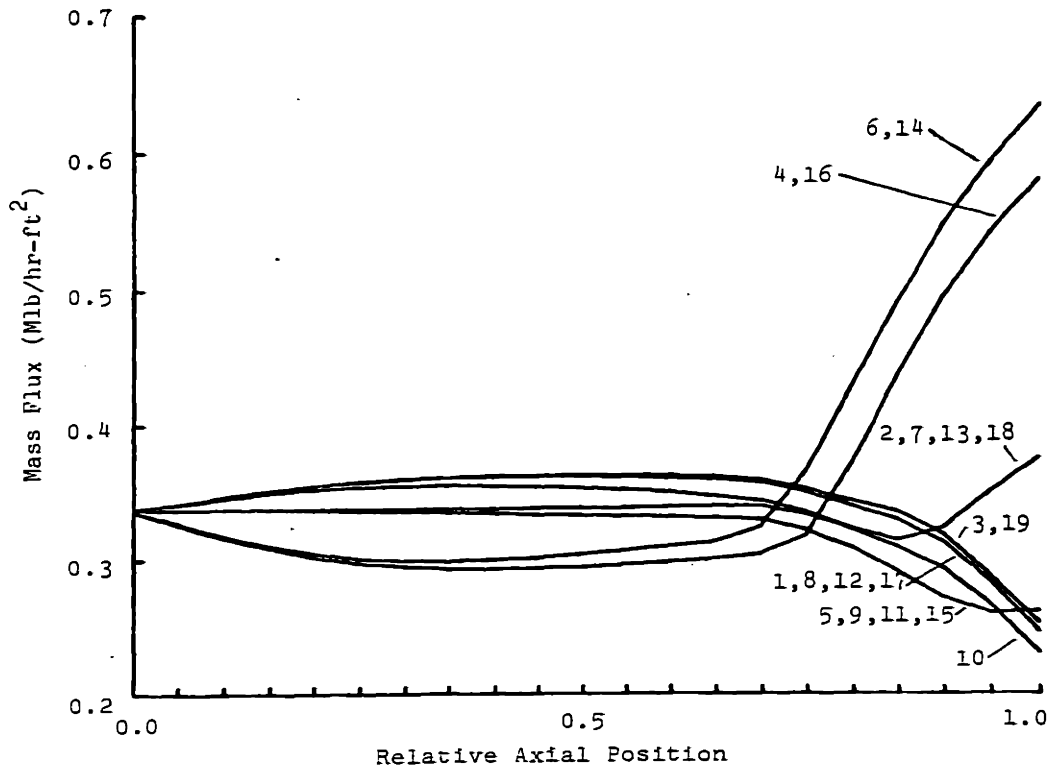


Figure IV-73

Axial Mass Flux Profiles  
 Nineteen Channel Bypass Case Two  
 $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$   
 $\bar{G} = 0.3369 \text{ Mlb/hr-ft}^2$

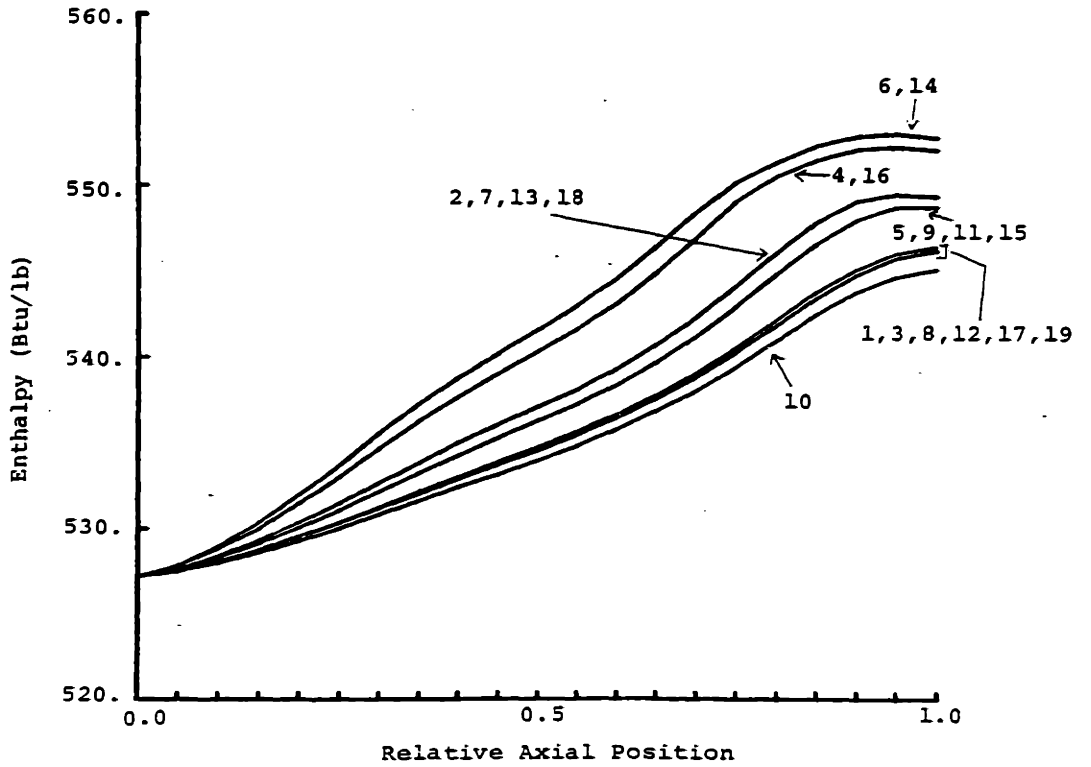


Figure IV-74

Axial Enthalpy Profiles

Nineteen Channel Bypass Case Two

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3369 \text{ Mlb/hr-ft}^2$$

becomes extremely large in N-N channels 6, 14, 4, and 16. The dip in channel 10 mass flux is about the same as in Fig. IV-69. Axial enthalpy profiles are shown in Fig. IV-74. The spread of axial enthalpy profiles is greater than seen for case one in Fig. IV-72. The old power distribution has a greater difference in enthalpy profiles because the old power distribution increases the power-to-flow ratio differences between the channel types (W-W, W-N and N-N). In general, the channels of case two behaved according to channel type. Near bypass channel outlets, flow is diverted from W-W channels to N-N channels via the W-N channels over the axial distance where boiling is present. COBRA-IIIC/MIT failures for case two were similar to those for case one. The gravitational pressure drop was at least 99% of the total pressure drop in channel 10 when failures occurred.

#### 2.c.iv Nineteen Channel Bypass Case Three Predictions

Case three has a control rod inserted approximately halfway into channel ten and uses the old power apportionment method. COBRA-IIIC/MIT predictions are shown in Figs. IV-75, IV-76 and IV-77 for  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup> and  $\bar{G} = 0.3650$  Mlb/hr-ft<sup>2</sup>. Axial mass flux profiles are shown in Fig. IV-75. Mass flux in channel 10 drops when the channel flow area becomes larger near  $x/L = 0.5$  and returns to the higher value in the top half of the bypass channel. Figure IV-76 shows axial enthalpy profiles. Channel 10 has a higher enthalpy relative to the other channels with the control rod inserted than it has without, as may be seen by comparing Figs. IV-76 and IV-74. Channel 10 has higher enthalpy because the mass flow rate is less over most of the axial length with the control rod inserted. Enthalpy profiles in channels other than 10 are nearly the same in Figs. IV-74 and IV-77, indicating that the effect of the control rod is not strongly felt by channels in the vicinity of 10 under the conditions analyzed. Figure IV-77 shows axial density profiles.

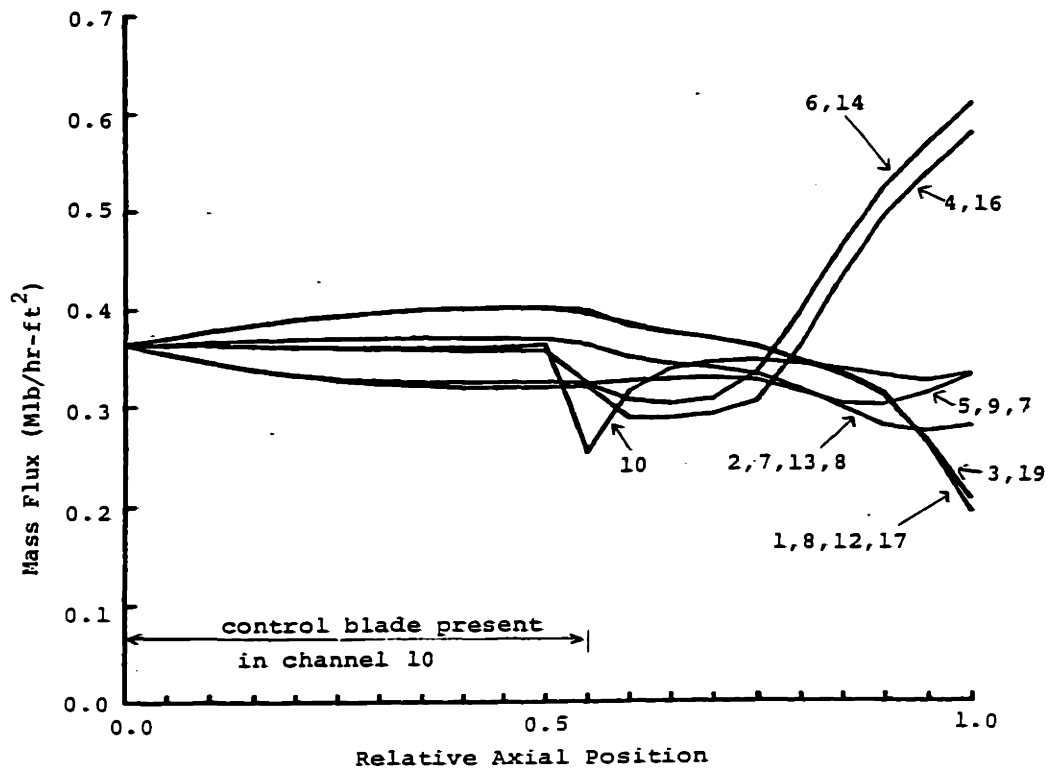


Figure IV-75

Axial Mass Flux Profiles

Four Channel Bypass Case Three

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.1668 \text{ Mlb/hr-ft}^2$$

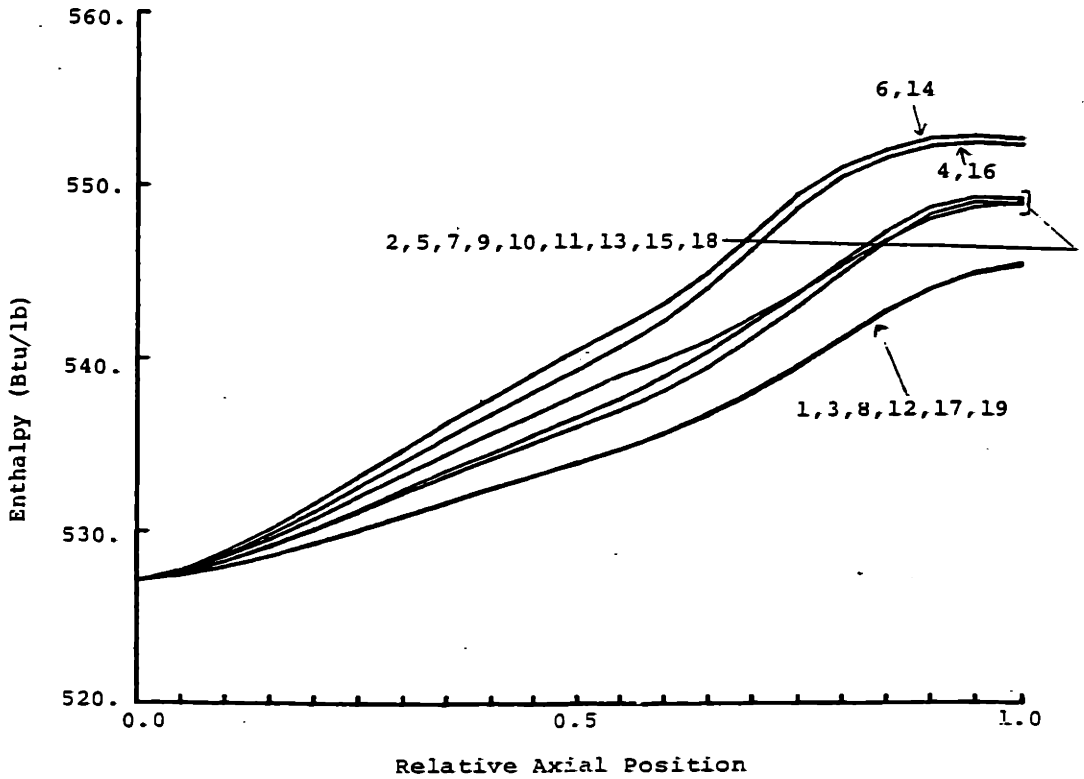


Figure IV-76

Axial Enthalpy Profiles

Nineteen Channel Bypass Case Three

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3650 \text{ Mlb/hr-ft}^2$$



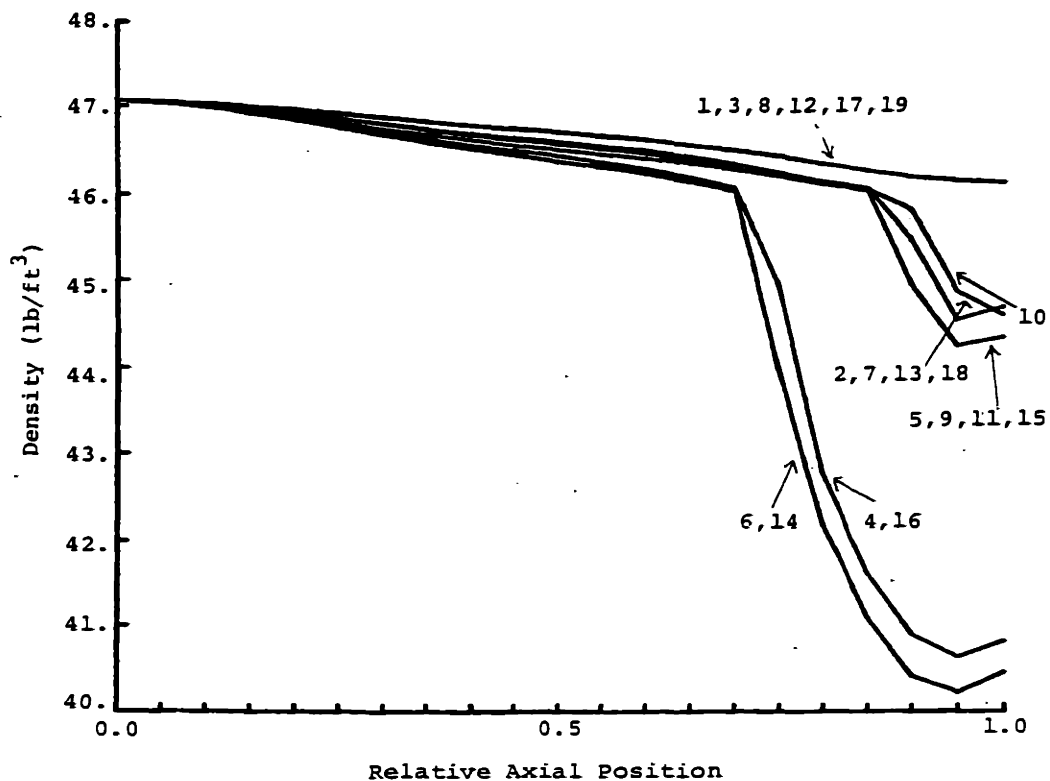


Figure IV-77

Axial Density Profile

Nineteen Channel Bypass Case Three

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3650 \text{ Mlb/hr-ft}^2$$

Boiling in channel 10 is indicated by a sharp dip in density. Case three failures occurred in the same way as for cases one and two.

#### 2.d Summary of Nineteen Channel Case Results

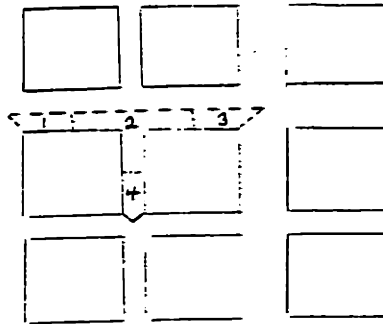
The nineteen channel case results are summarized as follows:

- COBRA-IIIC/MIT limits of applicability are similar for all four cases. Estimated full power and full flow conditions are outside the zone of COBRA-IIIC/MIT applicability.
- COBRA-IIIC/MIT failed when the gravitational component of pressure drop comprised at least 99% of the total pressure drop. This situation is conducive to recirculation and flow instability.
- Channels behaved according to bypass channel type (W-W, W-N and N-N). Near bypass channel outlets, flow was diverted from W-W channels to N-N channels via the W-N channels when boiling occurred.

### 3. Four Channel Test Case

#### 3.a Description of Four Channel Test Case Geometry and Operating Conditions

The four channel bypass case geometry represents the bypass flow region associated with two assembly halves, as shown in Fig. IV-78. The four channel geometry includes all three bypass channel types, W-W, W-N and N-N. Channels 1 and 2 are 1/8 sections of W-W channels. Channel 3 is a 1/2 section of a W-N channel. Channel 4 is a 1/4 section of a N-N channel. No control rod is inserted. The radial power distribution is flat. The same inlet enthalpy and system pressure used for the nineteen channel bypass cases were also used for the four channel test case.



<u>Channel</u>	<u>Area (in<sup>2</sup>)</u>	<u>Wetted Perimeter (in)</u>	<u>Heated Perimeter (in)</u>
1	1.090	2.719	2.719
2	3.195	10.880	10.880
3	1.090	2.719	2.719
4	1.052	5.438	5.438

Figure IV - 78

Four Channel Bypass Case Geometry

### 3.b Description of Modeling

The four channel bypass case was analyzed using COBRA-IIIC/MIT and THERMIT. All analyses used the new heat apportionment method. All analyses were made using the axial heat flux profile shown previously in Fig. IV-61. The four channel case was analyzed by COBRA-IIIC/MIT using the same options used for the nineteen channel bypass cases (see Table IV-10). Sixty axial nodes were again used for COBRA-IIIC/MIT analysis.

THERMIT represented the four channel case using the geometry shown in Fig. IV-79. THERMIT analysis used the options listed in Table IV-16. The axial length was divided into ten axial nodes to keep THERMIT computational time within reasonable limits. "Steady state" THERMIT predictions were obtained by running a transient calculation for two time periods. No transverse flow was calculated during the first time period. Then the calculation was restarted with reduced time step size and calculation of transverse flow. THERMIT analysis of the four channel case used boundary conditions similar to those used by COBRA. Inlet flows and outlet pressures were specified. Outlet pressures were equal. Inlet flows were equal. A pressure drop boundary condition would be a better approach. The physical situation has factors which are potential sources of hydrodynamic instability. Voiding is occurring near the outlets of parallel channels and the flow rate is low. The destabilizing factors are counterbalanced to some extent by the presence of flow restrictions near the channel inlets. However, this stabilizing factor was not accounted for in the THERMIT analysis since a pressure drop boundary condition was not used. A pressure drop boundary condition requires specification of loss coefficients for the inlet orifices.

IV-122

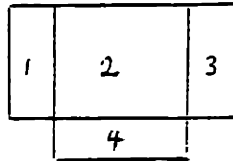


Figure IV-79

THERMIT Four Channel Bypass Case Representation

Table IV-16THERMIT Options Used for Bypass Analysis

## Thermal-Hydraulic Options In Use

Pressure boundary condition at core exit  
 Velocity boundary condition at core inlet  
 Nigratulin boiling model  
 MIT interfacial momentum exchange model  
 Constant property fuel pin model  
 Steady-state heat transfer calculation without CHF check  
 Constant heat flux boundary condition  
 Bias: critical heat flux correlation  
 Transverse friction model = Guntter-Snow  
 Transverse velocity used in transverse momentum calculations  
 Transverse flow is not calculated  
 No mixing model  
 Gravitational constant = -9.81000  
 Transverse hydraulic diameter = 0.30000E-01  
 Transverse friction multiplier = 0.50000E-01

## Friction Model

Axial  $f = 0.184 \cdot Re^{-0.200}$   
 Transverse  $f = 1.920 \cdot Re^{-0.145}$   
 Grid spacer  $k = 3.000 \cdot Re^{-0.100}$

## Iteration Control Parameters

Dump indicator (0/1)(no/yes) = 1  
 Max number of Newton iterations = 3  
 Max number of inner iterations = 20  
 Convergence crit. for Newton iter = 0.10000E-06  
 Convergence crit. for inner iter = 0.10000E-06

### 3.c Analysis Case Results

#### 3.c.i COBRA-IIIC/MIT Applicability Limits

COBRA-IIIC/MIT applicability limits were found for the four channel bypass case by varying power and flow rates. The most adverse combinations of power and flow for which COBRA four channel bypass case predictions were obtained are included in Fig. IV-80. COBRA results can be obtained over a wider range of conditions for the four channel case than for the nineteen channel cases. The greater geometrical simplicity of the four channel case is a factor which helps give COBRA the capability to analyze a wider range of conditions. Four channel and nineteen channel applicability limits become closer at low average heat fluxes. Table IV-17 lists four channel analyses made during determination of four channel case applicability limits. The particular four channel analyses contained in the limit line of Fig. IV-80 are described in Table IV-18.

#### 3.c.ii Four Channel Bypass Case Predictions

Four channel bypass case predictions were obtained using COBRA-IIIC/MIT and THERMIT. COBRA-IIIC/MIT predictions are shown in Figs. IV-81 through IV-87. THERMIT predictions are shown in Figs. IV-88 and IV-89. COBRA predictions will be discussed first. Predictions for  $\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$  and  $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$  are shown in Figs. IV-81, IV-82 and IV-83. Axial mass flux profiles are shown in Fig. IV-81. The profiles are close to the predictions obtained for the nineteen channel bypass case one under the same  $\bar{G}$  and  $\bar{q}''$ . Compare Fig. IV-81 with IV-65. The profiles reflect the behavior of different bypass channel types. Similarly, density and enthalpy profiles predicted for the four channel case match the predictions for different channel types obtained during nineteen channel bypass case one analysis. The similarity is seen by comparing Figs. IV-82 and IV-67 and Figs. IV-83 and IV-68. Thus, the four channel predictions

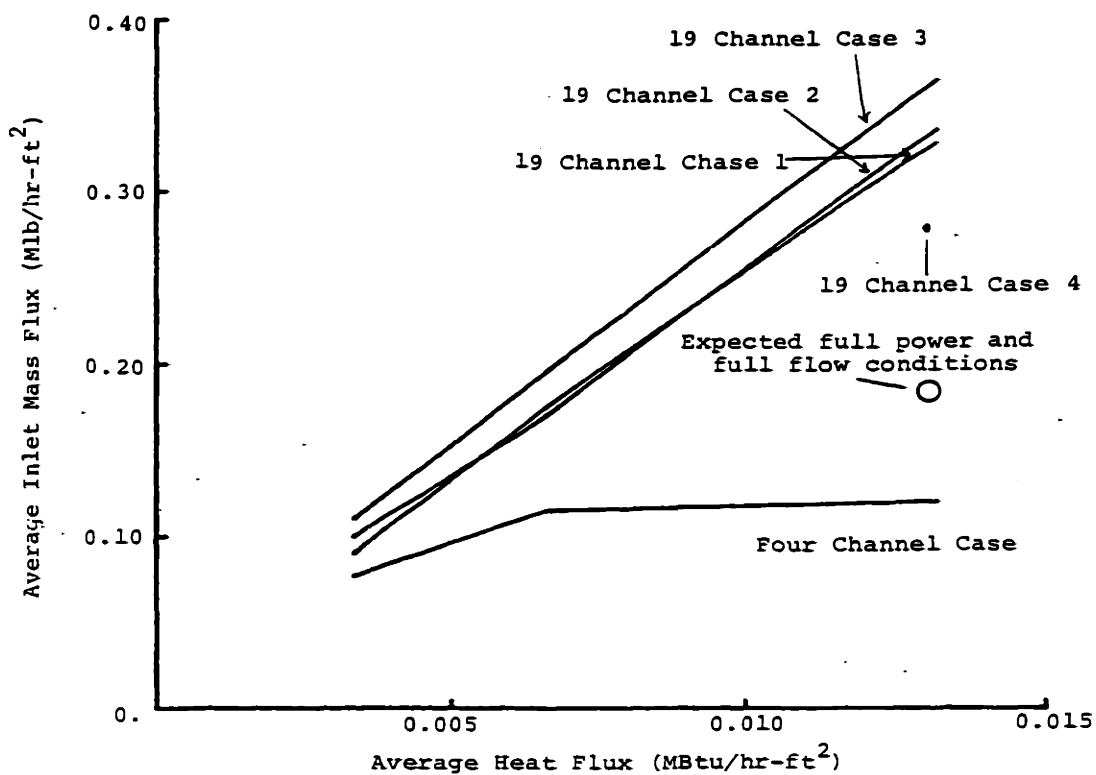


Figure IV-80

Mass Flux vs. Heat Flux

COBRA-IIIC/MIT Applicability Limits



Table IV-17

Pour Channel Bypass Case  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{g}$ (MB/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{max}$	$\alpha_{max}$
0.0132	0.0900	no			
"	0.1000	no			
"	0.1050	no			
"	0.1100	no			
"	0.1150	no			
"	0.1200	yes	19	0.065	0.577
"	0.1400	yes	15		
"	0.1700	yes	11		
"	0.2000	yes	12		
"	0.3290	yes	12	0.005	0.088
"	0.3368	yes	11	0.004	0.078
0.0066	0.1000	no			
"	0.1100	no			
"	0.1150	yes	17	0.018	0.268
"	0.1200	yes	15		
"	0.1400	yes	13		
"	0.1800	yes	10		

Table IV-17 (continued)

Four Channel Bypass Case  
 COBRA-IIIC/MIT Analysis Results

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{g}$ (Mlb/hr-ft <sup>2</sup> )	Success ?	Iterations	$X_{max}$	$\alpha_{max}$
0.0033	0.0700	no			
"	0.0750	no			
"	0.0775	yes	18	0.005	0.088
"	0.0800	yes	15		
"	0.0900	yes	9		

Table IV-18  
Four Channel Bypass Case  
Limits of COBRA-IIIC/MIT Applicability

$\bar{q}''$ (MBtu/hr-ft <sup>2</sup> )	$\bar{G}$ (Mlb/hr-ft <sup>2</sup> )	$X_{\max}$	$\alpha_{\max}$
0.0132	0.1200	0.065	0.577
0.0066	0.1150	0.018	0.268
0.0033	0.0775	0.005	0.088

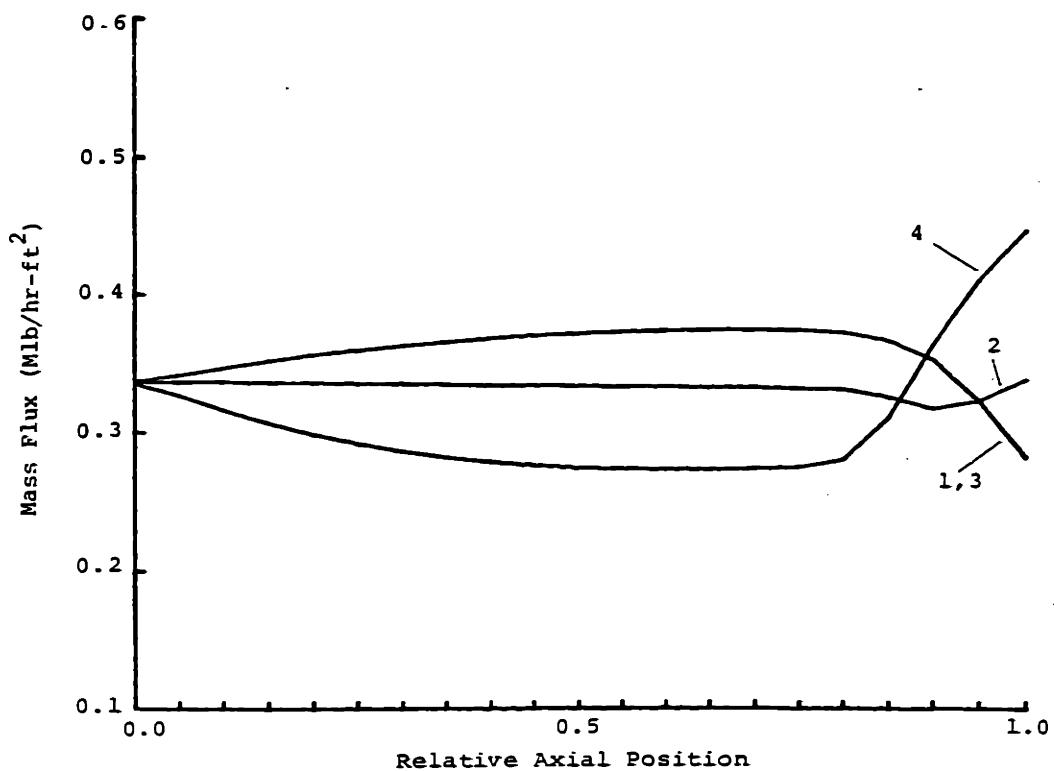


Figure IV-81

Axial Mass Flux Profiles

Four Channel Bypass Case

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$$

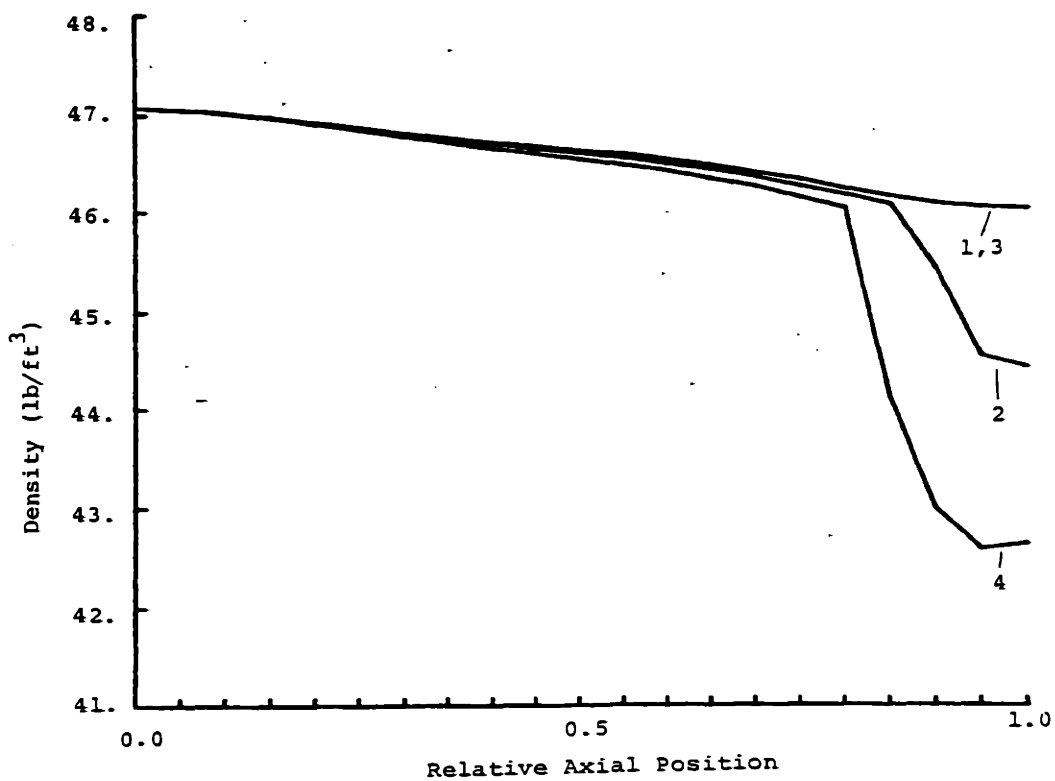


Figure IV-82

Axial Density Profiles  
Four Channel Bypass Case

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$$

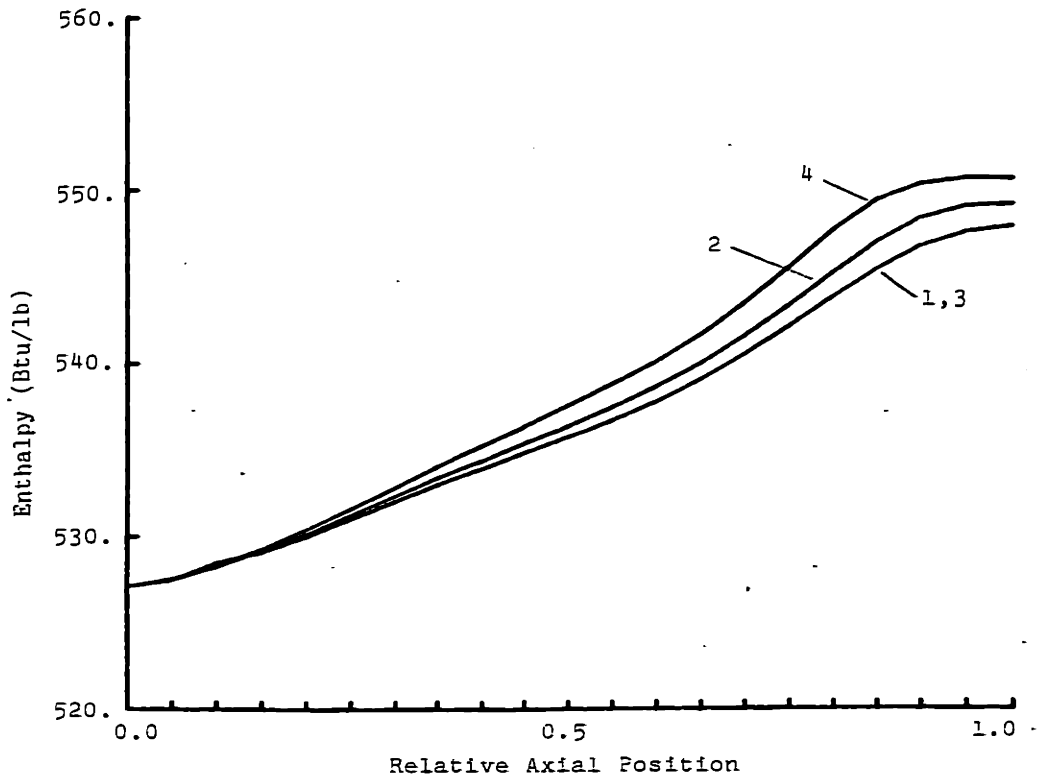


Figure IV-83

Axial Enthalpy Profiles  
Four Channel Bypass Case  
 $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$   
 $\bar{G} = 0.3368 \text{ Mlb/hr-ft}^2$

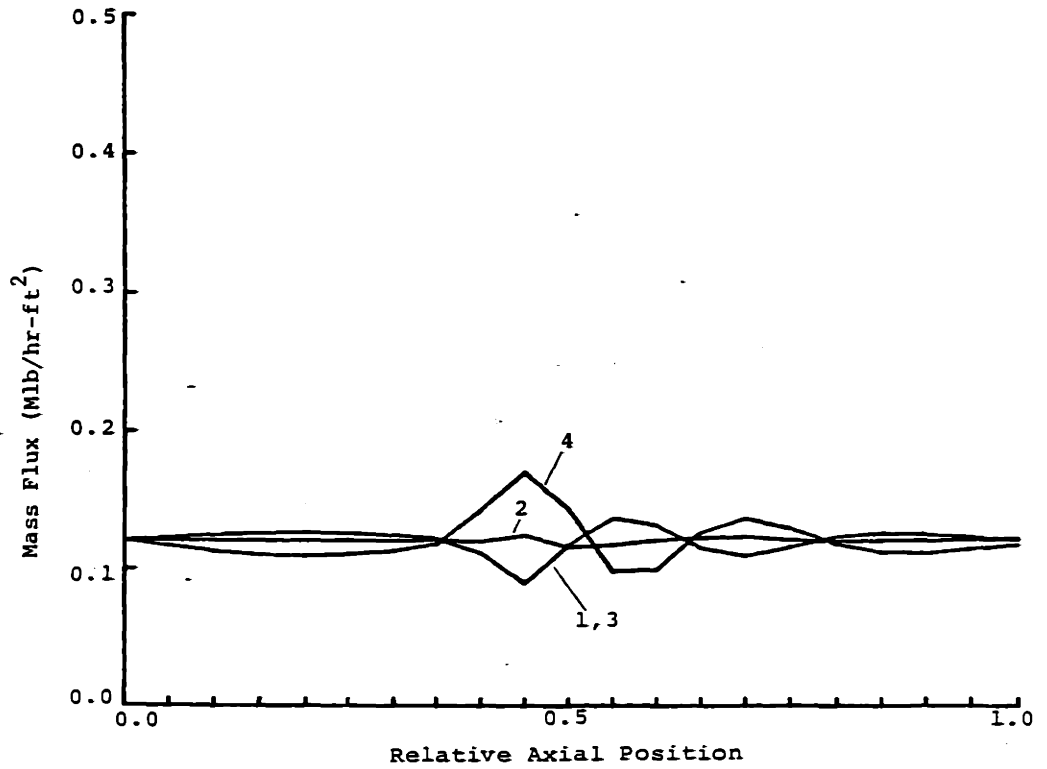


Figure IV-84

Axial Mass Flux Profiles

Four Channel Bypass Case

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.1200 \text{ Mlb/hr-ft}^2$$

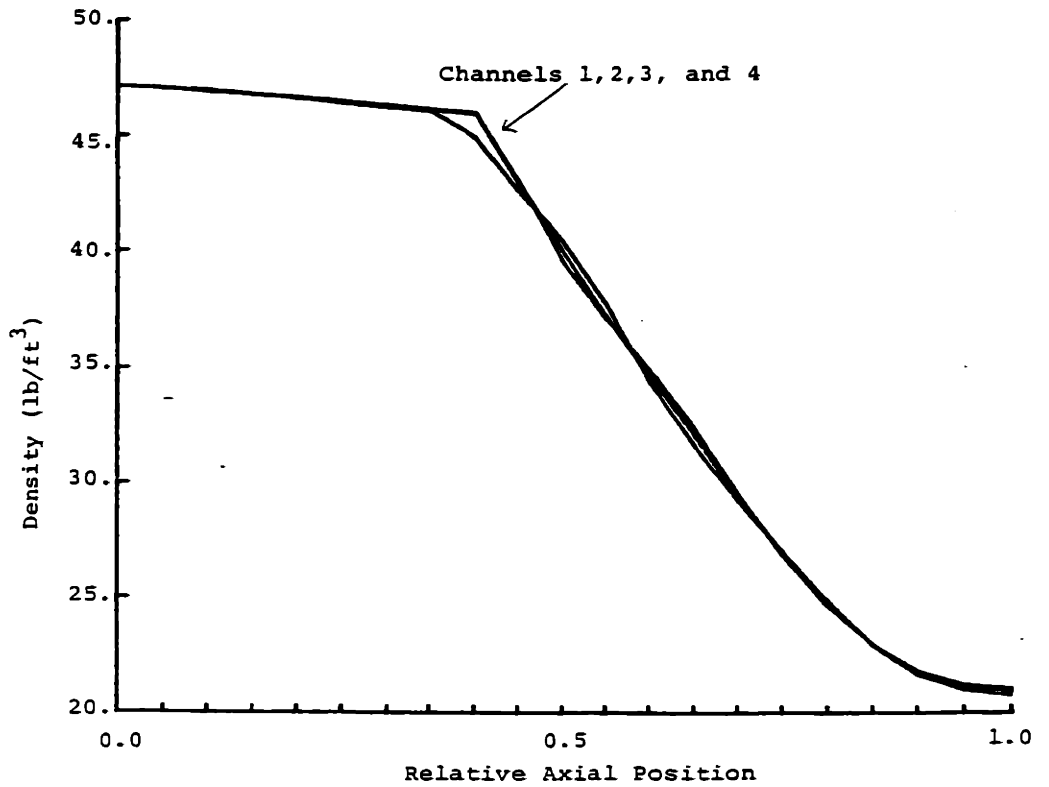


Figure IV-85

Axial Density Profiles

Four Channel Bypass Case

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.1200 \text{ Mlb/hr-ft}^2$$



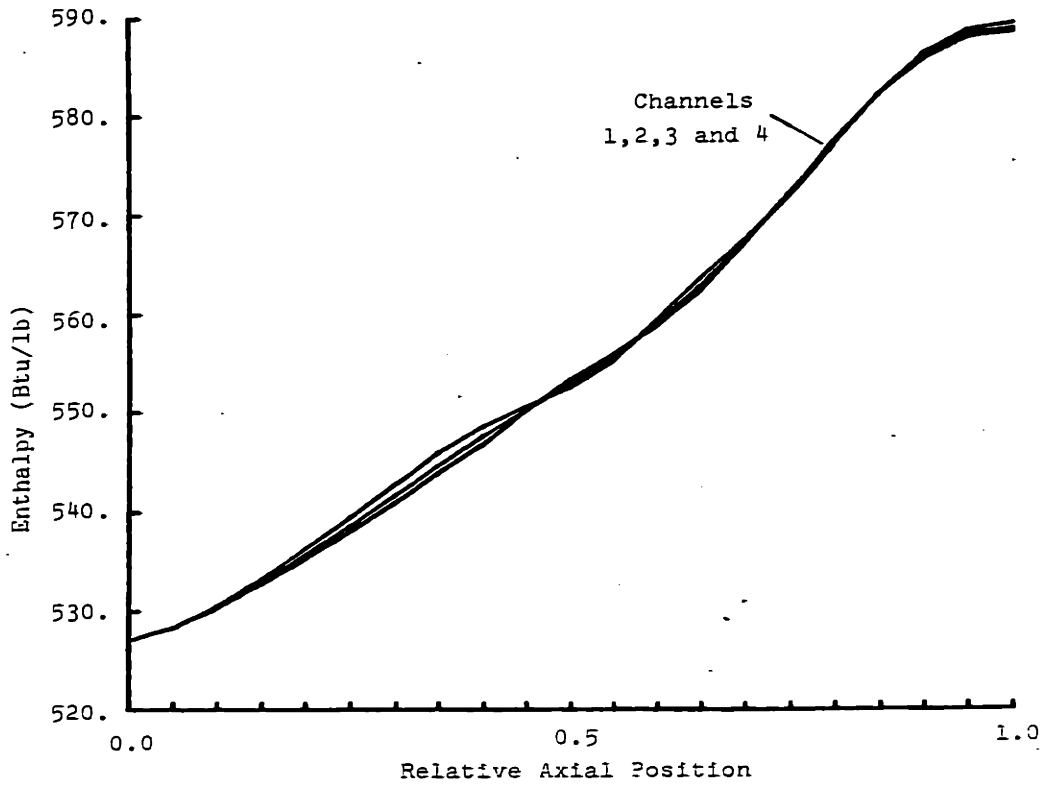


Figure IV-36

Axial Enthalpy Profiles  
Four Channel Bypass Case  
 $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$   
 $\bar{G} = 0.1200 \text{ Mlb/hr-ft}^2$

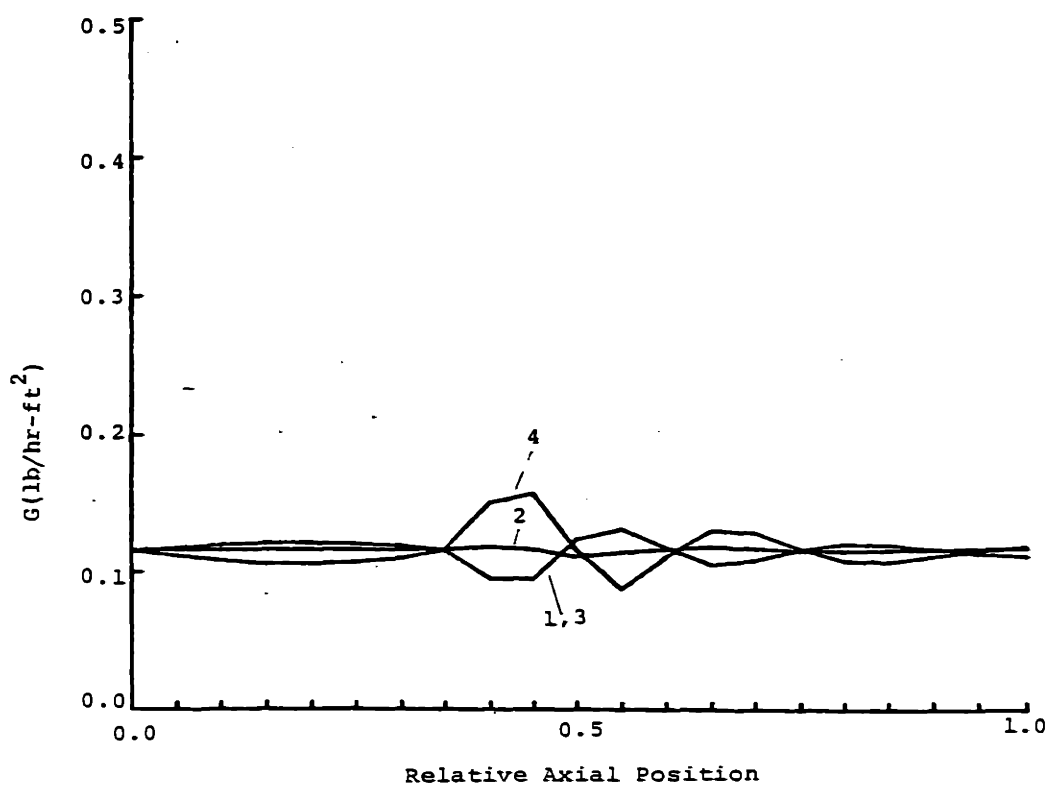


Figure IV-87

Axial Mass Flux Profiles

Four Channel Bypass Case

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.1150 \text{ Mlb/hr-ft}^2$$

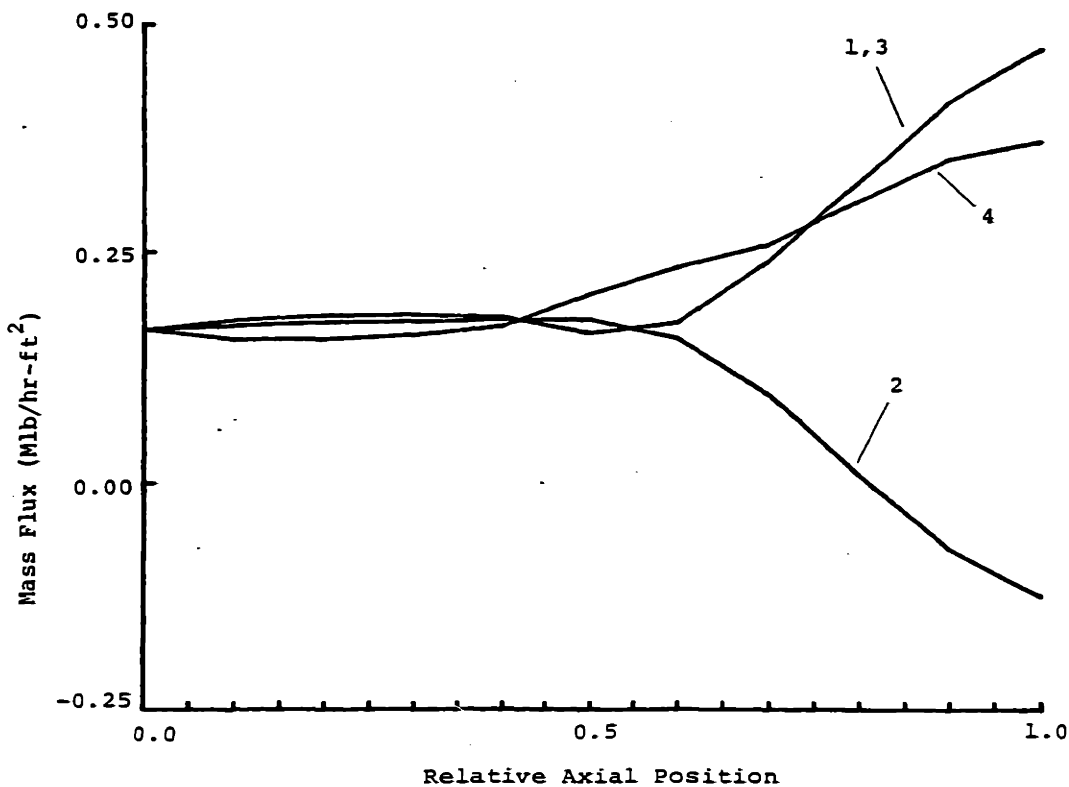


Figure IV-88

Axial Mass Flux Profiles

Four Channel Bypass Case

THERMIT Predictions

$$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$$

$$\bar{G} = 0.1668 \text{ Mlb/hr-ft}^2$$

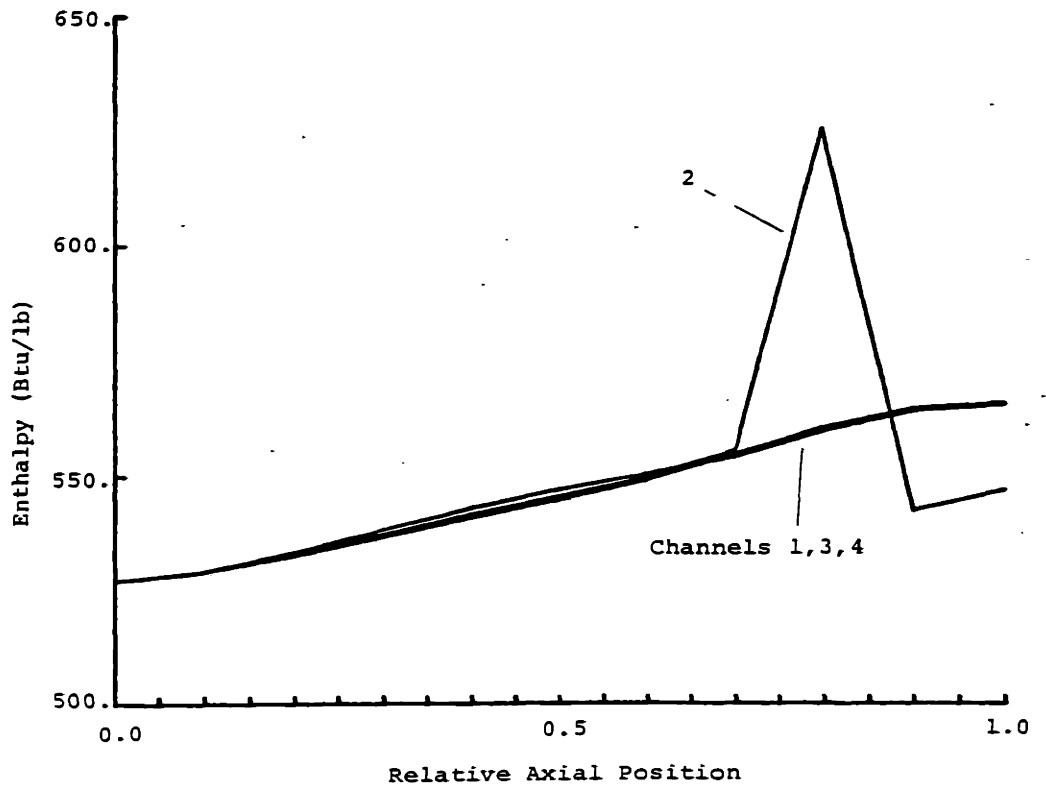


Figure IV-89

Axial Enthalpy Profiles  
Four Channel Bypass Case  
THERMIT Predictions  
 $\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$   
 $\bar{G} = 0.1668 \text{ Mlb/hr-ft}^2$

are similar to the nineteen channel case one analysis predictions for  $\bar{G} = 0.3368$  Mlb/hr-ft<sup>2</sup> and  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup>.

COBRA four channel bypass case predictions for  $\bar{G} = .1200$  Mlb/hr-ft<sup>2</sup>, the lowest flow rate for which predictions could be obtained with  $\bar{q}'' = 0.0132$  MBtu/hr-ft<sup>2</sup>, are shown in Figs. IV-84, IV-85 and IV-86. Axial mass flux profiles are shown in Fig. IV-84. These profiles are much different than those observed in other COBRA predictions. The start of boiling near  $x/L = 0.4$  produces flow oscillations that damp out by the end of the channels. The axial density and enthalpy profiles, shown in Figs. IV-85 and IV-86 are also much different than those observed in other COBRA predictions. The density and enthalpy predictions indicate that the channels are thermally communicating much more than previously observed in other bypass analysis predictions. The oscillating crossflow between channels 1 and 3 and channel 4 via channel 2 implied by the axial mass flux profiles shown in Fig. IV-84 is providing mixing between channels. Figure IV-87 shows axial mass flux profiles for  $\bar{G} = 0.1150$  Mlb/hr-ft<sup>2</sup>, a low rate at which COBRA failed to converge. The profiles are similar to those seen in Fig. IV-84 for  $\bar{G} = 0.1200$  Mlb/hr-ft<sup>2</sup>. The oscillation of flow along the axial direction is evidence of an unstable situation which is causing COBRA to fail. COBRA handles some degree of instability for the four channel geometry which it may not be able to tolerate for the nineteen channel geometry.

THERMIT predictions for the four channel case were obtained by performing transient calculations, as described earlier. When THERMIT calculations were stopped, the energy error was going from positive to negative, indicating a buildup of energy within the bypass channels, as would be expected when a stagnant pocket of vapor is building up within the four channels. THERMIT predictions for the four channel bypass case are shown in Figs. IV-88 and IV-89 for

$\bar{q}'' = 0.0132 \text{ MBtu/hr-ft}^2$  and  $\bar{G} = 0.1668 \text{ Mlb/hr-ft}^2$ . This power-flow combination is close to the estimated full power conditions. Lack of stability was seen in residual variations of THERMIT predictions with time. The transient THERMIT calculations were stopped while the solution was still varying, so the predictions shown in the figures are not actually steady state predictions. Axial mass flux profiles are shown in Fig. IV-88. Flow becomes negative in the upper half of channel 2. Liquid travels downward and vapor travels upward, resulting in a negative net mass flux in channel 2. Axial enthalpy profiles shown in Fig. IV-89 indicate a buildup of vapor near  $x/L = 0.75$  in channel 2. Since the THERMIT predictions shown in the figures are not actually steady state predictions, they have limited usefulness for comparison with COBRA-IIIC/MIT results. However, it is useful to compare general prediction trends. Comparison of Figs. IV-88 and IV-84 and Figs. IV-89 and IV-85 allows a rough comparison between THERMIT and COBRA predictions. The comparison is rough because different average mass fluxes are used in the THERMIT and COBRA predictions being compared. Similar general trends in THERMIT and COBRA predictions are seen in the predictions up to near  $x/L = 0.5$ . In the upper half, beyond  $x/L = 0.5$ , where THERMIT predictions show recirculation, there is no similarity between THERMIT and COBRA predictions.

### 3.d Summary of Four Channel Case Results

- COBRA-IIIC/MIT had a wider range of applicability for the four channel case than for any of the nineteen channel cases. However, predictions in the extended range of applicability showed signs of instability which were not assessed using THERMIT predictions since actual steady state THERMIT predictions within the extended range were not available.

- COBRA-IIIC/MIT four channel predictions were similar to nineteen channel case one predictions within the range of applicability for nineteen channels.
- THERMIT predictions in the vicinity of estimated full power conditions showed recirculation, oscillations, and countercurrent flow. A pressure drop boundary condition would be better than the boundary conditions used for THERMIT analysis.

4. Summary of Bypass Analysis Results and Recommendations for Future Work

The results of bypass analysis and recommendations for future work are summarized in the following:

- COBRA-IIIC/MIT cannot analyze bypass flow for some power and flow combinations.
- Estimated full power and full flow conditions are outside the COBRA-IIIC/MIT zone of applicability. Plugging of holes in the core plate has reduced the bypass flow through the E.I. Hatch reactor. Carlson and Gott analyses did not take into account the reduced bypass flow. Other operational conditions are likely to also be outside the zone of COBRA-IIIC/MIT applicability.
- THERMIT predictions indicate a possibility of some recirculation under full power and full flow conditions. THERMIT analysis should be performed using a pressure drop boundary condition rather than specified inlet flows and outlet pressures. Also, the variable axial nodalization capability of THERMIT should be used to give finer axial nodalization to the upper portion of the bypass channels and coarser axial nodalization to the lower portion. The upper portion of bypass channels can have boiling and consequently, large changes in density and large

crossflows. Variation of axial node size allows efficient allocation of computational resources.

The first two items will be discussed using Fig. IV-90 which contains one COBRA-IIIC/MIT limit of applicability line and a point indicating approximate full power and full flow conditions. The applicability limit line is selected where it is expected to be for analysis. The nineteen channel case three limit line is selected as the COBRA-IIIC/MIT limit line because a realistic case can have some rods inserted. The other three nineteen channel case gave no consideration to rod insertion and thus had larger zones of COBRA-IIIC/MIT applicability. The COBRA-IIIC/MIT four channel case limit of applicability line is not used in the figure because the four channel predictions in the extended zone of applicability looked questionable. The full power and full flow conditions are in the zone where COBRA-IIIC/MIT fails. Since expected bypass conditions are in the zone where COBRA-IIIC/MIT fails, another code having greater capabilities is needed. THERMIT is a code which offers greater capabilities.

THERMIT predictions were obtained using specified inlet velocities and outlet pressures as boundary conditions. During the attempt to obtain steady state predictions using a transient calculation, variations in predictions indicated hydrodynamic instability. For the purposes of better representation of the physical situation, which includes taking into account stabilizing factors, a pressure drop boundary condition should be used by THERMIT. Orifices at the inlets of the bypass channels should be represented in THERMIT analysis.

In summary, a code such as THERMIT, having the capability to use a pressure drop boundary condition, is needed for bypass analysis of the E.I. Hatch Unit I reactor.



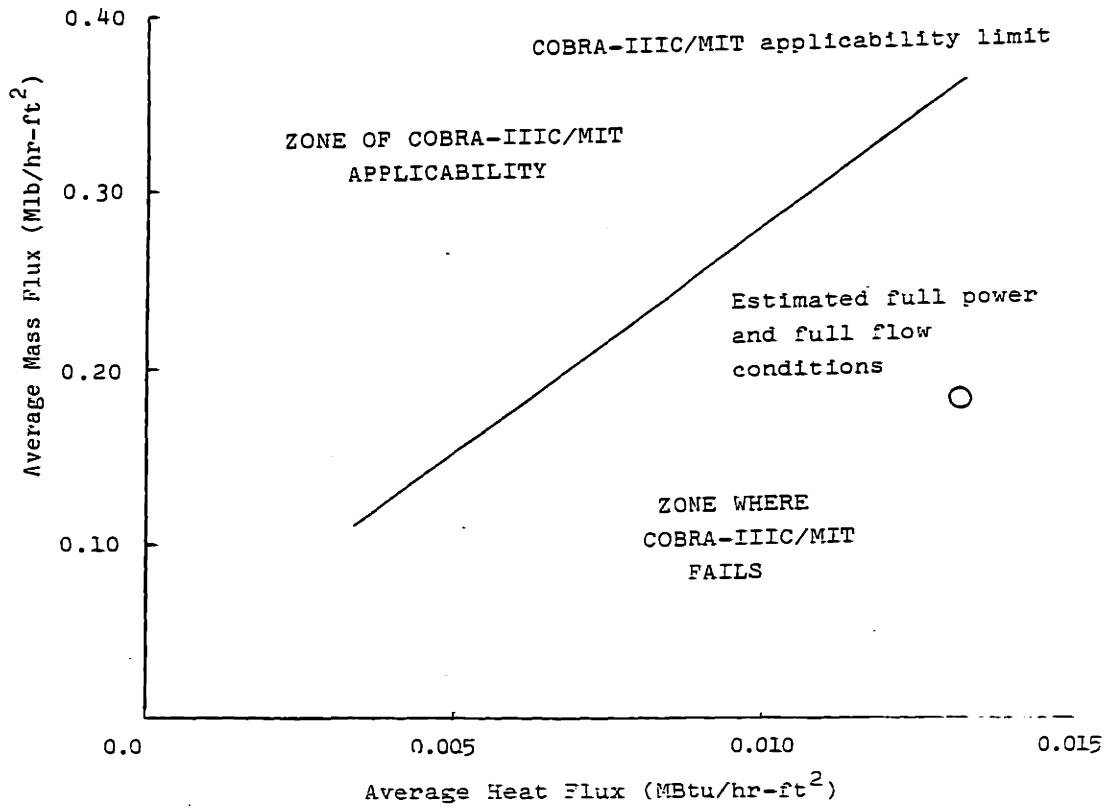


Figure IV-90

Mass Flux vs. Heat Flux  
COBRA-IIIC/MIT Applicability Limit

E. Summary of Testing and Application Results

The results of testing and application are summarized as follows:

- MDNBR, MCPFR, and MCHFR predictions were nearly the same for each case, even though various modeling options were used.
- Rod-to-coolant heat transfer predictions of the new heat transfer model vary smoothly in space and time. Discontinuous changes in predictions of the old heat transfer model can cause code failures during transient analysis of BWRs.
- Differences in fuel rod temperature predictions of the old and new fuel rod models make only small differences in fuel rod surface heat flux predictions.
- The new mixing model does not appear to significantly improve subchannel flow and enthalpy predictions for BWR conditions.
- CISE-4 MCPFR predictions are consistent with a best-estimate approach. Hench-Levy MCHFR predictions are conservative.
- Use of the Weisman transverse momentum option has no significant effect on steady state hot channel predictions of the single-pass method.
- COBRA-IIIC/MIT limits of applicability for BWR bypass analysis of the E.I. Hatch reactor have been obtained in terms of power and flow. Expected full power and full flow conditions are outside the zone of COBRA-IIIC/MIT applicability.
- THERMIT should be used with a pressure drop boundary condition for bypass analysis. THERMIT's variable

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axial nodalization feature should be used to give finer axial nodalization to the upper portion of bypass channels.

## V. DATA INPUT FOR THE IMPROVED VERSION OF COBRA-IIIC/MIT

The improved version of COBRA-IIIC/MIT has new calculation options that may be selected for use by input data. The three input data methods of COBRA-IIIC/MIT have been revised to allow use of new calculation options. Table V-1 gives the new options that may be selected by each input method.

The "New INPUT DATA Presentation" is the recommended input data method. It allows use of all new options and is convenient and well-documented. A limited selection of new options is available when either of the other two input methods is selected. Table V-1 also gives the IPILE options allowed by each of the three input methods. IPILE is a calculation option indicator. The value given for IPILE by input data determines the type of calculation performed. Table V-2 gives the features and uses of the different IPILE options. Old input data card decks may be expected to perform the same calculations when used by the improved version as they performed using COBRA-IIIC/MIT before improvement. Revisions of the input data methods have been made with the intent to have old card decks select old options when they are used with the improved version of COBRA-IIIC/MIT. There are ways for old card decks to mistakenly select new calculation options even though they selected old calculation options when used with an unimproved version of COBRA-IIIC/MIT. Although it is unlikely that old card decks will select new options, output should be checked to see that old options are selected when old decks are used with the improved version. A card-by-card description for each of the three input data methods is contained in Appendix M. Sample input and COBRA-IIIC/MIT output is included in Appendix O to facilitate understanding of data input for the improved version of COBRA-IIIC/MIT.

**Table V-1**  
**Input Data Methods for Improved Version of COBRA-IIIC/MIT**

Input Data Method	New Options Allowed	IPILE Options Allowed	Reference of Description for Input Data Method
Input Data Representation Based on that of COBRA-IIIC	New Mixing Model. Calculation of CPR using CISE correlation. Calculation of CHRR using Henschel Levy correlation.	IPILE = 0	App. 10 of Ref. 1
Simplified COBRA-IIIC Input Data Presentation to be Used for Assembly-to-Assembly Analysis of LWR	Same as above.	IPILE = 1 or 2	App. 11 of Ref. 1
New INPUT DATA Presentation	All new options available. New fuel rod, rod-to-coolant, heat transfer, and mixing models. Calculation of CPR using CISE. Calculation of CHRR using Henschel Levy or Biasi/Void-CHF. Transverse momentum coupling parameters may be used.	IPILE = 0, 1 or 2	App. 12 of Ref. 1

Table V-2

Features and Uses of IPILE Options

IPILE Option	Features	Uses
IPILE = 0	Gaps of various sizes may be used to interconnect coolant channels	Single-pass analysis Assembly-to-assembly analysis Subchannel-to-subchannel analysis
IPILE = 1	Gaps connecting coolant channels expected to be same size, except for channels split by "half-boundaries"	Assembly-to-assembly PWR analysis Subchannel-to-subchannel analysis
IPILE = 2	No interconnection between channels	Assembly-to-assembly BWR analysis

## VI. SUMMARY

Past research has indicated areas for improvement of COBRA-IIIC/MIT. The code has been improved by the addition of new options. New fuel rod, rod-to-coolant heat transfer, and mixing modeling options are now available. New critical power ratio and critical heat flux ratio calculation options and transverse momentum coupling parameters are also available in the improved COBRA-IIIC/MIT version.

The improvements have been tested individually and during application of the improved code to transient test cases. Testing mainly involved comparison of the predictions of different modeling options and in some instances, comparison of predictions with experimental measurements. The testing results provide an assessment of COBRA-IIIC/MIT capabilities in general, as well as the capabilities of individual options. Major testing results will be briefly discussed. MDNBR, MCPR and MCHFR predictions showed only small sensitivities to the fuel rod and heat transfer modeling options used for the test cases analyzed. Differences in predictions of the old and new heat transfer models resulted in different clad temperature predictions. Clad temperature varies more smoothly from one time step to the next with changing coolant conditions. Discontinuous change in old heat transfer model predictions caused failure of the flow solution to converge during transient BWR analysis. Fuel rod surface heat flux predictions of the old and new fuel rod models were close even though fuel rod temperature predictions showed some differences. The new mixing model did not improve subchannel flow and enthalpy predictions for BWR conditions. However, some improvement was seen in predictions for subcooled conditions. The CISE-4 MCPR predictions were in agreement with experimental CHF measurements. Hench-Levy MCHFR predictions were conservative for the CHF test cases. The new transverse momentum parameters had no significant effect on steady state hot channel predictions of the single-pass method.

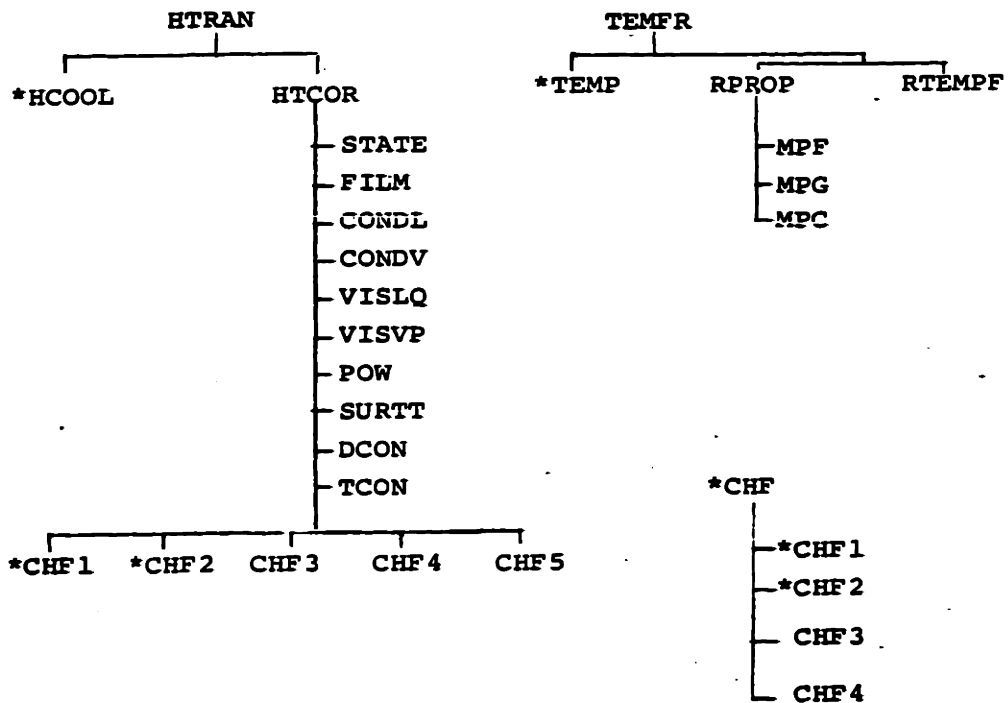
The improved version of COBRA-IIIC/MIT was applied to BWR bypass analysis. Two small test cases were analyzed to gain an understanding of how to analyze bypass flow in the E.I. Hatch Unit I reactor. COBRA-IIIC/MIT was used to analyze a nineteen channel test case geometry. COBRA-IIIC/MIT and THERMIT, a code with greater capabilities, were used to analyze a four channel test case geometry. COBRA-IIIC/MIT was limited in the power and bypass flow combinations it could analyze. Estimated full power and full flow conditions were found to be outside the zone of COBRA-IIIC/MIT applicability. Transient THERMIT analysis was performed in an attempt to obtain steady state predictions. Oscillation of predictions during this analysis indicated that THERMIT bypass analysis should be performed using a pressure drop boundary condition and modeling of flow restrictions at bypass channel inlets, rather than using only inlet flow and outlet pressure boundary conditions.



## APPENDIX A

### COBRA-IIIC/MIT Code Modifications

The COBRA-IIIC/MIT code has been modified during implementation of improvements. New subroutines have been added and old ones modified. Major new subroutines are contained within the subroutine structure shown in Figure A-1. New subroutines are described in Table A-1. Modifications of old subroutines are described in Table A-2. Subroutines are listed in the tables according to the order in which the subroutines appear in the listing of the improved version of COBRA-IIIC/MIT given in Appendix N.



Note: \* indicates old subroutines.

Figure A-1

New COBRA-IIIC/MIT Subroutine Structure

Table A-1New Subroutines

<u>Subroutine (or Function)</u>	<u>Description</u>
CHF3	calculation of critical heat flux using the Hench-Levy correlation
CHF4	calculation of critical power ratio using the CISE-4 correlation
HTRAN	oversees old and new rod-to-coolant heat transfer models
STATE	evaluates thermodynamics equations of state and their derivatives
TEMFR	oversees old and new fuel rod model calculations
INITRC	Initializes variables and arrays for new fuel rod model. Called by CALC before calculation of steady state.
RTEMPF	solves radial rod heat conduction for new fuel rod model
RPROP	finds fuel rod material and gap properties for new fuel rod model
MPF	material properties of fuel
MPG	gap conductance
MPC	material properties of clad
HTCOR	calculates rod-to-coolant heat transfer coefficient for new heat transfer model
FILM	calculates film boiling heat transfer coefficients for new heat transfer model
CHF5	calculation of critical heat flux using Biasi/CHF-Void correlation
POW	A function which evaluates $a^{**}b$ . It may be replaced by a fast, engineering accuracy exponentiation routine.
CONDL	liquid thermal conductivity
CONDV	steam thermal conductivity

Table A-1 (cont.)

<u>Subroutine (or Function)</u>	<u>Description</u>
VISLQ	liquid water viscosity
VISVP	steam viscosity
TCON	converts temperature from F to K
DCON	converts density from lb/ft**3 to kg/m**3
SURTT	surface tension of water

Table A-2Modifications of Old SubroutinesSubroutine

BAROC	COMMON COSAVE added to save CORAB array
CALC	Call to INITRC added. COMMONS LINK4, PPSV, REFP, and TIMEST added.
CURVE	COMMON INDSAV added to save index
INPRIN	New models indicated in printout. COMMONS FRDATA and LINK4 added.
EXPRIN	Type of CHF calculation indicated in printout.
MIX	New mixing model calculational option added.
PROP	Fuel rod surface temperature used to determine start of nucleate boiling and wall viscosity when rod-to-coolant heat transfer model is used. COMMON LINK4 added.
CARDS4	MC added to argument list. NK set to zero if IPILE=2.
CHAN	Modified to read in and print information regarding new models. COMMON FRDATA, GAPFAC, ITPSV, and LINK4 added.
CHF	Modified to call CHF3 and CHF4. CHF predictions made by CHF5 are obtained from the CHSAVE array. COMMON CHFSV added.
DIVERT	New transverse momentum parameters used in equations. COMMON GAPFAC added.
INDAT	Prints new model information. Fuel rod and rod-to-coolant heat transfer model indicators are initialized as zero. Elements of FACSL and FACSLK arrays are set to one. COMMON LINK4 added.
MODEL	IPILE added to argument list. Mixing model options are made available.
CORE	KS=1 and KMAX=80000 since DATA array set in MAIN program.
HEAT	Calls HTRAN rather than HCOOL. Calls TEMFR rather than TEMP. Iteration loop added. COMMON LINK4 and TIMEST added.
SEPRAT	COMMON REFP added.
VOID	COMMON PPSV added.

## APPENDIX B

### Methods Used by New Fuel Rod Model

#### B.1 Fuel and Cladding Material Properties

Calculation of fuel and cladding material properties is based on the MATPRO model (Ref. 15). The MATPRO model contains good fits to experimental data for fuel and clad material properties. However, some of the fits were formulated in terms of expressions which, although physically derived, were time consuming to compute. Therefore, the expressions were examined to find satisfactory fits which could be rapidly evaluated by a digital computer.

Cubic polynomials were developed to fit the temperature dependence of fuel  $\rho c_p$  within 2 percent over temperature from 300°K to 3000°K. The thermal conductivity of fuel was fit by a quadratic polynomial within 10 percent over 400°K to 2500°K. In each case there are separate, slightly different fits for uranium oxide and mixed oxide fuels.

Temperature-dependent clad material properties are also given by simple expressions in the new fuel rod model. The MATPRO model for thermal conductivity of Zircaloy is already a simple polynomial fit, and was taken over unchanged. The value of  $\rho c_p$  has been approximated by a linear fit from 300°K to 1190°K; this fit is within 5 percent of the data given in Ref. 13. (Clad temperatures would normally be far below 1190°K.) At 1190°K Zircaloy undergoes a transition fitted in the new model by two linear fits making a sharp, inverted vee corresponding to data in Ref. 15; above 1254°K, where the transition ends, few data are available, and a constant value is assumed as is recommended in Ref. 15.

#### B.2 Fuel-to-Clad Gap Heat Transfer Coefficient

The new fuel rod model calculates time-space behavior of gap conductance  $h_{\text{gap}}$ , using the MATPRO cracked-pellet model. This model calculates

$$h_{\text{gap}} = h_{\text{cond}} + h_{\text{contact}} + h_{\text{rad}} + h_{\text{press}}$$

where the four components on the right hand side represent, respectively, the effects of: thermal conductivity of the gas mixture of the gap; partial fuel-clad contact, supposed to change with burnup due to fuel pellet cracking and relocation; radiation heat transfer across the gap; and fuel pressing against clad if the gap is closed due to excessive fuel expansion. The gap heat transfer model has been added to COBRA-IIIC/MIT in a subroutine named MPG.

The four components of gap conductance will be briefly discussed. The first, gap gas conductivity, is computed in subroutine MPG by calculating a theoretical mixture conductivity for a mixture of four noble gases, helium, argon, krypton, and xenon. The presence of air and water vapor is neglected. The conductivity of helium is modified to represent the effect of a small gap on the statistical thermodynamics assumptions involved. The partial fuel-clad contact contribution is from the cracked-pellet model developed at INEL (Ref. 15); it involves a function of fuel burnup calculated once on the basis of input to MPG at the beginning of COBRA-IIIC/MIT calculations. The radiation heat transfer is based on standard formulas depending on the fuel and clad emissivities. The closed gap component is added on when the user-input gap width is less than the mean fuel-clad surface roughness; it takes the form  $h_{press} = CP_f^n$ , where  $C$ ,  $P_f$  (the fuel contact pressure against the clad), and the exponent  $n$  are user-specified input. The user-input dimensions are hot dimensions and are not recalculated to account for thermal expansion.

## APPENDIX C

### Description of Options and Logic Associated with Subroutine HEAT

Subroutine HEAT calculates the heat addition per unit length  $q'(I,J)$  for coolant nodes at axial position  $J$  of all channels  $I$ , from 1 to NCHANL. HEAT is called once for each axial level during the axial iteration scheme of COBRA-IIIC/MIT. HEAT may be used with or without a fuel rod model. When HEAT is used without a fuel rod model, the effect of heat capacity is ignored.

When a fuel rod model is used, the sequence of operations is as shown in Figure C-1. HEAT calculates fuel rod temperatures by first calling subroutine HTRAN to calculate a rod-to-coolant heat transfer coefficient. Then HEAT calls either subroutine TEMP (old fuel rod model) or subroutine TEMFR (new fuel rod model) to solve for the fuel rod temperature distribution. The calculation of rod-to-coolant heat transfer coefficient and the calculation of fuel rod temperatures have several options, as shown in Table C-1.

Subroutine HEAT has an inner iteration scheme to determine each steady state temperature distribution. This scheme is used at each axial level and for each pass through the reactor when either the temperature dependent property option or the new heat transfer model is used. The iteration is done either 50 times or until the centerline fuel temperature changes by less than an amount EPSF, which is user specified. If convergence is not reached in 50 iterations, the COBRA calculations are stopped and an error message is given.



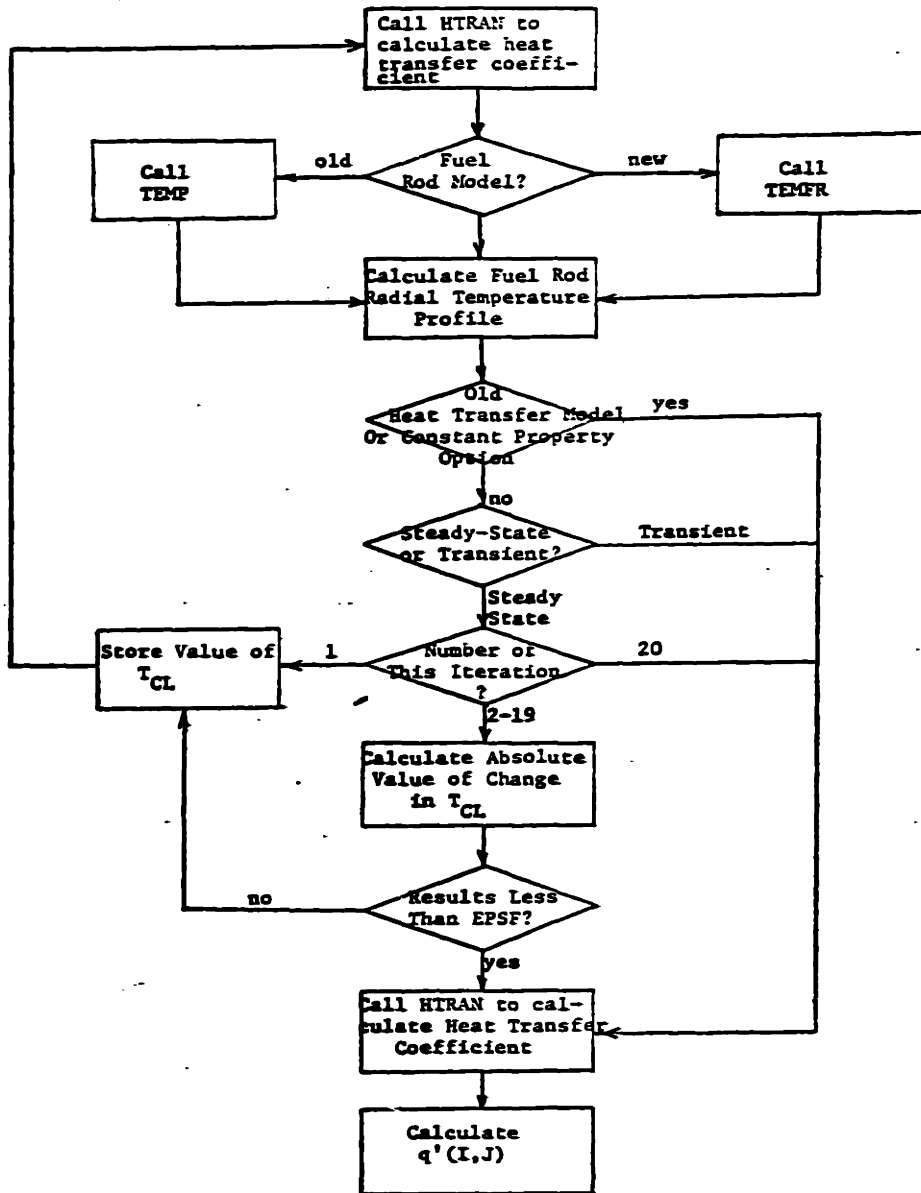


Figure C-1  
Flow Diagram of Logic Used in Subroutine  
HEAT when a Fuel Rod Model is Used

Table C-1

Available Options for Calculation of Heat Transfer Coefficient and Fuel Rod Temperatures

Option Indicator			Calculational Model Used		Heat Transfer Model
IFRM	IPROP	IHTM	Fuel Rod Model	Property Option	
0	0	0	Old	Constant properties, user input values of fuel and cladding properties and $h_{gap}$ .	Old
		1			New, pre-CHF only
		2			New, pre and post-CHF
1	0	0	New	Constant properties, user input values of fuel and cladding properties and $h_{gap}$ .	Old
		1			New, pre-CHF only
		2			New, pre and post CHF
1	1	0	New	Fuel and cladding properties calculated, user input value of $h_{gap}$ .	Old
		1			New, pre-CHF only
		2			New, pre and post CHF
1	2	0	New	Fuel and cladding properties and $h_{gap}$ calculated	Old
		1			New, pre-CHF only
		2			New, pre and post CHF

Note

Inner iteration on fuel rod temperature is used for all options except those which involve use of the constant property option (IPROP=0) and the old heat transfer model (IHTM=0).

## APPENDIX D

### New Heat Transfer Model.

The new heat transfer model calculates the rod-to-coolant heat transfer coefficient in subroutine HTRAN which is called by subroutine HEAT. The new heat transfer model is based on the BEEST package (Ref. 16). HTRAN calculates the heat transfer coefficient in two steps. First, it determines the heat transfer regime. Then, the correlation appropriate to the regime is used to calculate a heat transfer coefficient. The input to HTRAN is clad outer surface temperature and coolant temperature, pressure, velocity and void fraction. The heat transfer logic is given in Figure D-1. Correlations used by the new model are listed in Table D-1. The variable "IHTR" is a heat transfer regime indicator. "IHTM" is the heat transfer model indicator. IHTM equals either one or two when the new heat transfer model is used. If IHTM equals one, the new heat transfer model uses correlation and logic for pre-CHF conditions. When the IHTM equals two, the correlations and logic for pre- and post-CHF conditions are used.

Subroutine HTRAN computes fuel-to-fluid heat transfer coefficient using the following subroutines:

- STATE - calculates fluid properties as a function of temperature and pressures
- FILM - film boiling heat transfer coefficient
- CONDL - thermal conductivity of liquid water
- CONDV - thermal conductivity of dry steam
- VISLQ - viscosity of saturated liquid water
- MPC - thermal conductivity of cladding
- SURTTEN - surface tension of liquid water
- CHF1, }  
CHF2, } determines critical heat flux when IHTM = 2.  
CHF3, }  
CHF4, }  
or CHF5 }

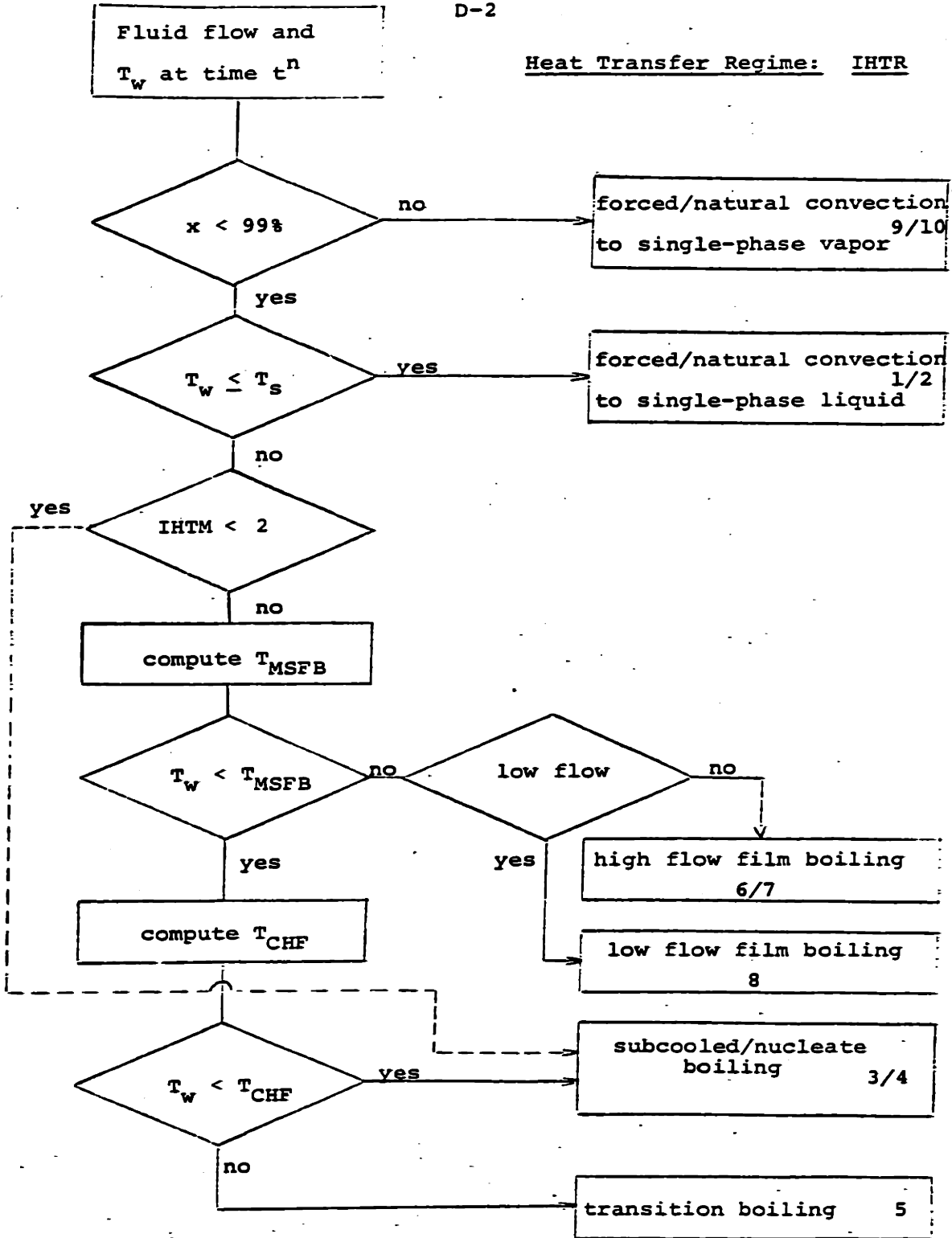


Figure D-1

Table D-1  
Heat Transfer Summary

IHTR	Regime	Correlation
1	forced convection to single-phase liquid	Sieder-Tate
2	natural convection to single-phase liquid	McAdams
3	subcooled boiling	Chen
4	nucleate boiling	Chen
5	transition	Interpolation between $q_{CHF}$ and $q_{MSFB}$
6	high P, high G film boiling	Groeneveld
7	low P, high G film boiling	modified Dittus-Boelter
8	low G film boiling	modified Bromley plus either McAdams vapor or high flow film boiling
9	forced convection to single-phase vapor	Sieder-Tate
10	natural convection to single-phase vapor	McAdams

## APPENDIX E

### Summary of Pre-CHF Correlations Used in Old and New Heat Transfer Models

The pre-CHF heat transfer correlations used in the old and new models are summarized in Tables E-1 and E-2. Table E-1 lists the correlations used. Table E-2 gives references, equations and range of data base for each correlation.

**Table E-1**

**Pre-CHE Correlations Used in  
the Old and New Heat Transfer Models**

Regime	Correlation Used and Selection Criterion	
	New Model	Old Model
Forced convection to single phase liquid	Sieder Tate Forced convection $x < 99\%$ $T_w \leq T_s$	Thom modified Dittus-Boelter $x \leq 0$ (Levy model not used) $x < x_d$ (Levy model used)
Natural convection to single phase liquid	McAdams Natural convection $x < 99\%$ $T_w \leq T_s$	Not considered
Local boiling or bulk boiling	Chen $x < 99\%$ $T_s < T_w < T_{MSFB}$	Thom modified Jens-Lottes $x > 0$ (Levy model not used) $x \geq x_d$ (Levy model used)

\*See list of nomenclature on page E-5.

Table E-2

Summary of Pre-CHF Correlations  
Used in New and Old Heat Transfer Models

Correlation	Ref.	Equation	Range of Data Base
Sieder Tate	31	$h = 0.023 \frac{k}{D} Re^{0.8} Pr^{0.33} (\mu/\mu_w)^{0.14}$ <p>Fluid properties at bulk fluid temperature, except <math>\mu_w</math> at <math>T_w</math></p> $Re = \frac{GD}{\mu} \quad Pr = \frac{\mu C_p}{k}$	Flow of water through tubes $10^2 < Re < 10^5$
McAdams	32	$h = 0.13k(Gr \cdot Pr)^{0.33}$ <p>Fluid properties should be at fluid film temperature</p> $Gr = \frac{\rho^2 g \beta (T_w - T_f)}{\mu}$	$10^9 < Gr \cdot Pr < 10^{12}$
Chen	33	$q'' = h_{FC}(T_w - T_f) + h_{NB}(T_w - T_B)$ $h_{FC} = 0.023 \frac{k_f}{D} Re_f^{0.4} Pr_f^{0.4}$ $h_{NB} = 0.00122S \left[ \frac{k_f C_p}{\sigma} \right]^{0.5} Pr_f^{-0.29}$ <p>* <math>\rho_f = 0.25 (P_w - P)^{0.75}</math></p> <p>* <math>C_p = \frac{C_p(T_w - T_s) \rho_f}{h_{fg} \rho_g} \Big]^{0.24}</math></p>	<p>Based on upflow and downflow through heated tubes and annuli. Originally developed for bulk boiling and two phase forced convective regimes. Extension to subcooled boiling regimes has produced satisfactory results (Ref. 3).</p> <p>P u - 505 psia  <math>V_{f,in}</math> 0.2 - 14.8 ft/sec  x 0 - 71%  <math>q''</math> .03 - 0.76 <math>\frac{MBTU}{hr-ft^2}</math></p>

Note: This eqn. is in SI units. All other eqns. in Table are in English units.



Table E-2 (cont.)

Correlation	Ref.	Equation	Range of Data Base
(Chen cont.)		$F = \begin{cases} 1 & \text{for } X_{tt}^{-1} \leq 0.1 \\ 2.35(X_{tt}^{-1} + 0.213)^{0.736} & \text{for } X_{tt}^{-1} > 0.1 \end{cases}$ $X_{tt}^{-1} = [x/(1-x)]^{0.9} (\rho_f/\rho_g)^{0.5} * (\mu_g/\mu_f)^{0.1}$ $S = \begin{cases} [1 + 0.12Re_{TP}^{1.14}]^{-1.0} & \text{for } Re_{TP} < 32.5 \\ [1 + 0.42Re_{TP}^{0.78}]^{-1.0} & \text{for } 32.5 \leq Re_{TP} \leq 70 \\ 0.1 & \text{for } Re_{TP} > 70 \end{cases}$ $Re_{TP} = 10^{-4} F^{1.25} (1-\alpha) (Re)_f$	
Thom modified Dittus-Boelter and Jens-Lottes	34	$h = 0.134 \frac{k_D}{D} Re^{0.65} Pr^{0.4}$ for forced convection to liquid $T_w = T_{sat} + \frac{0.072(q'')^{0.5}}{e^{P/1260}}$ $h = \frac{T_w - T_b}{q''}$ for local boiling	Based on upflow through heated tubes and annuli. Developed as a forced convective and subcooled boiling correlation. $P = 750 \text{ to } 2000 \text{ psia}$ $V_{f,in} = 5 \text{ to } 20 \text{ ft/sec}$ $q'' = 0 \text{ to } 0.5 \frac{\text{MBTU}}{\text{hr-ft}^2}$

Nomenclature for Tables E-1 and E-2Symbols

$C_p$	heat capacity	BTU/lb <sub>m</sub> °F
D	diameter	ft
g	gravitational acceleration	ft/hr <sup>2</sup>
G	mass flow rate	lb <sub>m</sub> /ft <sup>2</sup> hr
Gr	Grashof number $(\frac{\rho^2 g \beta [T_w - T]}{\mu^2})$	-
h	heat transfer coefficient	BTU/hr ft <sup>2</sup> °F
$h_{fg}$	latent heat of vaporization	BTU/lb
k	thermal conductivity	BTU/hr ft°F
P	pressure	psia
Pr	Prandtl number $(= \mu c_p / k)$	-
$q''$	heat flux	BTU/hr ft <sup>2</sup>
Re	Reynolds number $(= GD_h / \mu)$	-
T	temperature	°F
V	velocity	ft/sec
x	quality	-
$x_d$	quality at which bubble departure starts according to Levy model	-
$X_{tt}$	Martinelli parameter	-
$\alpha$	void fraction	-
$\beta$	thermal expansion coefficient	°F <sup>-1</sup>
$\mu$	viscosity	lb <sub>m</sub> /ft hr
$\rho$	density	lb <sub>m</sub> /ft <sup>3</sup>
$\sigma$	surface tension	lb <sub>f</sub> /ft

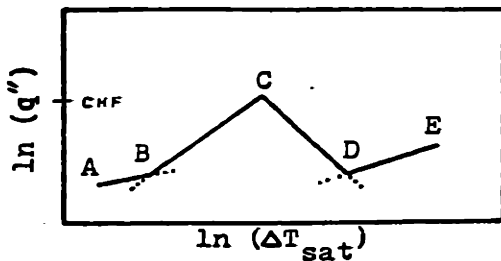
Subscripts

b bulk fluid  
f liquid phase  
s saturation  
g vapor phase  
w wall  
FC forced convection  
in inlet  
MSFB minimum stable film boiling  
NB nucleate boiling  
TP two phase

## APPENDIX F

### The COBRA-IV-I Heat Transfer Model

The COBRA-IV-I heat transfer model contains the capability to construct a complete boiling curve, as shown in the figure below, for each space and time step of the problem.



A-B forced convection

B-C subcooled and nucleate boiling, and forced convection vaporization

C-D transition boiling and transition pool boiling

D-E film boiling, low-pressure film boiling and pool film boiling

The heat transfer model contains the following correlations:

1. Dittus-Boelter
2. Thom (nucleate boiling heat transfer)
3. Schrock and Grossman
4. McDonough, Millich, and King
5. Groeneveld
6. Dougall and Rohsenow
7. Berenson

## APPENDIX G

### Beus Mixing Model

The Beus mixing model (Ref. 17) considers two regions on a plot mixing rate versus quality as shown in Figure G-1. The low quality region is referred to as the churn mixing region and corresponds to the bubbly slug flow regime, as shown in Figure G-2. The high quality region is referred to as the transition mixing region and corresponds to the annular flow regime. The two regions are divided by a location of peak mixing at which quality,  $x$ , equals  $x_c$ .

In the churn mixing region, the mixing model is based on a physical model which assumes that mixing is due to displacements of fluid between subchannels caused by movement of vapor slugs with respect to cocurrently flowing liquid. In this region, the experimental data studied by Beus indicates that the mixing rate increases steadily with quality and is given by the following equation\*:

$$W' = W_S + \beta_1 \left[ \frac{AG}{D_h} \right] \frac{\rho_l}{\rho_g} \left[ \frac{\gamma - 1}{\gamma} \right] x$$

where the slip ratio,  $\gamma$ , is obtained from the Smith correlation (Ref. 35).  $W_L$  and  $\beta_1$  are calculated using the following equations:

$$W_L = 0.0035 \mu_l Re_l^{.9}$$

$$\beta_1 = 0.04 \left[ \frac{S}{D_h} \right]^\lambda, \lambda = 1.5 .$$

The quality at which peak mixing occurs, and where transition mixing begins,  $x_c$  is determined by the following equation:

---

\* nomenclature is defined at the end of this appendix.

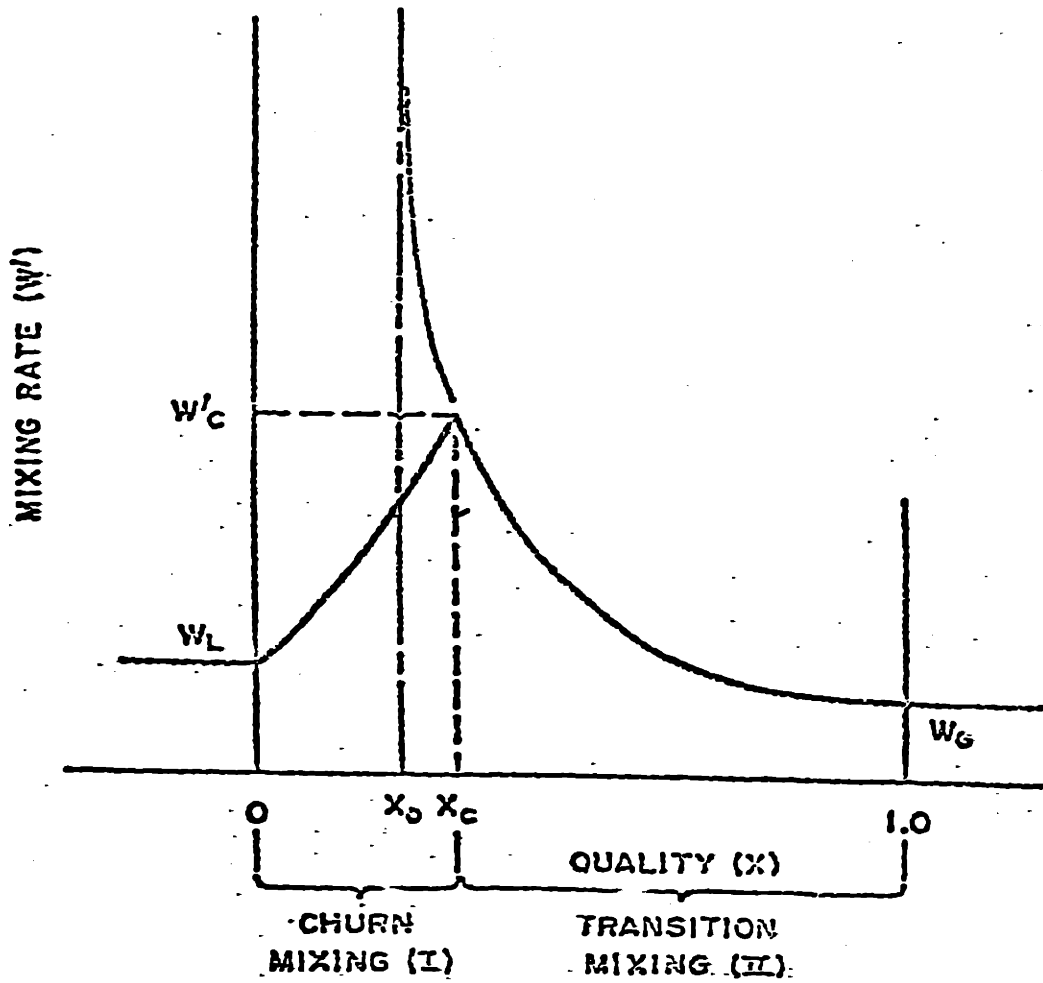


Figure G.1 (Fig. 4 of Ref. 17)

Plot of Mixing Model Showing Variation with Quality

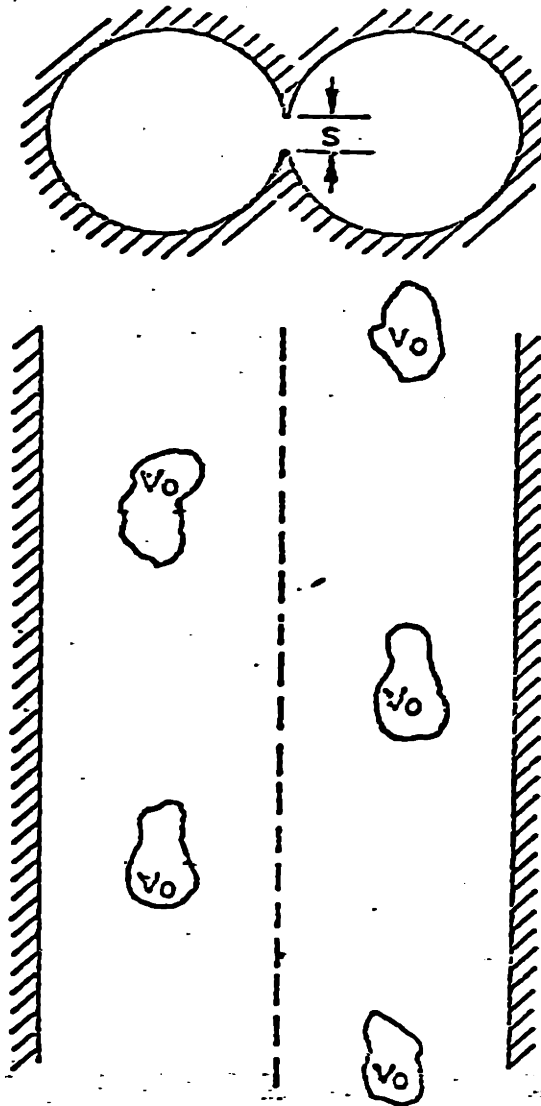


Figure G-2 (Based on Figure 2 of Ref. 17)

Idealized Subchannel Configuration

$$x_c = \frac{A_1 [g \rho_l D_h (\rho_l - \rho_g)]^{1/2}}{\left[ \frac{\rho_l}{\rho_g} \right]^{1/2} + A_2}$$

where,

$$A_1 = 0.4$$

$$A_2 = 0.6.$$

In the transition mixing region, the data studied by Beus indicated a smooth decline of mixing rate from the peak value to a constant value at high quality,  $W_G$ , as shown in Figure G-1. The Beus model constructs a hyperbolic curve to approximate the equation:

$$W'_{II} = W_G + [W'_c - W_G] \left[ \frac{1 - \frac{x_0}{x_c}}{\frac{x}{x_c} - \frac{x_0}{x_c}} \right]$$

where,

$$W'_c = W'_I[x_c]$$

$$W_G = .0035 \mu_g \text{Re}_g^{.9}$$

and

$$\frac{x_0}{x_c} = .57 \text{Re}^{.0417}$$

The values of  $\beta_1$ ,  $x_0/x_c$ ,  $W_L$  and  $W_G$  were obtained by least square fits to the studied data.



Nomenclature

A = subchannel flow area (ft<sup>2</sup>)

D<sub>h</sub> = hydraulic diameter (ft)

G = mass flux (lbm/hr-ft<sup>2</sup>)

L = channel length (ft)

T = temperature (°F)

W = mixing rate (lbm/hr-ft)

μ = viscosity (lbm/hr-ft)

ρ = density (lbm/hr-ft<sup>3</sup>)

x = quality

$$Re_l = (G \cdot D_h) / \mu_l$$

$$Re_g = (G \cdot D_h) / \mu_g$$

## APPENDIX H

### Summary of Correlations Provided for Calculation of DNBR, CHF and CPR

The correlations now provided in COBRA-IIIC/MIT for calculation of DNBR, CHF and CPR are summarized in Tables H-1 and H-2. Table H-1 lists the correlations provided. Table H-2 gives references, equations and range of data base for each correlation.

Table H-1  
Correlations Provided for Calculation  
Of DNBR, CHF and CPR

Option Indicator (NCHF)	Correlation	Quantity Calculated		
		DNBR	CHF	CPR
1	BAW-2	✓		
2	W-3	✓		
3	Hench-Levy		✓	
4	CISE-4			✓
5	Biasi/Void-CHF		✓	

Notes:

1. The new heat transfer model requires a CHF calculation in order to consider post-CHF heat transfer. Any of the correlations listed above can be used for this calculation. (Ref. discussion in Section II.B.1)
2. The W-3 correlation requires calculation of the start of local boiling. When the old heat transfer model is being used, the Thom modified Jens-Lottes correlation is used (Ref. Table B.2 of Appendix B). When the new heat transfer model is being used, the start of local boiling is determined by  $T_w > T_s$ .

Table H-2

Summary of Correlations Provided for Calculation of DNBR, CHF, and CPR\*

Correlation	Ref.	Equation	Range of Data Base
B6W-2	36	$\frac{q''_{CHF,EU}}{10^6}$ $= \{ (1.155 - 0.407D_e) [0.37 * 10^8 * (0.591G/10^6) \{0.83 + 0.685(p/10^3)^{-2}\} - 0.1521G_{CHF,fg}^H] \} / \{12.71 * (3.054G/10^6) [0.712 + 0.2073(p/10^3)^{-2}]\}$ <p>where <math>q''_{CHF,EU}</math> is in <math>BTU \text{ hr}^{-1} \text{ ft}^{-2}</math></p>	<p><math>P = 2000</math> to <math>2400</math> psia  <math>G = 0.75 * 10^6</math> to <math>4.0 * 10^6 \text{ lb hr}^{-1} \text{ ft}^{-2}</math>  <math>D_e = 0.2</math> to <math>0.5</math> in.  <math>X_{exit} = -0.03</math> to <math>0.20</math>  <math>L = 72</math> in.                      Geometry = rod bundles <math>72</math> in. long having <math>15</math> in. grid span</p>
	37	$F_c = \frac{q''_{CHF,EU}}{q''_{CHF,NU}}$ $= \frac{1.025C_0^k CHF q''(z) \exp[-C(k_{CHF} - z)] dz}{q''_{loc} * [1 - \exp(-C(k_{CHF,EU}))]}$ <p>where <math>z</math> is measured from the channel inlet</p> $C = \frac{0.249(1 - X_{CHF})^{7.82}}{(G/10^6)^{0.457}}$	<p><math>P = 2000</math> to <math>2400</math> psia  <math>G = 1 * 10^6</math> to <math>3.5 * 10^6 \text{ lb hr}^{-1} \text{ ft}^{-2}</math>  <math>D_e = 0.2</math> to <math>0.5</math> in.  <math>X_{exit} = 0.02</math> to <math>0.25</math></p>

\*See list of nomenclature on page C-9.

Table H-2 (cont.)

Correlation	Ref.	Equation	Range of Data Base
W-3.	38	$\frac{q''_{\text{crit,EU}}}{10^6} = \{(2.02 - 0.0004302p) + (0.1722 - 0.0000984p) * \exp[(18.177 - 0.004129p)X]\} * [(0.1484 - 1.596X) + 0.1729X X (G/10^6) + 1.037] * (1.157 - 0.869X)[0.2554 + 0.8357\exp(-3.151D_e)] [0.8258 + 0.000794(H_{\text{sat}} - K_{\text{in}})]$ <p>where <math>q''_{\text{CHF,EU}}</math> is in <math>\text{BTU hr}^{-1}\text{ft}^{-2}</math>.</p>	<p><math>p = 1000</math> to <math>2400</math> psia  <math>G = 1.0 * 10^6</math> to <math>5.0 * 10^6</math> <math>\text{lb hr}^{-1}\text{ft}^{-2}</math>  <math>D_e = 0.2</math> to <math>0.7</math>  <math>X_{\text{loc}} = -0.25</math> to <math>+0.15</math>  <math>L = 10</math> to <math>144</math> in.                      Heated perimeter = <math>0.88</math> to <math>1.00</math>                      Wetted perimeter = <math>0.88</math> to <math>1.00</math>                      Geometries = circular tube, rectangular channel, and bare rod-bundle</p>
	39	<p>Non-uniform flux shape factor:</p> $F_c = \frac{q''_{\text{DNB,EU}}}{q''_{\text{CHF,NU}}} = \frac{C}{q''_{\text{crit,NU}}(1 - e^{-Cl_{\text{crit}}})}$ $* \int_0^{\lambda} q''_{\text{crit}}(z) e^{-C(\lambda_{\text{crit}} - z)} dz$ <p>where <math>\lambda</math> is measured from start of local boiling.</p> $C = 0.15 \frac{(1 - X_{\text{crit}})^{4.31}}{(G/10^6)^{0.478}} \text{ in.}^{-1}$	<p><math>p = 1000</math> to <math>2400</math> psia  <math>G = 1.0 * 10^6</math> to <math>3.0 * 10^6</math> <math>\text{lb hr}^{-1}\text{ft}^{-2}</math>  <math>D_e = 0.2</math> to <math>0.7</math> in.  <math>X_{\text{exit}} \leq 0.15</math>  <math>L = 10</math> to <math>144</math> in.</p>

Table H-2 (cont.)

Correlation	Ref.	Equation	Range of Data Base
W-3 (cont.)	40	<p>Spacer-grid effect</p> $F_S = \frac{q''_{\text{crit, spacer}}}{q''_{\text{crit, bare rod bundle}}}$ $F_S = 1.0 + 0.03 \left( \frac{G}{10^6} \right) \left( \frac{TDC}{0.019} \right)^{0.35}$ <p>where TDC is thermal diffusion coefficient denoting the mixing caused by the spacer. Further, <math>TDC = c/(Va)</math>, where <math>c</math> is the eddy diffusivity, <math>V</math> is the axial velocity, and <math>a</math> is the gap between two adjacent fuel rods.</p>	rod bundles 8 to 14 ft. long
	41	$\frac{CHF_{\text{cold wall}}}{CHF_{W-3, D_h}} = 1.0 - Ru \left[ 13.76 - 1.372e^{1.78X} - 4.732 \left( \frac{G}{10^6} \right)^{-0.0535} - 0.0619 \left( \frac{P}{10^3} \right)^{0.14} - 8.509D_h^{0.107} \right]$ <p>where, <math>Ru = 1 - (D_e/D_h)</math> and <math>D_e</math> and <math>D_h</math> are the equivalent diameters based on wetted and heated perimeters, respectively.</p>	$X_{DNB} \leq 0.10$ $1.0 \leq G/10^6 \text{ lb hr}^{-1} \text{ ft}^{-2} \leq 5.0$ $L \geq 10 \text{ in.}$ Gap $\geq 0.10 \text{ in.}$
Hench-Levy	19	$(q''_c/10^6) = F \frac{RTU}{P \text{ hr-ft}^2}$ <p>for <math>(x_e) \leq 0.273 - 0.212 \text{ TANH}^2(3G/10^6)</math></p>	$P = 600 \text{ to } 1450 \text{ psia}$ $G = 0.2 \text{ to } 10^6 \text{ to } 1.6 \text{ to } 10^6 \text{ lb/h-ft}^2$ $D_e = 0.324 \text{ to } 0.485 \text{ in.}$ rod to rod and rod to wall spacings greater than 0.060 in.

Table H-2 (cont.)

Correlation	Ref.	Equation	Range of Data Base
Mench-Levy (cont.)		$(q''_c/10^6) = F_p [1.9 - 3.3 \langle x \rangle - 0.7 \text{TANH}^2 * (3G/10^6)], \text{ BTU hr}^{-1}\text{ft}^{-2}$ <p>for <math>0.273 - 0.212 \text{TANH}^2 (3G/10^6) \leq \langle x \rangle \leq 0.5 - 0.269 \text{TANH}^2 (3G/10^6) + 0.0346</math></p> <p>* <math>\text{TANH}^2 (\frac{2G}{10^6})</math></p> $(q''_c/10^6) = F_p [0.6 - 0.7 \langle x \rangle - 0.09 * \text{TANH}^2 (2G/10^6)], \text{ BTU hr}^{-1}\text{ft}^{-2}$ <p>for <math>\langle x \rangle \geq 0.5 - 0.269 \text{TANH}^2 (3G/10^6) + 0.0346 \text{TANH}^2 (\frac{2G}{10^6})</math></p> <p>where</p> $F_p = [1.1 - 0.1(\frac{P - 600}{400})^{1.25}]$	
CISE-4	21	$\langle x \rangle = \frac{D_h}{D_e} \left[ a \frac{L_B}{L_B + b} \right]$ <p>where</p> $a = \frac{1}{1 + 0.20(1 - P/P_{CR})^{-3}} \frac{a/10^6}{6} \text{ for } G < G^*$	<p>P = 720 to 1000 psia</p> <p>G = 0.8 to <math>3.0 \times 10^6 \text{ lb hr}^{-1}\text{ft}^{-2}</math></p> <p>L = 30 to 144 in.</p> <p>Rod O.D. = 0.40 to 0.78</p> <p>No. rods = 7 to 37</p>

Correlation	Ref.	Equation	Range of Data Base
CISE-4 (cont.)		<p>and</p> $a = \frac{1 - P/P_{CR}}{(1.35G/10^6)^{1/3}} \text{ for } G > G^*$ <p>where <math>G^* = 2.5 * 10^6 (1 - P/P_{CR})^3</math></p> <p>and</p> $b = 168(P_{CR}/P - 1)^{0.4} G/10^6 D_e^{1.4}$	
Biasi/Void-CHF	16	<p>For <math> G  \geq 10^6 \text{ lb hr}^{-1} \text{ ft}^{-2}</math> use the highest of the values of <math>q''_{CHF}</math> given by the following equations:</p> <p>1) <math>q''_{CHF} = 2.633(10^7)(30.48D)^{-n} G^{-1/6}</math>  <math>* [4.412F(p)G^{-1/6} - x]</math></p> <p>2) <math>q''_{CHF} = 1.181(10^9)H(p)(30.48D)^{-n}</math>  <math>* G^{-0.6}(1.0 - x)</math></p> <p>where</p> $F(p) = 0.7249 + 0.00683p \exp(-0.0021p)$ $H(p) = -1.159 + 0.01029p \exp(-0.00131p) + 130.4p(2103 + p)^{-1}$ $n = \begin{cases} 0.4 & \text{for } D \geq 0.0328 \text{ ft.} \\ 0.6 & \text{for } D < 0.0328 \text{ ft.} \end{cases}$	<p>Eqns. 1&amp;2 are based on the Biasi correlation (Ref.23). The range of data for this correlation is:</p> <p><math>P = 39</math> to <math>2058</math> psia</p> <p><math>G/10^6 = 0.074</math> to <math>4.4 \text{ lb hr}^{-1} \text{ ft}^{-2}</math></p> <p><math>D = 0.01</math> to <math>0.12</math> ft.</p> <p><math>L = 0.66</math> to <math>19.7</math> ft.</p> $X = \left( \frac{1}{1 + p_f/\rho_g} \right) \text{ to } 1.0$ <p>Note: Data base is for water in flow through vertical, uniformly heated tubes. The correlation is principally a dryout correlation and consequently is not expected to work well for low qualities and low flows.</p> <p>Eqn. 3 is based on the Void-CHF correlation (Ref.24). This correlation contains the physically based pool boiling CHF relationship of Zuber (Ref 42) Data base covers low flow upflow, downflow and counter-current flow conditions in Freon. Extension to water is justified on the basis of the proven wide range of applicability of the Zuber correlation.</p>



Table H-2 (cont.)

Correlation	Ref.	Equation	Range of Data Base
Biasi/Void-CHF (cont.)		<p>For <math>10^6 &gt;  G  &gt; 2 * 10^4 \text{ lb hr}^{-1} \text{ft}^{-2}</math> use a linear interpolation between the value obtained for "CHF" at <math>G = 10^6</math> and the value obtained by the following equation at <math>G = 2 * 10^4</math>:</p> $3) q''_{CHF} = (1 - \alpha) 0.9 \pi^{24} H_{fg}^{0.5} \rho_g^{0.25} * [\text{sg}(\rho_f - \rho_g)]^{0.25}$ <p>For <math>2 * 10^4 \geq  G  \geq 0</math> use Eqn. 3 with void fraction calculated for <math>G = +2 * 10^4</math>.</p> <p>Exception:</p> <p>For <math>P \geq 1200 \text{ psia}</math> and <math>x &gt; 0.5</math>, use Eqs. 1 and 2 for <math> G  \geq 2 * 10^5 \text{ lbs hr}^{-1} \text{ft}^{-2}</math>.            Use linear interpolation between Eqs. 1 and 2 at <math>G = 2 * 10^5</math> and Eqn. 3 at <math>G = 2 * 10^4</math>.</p>	

Nomenclature for Table H.2

a	Gap between two adjacent fuel rods	ft
C	Function of G and $X_{CHF}$ or $X_{crit}$	ft <sup>-1</sup>
CHF	Critical heat flux	BTU hr <sup>-1</sup> ft <sup>-2</sup>
D	Diameter of tube	ft
D <sub>e</sub>	Equivalent diameter based on wetted perimeter	ft
D <sub>h</sub>	Equivalent diameter based on heated perimeter	ft
F <sub>c</sub>	Flux shape factor	-
F <sub>p</sub>	Function of P	-
F <sub>s</sub>	Spacer grid factor	-
G	Mass velocity	lb hr <sup>-1</sup> ft <sup>-2</sup>
g	Acceleration of gravity	ft/sec <sup>2</sup>
g <sub>c</sub>	Conversion factor	ft/sec <sup>2</sup>
H	Enthalpy	BTU/lb
H <sub>fg</sub>	Latent heat of evaporation	BTU/lb
L	Length of heated channel	ft
L <sub>B</sub>	Boiling length	ft
L <sub>BC</sub>	Critical boiling length	ft
l <sub>CHF</sub>	Distance from start of local boiling to CHF location (W-3)	ft
l <sub>crit</sub>	Distance from channel inlet to critical heat flux location (B&W-2)	ft
l <sub>CHF,EU</sub>	Distance from start of local boiling to CHF location for equivalent uniform heat flux condition (W-3)	ft
p	Pressure	psia

$P_c$	Critical pressure	psia
$q''$	Critical heat flux	BTU hr <sup>-1</sup> ft <sup>-2</sup>
$q''_{crit}$		
$q''_c$		
$q''_{crit,EU}$	Critical heat flux for equivalent uniform heat flux	BTU hr <sup>-1</sup> ft <sup>-2</sup>
$q''_{CHF,EU}$		
$q''_{DNB,EU}$		
$q''_{crit,NU}$	Critical heat flux for non-uniform heat flux distribution	BTU hr <sup>-1</sup> ft <sup>-2</sup>
$q''_{CHF,NU}$		
$q''_{loc}$	Local heat flux	BTU hr <sup>-1</sup> ft <sup>-2</sup>
$v$	velocity	ft/hr
$\langle x_e \rangle$	Bundle average quality	-
$\langle x_e \rangle_c$	Bundle average critical quality	-
$x_{CHF}$	Quality at the critical heat flux location	-
$x_{DNB}$		
$x_{exit}$	Quality of channel exit	-
$x_{loc}$	Local quality	ft
$z$	Axial length	-
$\alpha$	Void fraction	ft <sup>2</sup> /hr
$\epsilon$	Eddy diffusivity or Reynolds flux	lb <sub>m</sub> /ft <sup>3</sup>
$\rho_f$	Density of saturated liquid	lb <sub>m</sub> /ft <sup>3</sup>
$\rho_g$	Density of saturated vapor	lb/ft
$\sigma$	Surface tension	

## APPENDIX I

### Description of the Three Transverse Momentum Options Provided in COBRA-IIIC/MIT

#### 1. The Old COBRA-IIIC/MIT Approach

The old COBRA approach (Ref. 43) is based on conserving transverse momentum in a control volume for the gap between two subchannels as shown in Figure I-1. By conservation of momentum, the following equation is obtained:

$$\frac{\partial}{\partial t} [W_{ij}] + \frac{\partial (u^* W_{ij})}{\partial x} = \frac{s}{l} (P_i - P_j) - F_{ij} \quad (\text{Eqn. I-1})$$

where

$$F_{ij} = \frac{K |W_{ij}| W_{ij}}{2 (S_{ij})^2 \rho^*} \frac{s}{l} \quad (\text{Eqn. I-2})$$

and

$W_{ij}$  = diversion crossflow between subchannels i and j  
(lb<sub>m</sub>/hr ft)

$u^*$  = effective velocity carried by diversion crossflow  
(ft/sec)

$x$  = axial distance (ft)

$s$  = width of gap between rods (ft)

$l$  = effective length of connection between subchannels (ft)

$P_i$  = pressure in channel i (lb<sub>f</sub>/ft<sup>2</sup>)

$P_j$  = pressure in channel j (lb<sub>f</sub>/ft<sup>2</sup>)

$K$  = crossflow resistance coefficient (dimensionless)

$S_{ij}$  = total gap width connecting channels i and j ( $S_{ij}=s$   
for subchannel analysis) [ft]

$\rho^*$  = density of the diversion crossflow (lb<sub>m</sub>/ft<sup>3</sup>)

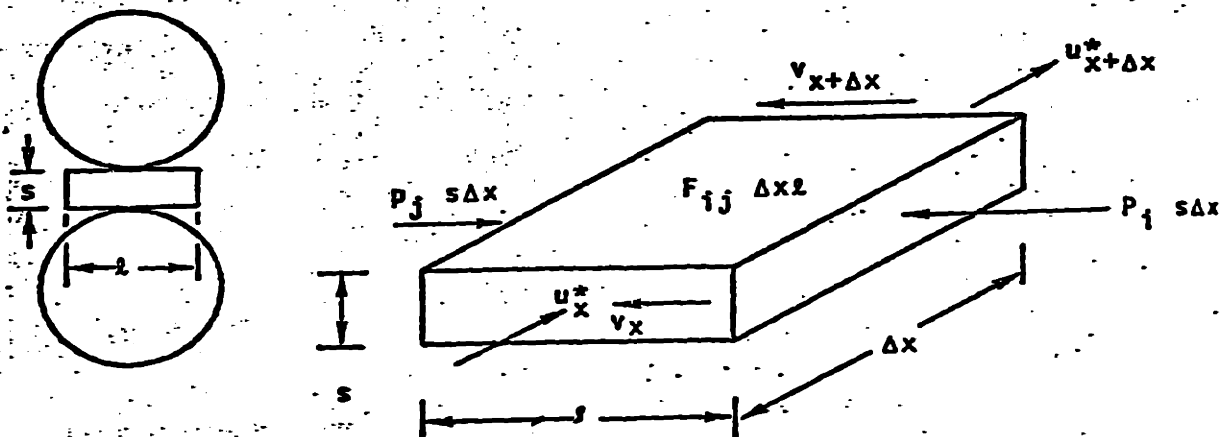


Figure I-1

COBRA Transverse Momentum Control Volume

## 2. The Weisman Approach

The Weisman approach (Ref.14) casts the transverse momentum equation in a more general form, allowing interconnection of different-sized channels.

$$\frac{\partial}{\partial t}[W_{ij}] + \frac{\partial(u^*W_{ij})}{\partial x} = \left(\frac{S}{L}\right)_{ij}(P_i - P_j) - F_{ij} \quad (\text{Eqn. I-3})$$

where

$$F_{ij} = \frac{K|W_{ij}|W_{ij}}{2(S_{ij})^2\rho^*} \left(\frac{S}{L}\right)_{ij}(N_r)_{ij} \quad (\text{Eqn. I-4})$$

and

$$S_{ij} = (N_g)_{ij}^s \quad (\text{Eqn. I-5})$$

$$L_{ij} = (N_r)_{ij}^l \quad (\text{Eqn. I-6})$$

where

$(N_g)_{ij}$  = number of gaps through which flow between channels  $i$  and  $j$  takes place

$(N_r)$  = number of rods between centers of channels  $i$  and  $j$ .

For subchannel or bundle-to-bundle analysis,  $N_g = N_r$  for all flow region interconnections. Thus, the Weisman approach reduces to the old COBRA approach for such analyses. Figure I-2 shows two interconnected regions of different size, a situation where the Weisman approach applies.

## 3. The Chiu Approach

The Chiu approach (Ref. 44) differs from the Weisman approach in the control volume used. Chiu uses the interaction of the adjacent rows of subchannels of two regions to represent the interaction between two regions, as shown in Figure I-3. This approach uses the following transverse momentum equation.

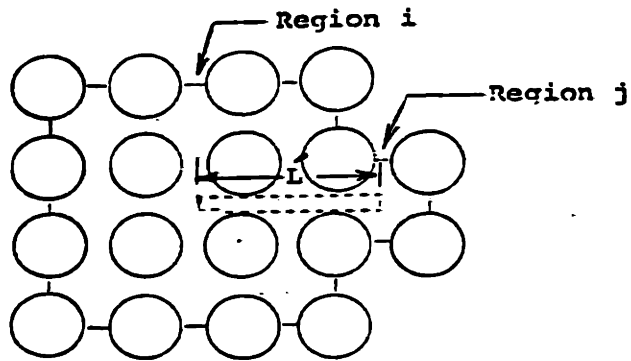


Figure I-2

Transverse Momentum Control  
Volume for Weisman Approach

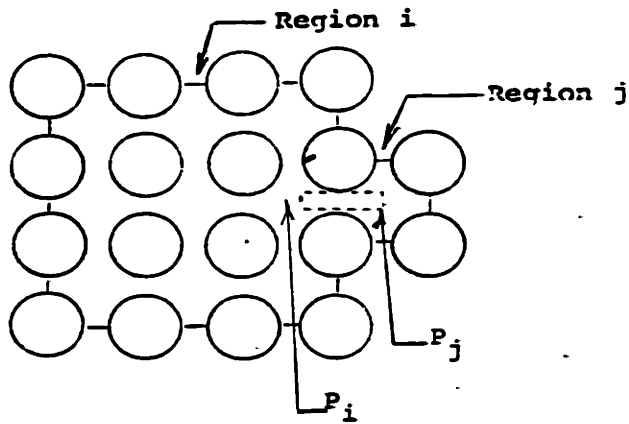


Figure I-3  
Transverse Momentum Control  
Volume for Chiu Approach



$$\frac{\partial}{\partial t}[W_{ij}] + \frac{\partial(u*W_{ij})}{\partial x} = \frac{S_{ij}}{l} \frac{(P_i - P_j)}{(N_p)_{ij}} - F_{ij} \quad (\text{Eqn. I-7})$$

where

$$F_{ij} = \frac{K|W_{ij}|W_{ij}}{2(S_{ij})^2 \rho^*} \frac{S_{ij}}{l} \quad (\text{Eqn. I-8})$$

and

$$(N_p)_{ij} = \frac{(P_i - P_j)}{(P_i - P_j)} \quad (\text{Eqn. I-9})$$

where

$(N_p)_{ij}$  = the pressure transport coefficient for subchannels adjacent to the boundary between subchannels i and j.

$P_i$  = pressure in interacting subchannel(s) of channel i adjacent to gap interconnection ij ( $\text{lb}_f/\text{ft}^2$ ).

$P_j$  = pressure in interacting subchannel(s) of channel j adjacent to gap interconnection ij ( $\text{lb}_f/\text{ft}^2$ ).

During the development of the single-pass method (Ref. 12), use of the pressure transport coefficient was found to have little effect upon COBRA-IIIC/MIT enthalpy predictions, especially in comparison to changes resulting from use of an enthalpy transport coefficient in COBRA's energy equation. Both pressure and enthalpy transport coefficients were found to be unnecessary for single-pass MDNBR analysis under conditions without strong crossflow.

APPENDIX J

Data Used for 1/8 Core Single-Pass Analysis Case  
(Ref. Section IV.B Part 5)

Operating Conditions

System reference pressure	2100 psia
Average mass flux	2.48 Mlb/hr ft <sup>2</sup>
Average heat flux	0.1695 MBTU/hr ft <sup>2</sup>
Inlet coolant temperature	541°F

Geometry

The 20 channel layout shown in Figure IV-30 is used for the analysis. The channels are 136.7 inches in length. The fuel rod pitch is 0.58 inches.

Grid Spacer Data

Nine grid spacers are modeled in each channel. The relative axial locations and associated drag coefficient are given below.

<u>x/L</u>	<u>Drag Coefficient</u>
0.0	1.105
0.090	.461
0.228	.461
0.366	.461
0.504	.461
0.642	.461
0.780	.461
0.918	.461
1.0	1.015

Power Distribution Data

A total of 24 fuel rods are modeled (Ref. Figure II-6 of Section II.E). The radial power factors used are listed below.

<u>Rod Number</u>	<u>Radial Power Factor</u>
1	0.9976
2	1.120
3	1.25
4	1.273
5	1.352
6	1.280
7	1.365
8	1.330
9	1.334
10	1.273
11	1.116
12	1.40
13	1.353
14	1.249
15	1.273
16	1.119
17	1.30
18	1.29
19	1.251
20	1.130
21	1.130
22	1.135
23	1.140
24	1.140

Each rod has the dimensions and consists of the same physical properties. These data are:

Fuel Diameter - 0.3765 in.

Clad O.D. - 0.44 in.

Clad Thickness - 0.028 in.

Fuel Density - 650. lb/ft<sup>3</sup>

Fuel Thermal Conductivity - 1.4 BTU/hr ft<sup>°F</sup>

Fuel Specific Heat - 0.08 BTU/lb<sup>°F</sup>

Clad Density - 410. lb/ft<sup>3</sup>

Clad Thermal Conductivity - 8.8 BTU/hr ft<sup>2</sup>°F  
Clad Specific Heat - 0.078 BTU/lb°F  
Fuel-Clad Gap Conductance - 600. BTU/hr ft<sup>2</sup>°F

The axial power distribution used is shown in Figure II-7  
(Ref. Section II.E).

Thermal Hydraulic Parameters

The following values were used for various other thermal  
hydraulic parameters.

Single Phase Friction -  $f = 0.184 \text{ Re}^{-0.2}$   
Two-Phase Friction - Homogeneous Model  
Two-Phase Slip - Equal to 1  
Subcooled Void Fraction - Levy Model  
Mixing  $\beta$  - 0.02  
K factor - 0.5  
s/l factor - 0.5

APPENDIX K

PWR Transient Test Case Data

Nine channels were used to represent the Maine Yankee core for the three pump loss of flow transient analyzed with COBRA-IIIC/MIT.

Operating Conditions

The following operating conditions were used:

- a) System Pressure - 2200. psia
- b) Average Inlet Mass Flux -  $2.29 \times 10^6$  lb/hr-ft<sup>2</sup>
- c) Average Heat Flux -  $0.1821 \times 10^6$  Btu/hr-ft<sup>2</sup>
- d) Inlet Coolant Temperature - 546. °F

Dimension of Channels

<u>Channel</u>	<u>Flow Area (in<sup>2</sup>)</u>	<u>Wetted Perimeter (in)</u>	<u>Heated Perimeter (in)</u>
1, 3, 5, & 7	0.1843	1.382	1.382
2 & 6	0.2309	1.695	1.178
4	0.0918	0.9083	0.5496
8	33.00	251.0	210.1
9	895.80	6813.0	6107.0

Channel Length = 136.7 in.

Channel Numbering Map

0	8	8	8	8	8
0	2	1	0	8	8
6	4	3	5	5	8
8	0	7	0	6	8
8	8	8	8	8	8
9	9	9	9	9	9

Gap Boundary Data

1-2	0.140
1-3	0.140
1-8	0.280
2-4	0.1396
2-8	0.2796
3-4	0.140
3-5	0.140
3-7	0.140
4-6	0.1396
5-6	0.140
5-8	0.280
6-8	0.2796
7-8	0.420
8-9	7.280

Power Distribution

The axial power distribution used is shown in Figure K-1. Fifteen fuel rods were modeled using the radial power factors given in Table K-1.

Fuel Rod Modeling

Fuel pin geometry is as follows:

Fuel Pellet Diameter - 0.44 in

Clad O.D. - 0.3675 in

Clad Thickness - 0.028 in

Some cases used constant and others used temperature-dependent fuel and clad properties. Constant fuel and clad properties used were:

Fuel Density - 650 lb/ft<sup>3</sup>

Fuel Thermal Conductivity - 1.5 Btu/hr-ft-°F

Fuel Specific Heat - 0.08 Btu/lb-°F

Clad Density - 410 lb/ft<sup>3</sup>

Clad Thermal Conductivity - 8.8 Btu/hr-ft-°F

Clad Specific Heat - 0.078 Btu/lb-°F

Fuel-Clad Gap Heat Transfer Coefficient - 600 Btu/hr-ft<sup>2</sup>-°F

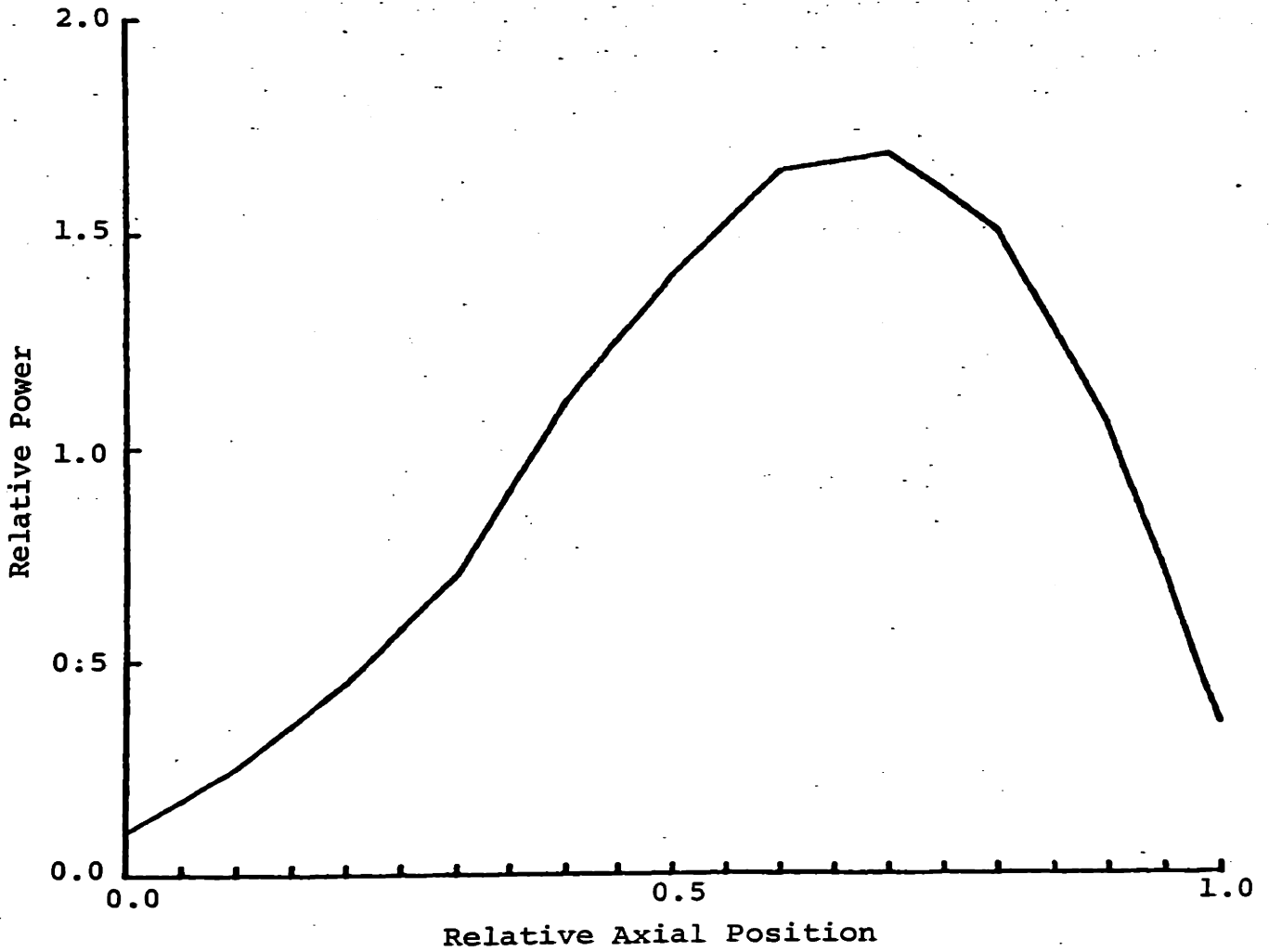


Figure K-1

Axial Power Distribution  
Loss of Flow Transient

Table K-1Radial Power Factors Used for PWR Transient Test Case

<u>Rod</u>	<u>Radial Power Factor</u>	<u>Fraction of Power(Channel)</u>			
1	1.475	.2564(1)	.7692(8)		
2	1.475	.2564(1)	.2564(2)	.5128(8)	
3	1.475	.3089(2)	.7166(8)		
4	1.475	.2564(1)	.2867(3)	.2564(7)	.2564(8)
5	1.611	.2442(1)	.2942(2)	.2730(3)	
6	1.475	.2867(3)	.2564(5)	.2564(7)	.2564(8)
7	1.475	.2867(3)	.2039(4)	.2564(5)	.3089(6)
8	1.475	.2564(5)	.7692(8)		
9	1.475	.2564(5)	.2564(6)	.5128(8)	
10	1.475	.3089(6)	.7166(8)		
11	1.475	.2564(7)	.7692(8)		
12	1.475	.2564(7)	.7692(8)		
13	1.264	168.2(8)			
14	0.9495	4716.(9)			
15	1.711	.1943(4)			



Spacer Friction Data

Nine grid spacers were modeled in each channel. The relative locations and associated drag coefficients are given in Table K-2.

Thermal-Hydraulic Models

The following thermal-hydraulic models were used for all cases:

Single-Phase Friction -  $f = 0.184 R_e^{-0.2}$

Two-Phase Friction - Homogeneous Model

Two-Phase Slip - Equal to 1

Subcooled Void Fraction - Levy Model

Mixing -  $\beta = 0.0062 \left(\frac{D}{S}\right) R_e^{-0.10}$

k factor - 0.5

Transverse Friction Factor, k - 0.5

s/l Factor - 0.5

Rod-to-coolant heat transfer was calculated using old model in some cases and new model for pre-CHF conditions in other cases.

Transient Forcing Functions

Transient forcing functions assumed are shown in Figure IV-30. Average inlet flow rate was specified for all cases. Average heat flux was specified for cases which used no fuel rod model. Average power level was specified for cases which used a fuel rod model.

Time Step Size

A time step size of 0.25 sec. was used for all cases.

K-6

Table K-2

Grid Spacer Data for PWR Transient Test Case

<u>x/L</u>	<u>Drag Coefficient</u>
0.0050	1.105
0.0877	0.4605
0.2194	0.4605
0.3511	0.4605
0.4828	0.4605
0.6144	0.4605
0.7461	0.4605
0.8778	0.4605
0.995	1.015

## Appendix L

### BWR Transient Test Case Data Shoreham Used to Represent the Turbine Trip Without Bypass Transient

#### Description of Input Used for COBRA-IIIC/MIT Analysis

Two channels were considered. One represented a "central hot" assembly and the other, a "central average" assembly. Both assemblies were 8x8. The channels were divided into twenty axial nodes.

#### Operating Conditions

The following operating conditions are used in all cases:

- a) System Pressure - 1031 psia
- b) Average Inlet Mass Flux -  $1.10 \times 10^6$  lb/hr-ft<sup>2</sup>
- c) Average Heat Flux -  $0.1512 \times 10^6$  Btu/hr-ft<sup>2</sup>

#### Dimensions of Channels

Two channels are used in each of the cases. The dimensions of both channels are as follows:

- a) Flow Area - 15.82 in<sup>2</sup>
- b) Wetted Perimeter - 118.25 in
- c) Heated Perimeter - 94.08 in
- d) Channel Length - 150.0 in

#### Power Distribution

The axial power distribution used in all cases is given in Figure L-1. Channel 1, used to represent a hot central assembly, has the radial peaking factor 1.4. Channel 2, used to represent an average central assembly, has the radial peaking factor of 1.04.

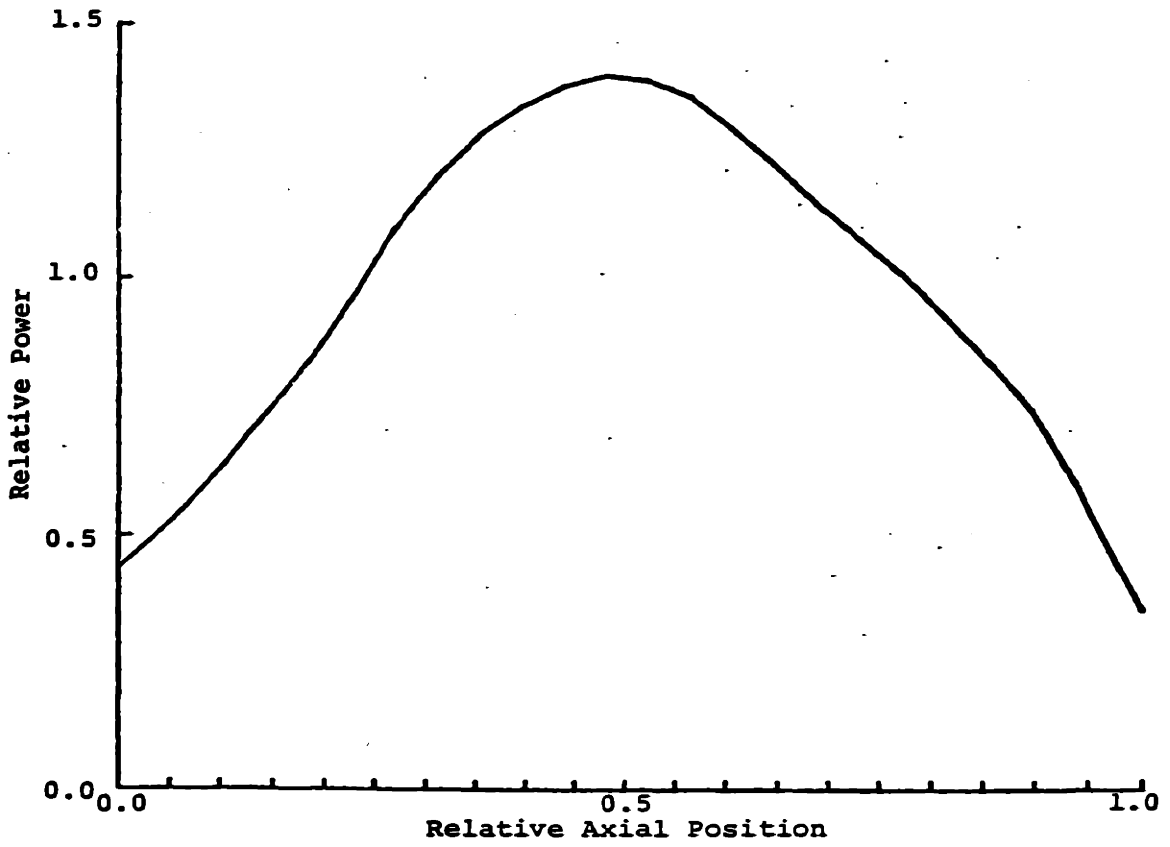


Figure L-1

Axial Power Distribution  
Turbine Trip Transient

Fuel Rod Modeling

Fuel pin geometry is as follows:

Fuel Pellet Diameter - 0.410 in

Clad O.D. - 0.483 in

Clad Thickness - 0.032 in

Cases were run with constant properties in some instances and temperature-dependent properties in others. The constant physical property used are as follows:

Fuel Density - 640.0 lb/ft<sup>3</sup>

Fuel Thermal Conductivity - 2.0 Btu/lb-ft-°F

Clad Specific Heat - 0.08 Btu/lb-°F

Clad Density - 405.0 lb/ft<sup>3</sup>

Clad Thermal Conductivity - 8.8 Btu/hr-ft-°F

Clad Specific Heat - 0.076 Btu/lb-°F

Fuel-clad Gap Conductance - 500.9 Btu/hr-ft<sup>2</sup>-°F

Rod to Coolant Heat Transfer Modeling

Some cases were analyzed using the old heat transfer model while others were analyzed using the new heat transfer model.

Spacer Data

Nine grid spacers are used to represent seven actual grid spacers, orificed fuel supports, and upper tie plates. Grid locations and coefficients for the two channels are as follows:

<u>Axial Location</u> (x/L)	<u>Grid Type</u>	<u>Grid Coefficient</u>	
		<u>Channel 1</u>	<u>Channel 2</u>
0.01	1	33.0	33.0
0.714	2	1.0	1.0
0.2143	2	1.0	1.0
9.3571	2	1.0	1.0
0.5000	2	1.0	1.0
0.6429	2	1.0	1.0
0.7857	2	1.0	1.0
0.9289	2	1.0	1.0
0.9900	3	10.0	19.0

#### Thermal Hydraulic Models

The following thermal-hydraulic models are used for all cases:

Single-Phase Friction -  $f = 0.184 \text{ Re}^{-0.2}$

Two-Phase Friction - Baroczy Model

Subcooled Void Fraction - Levy Model

Two-Phase Slip - Smith Model

Rod-to-coolant heat transfer calculated using old model in some cases and new model for pre-CHF conditions in other cases.

#### Transient Forcing Functions

Average heat flux, average inlet flow rate, and system reference power were varied as a function of time as is shown in Figure IV-39.



## APPENDIX M

### Input Data Methods for the Improved Version of COBRA-IIIC/MIT

The improved version of COBRA-IIIC/MIT has three methods for data input, as discussed in Section V. This appendix gives a card-by-card description of input for each of the three input data types. "Input Data Presentation Based on that of COBRA-IIIC/MIT" is described on pp. M-2 to M-21. "Simplified COBRA-IIIC/MIT Input Data Presentation to Be Used for Assembly to Assembly Analysis of LWR" is described on pp. M-22 to M-56. "New INPUT DATA Presentation" is described on pp. M-57 to M-121. This last methods is the recommended method.





M-2

Input Data Presentation Based on that of COBRA-IIIC  
(APPENDIX 10 of Ref. 1)

Card(s)	Type C1	Problem Array Size
Required to be present:		Always
FORTTRAN READ list:		MC MG MN MR MX
FORTTRAN FORMAT:		10I5
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
MC	1-5	I5	> No. of channels (NCHANL) in problem. NCHANL is set from NTHBOX on cards C5-C7, or in the original COBRA format, in Card Group 4.	--
MG	6-10	I5	> No. of gap interconnections [NK] between channels in problem. If this is not know, MG=2*MC is usually adequate but should be checked later. For a BWR, MG may be given as zero, when it is reset to 1 in CORE.	--
MN	11-15	I5	> No. of fuel nodal points in problem. This should be > (NODESF+1) on Card T1. If MN is given as zero, it is reset to 1 in CORE.	--
MR	16-20	I5	> No. of rods (NROD) in problem. For PWR and BWR, NROD=NCHANL, hence MR may be given=MC.	--
MX	21-25	I5	> No. of axial stations in problem. It may be given as NDX (Card C11) as it is increased by 1 immediately after reading in.	--

Notes:

- (1) MC to MX are used to set the array sizes in the dynamic storage, hence they should be set too big rather than too small.
- (2) Note that MC to MX are given in alphabetical order.
- (3) The maximum problem size is limited to 30,000 words by the dimension of the DATA array given in the MAIN program and the value of KMAX given in the CORE subroutine. Users can alter this limit with appropriate changes in their source programs.

Revised by J. Loomis  
May 1980

Card(s) Type	C2	Maximum Running Time
Required to be present	Always	
FORTTRAN READ list:	MAXT	
FORTTRAN FORMAT:	I5, 6E12.6	
Read from Subroutine:	INDAT	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
MAXT	1-5	I5	Maximum Running Time, Nominal value is 2000.

CG  
C  
O  
N  
T  
R  
O  
L

THE INPUT FOR A CASE REQUIRES A CASE CONTROL CARD FOLLOWED WITH UP  
TO 12 GROUPS OF INPUT INFORMATION. EACH OF THE 12 CARD GROUPS HAS A  
GROUP CONTROL CARD THAT IDENTIFIES THE GROUP NUMBER AND THE OPTIONS  
AVAILABLE FOR THAT GROUP.

GO TO THE CARD GROUP SPECIFIED BY NGROUP, IF THE DATA OF A CARD GROUP  
THE SAME AS THE PREVIOUS CASE, THEN THAT CARD GROUP AND ITS CONTROL  
MAY BE OMITTED.

## Card C3

Cards (s) Type C3

Required to be present

FORTRAN READ list

FORTRAN FORMAT

Read from subroutines

Case Control Card

Always

IPILE, KASE, J1, TEXT

I1, I4, I5, I7A4

INDAT

<u>Variables</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
IPILE	I1	1	= 0
KASE	I4	2 - 5	Run identification number. If > 0, calculation continue If < 0, calculation stops.
J1	I5	6 - 10	Printing option for standard COBRA output - as in COBR/ III-C. = 1 print entire output = 2 print only operation conditions
TEXT	I7A4	11 - 78	Alphanumeric information to identify case

Card Group 1	Always
Required to be present	NGROUP N1
FORTTRAN READ list	I5, I5
FORTTRAN FORMAT	INDAT
Read from subroutine	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
NGROUP	1-5	I-5	= 1 (to select Card Group 1)
N <sub>1</sub>	6-10	I-5	<p>≤ 0 : calculate physical properties from polynomials</p> <p>&gt; 1 : the physical properties are given in the next N<sub>1</sub> Cards as in the original COBRA.</p>

## Physical Properties

Required to be present	when N1(in Card Group 1) $\leq$ 0
FORTRAN READ list	N PH P2 N1
FORTRAN FORMAT	I5 F10.3 F10.3 I5
READ from subroutine	CARDS 1

<u>Variable</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
N	I5	1-5	= 1: PH defined as lowest pressure encountered in problem. = 2: PH defined as lowest enthalpy encountered in problem
PH	F10.3	6-15	Lowest pressure (psia) if N1 = 1 or lowest enthalpy (Btu/lb) if N1 = 2.
P2	F10.3	16-25	Highest pressure (psia) encountered in problem.
N1	I5	26-30	Number of pressure steps generated by polynomial (maximum 30)

Notes:



The lowest pressure encountered in the problem is defined as that at which the lowest enthalpy would be the saturation value. For example, at 1000 psia the saturation enthalpy is 543 Btu/lb. At an inlet subcooling of 100 Btu/lb, the enthalpy would be 443 Btu/lb and this would be the saturation value at a pressure of about 470 psia. Thus, one would require physical property data over the range 470 (or less) psia to 1000 psia in order to include data which covered the enthalpy range.

To avoid translating the lowest enthalpy to pressure, the option of giving the enthalpy is included. The program translates this value to a pressure which is safely below that required using the expression

$$p = 6h^3 (h-1.35) / (h - 0.35)$$

when  $p$  = calculated pressure (psia),  $h = 0.01H$ ,  $H$  = enthalpy (Btu/lb).

The values of  $p$ , so calculated, are given below and it may be seen that they are all less than  $P_{sat}$ , the tabled value of pressure corresponding to  $H$ .

H(Btu/lb)	181.2	300	400	500	600	700
p(psia)	11	101	279	589	1067	1749
$P_{sat}$ (psia)	15	103	319	745	1409	2236

In the original COBRA, the physical properties are read from cards into the arrays (PP(L), TT(L), etc., L = 1, N1). In the new version, the values of (PP(L), TT(L), etc., L = 1, N1) are generated within a Do Loop from 1 to N1 from the physical property polynomials. With the arrays set, the subsequent use of the values is the same in both versions of the code. Note: NPROP is set to N1 for storage of the size of the arrays.

M-11

Physical Properties

Required to be present:

When N1 (in Card Group 1) > 0

Read from Subroutine

CARDS 1

READ IN N1 CARDS OF FLUID PROPERTY DATA.  
EACH CARD CONTAINS -- SATURATION PRESSURE (PSIA), TEMPERATURE(DEG-F)

LIQUID SPECIFIC VOLUME (CU-FT/LB), VAPOR SPECIFIC VOLUME  
(CU-FT/LB)  
LIQUID ENTHALPY(BTU/LB), VAPOR ENTHALPY(BTU/LB), LIQUID VISCOSITY

(LB/FT-HR), LIQUID THERMAL CONDUCTIVITY(BTU/HR-FT-F) AND SURFACE  
TENSION(LB/FT), FORMAT(E5.2,F5.1,7F10.0). N1 MUST BE GREATER THAN  
ONE BUT NOT GREATER THAN THE PARAMETER MP.

THIS PROPERTY TABLE MUST HAVE PRESSURE HIGHER THAN OPERATING  
PRESS. AND LIQUID ENTHALPY LOWER THAN THE BUNDLE INLET ENTHALPY.

**CARD GROUP 2, FLOW CORRELATIONS**

READ IN UP TO FOUR SETS OF FRICTION FACTOR CORRELATION CONSTANTS THAT CORRESPOND TO THE SUBCHANNEL TYPES, FORMAT(12F5.3).

N1 IS THE SUBCOOLED VOID CORRELATION OPTION. N1=0, NO SUBCOOLED VOIDS. N1=1, LEVY SUBCOOLED VOID CORRELATION.

N2 IS THE BULK VOID CORRELATION OPTION. N2=0, HOMOGENEOUS MODEL.

N2 = 1, MODIFIED ARMAND MODEL. N2 =5, READ IN SLIP RATIO, FORMAT (5X,E10.5). N2=6, READ IN THE NUMBER OF TERMS AND COEFFICIENTS FOR UP TO A SIXTH ORDER POLYNOMIAL FUNCTION OF STEAM QUALITY, FORMAT (15,7E10.5).

N3 IS THE TWO-PHASE FRICTION GRADIENT MULTIPLIER OPTION. N3=0, HOMOGENEOUS. N3=1, ARMAND. N3=5, READ IN NUMBER OF TERMS AND COEFFICIENTS FOR UP TO A SIXTH ORDER POLYNOMIAL FUNCTION OF QUALITY FORMAT(15,7E10.5).

N4 IS AN OPTION TO INCLUDE A WALL VISCOSITY CORRECTION TO THE FRICTION FACTOR. IF N4=1, IT IS INCLUDED, OTHERWISE IT IS NOT.

**CARD GROUP 3, AXIAL HEAT FLUX TABLE**

READ IN N1 PAIR OF DATA FOR THE TABLE. EACH PAIR CONSISTS OF THE RELATIVE POSITION (X/L) AND THE CORRESPONDING RELATIVE HEAT FLUX (LOCAL FLUX/AVERAGE FLUX). EACH CARD ACCEPTS UP TO SIX PAIR OF DATA, FORMAT(12F5.3). N1 MUST BE GREATER THAN ONE BUT NOT GREATER THAN THE PARAMETER MP.

**CARD GROUP 4, SUBCHANNEL LAYOUT AND DIMENSIONS**

READ IN N1 CARDS OF SUBCHANNEL DATA CORRESPONDING TO THOSE SUBCHANNEL FOR WHICH DATA ARE BEING SUPPLIED. N2 IS THE TOTAL NUMBER OF SUBCHANNELS. FOR EACH OF THE N1 CARDS, READ IN THE SUBCHANNEL TYPE NUMBER (IF BLANK, IT IS ASSUMED TYPE 1), SUBCHANNEL IDENTIFICATION NUMBER, NOMINAL FLOW AREA(SQ-IN.), WETTED PERIMETER (IN.), HEATED PERIMETER(IN.) AND UP TO FOUR SETS OF ADJACENT SUBCHANNEL CONNECTING INFORMATION, FORMAT(I1,I4,3E5.2,4(I5,2E5.2)). EACH SET OF CONNECTING INFORMATION INCLUDES THE ADJACENT SUBCHANNEL NUMBER (NEGATIVE IF A LINE OF SYMMETRY SPLITS A GAP AT A BOUNDARY), NOMINAL GAP SPACING AND CENTROID-TO-CENTROID DISTANCE(IN.). IF SUBCHANNELS ARE INPUT IN ASCENDING ORDER, THEN ONLY HIGHER NUMBER SUBCHANNELS NEED TO BE IDENTIFIED AS CONNECTIONS. CENTROID DISTANCES ARE NOT REQUIRED IF THEY ARE NOT USED IN THE MIXING CORRELATIONS. N2 MUST BE GREATER THAN ONE BUT NOT GREATER THAN THE PARAMETER MC.

**CARD GROUP 5, SUBCHANNEL AREA VARIATION TABLE**

IF THERE ARE NO AREA VARIATIONS, OMIT THIS CARD GROUP.

READ N2 VALUES OF RELATIVE LOCATION(X/L) WHERE AREA FACTORS ARE GIVEN  
FORMAT(12F5.3). N2 MUST BE GREATER THAN ONE BUT NOT GREATER THAN  
THE PARAMETER ML.

READ N1 SETS OF AREA VARIATION FACTORS (LOCAL AREA/NOMINAL AREA).  
EACH SET CONSISTS OF SUBCHANNEL NUMBER AND N2 AREA VARIATION  
FACTORS, FORMAT(15/(12F5.3)). N1 IS LIMITED BY THE PARAMETER MA.  
IF N1 IS ZERO, AREA VARIATIONS ARE DELETED FOR SUCCEEDING CASES.

N3 IS THE NUMBER OF ITERATIONS FOR INSERTING AREA VARIATIONS.

IF N3 IS ZERO OR BLANK, N3 IS SET EQUAL TO 1.

**CARD GROUP 6, GAP SPACING VARIATION TABLE**

IF THERE ARE NO GAP VARIATIONS, OMIT THIS CARD GROUP.

READ N2 VALUES OF THE RELATIVE LOCATION(X/L) WHERE GAP FACTORS ARE  
GIVEN, FORMAT(12F5.3). N2 MUST BE GREATER THAN ONE BUT NOT GREATER  
THAN THE PARAMETER ML.

READ N1 SETS OF GAP SPACING FACTORS(LOCAL GAP/NOMINAL GAP).

EACH SET CONSISTS OF THE ADJACENT SUBCHANNEL NUMBERS FOR THE GAP  
N2 GAP VARIATION FACTORS, FORMAT(2I5/(12F5.3)). N1 IS LIMITED BY  
PARAMETER MS. IF N1 IS ZERO, GAP VARIATIONS ARE DELETED FOR SUCCEEDING  
CASES.

**CARD GROUP 7, SPACER DATA**

IF N1-1, WIRE WRAP FORCED DIVERSION CROSSFLOW MIXING IS INCLUDED, OTHERWISE, IT IS OMITTED.

READ ONE CARD CONTAINING THE WIRE WRAP PITCH (IN.), PIN DIAMETER AND WIRE DIAMETER (IN.), FORMAT (8E10.5).

IF N1-1, N5 IS AN OPTION TO SAVE OR USE A PREVIOUSLY COMPUTED CROSSFLOW SOLUTION. THE FLOW CONDITION MUST NOT CHANGE FOR THESE CASES NOR THE BASIC PROBLEM SETUP. THIS OPTION WOULD NORMALLY BE USED FOR CASES INVOLVING CHANGES IN POWER OR MIXING FOR NONDOLLING PROBLEMS.

N5=0, CROSSFLOW SOLUTION IS COMPUTED FOR EACH CASE.

N5-1, USE FIRST CASE SOLUTION FOR ALL SUCCEEDING CASES.

N5-2, WRITE SOLUTION TO TAPE AND USE FOR SUCCEEDING CASES.

N5-3, READ SOLUTION FROM TAPE AND USE FOR SUCCEEDING CASES. FOR EACH GAP, READ A CARD CONTAINING THE GAP NUMBER, THE EFFECTIVE FRACTION OF A PITCH FOR FORCING CROSSFLOW AND UP TO SIX RELATIVE PITCH LENGTHS IDENTIFYING THE LOCATION OF WRAPS CROSSING THROUGH A GAP USING A POSITIVE VALUE FOR WRAPS CROSSING FROM I TO J AND A NEGATIVE VALUE FOR CROSSINGS FROM J TO I WHERE I IS LESS THAN J. THE GAP NUMBERS ARE ASSIGNED IN THE ORDER THAT SUBCHANNEL PAIRS ARE IDENTIFIED IN CARD GROUP 4.

READ IN THE NUMBER OF WRAPS CONTAINED IN EACH SUBCHANNEL AT THE START OF THE BUNDLE IN ASCENDING SUBCHANNEL ORDER, FORMAT(10I5). USE ENOUGH CARDS TO SPECIFY ENTIRE WRAP INVENTORY.

IF N1-2, SPACER PRESSURE LOSSES AND FORCED FLOW DIVERSION ARE INCLUDED OTHERWISE, THEY ARE OMITTED.

N2 IS THE TOTAL NUMBER OF SPACER LOCATIONS.

N3 IS THE NUMBER OF SPACER TYPES.

N4 IS THE NUMBER OF ITERATIONS TO INSERT LOSS COEFFICIENTS OR THE WIREWRAP MIXING. IF N4 IS BLANK OR ZERO, ONE IS USED.

READ N2 RELATIVE LOCATIONS(X/L) WHERE SPACERS ARE LOCATED AND THE TYPE OF SPACER AT THAT LOCATION, FORMAT(6(F5.2,I5)).

READ N3 SETS OF DATA CORRESPONDING TO EACH SPACER TYPE. EACH SET CONSISTS OF A CARD FOR EVERY SUBCHANNEL. ON EACH CARD IS THE SUBCHANNEL NUMBER, SPACER LOSS COEFFICIENT, CONNECTION NUMBER OF GAP THROUGH WHICH FLOW IS FORCED, AND FRACTION OF FLOW DIVERTED, FORMAT(2(I5,E5.0)) IF THE CONNECTION NUMBER IS ZERO AND THE FLOW FRACTION IS ZERO, THEN THERE IS NO FORCED FLOW DIVERSION. THE FORCED CROSSFLOW HAS THE SAME SIGN AS THE FORCED FLOW FRACTION.



CARD GROUP 8, ROD LAYOUT, DIMENSIONS AND POWER FACTORS

READ IN N1 CARDS OF ROD LAYOUT DATA CORRESPONDING TO THOSE RODS FOR WHICH DATA ARE BEING SUPPLIED. N2 IS THE TOTAL NUMBER OF RODS. FOR EACH OF THE N1 CARDS, READ THE ROD TYPE, NUMBER, DIA. (IN.), RELATIVE ROD POWER (ROD POWER/AVERAGE ROD POWER) AND UP TO SIX SETS DATA FOR ROD-TO-SUBCHANNEL CONNECTIONS, FORMAT [I1, I4, IE5.2, 6 (I5, E5.0)] NUMBER AND FRACTION OF THE ROD POWER TO THAT SUBCHANNEL. THE NUMBER OF FUEL ROD TYPES ARE PRESENTLY LIMITED TO 2. N-1 INDICATES ROD FUEL. N-2 INDICATES PLATE FUEL. IN EACH CASE FOR PLATE FUEL THE ROD DIAMETER (ABOVE) IS THE PLATE THICKNESS AND THE FRACTION OF POWER TO A CHANNEL IS THE FRACTION OF THE CIRCUMFERENCE REQUIRED TO SPECIFY THE PLATE WIDTH FACING THE SUBCHANNEL.

N2 IS LIMITED BY THE PARAMETER MR.

N3 IS THE NUMBER OF RADIAL FUEL NODES INCLUDING THE CLADDING.

N4 IS THE TOTAL NUMBER OF FUEL TYPES.

FOR EACH FUEL TYPE, READ IN ON ONE CARD, THE THERMAL CONDUCTIVITY (B/HR-FT-F), SPECIFIC HEAT (B/LB-F), DENSITY (LB/FT3), AND PELLET DIAMETER (IN.) FOR THE FUEL, AND THE SAME FOR THE CLADDING EXCEPT FOR THICKNESS (I AND THE GAP COEFFICIENT (B/HR-FT]-F). THESE ARE ASSUMED CONSTANT. N5 IS AN OPTION TO SELECT A CRITICAL HEAT FLUX CORRELATION. IF N5=0, NO CHF CALCULATIONS ARE PERFORMED. IF N5=1, THE BAW-2 CORRELATION IS USED. IF N5=2, THE W-3 CORRELATION IS USED AND THE USER SHOULD VALIDATE THE TDC VALUE IN SUBROUTINE CHF. IF N5=3, THE HENCH-LEVY CORRELATION IS USED. IF N5=4, THE CISE-4 CORRELATION IS USED. OTHER CORRELATIONS OPTIONS CAN BE EASILY PROVIDED BY USERS.

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May 1980

CARD GROUP 9. CALCULATION VARIABLES

READ IN DIVERSION CROSSFLOW RESISTANCE FACTOR, TURBULENT MOMENTUM FACTOR, BUNDLE LENGTH(IN.), POSITION FROM VERTICAL(DEGREES), NUMBER OF AXIAL NODES, NUMBER OF TIME STEPS, TOTAL TRANSIENT TIME(SECONDS) MAXIMUM NUMBER OF ITERATIONS, ALLOWABLE FRACTION ERROR IN FLOW FORMAT CONVERGENCE AND TRANSVERSE MOMENTUM PARAMETER(S/L),  
FORMAT(4E5.2,2I5,E5.2,I5,4E5.2). IF THE NUMBER OF ITERATIONS, ALLOWABLE ERROR AND MOMENTUM PARAMETER ARE BLANK OR ZERO, THE PROGRAM USES 20., 1.E-3, AND .5, RESPECTIVELY.

N1 IS AN OPTION GIVING THE SPATIAL PRINTING INCREMENT. IF N1=1, STEP IS PRINTED. IF N2=2, EVERY OTHER STEP IS PRINTED, ETC. IF ZERO OR BLANK, THE PROGRAM SETS N1=1.

N2 IS AN OPTION GIVING THE TIME PRINTING INCREMENT AND IS SET UP SAME AS N1 ABOVE.

N3 IS A DEBUG PRINT OPTION. IF N3=0, NO DEBUG INFORMATION IS PRINT IF N3=1 A DEBUG PRINT IS MADE FOR EACH STEP OF THE CALCULATION. IT CAN GENERATE A LOT OF PAPER SO IT IS NOT NORMALLY USED.

CARD GROUP 10, TURBULENT MIXING CORRELATIONS

N1 IS THE OPTION FOR SUBCOOLED MIXING CORRELATIONS. FOR ANY N1<4  
 READ IN THE CONSTANTS A AND B, FORMAT (2F5.3).

THE OPTIONS ARE --

N1=0, W/GS=A

N1=1, W/GS=A\*RE\*\*B

N1=2, W/GD=A\*RE\*\*B

N1=3, W/GS=D/ZIJ\*A\*RE\*\*B

N1=4, NEW (BEUS) MIXING MODEL IS USED

NOTE THAT BETA = W/GS WHERE W IS THE TURBULENT CROSSFLOW.

N2 IS THE OPTION FOR TWO-PHASE MIXING. IF N2=1, TWO-PHASE MIXING  
 IS THE SAME AS FOR SUBCOOLED CONDITIONS. IF N2 IS GREATER THAN ONE  
 READ IN N2 PAIR OF DATA FOR A TABLE OF TWO-PHASE MIXING DATA.

EACH PAIR CONSISTS OF THE STEAM QUALITY AND THE CORRESPONDING VALUE  
 OF BETA. N2 IS LIMITED BY THE PARAMETER MP.

N3 IS THE OPTION FOR THERMAL CONDUCTION MIXING. IF N3=0, NO THERMA  
 CONDUCTION. IF N3=1, READ IN THE THERMAL CONDUCTION GEOMETRY FACTOR  
 FORMAT (F5.3).

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## CARD GROUP 11, OPERATING CONDITIONS

READ IN THE OPERATING PRESSURE(P5IA), INLET ENTHALPY(BTU/LB)  
OR INLET TEMPERATURE(DEG-F), MASS VELOCITY(M-LR/HR-SQ-FT) AND  
AVERAGE HEAT FLUX(M-BTU/HR-SQ-FT). (6P10.0)

N1 IS THE INLET ENTHALPY OPTION. IF N1=0, INLET ENTHALPY IS  
GIVEN. IF N1=1, INLET TEMPERATURE IS GIVEN. IF N1=2, READ IN THE  
INDIVIDUAL SUBCHANNEL INLET ENTHALPIES, FORMAT(12E5.0). IF N1=3,  
READ IN THE INDIVIDUAL SUBCHANNEL INLET TEMPERATURES, FORMAT(12E5.

N2 IS THE INLET FLOW DISTRIBUTION OPTION. IF N2=0, THE SUBCHANNELS  
ARE GIVEN THE SAME MASS VELOCITY. IF N2=1, THE INLET FLOW IS DIVID  
TO GIVE EQUAL PRESSURE GRADIENT IN THE SUBCHANNELS. IF N2=2, READ  
MASS VELOCITY FACTORS FOR EACH SUBCHANNEL, FORMAT(12E5.0).

N3, N4, N5 and N6 ARE OPTIONS FOR TRANSIENT FORCING FUNCTIONS. IF  
ANY OF THESE OPTION NUMBERS ARE ZERO OR BLANK, THE CORRESPONDING  
FORCING DATA IS NOT READ AND IS EXCLUDED FROM THE CALCULATIONS. EACH  
OF THESE NUMBERS GIVE THE NUMBER OF PAIRS OF TABULAR DATA TO BE RE  
FOR EACH FUNCTION. ALL DATA ARE READ AS PAIRS OF TIME(SECONDS)  
AND RELATIVE VALUE, FORMAT(12E5.0).

N3 IS THE OPTION FOR REFERENCE PRESSURE VERSUS TIME.

N4 IS THE OPTION FOR INLET ENTHALPY OR TEMPERATURE AS A FUNCTION OF  
TIME DEPENDING ON THE OPTION FOR INLET ENTHALPY OR TEMPERATURE.

N5 IS THE OPTION FOR INLET FLOW VERSUS TIME.

N6 IS THE OPTION FOR HEAT FLUX VERSUS TIME.

CARD GROUP 12, OUTPUT DISPLAY OPTIONS

N1 IS AN OPTION FOR PRINTING ANSWERS.

N1=0, PRINT SUBCHANNEL DATA ONLY.

N1=1, PRINT SUBCHANNEL DATA AND CROSSFLOWS.

N1=2, PRINT SUBCHANNEL DATA AND FUEL TEMPERATURES.

N1=3, PRINT SUBCHANNEL DATA, CROSSFLOWS AND FUEL TEMPERATURES.

N2 IS AN OPTION FOR SUBCHANNEL DATA PRINTOUT. IF N2=0, ALL SUBCHANNEL DATA ARE PRINTED. IF IT IS CALLED FOR BY N1. FOR N2 GREATER THAN 2 READ IN THE SUBCHANNEL NUMBERS FOR WHICH RESULTS ARE TO BE PRINTED FORMAT(36I2).

N3 IS AN OPTION FOR FUEL TEMPERATURE PRINTOUT. IF N3=0, DATA FOR ALL RODS ARE PRINTED IF CALLED FOR BY N1. FOR N3 GREATER THAN ZERO, READ IN N3 ROD NUMBERS FOR WHICH TEMPERATURES ARE TO BE PRINTED, FORMAT(36I2). IF CHF DATA IS CALLED FOR BY INPUT OPTION IT IS PRINTED FOR EACH SELECTED ROD PLUS A SUMMARY TO IDENTIFY THE ROD AND CHANNEL WITH THE MINIMUM CHF RATIO.

N4 IS AN OPTION FOR FUEL NODE PRINTOUT. IF N4=0, TEMPERATURES ARE PRINTED FOR EVERY NODE. FOR N4 GREATER THAN ZERO, READ IN N4 NODE NUMBERS TO BE PRINTED, FORMAT(36I2).

TO START A CALCULATION, READ A BLANK GROUP CONTROL CARD.

TO STOP THE CALCULATIONS, AFTER FINISHING A CASE, READ A BLANK CASE

\*\*\* END OF INPUT INSTRUCTIONS \*\*\*

UNITS - ALL COMPUTATIONS ARE DONE USING FT, LB, SEC, BTU AND DEG-F.  
UNIT CHANGES FOR INPUT AND OUTPUT ARE DONE IN THE PROGRAM.

M-22

Simplified COBRA-IIIC Input Data Presentation to be  
used for Assembly to Assembly Analysis of LWR  
(APPENDIX 11 of Ref. 1)

Card(s)	Type C1	Problem Array Size
Required to be present:		Always
FORTRAN READ list:		MC MG MN MR MX
FORTRAN FORMAT:		I0I5
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
MC	1-5	I5	> No. of channels (NCHANL) in problem. NCHANL is set from NTHBOX on cards C5-C7, or in the original COBRA format, in Card Group 4.	--
MG	6-10	I5	> No. of gap interconnections[NK] between channels in problem. If this is not know, MG=2*MC is usually adequate but should be checked later. For a BWR, MG may be given as zero, when it is reset to 1 in CORE.	--
MN	11-15	I5	> No. of fuel nodal points in problem. This should be > (NODESF+1) on Card T1. If MN is given as zero, it is reset to 1 in CORE.	--
MR	16-20	I5	> No. of rods (NROD) in problem. For PWR and BWR, NROD=NCHANL, hence MR may be given=MC.	--
MX	21-25	I5	> No. of axial stations in problem. It may be given as NDX (Card C11) as it is increased by 1 immediately after reading in.	--

Notes:

- (1) MC to MX are used to set the array sizes in the dynamic storage, hence they should be set too big rather than too small.
- (2) Note that MC to MX are given in alphabetical order.
- (3) The maximum problem size is limited to 30,000 words by the dimension of the DATA array given in the MAIN program and the value of KMAX given in the CORE subroutine. Users can alter this limit with appropriate changes in their source programs.

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Card(s) Type	C2	Maximum Running Time
Required to be present		Always
FORTTRAN READ list:		MAXT
FORTTRAN FORMAT:		I5, 6E12.6
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
MAXT	1-5	I5	Maximum Running Time, Nominal value is 2000.	C O N T R O L



THE INPUT FOR A CASE REQUIRES A CASE CONTROL CARD FOLLOWED WITH UP TO 12 GROUPS OF INPUT INFORMATION. EACH OF THE 12 CARD GROUPS HAS GROUP CONTROL CARD THAT IDENTIFIES THE GROUP NUMBER AND THE OPTIONS AVAILABLE FOR THAT GROUP.

GO TO THE CARD GROUP SPECIFIED BY NGROUP. IF THE DATA OF A CARD GROUP THE SAME AS THE PREVIOUS CASE, THEN THAT CARD GROUP AND ITS CONTROL MAY BE OMITTED.

Card(s) Type C3	Case Control Card
Required to be present	Always
FORTRAN READ list	1PILE, KASE, J1 TEXT
FORTRAN FORMAT:	I1, I4, I5, 17A4
Read from subroutine	INDAT

<u>Variable</u>	<u>Column</u>	<u>Format</u>	<u>Description:</u>
1PILE	1	I1	- 1: for PWR, with interconnected channels. - 2: for BWR, with separated channels.
KASE			} as in Appendix 10
J1			
TEXT			

## Card Group 1

Required to be present	Always
FORTRAN READ list	NGROUP      N1
FORTRAN FORMAT	I5, I5
Read from Subroutine	INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
NGROUP	1-5	I5	= 1 (to select Card Group 1)
N1	6-10	I5	<p>≤ 0: Calculate physical properties from polynomials.</p> <p>&gt; 1: the physical properties are given in the next N1 Cards as in the original COBRA.</p>

## Physical Properties

Required to be present	When N1(in Card Group 1) $\leq$ 0
FORTTRAN READ List	N PH P2 N1
FORTTRAN FORMAT	I5 F10.3 F10.3 I5
READ from subroutine	Cards 1

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
N	1-5	I5	= 1: PH defined as lowest pressure encountered in problem, = 2: PH defined as lowest enthalpy encountered in problem
PH	6-15	F10.3	Lowest pressure (psia if N1 = 1 or lowest enthalpy (Btu/lb) if N1 = 2)
P2	16-25	F10.3	Highest pressure (psia) encountered in problem
N1	26-30	I5	Number of pressure steps generated by polynomial (maximum 30).

The lowest pressure encountered in the problem is defined as that at which the lowest enthalpy would be the saturation value. For example, at 1000 psia the saturation enthalpy is 543 Btu/lb. At an inlet subcooling of 100 Btu/lb, the enthalpy would be 443 Btu/lb and this would be the saturation value at a pressure of about 470 psia. Thus, one would require physical property data over the range 470 (or less) psia to 1000 psia in order to include data which covered the enthalpy range.

To avoid translating the lowest enthalpy to pressure, the option of giving the enthalpy is included. The program translates this value to a pressure which is safely below that required using the expression

$$p = 6h^3(h - 1.35) / (h - 0.35)$$

when  $p$  = calculated pressure (psia),  $h = 0.01H$ ,  $H$  = enthalpy (Btu/lb).

The values of  $p$ , so calculated, are given below and it may be seen that they are all less than  $P_{sat}$ , the tabled value of pressure corresponding to  $H$ .

H(Btu/lb)	181.2	300	400	500	600	700
p(psia)	11	101	279	589	1067	1749
$P_{sat}$ (psia)	15	103	319	745	1409	2236

In the original COBRA, the physical properties are read from cards into the arrays (PP(L), TT(L), etc., L = 1, N1). In the new version, the values of (PP(L), TT(L), etc., L = 1, N1) are generated within a Do Loop from 1 to N1 from the physical property polynomials. With the arrays set, the subsequent use of the values is the same in both versions of the code. Note: NPROP is set to N1 for storage of the size of the arrays.

Physical Properties

Required to be present

When N1 (in the card group 1) > 0

READ IN N1 CARDS OF FLUID PROPERTY DATA.

EACH CARD CONTAINS -- SATURATION PRESSURE (PSIA), TEMPERATURE(DEC-F  
LIQUID SPECIFIC VOLUME(CU-FT/LB), VAPOR SPECIFIC VOLUME(CU-FT/LB),  
LIQUID ENTHALPY(BTU/LB), VAPOR ENTHALPY(BTU/LB), LIQUID VISCOSITY  
(LB/FT-HR), LIQUID THERMAL CONDUCTIVITY(BTU/HR-FT-F) AND SURFACE  
TENSION(LB/FT), FORMAT(E5.2,F5.1,7F10.0). N1 MUST BE GREATER THAN  
ONE BUT NOT GREATER THAN THE PARAMETER NF.

THIS PROPERTY TABLE MUST HAVE PRESSURE HIGHER THAN OPERATING PRESS.  
AND LIQUID ENTHALPY LOWER THAN THE BUNDLE INLET ENTHALPY.

**CARD GROUP 2, FLOW CORRELATIONS**

READ IN UP TO FOUR SETS OF FRICTION FACTOR CORRELATION CONSTANTS THAT CORRESPOND TO THE SUBCHANNEL TYPES, FORMAT(12F5.3).

N1 IS THE SUBCOOLED VOID CORRELATION OPTION. N1=0, NO SUBCOOLED VOIDS. N1=1, LEVY SUBCOOLED VOID CORRELATION.

N2 IS THE BULK VOID CORRELATION OPTION. N2=0, HOMOGENEOUS MODEL.

N2 = 1, MODIFIED ARMAND MODEL. N2=5, READ IN SLIP RATIO, FORMAT (5X,E10.5). N2=6, READ IN THE NUMBER OF TERMS AND COEFFICIENTS FOR UP TO A SIXTH ORDER POLYNOMIAL FUNCTION OF STEAM QUALITY, FORMAT (15,7E10.5).

N3 IS THE TWO-PHASE FRICTION GRADIENT MULTIPLIER OPTION. N3=0, HOMOGENEOUS. N3=1, ARMAND. N3=5, READ IN NUMBER OF TERMS AND COEFFICIENTS FOR UP TO A SIXTH ORDER POLYNOMIAL FUNCTION OF QUALITY, FORMAT(15,7E10.5).

N4 IS AN OPTION TO INCLUDE A WALL VISCOSITY CORRECTION TO THE FRICTION FACTOR. IF N4=1, IT IS INCLUDED, OTHERWISE IT IS NOT.

**CARD GROUP 3, AXIAL HEAT FLUX TABLE**

READ IN N1 PAIR OF DATA FOR THE TABLE. EACH PAIR CONSISTS OF THE RELATIVE POSITION (X/L) AND THE CORRESPONDING RELATIVE HEAT FLUX (Local flux/AVERAGE FLUX), EACH CARD ACCEPTS UP TO SIX PAIR OF DATA, FORMAT(12F5.3). N1 MUST BE GREATER THAN ONE BUT NOT GREATER THAN THE PARAMETER MP.



Card Group 4	(Channel Data) Card (1)
Required to be present	when IPILE=1 or 2
FORTRAN READ list	NGROUP
FORTRAN FORMAT	I5
READ from subroutine	INDAT

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
NGROUP	1-5	= 4 (To select Card Group 4)

**NOTE:** Once this card is read in the new subroutine CARDS 4 is entered for the remaining Read statements and Data processing of this Card Group 4.

## Card (2)

Required to be present:	when NGROUP = 4
FORTTRAN READ list:	N1, N2, NGRID, NGRIDT, NODESF, NFUEL, NCHF, IMAP, ITEXT
FORTTRAN FORMAT:	9I4
Read from subroutine:	CARDS4

<u>Variable</u>	<u>Columns</u>	
N1	1-4	Number of channel types [(max 15) see below]
N2	5-8	Total number of channels in problem
NGRID	9-12	Number of grid positions
NGRIDT	13-16	Number of types of grid
NODESF	17-20	Number of radial nodes on the fuel for center temperature calculation
NFUEL	21-24	Number of fuel types
NCHF	25-28	= 0 for no CHF calculations = 1 for B&W2 CHF correlation = 2 for W-3 correlation = 3 for Hench-Levy correlations = 4 for CISE-4 correlation
IMAP	29-32	= 1 to 4 to indicate method of presenting gap interconnection data [see Cards (9) below]
ITEXT	33-36	number of cards to be read in next which will be printed out as a message. If ITEXT=0, no message cards are read in

Note:

Channels are defined as being all of the same type if they have the same geometry, rod dimensions and grids and only differ in their power. More precisely, Cards (4) and (5) given later which define the geometry and grids must apply to all channels of the same type. In, for example, 1/4-core symmetry data, 1/4, 1/2 and whole channels would be different types.

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Required to be present	Card (3) when ITEXT > 0
FORTRAN READ list	TEXT
FORTRAN FORMAT	20A4
Read from Subroutine	CARDS#

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
TEXT	1-80	The array TEXT (20) is read and immediately printed in a DO loop from 1 to ITEXT. It is envisaged that a map of the channel numbering system could be printed as an aide-memory in a large problem.

Required to be present	Card (4)
FORTTRAN READ list	Always (being NGROUP=4) N, I, FRAC, AC(I), PW(I), PH(I) GAPS(I,1), DIST(I,1), DR(I), PHI(I,1), M
FORTTRAN FORMAT	I1, I4, 8E9.3, I2
READ from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
N	1	Selector for friction factor expression. If N=0 reset to 1.
I	2 - 5	Any channel number, preferably the first of the channel type being described.
FRAC	6-14	Factor by which AC, PW, PH should be multiplied. Thus for 1/4 channel, one may give FRAC = 0.25 and AC, PW, PH the same as for a whole channel.
AC	15-23	Channel flow area (in <sup>2</sup> )
PW	24-32	Channel wetted perimeter (in)
PH	33-41	Channel heated perimeter (in)
GAPS	42-50	Boundary gap dimensions (in)
DIST	51-59	Centroid-to-centroid channel distance (in). This is only required for a particular mixing correlation and may normally be given as zero.
DR	60-68	Rod diameter (in)
PHI	69-77	Number of rods in channel
M	78-79	Fuel type: = 1 for rod, = 2 for plate, Reset to 1 if M = 0

Card (5)

Required to be present

If NGRID &gt; 0

FORTRAN READ list

(CD (I.L), L=1, NGRIDT), (FXF(L),  
L=1, NGRIDT )

FORTRAN FORMAT

16 E5.3

Read from subroutine

CARDS<sup>4</sup>VariablecolumnsDescriptions

CD

Spacer loss coefficients

FXF

Fraction of axial flow forced  
across each boundary. It is  
not expected that this would  
be used in reactor problems hence  
nominal value = 0.0

If N1 (Card (2) ) is greater than one, cards describing channel and grid for channel type 2 will be given now, after these cards, the ones describing channel type 3 will be inputted and so on until the completion of the N1 channel types.

	Card (6)
Required to be present	Always
FORTTRAN READ list	(Radial (I), I=1, NROD)
FORTTRAN FORMAT	16 E5.3
Read from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
RADIAL	1-70	Radial power factor for rod I which is located in channel I. This is defined as the ratio of the rod power to that of the reactor average power.

**Notes:**

- a) NROD is the total number of rods, having set to NCHANL (total number of channels) which was itself set to N2 (Card (2) ).
- b) If all rods have the same power, RADIAL (1) alone may be given and is set negative. This triggers setting (RADIAL (I); I=1, NROD) = 1.0

	Card (7)
Required to be present	If NGRID > 0
FORTRAN READ list	(GRIDXL(L), IGRID(I), (I=1, NGRID)
FORTRAN FORMAT	3(E5.3, I5)
Read from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
GRIDXL		Relative location (z/L) where grids are located.
IGRID		Type of grid at GRIDXL

Required to be present	Card (3)
FORTRAN READ list	If N1 (Card(2)) > 1
FORTRAN FORMAT	JB(I)
Read from subroutine	20IN
	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
JB	1-80	List of channels of Type 2

**Notes:**

The first set given is the list of channel numbers in Type 2. The list is terminated by reading in a zero (or a blank space). Hence, if the last channel number comes at the end of a card, a blank card must follow in order to give the terminating zero. It is safer to make a habit of punching a final zero. Following Type 2, card(s) are read in for those channels in Type 3, then Type 4 etc. up to N1 Types.

Note that since the channel numbers for Type 1 are not read in, it is more economical to select Type 1 as that with the majority of channels.

An internal consistency check is made when reading in JB(I). If a set includes the channel number (I in Card (4)) for Type 1 or does not include that given for its own type in Card (4), an appropriate message is printed and the run terminated.

If N1 = 1, the JB cards above are not given.



## Card (9a)

Required to be present

only if IPILE = 1 (If IPILE = 2)  
 BWR case, no cards are given since  
 the channels are not connected) or  
 IMAP = 1 (Card (2) )

FORTRAN READ list

ICROSS, IDOWN

FORTRAN FORMAT

2I4

Read from subroutine

CARDS4

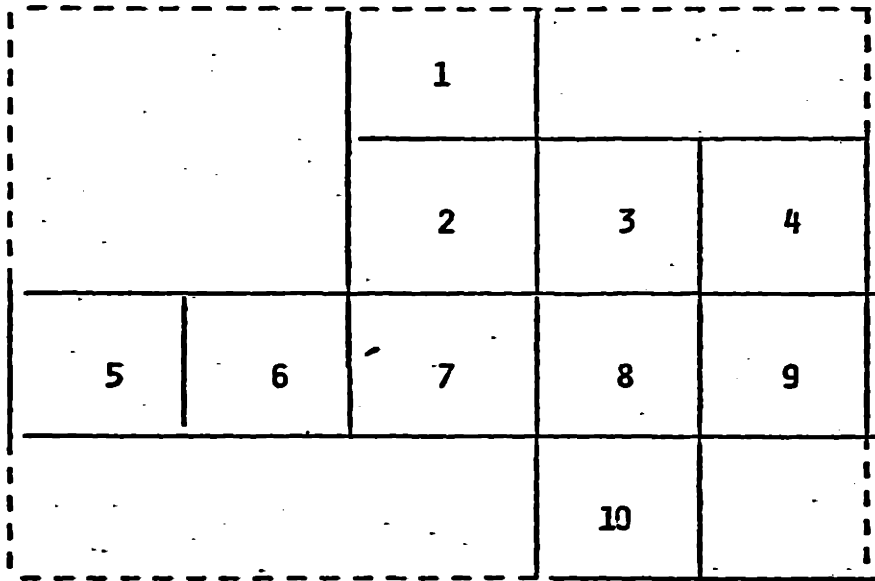
<u>Variable</u>	<u>Columns</u>	<u>Description</u>
ICROSS	1-4	) see notes below
IDOWN	5-8	

## NOTES

This option is only possible to use when the pattern of channel is rectangular. If this is the case, ICROSS is the number of columns and IDOWN the number of rows. For example, in the case represented in figure 1, ICROSS should be 4 and IDOWN 3. The maximum value for IDOWN and ICROSS is 20. The channels are sequentially numbered by the computer and the channel boundaries set in the IK, JK arrays; the order is that used to illustrate the case of IMAP = 4 (Card (9d) ).

1	2	3	4
5	6	7	8
9	10	11	12

Figure 1.  
Rectangular Matrix of Channels



**FIGURE 2**  
**Irregular Pattern of Channels**

Required to be present	Card (9b) Where IPILE = 1 and IMAP = 2
FORTRAN READ List	ISTART-IEND
FORTRAN FORMAT	2I4
Read from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
ISTART	1-4	First channel in each row
IEND	5-8	Last channel in each row

**Notes:**

One of these cards should be given for each row. Note that this method could not be used if there were an insert blank channel in any row; for this case use IMAP = 3. The maximum value of IEND is 20 and the maximum number of rows is also 20. If less than 20 rows are to be given, a blank card (or one with two zero) should be given after the last row.

The computer numbers the channels and the boundaries sequentially as illustrated in Figures 1 and 2.

Examples follow:

For Figure 1 the following cards should be inputed:

<u>ISTART</u>	<u>IEND</u>
1	4
1	4
1	4
0	0

For Figure 2 the following cards should be inputed:

<u>ISTART</u>	<u>IEND</u>
3	3
3	5
1	5
4	4
0	0

	Card (9c)
Required to be present	When IPILE = 1 and IMAP = 3
FORTRAN READ list	(MAAP(L), L= 1,20)
FORTRAN FORMAT	20 I4
Read from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
MAAP	1-80	The number of the channels making up a row

**Notes:**

One of these cards should be inputted for each row (maximum 20 rows). The value of MAAP represents the channel number with a zero indicating no channel. If less than 20 cards are to be used, the last should be all zeros (i.e., a blank card). The set of cards represents a map of the channel numbering system, which is thus under the control of the user. The boundary ordering is done by the computer.

**Examples:**

For pattern described in figure 1

1	2	3	4
5	6	7	8
9	10	11	12
0	0	0	0

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For pattern described in figure 2

0	0	1		
0	0	2	3	4
5	6	7	8	9
0	0	0	10	
0				

	Card (9d)
Required to be present	When IPILE = 1 and IMAP = 4
FORTRAN READ list	(IK(L), JK(L), L = 1, NK)
FORTRAN FORMAT	20 I4
Read from subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
IK		} See notes below
JK		

**Notes:**

IK, JK are the channel pairs defining each boundary in turn; NK = number of boundaries specified. The set of numbers are read in, 20 to a card, continuing on as many cards as necessary. They are terminated by a zero; if the final channel number is at the end of a card, the zero must be given on the next card. (Note, the value of NK is not known at the time of reading in IK, JK; it is set to the number of pairs read in). Thus, with IMAP = 4, both channel and boundary numbering are under the control of the user. When listing the subchannel pairs, it is preferable to give the lower number first; this saves the computer reversing the order.



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Card (9d)

Examples 1

For case in figure 2:

1	2	2	3	3	4	2	7	3	8	4	9	5	6	6	7	7	8	8	9
8	10	0	0																

For Case in figure 1:

1	2	2	3	3	4	1	5	2	6	3	7	4	8	5	6	6	7	7
5	9	6	10	7	11	8	12	9	10	10	11	11	12	0	0			

Card (10)

Required to be present	When IPILE = 1
FORTTRAN Read list	JB(L), L = 1,
FORTTRAN FORMAT	20 I4
Read from Subroutine	CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
JB	1-80	List the identification number of the channels making up each "half-boundary", i.e. the boundaries that are split by a line of symmetry.

**Notes:**

Always terminate with a zero. If there are no half boundaries, give a single card with a zero. The parameter FACTOR(K) is set to 1.0 for full boundaries and 0.5 for "half-boundaries".

Card (11)

Required to be present  
FORTRAN READ list

When NODESF &gt; 0

(K FUEL(I), CFUEL(I),  
RFUEL(I), DFUEL(I),  
KCLAD(I), CCLAD(I),  
RCLAD(I), TCCLAD(I),  
HGAP(I), I=1, NFUELT )

FORTRAN FORMAT

16E5.3

Read from subroutine

CARDS4

<u>Variable</u>	<u>Columns</u>	<u>Description</u>
NFUEL	1-5	Fuel thermal conduction $\left( \frac{\text{BTU}}{\text{hrft}^2 \text{ } ^\circ\text{F}} \right)$
CFUEL	6-10	Fuel specific heat $\frac{\text{BTU}}{\text{lb}^{\circ}\text{F}}$
RFUEL	11-15	Fuel Density ( $\text{lb}/\text{ft}^3$ )
DFUEL	16-20	Pellet Diameter (in)
KCLAD	21-25	Cladding thermal conduction $\frac{\text{BTU}}{\text{hrft}^2 \text{ } ^\circ\text{F}}$
CCLAD	26-30	Cladding specific heat $\frac{\text{BTU}}{\text{lb}^{\circ}\text{F}}$
RCLAD	31-35	Cladding density ( $\text{lb}/\text{ft}^3$ )
TCCLAD	36-40	Cladding thickness (in)
HGAP	41-45	Fuel-cladding heat transfer coefficient ( $\text{BTU}/\text{ft}^2\text{-hr}^{\circ}\text{F}$ )

CARD GROUPS 5, 6, 9, 10, 11 AND 12 ARE READ IN BY SUBROUTINE INDATA WITH THE FOLLOWING FORMAT:

CARD GROUP 5, SUBCHANNEL AREA VARIATION TABLE

IF THERE ARE NO AREA VARIATIONS, OMIT THIS CARD GROUP.

READ N2 VALUES OF RELATIVE LOCATION(X/L) WHERE AREA FACTORS ARE GIVEN  
FORMAT (12F5.3). N2 MUST BE GREATER THAN ONE BUT NOT GREATER THAN  
THE PARAMETER ML.

READ N1 SETS OF AREA VARIATION FACTORS(LOCAL AREA/NOMINAL AREA).  
EACH SET CONSISTS OF SUBCHANNEL NUMBER AND N2 AREA VARIATION  
FACTORS, FORMAT(15/(12F5.3)). N1 IS LIMITED BY THE PARAMETER MA.

IF N1 IS ZERO, AREA VARIATIONS ARE DELETED FOR SUCCEEDING CASES.

N3 IS THE NUMBER OF ITERATIONS FOR INSERTING AREA VARIATIONS.

IF N3 IS ZERO OR BLANK, N3 IS SET EQUAL TO 1.

CARD GROUP 6, GAP SPACING VARIATION TABLE

IF THERE ARE NO GAP VARIATIONS, OMIT THIS CARD GROUP.

READ N2 VALUES OF THE RELATIVE LOCATION(X/L) WHERE GAP FACTORS ARE  
GIVEN, FORMAT(12F5.3). N2 MUST BE GREATER THAN ONE BUT NOT GREATER  
THAN THE PARAMETER ML.

READ N1 SETS OF GAP SPACING FACTORS(LOCAL GAP/NOMINAL GAP).

EACH SET CONSISTS OF THE ADJACENT SUBCHANNEL NUMBERS FOR THE GAP  
N2 GAP VARIATION FACTORS, FORMAT(215/(12F5.3)). N1 IS LIMITED BY THE  
PARAMETER MS. IF N1 IS ZERO, GAP VARIATIONS ARE DELETED FOR SUCCEEDING  
CASES.

CARD GROUP 9, CALCULATION VARIABLES

READ IN DIVERSION CROSSFLOW RESISTANCE FACTOR, TURBULENT MOMENTUM FACTOR, BUNDLE LENGTH(IN.), POSITION FROM VERTICAL(DEGREES), NUMBER OF AXIAL NODES, NUMBER OF TIME STEPS, TOTAL TRANSIENT TIME(SECONDS) MAXIMUM NUMBER OF ITERATIONS, ALLOWABLE FRACTION ERROR IN FLOW FORM CONVERGENCE AND TRANSVERSE MOMENTUM PARAMETERS(S/L), FORMAT(4E5.2, 2I5, E5.2, I5, 4E5.2). IF THE NUMBER OF ITERATIONS, ALLOWABLE ERROR AND MOMENTUM PARAMETER ARE BLANK OR ZERO, THE PROGRAM USES 20., 1.E-3, AND .5, RESPECTIVELY.

N1 IS AN OPTION GIVING THE SPATIAL PRINTING INCREMENT. IF N1=1, EVERY STEP IS PRINTED. IF N2=2, EVERY OTHER STEP IS PRINTED, ETC. IF N IS ZERO OR BLANK, THE PROGRAM SETS N1=1.

N2 IS AN OPTION GIVING THE TIME PRINTING INCREMENT AND IS SET UP THE SAME AS N1 ABOVE.

N3 IS A DEBUG PRINT OPTION. IF N3=0, NO DEBUG INFORMATION IS PRINTED IF N3=1, A DEBUG PRINT IS MADE FOR EACH STEP OF THE CALCULATION. IT CAN GENERATE A LOT OF PAPER SO IT IS NOT NORMALLY USED.

CARD GROUP 10, TURBULENT MIXING CORRELATIONS

N1 IS THE OPTION FOR SUBCOOLED MIXING CORRELATIONS. FOR N1<4 READ IN THE CONSTANTS A AND B, FORMAT(2F5.3).

THE OPTIONS ARE --

- N1=0, W/GS=A
- N1=2, W/GS=A\*RE\*\*B
- N1=2, W/GD=A\*RE\*\*B
- N1=3, W/GS=D/AIJ\*A\*RE\*\*B
- N1=4, NEW (BUES) MIXING MODEL IS USED

NOTE THAT BETA = W/GS WHERE W IS THE TURBULENT CROSSFLOW.

N2 IS THE OPTION FOR TWO-PHASE MIXING. IF N2=1, TWO-PHASE MIXING IS THE SAME AS FOR SUBCOOLED CONDITIONS. IF N2 IS GREATER THAN ONE READ IN N2 PAIR OF DATA FOR A TABLE OF TWO-PHASE MIXING DATA. EACH PAIR CONSISTS OF THE STEAM QUALITY AND THE CORRESPONDING VALUE OF BETA. N2 IS LIMITED BY THE PARAMETER MP.

N3 IS THE OPTION FOR THERMAL CONDUCTION MIXING. IF N3=0, NO THERMAL CONDUCTION. IF N3=1, READ IN THE THERMAL CONDUCTION GEOMETRY FACTOR FORMAT (F5.3).

CARD GROUP 11, OPERATING CONDITIONS

READ IN THE OPERATING PRESSURE (PSIA), INLET ENTHALPY (BTU/LB) OR INLET TEMPERATURE (DEG-F), MASS VELOCITY (M-LB/HR-SQ-FT) AND AVERAGE HEAT FLUX (M-BTU/HR-SQ-FT). (6F10.0)

N1 IS THE INLET ENTHALPY OPTION. IF N1=0, INLET ENTHALPY IS GIVEN. IF N1=1, INLET TEMPERATURE IS GIVEN. IF NL=2, READ IN THE INDIVIDUAL SUBCHANNEL INLET ENTHALPIES, FORMAT (12F5.0). IF N1=3, READ IN THE INDIVIDUAL SUBCHANNEL INLET TEMPERATURES, FORMAT (12E5.0).

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N2 IS THE INLET FLOW DISTRIBUTION OPTION. IF N2=0, THE SUBCHANNELS ARE GIVEN THE SAME MASS VELOCITY. IF N2=1, THE INLET FLOW IS DIVIDED TO GIVE EQUAL PRESSURE GRADIENT IN THE SUBCHANNELS. IF N2=2, READ MASS VELOCITY FACTORS FOR EACH SUBCHANNEL, FORMAT(12E.50).

N3, N4, N5 and N6 ARE OPTIONS FOR TRANSIENT FORCING FUNCTIONS. IF ANY OF THESE OPTION NUMBERS ARE ZERO OR BLANK, THE CORRESPONDING FORCING DATA IS NOT READ AND IS EXCLUDE FROM THE CALCULATIONS. EACH OF THESE NUMBERS GIVE THE NUMBER OF PAIRS OF TABULAR DATA TO BE READ FOR EACH FUNCTION. ALL DATA ARE READ AS PAIRS OF TIME (SECONDS) AND RELATIVE VALUE, FORMAT (12E5.0).

N3 IS THE OPTION FOR REFERENCE PRESSURE VERSUS TIME.

N4 IS THE OPTION FOR INLET ENTHALPY OR TEMPERATURE AS A FUNCTION OF TIME DEPENDING ON THE OPTION FOR INLET ENTHALPY OR TEMPERATURE.

N5 IS THE OPTION FOR INLET FLOW VERSUS TIME.

N6 IS THE OPTION FOR HEAT FLUX VERSUS TIME.

CARD GROUP 12, OUTPUT DISPLAY OPTIONS

N1 IS AN OPTION FOR PRINTING ANSWERS.

N1=0, PRINT SUBCHANNEL DATA ONLY.

N1=1, PRINT SUBCHANNEL DATA AND CROSSFLOWS.

N1=2, PRINT SUBCHANNEL DATA AND FUEL TEMPERATURES.

N1=3, PRINT SUBCHANNEL DATA, CROSSFLOWS AND FUEL TEMPERATURES.

N2 IS AN OPTION FOR SUBCHANNEL DATA PRINTOUT. IF N2=0, ALL SUBCHAN DATA ARE PRINTED. IF IT IS CALLED FOR BY N1. FOR N2 GREATER THAN 2 READ IN THE SUBCHANNEL NUMBERS FOR WHICH RESULTS ARE TO BE PRINTED FORMAT(3612).

N3 IS AN OPTION FOR FUEL TEMPERATURE PRINTOUT. IF N3=0, DATA FOR ALL RODS ARE PRINTED IF CALLED FOR BY N1. FOR N3 GREATER THAN ZERO, READ IN N3 ROD NUMBERS FOR WHICH TEMPERATURES ARE TO BE PRINTED, FORMAT(3612). IF CHF DATA IS CALLED FOR BY INPUT OPTION IT IS PRINTED FOR EACH SELECTED ROD PLUS A SUMMARY TO IDENTIFY THE ROD AND CHANNEL WITH THE MINIMUM CHF RATIO.

N4 IS AN OPTION FOR FUEL NODE PRINTOUT. IF N4=0, TEMPERATURES ARE PRINTED FOR EVERY NODE. FOR N4 GREATER THAN ZERO, READ IN N4 NODE NUMBERS TO BE PRINTED, FORMAT(3612).

TO START A CALCULATION, READ A BLANK GROUP CONTROL CARD.

TO STOP THE CALCULATIONS, AFTER FINISHING A CASE, READ A BLANK CASE

\*\*\*\* END OF INPUT INSTRUCTIONS \*\*\*\*

UNITS - ALL COMPUTATIONS ARE DONE USING FT, LB, SEC, BUT AND DEG-F.  
UNIT CHANGES FOR INPUT AND OUTPUT ARE DONE IN THE PROGRAM.



M-57

New INPUT DATA Presentation  
(APPENDIX 12 of Ref. 1)

Card(s)	Type C1	Problem Array Size
Required to be present:		Always
FORTTRAN READ list:		MC MG MN MR MX
FORTTRAN FORMAT:		10I5
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
MC	1-5	I5	> No. of channels (NCHANL) in problem. NCHANL is set from NTHBOX on cards C5-C7, or in the original COBRA format, in Card Group 4.	--
MG	6-10	I5	> No. of gap interconnections[NK] between channels in problem. If this is not know, MG=2*MC is usually adequate but should be checked later. For a BWR, MG may be given as zero, when it is reset to 1 in CORE.	--
MN	11-15	I5	> No. of fuel nodal points in problem. This should be > (NODESF+1) on Card T1. If MN is given as zero, it is reset to 1 in CORE.	--
MR	16-20	I5	> No. of rods (NROD) in problem. For PWR and BWR, NROD=NCHANL, hence MR may be given=MC.	--
MX	21-25	I5	> No. of axial stations in problem. It may be given as NDX (Card C11) as it is increased by 1 immediately after reading in.	--

Notes:

- (1) MC to MX are used to set the array sizes in the dynamic storage, hence they should be set too big rather than too small.
- (2) Note that MC to MX are given in alphabetical order.
- (3) The maximum problem size is limited to 30,000 words by the dimension of the DATA array given in the MAIN program and the value of KMAX given in the CORE subroutine. Users can alter this limit with appropriate changes in their source programs.

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May 1980

Card(s) Type	C2	Maximum Running Time
Required to be present		Always
FORTRAN READ list:		MAXT
FORTRAN FORMAT:		I5, 6E12.6
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
MAXT	1-5	I5	Maximum Running Time, Nominal value is 2000.	C O N T R O L

Card(s) Type	C3	Case Control Card
Required to be present		Always
FORTRAN READ list:		IPILE KASE J1 TEXT
FORTRAN FORMAT:		I1, I4, I5, 17A4
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
IPILE	1	I1	= 0 for simplified method = 1 for PWR = 2 for BWR The value is unimportant if Card Group 20 is selected since it is overwritten on card T1.
KASE	2-5	I4	Run Identification Number -- as in COBRA IIIC. If > 0, calculation continues; if < , calculation stops.
J1	6-10	I5	Printing option for standard COBRA output--as in COBRA IIIC. = 0 print only new input = 1 print entire input = 2 print only operating conditions This option is only effective if NOPRIN = 0, i.e., N1 = 0 on card C4
TEXT	11-78		Alphanumeric information to identify Case.

C3  
CONTROL

Card(s) Type	C4	Select Card Group	20
Required to be present		Always	
FORTRAN READ list:		NGROUP	N1 N2 N3 N4 N5 N6
FORTRAN FORMAT:		7I5	
Read from Subroutine:		INDAT	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NGROUP	1-5	I5	= 20	
N1	6-10	I5	Printing trigger, MOPRIN, set to N1. N1=0, standard COBRA IIIC printing obtained as well as as "new" printout. N1=1, standard COBRA printing suppressed.	COBRA
N2-N6	11-35	I5	Leave blank	

**Notes:**

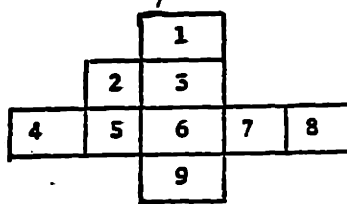
- (1) If NGROUP = 0, this acts as a trigger to stop reading Input Data and to start the hydraulic calculation (.e.g., after card T30).

Card(s) Type	C5	Channel Map parameter
Required to be present		Always
FORTRAN READ list:		IMAP ND1X ND2X
FORTRAN FORMAT:		14I5
Read from Subroutine:		CARD20

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
IMAP	1-5	IS	Selects method of reading channel into array NTHBOX (ND1X, ND2X). IMAP=1, 2 or 3	--
ND1X	6-10	IS	} Size of array NTHBOX, maximum values of each are 25.	--
ND2X	11-15	IS		

If IMAP =	1	2	3
Go to Card	C8	C6	C7

The channel numbering system is contained in the array NTHBOX (ND1X, ND2X) with a zero for each non-channel. The array is later used to define the interaction between adjacent channels. Thus a channel map:



would be represented in NTHBOX (5, 4) as

```

0 0 1 0 0
0 2 3 0 0
4 5 6 7 8
0 0 9 0 0
  
```

If IMAP=1, there are assumed to be ND1X  $\times$  ND2X channels numbered sequentially along each row, and column by column, to give a rectangular matrix. Thus IMAP=1, ND1X=4, ND2X=3 gives a channel map:

```

1 2 3 4
5 6 7 8
9 10 11 12
  
```

For IMAP=2, 3 more complicated channel maps may be specified.

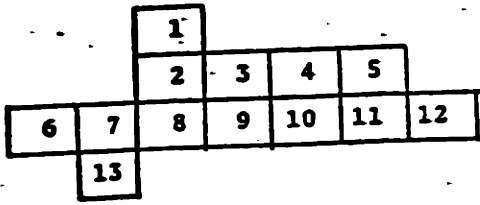
Card(s) Type	C6	Channel Map
Required to be present		Only if IMAP=2
FORTRAN READ list:		ISTART IFIN
FORTRAN FORMAT:	14I5	ND2X Cards of this type read.
Read from Subroutine:	CARD20	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
ISTART	1-5	I5	see below	--
IFIN	6-10	I5	" "	--

A total of ND2X cards of this type are read sequentially, one for each row of the channel map. Each card gives the start and finish of the row. For example, ISTART=3, IFIN=6 would imply a row 0 0 (N+1) (N+2) (N+3) (N+4) 0 0 etc. where channel N was the last channel in the previous row.  
 For IMAP=2, ND1X=7, ND2X=4, cards

- 3 3
- 3 6
- 1 7
- 2 2

would represent a channel map





Card(s) Type	C7	Channel Map
Required to be present		Only if IMAP=3
FORTTRAN READ list:		((NTHBOX (ND1, ND2), ND1=1, ND1X), ND2=1, ND2X)
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		CARD20

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NTHBOX	1-70	14I5	Channel identification number	--

If ND1X>14, the remaining numbers (i.e., 15-ND1X) are read on a continuation card. Note ND1X must not exceed 25. Each row of NTHBOX must start on a new card.

For IMAP=3, ND1X=7, ND2X=4, cards

```

0  0  1
0  0  2  3  4  5
6  7  8  9 10 11 12
0 13

```

would give the same channel map as that illustrating IMAP=2 (see card C6).

IMAP=3 could be used, either to specify a particular numbering system or when there are two channels in the same row separated by a "zero."

In the simplified method, (i.e. IPILE=0) cases as the one represented below may be required to be used. To input this kind of array only IMAP=3 is adequate. The cards needed are illustrated in the figure below.

1	2		3		4
5	6	7	8	9	10
	11	12	13	14	
15	16	17	18	19	20
	21	22	23	24	
25	26		27		28

IMAP=3, ND1X=6, ND2X=6 and

1	2	2	3	3	4
5	6	7	8	9	10
5	11	12	13	14	10
15	16	17	18	19	20
15	21	22	23	24	20
25	26	26	27	27	28

Card(s) Type	C8	Heat Flux
Required to be present	Always	
FORTTRAN READ list:	N1	AFLUX
FORTTRAN FORMAT:	(I5, 13E5.0)	
Read from Subroutine:	CARD20	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
N1	1-5	I5	<p>N1=0; trigger to read average nodal fuel powers after rest of data (Cards C12-14). NAX set to 0, IQP3 set to 0.</p> <p>N1=1; trigger to read average nodal fuel and coolant powers after rest of data (Cards 12-14). NAX set to 0, IQP3 set to 1.</p> <p>N1&gt;2; number of axial points at which heat flux profile will be given on following card C9. Maximum value of N1=30. NAX set to N1, IQP3 set to 2.</p>	
AFLUX	6-10	E5.0	<p>Reactor average heat flux in MBtu /ft<sup>2</sup>h. If N1=0 or 1, the value of AFLUX is irrelevant and may be given as zero.</p>	11

Card(s) Type	C9 Heat Flux Profile
Required to be present	Only if N1 on Card C8>2
FORTRAN READ list:	(Y(I), AXIAL (I), I=1, N1)
FORTRAN FORMAT:	(14E5.0)
Read from Subroutine:	READIN/CARD20

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
Y	1-70	E5.0	Normalised axial position along channel (x/L); $0 < Y < 1.0$	3
AXIAL	1-70	E5.0	Relative heat flux (local/average) corresponding to Y.	3

Card(s) Type	C10	Rod Power Factors
Required to be present	Only required if N1 on Card C8>2	
FORTTRAN READ list:	(RADIAL (I), I=1, NCHANL)	
FORTTRAN FORMAT:	(14E5.0)	
Read from Subroutine:	READIN/CARD20	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
RADIAL	1-70	14E5.0	Relative Rod Power (local/average)	8

NCHANL = No. of channels in problem ( $\leq$ MC in Card C1). It is set to the highest value of the channel map array NTHBOX -- see cards C5-C7.

#### Note

In the simplified method (IPILE=0) some subchannels are lumped together to create one channel, while others are treated as individual subchannels, (see figure below). For those every channel can be visualized as having only one rod that generates the whole power of the channel. In order to reduce the Input Data the power given to such a channel for its rod is specified here, while rods that share their power with several channels, will be described in Card T5a.

This system of entering the Data, reduces the cards required in the old presentation (do not forget that more than 150 channels can be used and only a few of them will be real subchannels) and only introduce the restriction that the lumped channels need to have the same identification number as its rod.

The following example clarifies all these points:

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1=channel $0_1$ =rod	2 $0_2$	3 $0_3$
4 $0_4$		9 $0_9$
10 $0_{10}$	11 $0_{11}$	12 $0_{12}$

For this case, card C10 should have the actual relative rod power for channels 1, 2, 3, 4, zero for 5, 6, 7, 8 and the actual values for 9, 10, 11, and 12.

The power given to channels 5, 6, 7 and 8 from rods  $\overline{13}$ ,  $\overline{14}$ ,  $\overline{5}$ ,  $\overline{6}$ ,  $\overline{7}$ ,  $\overline{8}$ ,  $\overline{15}$ ,  $\overline{16}$ , and  $\overline{17}$  will be specified later in card T5a.

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Card(s) Type	C11	Miscellaneous data
Required to be present	Always	
FORTRAN READ list:	Z	NDX NDT TTIME
FORTRAN FORMAT:	(E5.0, 2I5, 10E5.0)	
Read from Subroutine:	CARD20	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
Z	1-5	E5.0	Channel length (in.)	9
NDX	6-10	I5	Number of axial intervals	9
NDT	11-15	I5	Number of time steps NDT=0; steady state only NDT>0; steady state + transient	9
TTIME	16-20	E5.0	Total duration of transient (sec) The length of each time step is set to TTIME/NDT.	9

Card Type	T1	Channel Indicators
Required to be present:		Always
FORTTRAN READ list:		IPILE NCTYP NGRID NGRIDT NODESF NFXF IFRM IHTM IPROP
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
IPILE	1-5	I5	Iteration trigger=0 for simplified method=1 for PWR, =2 for BWR	
NCTYP	6-10	I5	No. of channel types to be read in; controls reading of cards T2-T4	--
NGRID	11-15	I5	No. of grid positions (maximum=10)	7
NGRIDT	16-20	I5	No. of grid types for each channel (maximum=5)	7
NODESF	21-25	I5	No. of fuel nodes	8
NFXF	26-30	I5	No. of "forced flow" types. Not in use; leave blank	--
IFRM	31-35	I5	Indicator for fuel rod model If IFRM=0, old model is used If IFRM=1, new model is used	--
IHTM	36-40	I5	Indicator for rod-to-coolant heat transfer model. If IHTM=0, old model is used. If IHTM=1, new model for pre-CHF conditions is used. If IHTM=2, new model for pre- and post-CHF conditions is used.	--
IPROP	41-45	I5	Indicator for new fuel rod properties (used when IFRM=1). IPROP=0, constant fuel and clad properties, hgap (gap conductance) constant. IPROP=1, temp-dep. fuel and clad, hgap constant. IPROP=2, temp-dep. fuel and clad, hgap calculated	--

Revised by J. Liu  
May 23, 1977

Revised by J. Loomis  
May 1980



Card Type	T1a
Required to be present:	When NODESF>0 and either IFRM=1 or IHTM>0
FORTTRAN READ list:	EPSF
FORTTRAN FORMAT:	(E8.0)
Read from Subroutine:	CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
EPSF	1-8	E8.0	Fuel rod temperature convergence criterion. If EPSF is given as zero, it is set to the default value 10 <sup>-2</sup> .	--

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May 1980

Card(s) Type	T2	Channel Data for Type I
Required to be present	Always	
FORTTRAN READ list:	N J FRAC GAP HNR DR A B C D	
FORTTRAN FORMAT:	(2I5, 8E5.0)	
Read from Subroutine:	CHAN	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
N	1-5	I5	Friction Indicator to select friction factor for channel (see T10). Nominal value=1, maximum=4.	4
J	6-10	I5	Indicator to define A, B, C, D below (=1 or 2)	--
FRAC	11-15	E5.0	Amount by which channel area, wetted and heated perimeters and number of heated rods are to be multiplied (see below).	--
GAP	16-20	E5.0	Effective rod gap for interconnection between channels (in.). If IPILE=0 this may be given as zero.	4
HNR	21-25	E5.0	No. of heated rods in fuel assembly.	8
DR	26-30	E5.0	Diameter of heated rods (in.)	8
<u>If J=1:</u>				
A	31-35	E5.0	Channel Flow Area (in <sup>2</sup> )	4
B	36-40	E5.0	Channel Wetted perimeter (in.)	4
C	41-45	E5.0	Channel heated perimeter (in.)	4
D	46-50	E5.0	Not used--leave blank	--
<u>If J=2:</u>				
A	31-35	E5.0	No. of unheated (e.g., control) rods	--
B	36-40	E5.0	Diameter of unheated rods (in.)	--
C	41-45	E5.0	Width of square assembly (in.)	--
D	46-50	E5.0	Radius of channel corners (in.)	--

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May 23, 1977

Notes

- (1) In COBRA IIIC, individual cards are read for each channel and rod. For PWR and BWR smeared assemblies, considerable simplification is possible because (a) there is a one-to-one correspondence between channels and rods, hence the data may be given together, and (b) many channels have identical geometries, hence one may give a typical geometry and specify to which channels it applies.
- (2) Channels are of the same type if they are described by the same data on cards T2, T3.
- (3) Cards T2, T3, T4 are read sequentially in a DO Loop I=1, NCTYP. Channels making up Types 2, NCTYP are specified on card T4. The unspecified channels are taken to be of Type 1, hence for economy, Type 1 should be defined as that which contains the majority of the channels.
- (4) The channel area and perimeters may either be given directly (J=1) or calculated from the dimensions of the assembly (J=2).
- (5) These parameters are multiplied by FRAC. Thus, if a line of symmetry divides a channel so that it is a half-channel, the data for a whole channel may be given and FRAC set to 0.5. Alternatively, data for a single channel may be given and FRAC

set to (say) 4.0 to obtain the parameters for a smeared group of 4 channels. If FRAC is given as zero, it is reset to 1.0.

(6) GAP is the "effective" gap between assemblies. For no internal resistance to mixing within an assembly, GAP could be considered to be the gap between individual rods \* the number of gaps. This would be reduced according to the internal resistance model used.

(7) Next card read is:

NCTYP=1	NGRID > 0	Card T 5
	NGRID = 0	Card T 5a
NCTYP>1;	I=1 (i.e., first type)	
	NGRID > 0	Card T3
	NGRID = 0	Card T2 for I=2
NCTYP>1:	I>1 (i.e., subsequent types)	
	NGRID > 0	Card T3
	NGRID = 0	Card T4

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Card(s) Type	T3	Grid Data for Channel Type I
Required to be present		Only if NGRID>0
FORTRAN READ list:		(CDG(L),L=1, NGRIDT)
FORTRAN FORMAT:		(14E5.0)
Read from Subroutine:		CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
CDG	1-70	14E5.0	Single phase grid coefficient for each grid type.	7

Card(s) Type	T4	Channels making up Type I
Required to be present		Only if I>1
FORTRAN READ list:		(JB(L), L=1, N)
FORTRAN FORMAT:		(14I5)
Read from Subroutine:		CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
JB	1-70	I5	Channel Identification Number for Type I	--

Notes:

(1) The channels of Type I are listed on one or more cards. A complete card is read and the numbers up to the first zero are taken as the relevant channels. The zero (or blank) must be given since it acts as a trigger, hence if the last channel number is at the end of a card, a blank card must follow to supply the terminating zero.

(2) Next card read is:

I = NCTYP      Card T5  
 I < NCTYP      Card T2

Card(s) Type	T5	Grid Positions
Required to be present		Only if NGRID>0
FORTRAN READ list:		(GRIDXL(I), IGRID(I), I=1, NGRID)
FORTRAN FORMAT:		(7(E5.0, I5))
Read from Subroutine:		CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
GRIDXL	1-70	E5.0	Fractional distance up channel (x/L) at which each grid is situated, i.e., $0 \leq \text{GRIDXL} \leq 1.0$	7
IGRID	1-70	I5	Grid Type; the coefficients for each type of grid were read by T5.	7

Card(s) Type	T5a	Indicators
Required to be present:		only if IPILE=0
FORTTRAN READ list:		NN11, NN22, NN33, NN44, ITMP
FORTTRAN FORMAT:		(4I5)
Read from Subroutine:		CHAN

<u>Variables</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NN11	1-5	I5	Cards of rod layout data to be read	8
NN22	5-10	I5	Total number of rods	8
NN33	10-15	I5	Number of radial fuel nodes including the cladding	8
NN44	15-20	I5	Total number of fuel types	8
ITMP	21-25	I5	Transverse momentum coupling parameter indicator. Parameters read by card(s) T7a if ITMP=1. No parameters read if ITMP=0.	--

**Note:**

- (1) NN44 should equal 1 if IRFM=1 (on T1) because the new fuel rod model only considers cylindrical geometry.



Card(s) Type	T5b	Rod layout information
Required to be present:		only if IPILE=0 and NN11 > 0
FORTTRAN READ list:		N, I, DR(I), RADIA(I), (LR(I,L), PHI(I,L)), L=1,6)
FORTTRAN FORMAT		(I1, I4, 2E5.0, 6(I3, E7.0))
Read from Subroutine:		CHAN

<u>Variables</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
N	1	I1	Fuel rod type (1)
I	2-5	I4	Identification number of the rod
DR(I)	6-10	E5.0	Rod diameter (in)
RADIA(I)	11-15	E5.0	Relative rod power (rod power/average rod power)
(2) { LR(I,L)		I3	Adjacent channel number
PHI(I,L)		E7.0	Fraction of the rod power to that channel

Then one card for every rod considered is required.

(1) N=1 indicates rod fuel

N=2 indicates plate fuel

(2) This block is repeated 6 times (L=1,6)

Card(s) Type	T6	Fuel temperature data
Required to be present	Only if NODESF>0	
FORTRAN READ list:	KF(I), CF(I), RF(I), DF(I), KC(I), CC(I), C(I), TC(I), HG(I), I=(1,NW44)	
FORTRAN FORMAT:	(14E5.0)	
Read from Subroutine:	CHAN	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
KF	1-5	ES.0	Fuel thermal conductivity (Btu/hr ft <sup>2</sup> F)	8
CF	6-10	ES.0	Fuel specific heat (Btu/lb °F)	8
RF	11-15	ES.0	Fuel density (lb/ft <sup>3</sup> )	8
DF	16-20	ES.0	Pellet diameter (inch)	8
KC	21-25	ES.0	Clad thermal conductivity (Btu/hr ft <sup>2</sup> F)	8
CC	26-30	ES.0	Clad specific heat (Btu/lb °F)	8
RC	31-35	ES.0	Clad density (lb/ft <sup>3</sup> )	8
TC	36-40	ES.0	Clad thickness (inch)	8
HG	41-45	ES.0	Fuel-to-clad heat transfer coef- ficient (Btu/ft <sup>2</sup> hr °F)	8

Note:

(1) Fuel temperature data must be given even when IPROP>0

Note added by J. Loomis  
May 1980

Card Type	T6a
Required to be present:	When NODESF>0 and IFRM=1
FORTTRAN READ list:	NCF, NCC, THG
FORTTRAN FORMAT:	(2I5, 8E5.0)
Read from Subroutine:	CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NCF	1-5	I5	Number of radial clad cells	--
NCC	6-10	I5	Number of radial fuel cells	--
THG	11-15	I5	Gap thickness(in)	--

Added by J. Loomis  
May 1980

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Card Type	T6b
Required to be present:	When NODESF>0 and IFRM=1 and IPROP>0
FOTRAN READ list:	FTD, FPUO2
FOTRAN FORMAT:	(14E5.0)
Read from Subroutine:	CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
FTD	1-5	E5.0	Fraction of theoretical density of fuel	--
EPUO2	6-10	E5.0	PUO2 content, volume fraction	--

Added by J. Loomis  
May 1980

Card Type	T6c
Required to be present:	When NODESF>0, IFRM=1, and IPROP=2
FORTTRAN READ list:	BURN, CPR, EXPR, FPRESS, GRGH, GMIX, PGAS
FORTTRAN FORMAT:	(14E5.0)
Read from Subroutine:	CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
BURN	1-5	E5.0	Burnup, MWD/MTU	--
CPR	6-10	E5.0	Coefficient of fuel pressure on clad for gap conductance model	--
EXPR	11-15	E5.0	Exponent for fuel pressure on clad	--
FPRESS	16-20	E5.0	Fuel pressure on clad for gap conductance model (psia)	--
GRGH	21-25	E5.0	RMS of fuel and clad roughness (in) -- GRG set equal to $1.6 \times 10^{-5}$ in. if GRGH given as 0.	--
GMIX(1)	26-30	E5.0	Mole fraction of helium	--
GMIX(2)	31-35	E5.0	Mole fraction of argon	--
GMIX(3)	36-40	E5.0	Mole fraction of krypton	--
GMIX(4)	46-45	E5.0	Mole fraction of xenon	--
PGAS	46-50	E5.0	Pressure of gas mixture in gap (psia)	--

Note: The four elements of GMIX must sum to 1.0.

Added by J. Loomis  
May 1980

Card(s) Type	T6d	Effective rod gap for interconnection between channels (in)
Required to be present:		Only if IPILE=0
FORMAT READ list:		(GAPREC(I), I=1, NK) where NK is the total number of gap interconnections
FORTTRAN FORMAT:		14E5.0.
Read from Subroutine:		CHAN

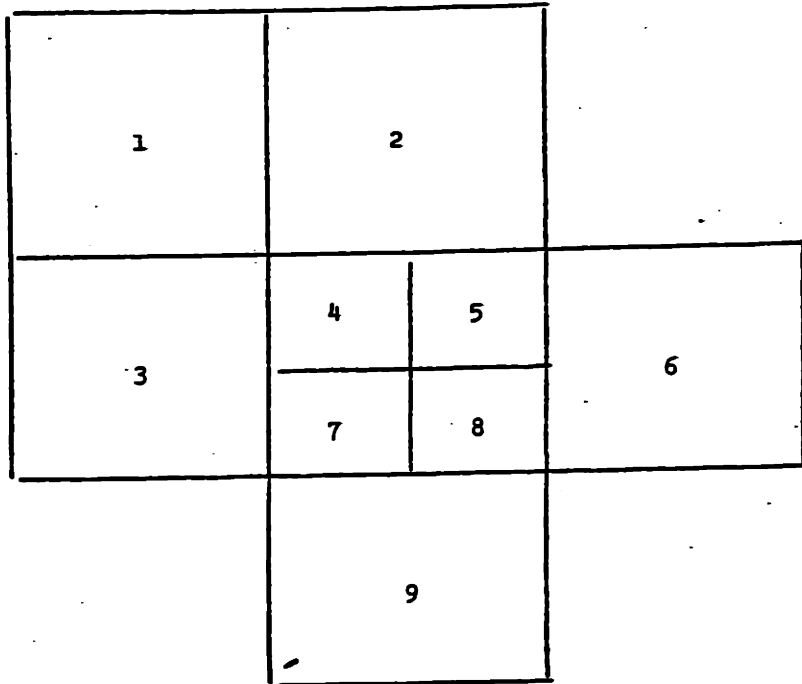
<u>Variable</u>	<u>Columns</u>	<u>Description</u>
GAPREC	1-70	Effective rod gap for interconnection between channels (in)

Notes

In order to give to each boundary its gap these gaps should be inputted in the same order as the boundaries are established. Then a few words are required to know how the boundaries are established.

For the following case the boundaries are established for the code as follows:

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Boundary number

1 2 3 4 5 6 7 8 9 10 11 12 13 14

Pair of channels making up each boundary

1-2 1-3 2-4 2-5 3-4 4-5 5-6 4-7 5-8 3-7 7-8 8-6 7-9 8-9

and in general the boundaries are established by going from left to right in each row and from top to bottom between two consecutive rows.

Card(s) Type	T7	Transverse Momentum Coupling Parameters
Required to be present:		When IPILE=0 and ITMP (on card T5a)=1
FORTTRAN READ list:		(FACSL(I), FACSLK(I), I=1,NK)
FORTTRAN FORMAT:		(14E5.0)
Read from subroutine:		CHAN

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
FACSL(I)		E5.0	Coupling parameter for gap I. May be set equal to the ratio of the number of inter-rod gaps at the boundary between the two regions separated by gap I, divided by the number of rows of rods separating the centroids of the two inter-connected regions.	--
FACSLK(I)		E5.0	Second type of coupling parameter. May be set equal to the number of inter-rod gaps at the boundary of the two regions separated by gap I.	--

**Note:** The suggestions given in the above descriptions are for use of the "Weisman approach" for transverse momentum modeling which is discussed in Section III.F. FACSL corresponds to  $(N_g/N_r)_{ij}$  and FACSLK corresponds to  $(N_r)_{ij}$ . The transverse momentum parameters could, alternatively, be used for the "Chiu approach."

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May 1980

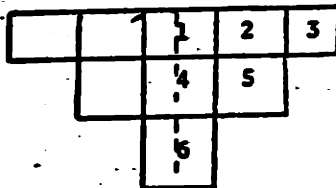


Card(s) Type	T7a	PWR "Half-Boundaries"
Required to be present	Only if IPILE=1	
FORTRAN READ list:	(II(L), JJ(L), L=1, N) where II(N)=0	
FORTRAN FORMAT:	(14I5)	
Read from Subroutine:	CHAN	

Variable	Columns	Format	Description	CG
II	1-70	I5	II(L), JJ(L) are the channel identification numbers which define the Lth "half-boundary."	--
JJ	1-70	I5		--

**Notes:**

(1) A "half-boundary" is one cut by a line of symmetry. In the example below the channel pairs defining the half-boundaries are 1 and 4, 4 and 6.



(2) The list of "half-boundaries" is terminated by a zero. If the list finishes at the end of a card, a blank card should follow to provide the zero-trigger.

(3) If there are no half-boundaries, give a blank card.

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Card(s) Type	T8	Hydraulic Model Indicators
Required to be present	Always	
FORTTRAN READ list:	N1 N2 N3 N4 N5 N6 N7 N8 N9	
FORTTRAN FORMAT:	(14I5)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
N1	1-5	I5	Mixing Indicator	--
N2	6-10	I5	Single Phase Friction Indicator	--
N3	11-15	I5	Two Phase Friction Indicator	--
N4	16-20	I5	Void Indicator	--
N5	21-25	I5	Inlet Flow Indicator	--
N6	26-30	I5	Parameter Indicator	--
N7	31-35	I5	Iteration Indicator	--
N8	36-40	I5	Physical Property Indicator	--
N9	41-45	I5	Coupling parameter in the mixing term of the energy equation	--

Notes:

(1) If all N1-N9 given as zero (i.e., blank card) a preset hydraulic model is obtained and the next card read is T20. If any are given positive, the appropriate part of the model may be changed by giving extra card(s).

(2) The preset model is defined in the card-descriptions following for the appropriate Indicator=0.

(3) N9 = 0 means that no coupling parameter will be used.

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Card(s) Type	T9	Mixing Model
Required to be present:	Only if N1 (on T8) > 0 and N1 < 3	
FORTTRAN READ list:	ABETA BBETA	
FORTTRAN FORMAT:	(14E5.0)	
Read from subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
ABETA	1-5	E5.0	$\beta = W/(G*S) = ABETA*(RE**BBETA)$ if N1=1 $W/(G*D) = ABETA*(RE**BBETA)$ if N1=2 The new mixing model is used if N1=3	10
BBETA	6-10	E5.0		

Notes:

- (1) If N1=0, then ABETA=0.02, BBETA=0.0, and  $W/(G*S) = ABETA*(RE**BBETA)$
- (2) Thermal conduction between channels is suppressed for all N1.
- (3) The new mixing model is described in Section III.D.
- (4) W is the mixing rate  
 RE is an average Reynolds number for the gap  
 S is the gap width  
 D is an average hydraulic diameter

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Card(s) Type	T10	Single Phase Friction Model
Required to be present	Only if $N2(\text{on } T8) > 0$	
FORTTRAN READ list:	NVISCW, (A(J), B(J), C(J), J=1, 4)	
FORTTRAN FORMAT:	(I5, 13E5.0)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NVISCW	1-5	I5	=1, if the wall viscosity correction to the single phase friction factor is required. =0, if not required.	2
A	6-65	E5.0	The single phase friction factor is calculated as $A*(RE**B)+C$ , where RE=Reynolds Number.	2
B	6-65	E5.0		
C	6-65	E5.0		

**Notes:**

(1) The friction factor defined by A(J), B(J), C(J) is applied to those channels with that value of J on card T2. If all channels have the same friction factor, J is given as 1 on card T2 for all channel types and only A(1), B(1), C(1) given on card T10.

(2) If  $N2=0$ , NVISC is set to 0 and the smooth tube friction factor is used, i.e.,  $A=0.184$ ,  $B=-0.2$  and  $C=0.0$  for all  $J=1,4$ .

Card(s) Type	T11	Two Phase Friction Model
Required to be present	Only if N3 (on T8)>0	
FORTTRAN READ list:	J4	
FORTTRAN FORMAT:	(14I5)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
J4	1-5	I5	Two phase friction correlation trigger	2
J4=0	Homogeneous Theory			
-1	Armand			
-2	Baroczy			
-3,4	Not in use			
-5	Polynomial in quality			

Note:

If N3=0, J4 is set to 0.

Card(s) Type	T12	Two phase friction polynomial
Required to be present	Only if J4 (on T11) = 5	
FORTTRAN READ list:	NF	(AF(L), L=1, NF)
FORTTRAN FOPMAT:	(I5, 13E5.0)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NF	1-5	I5	No. of terms in polynomial (max=7)	2
AF	6-40	E5.0	Polynomial coefficients	2

Notes:

(1) The two phase friction multiplier is calculated as

$$\sum_{l=1}^{l=NF} (AF(l) X^{l-1})$$

where X = quality (0 < X < 1)

Card(s) Type	T13	Void Fraction Model
Required to be present	Only if N4 (on T8) > 0	
FORTTRAN READ list:	J2	J3
FORTTRAN FORMAT:	(14I5)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
J2	1-5	I5	Subcooled Void Indicator	2
J3	6-10	I5	Slip Ratio Indicator	2
J2=0	no subcooled void			
-1	Levy subcooled void correlation			
J3=0	Slip Ratio=1			
-1	Armand Slip Ratio Correlation			
-2	Smith Slip Ratio Correlation			
-3,4	Not in use			
-5	Slip ratio given (T14)			
-6	Void fraction as a polynomial in quality (T14)			

Note:

If N4=0, J2 and J3 are both set to 0.

Card(s) Type	T14	Slip Ratio
Required to be present	Only if J3(on T13)=5 or 6	
FORTTRAN READ list:	NV (AV(L), L=1, NV)	
FORTTRAN FORMAT:	(I5, 13E5.0)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NV	1-5	I5	No. of terms in polynomial ( $\leq 7$ )	2
AV	6-40	E5.0	Polynomial coefficients	2

A polynomial  $\sum_{r=1}^{r=Nv} (AV(r)X^r)$  is calculated where X=quality ( $0 \leq X \leq 1$ ).

For J3=5, NV should be set to 1 and only one value of AV read in. The slip ratio is taken as AV(1).

For J3=6, up to 7 values of AV may be read in and the void fraction is calculated as a polynomial in X, namely:

$$\sum_{r=1}^{r=Nv} (AV(r)X^{r-1})$$



Card(s) Type	T15	Inlet Flow Model
Required to be present		Only if N5 (on T8) > 0
FORTRAN READ list:		IG
FORTRAN FORMAT:		(14I5)
Read from Subroutine:		MODEL

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
IG	1-5	I5	Inlet Flow Indicator	11
IG = 0			Inlet mass velocity same for all channels	
IG = 1			Inlet mass velocities for channels calculated to give same inlet pressure gradient	
IG = 2			Inlet mass velocities given (on T16)	

Note

(1) If N5 = 0, IG set to 0.

Card(s) Type	T16	Inlet Flow Distribution
Required to be present		Only if IG (on T15) = 2
FORTTRAN READ list:		(GR(I), I=1, NCHANL)
FORTTRAN FORMAT:		(14E5.0)
Read from Subroutine:		READIN/MODEL

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
GR	1-70	ES.0	Inlet Mass Velocity Ratio (local/ average) for all NCHANL channels	11

Card(s) Type	T17	Parameters
Required to be present:		Only if N6 (on T8) > 0
FORTTRAN READ list:		NCHF KIJ FTM SL THETA
FORTTRAN FORMAT:		(I5, 13E5.0)
Read from Subroutine:		MODEL

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NCHF	1-5	I5	Critical Heat Flux Correlation indicator. (1)	8
KIJ	6-10	E5.0	Cross-Flow Resistance Coefficient, k.	9
FTM	11-15	E5.0	Turbulent Momentum Factor, $f_t$ .	9
SL	16-20	E5.0	Transverse Momentum Factor, S/L	9
THETA	21-25	E5.0	Inclination of channel to vertical (degrees).	9

- (1) If NCHF=0 no CHF calculations are performed  
 If NCHF=1 the BAW-2 correlations is used  
 If NCHF=2 the W-3 correlation is used  
 If NCHF=3 the Hench-Levy correlation is used  
 If NCHF=4 the CISE-4 correlation is used  
 If NCHF=5 the Biasi/Void-CHF correlation is used

Note:

- (1) If N6=0; NCHF set to 0, KIJ to 0.5, FTM to 0.0, SL to 0.5 and THETA to 0.0 (i.e. vertical).  
 (2) If NCHF=5 then IHTM must equal 2 on card T1.

Revised by J. Loomis  
 May 1980

Card(s) Type	T18	Convergence Criteria
Required to be present		Only if N7 (on T8) > 0
FORTTRAN READ list:		NTRIES FERROR
FORTTRAN FORMAT:		(I5, 13 E5.0)
Read from Subroutine:		MODEL

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NTRIES	1-5	I5	Maximum permissible number of hydraulic iterations	9
FERROR	6-10	E5.0	Flow convergence criterion	9

Note

(1) If N7=0, NTRIES set to 20 and FERROR to 0.001.

Card(s) Type	T19	Physical Properties
Required to be present	Only if N8 (on T8) > 0	
FORTTRAN READ list:	NPROP N PH P2	
FORTTRAN FORMAT:	(2I5, 2E5.0)	
Read from Subroutine:	MODEL	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NPROP	1-5	I5	No. of pressure points in physical property table for interpolating between (Minimum=2, Maximum=30).	1
N	6-10	I5	= -1 or 2 (see PH below)	--
PH	11-15	E5.0	<u>N=1</u> , PH=lowest pressure (psia) in problem. <u>N=2</u> , PH=lowest enthalpy (Btu/lb) in problem, from which the lowest pressure is calculated (see below).	--
P2	16-20	E5.0	Highest pressure in problem (psia)	--

### Notes

(1) From this card, a table containing NPROP equi-spaced values of pressure from P1 (see below) to P2 is constructed giving relevant physical properties--calculated from polynomial expressions--at each pressure. Physical properties at intermediate pressures are found by linear interpolation.

(2) It is important that the table spans the physical property range of the problem. For example, with inlet subcooling, the inlet enthalpy would correspond to a pressure lower than the reference value; the pressure would be that at which the enthalpy was the saturation value. Hence the first pressure in the table should be lower than the value corresponding to the lowest steady state or transient enthalpy encountered, so that the other physical properties at that enthalpy may be properly interpolated. If N=1, P1 is given as P1, the lowest pressure in the problem and if N=2, as the lowest enthalpy--the lowest pressure P1 is then calculated from PH.

(3) If N8=0, NPROP is set to 30 and P1, P2 calculated by the computer.

Revised by J.Liu  
May 23, 1977

Card(s) Type	T19a	Coupling parameters
Required to be present		Only if N9 (on T8) > 0
FORTTRAN READ list:		(ENEH(K), K=1,NK) where NK=total number of boundaries
FORTTRAN FORMAT:		14E5.0
Read from Subroutine:		MODEL

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
ENEH	1-70	E5.0	Coupling parameter introduce in the mixing term of the energy conservation equation.

**Note:** The order in which these coupling parameters should be entered is the same as the one described in card T6d for interconnection between channels.

Card(s) Type	T20	Steady State Operating Conditions
Required to be present	Always	
FORTRAN READ list:	IH HIN GIN PEXIT	
FORTRAN FORMAT:	(I5, 13E5.0	
Read from Subroutine:	OPERA	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
IH	1-5	I5	Inlet Enthalpy Indicator	11
HIN	6-10	E5.0	IH=0: Inlet Enthalpy (Btu/lb) IH=1: Inlet Temperature (°F) IH=2,3: HIN not used, set to zero (see T21)	11
GIN	11-15	E5.0	Average <sub>2</sub> Inlet Mass Velocity (Mlb/ft <sup>2</sup> ·hr)	11
PEXIT	16-20	E5.0	System pressure (psia)	11

Card(s) Type	T21	Inlet Enthalpy Distribution
Required to be present		Only if IH = 2 or 3
FORTRAN READ list:		(A(I), I=1, NCHANL)
FORTRAN FORMAT:		(14E5.0)
Read from Subroutine:		READIN/OPERA

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
A	1-70	E5.0	IH=2: Inlet enthalpies for each channel (Btu/lb)	11
			IH=3: Inlet temperatures for each channel ( $^{\circ}$ F)	



Card(s) Type	T22	Transient Indicators
Required to be present		Always
FORTTRAN READ list:		NP NH NG NQ
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		OPERA

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NP	1-5	I5	No. of points at which pressure transient forcing function will be given (T23). Maximum=30	11
NH	6-10	I5	As NP but inlet enthalpy (T24). Maximum=30	11
NG	11-15	I5	As NP but inlet flow (T25). Maximum=30	11
NQ	16-20	I5	As NP but channel power (T25a). Maximum=30	11

### Notes

(1) NQ is only given in COBRA but not in MEKIN (leave NQ blank) as in MEKIN, the transient channel power is obtained from the Neutronics.

(2) If only steady state calculations are required, T22 may be a blank card.

Card(s) Type	T23	Pressure Transient Forcing Function
Required to be present	Only if NP>1 (T22)	
FORTTRAN READ list:	(YP(I), FP(I), I=1, NP)	
FORTTRAN FORMAT:	(14E5.0)	
Read from Subroutine:	READIN/OPERA	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
YP	1-70	E5.0	Time (seconds)	11
FP	1-70	E5.0	Ratio of transient to steady state pressure at time YP	11

Notes

- (1) YP(1), FP(1) should be given as 0.0 and 1.0 respectively.
- (2) The value of FP at a time intermediate between two values of YP is found by linear interpolation.

Card(s) Type	T24	Inlet Enthalpy Transient Forcing Function
Required to be present		Only if NH>1 (T22)
FORTTRAN READ list:		(YH(I), FH(I), I=1, NH)
FORTTRAN FORMAT:		(14E5.0)
Read from Subroutine:		READIN/OPERA

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
YH	1-70	E5.0	Time (seconds)	11
FH	1-70	E5.0	Ratio of transient to steady state enthalpy or temperature (depending on IH--card T20) at time Y.H.	11

Notes

(1) As for card-T23, but YH, FH instead of YP, FP.

Card(s) Type	T25	Inlet Flow Transient Forcing Function
Required to be present		Only if NG > 1 (T22)
FORTTRAN READ list:		(YG(I), FG(I), I=1, NG)
FORTTRAN FORMAT:		(14E5.0)
Read from Subroutine:		READIN/OPERA

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
YG	1-70	E5.0	Time (seconds)	11
FG	1-70	E5.0	Ratio of transient to steady state average mass velocity at time YG	11

Notes

- (1) As for card T23, but YG, FG instead of YP, FP.

Card(s) Type	T25a	Inlet Power Transient Forcing Function
Required to be present	Only if NQ > 1 (T22) and IQP3=2 (C8)	
FORTTRAN READ list:	(YQ(I), FQ(I), I=1, NQ)	
FORTTRAN FORMAT:	(14E5.0)	
Read from Subroutine:	READIN/OPERA	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
YQ	1-70	E5.0	Time (seconds)	11
FQ	1-70	E5.0	Ratio of transient to steady state channel power at time YQ	11

Notes

(1) As for card T23, but YP, FQ instead of YP, FP.

Card(s) Type	T26	"Debug" Option
Required to be present	Always	
FORTTRAN READ list:	KDEBUG	
FORTTRAN FORMAT:	(14I5)	
Read from Subroutine:	TABLES	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
KDEBUG	1-5	I5	"Debug" option =0: normal--no test printing =1: "debug"--with test printing	9

Card(s) Type	T27	Output Printing
Required to be present		Always
FORTTRAN READ list:		NSKIPX NSKIPT NOUT NPCHAN NPROD NPNODE
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		TABLES

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
NSKIPX	1-5	I5	Axial print option =0 or 1: every axial step printed >1 : each (NSKIPX)th step printed	9
NSKIPT	6-10	I5	Time step option As for NSKIPX but time (not axial) steps	9
NOUT	11-15	I5	=0: print channel results only =1: channel + cross flow tables =2: channel + fuel temperature tables =3: channel + cross flow + fuel temperature tables	12
NPCHAN	16-20	I5	=0: all channels printed >1: read in NPCHAN channels to be printed	12
NPROD	21-25	I5	As for NPCHAN but rods instead of channels	12
NPNODE	26-30	I5	As for NPCHAN but radial fuel nodes instead of channels	12

M-112

Card(s) Type	T28	Channels to be printed
Required to be present		Only if NPCHAN (T27) $\geq$ 1
FORTTRAN READ list:		(PRINTC(I), I=1, NPCHAN)
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		TABLES

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
PRINTC	1-70	I5	Identification Number of channels to be printed.	12



M-113

Card(s) Type	T29	Rods to be printed
Required to be present		Only if NPROD (T27) > 1
FORTRAN READ list:		(PRINTR(I), I=1, NPROD)
FORTRAN FORMAT:		(14I5)
Read from Subroutine:		TABLES

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
PRINTR	1-70	I5	Identification Number of rods to be printed.	12

M-114

Card(s) Type	T30	Fuel nodes to be printed
Required to be present		Only if NPNODE (T27) $\geq 1$
FORTTRAN READ list:		(PRINTN(I), I=1, NPNODE)
FORTTRAN FORMAT:		(14I5)
Read from Subroutine:		TABLES

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
PRINTN	1-70	I5	Radial fuel nodes to be printed 1=rod center, (NODESF + 1)=outer clad surface	12

Card(s) Type	C4	End Input Data, start calculation
Required to be present		Always
FORTRAN READ list:		BLANK CARD
FORTRAN FORMAT:		
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
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Note:

At this point in the calculation, control returns to reading Card C4. If NGROUP = 1-12, more Input Data are read in the original COBRA format, these later data overwriting what has already been read in. If NGROUP = 0, calculation starts.

C  
O  
N  
T  
R  
O  
L

Card(s) Type	C12	Nodal Power Multiplier
Required to be present		Only if IQP3 (C8) = 0 or 1.
FORTTRAN READ list:		ZM
FORTTRAN FORMAT:		(8E10.0)
Read from Subroutine:		QPR3

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
ZM	1-10	E10.0	Nodal Power Multiplier	--
ZM= -2.0:			Reset to 1000.0/3.6 (MBtu/hr to Btu/s)	
ZM= -1.0:			Reset to 3413.0/3.6 (MW to Btu/s)	
ZM> 0.0:			ZM unchanged	

The nodal powers given on cards C13, C14 are all multiplied by ZM. This allows, for example, units to be converted.

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May 23, 1977

M-117

Card(s) Type	C13	Fuel Nodal Powers
Required to be present	Only if IQP3 (C8) = 0 or 1	
FORTTRAN READ list:	((QF(I,J), J=1, NDX), I=1, NCHANL)	
FORTTRAN FORMAT:	(8E10.0)	
Read from Subroutine:	QPR3	

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
QF	1-80	8E10.0	Average Fuel Nodal Power for Channel I, axial interval J to (J+1)	--

The power for each channel I (I=1, NCHANL) is read in turn. Each channel-set, i.e., J=1, NDX, starts on a new card, continuing onto the next card if NDX > 8. The units of QF in the calculation are Btu/sec. They may be read in those units (when ZM=1.0 on C12) or converted using ZM. NDX is read on card C11.

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May 23, 1977

M-118

Card(s) Type	C14	Coolant Nodal Powers
Required to be present		Only if IQP3 (C8) = 1
FORTTRAN READ list:		((QC(I,J), J=1, NDX), I=1, NCHANL)
FORTTRAN FORMAT:		(8E10.0)
Read from Subroutine:		QPR3

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
QC(I,J)	1-80	8E10.0	Average Nodal Power deposited in Coolant for channel I, axial interval J to J+1.	--

As for card C13, but QC instead of QF.

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Card(s) Type	C13	<u>Transient Fuel Nodal Power</u>
Required to be present		Only if IQP3 = 0 or 1 and NDT > 1
FORTTRAN READ list:		((QF(I,J), J=1, NDX), I=1, NCHANL)
FORTTRAN FORMAT:		(8E10.0)
Read from Subroutine:		QPR3

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
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Cards C13 and (if IQP3=1) C14 are read for the first transient time step, then both sets of cards for the next time step, etc. until data for all time steps have been given. --

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May 23, 1977

Card(s) Type	C14	<u>Transient</u> Coolant Nodal Power
Required to be present		Only if IQP3= 1 and NDT > 1
FORTTRAN READ list:		((QC(I,J), J=1, NDX), I=1, NCHANL)
FORTTRAN FORMAT:		(8E10.0)
Read from Subroutine:		QPR3

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>
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See last card, "transient" C13.

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Card(s) Type	C3	Next case or End
Required to be present		Always
FORTRAN READ list:		IPILE KASE J1 TEXT
FORTRAN FORMAT:		(I1, I4, I5, 17A4)
Read from Subroutine:		INDAT

<u>Variable</u>	<u>Columns</u>	<u>Format</u>	<u>Description</u>	<u>CG</u>
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See earlier C3

Note

-At the end of the calculation, control returns again to the read statement for card C3.

If KASE > 0; the next case is read.

If KASE = 0 (e.g., a blank card), calculation stops.



```

6  SHCFR,$KCFRC,$KCFRR,$NTYPE,$NWRAP,$NWRPS,$P,$PERIM,$PM,C0B00560
7  $PHI,$PRINT,$PRINT,$SPIN,$SPM,$SPWRF,$SOC,$SOF,$SOPRIM,C0B00570
8  $QUAL,$RADIA,$RHO,$RHOOL,$SP,$ST,$STDUMY,$STINLE,$STROD,C0B00580
9  $SU,$SUH,$SUSAVE,$SUSTAR,$SV,$SVISC,$SVISCH,$SVP,$SVPA,C0B00590
A  $SW,$SHOLD,$SNP,$SNSAVE,$SX,$SKCROS,$SSA,$SBB,$SKPOLD,C0B00600
C  COMMON DATA(1)
   LOGICAL LDAT(1)
   INTEGER IDAT(1)
   EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C  C
C  DIMENSION AFAC(10),GFAC(10)
C  C
C  CALCULATE CHANNEL AREA IF REQUIRED.
   DO 5 I=1,NCHANL
   DATA($A +I)=DATA($AN +I)
   DATA($DHYD +I)=DATA($DHYDN+I)
5  DO 6 K=1,NK
   DATA($CAP +K)=DATA($CAPH +K)
   IF(NAXL.EQ.0) GO TO 101
   DO 100 I=1,NCHANL
   JJ=IDAT($IDARE+I)
   IF(JJ.LT.1) GO TO 100
   DO 10 K=1,NAXL
10  AFAC(K) = AFAC(JJ,K)
   CALL CURVE(FF,(DATA($X+J)/Z),AFAC,AXL,NAXL,ERROR,1)
   IF(ERROR.GT.1) GO TO 1000
   IF(DDT.LT.100.) GO TO 20
   DUMY = FLOAT(ITERAT)/FLOAT(NARAMP)
   IF(DUMY.GT.1.) DUMY = 1.
   IF(FF.LE.0.) GO TO 1000
   FF = 1.-(1.-FF)*DUMY
20  DATA($A +I)=DATA($AN +I)*FF
   DATA($DHYD +I)=DATA($DHYDN+I)*FF
100 CONTINUE
101 IF(JG.NE.1) GO TO 110
C  MODIFY AREA AND HYDRAULIC DIAMETER FOR WIRE WRAPS IN SUBCHANNELS.
   DO 102 I=1,NCHANL
   DATA($A+I)=DATA($A+I)-FLOAT(IDAT($NWRAP+I))*PI*THICK**2*0.25
102 DATA($DHYD+I)=4.*DATA($A+I)/(DATA($PERIM+I)+FLOAT(IDAT($NWRAP+I))*
1  PI*THICK)
C  C
C  CALCULATE GAP SPACING IF REQUIRED.
110 IF(NGXL.EQ.0) GO TO 210
   DO 200 K=1,NK
   L=IDAT($IDGAP+K)
   IF(L.LT.1) GO TO 200
   DO 120 I=1,NGAL
120 CALL CURVE(FF,(DATA($X+J)/Z),GFAC,GAPXL,NGXL,IERROR,1)
   IF(ERROR.GT.1) GO TO 1000
   IF(FF.LE.0.) GO TO 1000
   DATA($GAP +K)=DATA($GAPN +K)*FF

```

```

200 CONTINUE
210 RETURN
1000 ERROR = 9
      RETURN
      END
SUBROUTINE BAROC(IPART,P,Q,CWV,FMULT,PP1)
      COMMON/COSAVE/CORAB
      DIMENSION A1(4),A2(4),CORAB(14,7),COEF(12,8),DAT(12,5,5),X(5)
      GG(7),QQ(14),PP(8),ZNN(3,6)
      DATA I3/6/
      DATA ZNN/1.2621,0.6749,0.073,1.9551,1.0043,0.1097,1.4985,0.8408,
10.0971,0.7965,0.5531,0.0673,0.771,0.5638,0.0713,0.4838,0.4793,
20.0657/
      DATA PP/0.0001,0.001,0.004,0.01,0.03,0.1,0.3,1.0/
      DATA GG/0.0,0.25,0.5,1.0,2.0,3.0,1000.0/
      DATA QQ/0.0,0.001,0.01,0.035,0.05,0.075,0.1,0.15,0.2,
10.3,0.4,0.6,0.8,1.0/
      DATA COEF/2.2,9.2,26.5,47.0,99.0,163.0,376.0,630.0,1300.0,2050.0,
1 4300.0,6600.0,
2 2.15,8.8,22.8,34.2,48.2,70.0,108.0,148.0,240.0,330.0,538.0,760.0,
3 2.08,7.8,16.3,22.8,29.0,36.0,49.5,63.0,86.0,110.0,155.0,203.0,
4 1.59,4.8,9.6,12.4,16.0,20.0,27.0,33.5,43.5,53.0,69.0,85.0,
5 1.12,1.81,3.45,4.7,6.1,7.9,11.0,13.2,17.3,21.2,26.0,30.0,
6 1.04,1.22,1.78,2.05,2.5,2.8,3.6,4.2,5.5,6.5,8.0,9.1,
7 1.01,1.06,1.26,1.36,1.5,1.59,1.77,1.93,2.25,2.48,2.86,3.2,4.2+1.0/CORAB(3,8)
      DATA DAT/1.669,1.669,1.626,1.6,1.59,1.58,1.58,1.58,1.534,
1 1.492,1.362,1.178,
2 1.16,1.158,1.059,1.0,1.21,1.42,1.42,1.324,1.234,1.139,1.103,
3 1.22,1.307,1.355,1.384,1.502,1.36,1.36,1.33,1.34,1.162,1.086,
4 1.1,1.166,1.42,1.572,1.695,1.818,1.818,1.818,1.619,1.619,1.445,
5 1.204,1.07,12+1.0,
6 1.3,1.33,1.311,1.31,1.3,1.3,1.3,1.304,1.308,1.284,1.26,1.2,1.1,
7 1.13,1.25,1.17,1.12,1.148,1.276,1.256,1.236,1.195,1.153,1.11,1.07,
8 1.1,1.15,1.15,1.214,1.21,1.219,1.223,1.24,1.235,1.23,1.13,1.13,1.084,
9 1.078,1.086,1.232,1.32,1.334,1.460,1.472,1.596,1.457,
A 1.318,1.164,1.061,1.2+1.0,60+1.0,
B 0.75,0.74,0.749,0.754,0.752,0.75,0.736,0.722,0.746,0.77,0.82,0.91,
C 0.864,0.66,0.676,0.686,0.704,0.721,0.746,0.75,0.788,0.806,0.86,
D 0.932,0.905,0.88,0.829,0.798,0.805,0.812,0.788,0.764,0.73,
E 0.696,0.705,0.82,
F 0.97,0.912,0.817,0.76,0.73,0.7,0.665,0.63,0.602,0.574,0.574,0.7,
G 12+1.0,0.63,0.61,0.625,0.634,0.634,0.634,0.606,0.598,0.624,0.65,
H 718,836,78,404,501,512,551,59,605,62,667,714,782,88,
I 855,81,741,7,701,702,673,643,593,542,542,59,937,884,
J 769,7,671,642,587,540,493,454,454,58,12+1.0/
      DATA A2/0.220112,-0.299745,0.440706,-0.325823/
      DATA A1/2.46896E-04,1.95508E-01,-3.14163E-02,2.64363E-01/
      DATA X/-8.25483,-5.572754,-2.8647,-1.619488,0.0/
      ZLINE(XA,YA,XC,XB)={(YA-YC)*XB+(YC-YA-YA*XC)}/(XA-XC)
      ZRECT(X1,X2,Y1,Y2,Z11,Z12,Z21,Z22,XX,YY) =
1 ((Y2-YY)*(Z11+(X2-XA) + Z21*(XX-X1))

```

COB01110  
COB01120  
COB01130  
COB01140  
COB01150  
COB01160  
COB01170  
COB01180  
COB01190  
COB01200  
COB01210  
COB01220  
COB01230  
COB01240  
COB01250  
COB01260  
COB01270  
COB01280  
COB01290  
COB01300  
COB01310  
COB01320  
COB01330  
COB01340  
COB01350  
COB01360  
COB01370  
COB01380  
COB01390  
COB01400  
COB01410  
COB01420  
COB01430  
COB01440  
COB01450  
COB01460  
COB01470  
COB01480  
COB01490  
COB01500  
COB01510  
COB01520  
COB01530  
COB01540  
COB01550  
COB01560  
COB01570  
COB01580  
COB01590  
COB01600  
COB01610  
COB01620  
COB01630  
COB01640  
COB01650



```

18 IF(J.EQ.1) J=2
   ZN1 = ALOG((COEF(M,J-1) - 1.0 + QQ(I))*PP(J-1))/ALOG(QQ(I))
   ZN2 = ALOG((COEF(M,J) - 1.0 + QQ(I))*PP(J))/ALOG(QQ(I))
   ZN = ZLINE(ALOG(PP(J-1)),ALOG(ZN1),ALOG(ZN)),ALOG(PP(J)),ALOG(ZN2),PPI)
   ZN = EXP(ZN)
19 CORAB(I,A) = 1.0 - QQ(I) + (QQ(I)*ZN)/PX
22 CONTINUE

C
C   SET CORAB MATRIX USING MASS VELOCITY CORRECTION FACTOR.
   INDI=1.0
   BIT=0.15
30 IF(PPI.LT.X(INDI+1)) GO TO 32
   INDI=INDI+1
   GO TO 30
32 IND2=0.0
   DO 34 K=2,4
34 IF((PPI.GT.(X(K)-BIT)).AND.(PPI.LT.(X(K)+BIT))) IND2=K
   DO 38 I=1,IMAX
   N=I-1
   DO 38 J=1,7
   IF((I.EQ.1).AND.(J.LT.7)) GO TO 35
   IF((I.EQ.IMAX).AND.(J.LT.7)) GO TO 35
   M=J-1
   IF(J.EQ.1) M=J
   IF(J.EQ.7) GO TO 37
   YY=ZLINE(X(INDI),DAT(N,INDI,M),X(INDI+1),DAT(N,INDI+1,M)
   ),PPI)
   IF(IND2.EQ.0.0) GO TO 36
   X1=X(IND2)-BIT
   X2=X(IND2)+BIT
   Y1=ZLINE(X(IND2-1),DAT(N,IND2-1,M),X(IND2),DAT(N,IND2,M),X1)
   Y2=ZLINE(X(IND2),DAT(N,IND2,M),X(IND2+1),DAT(N,IND2+1,M),X2)
   YY=0.5*(ZLINE(X1,Y1,X2,Y2,PPI)+YY)
   GO TO 36
35 YY=1.0
36 CORAB(I,J)=YY*CORAB(I,A)
   GO TO 38
37 CORAB(I,J)=1.0
38 CONTINUE
   RETURN

C
C   INTERPOLATE IN CORAB ARRAY TO FIND MULTIPLIER.
   G=GWV+1.0E-06
41 IF(G.GE.1000.0)G = 1000.0
   INDI=1
42 IF(Q.LE.QQ(INDI)) GO TO 44
   INDI=INDI+1
   GO TO 42
44 CONTINUE
   IND2=1
46 IF(G.LT.GG(IND2)) GO TO 48
   IND2=IND2+1
   GO TO 46
48 G2=GG(IND2)
   G1=GG(IND2-1)

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COB02210  
COB02220  
COB02230  
COB02240  
COB02250  
COB02260  
COB02270  
COB02280  
COB02290  
COB02300  
COB02310  
COB02320  
COB02330  
COB02340  
COB02350  
COB02360  
COB02370  
COB02380  
COB02390  
COB02400  
COB02410  
COB02420  
COB02430  
COB02440  
COB02450  
COB02460  
COB02470  
COB02480  
COB02490  
COB02500  
COB02510  
COB02520  
COB02530  
COB02540  
COB02550  
COB02560  
COB02570  
COB02580  
COB02590  
COB02600  
COB02610  
COB02620  
COB02630  
COB02640  
COB02650  
COB02660  
COB02670  
COB02680  
COB02690  
COB02700  
COB02710  
COB02720  
COB02730  
COB02740  
COB02750

```

G3=G
IF(G,LE,1.0) GO TO 50
G1=1.0/G1
G2=1.0/G2
G3=1.0/G3
50 CONTINUE
C
Z11 = CORAB(IND1-1,IND2-1)
Z12 = CORAB(IND1-1,IND2 )
Z21 = CORAB(IND1 ,IND2-1)
Z22 = CORAB(IND1 ,IND2 )
X1 = QQ(IND1-1)
X2 = QQ(IND1 )
XX = Q
FMULT = ZRECT(X1,X2,G1,G2,Z11,Z12,Z21,Z22,XX,G3)
PPI=ALOG10(EXP(PPI))
RETURN
C
1001 FORMAT(' PRESSURE = ', 1PE15.4, ' OUTSIDE VALID RANGE OF 11.43 TOC0802940
1 3204 PSIA')
END
FUNCTION BVOID(I,J)
C BVOID CALCULATES THE BULK VOID FRACTION GIVEN A QUALITY.
C
C
C
IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL, BBETA , DIA , DT , DK ,
1 ELEV , FERROR, FLO , FTM , GC , GK , GRID , HSURF , MF ,
2 HFG , HG , I2 , I3 , TERROR, IQP3 , ITERAT, J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , MDEBUG, KF , KIJ ,
4 NAFAC, NARAMP, NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRIDT, NGTYPE , NGXL , NK , NNODES , NODESF, NPROP ,
6 NRAMP , NROD , NSCBC , NV , NVISCM, PI , PITCH , POWER , PREF ,
7 QAX , RHDF , RHOG , SIGMA , SL , TF , TFLUID, THETA , THICK ,
8 UF , VF , VFG , VG , Z
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BK(30) , CC(4) , CCLAD(2) , CFUEL(2) , OFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HMG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KFUEL(30) , NCH(10) , NGAP(9) ,
4 PPI(30) , RCLAD(2) , RFUEL(2) , SSGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , Y(30) , Y(30)
C
C
LOGICAL GRID
REAL KIJ , KF , KMF , KCLAD , KFUEL
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MK ,
1 SA , SAA , SAC , SALPHA, SAN , SANSWE, SB ,
2 $CHAN, $CD , $CHFR , $CON , $COND , $CP , $D , $DC , $DFDX ,
3 $DIX , $DHYD , $DIST , $DPOX , $DPK , $DUR , $DR , $F ,
4 $FACTO, $FDIV , $FINLE, $FLUX , $FMULT, $FOLD , $FSP , $FSPLI, $FKFLO,
5 $GAP , $GAPN , $GAPS , $H , $HFILM, $HINLE, $HOLD , $HPERI, $IDARE ,
6 $IDFUE, $IDGAP, $IK , $JBOIL, $JK , $SLC , $LENGT, $LOCA , $LR ,

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C0802760
C0802770
C0802780
C0802790
C0802800
C0802810
C0802820
C0802830
C0802840
C0802850
C0802860
C0802870
C0802880
C0802890
C0802900
C0802910
C0802920
C0802930
C0802940
C0802950
C0802960
C0802970
C0802980
C0802990
C0803000
C0803010
C0803020
C0803030
C0803040
C0803050
C0803060
C0803070
C0803080
C0803090
C0803100
C0803110
C0803120
C0803130
C0803140
C0803150
C0803160
C0803170
C0803180
C0803190
C0803200
C0803210
C0803220
C0803230
C0803240
C0803250
C0803260
C0803270
C0803280
C0803290
C0803300

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LOGICAL GRID
REAL.  KIJ, KF, KMF, KCLAD, KFUEL
C
C
COMMON /COBRA3/  MA, MC, MG, MN, MR, MS, MX
      $$$, SA, SAA, SAC, SALS, SALSME, SB
1  SCCIAM, SCD, SCHFR, SCGN, SCND, SCD, SDFDX
2  SDHX, SDHYD, SDYDN, SDIST, SDPX, SDPK, SDUR, SDR, SF
3  SFACID, SFDIV, SFINLE, SFLOX, SFULT, SFOLD, SFSP, SF SPLIT, SFAFLO, COB03940
4  SGAP, SGAPN, SGAPS, SH, SHFILM, SHINLE, SHOLD, SHPERI, SIDARE, COB03950
5  $IDFUE, $IDGAP, $IK, $JBOIL, $JK, $LC, $LENGT, $LOCA, $LN, COB03960
6  $MCHFR, $MCFRC, $MCFRR, $NTYPE, $NHRAP, $NHRPS, $P, $PERIM, $PH, COB03970
7  $PHI, $PRNTC, $PRNTR, $PRNTN, $PW, $PWRF, $QC, $QF, $QPRIM, COB03980
8  $QUAL, $RADIA, $RHO, $RHOOL, $SSP, $T, $TDUMY, $TINLE, $TROD, COB03990
9  $U, $UH, $USAVE, $USTAR, $V, $VISC, $VISCH, $VP, $VPA, COB04000
A  $W, $WOLD, $WP, $WSAVE, $X, $XCROS, $$A, $$B, $APOLD, COB04010
      COB04020
C
COMMON /LINK4 /IFRM, IHTM, IPROP, NCC, NCF, NDM1, NDS, NGP
      COB04030
C
COMMON /TIMEST/ NT
      COB04040
C
COMMON /REFP/ PO
      COB04050
C
COMMON /PPSV/ PPI
      COB04060
C
COMMON /DATA(1)
      COB04070
LOGICAL LDAT(1)
      COB04080
INTEGER IDAT(1)
      COB04090
EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
      COB04100
EQUIVALENCE (NCHAN, NCHANL)
      COB04110
C
COMMON /LINK2 /CROSS(6), DATE(2), FG(30), FH(30), FP(30), FO(30), FM(9),
      COB04120
1  JM(9), OUTPUT(10), PRINT(12), TEXT(17), TIME(3), YG(30), YH(30), YP(30),
      COB04220
2  YQ(30)
COMMON /LINK3 /DAX, ETIME, GIN, HIN, IB, IG, IN, ISAVE, JUMP, KASE, KT, MAXT,
      COB04230
1  NDT, NDXP1, NFUELT, NG, NH, NJUMP, NOUT, NP, NPCHAN, NPNODE, NPROD, NO, NR,
      COB04240
2  NSKIPT, NSKIPTX, NTRIES, PEXIT, PHTOT, SAVEDT, TIN, TTIME, ZZ
COMMON /ISAVER /ISART
      COB04250
INTEGER SIGNAL(18)
      COB04260
DATA SIGNAL /4HMAIN, 4HDIFF, 4HDVRT, 4HRIX,
      COB04270
14HSCHM, 4HFORC, 4HVOID, 4HSPLT, 4HAREA, 4HCURY, 4HPROP,
      COB04280
24HDCOR, 4HSOLV, 4HHEAT, 4HTEMP, 4HCOL, 4HGAUS, 4HCLJ /
      COB04290
HYDRAULIC CONTROL ( COBRA CARDS MAIN0360-MAIN1820 AND 2340 - 2410) COB04300
      COB04310
C
C
C START SUBCHANNEL FLOW AND ENTHALPY CALCULATIONS.
      COB04320
400  KT = NSKIPT
      COB04330
      IPILE = J7
      DT = SAVEDT
      COB04340
401  DATA(X+J)=DX*FLOAT(J-1)
      COB04350
      NDTPI = NDT+1
      COB04360
      CALL PRINTM (0)
      COB04370
      COB04380
      COB04390
      COB04400

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CC      COB04410
CC      COB04420
CC      COB04430
CC      COB04440
CC      COB04450
CC      COB04460
CC      COB04470
CC      COB04480
CC      COB04490
CC      COB04500
CC      COB04510
CC      COB04520
CC      COB04530
CC      COB04540
CC      COB04550
CC      COB04560
CC      COB04570
CC      COB04580
CC      COB04590
CC      COB04600
CC      COB04610
CC      COB04620
CC      COB04630
CC      COB04640
CC      COB04650
CC      COB04660
CC      COB04670
CC      COB04680
CC      COB04690
CC      COB04700
CC      COB04710
CC      COB04720
CC      COB04730
CC      COB04740
CC      COB04750
CC      COB04760
CC      COB04770
CC      COB04780
CC      COB04790
CC      COB04800
CC      COB04810
CC      COB04820
CC      COB04830
CC      COB04840
CC      COB04850
CC      COB04860
CC      COB04870
CC      COB04880
CC      COB04890
CC      COB04900
CC      COB04910
CC      COB04920
CC      COB04930
CC      COB04940
CC      COB04950

CC      CALL TIMING(ICPU)
CC      TSTART=FLOAT(ICPU)/100.

CC      INITIALIZE FUEL ROD VARIABLES IF NEW FUEL ROD MODEL USED
CC      IF (IFIM.EQ.0) GO TO 409
CC      CALL INTRC

C      START TRANSIENT DO LOOP
C      409 DO 500 NT=1,NOTPI
C          CALL PRINTIM (1)
C          IERROR = 0
C          IF (IQP3.GT.1) GO TO 710
C          CALL QPR3(NCHANL, KASE,TEXT,DATE,TIME,DATA($K+1))
C          710 CONTINUE
C          DT = SAVEDT
C          IF(NT.EQ.1) DT = 1.E+10
C          ETIME = DT*FLOAT(NT-1)
C          ESTABLISH CHANNEL BOUNDARY CONDITIONS AND FORCING FUNCTION VALUES.
C          SET TRANSIENT PRESSURE
C          DUMY = 1.
C          IF(NP.GT.1)
C              1CALL CURVE (DUMY,ETIME,FP,YP,MP,IERROR,1)
C          IF(IERROR.GT.1) GO TO 505
C          PREF = DUMY*PEX1T
C          CALL PROP(1,1)
C          IF(IERROR.GT.1) GO TO 505

C          SET TRANSIENT INLET ENTHALPY
C          DUMY = 1.
C          IF(NH.GT.1)
C              1CALL CURVE (DUMY,ETIME,FH,YH,NH,IERROR,1)
C          IF(IERROR.GT.1) GO TO 505
C          DO 402 I=1,NCHANL
C              DATA($HOLD+I)=DATA($H +I)
C              DATA($H +I)=DATA($HINLE+I)*DUMY
C              IF(IN.EQ.1 .OR. IN.EQ.3)
C                  1CALL CURVE(DATA($H+I),DATA($FINLE+I)*DUMY,MHF,TT,NPROP,IERROR,1)
C          402 CONTINUE

C          SET TRANSIENT INLET FLOW
C          DUMY = 1.
C          IF(NG.GT.1)
C              1 CALL CURVE(DUMY,ETIME,FG,YG,NG,IERROR,1)
C          IF(IERROR.GT.1) GO TO 505
C          IF ( (IPILE.EQ.2) .AND. (NT.GT.1) ) GO TO 404
C          STEADY STATE AND PWR.
C          DO 403 I=1,NCHANL
C              DATA($FOLD+I)=DATA($F+I)
C              DATA($F +I)=DATA($FINLE+I)*DUMY
C          403

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GO TO 407
C/IR. UPDATE INLET FLOW FOR DUMMY AND LAST TRANSIENT.
404 SUMSS = 0.0
SUMTR = 0.0
DO 405 I=1,NCHANL
SUMSS = SUMSS + DATA($FINLE+I)
405 SUMTR = SUMTR + DATA($F+I)
WV = DUMY+SUMSS/SUMTR
DO 406 I=1,NCHANL
DATA($FOLD+I)=DATA($F+I)
406 DATA($F+I) = WV*DATA($F+I)
407 CONTINUE
C
C SET TRANSIENT POWER
DUMV = 1.
IF(NQ.GT.1)
1CALL CURVE (DUMY,ETIME,FQ,YO,NO,IERROR,1)
IF(IERROR.GT.1) GO TO 505
POWER = DUMV
C
C SET BAROCZY PRESSURE DROP ARRAY
IF (J4.EQ.2) CALL BAROC(1,PREF,0.0,0.0,RUB,PPF)
C
C BEGIN ITERATION TO OBTAIN SOLUTION.
DO 430 NN=1,NTRIES
CALL PRINTM (2)
DO 410 I=1,NCHANL
IDAT($NRAP+I)=IDAT($NRPS+I)
ITERAT = NN
CALL SCHEME(JUMP,DATA($AAA+I))
CALL PRINTM (6)
IF(IERROR.GT.1) GO TO 440
CALL TURNS(ICPU)
MTIME=ITIX(FLOAT(ICPU)/100.-TSTART)
IF(MTIME.LT.MAXT) GO TO 429
WRITE(13,102)
GO TO 440
429 IF(JUMP.LT.1 .OR. JUMP.GT.3) GO TO 505
GO TO (430,440,440),JUMP
430 CONTINUE
WRITE(13,22) NTRIES
IERROR = 1
C
C SET CONDITIONS FOR NEXT TIME STEP
440 IF(JUMP.EQ.3) GO TO 441
CALL PRINTM (7)
IF(NJUMP.GT.0) JUMP = 3
IF(NJUMP.NE.2) GO TO 441
REWIND 18
WRITE(18) ((DATA($H+MG*(J-1)),I=1,MG),J=1,MX),
((DATA($P+MC*(J-1)),I=1,MC),J=1,MA),
2 ((DATA($RH0+I*MC*(J-1)),I=1,MC),J=1,MX),
3 ((DATA($F +I*MC*(J-1)),I=1,MC),J=1,MX)
END FILE 18
REWIND 18

```

- COB04960
- COB04970
- COB04980
- COB04990
- COB05000
- COB05010
- COB05020
- COB05030
- COB05040
- COB05050
- COB05060
- COB05070
- COB05080
- COB05090
- COB05100
- COB05110
- COB05120
- COB05130
- COB05140
- COB05150
- COB05160
- COB05170
- COB05180
- COB05190
- COB05200
- COB05210
- COB05220
- COB05230
- COB05240
- COB05250
- COB05260
- COB05270
- COB05280
- COB05290
- COB05300
- COB05310
- COB05320
- COB05330
- COB05340
- COB05350
- COB05360
- COB05370
- COB05380
- COB05390
- COB05400
- COB05410
- COB05420
- COB05430
- COB05440
- COB05450
- COB05460
- COB05470
- COB05480
- COB05490
- COB05500

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441 DD 445 J=1,NDXPI
      DD 443 K=1,NK
      DATA$WOLD+K*MG*(J-1))=
1 DATA$W +K*MG*(J-1))
443 CONTINUE
      DD 444 I=1,NCHARL
      DATA($FOLD +I*MC*(J-1))=DATA($F +I*MC*(J-1))
      DATA($HOLD +I*MC*(J-1))=DATA($H +I*MC*(J-1))
      DATA($RHOLD +I*MC*(J-1))=DATA($RHD +I*MC*(J-1))
444 CONTINUE
445 CONTINUE
      CALL EXPIN
      IF(KT.GE.NSKIPT) KT=0
      IF(ISAVE.GT.0) GO TO 505
      IF(IEPERR.GT.0) GO TO 505
500 CONTINUE
      CALL PRINTIM (B)
C
C END OF PROBLEM, LOOK FOR NEW CASE
      GO TO 990
505 WRITE(13,55) SIGNAL(IEPERR)
      WRITE(13,55) SIGNAL(ISAVE)
990 RETURN
C
22 FORMAT (23#OFAILURE INTEGRATION IN,14,17# ITERATIONS AT X=
1,FB.4,21#)
55 FORMAT (10# ERROR IN ,A6,' ** CALCULATION FOR THIS CASE STOPPED',
1)
102 FORMAT('///' * * * ABNORMAL EXIT THROUGH MAXIMUM TIME * * '///)
      END
SUBROUTINE CARDS1(PP,TT,VVF,VVG,MHF,MHG,UUF,KKF,SSIGMA,N1,I2)
DIMENSION PP(1),TT(1),VVF(1),VVG(1),MHF(1),MHG(1),UUF(1),KKF(1),
1 SSIGMA(1)
      REAL KKF
C
      I2=5
C MEMIN NEW PHYS PROP FROM CARDS OR POLYNOMIALS
      IF (N1.LE.0) GO TO 6
      READ(12,4) (PP(1),TT(1),VVF(1),VVG(1),MHF(1),MHG(1),UUF(1),
1 KKF(1),SSIGMA(1),I=1,N1)
4 FORMAT (E5.2,F5.1,7F10.0)
      RETURN
C
C P2 TO BE HIGHER THAN OPERATING PRESSURE
C N=1,PH TO BE LOWER THAN P FOR H-IN
C N=2,PH TO BE LOWER THAN H-IN.
C N1=NUMBER OF PRESSURE INTERPOLATION STEPS
6 READ(12,8) N,PH,P2,N1
8 FORMAT (15,2F10.3,15)
      P1=PH
      IF(N.EQ.1) GO TO 10
      P1=10.0
      IF(PH.LT.161.3) GO TO 10
      H=0.01*PH;
      P1=6.0*H*H*(H-1.35)/(H-0.35)

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COB05510  
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COB05610  
COB05620  
COB05630  
COB05640  
COB05650  
COB05660  
COB05670  
COB05680  
COB05690  
COB05700  
COB05710  
COB05720  
COB05730  
COB05740  
COB05750  
COB05760  
COB05770  
COB05780  
COB05790  
COB05800  
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COB05950  
COB05960  
COB05970  
COB05980  
COB05990  
COB06000  
COB06010  
COB06020  
COB06030  
COB06040  
COB06050

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10 IF(NI.LT.3) NI=3
A=(P2-P1)/(N1-1)
DO 12 I=1,NI
P=PI*(I-1.0)*A
PP(I)=P
TT(I)=SATTEM(P)
RL=ROLIQ(P)
VVF(I)=1.0/RL
RG=ROVAP(P)
VVG(I)=1.0/RG
H=HLIQ(P)
HHF(I)=H
HHG(I)=HVAP(P)
CALL HAPROP(P,H,CP,UUF(I),KKF(I))
CALL SURTEN(P,RL,RG,SSIGMA(I))
12 CONTINUE
RETURN
END
SUBROUTINE CARD20(NOPRIN)
C
C
IMPLICIT INTEGER (S)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , MF
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , J1 , J2
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ
4 NAFAC , NARAMP , NAX , NAXL , NBBC , NCHANL , NCHF , NDX , NF
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , NPROP ,
6 NQAP , NROD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4 PP(30) , RCLAD(2) , RFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)
C
C
LOGICAL GRID
REAL KIJ , KF , KKF , KCLAD , KFUEL
C
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MK
1 $$$ , SA , SAAA , SAC , SALPHA , SAN , SANSNE , SB
1 SCCHAN , SCD , SCHFR , SCON , SCOND , SCP , SD , SDC , SDFDX
2 SDHX , SDHYO , SDHYDN , SDIST , SDPDX , SDPK , SDUR , SDR , SF
3 SFACTO , SFDIV , SFINLE , SFUX , SFMULT , SFOLD , SFSP , SFSPLI , SFKFLD ,
4 SCAP , SCAPN , SCAPS , SH , SHFILM , SHINLE , SHOLD , SHPERI , SIDARE ,
5 SIDFUE , SIDGAP , SIK , SJBUIL , SJK , SLENGT , SLOCA , SLR ,
6 SWCFER , SWCFRC , SWCFRR , SNTYPE , SNWRAP , SNWRPS , SP , SPERIM , SPH
7 SPHI , SPRINTC , SPRINTR , SPRINT , SPW , SPWF , SOC , SQF , SOPRIM ,
8 SQUAL , SPRINTC , SRHO , SRHOOL , SSP , ST , $TDUMY , $TINLE , $TROD ,
9 SU , $UH , $USAVE , $USTAR , $V , $VISC , $VISCH , $VP , $VPA ,

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```

C      A SW  .SHOLD ,SMP  ,SWSAVE ,SX  ,$XCROS,$$A  ,$$B  ,$XPOLO  COB06610
C      COMMON DATA(1) COB06620
C      LOGICAL LDAT(1) COB06630
C      INTEGER IDAT(1) COB06640
C      EQUIVALENCE (DATA(1),IDAT(1),LDAT(1)) COB06650
C      COB06660
C      COB06670
C      COB06680
C      COB06690
C      COB06700
C      COB06710
C      COB06720
C      COB06730
C      COB06740
C      COB06750
C      COB06760
C      COB06770
C      COB06780
C      COB06790
C      COB06800
C      COB06810
C      COB06820
C      COB06830
C      COB06840
C      COB06850
C      COB06860
C      COB06870
C      COB06880
C      COB06890
C      COB06900
C      COB06910
C      COB06920
C      COB06930
C      COB06940
C      COB06950
C      COB06960
C      COB06970
C      COB06980
C      COB06990
C      COB07000
C      COB07010
C      COB07020
C      COB07030
C      COB07040
C      COB07050
C      COB07060
C      COB07070
C      COB07080
C      COB07090
C      COB07100
C      COB07110
C      COB07120
C      COB07130
C      COB07140
C      COB07150

```

COMMON/LINK3/DXX,ETIME,GIN,HIN,IB,IG,IN,ISAVE,JUMP,KASE,KY,MAXT,  
1 NDT,NDXPI,RFUELT,NG,NH,NUJMP,NDOUT,NP,NFCHAN,NPNODE,NP ROD,NO,NR,  
2 NSKIPT,NSKIPTX,NTRIES,PEXIT,PHTOT,SAVEDT,TIN,TTIME,ZZ  
DIMENSION NTHBOX(25,25)  
DIMENSION CARD(20)

C SIMULATE NEUTRONIC INPUT TO MEKIN ITHO

C

C

WRITE (13,1010)  
DO 2 NDI = 1,20  
DO 2 ND2 = 1,20

2 NTHBOX(ND1,ND2) = 0  
NTHBX = 0  
READ (12,1001) CARD, IMAP, ND1X, ND2X  
WRITE (13,1011) CARD  
IF ( (ND1X.LE.25) .AND. ( ND2X.LE.25) ) GO TO 4  
WRITE (13,1012) ND1X,ND2X  
STOP

C

4 IF (IMAP-2) 6,10,14  
IMAP = 1. RECTANGULAR MATRIX

6 DO 8 ND2 = 1,ND2X  
DO 8 NDI = 1,ND1X  
NTHBX = NTHBX+1

8 NTHBOX(ND1,ND2) = NTHBX  
GO TO 18

C

C IMAP = 2. GIVE START AND END OF EACH ROW.  
DO 12 ND2=1,ND2X  
READ (12,1001) CARD, ISTART, IFIN  
WRITE (13,1013) ND2, CARD  
DO 12 NDI=1,ND1X  
IF ( (ND1.LT.ISTART) .OR. (ND1.GT.IFIN) ) GO TO 12  
NTHBX = NTHBX+1  
NTHBOX(ND1,ND2) = NTHBX  
CONTINUE  
GO TO 18

12

C

C IMAP = 3. READ NTHBOX  
MAXRD = 14  
MPI = MAXRD+1  
MORE = ND1X - MAXRD  
DO 16 ND2 = 1,ND2X  
READ (12,1001) CARD, (NTHBOX(ND1,ND2),ND1=1,MAXRD)  
WRITE (13,1014) ND2, CARD  
IF (MORE.LE.0) GO TO 15  
READ (12,1001) CARD, (NTHBOX(ND1,ND2),ND1=MPI,ND1X)

14

C

```

WRITE (13,1014) ND2, CARD
15 DO 16 ND1=1,ND1X
  IF (NTHBX(ND1,ND2).GT.NTHBXX) NTHBXX=NTHBX(ND1,ND2)
16 CONTINUE
C
C READ HEAT FLUX PARAMETERS.
18 READ (12,1003) CARD, NI, AFLUX
WRITE (13,1015) CARD
IF (NI.GT.1) GO TO 22
  IOP3 = NI
DO 20 I=1,NTHBXX
  DATA($RADIO+1) = 1.0
  GO TO 24
22 NAX = NI
CALL READIN(B,NAX,Y,AXIAL,CARD,2)
CALL READIN(9,NTHBXX,DATA($RADIO+1),CARD,CARD,1)
C
24 READ (12,1004) CARD,Z, NDX, NDT, TTIME
WRITE (13,1016) CARD
C
CALL ITHO(NTHBX,NTHBXX,ND1X,ND2X)
IF (NOPRIN.EQ.0) CALL TIDY
CALL PRECAL
RETURN
C
1001 FORMAT(20A4, T1, 14I5)
1003 FORMAT(20A4, T1, I5, 13E5.0)
1004 FORMAT(20A4, T1, E5.0, 2I5, 10E5.0)
1010 FORMAT(IH1, 42X, 'COBRA INPUT DATA', /, 43X,
1 '-----', /, /, ' NB. DATA READ FROM CARD20 WOULD BE REACARD07450
2 ' OR SET WITH THE NEUTRONICS DATA IN MEKIN', /, /, ' CARD IMAGES', /, /, '.....3.....
3 /, 2X, '-----', /, 32X, '0.....1.....2.....3.....
4.....4.....5.....6.....7.....8'
1011 FORMAT(' IMAP ND1X ND2X', 14X, '...', 20A4, '...', CARD20')
1012 FORMAT(' INPUT DATA ERROR IN CARD20. ND1X, ND2X = ', 2I5,
1 ' IE GREATER THAN 25 FOR EACH ALLOWED')
1013 FORMAT(' ND2=', I3, ' ISTART I FIN', 9X, '...', 20A4, '...', CARD20')
1014 FORMAT(' ND2=', I3, ' NTHBX', 14X, '...', 20A4, '...', CARD20')
1015 FORMAT(' NAX AFLUX', 19X, '...', 20A4, '...', CARD20')
1016 FORMAT(' Z NDX NDT TTIME', 13X, '...', 20A4, '...', CARD20')
C
END
FUNCTION CHF1(N,I,J)
C
C IMPLICIT INTEGER (9)
COMMON /COBRA1/ ABETA, AFLUX, ATOTAL, BBETA, DIA, DT, DX
1 ELEV, FERROR, FLO, FIM, GC, GK, GRID, HSURF, HF
2 HFG, HG, I2, I3, IERROR, IOP3, ITERAT, J1, J2
3 J3, J4, J5, J6, J7, KDEBUG, KF, KI, J
4 HAFAC, NARAMP, NAX, NAXL, NBBC, NCHAN, NCHF, NDX, NF
5 NGAPS, NGRID, NGRIDT, NGTYPE, NGXL, NK, NODES, NODEST, NPROP,
6 NRAMP, NROD, NSCBC, NV, NVISCN, PI, PITCH, POWER, PREF,
7 QAX, RHOF, RHOG, SIGMA, SL, TF, TFLUID, THETA, THICK,
8 UF, VF, VFG, VG, Z

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COB07160  
COB07170  
COB07180  
COB07190  
COB07200  
COB07210  
COB07220  
COB07230  
COB07240  
COB07250  
COB07260  
COB07270  
COB07280  
COB07290  
COB07300  
COB07310  
COB07320  
COB07330  
COB07340  
COB07350  
COB07360  
COB07370  
COB07380  
COB07390  
COB07400  
COB07410  
COB07420  
COB07430  
COB07440  
COB07450  
COB07460  
COB07470  
COB07480  
COB07490  
COB07500  
COB07510  
COB07520  
COB07530  
COB07540  
COB07550  
COB07560  
COB07570  
COB07580  
COB07590  
COB07600  
COB07610  
COB07620  
COB07630  
COB07640  
COB07650  
COB07660  
COB07670  
COB07680  
COB07690  
COB07700

```

C
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
1 AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2 GAPXL(10), GFACT(9,10), GRIDXL(10), HGAP(2), MHF(30), MHG(30),
3 IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9),
4 PP(30), RCLAD(2), RFUEL(2), SIGMA(30), TCLAD(2), UUF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), YI(30)
C
C
LOGICAL GRID
REAL KIJ, KF, KMF, KCLAD, KFUEL
C
C
COMMON /COBRA3/ MA, MC, MG, MN, MR, MS, MX,
1 $$$, SA, SAAA, SAC, SALPHA, SAN, SANSWE, SB,
2 SCCHAN, SCD, SCHFR, SCON, SCOND, SCP, SD, SDC, SDFDX,
3 $OHDX, $DRYD, $DRYDN, $DIST, $DPDX, $DPK, $DUR, $DR, $F,
4 $FACT, $FDIV, $FINLE, $FLUX, $FAULT, $FOLD, $FSP, $FSPLI, $FAPLO,
5 $GAP, $GAPN, $GAPS, $H, $HFILM, $HINLE, $HOLD, $HPERI, $IDARE,
6 $IDFUE, $IDGAP, $IK, $JBOIL, $JK, $SLC, $LENGT, $LOCA, $LR,
7 $MCHFR, $MCFRC, $MCFRR, $NTYPE, $NMRAP, $NMRPS, $P, $PERIM, $PH,
8 $PHI, $PRINTC, $PRINTR, $PRINTN, $PW, $PHRF, $QC, $QF, $QPRIM,
9 $QUAL, $RADIA, $RHOD, $RHODL, $SSP, $T, $TOUMY, $TINLE, $TROD,
A $U, $UH, $USAVE, $USTAR, $V, $VISC, $VISCW, $VP, $VPA,
$W, $WOLD, $WMP, $WSAVE, $X, $XCROSS, $XA, $XB, $XAPOLD
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
C
C
C
BAW-2 CHF CORRELATION
DATA A0, B0, A1, A2, A3, A4, A5, A6, A7, A8, A9 / 1.15509, 4.8844,
1 0.3702E+8, 2.1289E-3, 0.83040, 0.68479E-3, 4.5756E+4, 1.0998E-2,
2 0.71185, 0.20729E-3, 547.49/
REAL KD
DATA A21, A22, A23, KD / 2.9840, 7.82293, 0.45758, 1.02508 /
QA=DATA($A+1)
QP=DATA($PERIM+1)
QF=DATA($FI+MC*(J-1))
QH=DATA($HI+MC*(J-1))
RAT=QF/QA
DE=4.+QA/QP
XX=(QH-HF)/HFG
CHF1=(A0-B0*DE)*(A1*(A2+RAT)**(A3+A4*(PREF-2000.)))
1 -A9*RAT+XX*HFG)/(A5*(A6+RAT)**(A7+A8*(PREF-2000.)))
C AXIAL FLUX CORRECTION FACTOR
FAXIAL = 1,
IF(J.EQ.1) GO TO 10
C=A21*(1.-XX)**A22/(RAT*.0036)**A23
SUM = 0.
J5 = 2
DO 5 JJ=J5, J
5 SUM=SUM+DATA($FLUX+N+MR*(JJ-1))*(EXP(C*DATA($X+JJ)))-

```





```

C0808810
C0808820
C0808830
C0808840
C0808850
C0808860
C0808870
C0808880
C0808890
C0808900
C0808910
C0808920
C0808930
C0808940
C0808950
C0808960
C0808970
C0808980
C0808990
C0809000
C0809010
C0809020
C0809030
C0809040
C0809050
C0809060
C0809070
C0809080
C0809090
C0809100
C0809110
C0809120
C0809130
C0809140
C0809150
C0809160
C0809170
C0809180
C0809190
C0809200
C0809210
C0809220
C0809230
C0809240
C0809250
C0809260
C0809270
C0809280
C0809290
C0809300
C0809310
C0809320
C0809330
C0809340
C0809350

II=IDAT($IK+K)
JJ=IDAT($JK+K)
RSTAR=DATA($RHO+II+MC*(J-1))
RSTAR=DATA($RHO+JJ+MC*(J-1))
IF(DATA($W+K+MG*(J-1)).LT.0.0)
WHIN=ABS(DATA($W+K+MG*(J-1)))
IF(WHIN.LT..001) WHIN = .001
CIJ=KI*WHIN*0.5/GC/RSTAR/GGG/GGG
CIJ=CJ/J/DATA($FACTO+K)**2
RETURN
1000 IERROR = 18
RETURN
END
SUBROUTINE CURVE (FX,X,F,Y,N,J,ISAVE)
DIMENSION F(30), Y(30)
C FX - QUANTITY TO BE FOUND
C X - INDEPENDENT VARIABLE
C F - INPUT ARRAY FOR THE ORDINATE (MONOTONIC WITH Y)
C Y - INPUT ARRAY FOR THE ABCISSA (MONOTONIC INCREASE)
C N - NUMBER OF F(I) OR Y(I) VALUES
C J - ERROR SIGNAL, J=10
C THE INDEX I IS SAVED IN COMMON INDSAV
COMMON/INDSAV/I
DATA I3/6/
1 FORMAT(49H TABULAR LOOKUP FAILED IN SUBROUTINE CURVE, FX = E12.6,
1 6H X = E12.6 / (10E12.4))
IF(ISAVE.LT.1 .OR. ISAVE.GT.2) GO TO 70
GO TO (10,50),ISAVE
10 DO 20 I=1,N
IF(X-Y(I)) 30,15,20
15 IF(I.EQ.N) GO TO 40
20 CONTINUE
GO TO 60
30 IF(I.EQ.1) GO TO 60
40 B = (X-Y(I-1))/(Y(I)-Y(I-1))
50 FX = F(I-1) + B*(F(I)-F(I-1))
RETURN
60 WRITE(19,1) FX,X,(F(I),Y(I),I=1,N)
70 J = 10
RETURN
END
SUBROUTINE DECOMP (NM,IERROR,LMAX,MID,UL,X,B,NK)
DIMENSION UL(NK,1),X(1),B(1)
C SIMPLIFIED VERSION OF DECOMP WITH NO PIVOTING
C STORE DIAGONAL BAND OF AAA MATRIX. POSITION (K,L) IN SQUARE
C ARRAY BECOMES (K,(MID-K+L)) IN NEW ARRAY.
C
N = NM
IF(N.EQ.1) RETURN

```

```

DATA I3/6/
NM1 = N-1
DO 17 K = 1,NM1
PIVOT = UL(K,MID)
KPI = K+1
LIMIT = MIND(N,(K+MID-1))
DO 16 I = KPI,LIMIT
KK = MID+K-1
EM = -UL(I,KK)/PIVOT
UL(I,KK) = -EM
IF (EM) 20,16,20
20 DO 21 J=KPI,LIMIT
J1 = MID-I+J
JK = MID-K+J
21 UL(I,J1) = UL(I,J1) + EM*UL(K,JK)
16 CONTINUE
17 CONTINUE
C
IF (UL(N,MID)) 19,18,19
18 WRITE(13,112)
100 WRITE(13,113) ((UL(K,L),L=1,NN),K=1,NN)
112 FORMAT(7E14.8)
113 FORMAT(54HOSINGULAR MATRIX IN DECOMPOSE. ZERO DIVIDE IN SOLVE.
ERROR = 12
19 RETURN
END
SUBROUTINE DOY(A)
DIMENSION A(2),DATIM(S)
CALL WHEN(DATIM)
A(1)=DATIM(1)
A(2)=DATIM(2)
RETURN
END
SUBROUTINE EXPRIN
C
IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA ,AFLUX ,ATOTAL,BBETA ,DIA ,DT ,DX
1 ELEV ,FERROR,FLO ,FTM ,GC ,GK ,GRID ,HSURF ,HF
2 HFG ,HG ,I2 ,I3 ,IERROR,IQP3 ,ITERAT ,J1 ,J2
3 J3 ,J4 ,J5 ,J6 ,J7 ,KDEBUG,KF ,KIJ ,
4 NAFCT,NARAMP,NAX ,NAXL ,NBBC ,NCHAN ,NCHF ,NDX ,NF
5 NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGXL ,NK ,NODES ,NODESF ,MPROP
6 NRAMP ,NRORD ,NSCBC ,NV ,NVISCV,PI ,PITCH ,POWER ,PREF ,COB09790
7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK ,COB09800
8 UF ,VF ,VFG ,VG ,Z
C
COMMON /COBRA2/ AA(4) ,AF(7) ,AFACT(10,10) ,AV(7) ,AXIAL(30)
1 AXL(10) ,BB(4) ,BX(30) ,CC(4) ,CCLAD(2) ,CFUEL(2) ,DFUEL(2) ,
2 GAPXL(10) ,GFACT(9,10) ,GRIDXL(10) ,HGAP(2) ,HMF(30) ,HMG(30) ,
3 IGRID(10) ,KCLAD(2) ,KFUEL(2) ,KNF(30) ,NCH(10) ,NGAP(9) ,
4 PP(30) ,RCLAD(2) ,RFUEL(2) ,SSIGMA(30) ,TCLAD(2) ,UUF(30) ,
5 VVF(30) ,VVG(30) ,XQUAL(30) ,Y(30) ,TT(30)
C
COB09360
COB09370
COB09380
COB09390
COB09400
COB09410
COB09420
COB09430
COB09440
COB09450
COB09460
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COB09720
COB09730
COB09740
COB09750
COB09760
COB09770
COB09780
COB09790
COB09800
COB09810
COB09820
COB09830
COB09840
COB09850
COB09860
COB09870
COB09880
COB09890
COB09900

```

```

C
C
LOGICAL GRID
REAL KIJ, KF, KKF, KCLAD, KFUEL
C
COMMON /COBRA3/ MA, MC, MG, MN, MR, MS, MX
$$$ ,SA, SAAA, SAC, SALPHA, SAN, SANSWE, SB
1 SCCHAN, SCD, SCHFR, SCON, SCP, SD, SDC, SDFDX, COB09970
2 SDHX, SDHYD, SDHYON, SDIST, SDPK, SDUR, SDR, SF, COB09980
3 SFACTD, SFDIV, SFINLE, SFUX, SFMULT, SFOLD, SFSP, SFSPLI, SFKFLD, COB09990
4 SGAP, SGAPN, SGAPS, SH, SHFILM, SHINLE, SHOLD, SHPERI, SIDARE, COB10000
5 SIDFUE, SIDGAP, SIK, SJOIL, SJK, SLC, SLENGT, SLOCA, SLR, COB10010
6 SMCHFR, SMCFRC, SMCFRR, SMTYPE, SMWRAP, SMWRPS, SP, SPERIM, SPH, COB10020
7 SPHI, SPRTIC, SPRTIR, SPRTIN, SPW, SPWRF, SOC, SOF, SOPRIM, COB10030
8 SQUAL, SRADIA, SRHO, SRHOOL, SSP, ST, STOUNY, STINLE, STROD, COB10040
9 SU, SUH, SUSAVE, SUSTAR, SV, SVISC, SVISCH, SVP, SVPA, COB10050
A SW, SWHOLD, SWP, SWSAVE, SX, SXCROS, SSSA, SSSB, $XPOLD, COB10060
COB10070
COB10080
COB10090
COB10100
COB10110
COB10120
COB10130
COB10140
COB10150
COB10160
COB10170
COB10180
COB10190
COB10200
COB10210
COB10220
COB10230
COB10240
COB10250
COB10260
COB10270
COB10280
COB10290
COB10300
COB10310
COB10320
COB10330
COB10340
COB10350
COB10360
COB10370
COB10380
COB10390
COB10400
COB10410
COB10420
COB10430
COB10440
COB10450
COMMON /LINK2/ CROSS(6), DATE(2), FGI(30), FHI(30), FPI(30), FQ(30), IM(9),
1 JUM(9), OUTPUT(10), PRINT(12), TEXT(17), TIME(3), YG(30), YH(30), YP(30),
2 YQ(30)
COMMON /LINK3/ DX, ETIME, GIN, HIN, IB, IG, IN, ISAVE, JUMP, KASE, KT, MAXT,
1 NDT, NDXP1, NFUEL, NG, NH, NUJMP, NOUT, NP, NPCHAN, NPNODE, NPROD, NO, NR,
2 NSKIPT, NSKIPT, NTRIES, PEXIT, PHTOT, SAVEDT, TIN, TTIME, ZZ
DIMENSION CHFCOR(5), CHFLBL(5)
DATA CHFCOR /4HBAW2, 4HW-3, 4HH-L, 4HC-4, 4HB-VC/
DATA CHFLBL /4HD, BR, 4HDNR, 4HCNFR, 4HCPR, 4HCNFR/
DATA H1, H2, H3, H4, H5 / 1H, 1H, 1H, 4H W(, 4H)WP( /
PRINT OUTPUT (COBRA CARDS MAIN1822 - MAIN2331)
ISAVE = IERROR
IERROR = 0
IF(HCHF.GT.0 .AND. ISAVE.EQ.0) CALL CHF(3, NDXP1)
KT = KT+1
IF(KT.LT.NSKIPT) GO TO 500
CALL TGD(TIME)
C
C
PRINT RESULTS
IF(ETIME.GT.0.) GO TO 457
C
COMPUTE MASS AND ENERGY BALANCE
FLOIN = 0.
FLOOUT = 0.
ENGIN = 0.
ENGTOT = 0.
NDXP1 = NDXP1+1
DO 448 I=1, NCHANL
FLOIN = FLOIN +DATA($F +1)MC*(NDXP1-1))
FLOOUT = FLOOUT +DATA($F +1)MC*(NDXP1-1))
ENGIN = ENGIN +DATA($F +1)*DATA($H+1)

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```

448 ENGOUT=ENGOUT+DATA($F +I*MC*(NDXP1-1))*
      DATA($H +I*MC*(NDXP1-1))
      FLOERR = FLOUT - FLOIN
      ENGADD = AFLUX*Z*PHTOT/.0036
      ENGERR = ENGOUT - ENGIN - ENGADD
      WRITE(13,99) KASE,TEXT,DATE,TIME,FLOIN,ENGIN,FLOUT,ENGADD,FLOERR,
      1 ENGOUT,ENGERR
      C PREPARE CHANNEL EXIT SUMMARY
      J = NDXP1
      DO 450 I=1,NCHANL
      OUTPUT(1) = TF
      IF(DATA($H+I*MC*(J-1)).LT.HF) CALL CURVE(OUTPUT(1),
      1 DATA($H+I*MC*(J-1)),TT,HF,NPROP,IERROR,1)
      OUTPUT(2)=(DATA($H+I*MC*(J-1))-HF)/HFG
      IF(OUTPUT(2).LT.0.) OUTPUT(2) = 0.
      OUTPUT(3)=(RHOF-DATA($H+I*MC*(J-1)))/(RHOF-RHOG)
      IF(OUTPUT(3).LT.0.) OUTPUT(3) = 0.
      OUTPUT(4)=DATA($F+I*MC*(J-1))/DATA($AN+I)*.0036
      WRITE(13,100) I,DATA($H+I*MC*(J-1)),OUTPUT(1),DATA($RHO+I*
      1 MC*(J-1)),OUTPUT(2),OUTPUT(3),
      2 OUTPUT(4)
      450 CONTINUE
      IF(IERROR.GT.1) GO TO 505
      C COMPUTE BUNDLE AVERAGED RESULTS
      452 WRITE(13,25) KASE,TEXT,DATE,TIME
      WRITE(13,101)
      DO 456 J=1,NDXP1,NSKIPX
      SAVE1 = 0.
      SAVE2 = 0.
      SAVE3 = 0.
      SAVE4 = 0.
      DO 454 I=1,NCHANL
      SAVE1=SAVE1+DATA($F+I*MC*(J-1))*DATA($AN+I)
      SAVE2=SAVE2+DATA($H+I*MC*(J-1))*DATA($F+I*MC*(J-1))
      SAVE3=SAVE3+DATA($F+I*MC*(J-1))
      SAVE4=SAVE4+DATA($RHO+I*MC*(J-1))*DATA($AN+I)
      OUTPUT(1)=DATA($X+J)*12.
      OUTPUT(2) = SAVE1/ATOTAL/144.
      OUTPUT(3) = SAVE2/SAVE3
      OUTPUT(4) = TF
      IF(OUTPUT(3).LT.HF) CALL CURVE(OUTPUT(4),OUTPUT(3),TT,HF,NPROP,
      1 IERROR,1)
      IF(IERROR.GT.1) GO TO 505
      OUTPUT(5) = SAVE4/ATOTAL
      OUTPUT(6) = 0.
      IF(OUTPUT(3).GT.HF) OUTPUT(6) = (OUTPUT(3)-HF)/HFG
      OUTPUT(7) = 0.
      IF(OUTPUT(5).LT.RHOF) OUTPUT(7) = (RHOF-OUTPUT(5))/(RHOF-RHOG)
      OUTPUT(8) = SAVE3
      OUTPUT(9) = SAVE3/ATOTAL*.0036
      WRITE(13,81) (OUTPUT(1),I1=1,9)
      456 CONTINUE
      IF(IERROR.GT.1) GO TO 505
      C PRINT CHANNEL AND ROD RESULTS AS DEFINED BY OUTPUT OPTIONS

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COB10460  
COB10470  
COB10480  
COB10490  
COB10500  
COB10510  
COB10520  
COB10530  
COB10540  
COB10550  
COB10560  
COB10570  
COB10580  
COB10590  
COB10600  
COB10610  
COB10620  
COB10630  
COB10640  
COB10650  
COB10660  
COB10670  
COB10680  
COB10690  
COB10700  
COB10710  
COB10720  
COB10730  
COB10740  
COB10750  
COB10760  
COB10770  
COB10780  
COB10790  
COB10800  
COB10810  
COB10820  
COB10830  
COB10840  
COB10850  
COB10860  
COB10870  
COB10880  
COB10890  
COB10900  
COB10910  
COB10920  
COB10930  
COB10940  
COB10950  
COB10960  
COB10970  
COB10980  
COB10990  
COB11000

```

457 DO 460 JJ=1,RPCHAN
      I=IDAT($PRINC+JJ)
      WRITE(13,25) MASE, TEXT, DATE, TIME
      WRITE(13,80) ETIME, I
      WRITE(13,82)
      DO 458 J=1,NDXPI,NSKIPX
        OUTPUT(1)=DATA($X+J)*12.
        OUTPUT(3)=DATA($H+I+MC*(J-1))
        OUTPUT(2)=DATA($P+I+MC*(J-1))/144.
        OUTPUT(4) = TF
        IF( DATA($H+I+MC*(J-1)).LT.HF)CALL CURVE(OUTPUT(4)).
      1 DATA($H+I+MC*(J-1)),TT,HF,NPROP,ERROR,1)
        OUTPUT(5)=DATA($RHO+I+MC*(J-1))
        OUTPUT(6) = 0.
        IF (DATA($H+I+MC*(J-1)).GT.HF) OUTPUT(6)=(
      1 DATA($H+I+MC*(J-1))-HF)/HFG
        OUTPUT(7) = 0.
        IF (DATA($RHO+I+MC*(J-1)).LT.RHOF) OUTPUT(7)=(RHOF-
      1 DATA($RHO+I+MC*(J-1)))/(RHOF-RHOG)
        OUTPUT(8)=DATA($F+I+MC*(J-1))
        OUTPUT(9)=DATA($F+I+MC*(J-1))/DATA($AN+I)*.0036
        WRITE(13,81) (OUTPUT(II),II=1,9)
458 CONTINUE
460 CONTINUE
      IF(ROUT.LT.1) GO TO 499
      IF(ROUT.EQ.2) GO TO 470
      DO 465 M=1,NN,10
        MM = M+9
        IF(RK,LE,MM) MM=MK
        WRITE(13,31) MASE,TEXT,DATE,TIME,H7,(H6,H1,IOAT($IK+K),H2,IOAT($JK+
      1 K),H3,K=M,MM)
        DO 465 J=1,NDXPI,MSKIPX
          XDUMY=DATA($X+J)*12.
        WRITE(13,30) XDUMY,(DATA($M+K+MG*(J-1)),K=M,MM)
465 CONTINUE
      IF(ROUT.EQ.1) GO TO 499
470 IF(INPROD.LT.1) GO TO 4990
      DO 485 NN=1,NPROD
        N=IDAT($PRINTR+NI)
        NDUMY=IDAT($IDFUE+N)
        II=1
        IF(INCHF.GT.0) II=NCHF
        WRITE(13,94) MASE, TEXT, DATE, TIME, ETIME, N, NDUMY,CHFLBL(II),
      1 (H8,IOAT($PRINTN+I),H3,I=1,NPNODE)
        DO 483 J=1,NDXPI,NSKIPX
          XDUMY=DATA($X+J)*12.
        DO 480 II=1,NPNODE
          I=IDAT($PRINTN+II)
          DATA($DUMY+II)=
      1 DATA($IROD+I+MN*(N-1)+MR*(J-1))
          DFUX=DATA($FLUX+N+MR*(J-1))*0.0036
          IF(IDAT($CHAN+NR*(J-1)).EQ.0) DATA($CHFR+N+MR*(J-1))=0.
          IF(NODESF.GT.1) WRITE(13,95) XDUMY,DFUX,DATA($CHFR+N+MR*(J-1)),
      1 IDAT($CCCHAN+NR*(J-1)),(DATA($DUMY+I),I=1,NPNODE)

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COB11010  
COB11020  
COB11030  
COB11040  
COB11050  
COB11060  
COB11070  
COB11080  
COB11090  
COB11100  
COB11110  
COB11120  
COB11130  
COB11140  
COB11150  
COB11160  
COB11170  
COB11180  
COB11190  
COB11200  
COB11210  
COB11220  
COB11230  
COB11240  
COB11250  
COB11260  
COB11270  
COB11280  
COB11290  
COB11300  
COB11310  
COB11320  
COB11330  
COB11340  
COB11350  
COB11360  
COB11370  
COB11380  
COB11390  
COB11400  
COB11410  
COB11420  
COB11430  
COB11440  
COB11450  
COB11460  
COB11470  
COB11480  
COB11490  
COB11500  
COB11510  
COB11520  
COB11530  
COB11540  
COB11550

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IF (MODESF.LT.1) WRITE(13,95) XDUY,DFLUX,DATA(SCHFR+NR*(J-1)), COB11560
1 IDAT(SCCHAN+NR*(J-1)) COB11570
483 CONTINUE COB11580
485 CONTINUE COB11590
490 IF(NCHF.LT.1) GO TO 499 COB11600
WRITE(13,96) KASE,TEXT,DATE,TIME,ETIME,CHFCOR(NCHF),CHFLBL(NCHF) COB11610
DO 4995 J=1,NDXPI,NSKIP COB11620
XDUY=DATA($X+J)*12. COB11630
N= IDAT($CFRR+J) COB11640
DFLUX = 0. COB11650
IF(N.NE.0) DFLUX=DATA($FLUX+NR*(J-1))*0.0036 COB11660
IF(N.EQ.0) DATA($MCHFR+J)=0. COB11670
WRITE(13,97) XDUY,DFLUX,DATA($MCHFR+J),IDAT($MCFRR+J), COB11680
1 IDAT($MCFRC+J) COB11690
4995 CONTINUE COB11700
499 WRITE(13,75) ITERAT COB11710
500 CONTINUE COB11720
505 CONTINUE COB11730
RETURN COB11740
COB11750
COB11760
COB11770
COB11780
C
25 FORMAT(17) CHANNEL RESULTS / COB11790
1 5H CASE15,5X17A4, 9H DATE 2A4,7H TIME 2A4,A1// COB11800
30 FORMAT (F7.1,10F10.5) COB11810
31 FORMAT (68H) DIVERSION CROSSFLOW BETWEEN ADJACENT CHANNELS, W(I,J), COB11820
1 (LB/SEC-FT). COB11830
1 // 5H CASE15, 5X, 17A4, COB11840
29H DATE 2A4,7H TIME 2A4,A1 /// COB11850
3 5X,A1,2X,10(2X,A1,A1,12,A1,12,A1) COB11860
75 FORMAT (// 14H ITERATIONS = 14) COB11870
80 FORMAT (8H TIME = F8.5, 9H SECONDS COB11880
1 20H DATA FOR CHANNEL 13//) COB11890
81 FORMAT (F6.1,F12.2,2F12.2,F10.2,2F9.3,F11.4,F12.4) COB11900
82 FORMAT (' DISTANCE DELTA-P ENTHALPY TEMPERATURE DENSITY COB11910
1 EQUIL VOID FLOW MASS FLUX/' (IN.) (PSI) (MLB/MCOB11920
1 BTU/LB) (DEG-F) (LB/CU-FT) QUALITY FRACTION (LB/SEC) COB11930
1R-FT2') COB11940
94 FORMAT (5H) CASE15,5X17A4,9H DATE 2A4,7H TIME 2A4,A1// COB11950
1 4H TIME = F8.5,9H SECONDS COB11960
2 28H TEMPERATURE DATA FOR ROD 13, COB11970
3 12H, FUEL TYPE 12// COB11980
4 ' DISTANCE FLUX 'A4,' CHANNEL TEMP', COB11990
5 'ERATURE(F)'/,22H (IN.) (MBTU/HR-FT2) 13X,10(4X,A2,12,A1) COB12000
95 FORMAT (F8.1,F9.4,F9.3,14,5X,10(F9.1)) COB12010
96 FORMAT (5H) CASE15,5X17A4,9H DATE 2A4,7H TIME 2A4,A1// COB12020
1 8H TIME = F8.5,9H SECONDS // COB12030
2A7, ' CRITICAL HEAT FLUX SUMMARY'/ COB12040
3 ' DISTANCE FLUX M',A4,' ROD CHANNEL' COB12050
97 FORMAT (F8.1,2F8.3,2I8) COB12060
99 FORMAT (' CHANNEL EXIT SUMMARY RESULTS'/ COB12070
1 5H CASE15,5X17A4, 9H DATE 2A4,7H TIME 2A4,A1// COB12080
2' MASS BALANCE -- ,17X, ,/ COB12090
410X,' ENERGY BALANCE -- ,E12.5,' LB/SEC', COB12100
3,' MASS FLOW IN ,E12.5,' BTU/SEC'./ COB12110
410X,' FLOW ENERGY IN ,E12.5,' LB/SEC'./ COB12120
3' MASS FLOW OUT ,E12.5,' LB/SEC'. COB12130

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410X, ENERGY ADDED ,E12.5, BTU/SEC', /
3, MASS FLOW ERROR ,E12.5, LB/SEC', /
410X, FLOW ENERGY OUT ,E12.5, BTU/SEC', /
449X, ENERGY ERROR ,E12.5, BTU/SEC', //
7, CHANNEL ENTHALPY TEMPERATURE DENSITY EQUIL VOID FLOW
8 MASS FLUX', /
9, (ND.) (BTU/LB) (DEG-F) (LB/FT3) QUALITY FRACTION (LB/SEC) COB12110
1 (MLB/HR-FT2) COB12120
100 FORMAT (I6,2F10.2,F10.2,2F9.3,F10.4,F12.4) COB12130
101 FORMAT (' BUNDLE AVERAGED RESULTS',/) COB12140
END COB12150
SUBROUTINE FIZPRP(IPART,NPROP) COB12160
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30), COB12180
1 AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2), COB12190
2 GAPAL(10), GFACT(9,10), GRIDXL(10), HGAP(2), HHF(30), HHG(30), COB12200
3 IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9), COB12210
4 PP(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30), COB12220
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30) COB12230
C REAL KKF COB12240
C IPART = 1, SET PHYSICAL PROPERTIES COB12250
C IPART = 2, PRINT PHYSICAL PROPERTIES COB12260
C CODING SAME AS FOR COBRA COB12270
C IF (IPART.EQ.2) GO TO 10 COB12280
ENTER WITH NPROP PMAX (=PP(1)) PMIN (=PP(2)) SET IN OPERA OR MODEL COB12290
P1 = PP(1) COB12300
P2 = PP(2) COB12310
6 A = (P2-P1)/FLOAT(NPROP-1) COB12320
DO 8 I=1,NPROP COB12330
P=P1+(I-1.0)*A COB12340
PP(I)=P COB12350
TT(I)=SATTEM(P) COB12360
RL=ROLIQ(P) COB12370
VVF(I)=1.0/RL COB12380
RG=ROVAP(P) COB12390
VVG(I)=1.0/RG COB12400
H=HLIQ(P) COB12410
HHF(I)=H COB12420
HHG(I)=HVAP(P) COB12430
CALL HAPROP(P,H,CP,UUF(I),KKF(I)) COB12440
CALL SURTEN(P,RL,RG,SSIGMA(I)) COB12450
8 CONTINUE COB12460
RETURN COB12470
C 10 WRITE (6,1003) COB12480
WRITE (6,1004) (PP(I),TT(I),VVF(I),VVG(I),HHF(I),HHG(I),UUF(I), COB12490
1 KKF(I),SSIGMA(I),I=1,NPROP) COB12500
RETURN COB12510
1003 FORMAT(////, ' PHYSICAL PROPERTIES', /, 2X, '-----', COB12520
1 //, 4X, 'P', 9X, 'T', 8X, 'VVF', 8X, 'VVG', 8X, 'HF', 8X, 'HG', COB12530
2 7X, 'VISC', 8X, 'KF', 6X, 'SIGMA', /) COB12540
COB12550
COB12560
COB12570
COB12580
COB12590
COB12600
COB12610
COB12620
COB12630
COB12640
COB12650

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GO TO (100,200),J6
C FORCED DIVERSION CROSSFLOW FROM WIRE WRAPS
100 IF(PITCH.LE.0.) GO TO 1000
NN = Z/PITCH
NN = NN+1
DO 115 K=1,NN
  IF(DATA(SX+J).LE.PITCH*FLOAT(K)) GO TO 118
115 CONTINUE
118 PL = K-1
C PL IS THE PITCH LENGTH CONTAINING X(J).
C FIND THE WRAP CROSSINGS IN DX.
DO 130 K=1,NK
  II=IDAT(SIK+K)
  JJ=IDAT(SJK+K)
  DO 130 L=1,6
    IF(DATA(SXCROS+K*MG+(L-1)) 119,130,119)
    IF(DATA(SXCROS+K*MG+(L-1))+(PL)*PITCH
    IF(XC.GT.DATA(SX+J).OR.
    1 XC.LE.DATA(SX+JM)) GO TO 130
    LDAT($DIV+K) = .TRUE.
C ADD AND SUBSTRACT WIRE WRAPS FROM SUBCHANNEL AT EACH WRAP CROSSING.
IF(DATA(SXCROS+K*MG+(L-1)) 120,130,121)
120 IDAT(SHWRAP+II)=IDAT(SNWRAP+II)+1
  IDAT(SHWRAP+JJ)=IDAT(SNWRAP+JJ)-1
  GO TO 123
121 IDAT(SHWRAP+II)=IDAT(SNWRAP+II)-1
  IDAT(SHWRAP+JJ)=IDAT(SNWRAP+JJ)+1
123 DUMY = FLOAT(ITERAT)/FLOAT(NRAMP)
  IF(DUMY.GT.1.) DUMY = 1.
  DATA(SWK+MG*(J-1))=DATA(SGAP+K)*PI*(DIA+THICK)*DATA($DUR+K)/DX
  1 *DUMY
124 IF(DATA(SXCROS+K*MG+(L-1)) 124,130,125)
  DATA(SWK+MG*(J-1))=-DATA(SWK+MG*(J-1))*DATA($F+J+MC*(J-1))/
  DATA($A+JJ)
  DATA(SWK+MG*(J-1))=DATA(SWK+MG*(J-1))*DATA($FACTO+K)
  GO TO 130
125 DATA(SWK+MG*(J-1))=DATA(SWK+MG*(J-1))*DATA($F+II+MC*(J-1))/
  DATA($A+II)
  DATA(SWK+MG*(J-1))=DATA(SWK+MG*(J-1))*DATA($FACTO+K)
130 CONTINUE
  RETURN
200 IF(.NOT.GRID) RETURN
DO 230 K=1,NKK
  IF(ABS(DATA($XFLO+K*MG+(NGTYPE-1)).LT.1.0E-10) GO TO 230
C ZERO FORCED FLOW FRACTION DOES NOT BLOCK THE NATURAL DIVERSION CROSSF
  II=IDAT(SIK+K)
  JJ=IDAT(SJK+K)
  LDAT($DIV+K) = .TRUE.
  IF(NRAMP.LE.0) GO TO 1000
  DUMY = FLOAT(ITERAT)/FLOAT(NRAMP)
  IF(DUMY.GT.1.) DUMY = 1.
  DUMY=DUMY*DATA($XFLO+K*MG+(NGTYPE-1))/DX
  IF(DUMY.GT.0.) DATA(SWK+MG*(J-1))=DUMY*DATA($F+II+MC*(J-1))
  IF(DUMY.LT.0.) DATA(SWK+MG*(J-1))=DUMY*DATA($F+JJ+MC*(J-1))

```

COB13210  
COB13220  
COB13230  
COB13240  
COB13250  
COB13260  
COB13270  
COB13280  
COB13290  
COB13300  
COB13310  
COB13320  
COB13330  
COB13340  
COB13350  
COB13360  
COB13370  
COB13380  
COB13390  
COB13400  
COB13410  
COB13420  
COB13430  
COB13440  
COB13450  
COB13460  
COB13470  
COB13480  
COB13490  
COB13500  
COB13510  
COB13520  
COB13530  
COB13540  
COB13550  
COB13560  
COB13570  
COB13580  
COB13600  
COB13610  
COB13620  
COB13630  
COB13640  
COB13650  
COB13660  
COB13670  
COB13680  
COB13690  
COB13700  
COB13710  
COB13720  
COB13730  
COB13740  
COB13750



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2  GAPXL(10), GFACT(9,10), GRIDXL(10), HGAP(2), HHF(30), HHG(30), COB14310
3  IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9), COB14320
4  PP(30), RCLAD(2), RFUEL(2), SIGMA(30), TCLAD(2), UUF(30), COB14330
5  VVF(30), VVG(30), XQUAL(30), Y(30), TT(30) COB14340
C COB14350
C COB14360
C COB14370
C COB14380
C COB14390
C COB14400
C COB14410
C COB14420
C COB14430
C COB14440
C COB14450
C COB14460
C COB14470
C COB14480
C COB14490
C COB14500
C COB14510
C COB14520
C COB14530
C COB14540
C COB14550
C COB14560
C COB14570
C COB14580
C COB14590
C COB14600
C COB14610
C COB14620
C COB14630
C COB14640
C COB14650
C COB14660
C COB14670
C COB14680
C COB14690
C COB14700
C COB14710
C COB14720
C COB14730
C COB14740
C COB14750
C COB14760
C COB14770
C COB14780
C COB14790
C COB14800
C COB14810
C COB14820
C COB14830
C COB14840
C COB14850

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LOGICAL GRID
REAL KIJ, MF, MKF, KCLAD, KFUEL

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COMMON /COBRA3/ MA ,MC ,MG ,MN ,MR ,MS ,MX ,MY
  $S ,SA ,SAA ,SAC ,SALPHA,SAN ,SANSHE,SB
1  $CCHAN,$CD ,SCHFR ,$CON ,$COND ,$CP ,SD ,SOC ,$OFDX ,
2  $DHDX ,SDHYD ,SDHYD ,SDIST ,SDPDX ,SDPK ,SDUR ,$DR ,$F ,
3  $FACTO,$FDIV ,$FINLE,$FLUX ,$FMULT,$FOLD,$FSP ,$FSPLI,$FXFLO,
4  $GAP ,$GAPN ,$GAPS ,SH ,SHFILM,SHINLE,SHOLD ,$HPERI,$IDARE,
5  $IDFUE,$IDGAP,$IK ,$JBDIL,$JK ,$LC ,$LENGT,$LOCA,$LR ,
6  $MCHFR,$MCFRC,$MCFRR,$MTC,$MTRAP,$MWRPS,$P ,$PERIM,$PH ,
7  $PHI ,$PRNTC,$PRINTR,$PRINTN,$PW ,$SPRF ,SOC ,$OF ,$OPRIM,
8  $SUAL ,$RADIA,$RHO ,$RHOO,$SP ,$T ,$TDUINY,$TINLE,$TPROD ,
9  $U ,$UH ,$USAVE,$USTAR,$V ,$VJSC ,$VJSCW,$VP ,
A  $W ,$WOLD ,$WP ,$WSAVE,$X ,$XCROS,$XA ,$$B ,$$POLD

```

```

COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))

```

```

C COB14530
C COB14540
C COB14550
C COB14560
C COB14570
C COB14580
C COB14590
C COB14600
C COB14610
C COB14620
C COB14630
C COB14640
C COB14650
C COB14660
C COB14670
C COB14680
C COB14690
C COB14700
C COB14710
C COB14720
C COB14730
C COB14740
C COB14750
C COB14760
C COB14770
C COB14780
C COB14790
C COB14800
C COB14810
C COB14820
C COB14830
C COB14840
C COB14850

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```

IF (N+1) 6,4,2
2  IF (DATA(SQUAL+1).GT.0.0) GO TO 6
   SINGLE PHASE AND ENTRY FROM PROP (N=-1)
4  RE=DATA(SF+1+NC*(JJ-1))/DATA(SA+I)*DATA(SDHYD+1)/DATA(SVISC+1)
   IF(RE.LT.2000.) RE = 2000.
   PR=DATA(SCP+1)*DATA(SVISC+1)/DATA(SCON+1)
   HCOOL =0.023*DATA(SCON+1)/DATA(SDHYD+1)*RE**.8*PR**.4
   HCOOL = 0.134*DATA(SCON+1)/DATA(SDHYD+1)*RE**.65*PR**.4
   RETURN
C TWO PHASE AND ENTRY FROM PROP(N=-2)
6  F1 = 3600.0*DATA(SOPRIM+1)/DATA(SHPERI+1)
   IF(F1.LT.0.) F1=ABS(F1)
   DTSAT = 0.072*(F1**0.5)*EXP(-PREF/1260.0)
   IF (N.GE.0) GO TO 8
   HCOOL = DTSAT
   RETURN
8  DTOT = DTSAT + TF - DATA(ST+I)
   HCOOL = FI/(3600.0*DTOT)
   RETURN
END
FUNCTION HLIQ(P)
  MEKIN NEW. AUGUST 1974
  U=ALOG(P)
  IF(P.LE.265.0) GO TO 2
  U=U-7.0

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```

HLIQ=((((((-0.5872871D00)*U+0.11490811D01)*U+0.74153448D01)*U
1+0.1080109D02)*U+0.13091584D02)*U+0.37492429D02)*U
2+0.16078158D03)*U+0.55715337D03
RETURN
2 HLIQ=((((((-0.4771D-04)*U+0.04618D-03)*U-0.533926D-02)*U
1+0.1203737D-01)*U+0.908507D-02)*U-0.6628012D-01)*U
2+0.41031089D-01)*U+0.28766511D-00)*U+0.22225855D01)*U
3+0.33320422D02)*U+0.69795537D02
RETURN
END
C FUNCTION HVAP(P)
MEKIN NEW, AUGUST 1974
U=ALOG(P)
IF(P.LE.450.0) GO TO 2
U=U-7.0
HVAP=((((((-0.37170416D01)*U-0.91118126D01)*U-0.2444781D02)*U
1-0.27217176D02)*U-0.44206696D02)*U-0.46351642D02)*U
2+0.11876082D04
RETURN
2 HVAP=((((((-0.3674D-04)*U-0.5662D-03)*U+0.43507598D-02)*U
1-0.1453504D-01)*U+0.22775919D-01)*U
2+0.85550917D0)*U+0.14228318D02)*U+0.11059625D04
RETURN
END
C SUBROUTINE INPRIN
C
C IMPLICIT INTEGER (S)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , HF
2 HFG , HG , I2 , I3 , IERROR , IQP3 , ITERAT , JI , J2
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ ,
4 NFACT , NARAMP , MAX , MAXL , NBBC , NCHAN , NCHF , NDX , NF
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NNODES , NNODESF , NPROP
6 HRAMP , NROD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
8 UF , VF , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4 PP(30) , RCLAD(2) , REFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)
C
C LOGICAL GRID
REAL KIJ , MF , MKF , KCLAD , KFUEL
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MK
1 $$$ , SA , SAAA , SAC , SALPHA , SAN , SANSWE , SB
1 SCCHAN , SCD , SCHFR , SCON , SCONP , SCP , SD , SDC , SDFDX ,
2 SDHDX , SDHYD , SDHYDN , SDIST , SDPDX , SDPK , SDUR , SDR , SF
3 SFACTO , SFDIV , SFINLE , $FLUX , $FMULT , $FOLD , $FSP , $FSPLI , $FVFLQ , COB14860
COB14870
COB14880
COB14890
COB14900
COB14910
COB14920
COB14930
COB14940
COB14950
COB14960
COB14970
COB14980
COB14990
COB15000
COB15010
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COB15140
COB15150
COB15160
COB15170
COB15180
COB15190
COB15200
COB15210
COB15220
COB15230
COB15240
COB15250
COB15260
COB15270
COB15280
COB15290
COB15300
COB15310
COB15320
COB15330
COB15340
COB15350
COB15360
COB15370
COB15380
COB15390
COB15400

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4  SGAP ,SGAPN ,SGAPS ,SH ,SHFILM,SHINLE,SHOLD ,$PERI,$IDARE, COB15410
5  $IDFUE,$IDGAP,$IK ,SUBOIL,SUK ,SLC ,$LENGT,SLOCA ,$LR , COB15420
6  $VCHFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP,$NWRPS,$P ,SPERIM,$PH , COB15430
7  $PHI ,SPRNTC,$SPRTR,$SPRNTN,$SPW ,SPWRF ,SOC ,SOF ,SOPRIM, COB15440
8  $QUAL ,$RADIA,$RHO ,$RHOOL,$SP ,ST ,STDUMY,$TINLE,$TROD, COB15450
9  $U ,SUH ,$USAVE,$USTAR,$V ,SVISC ,$VISCH,$VPA ,SVPA , COB15460
A  $W ,$WOLD,$WP ,$WSAVE,$X ,$XCRDS,$XA ,$$B ,$XPOLD , COB15470
C  COB15480
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
EQUIVALENCE (NCHAN,NCHANL)
C
LOGICAL PRINT
COMMON/LINK2/CROSS(6),DATE(2),FG(30),FH(30),FP(30),FQ(30),FM(9),
1 JH(9),OUTPUT(10),PRINT(12),TEXT(17),TIME(3),YG(30),YH(30),YP(30), COB15500
2 YQ(30) COB15510
COMMON/LINK3/DXX,ETIME,GIN,HIN,IB,IG,IN,ISAVE,JUMP,KASE,KT,MAKT, COB15520
1 NDT,NDXPI,NFUELT,NG,NH,NJUMP,NOUT,NP,NPCHAN,NPNODE,NPROD,NQ,NR, COB15530
2 NSKIPT,NSKIPIX,NTRIES,PEXIT,PHYTOT,SAVEDT,TIN,TTIME,ZZ COB15540
C  COB15550
COMMON/LINK4/IFRM,IHTM,IPROP,NCC,NCF,NDMI,NDS,NGP COB15560
C
COMMON/FRDATA/BURN,OPR,EFB,EPF,EXPR,FPRESS,FPUO2,FRAC,FTD, COB15600
1 GRX(4),GRCH,PGAS,RADR,RDELT,THC,THG COB15650
C
DATA H1,H2,H3,H4,H5 / 1H(, 1H(, 1H(, 4H W(, 4H)WP( / COB15670
DATA H6, H7, H8 / 1H(, 1H(, 1H(, 2HT( / COB15680
C PRINT INPUT DATA (COBRA CARDS MAIN8840 - MAIN0350) COB15690
C SET UP VARIABLES FOR OUTPUT PRINTOUT COB15700
C
250 DD 251 I=1,NCHANL COB15720
DATA($A +I)=DATA($AN +I) COB15730
DATA($DHYD +I)=DATA($DHYDN+I) COB15740
DD 252 K=1,NK COB15750
DATA($GAP +K)=DATA($GAPN +K) COB15760
IF(NPCHAN.GT.0) GO TO 257 COB15770
NPCHAN = NCHANL COB15780
DD 256 I=1,NCHANL COB15790
IDAT($PRINTC+I)=I COB15800
257 IF(NPROD.GT.0) GO TO 259 COB15820
NPROD = NROD COB15830
DD 258 N=1,NROD COB15840
IDAT($PRINTN)=N COB15850
259 IF(NPNODE.GT.0) GO TO 261 COB15860
NN = NODESF+1 COB15870
NPNODE = NN COB15880
DD 260 I=1,NN COB15890
IDAT($PRINT+I)=I COB15900
C
C OUTPUT OF INPUT DATA COB15910
C
261 IF(.NOT.PRINT(1)) GO TO 265 COB15920
WRITE(13,13) (PP(1),TT(1),VVV(1),VVV(1),VVG(1),HHF(1),HHG(1),UUUF(1), COB15940
COB15950

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1KKF(I),SSICMA(I),I=1,NPROP)
265 IF(.NOT.PRINT(2)) GO TO 270
WRITE(13,28)
DO 266 J=1,4
IF(AA(J).GT.0.,OR.CC(J).GT.0.) WRITE(13,29) J,AA(J),BB(J),CC(J)
266 CONTINUE
IF(NVISCW.EQ.0) WRITE(13,61)
IF(NVISCW.EQ.1) WRITE(13,62)
WRITE(13,44)
IF(J2.EQ.0) WRITE(13,45)
IF(J2.EQ.1) WRITE(13,46)
IF(J3.EQ.0) WRITE(13,47)
IF(J3.EQ.1) WRITE(13,48)
IF(J3.EQ.5) WRITE(13,49) AV(1)
IF(J3.EQ.6) WRITE(13,57) NV,(AV(I),I=1,NV)
IF(J4.EQ.0) WRITE(13,58)
IF(J4.EQ.1) WRITE(13,59)
IF(J4.EQ.5) WRITE(13,60) NF,(AF(I),I=1,NF)
270 IF(.NOT.PRINT(3)) GO TO 275
WRITE(13,6) Y(1),AXIAL(1),I=1,MAX)
275 IF(.NOT.PRINT(4)) GO TO 280
WRITE(13,12)
DO 277 I=1,NCHANL
IF((DATA(SAC+I).LT.9.99).AND.
1 (DATA(SPW+I).LT.9.99)) GO TO 276
WRITE(13,1003) I,IDAT(SHTYPE+I),DATA(SAC+I),DATA(SPW+I),
1 DATA(SPH+I),DATA(SDC+I),(IDAT(SLC+I+MC*(L-1))),
2 DATA($GAPS+I*MG*(L-1)),DATA($DIST+I*MC*(L-1)),L=1,4)
GO TO 277
276 WRITE(13,1004) I,IDAT(SHTYPE+I),DATA(SAC+I),DATA(SPW+I),
1 DATA(SPH+I),DATA(SDC+I),(IDAT(SLC+I+MC*(L-1))),
2 DATA($GAPS+I*MG*(L-1)),DATA($DIST+I*MC*(L-1)),L=1,4)
277 CONTINUE
280 IF(MAXL.LT.1) GO TO 285
IF(.NOT.PRINT(5)) GO TO 285
N=1
NN=10
DO 284 LL=1,4
IF(NN.GT.NAFACT) NN = NAFACT
WRITE(13,19) (HI,NCH(J),H3,J=N,NN)
DO 283 I=1,MAXL
N=N+1
NN=NN+10
IF(NN.GE.NAFACT) GO TO 285
284 CONTINUE
285 IF(NCXL.LT.1) GO TO 290
IF(.NOT.PRINT(6)) GO TO 290
N = 1
NN= 10
DO 289 LL = 1,6
IF(NN.GT.NGAPS) NN=NGAPS
DO 286 M=N,NN
K = NGAP(M)
IM(M)=IDAT($IK+K)

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- COR15960
- COR15970
- COR15980
- COR15990
- COR16000
- COR16010
- COR16020
- COR16030
- COR16040
- COR16050
- COR16060
- COR16070
- COR16080
- COR16090
- COR16100
- COR16110
- COR16120
- COR16130
- COR16140
- COR16150
- COR16160
- COR16170
- COR16180
- COR16190
- COR16200
- COR16210
- COR16220
- COR16230
- COR16240
- COR16250
- COR16260
- COR16270
- COR16280
- COR16290
- COR16300
- COR16310
- COR16320
- COR16330
- COR16340
- COR16350
- COR16360
- COR16370
- COR16380
- COR16390
- COR16400
- COR16410
- COR16420
- COR16430
- COR16440
- COR16450
- COR16460
- COR16470
- COR16480
- COR16490
- COR16500

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286 JMI(M)= IDAT($JK+K)
WRITE (13,20) (M1, JM(M), H2, JM(M), H3, M=N, NN)
DO 287 L=1, NGXL
287 WRITE (13,38) GAPXL(L), (GFACT(M,L), M=N, NN)
N=H+10
NH=NH+10
IF(N, GE, NGAPS) GO TO 290
289 CONTINUE
290 IF(.NOT. PRINT(7)) GO TO 300
IF(J6, EQ, 0) GO TO 300
IF(J6, GT, 1) GO TO 296
PITCH = PITCH*12.
DIA = DIA*12.
THICK = THICK*12.
WRITE(13,69) PITCH, THICK, DIA
PITCH = PITCH/12.
DIA = DIA/12.
THICK = THICK/12.
WRITE(13,70) (K, H1, IDAT($IK+K), H2, IDAT($JK+K), H3, DATA(SOUR+K),
1 DATA($XCROSS+K+MG*(L-1)), L=1, 6), K=1, NK)
WRITE(13,74) (IDAT($WRAP+I), I=1, NCHANL)
GO TO 300
296 WRITE(13,71) (IGRID(I), I=1, NGRID)
WRITE(13,72) (GRIDXL(I), I=1, NGRID)
DO 297 L=1, NGRID
297 WRITE(13,73) L, (I, DATA($CD+I+MG*(L-1)), I=1, NCHANL)
DO 299 I=1, NGRID
I1 = 0
DO 298 K=1, NK
IF(ABS(DATA($XFLO+K+MG*(I-1)))-GT.0) I1=1
298 CONTINUE
IF(I1, EQ, 0) GO TO 299
WRITE(13,76) I, (KK, H1, IDAT($IK+KK), H2, IDAT($JK+KK), H3,
1 DATA($XFLO+KK+MG*(I-1)), KK=1, NK)
299 CONTINUE
300 IF(.NOT. PRINT(8)) GO TO 305
WRITE(13,15) (I, IDAT($DFUE+I), DATA($DR+I),
1 DATA($RADIA+I), (DATA($PHI+I+NR), COBI6880)
1 (L-1), IDAT($LR+I+MR*(L-1)), L=1, 6), I=1, NRDD)
IF(NODESF, LT, 1) GO TO 305
DO 301 I = 1, NFUEL
KFUEL(I) = KFUEL(I)*3600.
KCLAD(I) = KCLAD(I)*3600.
DFUEL(I) = DFUEL(I)*12.
TCLAD(I) = TCLAD(I)*12.
HGAP(I) = HGAP(I)*3600.
301 CONTINUE
WRITE(13,77) NODESF
WRITE(13,78) (J, KFUEL(J), CFUEL(J), RFUEL(J), DFUEL(J), KCLAD(J),
1 KCLAD(J), RCLAD(J), TCLAD(J), MGAP(J), J=1, NFUEL)
DO 302 I = 1, NFUEL
KFUEL(I) = KFUEL(I)/3600.
KCLAD(I) = KCLAD(I)/3600.
DFUEL(I) = DFUEL(I)/12.
TCLAD(I) = TCLAD(I)/12.
COBI6510
COBI6520
COBI6530
COBI6540
COBI6550
COBI6560
COBI6570
COBI6580
COBI6590
COBI6600
COBI6610
COBI6620
COBI6630
COBI6640
COBI6650
COBI6660
COBI6670
COBI6720
COBI6730
COBI6740
COBI6750
COBI6760
COBI6770
COBI6780
COBI6790
COBI6800
COBI6810
COBI6820
COBI6830
COBI6840
COBI6850
COBI6860
COBI6870
COBI6880
COBI6890
COBI6900
COBI6910
COBI6920
COBI6930
COBI6940
COBI6950
COBI6960
COBI6970
COBI6980
COBI6990
COBI7000
COBI7010
COBI7020
COBI7030
COBI7040
COBI7050

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HGAP(I) = HGAP(I)/3600.
302 CONTINUE
305 IF(.NOT.PRINT(9)) GO TO 310
WRITE(13,18) K1J,PTM,SL,ZZ,THE TA,NDX,DIX,NDT,TTIME,DT,NTRYIES,
1 FERROR
310 IF(IFRM.EQ.0.AND.IHTM.EQ.0) GO TO 307
WRITE(13,17) EPSF
307 IF(.NOT.PRINT(10))GO TO 315
WRITE(13,35)
IF(NSCBC.EQ.1) WRITE(13,32) ABETA
IF(NSCBC.EQ.2) WRITE(13,33) ABETA, BBETA
IF(NSCBC.EQ.3) WRITE(13,34) ABETA, BBETA
IF(NSCBC.EQ.4) WRITE(13,39) ABETA, BBETA
IF(NSCBC.EQ.4) WRITE(13,41)
IF(NDBC-1) 311,311,312
311 IF (NSCBC.NE.4) WRITE(13,36)
GO TO 314
312 WRITE (13,37) (XQUAL(I),BX(I),I=1,NBBC)
314 IF(J5.EQ.1) WRITE(13,65) GK
315 IF(.NOT.PRINT(11)) GO TO 318
WRITE(13,21) PEXIT,H'N,G'N,TIN,AFUX
IF(IN.EQ.0) WRITE(13,87)
IF(IN.EQ.1) WRITE(13,88)
IF(IN.EQ.2) WRITE(13,89)
IF(IN.EQ.3) WRITE(13,90)
IF(IG.EQ.0) WRITE(13,91)
IF(IG.EQ.1) WRITE(13,92)
IF(IG.EQ.2) WRITE(13,93)
IF(NP.GT.1) WRITE(13,83) (YP(I),FP(I),I=1,NP)
IF(NH.GT.1) WRITE(13,84) (YH(I),FH(I),I=1,NH)
IF(NG.GT.1) WRITE(13,85) (YG(I),FG(I),I=1,NG)
IF(NQ.GT.1) WRITE(13,86) (YQ(I),FQ(I),I=1,NQ)
318 IF(ADDEB) 400,400,319
319 WRITE(13,50) ((IDAT(SLC+I*MC*(L-1)),I=1,NCHANL),L=1,4)
WRITE(13,50) (IDAT($IK+K),IDAT($JK+K),K=1,NK)
WRITE(13,51) (DATA($FACTO+K),K=1,NK)
WRITE(13,50) ((IDAT($LR+NR*(L-1)),NR=1,NROD),L=1,6)
WRITE(13,51) ((DATA($PWF+I*MC*(NR-1)),NR=1,NROD),I=1,NCHANL)
WRITE(13,51) (DATA($D+NR)
1 (DATA($RDIA+NR)
400 CONTINUE
RETURN
C
6 FORMAT (23H)HEAT FLUX DISTRIBUTION /23H X/L RELATIVE FLUX /
1(F7.3,F12.3)
12 FORMAT(22H)SUBCHANNEL INPUT DATA /
1109H CHANNEL TYPE AREA WETTED HEATED HYDRAULIC (ADJ)COB17520
2ACENT CHANNEL NO., SPACING, CENTROID DISTANCE) /
3 55H NO. (SQ-IN) PERIM. PERIM. DIAMETER / COB17540
4 25X, 30H (IN) (IN) / COB17550
13 FORMAT(22H)FLUID PROPERTY TABLE ./,
1 60H P T VF VG HF HG
1 30H VISC. KF SIGMA ./,
1 (F8.1,F10.2,F8.5,F12.5,2F10.2,3F10.5)
15 FORMAT(15H)ROD INPUT DATA / 96H ROD TYPE DIA RADIAL POWER COB17600

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COB17060

COB17070

COB17080

COB17090

COB17100

COB17110

COB17120

COB17130

COB17140

COB17150

COB17160

COB17170

COB17180

COB17190

COB17200

COB17210

COB17220

COB17230

COB17240

COB17250

COB17260

COB17270

COB17280

COB17290

COB17300

COB17310

COB17320

COB17330

COB17340

COB17350

COB17360

COB17370

COB17380

COB17390

COB17400

COB17410

COB17420

COB17430

COB17440

COB17450

COB17460

COB17470

COB17480

COB17490

COB17500

COB17510

COB17520

COB17530

COB17540

COB17550

COB17560

COB17570

COB17580

COB17590

COB17600

1 FRACTION OF POWER TO ADJACENT CHANNELS (ADJ. CHANNEL NO.) / COB17610  
 2 30H NO. (IN) FACTOR / (215,F8.4,F9.4,F11.4,1H(12, COB17620  
 11H)F9.4,1H(12,1H)F9.4,1H(12,1H)F9.4,1H(12,1H)F9.4, COB17630  
 11H(12,1H)) COB17640  
 17 FORMAT (/,) FUEL ROD TEMP. CONVERGENCE CRITERIA = ',E10.5//) COB17650  
 18 FORMAT (23HOCALCULATION PARAMETERS / COB17660  
 2 28H CROSSFLOW RESISTANCE,KIJ F8.3/ COB17670  
 3 28H MOMENTUM TURBULENT FACTORF8.4 / COB17680  
 4 28H CHANNEL LENGTH F8.3/ COB17690  
 4 28H CHANNEL ORIENTATION F8.2,8H INCHES / COB17700  
 5 28H NUMBER OF AXIAL NODES F8.1,8H DEGREES/ COB17710  
 6 28H NODE LENGTH F8.3,7H INCHES / COB17720  
 7 28H NUMBER OF TIME STEPS 18/ COB17730  
 8 28H TOTAL TRANSIENT TIME F8.3,8H SECONDS/ COB17740  
 X 28H TIME STEP F8.4,8H SECONDS/ COB17750  
 1 28H ALLOWABLE ITERATIONS 18/ COB17760  
 2 28H FLOW CONVERGENCE FACTOR E10.5// COB17770  
 19 FORMAT (50H0 X/L AREA VARIATION FACTORS FOR SUBCHANNEL (I) / COB17780  
 1 7X,10(3X,A1,12,A1,1X)) COB17790  
 20 FORMAT (69H0 X/L GAP SPACING VARIATION FACTORS FOR ADJACENT SUB COB17800  
 1 CHANNELS (I,J) / 7X,10(1X,A1,12,A1,12,A1)) COB17810  
 21 FORMAT (22HOPERATING CONDITIONS / COB17820  
 1 25H SYSTEM PRESSURE = ',F8.1,5H PSIA / COB17830  
 2 25H INLET ENTHALPY = ',F8.1,7H BTU/LB / COB17840  
 3 25H AVG. MASS VELOCITY = ',F8.3,21H MILLION LB/(HR-SQFT) / COB17850  
 2 25H INLET TEMPERATURE = ',F8.1,10H DEGREES F / COB17860  
 4 25H AVG. HEAT FLUX = ',F8.6,22H MILLION BTU/(HR-SQFT) ) COB17870  
 28 FORMAT (/29H FRICTION FACTOR CORRELATION ) COB17880  
 29 FORMAT ( 16H CHANNEL TYPE 13,11H FRICT = F5.3,6H-RE\*\*(F5.3, COB17900  
 14H) + F6.4 ) COB17910  
 32 FORMAT (31H SUBCOOLED MIXING, BETA = F6.4) COB17920  
 33 FORMAT (31H SURCOOLED MIXING, BETA = F6.4,6H-RE\*\*(F6.4,1H)) COB17930  
 34 FORMAT (31H SUBCOOLED MIXING, BETA = F6.4,12H\*(D/S)\*RE\*\*(F6.4, COB17940  
 1 1H)) COB17950  
 35 FORMAT (20H MIXING CORRELATIONS ) COB17960  
 36 FORMAT (54H BOILING MIXING, BETA IS ASSUMED SAME AS SUBCOOLED) COB17970  
 37 FORMAT (55H BOILING MIXING, BETA IS A FUNCTION OF STEAM QUALITY/ COB17980  
 1 25H BETA(X) / (F12.3,F13.6)) COB17990  
 38 FORMAT (F6.3,10F8.3) COB18000  
 39 FORMAT (31H SUBCOOLED MIXING, BETA = F6.4,12H\*(D/L)\*RE\*\*(F6.4, COB18010  
 1 1H)) COB18020  
 41 FORMAT (1M ' NEW(BEUS) MIXING MODEL USED') COB18030  
 44 FORMAT ( / 20H TWO-PHASE FLOW CORRELATIONS ) COB18040  
 45 FORMAT ( 33H NO SUBCOOLED VOID CORRELATION ) COB18050  
 46 FORMAT ( 35H LEVY SUBCOOLED VOID CORRELATION) COB18060  
 47 FORMAT ( 31H HOMOGENEOUS BULK VOID MODEL) COB18070  
 48 FORMAT ( 41H MODIFIED ARMAND BULK VOID CORRELATION ) COB18080  
 49 FORMAT ( 50H HOMOGENEOUS BULK VOID MODEL WITH SLIP RATIO OF. COB18090  
 1 F6.2 ) COB18100  
 50 FORMAT (2015) COB18110  
 51 FORMAT (8E12.3) COB18120  
 57 FORMAT ( 33H BULK VOID FRACTION GIVEN AS A 12,56H TERM POLYNOCOB18130  
 1 TIAL FUNCTION OF QUALITY WITH COEFFICIENTS OF/ 10X,7E10.4) COB18140  
 58 FORMAT ( 41H HOMOGENEOUS MODEL FRICTION MULTIPLIER ) COB18150

59 FORMAT( 30H ARMAND FRICTION MULTIPLIER) COB18160  
 60 FORMAT( 34H FRICTION MULTIPLIER GIVEN AS A 12,57H TERM POLYNOCOB18170  
 10RIAL FUNCTION OF QUALITY WITH COEFFICIENTS OF/ 10X,7E10.4) COB18180  
 61 FORMAT(65H WALL VISCOSITY CORRECTION TO FRICTION FACTOR IS NOT COB18190  
 1)INCLUDED ) COB18200  
 62 FORMAT(65H WALL VISCOSITY CORRECTION TO FRICTION FACTOR IS INCLCOB18210  
 1)UDED ) COB18220  
 65 FORMAT(42H CONDUCTION MIXING, GEOMETRY FACTOR = F6.4) COB18230  
 69 FORMAT ( /62H WIRE WRAP SPACER DATA FOR FORCED DIVERSION CROSSFLOWCOB18240  
 1 MIXING //20H WRAP PITCH = F6.1,7H INCHES / COB18250  
 2 WRAP THICKNESS = F6.4,7H INCHES / COB18260  
 3 PIN DIAMETER = F6.4,7H INCHES // COB18270  
 70 FORMAT (23H WRAP CROSSING DATA / COB18280  
 1 GAP SUBCHANNEL MIXING RELATIVE LOCATION COB18290  
 2 / 60H NO. PAIR NO. PARAMETER OF WRAP CROSSINGS COB18300  
 3 / (110,4X,A1,12,A1,12,A1,F11.4,6F10.4)) COB18310  
 71 FORMAT ( /12H SPACER DATA / 20H SPACER TYPE NO. ,10I6 ) COB18320  
 72 FORMAT ( 21H LOCATION (X/L) ,10F6.3 ) COB18330  
 73 FORMAT (15HO SPACER TYPE 12 / COB18340  
 1 62H CHANNEL DRAG CHANNEL DRAG CHANNEL DRAG CHANNEL DRAGCOB18350  
 2/6:4H NO. COEFF. NO. COEFF. NO. COEFF. NO. COEFFCOB18350  
 3F. / (3X.4 (16,F9.3))) COB18370  
 74 FORMAT (46H INITIAL WRAP INVENTORY FOR EACH SUBCHANNEL / (10I5))COB18380  
 76 FORMAT (43HO FLOW DIVERSION FACTORS FOR SPACER TYPE ,12,/. COB18390  
 1 5X,46HGAP CHANNEL FRACTION GAP CHANNEL FRACTION // COB18400  
 2 5X,46HNO. PAIR DIVERTED NO. PAIR DIVERTED // COB18410  
 3 (2(5X,13,1X,A1,12,A1,12,A1,F9.4))) COB18420  
 77 FORMAT(39H THERMAL PROPERTIES FOR FUEL MATERIAL COB18430  
 1 18,18H RADIAL FUEL NODES / FUEL PROPERTIES ,25X,15HCLAD PROPERTIES, COB18440  
 1 37H TYPE COND. SP. HEAT DENSITY DIA. COB18450  
 2 / ,50H 50H COND. SP. HEAT DENSITY THICK. GAP COND. / COB18470  
 3 4 40H NO. (B/HR-FT-F) (B/LB-F) (LB/FT3) (IN.) COB18480  
 5 52H(B/HR-FT-F) (B/LB-F) (LB/FT3) (IN.) (B/HR-FT2-F))COB18490  
 78 FORMAT(17,2X,F7.2,F11.4,F11.1,F9.4,2X,F7.2,F11.4,F11.1,F9.4,2X, COB18500  
 1 F9.2) COB18510  
 83 FORMAT (33H FORCING FUNCTION FOR PRESSURE / COB18520  
 1 TIME PRESSURE / COB18530  
 2 (SEC) FACTOR / (F10.4,F13.4)) COB18540  
 84 FORMAT (30H FORCING FUNCTION FOR INLET ENTHALPY// COB18550  
 1 TIME INLET ENTHALPY / COB18560  
 2 (SEC) FACTOR / (F10.4,F13.4)) COB18570  
 85 FORMAT (38H FORCING FUNCTION FOR INLET FLOW / COB18580  
 1 TIME INLET FLOW / COB18590  
 2 (SEC) FACTOR / (F10.4,F13.4)) COB18600  
 86 FORMAT (30H FORCING FUNCTION FOR HEAT FLUX / COB18610  
 1 TIME HEAT FLUX / COB18620  
 2 (SEC) FACTOR / (F10.4,F13.4)) COB18630  
 87 FORMAT(30H UNIFORM INLET ENTHALPY ) COB18640  
 88 FORMAT(35H UNIFORM INLET TEMPERATURE ) COB18650  
 89 FORMAT(45H INDIVIDUAL SUBCHANNEL ENTHALPY SPECIFIED ) COB18660  
 90 FORMAT(50H INDIVIDUAL SUBCHANNEL TEMPERATURE SPECIFIED ) COB18670  
 91 FORMAT(35H UNIFORM INLET MASS VELOCITY ) COB18680  
 92 FORMAT(50H FLOWS SPLIT TO GIVE EQUAL PRESSURE GRADIENT ) COB18690  
 93 FORMAT(45H INDIVIDUAL SUBCHANNEL FLOWS SPECIFIED ) COB18700

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1003 FORMAT(15,17,4F10.4,4X,4(1H(13,1H,FS.3,1H,FS.3,1H)))
1004 FORMAT(15,17,4F10.6,4X,4(1H(13,1H,FS.3,1H,FS.3,1H)))
END
SUBROUTINE ITHO(NTHBOX,NTHBXX,ND1X,ND2X)
C
C
C      IMPLICIT INTEGER (5)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL,BBETA ,DIA ,DT ,DX
1  ELEV ,FERROR,FLO ,FTM ,GC ,GK ,GRID ,HSURF ,HF
2  HFG ,HG ,I2 ,I3 ,IERROR,IOP3 ,ITERAT,J1 ,J2
3  J3 ,J4 ,J5 ,J6 ,J7 ,KOEBUG,KF ,KIJ ,KF
4  NAFAC,NARAMP,NAX ,NAXL ,NBBC ,NCHANL,NCHF ,NDX ,NF
5  NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGKL ,NK ,NODES ,NODESF ,NPROP
6  NRAMP ,NROD ,NSCBC ,NV ,NVISCH,PI ,PITCH ,POWER ,PREF
7  QAX ,RHDF ,RHOG ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK
8  UF ,VF ,VFG ,VG ,Z
C
C      LOGICAL GRID
REAL KIJ , MF , MKF , KCLAD , KFUEL
C
C      DIMENSION NTHBOX(25,25)
DIMENSION CARD(20)
C
C      CONTROL FOR THERMAL-HYDRAULIC INPUT DATA
C
C      WRITE (13,1001)
WRITE (13,1002)
CALL CHAN(1,NTHBOX,NTHBXX,ND1X,ND2X,CARD)
CALL MODEL(1,CARD,IPILE)
CALL OPERA(1,CARD)
CALL FIZPRP(1,NPROP)
CALL TABLES(CARD)
WRITE (13,1002)
CALL CORE3
IF (IERROR.EQ.0) GO TO 2
WRITE (13,1004)
RETURN
C
C      CONTINUE
WRITE (13,1003)
CALL OPERA(2,CARD)
CALL CHAN(2,NTHBOX,NTHBXX,ND1X,ND2X,CARD)
CALL MODEL(2,CARD,IPILE)
CALL FIZPRP(2,NPROP)
RETURN
C
1001 FORMAT(1H1, 42X, ' THERMAL - HYDRAULIC INPUT DATA', /, 43X,
1  '-----', /, /, /, /, ' CARD IMAGES', /, 2X,
2  '-----')
1002 FORMAT(32X, '0', .....1, .....2, .....3, .....4, .....5, .....6, .....7, .....8)
1003 FORMAT(1H1, 42X, ' PROCESSED INPUT DATA', /, 43X,
1  '-----', /, ' * = SET IN NEUTRONICS (CARD20)',
2  '////')
1004 FORMAT(' ERROR SIGNAL IN ITHO')

```

COB18710  
COB18720  
COB18730  
COB18740  
COB18750  
COB18760  
COB18770  
COB18780  
COB18790  
COB18800  
COB18810  
COB18820  
COB18830  
COB18840  
COB18850  
COB18860  
COB18870  
COB18880  
COB18890  
COB18900  
COB18910  
COB18920  
COB18930  
COB18940  
COB18950  
COB18960  
COB18970  
COB18980  
COB18990  
COB19000  
COB19010  
COB19020  
COB19030  
COB19040  
COB19050  
COB19060  
COB19070  
COB19080  
COB19090  
COB19100  
COB19110  
COB19120  
COB19130  
COB19140  
COB19150  
COB19160  
COB19170  
COB19180  
COB19190  
COB19200  
COB19210  
COB19220  
COB19230  
COB19240  
COB19250



```

VBAR=DATA(SVISC+II)+DATA(SVISC+JJ)
DAVG=4.*ABAR/PBAR
GAVG=FBAR/ABAR
XAVG=0.
IF(AMAX1(DATA(SQUAL+II),DATA(SQUAL+JJ)).GT.0.) XAVG=0.5*QBAR
IF(XAVG.GT.0.)AND.(NBBC.GE.2) GO TO 80
UAVG=0.5*VBAR
IF(NSCBC.EQ.1) RE = GAVG*DAVG/UAVG
IF(NSCBC.EQ.0) DATA(SWP+K)=DATA(SGAP+K)*GAVG*ABETA
IF(NSCBC.EQ.1) DATA(SWP+K)=DATA(SGAP+K)*GAVG*ABETA*RE*BBETA
IF(NSCBC.EQ.2) DATA(SWP+K)=DAVG
IF(NSCBC.EQ.3) AND.(DATA(SLENGT+K).LE.0.) GO TO 1000
IF(NSCBC.EQ.3) AND.(DATA(SWP+K)=DATA(SGAP+K)/DATA(SLENGT+K))*DAVG*GAVG
1 *ABETA*RE*BBETA
IF(NSCBC.EQ.4) GO TO 50
DATA(SWP+K)=DATA(SWP+K)*DATA(SFACTO+K)
GO TO 100
CC BEUS MIXING MODEL USED WHEN NSCBC=4
50 WL=0.0035*UAVG*RE*0.9
ARBAR=ABAR*0.5
BI=0.01*(DATA(SGAP+K)/DAVG)**1.5
XC=(0.4/GAVG*SQRT(32.2*RHOF*DAVG*(RHOF-RHOG))+0.6)/(SQRT(RHOF
1 /RHOG)+0.6)
2 -1.))
GAM=XAVG/(1.-XAVG)*(1.-ALP)/ALP*RHOF/RHOG
IF(XAVG.GT.XC) GO TO 55
52 DATA(SWP+K)=WL+BI*ARBAR*GAVG/DAVG*RHOF/RHOG*(GAM-1.)/GAM*XAVG
GO TO 100
55 XDXC=0.57*RE**0.0417
TK=0.5556*(TF+459.67)
VISCG=0.672*VISUP(TK)
WG=0.0035*VI SCG*(GAVG*GAM*DAVG/VISCG)**0.9
WC=WL+BI*ARBAR*GAVG/DAVG*RHOF/RHOG*(GAM-1.)/GAM*XC
DATA(SWP+K)=(WG+(WC-WG)*(1.-XDXC)/(XAVG/XC-XDXC))*DATA(SFACTO+K)
GO TO 100
80 CALL CURVE (XBETA,XAVG,BK,QUAL,HDBC,ERROR,1)
IF(ERROR.GT.1) GO TO 1000
100 DATA(SWP+K)=
GAVG*DAVG*XBETA *DATA(SFACTO+K)
DAVG*0.5*(DATA(SCON+II)+DATA(SCON+JJ))
IF(DATA(SLENGT+K).LE.0) GO TO 1000
DATA(SCONB+K)=CAVG*DATA(SGAP+K)/DATA(SLENGT+K)*GK*DATA(SFACTO+K)
240 CONTINUE
RETURN
1000 ERROR = 4
RETURN
END
SUBROUTINE OPERA(IPART,CARD)
C
C

```

COB19810  
COB19820  
COB19830  
COB19840  
COB19850  
COB19860  
COB19870  
COB19880  
COB19890  
COB19900  
COB19910  
COB19920  
COB19930  
COB19940  
COB19950  
COB19960  
COB19970  
COB19980  
COB19990  
COB20000  
COB20010  
COB20020  
COB20030  
COB20040  
COB20050  
COB20060  
COB20070  
COB20080  
COB20090  
COB20100  
COB20110  
COB20120  
COB20130  
COB20140  
COB20150  
COB20160  
COB20170  
COB20180  
COB20190  
COB20200  
COB20210  
COB20220  
COB20230  
COB20240  
COB20250  
COB20260  
COB20270  
COB20280  
COB20290  
COB20300  
COB20310  
COB20320  
COB20330  
COB20340  
COB20350

CONVERSATIONAL MONITOR SYSTEM

FILE: COBR3C FORTRAN A

```

IMPLICIT INTEGER ($)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX ,
1 ELEV , FERROR , FLD , FTM , GC , GK , GRID , HSURF , MF ,
2 HFG , HG , I2 , I3 , IERROR , IQP3 , ITERAT , J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ ,
4 NAFAC , NARAMP , MAX , NAXL , NBBC , NCHANL , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRDIT , NGTYPE , NGXL , NK , NNODES , NODESF , NPROP ,
6 NRAMP , NRDD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VF , VFG , VG , Z ,
COB20360
COB20370
COB20380
COB20390
COB20400
COB20410
COB20420
COB20430
COB20440
COB20450
COB20460
COB20470
COB20480
COB20490
COB20500
COB20510
COB20520
COB20530
COB20540
COB20550
COB20560
COB20570
COB20580
COB20590
COB20600
COB20610
COB20620
COB20630
COB20640
COB20650
COB20660
COB20670
COB20680
COB20690
COB20700
COB20710
COB20720
COB20730
COB20740
COB20750
COB20760
COB20770
COB20780
COB20790
COB20800
COB20810
COB20820
COB20830
COB20840
COB20850
COB20860
COB20870
COB20880
COB20890
COB20900

COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KNF(30) , NCH(10) , NGAP(9) ,
4 PP(30) , RCLAD(2) , RFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)

LOGICAL GRID
REAL KIJ , KF , MKF , KCLAD , KFUEL

COMMON /COBRA3/ MA , MC , MG , MH , MI , MJ , MK , ML , MM , MN , MO ,
1 MS , MP , MQ , MR , MS , MT , MU , MV , MW , MX ,
2 MY , MZ , NA , NB , NC , ND , NE , NF , NG , NH , NI , NJ , NK ,
3 NL , NM , NN , NO , NP , NQ , NR , NS , NT , NU , NV , NW , NX ,
4 NY , NZ , OA , OB , OC , OD , OE , OF , OG , OH , OI , OJ , OK ,
5 OL , OM , ON , OO , OP , OQ , OR , OS , OT , OU , OV , OW , OX ,
6 OY , OZ , PA , PB , PC , PD , PE , PF , PG , PH , PI , PJ , PK ,
7 PL , PM , PN , PO , PP , PQ , PR , PS , PT , PU , PV , PW , PX ,
8 PY , PZ , QA , QB , QC , QD , QE , QF , QG , QH , QI , QJ , QK ,
9 QL , QM , QN , QO , QP , QQ , QR , QS , QT , QU , QV , QW , QX ,
A QY , QZ , RA , RB , RC , RD , RE , RF , RG , RH , RI , RJ , RK ,
1 RL , RM , RN , RO , RP , RQ , RR , RS , RT , RU , RV , RW , RX ,
2 RY , RZ , SA , SB , SC , SD , SE , SF , SG , SH , SI , SJ , SK ,
3 SL , SM , SN , SO , SP , SQ , SR , SS , ST , SU , SV , SW , SX ,
4 SY , SZ , TA , TB , TC , TD , TE , TF , TG , TH , TI , TJ , TK ,
5 TL , TM , TN , TO , TP , TQ , TR , TS , TT , TU , TV , TW , TX ,
6 TY , TZ , UA , UB , UC , UD , UE , UF , UG , UH , UI , UJ , UK ,
7 UL , UM , UN , UO , UP , UQ , UR , US , UT , UU , UV , UW , UX ,
8 UY , UZ , VA , VB , VC , VD , VE , VF , VG , VH , VI , VJ , VK ,
9 VL , VM , VN , VO , VP , VQ , VR , VS , VT , VU , VV , VW , VX ,
A VY , VZ , WA , WB , WC , WD , WE , WF , WG , WH , WI , WJ , WK ,
1 WL , WM , WN , WO , WP , WQ , WR , WS , WT , WU , WV , WW , WX ,
2 WY , WZ , XA , XB , XC , XD , XE , XF , XG , XH , XI , XJ , XK ,
3 XL , XM , XN , XO , XP , XQ , XR , XS , XT , XU , XV , XW ,
4 XX , XY , XZ , YA , YB , YC , YD , YE , YF , YG , YH , YI , YJ ,
5 YK , YL , YM , YN , YO , YP , YQ , YR , YS , YT , YU , YV ,
6 YW , YX , YY , YZ , ZA , ZB , ZC , ZD , ZE , ZF , ZG , ZH ,
7 ZI , ZJ , ZK , ZL , ZM , ZN , ZO , ZP , ZQ , ZR , ZS , ZT ,
8 ZU , ZV , ZW , ZX , ZY , ZZ
COB20730
COB20740
COB20750
COB20760
COB20770
COB20780
COB20790
COB20800
COB20810
COB20820
COB20830
COB20840
COB20850
COB20860
COB20870
COB20880
COB20890
COB20900

COMMON /LINK2/ CROSS(6) , DATE(2) , FG(30) , FH(30) , FI(30) , FJ(30) , FK(30) , FL(30) , FM(30) , FN(30) , FO(30) , FP(30) , FQ(30) , FR(30) , FS(30) , FT(30) , FU(30) , FV(30) , FW(30) , FX(30) , FY(30) , FZ(30) ,
1 GM(9) , OUTPUT(10) , PRINT(12) , TEXT(17) , TIME(3) , YG(30) , YH(30) , YI(30) , YJ(30) , YK(30) , YL(30) , YM(30) , YN(30) , YO(30) , YP(30) , YQ(30) , YR(30) , YS(30) , YT(30) , YU(30) , YV(30) , YW(30) , YX(30) , YZ(30) ,
2 ZQ(30)

COMMON /LINK3/ DX , ETIME , GIN , HIN , IB , IG , IN , ISAVE , JUMP , KASE , KT , MAXT ,
1 NDT , NDXPT , NFUEL , NG , NH , NJUMP , NOUT , NP , NPCHAN , NPNODE , NPROD , NQ , NR ,
2 NSKIPT , NSKIPX , NTRIES , PEXIT , PHOTOT , SAVEDT , TIN , TTIME , ZZ
DIMENSION CARD(20)

IPART=1 READ OPERATING CONDITIONS
IPART=2 PRINT OPERATING CONDITIONS
COBRA AND MEKIN SAME CODING EXCEPT IMEXIN = 0

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```

IMEKIN = 0
IF (IPART.EQ.2) GO TO 10
READ (12,1001) CARD, IN, HIN, GIN, PEXIT
WRITE (13,1011) CARD
IF ( (NDX.LE.0) .OR. (Z.LE.0.0) ) GO TO 30
PREF = PEXIT
IF (IN.LT.2) GO TO 2
IF (IN.EQ.2) CALL READIN(2,NCHANL,DATA($HINLE+1),CARD,CARD,1)
IF (IN.EQ.3) CALL READIN(3,NCHANL,DATA($HINLE+1),CARD,CARD,1)
2 READ(12,1002) CARD,NP,NH,NG,NQ
WRITE (13,1012) CARD
IF (NP.GT.1) CALL READIN(4,NP,YP,FP,CARD,2)
IF (NH.GT.1) CALL READIN(5,NH,YH,FH,CARD,2)
IF (NG.GT.1) CALL READIN(6,NG,YG,FG,CARD,2)
IF (NQ.GT.1) CALL READIN(7,NQ,YQ,FQ,CARD,2)
IF (NPROP.GT.0) GO TO 9
C SET MAX AND MIN PRESSURES FOR PHYSICAL PROPERTIES IN FIZPRP.
ZMIN = 1.0
IF (NH.LE.1) GO TO 4
DO 3 I=1,NH
IF (FH(I).LT.ZMIN) ZMIN = FH(I)
3 CONTINUE
4 WV = HIN
IF (IN.LT.2) GO TO 6
WV = 1000.0
DO 5 I=1,NCHANL
IF (DATA($HINLE+1).LT.WV) WV=DATA($HINLE+1)
5 CONTINUE
WV CORRESPONDS TO MIN HIN OR TIN AT STEADY STATE
6 R = 0.01*WV*ZMIN
IF (R.LT.4.5) R = R*(1.0-0.1*(4.5-R))
C SET PP(1) TO PRESSURE LOWER THAN MIN IN PROBLEM FOR FIZPRP
PP(1) = 10.0
IF (R.GT.2.0) PP(1) = 6.0*R*R*(R-1.35)/(R-0.35)
ZMAX = 1.0
IF (NP.LE.1) GO TO 8
ZMIN = 1.0E06
DO 7 I=1,NP
IF (FP(I).GT.ZMAX) ZMAX = FP(I)
IF (FP(I).LT.ZMIN) ZMIN = FP(I)
7 CONTINUE
IF (ZMIN*PREF.LT.PP(1)) PP(1) = ZMIN*PREF
C PP(2) = ZMAX*PREF + 0.01
NPROP = 30
9 CONTINUE
C
C SET TIME AND NDT FOR MEKIN ONLY
IF (IMEKIN.EQ.0) RETURN
TIME = 1.0
NDT = 1
IF ((NP+NH+NG+NQ).LE.0) NDT=0
RETURN
C 10 WRITE (13,1020) PEXIT,GIN

```

COB20910  
COB20920  
COB20930  
COB20940  
COB20950  
COB20960  
COB20970  
COB20980  
COB20990  
COB21000  
COB21010  
COB21020  
COB21030  
COB21040  
COB21050  
COB21060  
COB21070  
COB21080  
COB21090  
COB21100  
COB21110  
COB21120  
COB21130  
COB21140  
COB21150  
COB21160  
COB21170  
COB21180  
COB21190  
COB21200  
COB21210  
COB21220  
COB21230  
COB21240  
COB21250  
COB21260  
COB21270  
COB21280  
COB21290  
COB21300  
COB21310  
COB21320  
COB21330  
COB21340  
COB21350  
COB21360  
COB21370  
COB21380  
COB21390  
COB21400  
COB21410  
COB21420  
COB21430  
COB21440  
COB21450



```

C      SET HINLET = H OR T ACCORDING TO IN
      IF (IN-1) 12,14,20
12     WRITE (13,1021) IN,MIN
      GO TO 16
14     WRITE (13,1022) IN,MIN
16     DO 18 I=1,NCHANL
18     DATA($HINLE*I) = MIN
      GO TO 22
20     IF (IN.EQ.2) WRITE (13,1023) IN,(I,DATA($HINLE*I),I=1,NCHANL)
      IF (IN.EQ.3) WRITE (13,1024) IN,(I,DATA($HINLE*I),I=1,NCHANL)
22     WRITE (13,1025) Z,NDX
      Z = Z/12.0
      IF (NDT.GT.0) GO TO 24
      WRITE (13,1026)
      GO TO 26
24     IF (IMCKIN.EQ.0) WRITE (13,1027) NDT,TTIME
      IF (NP.GT.1) WRITE(13,1028) (YP(I),FP(I),I=1,NP)
      IF (NH.GT.1) WRITE(13,1029) (YH(I),FH(I),I=1,NH)
      IF (NG.GT.1) WRITE(13,1030) (YG(I),FG(I),I=1,NG)
      IF (NQ.GT.1) WRITE(13,1031) (YQ(I),FQ(I),I=1,NQ)
26     RETURN
30     WRITE (13,1040)
      STOP
C
1001  FORMAT(20A4, T1, I5, 13E5.0)
1002  FORMAT(20A4, T1, 14I5)
C
1011  FORMAT(' IN (OR T) IN GIN PEXIT', 6X, '...', 20A4, '... OPERA')
1012  FORMAT(' TRANS INDIC FOR P H G Q', 5X, '...', 20A4, '... OPERA')
C
1020  FORMAT(43X, 'OPERATING CONDITIONS', /, 43X,
1 '-----', //, ' PRESSURE', 20X, '(PSIA)', 9X, '...',
2 F10.2, /, ' AV. INLET MASS VELOCITY', 5X, '(MLB/SQFT.HR)', 2X,
3 '...', F12.4)
1021  FORMAT(' IN=', 12, ' INLET ENTHALPY', 7X, '(BTU/LB)', 7X, '...',
1 F11.3)
1022  FORMAT(' IN=', 12, ' INLET TEMPERATURE', 4X, '(DEG F)', 8X, '...',
1 F11.3)
1023  FORMAT(' IN=', 12, ' INLET ENTHALPIES', 5X, '(BTU/LB)', 7X, '...',
1 /,(5X,6(15,5X,F10.3)/))
1024  FORMAT(' IN=', 12, ' INLET TEMPERATURES', 3X, '(DEG F)', 8X, '...',
1 /,(5X,6(15,5X,F10.3)/))
1025  FORMAT(' *CHANNEL LENGTH', 14X, '(IN)', 11X, '...', F10.2, /,
1 ' *NO. OF AXIAL INTERVALS', 21X, '...', 17)
1026  FORMAT(' NO TRANSIENT CALCULATION')
1027  FORMAT(' *NO. OF TIME STEPS', 26X, '...', 17, /,
1 ' *TOTAL TIME OF TRANSIENT', 5X, '(SEC)', 10X, '...', F10.2)
1028  FORMAT (/, 33H FORCING FUNCTION FOR PRESSURE /
1 23H PRESSURE /
2 23H (SEC) / (F10.4,F13.4))
1029  FORMAT (/, 38H FORCING FUNCTION FOR INLET ENTHALPY /
1 28H TIME INLET ENTHALPY /
2 23H (SEC) / (F10.4,F13.4))
1030  FORMAT (/, 38H FORCING FUNCTION FOR INLET FLOW /
1 28H TIME INLET FLOW /

```

COB21460  
COB21470  
COB21480  
COB21490  
COB21500  
COB21510  
COB21520  
COB21530  
COB21540  
COB21550  
COB21560  
COB21570  
COB21580  
COB21590  
COB21600  
COB21610  
COB21620  
COB21630  
COB21640  
COB21650  
COB21660  
COB21670  
COB21680  
COB21690  
COB21700  
COB21710  
COB21720  
COB21730  
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COB21860  
COB21870  
COB21880  
COB21890  
COB21900  
COB21910  
COB21920  
COB21930  
COB21940  
COB21950  
COB21960  
COB21970  
COB21980  
COB21990  
COB22000

```

2      23H      (SEC)      FACTOR / (F10.4,F13.4)
1031 FORMAT (/, 38H FORCING FUNCTION FOR HEAT FLUX /,
1      38H      TIME      HEAT FLUX /,
2      23H      (SEC)      FACTOR / (F10.4,F13.4)
1040 FORMAT (' INPUT DATA ERROR. NDX OR Z .LT.O. STOP (OPERA)')
END
SUBROUTINE PRECAL

```

```

C
C
C      IMPLICIT INTEGER ($ )
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1      ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , HF
2      HFG , HG , I2 , I3 , IERROR , IQP3 , ITERAT , J1 , J2
3      J3 , J4 , J5 , J6 , J7 , KDEBUG , MF , MIJ ,
4      NAFAC , NARAMP , MAX , MAXL , NBBC , NCHANL , NCHF , NDX , NF
5      NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NNODES , NODESF , NPROP
6      NRAMP , NROD , NSCBC , NV , NVISWC , PI , PITCH , POWER , PREF
7      OAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
8      UF , VFG , YG , Z

```

```

C
C      COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1      AXL(10) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2      GAPXL(10) , GFAC(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3      IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4      PP(30) , RCLAD(2) , RFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
5      VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)

```

```

C
C      LOGICAL GRID
REAL      KIJ , MF , MKF , KCLAD , KFUEL

```

```

C
C      COMMON /COBRA3/ MA , MC , MIG , MIN , MR , MS , MX
1      $$$ , SA , SAAA , SAC , SALPHA , SAN , SANSME , SD
1      SCCHAN , SCD , SCHFR , SCON , SCOND , SCP , SD , SDC , SDFDX
2      SDHX , SDHYD , SDHYDN , SDIST , SDPDX , SDPK , SDUR , SDR , SF
3      SFACX , SFDIV , SFINLE , SFELUX , SFMULT , SFOLD , SFSP , SFSPLI , SFXFLO , COB22370
4      SCAP , SCAPN , SGAPS , SH , SHFILM , SHINLE , SHOLD , SHPERI , $IDARE , COB22370
5      SIDFUE , SIDGAP , SIK , $JBOIL , $JK , $LC , $LENGT , $LOCA , $LR , COB22390
6      $MCFR , $ACFRC , $ACFRR , $NTYPE , $NWRAP , $NWRPS , $P , $PERIM , $PH
7      $PHI , $PRINC , $PRINT , $PRINT , $PW , $PWR , $QC , $OF , $OPRIM , COB22410
8      $QUAL , $RADIA , $RHD , $RHOOL , $SP , $ST , $STOUNY , $TINLE , $TROD , COB22420
9      $U , $UH , $USAVE , $USTAR , $V , $VISC , $VISCH , $VVP , $VPA , COB22440
A      $W , $WOLD , $WP , $WSAVE , $X , $XACROS , $XA , $XB , $XPOLD , COB22450

```

```

C
C      COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))

```

```

C
C      COMMON /LINK3/ DXA , ETIME , GIN , HIN , I8 , IG , IN , ISAVE , JUMP , KASE , KY , MAKT ,
1      NDT , NDAPI , NFUEL , NG , NH , NJUMP , NOUT , NP , NPCHAN , NPNODE , NP ROD , NO , NR ,
2      NSKIPT , NSKIPIX , NTRIES , PEXIT , PHOTOT , SAVEDT , TIN , TTIME , ZZ

```

```

C

```

```

C   PREPARE TO START CALCULATION ( IN CALC)
CALL PROP(1,1)
IF (IERROR.GT.0) GO TO 20
NDXPI = NDX + 1
DX = Z/LOAT(NDX)
DXK = DX*12.0
DT = 0.0
IF ( (NOT.GT.0) .AND. (TTIME.LE.0.0) ) NDT=0
IF (NDT.GT.0) DT = TTIME/LOAT(NDT)
SAVEI = DT
SET HINLET
HIN = DATA($HINLE+1)
IF ( (IN.EQ.0) .OR. (IN.EQ.2) ) GO TO 10
IF (IN.GE.3) GO TO 6
:TIN = HIN
CALL CURVE(HIN,TIN,MHF,TT,NPROP,IERROR,1)
IF (IERROR.GT.0) GO TO 20
DO 4 I=1,NCHANL
DATA($HINLE+I) = TIN
4 DATA($HINLE+I) = HIN
GO TO 10
6 DO 8 I=1,NCHANL
DATA($HINLE+I) = DATA($HINLE+I)
CALL CURVE(DATA($HINLE+I),DATA($HINLE+I),MHF,TT,NPROP,IERROR,1)
IF (IERROR.GT.0) GO TO 20
8 CONTINUE
C   SET FINLET
10 WV = GIN/0.0036
FLO = WV*ATOTAL
WV1 = 1.0
DO 12 I=1,NCHANL
IF (IG.EQ.2) WV1 = DATA($FINLE+I)
12 DATA($FINLE+I) = WV*WV1+DATA($AN+I)
IF (IG.EQ.1) CALL SPLIT
RETURN
20 WRITE (13,1001)
RETURN
C
1001 FORMAT(' PRECAL ERROR SIGNAL AFTER CALLING CURVE OR PROP')
END
SUBROUTINE PROP(IPART,J)
C THIS PROCEDURE CONTAINS THE COMMON AND TYPE STATEMENTS SHARED BY THE COB22970
C MAJOR SUBROUTINES OF COBRA-111C.
C
C
C
C   IMPLICIT INTEGER (9)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL,BBETA , DIA , DT , DX
1 ELEV , FERROR,FLO , FTM , GC , GK , GRID , HSURF , HF
2 HFG , HG , I2 , I3 , IERROR,IQ3 , ITERAT,J1 , J2
3 J3 , J4 , J5 , J6 , J7 , KDEBUG,KF , KIJ ,
4 NAFACT,NARAMP,NAK , NAXL , NBBC , NCHAN , NCHF , NDX , NF
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF,NPROP , COB23070
6 NRAMP , NRDD , NSCBC , NV , NVISCH,PI , PITCH , POWER , PREF , COB23080
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID,THETA , THICK , COB23090
8 UF , VF , VFG , VG , VZ

```

```

C
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
1 AXL(10), BX(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2 GAPAL(10), GFACT(9,10), GRIDXL(10), HGAP(2), HHF(30), HHG(30),
3 IGRID(10), KCLAD(2), KFUEL(2), KNF(30), NCH(10), NGAP(9),
4 PP(30), RFLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), Y(30), Y(30)
COB23110
COB23120
COB23130
COB23140
COB23150
COB23160
COB23170
COB23180
COB23190
COB23200
COB23210
COB23220
COB23230
COB23240
COB23250
COB23260
COB23270
COB23280
COB23290
COB23300
COB23310
COB23320
COB23330
COB23340
COB23350
COB23360
COB23370
COB23380
COB23390
COB23400
COB23410
COB23420
COB23430
COB23440
COB23450
COB23460
COB23470
COB23480
COB23490
COB23500
COB23510
COB23520
COB23530
COB23540
COB23550
COB23560
COB23570
COB23580
COB23590
COB23600
COB23610
COB23620
COB23630
COB23640
COB23650
C
LOGICAL GRID
REAL KI,J, KF, MKF, KCLAD, KFUEL
C
COMMON /COBRA3/ MA ,MC ,MG ,MH ,MI ,MJ ,MK ,ML ,MM ,MN ,MO ,MP ,MQ ,MR ,MS ,MT ,MU ,MV ,MW ,MX
1 $$ ,SA ,SAA ,SAC ,SALPHA ,SAN ,SANSWE ,SD ,SDC ,SDFDX ,SDFDI ,SDFDZ ,SDFDY ,SDFDX ,SDFDI ,SDFDZ
2 SCCHAN ,SCD ,SCHFR ,SCON ,SCOND ,SCP ,SD ,SDR ,SDR ,SDFDX ,SDFDI ,SDFDZ
3 $DHDX ,SDHYD ,SDHYD ,SDIST ,SDPDX ,SDPK ,SDUR ,SDR ,SDFDX ,SDFDI ,SDFDZ
4 $FACTO ,SFDIV ,SFILE ,SFLUX ,SFMULT ,SFOLD ,SFSP ,SFSPLI ,SFMFLO ,SFBPERI ,SDARE ,SFBPERI ,SDARE ,SFBPERI ,SDARE
5 $DIFUE ,SDIGAP ,SJK ,SUDOIL ,SJM ,SLC ,SLENGT ,SLOCA ,SLR ,SLENGT ,SLOCA ,SLR
6 $MCHFR ,SMCFRC ,SMCFRR ,SMCTYP ,SMWRAP ,SMWRPS ,SP ,SPERIM ,SPH ,SPERIM ,SPH
7 $PHI ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC ,SPRNTC
8 $QUAL ,SRADIA ,SRHO ,SRHOOL ,SSP ,ST ,STOUMY ,STINLE ,STROD ,STOUMY ,STINLE ,STROD
9 $U ,SUH ,SUSAVE ,SUSTAR ,SV ,SVISC ,SVISCH ,SVP ,SVPA ,SVISC ,SVISCH ,SVP ,SVPA
A $W ,SWOLD ,SWP ,SWSAVE ,SX ,SXACROS ,SXA ,SXB ,SXPOLD ,SXACROS ,SXA ,SXB ,SXPOLD
C
COMMON/LINK4/IFRM, IRTM, IPROP, NCC, NCF, NDM1, NDS, NGP
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
C
EQUIVALENCE (NCHAN, NCHANL)
C
1 FORMAT(' PROP. REYNOLDS NO. IN CHAN ', I3, ' J = ', I3,
1 ' IS TOO LOW. RE = ', IPE10.3, ' SX = ', F, ' VISC = ', 2E15.4)
5 FORMAT(60H FAILURE OF SUBROUTINE PROP, PRESSURE TOO LOW FOR TABLE
1P = E12.5 / (10E10.4))
6 FORMAT(61H FAILURE OF SUBROUTINE PROP, PRESSURE TOO HIGH FOR TABLE
1P = E12.5 / (10E10.4))
7 FORMAT(40H TABLE LOOKUP FAILED IN SUBROUTINE PROP )
NPROP = NPROP
IF(IPART.LT.1 .OR. IPART.GT.2) GO TO 1001
GO TO (9,100), IPART
C
PART 1, CALCULATION OF SATURATED PROPERTIES
9 DO 10 I=1,NPROP
IF(PREF.LT.-PP(I)) GO TO 20
10 CONTINUE
20 GO TO 200
IF(I.GT.1) GO TO 40
GO TO 210
40 VALUE = (PREF-PP(I-1))/(PP(I)-PP(I-1))
HF = HHF(I-1) + VALUE*( HHF(I) - HHF(I-1))

```

```

HG = HHG(I-1) + VALUE*( HHG(I)- HHG(I-1))
VF = VVF(I-1) + VALUE*( VVF(I)- VVF(I-1))
VG = VVG(I-1) + VALUE*( VVG(I)- VVG(I-1))
UF = UUF(I-1) + VALUE*( UUF(I)- UUF(I-1))
TF = TT(I-1) + VALUE*( TT(I)- TT(I-1))
KF = KHF(I-1) + VALUE*( KHF(I)- KHF(I-1))
SIGMA = SSIGMA(I-1) + VALUE*(SSIGMA(I)-SSIGMA(I-1))
HFG = HG-HF
VFG = VG-VF
RHOG = 1./VG
RHOF = 1./VF
RETURN

C PART 2. CALCULATE LIQUID PROPERTIES AND PARAMETERS
100 NCHAN = NCHANL
IF(J.GT.1) GO TO 102
DO 101 I=1,NCHAN
101 IDAT(SUBOIL+I)=0
102 DO 150 I=1,NCHAN
DATA($VISC+I)=UF
DATA($CON +I)=KF
DATA($S. +I)=VF
HH=DATA($H+I)+MC*(J-1)
IF(NH.GT.HF) GO TO 105
CALL CURVE(DATA($VISC+I),HH,UUF,MHF,NPROP, I,ERROR, 1)
IF(IEORR.GT.1) GO TO 1000
CALL CURVE(DATA($V +I),MH,VVF,MHF,NPROP, I,ERROR, 2)
CALL CURVE(DATA($T +I),MH,TT,MHF,NPROP, I,ERROR, 2)
CALL CURVE(DATA($CON +I),MH,KKF,MHF,NPROP, I,ERROR, 2)
TM=DATA($T +I)-1.
CALL CURVE (HM, TM, MHF, TT, NPROP, I, ERROR, 1)
IF(IEORR.GT.1) GO TO 1000
DATA($CP+I)=HM-HM
DATA($VISC +I)=DATA($VISC +I)/3600.
DATA($CON +I)=DATA($CON +I)/3600.
RE=DATA($F+I)+MC*(J-1)/DATA($A+I)+DATA($DHYD+I)/DATA($VISC+I)
IF(RE.LT.0.) WRITE(13,1) I,J,RE,DATA($F+I)+MC*(J-1),DATA($VISC+I)
IF(RE.LT.2000.) RE = 2000.
PR=DATA($CP+I)+DATA($VISC+I)/DATA($CON+I)
IF(DATA($H+I)+MC*(J-1)).GT.HF.AND.IDAT($SUBOIL+I).NE.0)
GO TO 120
1 IF(INITIA.NE.0.AND.J.NE.1) GO TO 108
DATA($FILM+I)=0.023*DATA($CON+I)/DATA($DHYD+I)+RE**.8*PR**.4
DATA($FILM+I) = HCOOL(-1,J)
DTWALL=DATA($OPRIM+I)/DATA($PERI+I)/DATA($FILM+I)
C DETERMINE THE START OF NUCLEATE BOILING
IF(IDAT($SUBOIL+I).GT.0) GO TO 106
IF(DATA($OPRIM+I).LT.0.0) GO TO 106
TLBOIL=TF-DTWALL+60.*EXP(-PREF/900.)*(DATA($OPRIM+I)/
TLBOIL=TF-DTWALL+ HCOOL(-2,I,J)
IF(DATA($T+I).GE.TLBOIL.AND.NCHF.NE.4) IDAT($SUBOIL+I)=J
IF(NCHF.EQ.4.AND.DATA($H+I)+MC*(J-1)).GE.HF) IDAT($SUBOIL+I)=J

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COB23660  
COB23670  
COB23680  
COB23690  
COB23700  
COB23710  
COB23720  
COB23730  
COB23740  
COB23750  
COB23760  
COB23770  
COB23780  
COB23790  
COB23800  
COB23810  
COB23820  
COB23830  
COB23840  
COB23850  
COB23860  
COB23870  
COB23880  
COB23890  
COB23900  
COB23910  
COB23920  
COB23930  
COB23940  
COB23950  
COB23960  
COB23970  
COB23980  
COB23990  
COB24000  
COB24010  
COB24020  
COB24030  
COB24040  
COB24050  
COB24060  
COB24070  
COB24080  
COB24090  
COB24100  
COB24110  
COB24120  
COB24130  
COB24140  
COB24150  
COB24160  
COB24170  
COB24180  
COB24190  
COB24200

```

106 TWALL=DATA(I+1)+DTWALL
GO TO 110
C
108 SAVE=0.
SUM=0.
DO 109 NN=1,NROD
DUM/=DATA($PWR+I+MC*(NN-1))
IF(DUMY.LE.0.) GO TO 109
SUM=SUM+DUMY*DATA($TROD+NODESF+I+MN*(NN-1+MR*(J-1)))
SAVE=SAVE+DATA($PWR+I+MC*(NN-1))
109 CONTINUE
IF(SAVE.EQ.0.) GO TO 120
TWALL=SUM/SAVE
IF(IDAT($JBOIL+1).NE.0) GO TO 112
IF(TWALL.GE.TF.AND.NCHF.NE.4) IDAT($JBOIL+1)=J
IF(DATA($M+I+MC*(J-1)).GE.MF.AND.NCHF.EQ.4) IDAT($JBOIL+1)=J
CC
110 CONTINUE
112 IF(TWALL.LT.TF) CALL CURVE(DATA($VISCW+I),TWALL,UUF,TT,NPROP,
1 ERROR,1)
IF(IEERROR.GT.1) GO TO 1000
L=IDAT($NTYPE+I)
DATA($FSP+I)=AA(L)+RE*BB(L)+CC(L)
DATA($VISCW+I)=DATA($VISCW+1)/3600.
IF(NVISCW.EQ.1)
1 DATA($FSP+I)=DATA($FSP+1)*(1.+DATA($HPERI+1)/DATA($PERIM+1)+
2 ((DATA($VISCW+1)/DATA($VISCW+1))-0.6-1.0))
150 CONTINUE
RETURN
200 WRITE(13,6) PREF,PP
GO TO 1001
210 WRITE(13,5) PREF,PP
GO TO 1001
1000 WRITE(13,7)
1001 ERROR = 11
RETURN
ERID
SUBROUTINE QPR3(NCHANL, IKASE,TEXT,DATE,TIME,X)
C
C
IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DA
1 ELEV , FERROR , FLO , FTM , GC , GRID , HSURF , MF
2 HFG , HG , I2 , I3 , FERROR , IQP3 , ITERAT , J1 , J2
3 J3 , J4 , J5 , J6 , J7 , NDEBUB , MF , M1 , J
4 NAFACT , NARAMP , NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGAL , NK
6 NRAMP , NROD , NSCBC , NV , NVISCW , PI , PITCH , POWER , PREF
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
8 UF , VF , VFG , VG , Z
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AKL(10) , BB(4) , BX(30) , CC(4) , CGLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFAC(9,10) , GRIDXL(10) , NGAP(2) , MHF(30) , MHG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , NKF(30) , NCH(10) , NGAP(9) ,

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4 PP(30), RCLAD(2), RFUEL(2), SSIGNA(30), TCLAD(2), UNF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30)
C
C
LOGICAL GRID
REAL KIJ, KF, KKF, KCLAD, KFUEL
C
C
COMMON /COBRA3/ MA, MC, MG, MN, MR, MS, MK
1 $$$, SA, SAA, SAC, SALPHA, SAN, SANSME, SB
1 SCCHAN, SCD, SCHFR, SCON, SCP, SD, SDC, SDFDX
2 SDHX, SDHYD, SDHYD, SDIST, SDPK, SDUR, SDR, SF
3 SFACD, SFDIV, SFINLE, SFILUX, SFMULT, SFOLD, SFSP, SFSPLI, SFKFLO,
4 SGAP, SGAPN, SGAPS, SH, SHFILM, SHINLE, SHOLD, SHPERI, SIDARE,
5 SIDFUE, SIDGAP, SIA, SUDOIL, SUK, SLENGT, SLOCA, SLR,
6 SHCFR, SHCFRC, SHCFRR, SNTYPE, SHWRAP, SNWRPS, SP, SPERIM, SPH
7 SPHI, SPRHIC, SPRNTR, SPRNTR, SPW, SPWRF, SOC, SOF, SOPRIM,
8 SQUAL, SRADIA, SRHO, SRHOOL, SSP, ST, STDUMY, STINLE, STROD,
9 SU, SUH, SUSAVE, SUSTAR, SV, SVISC, SVISCH, SVP, SUPA,
A SW, SWOLD, SWP, SWSAVE, SX, SXCROS, SZA, SSB, SAPOLD
C
COMMON/LINK3/DXX,ETIME,GIN,HIN,IB,IG,IN,ISAVE,JUMP,KASE,KT,MAXT,
1 NDT,NDXP1,NFUFLT,NG,NH,NJUMP,NOUT,NP,NPCHAN,NPNODE,NPROD,NQ,NR,
2 NSKIPT,NSKIPX,NTRIES,PEXIT,PHTOT,SAVEED,TIN,TTIME,ZZ
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
C
DIMENSION JB(10),TEXT(17),DATE(2),TIME(3),X(1)
READ (12,700) ZM
WRITE (13,1000) ZM
IF (ZM-GE.0.0) GO TO 505
IF (ZM.LT.-1.01) GO TO 500
ZM = 3*13.0/3.6
GO TO 505
500 ZM = 1000.0/3.6
505 WRITE (13,1001) ZM
NDX:P1=NDX+1
DO 601 I=1,NCHANL
READ(12,700) (DATA(SOF+I+MC*(J-1)),J=2,NDXP1)
IF(IQP3.EQ.0) GO TO 705
DO 602 I=1,NCHANL
READ(12,700) (DATA(SOC+I+MC*(J-1)),J=2,NDXP1)
705 CONTINUE
C
C
PRINT INPUT FUEL NODAL POWERS
WRITE(13,650)
DO 621 I=1,NCHANL,10
DO 5 K=1,10
5 JB(K)=I+K-1
I=I+9
IF(NCHANL.LE.11) II=NCHANL

```

COB24760  
COB24770  
COB24780  
COB24790  
COB24800  
COB24810  
COB24820  
COB24830  
COB24840  
COB24850  
COB24860  
COB24870  
COB24880  
COB24890  
COB24900  
COB24910  
COB24920  
COB24930  
COB24940  
COB24950  
COB24960  
COB24970  
COB24980  
COB24990  
COB25000  
COB25010  
COB25020  
COB25030  
COB25040  
COB25050  
COB25060  
COB25070  
COB25080  
COB25090  
COB25100  
COB25110  
COB25120  
COB25130  
COB25140  
COB25150  
COB25160  
COB25170  
COB25180  
COB25190  
COB25200  
COB25210  
COB25220  
COB25230  
COB25240  
COB25250  
COB25260  
COB25270  
COB25280  
COB25290  
COB25300

```

L=II-I+1
WRITE(13,655) (JB(K),K=1,L)
DO 621 J=1,NDX
WRITE(13,30) J , (DATA($QF+K+MC*(J )),K=1,II)
621 CONTINUE
C
C MULTIPLY FUEL POWERS BY ZM
SUMF = 0.0
DO 630 I=1,NCHANL
DATA($RADIA+I) = 0.0
DO 630 J=2,NDXPI
DATA($QF+I+MC*(J-1))=DATA($QF+I+MC*(J-1))*ZM
DATA($RADIA+I) = DATA($RADIA+I) + DATA($QF+I+MC*(J-1))
630 SUMF = SUMF + DATA($QF+I+MC*(J-1))
SUMC = 0.0
IF(10P3.EQ.0) GO TO 645
C
C PRINT INPUT COOLANT NODAL POWERS
WRITE(13,660)
DO 622 I=1,NCHANL,10
CC 6 K=1,10
6 JB(K)=I+K-1
II=I+9
IF(NCHANL.LE.II) II=NCHANL
L=II-I+1
WRITE(13,655) (JB(K),K=1,L)
DO 622 J=1,NDX
WRITE(13,30) J , (DATA($QF+K+MC*(J )),K=1,II)
622 CONTINUE
C
C MULTIPLY COOLANT POWERS BY ZM
DO 640 I=1,NCHANL
DO 640 J=2,NDXPI
DATA($QF+I+MC*(J-1))=DATA($QF+I+MC*(J-1))*ZM
DATA($RADIA+I) = DATA($RADIA+I) + DATA($QF+I+MC*(J-1))
640 SUMC = SUMC + DATA($QF+I+MC*(J-1))
C
C PRINT FUEL AND COOLANT SUMMED POWERS.
645 SUMT = SUMF+SUMC
WV = FLOAT(NCHANL)/SUMT
DO 647 I=1,NCHANL
647 DATA($RADIA+I) = DATA($RADIA+I)*WV
WRITE (13,1004) (DATA($RADIA+I),I=1,NCHANL)
WV = 3.6/3413.0
SUMF1 = WV*SUMF
SUMC1 = WV*SUMC
SUMT1 = WV*SUMT
SUMF = 0.001*SUMF1
SUMC = 0.001*SUMC1
SUMT = 0.001*SUMT1
WRITE (13,1002) SUMF1,SUMF, SUMC1,SUMC, SUMT1,SUMT
AFLUX = SUMT1*3.413/(PHTOT*Z)
WRITE(13,1003) AFLUX
RETURN
C

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COB25310  
COB25320  
COB25330  
COB25340  
COB25350  
COB25360  
COB25370  
COB25380  
COB25390  
COB25400  
COB25410  
COB25420  
COB25430  
COB25440  
COB25450  
COB25460  
COB25470  
COB25480  
COB25490  
COB25500  
COB25510  
COB25520  
COB25530  
COB25540  
COB25550  
COB25560  
COB25570  
COB25580  
COB25590  
COB25600  
COB25610  
COB25620  
COB25630  
COB25640  
COB25650  
COB25660  
COB25670  
COB25680  
COB25690  
COB25700  
COB25710  
COB25720  
COB25730  
COB25740  
COB25750  
COB25760  
COB25770  
COB25780  
COB25790  
COB25800  
COB25810  
COB25820  
COB25830  
COB25840  
COB25850





```

20 CONTINUE
RETURN
30 WRITE (6,1030) IVAR,IVMAX,CARD
RETURN
C
1000 FORMAT(20A4, T1, 14E5.0)
1001 FORMAT(' INLET FLOW SPLIT', 12X, '...', 20A4, '...', READIN (MODEL))
1002 FORMAT(' INLET ENTHALPIES', 12X, '...', 20A4, '...', READIN (OPERA))
1003 FORMAT(' INLET TEMPERATURES', 10X, '...', 20A4, '...', READIN (OPERA))
1004 FORMAT(' PRESSURE TRANSIENT', 10X, '...', 20A4, '...', READIN (OPERA)
1A))
1005 FORMAT(' INLET ENTHALPY TRANSIENT', 4X, '...', 20A4, '...', READIN (COD20520
1OPERA))
1006 FORMAT(' INLET FLOW TRANSIENT', 8X, '...', 20A4, '...', READIN (OPECOD26540
10A))
1007 FORMAT(' INLET POWER TRANSIENT', 7X, '...', 20A4, '...', READIN (OPCOU26560
1ERA))
1008 FORMAT(' AXIAL HEAT FLUX', 13X, '...', 20A4, '...', READIN(CARD20))
1009 FORMAT(' RADIAL POWERS', 15X, '...', 20A4, '...', READIN(CARD20))
1011 FORMAT(30X, '...', 20A4, '...', CONTINUED)
1030 FORMAT(' IVAR = ', 13, ' NOT 0 - ', 13, 6X, '...', 20A4, '...', REACOD265510
1DIR')
END
C
FUNCTION ROLIQ(P)
MEKIN NEW, AUGUST 1974
U=ALOG(P)
IF(P.LE.450.0) GO TO 2
U=U-7.0
VLIQ=((((-0.26381D-03)*U+0.142678D-02)*U+0.21252D-02)*U
+0.2196328D-1
ROLIQ=1.0/VLIQ
RETURN
END
VLIQ=(((((0.468D-08)*U-0.747D-07)*U+0.39696D-06)*U
1-0.36945D-06)*U-0.204944D-05)*U+0.67462798D-05)*U
2+0.33132739D-04)*U+0.10394514D-03)*U+0.16140836D-1
ROLIQ=1.0/VLIQ
RETURN
END
C
FUNCTION ROVAP(P)
MEKIN NEW, AUGUST 1974
U=ALOG(P)
IF(P.LE.450.0) GO TO 2
U=U-7.0
PVG=(((((0.47458752D01)*U-0.65913524D01)*U-0.22430605D02)*U
1-0.27067054D02)*U-0.53007282D02)*U-0.61514691D02)*U
2+0.43997464D03
ROVAP=P/PVG
RETURN
END
PVG=(((((0.186D-05)*U-0.12008D-03)*U+0.67223D-03)*U
1-0.307139D-02)*U-0.631126D-02)*U+0.60001629D-01)*U
2+0.11039315D01)*U+0.19257401D02)*U+0.33360056D03
ROVAP=P/PVG
RETURN
END

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COB26410  
COB26420  
COB26430  
COB26440  
COB26450  
COB26460  
COB26470  
COB26480  
COB26490  
COB26500  
COB26510  
COB26520  
COB26530  
COB26540  
COB26550  
COB26560  
COB26570  
COB26580  
COB26590  
COB26600  
COB26600  
COB26620  
COB26630  
COB26640  
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COB26670  
COB26680  
COB26690  
COB26700  
COB26710  
COB26720  
COB26730  
COB26740  
COB26750  
COB26760  
COB26770  
COB26780  
COB26790  
COB26800  
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COB26940  
COB26950



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COB27510
COB27520
COB27530
COB27540
COB27550
COB27560
COB27570
COB27580
COB27590
COB27600
COB27610
COB27620
COB27630
COB27640
COB27650
COB27660
COB27670
COB27680
COB27690
COB27700
COB27710
COB27720
COB27730
COB27740
COB27750
COB27760
COB27770
COB27780
COB27790
COB27800
COB27810
COB27820
COB27830
COB27840
COB27850
COB27860
COB27870
COB27880
COB27890
COB27900
COB27910
COB27920
COB27930
COB27940
COB27950
COB27960
COB27970
COB27980
COB27990
COB28000
COB28010
COB28020
COB28030
COB28040
COB28050

XX=ALOG(P)
U=DBLE(XX)
IF(P.LE.450.0) GO TO 2
U=U-7.000
XATTEM=(((1-0.160742250-00-U-0.6967857500)*U+0.6178111900)*U
1+0.14657783002)*U+0.12405875003)*U+0.55599496003
SATTEN=SNGL(XATTEM)
RETURN
2 XATTEM=((-0.1900-05+U+0.14050-04)*U-3.2650-5)*U+
1-2.39070-3)
XATTEM=(((XATTEM*U+0.4346180-02)*U+0.1736300400)*U+0.22808149001)
XATTEM=(((XATTEM*U+0.39446776002)*U+0.1018249403)
SATTEN=SNGL(XATTEM)
RETURN
END
SUBROUTINE SOLVE(NN,LMAX,MID,UL,X,B,NK)
DIMENSION UL(NK,1),X(1),B(1)
STORE DIAGONAL BAND OF AAA MATRIX. POSITION (K,L) IN SQUARE
ARRAY BECOMES (K,(MID-K+L)) IN NEW ARRAY.
N = NN
IF(N.EQ.1) GO TO 5
NPI = N+1
C
X(1) = B(1)
DO 2 I = 2,N
IM1 = I-1
SUM = 0.0
JMIN = MAXO(1,(I-MID+1))
DOUBLE PRECISION MAY BE REQUIRED FOR INNER LOOP.
DO 1 J = JMIN,IM1
JJ = MID-I+J
1 SUM = SUM + UL(I,JJ)*X(J)
2 X(I) = B(I) - SUM
C
X(N) = X(N)/UL(N,MID)
DO 4 IBACK = 2,N
I = NPI-IBACK
1 GOES (N-1),...,1
IPI = I+1
SUM = 0.0
DOUBLE PRECISION MAY BE REQUIRED FOR INNER LOOP.
JMAX = MINO(N,(I-MID-1))
DO 3 J = IPI,JMAX
JJ = MID-I+J
3 SUM = SUM + UL(I,JJ)*X(J)
4 X(I) = (X(I)-SUM)/UL(I,MID)
RETURN
5 X(1) = B(1)/UL(1,MID)
RETURN
END
SUBROUTINE SPLIT
THIS PROCEDURE CONTAINS THE COMMON AND TYPE STATEMENTS SHARED BY THE COB28020
MAJOR SUBROUTINES OF COBRA-IIIC.
C
C
C

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```

IMPLICIT INTEGER (5)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1  ELEV , FERROR , FLO , FTW , GC , GK , GRID , HSURF , MF ,
2  HFG , HG , I2 , I3 , IERROR , IQP3 , ITERAT , J1 , J2 ,
3  J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ ,
4  NAFAC , NARAMP , NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF
5  NGAPS , NGRID , NGRID , NGTYPE , NGXL , NK , NODES , NODESF , NPROP
6  NRAMP , NRDD , NSCBC , NV , NVISCH , PI , PITCH , POWER , PREF
7  QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
8  VAF , VAF , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30)
1  AXL(10) , BB(4) , BK(30) , CC(4) , CCLAD(2) , CFUEL(2) ,
2  GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3  IGRID(10) , KCLAD(2) , KFUEL(2) , KMF(30) , NCH(10) , NGAP(9) ,
4  PI(30) , RCLAD(2) , RFUEL(2) , SIGMA(30) , TCLAD(2) , UUF(30) ,
5  VV(30) , VVG(30) , XQUAL(30) , Y(30) , Y(30) , Y(30)
C
LOGICAL GRID
REAL KIJ , KF , KMF , KCLAD , KFUEL
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MX
1  SCCHAN , SCD , SCHR , SCON , SCND , SCP , SD , SANSWE , SB
2  SDHX , SDHYD , SDHYDM , SDIST , SDPX , SDPK , SDUR , SDR , SF
3  $FACTO , $FDIV , $FINLE , $FLUX , $FRULT , $FOLD , $FSP , $FSPLI , $FXFLO ,
4  $GAP , $GAPN , $GAPS , $H , $HFLM , $HINLE , $HOLD , $HPERI , $IDARE ,
5  $IDUE , $IDGAP , $IK , $JBOIL , $JK , $LENGT , $LOCA , $LR ,
6  $MCHFR , $MCFRC , $MCFRR , $NTYPE , $NWRAP , $NWRAPS , $P , $PERIM , $PH
7  $PHI , $PRNTC , $PRNTR , $PRINT , $PM , $PMRF , $QC , $QF , $OPRIM
8  $QUAL , $RADIO , $RHO , $RHODL , $SP , $T , $TDUMY , $TINLE , $TROD ,
9  $U , $UH , $USAVE , $USTAR , $V , $VISC , $VISCW , $VP , $VPA
A  $W , $WOLD , $WP , $WSAVE , $X , $XCROS , $XA , $XB , $XPOLD
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
EQUIVALENCE (NCHAN,NCHANL)
C CORRECT FLOW ESTIMATE BY ITERATION. THIS PROCEDURE ASSUMES THERE IS NO CORRECT FLOW
C DENSITY CHANGE WITH LENGTH AND THAT NO DIVERSION CROSSFLOW IS OCCURRING
C CONVERGENCE TOLERANCE IS E.
E=0.005
SAVEDT = DT
DT = 1.E+10
DO 10 I=1,NCHANL
DATA($F+I)=DATA($FINLE+I)
DATA($H+I)=DATA($HINLE+I)
DO 100 N=1,200
CALL PROP(2,1)
IF(ITEROR.GT.1) GO TO 1000

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10

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CALL VOID(1)
DO 15 I=1,NCHANL
DATA(SVP+I)=DATA(SVP+1)/DATA(SA+I)
15 IF(1ERROR.GT.1) GO TO 1000
IF(IFM.GT.0.) CALL MIX(1)
IF(1ERROR.GT.1) GO TO 1000
CALL DIFFER(3,1)
IF(1ERROR.GT.1) GO TO 1000
DPAVG = 0.
DO 20 I=1,NCHANL
DPAVG=DPAVG+DATA(SDPDX+I)*DATA(SA+I)
20 DPAVG = DPAVG/ATOTAL
J=2
FTOT = 0.
DO 30 I=1,NCHANL
DELTA F=(DPAVG-DATA(SDPDX+I))*0.5/DATA(SDPDX+I)*DATA(SF+I)
IF(IFM.GT.0.) DELTA F = DELTA F*0.5
FSAVE =DATA(SF+I)
DATA(SF+I)=DATA(SF+I)+DELTA F
IF (DATA(SF+I).LT.0.) GO TO 1000
IF(ABS (DATA(SF+I)-FSAVE)/FSAVE.GT. E) J=J+1
FTOT=FTOT+DATA(SF+I)
30 CONTINUE
DO 40 I=1,NCHANL
DATA(SF +I)=DATA(SF+I)*FLO/FTOT
40 DATA(SFINE+I)=DATA(SF+I)
IF(J.GT.1) GO TO 120
100 CONTINUE
1000 WRITE(13,1) (I,DATA(SF+I),DATA(SDPDX+I),I=1,NCHANL)
1 FORMAT (40H FLOW SPLIT TO GIVE EQUAL DP/DX FAILED / (15,2E14.6))
1 ERROR = 8
120 DT = SAVEDT
RETURN
END
SUBROUTINE SURTEIN(P,RL,RG,ST)
C MEKIN HEW. AUGUST 1974
X=RL-RG
X=0.000001*X**4
ST=X*(4.60+1.84/EXP(0.685*X))+0.232+EXP(1.56*(X-15.0))
RETURN
END
SUBROUTINE TEMP (T,DUM,N,JJ,A,B)
C SUBROUTINE TEMP CALCULATES THE TRANSIENT TEMPERATURE DISTRIBUTION
C IN A CYLINDRICAL OR PLATE NUCLEAR FUEL ELEMENT WHERE THE LARGEST
C NUMBER NODE IS THE CLADDING. FOR TRANSIENT CALCULATIONS, FLUID
C DATA AT T IS USED TO CALCULATE THE TEMPERATURE AT T+DT BY USING
C A STABLE IMPLICIT NUMERICAL TECHNIQUE.
C SIMULTANEOUS EQUATIONS ARE SOLVED USING A COMPACT ELIMINATION
C SCHEME FOR TRI-DIAGONAL MATRICES.
C THE VALUE OF T UPON ENTRY IS THE TEMPERATURE AT ORIGINAL TIME.
C AT EXIT T IS THE TEMPERATURE DELTA-T LATER IN TIME.

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COB28610  
COB28620  
COB28630  
COB28640  
COB28650  
COB28660  
COB28670  
COB28680  
COB28690  
COB28700  
COB28710  
COB28720  
COB28730  
COB28740  
COB28750  
COB28760  
COB28770  
COB28780  
COB28790  
COB28800  
COB28810  
COB28820  
COB28830  
COB28840  
COB28850  
COB28860  
COB28870  
COB28880  
COB28890  
COB28900  
COB28910  
COB28920  
COB28930  
COB28940  
COB28950  
COB28960  
COB28970  
COB28980  
COB28990  
COB29000  
COB29010  
COB29020  
COB29030  
COB29040  
COB29050  
COB29060  
COB29070  
COB29080  
COB29090  
COB29100  
COB29110  
COB29120  
COB29130  
COB29140  
COB29150

```

C
  IMPLICIT INTEGER (S)
  COMMON /COBRA1/ ABETA ,AFLUX ,ATOTAL,BBETA ,DIA ,DT ,DK ,
  1 ELEV ,FERROR,FLO ,FTM ,GC ,GK ,GRID ,HSURF ,MF ,
  2 HFG ,HG ,I2 ,I3 ,IERROR,IQ3 ,ITERAT,J1 ,J2 ,
  3 J3 ,JA ,J5 ,J6 ,J7 ,KDEBUG,KF ,KIJ ,
  4 NAFAC,NARAMP,NAX ,NBBG ,NCHAN ,NCHF ,NDX ,NF ,
  5 NGAPS ,NGRID ,NGRIDT,NGTYPE,NGXL ,NK ,NODES ,NODESF,NPROP ,
  6 NRAMP ,NRDD ,NSCBC ,NV ,NVISCM,PI ,PITCH ,POWER ,PREF ,
  7 OAX ,RHDF ,RHOG ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK ,
  8 UF ,VF ,VFG ,VG ,Z ,
  COB29160
  COB29170
  COB29180
  COB29190
  COB29200
  COB29210
  COB29220
  COB29230
  COB29240
  COB29250
  COB29260
  COB29270
  COB29280
  COB29290
  COB29300
  COB29310
  COB29320
  COB29330
  COB29340
  COB29350
  COB29360
  COB29370
  COB29380
  COB29390
  COB29400
  COB29410
  COB29420
  COB29430
  COB29440
  COB29450
  COB29460
  COB29470
  COB29480
  COB29490
  COB29500
  COB29510
  COB29520
  COB29530
  COB29540
  COB29550
  COB29560
  COB29570
  COB29580
  COB29590
  COB29600
  COB29610
  COB29620
  COB29630
  COB29640
  COB29650
  COB29660
  COB29670
  COB29680
  COB29690
  COB29700

C
  LOGICAL GRID
  REAL KIJ ,KF ,KXF ,KCLAD ,MFUEL

C
  COMMON /COBRA3/ MA ,MC ,MG ,MN ,MR ,MS ,MX ,
  1 $$$ ,SA ,SAAA ,SAC ,SALPHA,SAN ,SANSNE,$B ,
  1 SCCHAN,$CD ,SCHFR,$CON ,SCOND,$CP ,SD ,SDC ,SDFA ,
  2 $DHUX,$DHVD,$DHVDN,$DIST ,SDPK ,SDUR ,SDR ,SF ,
  3 $FACTO,$FDIV,$FINLE,$FLUX,$FRULT,$FOLD,$FSP,$FSPLI,$FXFLO,
  4 $GAP,$GAPP,$GAPS,$H,$SHFILM,$SHINLE,$SHOLD,$SHPERI,$SIDARE,
  5 $IDTUE,$IDCAP,$IK,$SBOIL,$JK,$SLC,$SLENGT,$LOCA,$LR ,
  6 $MCHFR,$MFCRC,$MFCRR,$SNTYPE,$NWRAP,$NWRPS,$P,$PERIM,$PH ,
  7 $PHI,$SPRNTC,$SPRNTR,$SPRNTL,$SPW,$SPWRP,$SQ,$SQF,$SQPRIM,
  8 $QUAL,$RADIA,$RHO,$RHOOL,$SP,$ST,$STOUMY,$TINLE,$TOD ,
  9 $U,$UH,$USAVE,$USTAR,$V,$VISC,$VISCW,$VP,$VPA ,
  A $W,$WOLD,$WP,$WSAVE,$X,$XCROSS,$XA,$XSB,$XAPOLD

C
  COMMON DATA(1)
  LOGICAL LDAT(1)
  INTEGER IDAT(1)
  EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))

C
  DIMENSION A(3,1),B(1),T(1)
  REAL KFDR2

C
  SETUP A MATRIX OF THE FORM A*T=B WHERE ONLY THE 3 DIAGONALS OF
  A ARE STORED.
  NMI = NODESF-1
  NP1 = NODESF+1
  IF(NODESF.LE.0) GO TO 1000
  J=IDAT($IDFUE+N)
  DR = DFUEL(J)*.5/FLOAT(NMI)
  DR2 = DR**2
  RCFUL = RFUEL(J)*CFUEL(J)/DT

```

```

KFDR2 = KFUEL(J)/DR2
HGAPI = 1./((1./HGAPI(J) + TCLAD(J)/KCLAD(J))
QCLAD = 0.
C J IS THE FUEL TYPE CODE. CYLINDRICAL FUEL, J=1. PLATE FUEL, J=2.
IF(J.EQ.2) GO TO 101
C
C THIS SECTION FOR CYLINDRICAL FUEL RODS.
QFUEL=DATA($FLUX+HMR*(JU-1))*4.*DATA($D+N)/DFUEL(J)**2
DO 100 I=1,NPI
IF(I.GT.1) GO TO 10
A(2,I) = RCFUEL + 4.*KFDR2
A(3,I) = -4.*KFDR2
GO TO 80
10 IF(I.GT.NM1) GO TO 20
A(1,I) = -KFDR2*(1.-1./FLOAT(2*I-2))
A(2,I) = RCFUEL + 2.*KFDR2
A(3,I) = -KFDR2*(1.+1./FLOAT(2*I-2))
GO TO 80
20 IF(I.EQ.NPI) GO TO 30
A(1,I) = -2.*KFDR2
A(2,I) = RCFUEL + 2.*KFDR2 + 2.*HGAPI/DR + HGAPI/DR/FLOAT(I-1)
A(3,I) = -(2.*HGAPI/DR + HGAPI/DR/FLOAT(I-1))
GO TO 80
30 A(1,I)=-HGAPI/TCLAD(J)*DFUEL(J)/DATA($D+N)
A(2,I) = RCLAD(J)*CCLAD(J)/DT+HGAPI/TCLAD(J) * DFUEL(J)/DATA($D+N)
1 + HSURF/TCLAD(J)
80 IF(I.EQ.NPI) GO TO 90
B(1) = QFUEL + RCFUEL*T(I)
GO TO 100
90 B(1) = QCLAD + RCLAD(J)*CCLAD(J)/DT*T(I) + HSURF/TCLAD(J)*TFLUID
100 CONTINUE
C SOLVE FOR TEMPERATURES
CALL GAUSS(1,NPI,A,B,T)
RETURN
C
C THIS SECTION FOR FLAT PLATE FUEL.
101 QFUEL=DATA($FLUX+HMR*(JU-1))*2./DFUEL(J)
DO 200 I=1,NPI
IF(I.GT.1) GO TO 110
A(2,I) = RCFUEL + KFDR2*2.
A(3,I) = -2.*KFDR2
GO TO 180
110 IF(I.GT.NM1) GO TO 120
A(1,I) = -KFDR2
A(2,I) = RCFUEL + 2.*KFDR2
A(3,I) = -KFDR2
GO TO 180
120 IF(I.EQ.NPI) GO TO 130
A(1,I) = -2.*KFDR2
A(2,I) = RCFUEL + 2.*KFDR2 + 2.*HGAPI/DR
A(3,I) = -2.*HGAPI/DR
GO TO 180
130 A(1,I) = -HGAPI/TCLAD(J)
A(2,I) = RCLAD(J)*CCLAD(J)/DT + HGAPI/TCLAD(J) + HSURF/TCLAD(J)
180 IF(I.EQ.NPI) GO TO 190

```

COB29710  
COB29720  
COB29730  
COB29740  
COB29750  
COB29760  
COB29770  
COB29780  
COB29790  
COB29800  
COB29810  
COB29820  
COB29830  
COB29840  
COB29850  
COB29860  
COB29870  
COB29880  
COB29890  
COB29900  
COB29910  
COB29920  
COB29930  
COB29940  
COB29950  
COB29960  
COB29970  
COB29980  
COB29990  
COB30000  
COB30010  
COB30020  
COB30030  
COB30040  
COB30050  
COB30060  
COB30070  
COB30080  
COB30090  
COB30100  
COB30110  
COB30120  
COB30130  
COB30140  
COB30150  
COB30160  
COB30170  
COB30180  
COB30190  
COB30200  
COB30210  
COB30220  
COB30230  
COB30240  
COB30250



```

R(1) = OFUEL + RCFUEL*(1)
GO TO 200
190 D(1) = OCLAD + RCLAD(J)*CCLAD(J)/DT*(1) + HSURF/TCLAD(J)*TFLUID
200 CONTINUE
C SOLVE FOR TEMPERATURES
CALL GAUSS(1,NPI,A,B,T)
RETURN
1000 IERROR = 15
RETURN
END
SUBROUTINE TIDY
C
C IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX ,
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , MF ,
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ ,
4 NFACT , NARAMP , NAX , NAXL , NBBC , NCHANL , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , NPROP ,
6 NRAMP , NRDD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 OAX , RHDF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VF , VFG , VG , Z
C
C LOGICAL GRID
REAL KIJ , KF , KKF , KCLAD , KFUEL
C
COMMON /COBRA3/ MA , MC , MG , MH , MI , MJ , MK ,
1 $$$ , SA , SAA , SAC , SALPHA , SAN , SANSWE , SB ,
2 SCCHAN , SCD , SCHFR , SCON , SCND , SCP , SD , SDC , SDFDX ,
3 SDHDX , SDHYD , SDHYDN , SDIST , SDPDX , SDPK , SDUR , SDR , SF ,
4 $FACTO , $FDIV , $FHLE , $FLUX , $FMULT , $FOLD , $FSP , $FSPLI , $FXFLO ,
5 SGAP , SGAPN , SGAPS , SH , SHFILM , SHINLE , $HOLD , $HPERI , $IDARE ,
6 $IDFUE , $IDGAP , $IK , $JBOIL , $JK , $LENGT , $LOCA , $LR ,
7 SMCHFR , $CFERC , $CFERR , $NTYPE , $NWRAP , $NWRPS , $P , $PERIM , $PH ,
8 $PHI , $PRITC , $PRITR , $PRINTN , $PW , $PWRF , $SOC , $SOF , $SOPRIM ,
9 $SU , $SUAL , $RADIA , $RH0 , $RH0OL , $SP , $ST , $TDUMY , $TINLE , $TROD ,
A $W , $WOLD , $WP , $WSAVE , $X , $XCROSS , $XA , $XB , $XPOLD
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
COMMON /LINK3/ DAX , ETIME , GIN , HIN , IG , IN , ISAVE , JUMP , KASE , KT , MART ,
1 NDT , NDP1 , NFUEL , NG , NH , NUIMP , NOUT , NP , NPCHAN , NPNODE , NP ROD , NQ , NR ,
2 NSKIPT , NSKIPTX , NTRIES , PEXIT , PHOTOT , SAVEDT , TIN , TTIME , ZZ
DIMENSION NTHBOX(20,20)
C
C TIDY UP FOR INPRIN.
C ZZ = 12.0*Z
    
```

```

DO 4 I=1,NCHANL
DATA($AC+I) = 144.0*DATA($A+I)
DATA($PW+I) = 12.0*DATA($PERIM+I)
DATA($PH+I) = 12.0*DATA($SPERI+I)
DATA($DC+I) = 12.0*DATA($DHYD+I)
DATA($DR+I) = 12.0*DATA($D+I)
DO 4 L=1,4
IDAT($LC+I+MC*(L-1)) = 0
DATA($DIST+I+MC*(L-1)) = 0.0
DATA($GAPS+I+MG*(L-1)) = 0.0
4 CONTINUE
C
IF (NK.EQ.0) RETURN
DO 12 K=1,NK
I = IDAT($IK+K)
J = IDAT($JK+K)
DO 8 L=1,4
IF (IDAT($LC+I+MC*(L-1)).EQ.0) GO TO 10
8 CONTINUE
WRITE (6,2004) K,J,I
10 IDAT($LC+I+MC*(L-1)) = J
DATA($DIST+I+MC*(L-1)) = DATA($LENGT+K)*12.0
12 DATA($GAPS+I+MG*(L-1)) = DATA($GAP+K)*12.0
C
RETURN
2004 FORMAT(' CARDS4 GAP CONNECTION ', I3, ' CHANNEL ', I3,
1 ' IS 5TH ADJACENT TO ', I3)
END
SUBROUTINE TOD(A)
DIMENSION A(3),DATIM(5)
CALL WHEN(DATIM)
A(1)=DATIM(3)
A(2)=DATIM(4)
A(3)=DATIM(5)
RETURN
END
SUBROUTINE ACOL(IFROM,JK,JKWMAX,LOCA,MA,MS,NK,MG,IPILE)
DIMENSION IK(1),JK(1),LOCA(MG,14)
SET LOCA, DEFINING INTERACTING BOUNDARIES
C IFROM = 1, CALLED FROM CARDS4,
C LOCA(IK,1)=K. LOCA(JK,L),L=2,7 SPECIFIES UP TO LOCA(K,8)
C BOUNDARIES ADJACENT TO CHANNELS DEFINING BOUNDARY K.
DO 8 K=1,NK
IF (IPILE.GT.0) GO TO 107
DO 103 L=2,13
103 LOCA(K,L)=0
GO TO 110
107 DO 3 L=2,7
3 LOCA(K,L)=0
110 N=1
LOCA(K,1) = K
11 = IK(K)
JJ = JK(K)
4 DO 7 KK=1,NK
111 = IK(KK)

```

```

COB30810
COB30820
COB30830
COB30840
COB30850
COB30860
COB30870
COB30880
COB30890
COB30900
COB30910
COB30920
COB30930
COB30940
COB30950
COB30960
COB30970
COB30980
COB30990
COB31000
COB31010
COB31020
COB31030
COB31040
COB31050
COB31060
COB31070
COB31080
COB31090
COB31100
COB31110
COB31120
COB31130
COB31140
COB31150
COB31160
COB31170
COB31180
COB31190
COB31200
COB31210
COB31220
COB31230
COB31240
COB31250
COB31260
COB31270
COB31280
COB31290
COB31300
COB31310
COB31320
COB31330
COB31340
COB31350

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CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

```

IF (III.GT.II) GO TO 7
JJJ = JK(KK)
IF ( (II.EQ.III) .OR. (II.EQ.JJJ) ) GO TO 6
GO TO 7
6 IF ( (III+JJJ - II-JJ) .EQ. 0) GO TO 7
N = N+1
LL = III
IF (II.EQ.III) LL=JJJ
MV = FLOAT(II-LL)/FLOAT(II-JJ)
LOCA(K,N) = MK
IF (MV.LT.0.0) LOCA(K,N)=-MK
7 CONTINUE
IF (IPILE.GT.0) GO TO 108
LOCA(K,14)=N
GO TO 109
108 LOCA(K,8)=N
109 IF(II.GE.JJ) GO TO 8
II = JK(K)
JJ = IK(K)
GO TO 4
8 CONTINUE
C
C FIND STRIPE WIDTH FOR AAA MATRIX IN DIVERT
MAX = 0
DO 10 K=1,NK
N=LOCA(K,8)
IF (IPILE.GT.0) GO TO 111
N=LOCA(K,14)
111 DO 10 L=2,N
LKL = ABS(LOCA(K,L))
J = IABS(K-LKL)
IF (J.LT.MAX) GO TO 10
MAX = J
KIMAX = K
10 CONTINUE
MS = 2*MAX + 1
CALL CORE2(MS,NK)
RETURN
END
SUBROUTINE CARDS4(AC,DC,DIST,DR,GAPS,LC,MA,MG,N1,N2,NCHF,NFUEL,
1 PH,PHIOT,PRINT,PW,MC)
C
C-----NOTE THAT THESE COMMON AREAS ARE NOT IDENTICAL WIT THOSE
C
C IN OTHER ROUTINES
C
C
IMPLICIT INTEGER ($)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL,BBETA , DIA , DT , DX
1 ELEV , FERROR,FLO , FTM , GC , GK , GRID ,HSURF , HF
2 HFG ,HG ,I2 ,I3 , IERROR,IOP3 ,ITERAT ,J1 ,J2
3 J3 ,J4 ,J5 ,J6 ,J7 ,KDEBUG,KF ,KIJ ,
4 NAFACT,NARAMP,NAX ,NAXL ,NBBC ,NCHAN ,DUM1 ,NOX ,NF
5 NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGXL ,NK ,NODES ,NODESF ,NPROP ,COB31880
6 NRAMP ,NROD ,NSCBC ,NV ,NVISCH,PI ,PITCH ,POWER ,PREP ,COB31900

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COB31360  
COB31370  
COB31380  
COB31390  
COB31400  
COB31410  
COB31420  
COB31430  
COB31440  
COB31450  
COB31460  
COB31470  
COB31480  
COB31490  
COB31500  
COB31510  
COB31520  
COB31530  
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COB31500  
COB31590  
COB31600  
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COB31650  
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COB31780  
COB31790  
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COB31860  
COB31870  
COB31880  
COB31890  
COB31900

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7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK , COB31910
8 UF ,VF ,VFG ,VG ,Z , COB31920
C COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30), COB31940
1 AX(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2), COB31950
2 GAPXL(10), GFAC(9,10), GRIDXL(10), HGAP(2), HHF(30), HHC(30), COB31960
3 IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9), COB31970
4 PPI(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30), COB31980
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30) COB31990
C COB32000
C LOGICAL GRID,PRINT
C REAL MIJ, KF, KKF, KCLAD, KFUEL
C COB32010
C COB32020
C COB32030
C COB32040
C COB32050
C COMMON /COBRA3/ DUM2 ,DUMC ,DUM3 ,MIN ,MR ,MS ,MX , COB32060
1 $$$ ,SA ,SAA ,SAC ,SALPHA,$AN ,SANSME,$O , COB32070
2 SCCHAN,$CD ,SCHFR ,$CON ,SCOND ,SCP ,SD ,SDC ,SDFDX , COB32080
3 SDHX ,SDHYD ,SDHYDN ,SDIST ,SDPDX ,SDPK ,SDUR ,SDR ,SF , COB32090
4 SFACIO,$DIV ,$FILE,$FLUX ,$FMULT,$FOLD,$FSP ,$FSPLI,$FXFLO, COB32100
5 $GAP , $GAPN , $CAPS , $H , $HFILM,$HINLE,$HOLD , $HPERI,$IDARE, COB32110
6 SIDFUE,$IDGAP,$IK , $JBOIL,$JK , $LC , $LENGT,$LOCA , $LR , COB32120
7 $MCHFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP,$NWRPS,$P , $PERIM,$PH , COB32130
8 $PHI , $PRHTC,$PRNTR , $PRNTH , $PM , $PMRF , $SOC , $SF , $SQPRIM, COB32140
9 $QUAL , $RADIA,$RHO , $RHODL,$SP , $ST , $STDUMY,$TINLE,$TROD , COB32150
A $U , $UH , $USAVE , $USTAR,$V , $VISC , $VISCV,$VP , $VPA , COB32160
A $W , $WOLD , $WP , $WSAVE,$X , $XCROSS,$XA , $XB , $XPOLD , COB32170
C COB32180
C COB32190
C COB32200
C COB32210
C COB32220
C COB32230
C COB32240
C COB32250
C COB32260
C COB32270
C COB32280
C COB32290
C COB32300
C COB32310
C COB32320
C COB32330
C COB32340
C COB32350
C COB32360
C COB32370
C COB32380
C COB32390
C COB32400
C COB32410
C COB32420
C COB32430
C COB32440
C COB32450

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```

COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C EQUIVALENCE (NCHAN,NCHANL)
C DIMENSION AC(1),DC(1),DR(1),PH(1),PRINT(12),PH(1),DIST(1),
1 GAPS(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),LC(1),
2 TEXT(20),MAAP(2,20)
C MEKIN - ENTERED FOR PMR AND BWR SIMPLIFIED INPUT DATA.
C COMBINES CARD GROUPS 4, 7, 8 IE CHAN GEOMETRY, SPACERS AND RODS.
C READ (A) INDICATORS, (B) CHAN GEOM + SPACERS FOR EACH GROUP,
C (C) ROD POWERS, (D) SPACER X/L, (E) CHANNELS IN GROUPS 2,3 ETC.
C (F) GAP CONNECTIONS, (G) FUEL DATA
C READ INDICATORS. INITIALISE
C READ (12,1001) NI,N2,NGRID,NGRIDT,MODESF,NFUELT,NCHF,IMAP,ITEXT
C IF (NI.LE.15) GO TO 1
C WRITE (13,2001)
C IERROR = 1
C RETURN
C 1 IF (ITEXT.LE.0) GO TO 3
C DO 2 I=1,ITEXT
C READ (12,1005) TEXT
C 2 WRITE(13,1005) TEXT

```

```

3 NCHANL = N2
  NROD = N2
  J6 = 2
  NRAMP = 1
  GRID = .FALSE.
  NGRT = MAXO(NGRIDT,1)
  IPILE = J7
  DO 4 I=1,NCHANL
    DO 4 L=1,6
      IDAT($LR+I+MR*(L-1))=0
      DATA($PHI+I+MR*(L-1))=0.
      IF (L.GT.4) GO TO 4
      LC(I,L) = 0
      GAPS(I,L) = 0.0
      DIST(I,L) = 0.0
4 CONTINUE
C
C READ GEOM AND SPACER DATA FOR EACH CHANNEL GROUP. SET GROUP 1
DO 10 J=1,N1
  READ (12,1002) N1,FRAC,AC(I),PH(I),PH(I),GAPS(I,1),DIST(I,1),
1 DR(I),DATA($PHI+I),M
  DATA($CD+I)=0.
  FXF(1) = 0.0
  IF (FRAC.LE.0.0) FRAC = 1.0
  AC(I) = FRAC*AC(I)
  PW(I) = FRAC*PW(I)
  PH(I) = FRAC*PH(I)
  DATA($PHI+I)=FRAC*DATA($PHI+I)
  IF (NGRID.EQ.0) GO TO 6
  READ(12,1003) (DATA($CD+I+MC*(L-1)),L=1,NGRIDT), (FXF(L),L=1,NGRIDT)
1
6 IDAT($NTYPE+I)=J
  IFRIC(J) = MAXO(N,1)
  IDAT($IDFUE+I)=MAXO(M,1)
  IGROUP(J) = 1
  IF (J.GT.1) GO TO 10
C SET ALL CHANNELS TEMPORARILY TO GROUP 1 VALUES.
DO 8 K=1,NCHANL
  AC(K) = AC(I)
  PW(K) = PW(I)
  PH(K) = PH(I)
  GAPS(K,1) = GAPS(I,1)
  DIST(K,1) = DIST(I,1)
  DR(K) = DR(I)
  DATA($PHI +K)=DATA($PHI +I)
  IDAT($NTYPE+K)=1
  IDAT($IDFUE+K)=IDAT($IDFUE+I)
DO 8 L=1,NGRT
  DATA($CD+K+MC*(L-1))=
1 DATA($CD+I+MC*(L-1))
8 CONTINUE
10 CONTINUE
DO 12 K=1,NG
DO 12 L=1,NGRT
12 DATA($FXFLO+K+MG*(L-1))=FXF(L)

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COB32460  
COB32470  
COB32480  
COB32490  
COB32500  
COB32510  
COB32520  
COB32530  
COB32540  
COB32550  
COB32560  
COB32570  
COB32580  
COB32590  
COB32600  
COB32610  
COB32620  
COB32630  
COB32640  
COB32650  
COB32660  
COB32670  
COB32680  
COB32690  
COB32700  
COB32710  
COB32720  
COB32730  
COB32740  
COB32750  
COB32760  
COB32770  
COB32780  
COB32790  
COB32800  
COB32810  
COB32820  
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COB32840  
COB32850  
COB32860  
COB32870  
COB32880  
COB32890  
COB32900  
COB32910  
COB32920  
COB32930  
COB32940  
COB32950  
COB32960  
COB32970  
COB32980  
COB32990  
COB33000

```

C
C
C   READ ROD POWER FACTORS AND SPACER LOCATIONS.
II = #IHO(NROD,16)
READ(12,1003) (DATA($RADIA+1), I=1, II)
IF (NGRID.GT.0) READ(12,1004) (GRIDAL(I), IGRID(I), I=1, NGRID)
DO 14 I=1, NROD
14   DATA($RADIA+1)=1.0
GO TO 18
16 IF (NROD.GT.16) READ(12,1003) (DATA($RADIA+1), I=17, NROD)
18 IF (NGRID.GT.0) READ(12,1004) (GRIDAL(I), IGRID(I), I=1, NGRID)
C
C   READ CHANNEL NUMBERS NOT IN GROUP 1, SET DATA
JCHECK = 1
IF (N1.EQ.1) GO TO 28
DO 26 J=2, N1
ICHECK = 0
20 READ(12,1001) (JB(I), I=1, 20)
DO 22 J=1, 20
K = JB(J)
IF (K.LE.0) GO TO 24
I = IGROUP(J)
AC(K) = AC(I)
PW(K) = PW(I)
PH(K) = PH(I)
GAPS(K,1) = GAPS(I,1)
DIST(K,1) = DIST(I,1)
DR(K) = DR(I)
DATA($PHI+K)=DATA($PHI+I)
IDAT($N1YPE+K)=J
IDAT($IDFUE+K)=IDAT($IDFUE+I)
IF (K.EQ.1) ICHECK=1
IF (K.EQ.IGROUP(1)) JCHECK=0
DO 22 L=1, NGR1
DATA($CD+K+MC*(L-1))=
1DATA($CD+I+MC*(L-1))
22 CONTINUE
GO TO 20
24 IF (ICHECK.EQ.1) GO TO 26
WRITE(13,2002) J, IGROUP(J)
ERROR = 1
RETURN
26 CONTINUE
IF (JCHECK.EQ.1) GO TO 28
J = 1
WRITE(13,2002) J, IGROUP(J)
ERROR = 1
RETURN
C
C   SET ROD POWER FRACTIONS AND CHANNEL PARAMETERS
28 PHTOT = 0.0
ATOTAL = 0.0
DO 32 I = 1, NCHANL
DO 30 J=1, NROD
30 DATA($PWR+I+MC*(J-1))=0
DATA($PWR+I+MC*(I-1))-DATA($PHI +I)

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COB33010  
COB33020  
COB33030  
COB33040  
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COB33060  
COB33070  
COB33080  
COB33090  
COB33100  
COB33110  
COB33120  
COB33130  
COB33140  
COB33150  
COB33160  
COB33170  
COB33180  
COB33190  
COB33200  
COB33210  
COB33220  
COB33230  
COB33240  
COB33250  
COB33260  
COB33270  
COB33280  
COB33290  
COB33300  
COB33310  
COB33320  
COB33330  
COB33340  
COB33350  
COB33360  
COB33370  
COB33380  
COB33390  
COB33400  
COB33410  
COB33420  
COB33430  
COB33440  
COB33450  
COB33460  
COB33470  
COB33480  
COB33490  
COB33500  
COB33510  
COB33520  
COB33530  
COB33540  
COB33550

```

IDAT($LR+1)=I
DATA($D+1)=DR(I)/12.0
DATA($PERIM+1)=PW(I)/12.0
DATA($PERI+1)=PH(I)/12.0
DATA($AN +1)=AC(I)/144.0
DATA($A +1)=DATA($AN+1)
DC(I) = 4.*AC(I)/PW(I)
DATA($HYD +1)=DC(I)/12.0
DATA($DHYD+1)=DATA($HYD+1)
PHOT=PHOT+ DATA($PERI+1)
32 ATOTAL=ATOTAL+ DATA($AN+1)
C
C IF (IPILE.EQ.1) GO TO 34
C BWR. NO CHANNEL INTERACTION
C NSCBC = 0
C NBBC = 1
C J5 = 0
C ABETA = 0.0
C BUETA = 0.0
C GK = 0.0
C NK=0
C GO TO 120
C
C PWR. READ AND SET GAP CONNECTIONS (IE BOUNDARIES)
C IMAP=1 FOR RECTANGULAR MAP. SAY HOW MANY CHAN ACROSS AND DOWN.
C IMAP=2 FOR PWR MAP. GIVE START AND END OF EACH ROW. LAST ROW ALL 0
C IMAP=3 FOR CHANNEL-NUMBERED MAP. LAST ROW ALL 0.
C IMAP=4 FOR SPECIFYING CHANNEL BOUNDARY NUMBERS
34 NK = 0
IRAD = 0
ISIZE = 20
NEXT = 1
WRITE (13,3001) IMAP
IF (IMAP.EQ.4) GO TO 70
IF (IMAP=2) 40,42,48
40 READ (12,1001) ICROSS, IDOWN
ISTART = 1
IEND = ICROSS
GO TO 44
42 READ(12,1001) ISTART, IEND
44 J5 = 0
DO 46 J=1,ISIZE
MAAP(2,J) = 0
IF ( (J.LT.1START) .OR. (J.GT.IEND) ) GO TO 46
J5 = J5+1
MAAP(2,J) = J5
46 CONTINUE
GO TO 49
48 READ (12,1001) (MAAP(2,J),J=1,ISIZE)
C SET BOUNDARIES FOR IMAP = 1,2,3
49 JSMAX = 0
WRITE (13,3008)
DO 66 I=1,ISIZE
C SET BOUNDARIES ACROSS
DO 50 J=1,ISIZE

```

COB33560  
COB33570  
COB33580  
COB33590  
COB33600  
COB33610  
COB33620.  
COB33630  
COB33640  
COB33650  
COB33660  
COB33670  
COB33680  
COB33590  
COB33700  
COB33710  
COB33720  
COB33730  
COB33740  
COB33750  
COB33760  
COB33770  
COB33780  
COB33790  
COB33800  
COB33810  
COB33820  
COB33830  
COB33840  
COB33850  
COB33860  
COB33870  
COB33880  
COB33890  
COB33900  
COB33910  
COB33920  
COB33930  
COB33940  
COB33950  
COB33960  
COB33970  
COB33980  
COB33990  
COB34000  
COB34010  
COB34020  
COB34030  
COB34040  
COB34050  
COB34060  
COB34070  
COB34080  
COB34090  
COB34100

```

      MAAP(1,J) = MAAP(2,J)
      USMAX = MAXO(USMAX,MAAP(2,J))
      IF (MAAP(2,J).NE.0) JMAX=J
      IF (J.EQ.ISIZE) GO TO 50
      IF ( (MAAP(2,J).EQ.0) .OR. (MAAP(2,J+1).EQ.0) ) GO TO 50
      NK = NK+1
      IDAT($IK+NK) = MAAP(2,J)
      IDAT($JK+NK) = MAAP(2,J+1)
50  CONTINUE
      IF (I.GT.1) GO TO 51
      WRITE (13,3002) (MAAP(1,J),J=1,JMAX)
      JUMP = 1
      GO TO 64
51  IF (I.EQ.ISIZE) GO TO 66
      IF (IMAP-2) 52,54,60
52  IF (I.GE.IDOWN) ISTART = ISIZE+1
      GO TO 56
54  READ(12,1001) ISTART, IEND
56  DO 58 J=1,ISIZE
      MAAP(2,J) = 0
      IF ( (J.LT.ISTART) .OR. (J.GT.IEND) ) GO TO 58
      JS = JS+1
      MAAP(2,J) = JS
58  CONTINUE
      GO TO 62
60  READ(12,1001) (MAAP(2,J),J=1,ISIZE)
62  IC = NK
      SET BOUNDARIES DOWN
      DO 63 J=1,ISIZE
      IF (MAAP(2,J).NE.0) JMAX=J
      IF ( (MAAP(1,J).EQ.0) .OR. (MAAP(2,J).EQ.0) ) GO TO 63
      NK = NK+1
      IDAT($IK+NK) = MAAP(1,J)
      IDAT($JK+NK) = MAAP(2,J)
63  CONTINUE
      IF (I.C.EQ.NK) GO TO 68
      WRITE (13,3002) (MAAP(2,J),J=1,JMAX)
      SET HOLD TO PRINT MAP OF RADIAL POWERS
      JUMP = 2
64  IRAD = IRAD+1
      JB(IRAD) = JMAX
      DO 65 J=1,JMAX
      L = MAAP(JUMP,J)
      DATA($HOLD+IRAD+MG*(J-1))=-100.
      IF (L.LE.0) GO TO 65
      DATA($HOLD+IRAD+MG*(J-1))=DATA($RADIA+L)
65  CONTINUE
      IF (JUMP.EQ.1) GO TO 51
66  CONTINUE
68  CONTINUE
      PRINT RADIAL POWER MAP
      WRITE (13,3010)
      DO 69 I=1,IRAD
      JMAX = JB(I)
69  WRITE(13,3011) (DATA($HOLD+I+MG*(J-1)),J=1,JMAX)

```

```

COB34110
COB34120
COB34130
COB34140
COB34150
COB34160
COB34170
COB34180
COB34190
COB34200
COB34210
COB34220
COB34230
COB34240
COB34250
COB34260
COB34270
COB34280
COB34290
COB34300
COB34310
COB34320
COB34330
COB34340
COB34350
COB34360
COB34370
COB34380
COB34390
COB34400
COB34410
COB34420
COB34430
COB34440
COB34450
COB34460
COB34470
COB34480
COB34490
COB34500
COB34510
COB34520
COB34530
COB34540
COB34550
COB34560
COB34570
COB34580
COB34590
COB34600
COB34610
COB34620
COB34630
COB34640
COB34650

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CONVERSATIONAL MONITOR SYSTEM

FILE: CODRA3C FORTRAN A

```

IF (JSMAX, EQ, NCHANL) GO TO 76
WRITE (13,2006) JSMAX, NCHANL
ERROR = 1
RETURN

C
C SET BOUNDARIES FOR IMAP = 4
70 READ (12,1001) (JB(J), J=1,20)
DO 74 I=1,20
IF (JB(I).EQ.0) GO TO 76
IF (NEXT.EQ.0) GO TO 72
NK = NK+1
IDAT($IK+NK) = JB(I)
NEXT = 0
GO TO 74
72 IDAT($JK+NK) = JB(I)
NEXT = 1
74 CONTINUE
GO TO 70
76 DO 90 K=1, NK
I=IDAT($IK+K)
IF (IABS(I)-IABS(IDAT($JK+K))) 84,80,82
80 WRITE(13,2003) K, I, IDAT($JK+K)
IERROR = 1
RETURN
82 IDAT($IK+K) = IDAT($JK+K)
IDAT($JK+K) = I
GO TO 78
84 M = IDAT($JK+K)
DO 86 L=1,4
IF (LC(I,L).EQ.0) GO TO 88
CONTINUE
86 WRITE (13,2004) K, M, I
IERROR = 1
RETURN
88 LC(I,L) = M
NG = IDAT($NTYPE+1)
N = IGROUP(NG)
GAPS(I,L)=MAX1(GAPS(M,1),GAPS(N,1))
DIST(I,L)=DIST(N,1)
DATA($GAPN+K)=GAPS(I,L)/12.0
DATA($GAP +K)=DATA($GAPN+K)
DATA($LENGT+K)=DIST(I,L)/12.0
DATA($FACTO+K)=1.0
CONTINUE
90
C
C READ HALF-BOUNDARIES AND SET FACTOR(M) = 0.5
92 READ (12,1001) (JB(L),L=1,20)
IF (JB(1).EQ.0) GO TO 110
IEND = 100
MARK = 1
DO 98 M=1,10
L = 2*M - 1
JBL = JB(L)
IF (JBL-JB(L+1)) 98,94,96
94 IEND = M

```

COB34660  
COB34670  
COB34680  
COB34690  
COB34700  
COB34710  
COB34720  
COB34730  
COB34740  
COB34750  
COB34760  
COB34770  
COB34780  
COB34790  
COB34800  
COB34810  
COB34820  
COB34830  
COB34840  
COB34850  
COB34860  
COB34870  
COB34880  
COB34890  
COB34900  
COB34910  
COB34920  
COB34930  
COB34940  
COB34950  
COB34960  
COB34970  
COB34980  
COB34990  
COB35000  
COB35010  
COB35020  
COB35030  
COB35040  
COB35050  
COB35060  
COB35070  
COB35080  
COB35090  
COB35100  
COB35110  
COB35120  
COB35130  
COB35140  
COB35150  
COB35160  
COB35170  
COB35180  
COB35190  
COB35200

```

IF (JBL.EQ.0) GO TO 100
WRITE (13,2005) JBL,JB(L+1)
IERROR = 1
RETURN
96 JB(L) = JB(L+1)
JB(L+1) = JBL
98 CONTINUE
100 IC = MARK
DO 102 K=1,NK
IF (IDAT($JK+K).NE.JB(MARK)).OR.
1 (IDAT($JK+K).NE.JB(MARK +1))) GO TO 102
DATA($FACTO+K)=0.5
MARK = MARK+2
IF (MARK.EQ.IEND) GO TO 110
IF (MARK.GE.20) GO TO 92
102 CONTINUE
IF (IC.LT.MARK) GO TO 100
WRITE (13,2005) JB(MARK), JB(MARK+1)
IERROR = 1
RETURN
C
110 CALL ACOL(1, IDAT($IK+1), IDAT($JK+1), KMAX, IDAT($LOCA+1), MA, MS, NK,
1 MG, IPILE)
112 WRITE (13,3003) NK
M = 1
114 MM = MINO( (M+7), NK)
WRITE (13,3004) M, (IDAT($IK+K), IDAT($JK+K), K=M, MM)
M = MM+1
IF (M.LE.NK) GO TO 114
WRITE (13,3005) NK
M = 1
116 MM = MINO( (M+24), NK)
DO 118 L=1,8
118 WRITE (13,3006) L, (IDAT($LOCA+K+MG*(L-1)), K=M, MM)
M = MM+1
WRITE (13,3007)
IF (M.LE.NK) GO TO 116
L = MS+NK
WRITE (13,3009) MS, KMAX, L, MA
C
C SET NTYPE BACK TO INDICATE FRICTION TYPE
120 DO 122 I=1,NCHANL
NG=IDAT($NTYPE+I)
IDAT($VTYPE+I)=IFRIC(NG)
IF (LC(I,1).GT.0) GO TO 122
GAPS(I,1) = 0.0
DIST(I,1) = 0.0
CONTINUE
122
C
C READ FUEL DATA
IF (NODESF.EQ.0) GO TO 126
READ(12,1003) (NFUEL(I), CFUEL(I), RFUEL(I), DFUEL(I),
1 RCLAD(I), CCLAD(I), RCLAD(I), TCLAD(I), HGAP(I), I=1, NFUEL)
DO 124 I=1, NFUEL
NFUEL(I) = NFUEL(I)/3600.

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```

COB35210
COB35220
COB35230
COB35240
COB35250
COB35260
COB35270
COB35280
COB35290
COB35300
COB35310
COB35320
COB35330
COB35340
COB35350
COB35360
COB35370
COB35380
COB35390
COB35400
COB35410
COB35420
COB35430
COB35440
COB35450
COB35460
COB35470
COB35480
COB35490
COB35500
COB35510
COB35520
COB35530
COB35540
COB35550
COB35560
COB35570
COB35580
COB35590
COB35600
COB35610
COB35620
COB35630
COB35640
COB35650
COB35660
COB35670
COB35680
COB35690
COB35700
COB35710
COB35720
COB35730
COB35740
COB35750

```

```

KCLAD(1) = KCLAD(1)/3600.
DFUEL(1) = DFUEL(1)/12.
TCLAD(1) = TCLAD(1)/12.
HCAP(1) = HCAP(1)/3600.
124
C
C SET PRINT REQUIREMENTS
126 IF (J1.GT.1) RETURN
PRINT(4) = .TRUE.
PRINT(7) = .TRUE.
PRINT(8) = .TRUE.
RETURN
C
1001 FORMAT(2014)
1002 FORMAT(11,14,8E9.3,12)
1003 FORMAT(16E5.3)
1004 FORMAT(8(E5.3,15))
1005 FORMAT(20A4)
2001 FORMAT(' CARDS4 N1,GT,15')
2002 FORMAT(' CARDS4 CHANNEL GROUP',I3,' CHANNEL',I4,' INCORRECT')
2003 FORMAT(' CARDS4 GAP CONNECTION',I3,' I AND J SAME IE',I3)
2004 FORMAT(' CARDS4 GAP CONNECTION',I3,' CHANNEL',I3,
1,15 5TH ADJACENT TO',I3)
2005 FORMAT(' CARDS4 HALF-BOUNDARY',I4,' -',I4,' NOT IN BOUNDARY
1SET')
2006 FORMAT(' CARDS4 HIGHEST NUMBER CHANNEL FOUND TO BE',I3,
1, AND THIS NOT EQUAL TO NUMBER SPECIFIED, IE',I3)
3001 FORMAT(11H1,' CHANNEL DATA SET IN SUBROUTINE CARDS4 ( IMAP =',
1,12,' ),',//)
3002 FORMAT(/,2016)
3003 FORMAT(11H1,15,' BOUNDARIES AS BELOW (IK(K) - JK(K))',//)
3004 FORMAT(' ',I3,' ',8(6X,I3,' -',I3) )
3005 FORMAT(///,' LOCAL(K,B) ARRAY SET IN ACOL',5X,' K = 1 TO',I3,///)
3006 FORMAT(' ',I1,' ',2515)
3007 FORMAT(/)
3008 FORMAT(' CHANNEL NUMBERING MAP',//)
3009 FORMAT(///,' MAXIMUM OVERALL STRIPE WIDTH FOR ARRAY AAA IN DIVERCOB36110
11 =',I3,' FOR BOUNDARY NO.',I3,///,' REQUIRE',I6,' STORES COB36120
2 FOR AAA SIZE AND THIS OK SINCE LESS THAN',I6,' PROVIDED',//) COB36130
3010 FORMAT(11H1,' RADIAL POWER MAP (-100 OR *** INDICATES NO CHANN COB36140
1EL)',//)
3011 FORMAT(/,2016.3)
END
SUBROUTINE CHAN(IPART,NTHBOX,NTHBX,ND1X,ND2X)
C
C IMPLICIT INTEGER (S)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DA
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , MF ,
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KI , J ,
4 NAFAC , NARAMP , MAX , MAXL , NBBC , NCHANL , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NNODES , NODESF , NPROP ,
6 NRAMP , NR0D , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 OAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VF , VFG , VG , Z

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COB35760
COB35770
COB35780
COB35790
COB35800
COB35810
COB35820
COB35830
COB35840
COB35850
COB35860
COB35870
COB35880
COB35890
COB35900
COB35910
COB35920
COB35930
COB35940
COB35950
COB35960
COB35970
COB35980
COB35990
COB36000
COB36010
COB36020
COB36030
COB36040
COB36050
COB36060
COB36070
COB36080
COB36090
COB36100
COB36110
COB36120
COB36130
COB36140
COB36150
COB36160
COB36170
COB36180
COB36190
COB36200
COB36210
COB36220
COB36230
COB36240
COB36250
COB36260
COB36270
COB36280
COB36290
COB36300

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C
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
1  AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2  GAPAL(10), GFACT(10), GRIDXL(10), HGAP(2), HHF(30), HMG(30),
3  IGRID(10), KCLAD(2), KFUEL(2), MKF(30), NCH(10), NGAP(9),
4  PPI(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30),
5  VVF(30), VVG(30), XQUAL(30), Y(30), TT(30)
COB36310
COB36320
COB36330
COB36340
COB36350
COB36360
COB36370
COB36380
COB36390
COB36400
COB36410
COB36420
COB36430
COB36440
COB36450
COB36460
COB36470
COB36480
COB36490
COB36500
COB36510
COB36520
COB36530
COB36540
COB36550
COB36560
COB36570
COB36580
COB36590
COB36600
COB36610
COB36620
COB36630
COB36640
COB36650
COB36660
COB36670
COB36680
COB36690
COB36700
COB36710
COB36720
COB36730
COB36740
COB36750
COB36760
COB36770
COB36780
COB36790
COB36800
COB36810
COB36820
COB36830
COB36840
COB36850
COMMON /COBRA3/ MA ,MC ,MG ,MR ,MS ,MX
$$$ ,SA ,SAAA ,SAC ,SALPHA,$AN ,SANSHE,$U
1  SCCHAN,$CD ,SCHFR ,SCON ,SCOND ,SCP ,SD ,SDC ,SDFDX
2  $DHDX ,SDHYD ,SDHYDN ,SDIST ,SDPX ,SDPK ,SDUR ,SDR ,SF
3  $FACTO,$FDIV ,FINLE,$FLUX ,$FMULT,$FOLD,$FSP ,$FSPLI,$FKFLO,
4  $GAP ,$GAPN ,$GAPS ,SH ,SHFILM,$HINLE,$HOLD,$HPERI,$IDARE,
5  $IDFUE,$IDGAP,$IK ,SUBOIL,$UK ,SLC ,$LENGT,$LOCA,$LR
6  $MCHFR,$MCFRC,$MCFRR,$MTYPE,$MWRAP,$MWRPS,$P
7  $PHI ,$PRITC,$PRNTR,$PRNTN,$PW ,$PMRF ,SOC ,$QF ,$QPRIM,
8  $QUAL,$RADIA,$RHO ,$RHOO,$SP ,ST ,STDUNY,$TINLE,$TROD,
9  $U ,$UHV ,$USAVE,$USTAR,$SV ,SVISC ,SVISCH,$VPA
A $W ,$WOLD,$WP ,$WSAVE,$X ,$XCROS,$XA ,$$B ,$XPOLD
COB36560
COB36570
COB36580
COB36590
COB36600
COB36610
COB36620
COB36630
COB36640
COB36650
COB36660
COB36670
COB36680
COB36690
COB36700
COB36710
COB36720
COB36730
COB36740
COB36750
COB36760
COB36770
COB36780
COB36790
COB36800
COB36810
COB36820
COB36830
COB36840
COB36850
COMMON /FRDATA/ BURN,CPR,EFFB,EPF,EXPR,PRESS,PUO2,FRAC,FTD,
1  GMIX(4),GRGH,PGAS,RADR,$DELTE,THC,THG
C
COMMON /LINK4/IFRM,ITHM,IPROP,NCC,NCF,NDM1,NDS,NGP
C
COMMON /ITPSV/ITMP
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
COMMON /GAPAC/ FACSL(70), FACSLK(70)
C
COMMON /LINK2/ CROSS(6),DATE(2),FG(30),FH(30),FP(30),FQ(30),JM(9),
1  JN(9),OUTPUT(10),PRINT(12),TEAT(17),TIME(3),YI(30),YH(30),YP(30),
2  YO(30)
COMMON /LINK3/ DXX,ETIME,GIN,HIN,IG,IN,ISAVE,JUMP,KASE,KY,MAAT,
1  NDT,NDAP1,NFUEL,NG,NH,NMUMP,NOUT,NP,NPCHAN,NPNODE,NPROD,NQ,NR,
2  NSKIPT,NSKIPLA,NTRIES,PEXIT,PHOTOT,SAVEDT,TIN,TTIME,$Z
DIMENSION CARD(20),CDG(5),GP(250),JBSTOR(150),JB(20),
1  NTHBOX(25,25),GAPREC(400)
C
IPART = 1 READ CHANNEL INPUT DATA
IPART = 2 PRINT CHANNEL INPUT DATA
OWN-ARRAY MAX SIZES, CARD(20), CDG(NGRIDT), GP(INCHANL), JB(MAARD),
C
JBSTOR(INCTYP+3+NUMBER OF CHANNELS NOT OF TYPE 1)
C
DEFINE JBSTOR(L), L=1,NCTYP+1 = ARRAY POSITIONS STARTING EACH TYPE,
C
JBSTOR(INCTYP+2) = A CHANNEL NUMBER OF TYPE 1,

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CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

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C   CHN OF TYPE N IN JBSTOR(L),L=J,K WHERE J=JBSTOR(N),K=JBSTOR(N+1)-1 COB36860
C   IF (IPART.EQ.2) GO TO 102 COB36870
C   MAXRD = 14 COB36880
C   NFUELT = 1 COB36890
C   NCHANL = NTHBXX COB36900
C   ITEMP=0 COB36910
C   READ (12,1001) CARD,IPILE,NCTYP,NGRID,NGRIDT,NODESF,NF,XF,IFRM, COB36920
C   WRITE (13,1002) CARD COB36930
C   1 IHTM,IPROP COB36940
C   WRITE (13,1002) CARD COB36950
C   IF (NODESF.EQ.0) GO TO 2 COB36960
C   IF (IFRM.EQ.0.AND.IHTM.EQ.0) GO TO 2 COB36970
C   READ(12,2016) CARD,EPF COB36980
C   WRITE(13,2009) CARD COB36990
C   CC IF EPF=0, THEN SET TO DEFAULT VALUE COB37000
C   IF (EPF.EQ.0.) EPF=0.01 COB37010
C   2 HRDD = NCHANL COB37020
C   J6 = 2 COB37030
C   HRAMP = 1 COB37040
C   GRID = .FALSE. COB37050
C   NGRT = MAXO(NGRIDT,1) COB37060
C   J7 = IPILE COB37070
C   DO 1109 I=1,MC COB37080
C   DO 1109 J=1,MR COB37090
C   DATA($PARF+I+MC*(J-1))=0.0 COB37100
C   DO 4 I=1,MR COB37110
C   DO 4 L=1,6 COB37120
C   IDAT($LR+I*MR*(L-1))=0 COB37130
C   DATA($PHI+I*MR*(L-1))=0.0 COB37140
C   4 CONTINUE COB37150
C   READ AND SET CHANNEL DATA. (A) CHANNEL PARAMETERS, (B) GRID DATA, COB37160
C   (C) CHANNELS MAKING EACH TYPE (EXCEPT TYPE 1) COB37170
C   JBIC = NCTYP+2 COB37180
C   JBSTOR(1) = JBIC COB37190
C   JBSTOR(2) = JBIC+1 COB37200
C   DO 20 I=1,NCTYP COB37210
C   READ(12,1003)CARD,N,J,FRAC,GAP,WV,HRNUM,HRDI,CRNUM,CRDI,SIDE,CORN COB37220
C   WRITE (13,1004) I,CARD COB37230
C   IF (FRAC.LE.0.0) FRAC=1.0 COB37240
C   IF (J.EQ.2) GO TO 6 COB37250
C   CHAR=CRNUM COB37260
C   CHPW=CRDI COB37270
C   CHPH=SIDE COB37280
C   GO TO 8 COB37290
C   6 CHAR = SIDE+SIDE - 4.0*CORN+CORN - PI*(0.25*HRNUM*HRDI+HRDI COB37300
C   1 + 0.25*CRNUM*CRDI+CRDI - CORN*CORN) COB37310
C   CHPW=HRNUM*PI+HRDI COB37320
C   CHPW=CHPW+4.0*(SIDE-2.0*CORN)+2.0*PI*CORN+CRNUM*PI+CRDI COB37330
C   8 CHDI = 4.0*CHAR/CHPW COB37340
C   CDG(1)=0.0 COB37350
C   IF (NGRID.LE.0) GO TO 9 COB37360
C   READ(12,1005) CARD,(CDG(L),L=1,NGRIDT) COB37370
C   WRITE(13,1006) I,CARD COB37380
C   9 M=1 COB37390
C   IF(I.EQ.1) GO TO 12 COB37400

```

```

IFIRST=1
10 READ(12,1001) CARD,(JB(L),L=1,MAXRD)
IF(IFIRST.EQ.0) WRITE(13,1008) CARD
IF (IFIRST.EQ.0) GO TO 12
WRITE(13,1007) I,CARD
IFIRST=0
M=JB(1)
IF((M.GT.0).AND.(M.LE.NCHANL)) GO TO 12
ERROR=1
WRITE(13,2001) I,M
RETURN
12 DATA(SA+M) = CHAR*FRAC/144.0
DATA(SPERIM+M) = CHPW*FRAC/12.0
DATA(SIPERI+M) = CHPH*FRAC/12.0
DATA($PHI+M) = HRNUM*FRAC
DATA($DHYD+M) = CHDI/12.0
DATA($D+M) = HRDI/12.0
IDAT($TYPE+M) = MAX0(N,1)
GP(M) = GAPWV
DO 18 L=1,NCHANL
J=L
IF(L.EQ.1) GO TO 14
IF(L.GT.MAXRD) GO TO 10
J=JB(L)
IF(J.LE.0) GO TO 20
JBIC=JBIC+1
JBSTOR(JBIC)=J
JBSTOR(I+1) = DATA(SA+M)
DATA($AN+J) = DATA(SA+M)
DATA(SPERIM+J) = DATA(SPERIM+M)
DATA(SIPERI+J) = DATA(SIPERI+M)
DATA($DHYD+J) = DATA($DHYD+M)
DATA($DR+J) = DATA($D+M)*12.
DATA($D+J) = DATA($D+M)
GP(J) = GP(M)
IDAT($TYPE+J) = IDAT($TYPE+M)
IDAT($IDFUE+J) = 1
IF(DATA($RADIA+J).EQ.0.0) GO TO 17
DATA($PHI+J) = DATA($PHI+M)
DATA($PHI+J+NR*(1-1))=DATA($PHI+M)
DATA($PWR+J+MC*(J-1)) = DATA($PHI+M)
IDAT($LR+J) = J
IDAT($LR+J+NR*(1-1)) = J
17 CONTINUE
DO 16 K=1,NGRT
16 DATA(SCD+J+MC*(K-1)) = CDG(K)
18 CONTINUE
20 CONTINUE
C
SET CHANNEL OF TYPE 1 INTO JBSTOR
L = JBSTOR(2)
M = JBSTOR(NCTYP+1) - 1
DO 26 I=1,NCHANL
DO 24 J=L,M

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COB37410
COB37420
COB37430
COB37440
COB37450
COB37460
COB37470
COB37480
COB37490
COB37500
COB37510
COB37520
COB37530
COB37540
COB37550
COB37560
COB37570
COB37580
COB37590
COB37600
COB37610
COB37620
COB37630
COB37640
COB37650
COB37660
COB37670
COB37680
COB37690
COB37700
COB37710
COB37720
COB37730
COB37740
COB37750
COB37760
COB37770
COB37780
COB37790
COB37800
COB37810
COB37820
COB37830
COB37840
COB37850
COB37860
COB37870
COB37880
COB37890
COB37900
COB37910
COB37920
COB37930
COB37940
COB37950

```

```

IF (JBSTOR(J).EQ.1) GO TO 26
24 CONTINUE
JBSTOR(NCTYP+2) = 1
GO TO 28
26 CONTINUE
C
28 IF (NGRID.EQ.0) GO TO 30
READ GRID POSITIONS
READ (12,1009) CARD,(GRIDXL(I),IGRID(I),I=1,7)
WRITE (13,1010) CARD
IF (NGRID.LE.10) GO TO 29
WRITE (13,2007) NGRID
STOP
29 IF (NGRID.LE.7) GO TO 30
READ (12,1009) CARD,(GRIDXL(I),IGRID(I),I=8,NGRID)
WRITE (13,1010) CARD
READ ROD LAYOUT
C
30 IF (IPILE) 2031,2031,2032
2031 READ(12,2033) CARD,NN11,NN22,NN33,NN44,ITMP
2033 FORMAT(20A4,11,515)
WRITE(13,2034) CARD
2034 FORMAT(' INDICATORS ',14X,'***',20A4,'*** CHAN')
IF (IFRM.EQ.1.AND.NN44.NE.1) GO TO 146
NR0D=NR22
IF (NR11.EQ.0) GO TO 2102
DO 2181 J=1,NR11
READ (12,2035) CARD,N,I,DATA(SDR+I),DATA(SRADIA+I),(IDAT(SLR+I)+
1MR*(L-1)),DATA($PHI+I+MR*(L-1)),L=1,6)
2035 FORMAT(20A4,11,1,14,2E5.0,6(13,E7.0))
WRITE (13,2047) CARD
2047 FORMAT(' ROD DATA',20X,'***',20A4,'*** CHAN')
IDAT(SIDFUE+I)=N
IF(N.LT.1) IDAT(SIDFUE+I)=1
2181 CONTINUE
2182 DO 2185 I=1,NR0D
DO 2183 L=1,6
IF(1DA1($L+I+NR*(L-1))) 2184,2184,2183
2183 K=IDAT(SLR+I+MR*(L-1))
DATA ($P*RF+K+MC*(I-1))=DATA($PHI+I+MR*(L-1))
2184 CONTINUE
2185 DATA($D+I)=DATA(SDR+I)/12.
IF (J1.LE.1) PRINT(0)=.TRUE.
NODESF=NN33
MFUELT=NN14
2032 IF(NODESF.EQ.0) GO TO 34
READ FUEL THERMAL DATA
READ(12,1005) CARD, (KFUEL(I),CFUEL(I),RFUEL(I),DFUEL(I),
1 KCLAD(I),CCLAD(I),TCLAD(I),HGAP(I),I=1,NFUELT)
WRITE (13,1011) CARD
IF (IFRM.EQ.0) GO TO 31
READ(12,1003)CARD,NCF,NCC,THG
WRITE(13,2010)CARD
THG=THG/12.
IF ((NCF+NCC+1).NE.NODESF) GO TO 146
IF(NODESF.GT.21) GO TO 146
C

```

- COB37960
- COB37970
- COB37980
- COB37990
- COB38000
- COB38010
- COB38020
- COB38030
- COB38040
- COB38050
- COB38060
- COB38070
- COB38080
- COB38090
- COB38100
- COB38110
- COB38120
- COB38130
- COB38140
- COB38150
- COB38160
- COB38170
- COB38180
- COB38190
- COB38200
- COB38210
- COB38220
- COB38230
- COB38240
- COB38250
- COB38260
- COB38270
- COB38280
- COB38290
- COB38300
- COB38310
- COB38320
- COB38330
- COB38340
- COB38350
- COB38360
- COB38370
- COB38380
- COB38390
- COB38400
- COB38410
- COB38420
- COB38430
- COB38440
- COB38450
- COB38460
- COB38470
- COB38480
- COB38490
- COB38500

```

IF (IPROP.EQ.0) GO TO 31
READ(12,1005)CARD,FTD,FPUD2
WRITE(13,2012)CARD
IF(IPROP.LE.1) GO TO 31
READ(12,1005)CARD,BURN,CPR,EXPR,FPRESS,GRGH,GMIX,PGAS
WRITE(13,2014)CARD
IF((GMIX(1)+GMIX(2)+GMIX(3)+GMIX(4)).GT.1.01) GO TO 146
GRGH=GRGH/12.
31 DO 32 I=1,NFUFLT
KFUEL(I) = KFUEL(I)/3600.
KCLAD(I) = KCLAD(I)/3600.
DFUEL(I) = DFUEL(I)/12.
TCLAD(I) = TCLAD(I)/12.
32 HGAP(I) = HGAP(I)/3600.
C
C SET WHOLE-CHANNEL AREA AND PH
34 ATOTAL = 0.0
PHTOT = 0.0
DO 36 I=1,NCHANL
ATOTAL = ATOTAL + DATA($A+I)
PHTOT = PHTOT + DATA($PERI+I)
NK = 0
IF (IPILE.EQ.2) GO TO 99
C
C SET GAP BOUNDARY NUMBERING SYSTEM (PWR ONLY)
IF(PTLE.GT.0) GO TO 3010
DO 242 ND2=1,ND2X
DO 238 ND1=2,ND1X
I=NIHBOX(ND1-1,ND2)
J=NIHBOX(ND1,ND2)
IF((I.LE.0).OR.(J.LE.0)) GO TO 238
IF((I-J).EQ.0) GO TO 238
DO 5216 K=1,NK
IF((I.EQ.IDAT($IK+K)).OR.(I.EQ.IDAT($JK+K))) GO TO 5215
GO TO 5216
5215 IF((J.EQ.IDAT($JK+K)).OR.(J.EQ.IDAT($IK+K))) GO TO 238
5216 CONTINUE
NK=NK+1
IDAT($IK+NK) = I
IDAT($JK+NK) = J
238 CONTINUE
DO 240 ND1=1,ND1X
J=NIHBOX(ND1,ND2)
I=NIHBOX(ND1,ND2+1)
IF((I.LE.0).OR.(J.LE.0)) GO TO 240
IF((I-J).EQ.0) GO TO 240
DO 6216 K=1,NK
IF((I.EQ.IDAT($IK+K)).OR.(I.EQ.IDAT($JK+K))) GO TO 6215
GO TO 6216
6215 IF((J.EQ.IDAT($JK+K)).OR.(J.EQ.IDAT($IK+K))) GO TO 240
6216 CONTINUE
NK=NK+1
IDAT($IK+NK) = I
IDAT($JK+NK) = J

```

COB38510  
COB38520  
COB38530  
COB38540  
COB38550  
COB38560  
COB38570  
COB38580  
COB38590  
COB38600  
COB38610  
COB38620  
COB38630  
COB38640  
COB38650  
COB38660  
COB38670  
COB38680  
COB38690  
COB38700  
COB38710  
COB38720  
COB38730  
COB38740  
COB38750  
COB38760  
COB38770  
COB38780  
COB38790  
COB38800  
COB38810  
COB38820  
COB38830  
COB38840  
COB38850  
COB38860  
COB38870  
COB38880  
COB38890  
COB38900  
COB38910  
COB38920  
COB38930  
COB38940  
COB38950  
COB38960  
COB38970  
COB38980  
COB38990  
COB39000  
COB39010  
COB39020  
COB39030  
COB39040  
COB39050



```

240 CONTINUE
242 CONTINUE
GO TO 3020
3010 DO 42 ND2=1,ND2X
DO 36 ND1=2,ND1X
I=NTHBOX(ND1-1,ND2)
J=HTHBOX(ND1,ND2)
IF((I.LE.0).OR.(J.LE.0)) GO TO 38
NK=MK+1
IDAT($IK+NK) = I
IDAT($JK+NK) = J
38 CONTINUE
IF(ND2.EQ.ND2X) GO TO 42
DO 40 ND1=1,ND1X
J=NTHBOX(ND1,ND2)
I=NTHBOX(ND1,ND2+1)
IF((I.LE.0).OR.(J.LE.0)) GO TO 40
NK=MK+1
IDAT($IK+NK) = I
IDAT($JK+NK) = J
40 CONTINUE
42 CONTINUE
C
C SET GAP BOUNDARY PARAMETERS
3020 IF(PIPLE.GT.0) GO TO 9006
M=1
9014 MM=MINO((M+13),NK)
9007 READ (12,9007) CARD,(GAPREC(1),I=M,MM)
WRITE(13,9107) CARD
9107 FORMAT(' GAP INTERCONNECTIONS',8X,'****',20A4,'**** CHAN')
M=MM+1
IF(M.LE.NK) GO TO 9014
IF (ITMP.EQ.0) GO TO 9076
IF (NK.LE.70) GO TO 9012
WRITE(13,9010)
9010 FORMAT(1H,' ERROR DETECTED IN CHAN - TRANSVERSE ',
1 'COUPLING PARAMETER ARRAYS NOT LARGE ENOUGH FOR GREATER THAN',
2 '/, ' 70 GAP INTERCONNECTIONS.')
GO TO 146
C
C READ TRANSVERSE MOMENTUM COUPLING PARAMETERS
9012 M=1
9020 MM=MINO((M+6),NK)
READ(12,9007) CARD,(FACSLK(1),I=M,MM)
WRITE(13,9025) CARD
9025 FORMAT(' GAP FACTOR PAIRS',12X,'****',20A4,'**** CHAN')
M=MM+1
IF(M.LE.NK) GO TO 9020
9076 DO 9008 K=1,NK
9078 I=IDAT($IK+K)
IF (I-IDAT($JK+K)) 9084,9080,9082
9080 WRITE(13,2003) K,I,IDAT($JK+K)
ERROR=1
RETURN

```

```

COB39060
COB39070
COB39080
COB39090
COB39100
COB39110
COB39120
COB39130
COB39140
COB39150
COB39160
COB39170
COB39180
COB39190
COB39200
COB39210
COB39220
COB39230
COB39240
COB39250
COB39260
COB39270
COB39280
COB39290
COB39300
COB39310
COB39320
COB39330
COB39340
COB39350
COB39360
COB39370
COB39380
COB39390
COB39400
COB39410
COB39420
COB39430
COB39440
COB39450
COB39460
COB39470
COB39480
COB39490
COB39500
COB39510
COB39520
COB39530
COB39540
COB39550
COB39560
COB39570
COB39580
COB39590
COB39600

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```

9082 IDAT($IK+K)=IDAT($JK+K)
IDAT($JK+K)=I
GO TO 9078
9084 M=IDAT($JK+K)
DATA($GAPN+K)=GAPREC(K)/12.
DATA($GAP+K)=DATA($GAPN+K)
DATA($LENGT+K)=0.0
DATA($FACTO+K)=1.0
9008 CONTINUE
GO TO 9009
9006 DO 90 K=1,NK
78 I = IDAT($IK+K)
IF (I-IDAT($JK+K)) 84,80,82
80 WRITE (13,2003) K,I,IDAT($JK+K)
IERROR = 1
RETURN
82 IDAT($IK+K) = IDAT($JK+K)
IDAT($JK+K) = I
GO TO 78
84 M = IDAT($JK+K)
DATA($GAPN+K) = 0.5*(GP(I)+GP(M))/12.0
DATA($GAP+K) = DATA($GAPN+K)
DATA($LENGT+K) = 0.0
DATA($FACTO+K) = 1.0
90 CONTINUE
9009 CONTINUE
C
C SET LOCA ARRAY
C DYNAMIC STORAGE CALL TO CORE2 FROM ACOL TO SET MA, MS IF GAPS.
C CALL ACOL(1, IDAT($IK+1), IDAT($JK+1), NMAX, IDAT($LOCA+1), MA, MS, NK,
1MG, IPILE)
C
C IF (IPILE.EQ.0) GO TO 99
C READ HALF-BOUNDARIES AND SET FACTOR(K)=0.5
NMAX=NAXRD/2
92 READ(12,1001) CARD, (JB(L),L=1,MAXRD)
MM = 0
DO 98 M=1,MMAX
MM = MM+1
L=2*MM-1
IF(JB(L).LE.0) GO TO 99
I=MINO(JB(L),JB(L+1))
J=MAXO(JB(L),JB(L+1))
DO 94 K=1,NK
IF ( (I.EQ.IDAT($IK+K)) .AND. (J.EQ.IDAT($JK+K)) ) GO TO 96
94 CONTINUE
IERROR=1
WRITE(13,2005) MM,I,J
RETURN
96 DATA($FACTO+K) = 0.5
98 CONTINUE
GO TO 92
C
C READ FORCED FLOW BOUNDARIES HERE IF PROGRAMMED LATER

```

COB39610  
COB39620  
COB39630  
COB39640  
COB39650  
COB39660  
COB39670  
COB39680  
COB39690  
COB39700  
COB39710  
COB39720  
COB39730  
COB39740  
COB39750  
COB39760  
COB39770  
COB39780  
COB39790  
COB39800  
COB39810  
COB39820  
COB39830  
COB39840  
COB39850  
COB39860  
COB39870  
COB39880  
COB39890  
COB39900  
COB39910  
COB39920  
COB39930  
COB39940  
COB39950  
COB39960  
COB39970  
COB39980  
COB39990  
COB40000  
COB40010  
COB40020  
COB40030  
COB40040  
COB40050  
COB40060  
COB40070  
COB40080  
COB40090  
COB40100  
COB40110  
COB40120  
COB40130  
COB40140  
COB40150

```

99 DO 100 K=1,NK
   DO 100 L=1,5
  100 DATA(SXFLO+K*MG-(L-1)) = 0.0
      IF (NFXF.EQ.0) GO TO 101
      WRITE (13,1013)
      IERROR = 1
  101 CONTINUE
      RETURN
C
C      IPART = 2.      PRINT CHANNEL DATA
C  102 IPILE=J7
      WRITE(13,1040) IPILE,NCHANL,NCTYP,NGRID,NGRIDT,MODESF,NFXF
      IF(NODESF.GT.0) WRITE(13,1045) IFRM,INTM,IPROP
      WRITE(13,1050)
C
C      DRAW MAP OF CHANNELS AND CHECK TOTAL
      NUMCH=0
C  106 DO 106 ND2=1,ND2X
      IMAX=0
      DO 104 ND1=1,ND1X
        NUMCH=MAX0(NUMCH,NTHBOX(ND1,ND2))
        IF(NTHBOX(ND1,ND2).GT.0) IMAX=ND1
C  104 CONTINUE
      IF(IMAX.EQ.0) GO TO 108
      WRITE(13,1052) (NTHBOX(I,ND2),I=1,IMAX)
C  106 CONTINUE
C  108 IF(NUMCH.EQ.NCHANL) GO TO 110
      IERROR=1
      WRITE(13,2006) NUMCH,NCHANL
      RETURN
C
C      PRINT CHANNEL NUMBER IN EACH TYPE
C  110 IF (NCTYP.EQ.1) GO TO 115
      WRITE (13,1053)
      DO 114 J=2,NCTYP
        L=JBSTOR(J)
        M=JBSTOR(I+1) - 1
        WRITE(13,1054) I,(JBSTOR(K),K=L,M)
C  114 CONTINUE
C
C      PRINT CHANNEL DATA FOR EACH TYPE
C  115 WRITE(13,1055)
      DO 116 I=1,NCTYP
        L=JBSTOR(I)
        J=JBSTOR(L)
        DROD = DATA(D0+J)+12.0
C  116 WRITE(13,1056) I,IDAT($NTYPE+J),DATA($A+J),DATA($PERIM+J),
      1 DATA($PER1+J), DATA($PHI+J), DROD, GP(J)
C
C      PRINT GRID DATA
      IF(NGRID.GT.0) GO TO 118
      WRITE(13,1057)
      GO TO 124
C  118 WRITE(13,1058) NGRID,NGRIDT,(IGRID(I),GRIDAL(I),I=1,NGRID)
      WRITE(13,1059) NGRID

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```

COB40160
COB40170
COB40180
COB40190
COB40200
COB40210
COB40220
COB40230
COB40240
COB40250
COB40260
COB40270
COB40280
COB40290
COB40300
COB40310
COB40320
COB40330
COB40340
COB40350
COB40360
COB40370
COB40380
COB40390
COB40400
COB40410
COB40420
COB40430
COB40440
COB40450
COB40460
COB40470
COB40480
COB40490
COB40500
COB40510
COB40520
COB40530
COB40540
COB40550
COB40560
COB40570
COB40580
COB40590
COB40600
COB40610
COB40620
COB40630
COB40640
COB40650
COB40660
COB40670
COB40680
COB40690
COB40700

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```

ITMAX = 1
IF (NEXF.GT.0) ITMAX = 2
DO 122 ITR=1,ITMAX
DO 120 I=1,NCTYP
L=UBSTOR(I)
J=UBSTOR(L)
IF(ITR.EQ.1) WRITE(13,1060) I,(DATA(SCD+J*MG*(K-1)),K=1,NGRIDT)
IF(ITR.EQ.2) WRITE(13,1060) I,(DATA(SFXFLO+J*MG*(K-1)),K=1,
1 NGRIDT)
120 CONTINUE
IF(ITR.LT.ITMAX) WRITE(13,1061) NGRIDT
122 CONTINUE
C
124 IF(IPILE.GT.0) GO TO 125
WRITE(13,2008) (I,1DAT($OFUE+1),DATA(SDR+1),DATA($RADIA+1),
1(DATA($PHI+1+HR*(L-1)),IDAT($LR+1+HR*(L-1)),L=1,6),I=1,NMOD)
PRINT FUEL THERMAL DATA
125 IF(INODSF.EQ.0) GO TO 130
WRITE (13,1062) NODESF
DO 126 J=1,NFUEL
WV1 = KFUEL(J)*3600.0
WV2 = DFUEL(J)*12.0
WV3 = KCLAD(J)*3600.0
WV4 = TCLAD(J)*12.0
WV5 = HGAP(J)*3600.0
126 WRITE(13,1063) J,WV1,CFUEL(J),RFUEL(J),WV2,WV3,CCLAD(J),RCLAD(J),
1 WV4,WV5
WV6=THG*12.
IF(IFRM.EQ.1) WRITE(13,1080) NCF,NCC,WV6
IF(IPROP.GE.1) WRITE(13,1082) FTO,FPUD2
IF(IPROP.EQ.2) WRITE(13,1084) BURN,CPR,EXPR,FPRESS,GRGH,GHIX,PGAS
IF(IHTM.EQ.1) WRITE(13,1090)
IF(IHTM.EQ.2) WRITE(13,1092)
C
130 IF (IPILE.EQ.2) GO TO 144
IF(IPILE.EQ.0) GO TO 132
DO 131 K=1,NK
131 GAPREC(K)=DATA($GAPN+K)+12.0
PRINT ARRAYS IK, JK AND LOCA
132 WRITE (13,1064)
WRITE (13,1065) NK
M = 1
134 N=M*MINO((M+5),NK)
WRITE (13,1066) M,(1DAT($IK+K),IDAT($JK+K),GAPREC(K),K=M,MM)
M = MM+1
IF (M.LE.NK) GO TO 134
WRITE (13,1067) NK
M = 1
136 NN = MINO((M+24),NK)
IF (IPILE.GT.0) GO TO 4207
DO 8138 L=1,14
8138 WRITE(13,1068) L,(1DAT($LOCA+K+MG*(L-1)),K=M,MM)
GO TO 4208
4207 DO 138 L=1,8
138 WRITE (13,1068) L, (1DAT($LOCA+K+MG*(L-1)),K=M,MM)

```

```

COB40710
COB40720
COB40730
COB40740
COB40750
COB40760
COB40770
COB40780
COB40790
COB40800
COB40810
COB40820
COB40830
COB40840
COB40850
COB40860
COB40870
COB40880
COB40890
COB40900
COB40910
COB40920
COB40930
COB40940
COB40950
COB40960
COB40970
COB40980
COB40990
COB41000
COB41010
COB41020
COB41030
COB41040
COB41050
COB41060
COB41070
COB41080
COB41090
COB41100
COB41110
COB41120
COB41130
COB41140
COB41150
COB41160
COB41170
COB41180
COB41190
COB41200
COB41210
COB41220
COB41230
COB41240
COB41250

```

```

4208 M=MM*1
WRITE (13,1069)
IF (M.LE.NK) GO TO 138
L = MS*NK
WRITE (13,1070) MS,MMAX,L,MA
IF (1TMP.EQ.0) GO TO 139

C PRINT TRANSVERSE MOMENTUM COUPLING PARAMETERS
WRITE(13,1076)
WRITE(13,1078) (K,FACSL(K),FACSLK(K),K=1,NK)

C
C PRINT HALF-BOUNDARIES
139 IC = 0
DO 140 K=1,NK
IF (DATA($FACTO+K).EQ.1.0) GO TO 140
IC = IC+1
JBSTOR(IC) = K
140 CONTINUE
IF (IC.GT.1) GO TO 142
WRITE (13,1072)
GO TO 144
142 WRITE (13,1073) (JBSTOR(K),K=1,IC)
144 CONTINUE
WRITE (13,1074)
RETURN

C
146 WRITE (13,1000)
ERROR=1
RETURN

C
1000 FORMAT (1H, ' INPUT ERROR DETECTED BY CHAN. ')
1001 FORMAT (20A4, T1, 1415)
1002 FORMAT (' INDICATORS', 18X, '...', 20A4, '...', 'CHAN')
1003 FORMAT (20A4, T1, 215, 8E5.0)
1004 FORMAT (' CHANNEL DATA, TYPE', 13, 7X, '...', 20A4, '...', 'CHAN')
1005 FORMAT (20A4, T1, 14E5.0)
1006 FORMAT (' GRID DATA, TYPE', 13, 10X, '...', 20A4, '...', 'CHAN')
1007 FORMAT (' CHANNELS OF TYPE', 13, 9X, '...', 20A4, '...', 'CHAN')
1008 FORMAT (30X, '...', 20A4, '...', 'CHAN')
1009 FORMAT (20A4, T1, 7(E5.0, 15))
1010 FORMAT (' GRID POSITIONS', 14X, '...', 20A4, '...', 'CHAN')
1011 FORMAT (' FUEL THERMAL DATA', 11X, '...', 20A4, '...', 'CHAN')
1012 FORMAT (' HALF-BOUNDARY CHANNEL PAIRS', 1X, '...', 20A4, '...', 'CHAN')
1013 FORMAT (' FORCED FLOW NOT PROGRAMMED. STOP CALCULATION IN CHAN')
1040 FORMAT (///, 43X, 'CHANNEL, ROD AND GRID DATA', /, 43X,
1 //, ' REACTOR TYPE', 8X,
2 ' ', 13, 5X, '(1=PWR, 2=BWR)', /, ' NO. FUEL ASSEMBLIES', 13, COB41720
3 /, ' NO. ASSEMBLY TYPES', 13, /, ' NO. GRIDS', 11X, COB41730
4 13, /, ' NO. GRID TYPES', 6X, ' ', 13, /, ' NO. FUEL MODES', 6X, COB41740
5 ' ', 13, /, ' NO. FCO FLOW TYPES', 2X, ' ', 13, /)
1045 FORMAT (1H, ' FUEL ROD MODEL IND.', 13, /,
1 ' HEAT TRANSFER MODEL IND.', 13, /,
2 ' FUEL ROD PROP. IND.', 13, /)
1050 FORMAT (///, ' CHANNEL DATA', /, ' -----', 15X,
1 ' CHANNEL NUMBERING MAP', /)

```

COB41260

COB41270

COB41280

COB41290

COB41300

COB41310

COB41320

COB41330

COB41340

COB41350

COB41360

COB41370

COB41380

COB41390

COB41400

COB41410

COB41420

COB41430

COB41440

COB41450

COB41460

COB41470

COB41480

COB41490

COB41500

COB41510

COB41520

COB41530

COB41540

COB41550

COB41560

COB41570

COB41580

COB41590

COB41600

COB41610

COB41620

COB41630

COB41640

COB41650

COB41660

COB41670

COB41680

COB41690

COB41700

COB41710

COB41720

COB41730

COB41740

COB41750

COB41760

COB41770

COB41780

COB41790

COB41800



```

CC
1084 FORMAT(/,/, ' GAP HEAT TRANSFER COEFFICIENTS WILL BE ',
1 ' CALCULATED USING FUEL ROD TEMPERATURES. ',/,
2 ' BURNUP (MWD/MTU) =', E12.5,/,
3 ' COEFF. OF FUEL PRESSURE =', E12.5,/,
4 ' EXPONENT OF FUEL PRESSURE =', E12.5,/,
5 ' FUEL PRESSURE =', E12.5,/,
6 ' GAP ROUGHNESS, RMS (FT) =', E12.5,/,
7 ' HELIUM FRACTION =', E12.5,/,
8 ' ARGON FRACTION =', E12.5,/,
9 ' KRYPTON FRACTION =', E12.5,/,
1 ' KENDON FRACTION =', E12.5,/,
2 ' GAP GAS PRESSURE (PSIA) =', E12.5)

CC
1090 FORMAT(/,/, ' ROD-TO-COOLANT HEAT TRANSFER USING NEW MODEL FOR ',
1 ' PRE-CHF CONDITIONS')
1092 FORMAT(/,/, ' ROD-TO-COOLANT HEAT TRANSFER USING NEW MODEL FOR ',
1 ' PRE- AND POST-CHF CONDITIONS')

C
2001 FORMAT(' INPUT DATA ERROR IN ITHO. FIRST CHANNEL OF TYPE', I3,
1 ' IS', I3)
2003 FORMAT(' ITHO GAP CONNECTION ', I3, ' I AND J SAME IE ', I3)
2005 FORMAT(' ITHO HALF-BOUNDARY ', I4, ' - ', I4, ' NOT IN BOUNDARY $COB42500
1E')
2006 FORMAT(' ITHO HIGHEST NUMBER CHANNEL FOUND TO BE ', I3,
1 ' AND THIS NOT EQUAL TO NUMBER SPECIFIED, IE ', I3)
2007 FORMAT(' GRID GIVEN AS ', I3, ' THIS TOO LARGE AS MAX ALLOWED)COB42620
1E')
2008 FORMAT(/,/, ' ROD INPUT DATA',/, ' --- -----',/,
1 ' ROD TYPE DIA RADIAL POWER FRACTION OF POWER TO ADJA',/,
2 ' CENT CHANNELS (ADJ. CHANNEL NO.)',/, ' NO. NO. (IN) ',/,
3 ' FACTOR',/(2I5,F8.4,F9.4,F11.4,1H(12,1H)F9.4,1H(12,1H)F9.4,
4 1H(12,1H)F9.4,1H(12,1H)F9.4,1H(12,1H)F9.4,1H(12,1H))

C
2009 FORMAT(1H,1X,'EPSF',24X,'...',20A4,'... CHAN')
2010 FORMAT(1H,1X,'NCF,NCC,THG,15X,'...',20A4,'... CHAN')
2012 FORMAT(1H,1X,'FTD,FPUO2',19X,'...',20A4,'... CHAN')
2014 FORMAT(1H,1X,'GAP DATA',20X,'...',20A4,'... CHAN')
2016 FORMAT(20A4, T1, EB.0)

C
END
SUBROUTINE PRINTM(IN)
RETURN
END

SUBROUTINE CHF(JSTART,JEND)
C CHF SEARCHES COBRA-IIIC OUTPUT AT THE END OF EACH TIME STEP FOR
C THE OCCURRENCE OF CRITICAL HEAT FLUX. THE SEARCH IS MADE ON EACH ROD
C AT A SPECIFIED AXIAL LOCATION RANGE BY CONSIDERING EACH ROD AND THE
C ADJACENT CHANNELS.
C ALTHOUGH THE BAN-2 AND W-3 CORRELATIONS ARE INCLUDED, USERS SHOULD
C PROGRAM OTHER CORRELATIONS OF THEIR CHOICE AS OPTIONS.
C
C
C IMPLICIT INTEGER ($)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX

```

COB42360  
COB42370  
COB42380  
COB42390  
COB42400  
COB42410  
COB42420  
COB42430  
COB42440  
COB42450  
COB42460  
COB42470  
COB42480  
COB42490  
COB42500  
COB42510  
COB42520  
COB42530  
COB42540  
COB42550  
COB42560  
COB42570  
COB42580  
COB42590  
COB42600  
COB42610  
COB42620  
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COB42720  
COB42730  
COB42740  
COB42750  
COB42760  
COB42770  
COB42780  
COB42790  
COB42800  
COB42810  
COB42820  
COB42830  
COB42840  
COB42850  
COB42860  
COB42870  
COB42880  
COB42890  
COB42900

```

1 ELEV ,FERROR,FLO ,FTM ,GC ,GK ,GRID ,HSURF ,HF ,COB42910
2 HFG ,HG ,I2 ,I3 ,IERROR, IOP3 ,ITERAT, J1 ,J2 ,COB42920
3 J3 ,J4 ,J5 ,J6 ,J7 ,KDEBUG, KF ,KLJ ,COB42930
4 NAFAC, NARAMP, NAX ,MAXL ,NBBC ,NCHAN ,NCHF ,NDX ,NF ,COB42940
5 NGAPS ,NGRID ,NGRIDT, NGTYPE, NGXL ,NK ,NODES ,NODESF, NPROP ,COB42950
6 NRAM, NROD ,NSCBC ,NV ,NVIS, CW, PI ,PITCH ,POWER ,PREF ,COB42960
7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID, THETA ,THICK ,COB42970
8 UF ,VF ,VFG ,VG ,VZ ,COB42980
9 COB42990
C COMMON /COBRA2/ AA(4), AF(7), AFAC(10,10), AV(7), AXIAL(30),
1 AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2 GAPXL(10), GFAC(9,10), GRIDXL(10), HGAP(2), HMF(30), MHG(30),
3 IGRID(10), KCLAD(2), KFUEL(2), KMF(30), NCH(10), NGAP(9),
4 PP(30), RCLAD(2), REUEL(2), SSIGMA(30), TCLAD(2), UUF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30)
C
C LOGICAL GRID
REAL KIJ, KF, KKF, KCLAD, KFUEL
C
C COMMON /COBRA3/ MA ,MC ,MG ,MH ,MI ,MJ ,MK ,MS ,MX ,MY ,MZ ,
1 $$$ ,SA ,SAAA ,SAC ,SALPHA, SAN ,SANSME, SB ,SCM43130
2 SCCHAN, SCD ,SCHFR ,SCON ,SCOND ,SCP ,SD ,SDC ,SDFDX ,COB43140
3 SDHX ,SDHYD ,SDHYD, SDIST ,SDPDX ,SDPK ,SDUR ,SDR ,SF ,COB43150
4 $FAC10, $FDIV ,$FINLE, $FLUX ,$FMULT, $FOLD ,$FSP ,$FSPLI, $FAFLO, COB43160
5 $GAP ,$GAPN ,$GAPS ,SH ,$HFILM, $HINLE, $HOLD ,$HPERI, $IDARE, COB43170
6 $IDFUE, $IDGAP, $IK , $JBOIL, $JK , $K , $LENGT, $LOCA , $LW , COB43180
7 $MCFR, $MCFRC, $MCFRR, $NTYPE, $NWRAP, $NWRPS, $P , $PERIM, $PH , COB43190
8 $PH1 , $PRNTC, $PRNTR, $PRNTR, $PW , $PWRF , $SOC , $OF , $OPRIM, COB43200
9 $QUAL , $RADIA, $RHO , $RHOOL, $SP , $ST , $TDUMY, $TINLE, $TROD , COB43210
A $U , $UHS , $USAVE , $USTAR , $V , $VISC , $VISCW, $VP , $VPA , COB43220
A $W , $WOLD , $WP , $WSAVE, $X , $XCROS, $XA , $XB , $XPOLD , COB43230
C COMMON /CHFSV/ CHSAVE(20,20,31)
C
C COMMON /LINK4/ IFRM, IMTM, IPROP, MCC, NCF, NDM1, NDS, NGP
C
C COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
C
C NDXP1 = NDX + 1
DO 100 J=1, NDXP1
DATA(SMCFR+J)=10.0
IDAT(SMCFRC+J)=0
IDAT(SMCFRR+J)=0
DO 100 N=1, NROD
DATA(SCFR +N*MR-(J-1))=10.
IDAT(SCCHAN+N*MR*(J-1))=0
100 CONTINUE
2000 FORMAT(1H , ' ERROR DETECTED IN CHF - ',

```



```

1  NCHF=5 AND IHFM DOES NOT = 2.)
DO 500 J=JSTART,JEND
  CHFR0D = 0
  DO 300 N=1,NROD
    XMCNFR = 10.
    IF(DATA($FLUX+N*MR*(J-1)).LE.0.0) GO TO 300
    DD 290 L=1,6
    IF(IDAT($LR +N*MR*(L-1))) 200,290,200
  C  CALCULATE CHF RATIO FOR ROD N FACING CHANNEL I.
  200 I= IDAT($LR +N*MR*(L-1))
    XCHFR = 0.
    IF(NCHF.EQ.1) XCHFR = CHF1(N,I,J)/DATA($FLUX+N*MR*(J-1))
    IF(NCHF.EQ.2) XCHFR = CHF2(N,I,J)/DATA($FLUX+N*MR*(J-1))
    IF(NCHF.EQ.3) XCHFR = CHF3(N,I,J)/DATA($FLUX+N*MR*(J-1))
    IF(NCHF.EQ.4) XCHFR = CHF4(N,I,J)
  CC OPTION NCHF=5 OPERATIONAL ONLY IF IHFM=2
  CC BECAUSE CHSAVE CALCULATED IN HTCOR AND SAVED
    IF (NCHF.EQ.5.AND.IHFM.EQ.2)
      1 XCHFR = CHSAVE(N,I,J)/DATA($FLUX+N*MR*(J-1))
    IF(XCHFR.LE.0.) GO TO 1000
  C  CALCULATE MINIMUM CHF RATIO FOR ROD N FACING CHANNEL I.
    IF(XCHFR.GT. DATA($CHFR+N*MR*(J-1))) GO TO 290
    DATA($CHFR+N*MR*(J-1))=XCHFR
    IDAT($CCHAN+N*MR*(J-1))=1
  CHFR0D = N
  290 CONTINUE
  C  DETERMINE MINIMUM CHF RATIO AT AXIAL LOCATION J.
    XMCNFR =DATA($CHFR+N*MR*(J-1))
    IF(XMCNFR.GT. DATA($MCHFR+J ) ) GO TO 300
    DATA($MCHFR+J)=XMCNFR
    IDAT($ICFR+J)=CHFR0D
    IDAT($ICFR+J)=IDAT($CCHAN+N*MR*(J-1))
  300 CONTINUE
  500 CONTINUE
  RETURN
  1000 PRINT 1
    1 FORMAT (' ERROR IN CHF ROUTINE ')
  RETURN
  END
  C
  C
  FUNCTION CHF2(N,I,J)
  IMPLICIT INTEGER ($)
  COMMON /COBRA1/ ABETA , AFLUX , ATOTAL, BBETA , DIA , DT , DX
  1 ELEV , FERROR, FLO , FTM , GC , GK , GRID , HSURF , HF
  2 HFG , HG , I2 , I3 , IERROR, IOP3 , ITERAT, J1 , J2
  3 J3 , J4 , J5 , J6 , J7 , KDEBUG, KF , KIJ ,
  4 NAFACT, NARAMP, NAX , NAXL , NBBBC , NCHAN , NCHF , NDX , NF
  5 NGAPS , NGRID , NGRIDT, NGTYPE, NGXL , NK , NODESF, NPROP ,
  6 NRAMIP , NROD , NSCBC , NV , NVISCH, PI , PITCH , POWER , PREF ,
  7 QAX , RHOF , RHOG , RHOGG , SIGMA , SL , TF , TFLUID, THETA , THICK
  8 UF , VF , VFG , VG , Z
  COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
  1 AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),

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COB43460  
COB43470  
COB43480  
COB43490  
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COB43590  
COB43600  
COB43610  
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COB43880  
COB43890  
COB43900  
COB43910  
COB43920  
COB43930  
COB43940  
COB43950  
COB43960  
COB43970  
COB43980  
COB43990  
COB44000

```

2  GAPL(10), GFACT(9,10), GRIDXL(10), HGAP(2), MHF(30), HHG(30), COB44010
3  IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9), COB44020
4  PP(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30), COB44030
5  VVF(30), VVG(30), XQUAL(30), Y(30), TT(30), COB44040
C COB44050
C COB44060
C COB44070
C COB44080
C COB44090
C COB44100
C COB44110
C COB44120
C COB44130
C COB44140
C COB44150
C COB44160
C COB44170
C COB44180
C COB44190
C COB44200
C COB44210
C COB44220
C COB44230
C COB44240
C COB44250
C COB44260
C COB44270
C COB44280
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C COB44330
C COB44340
C COB44350
C COB44360
C COB44370
C COB44380
C COB44390
C COB44400
C COB44410
C COB44420
C COB44430
C COB44440
C COB44450
C COB44460
C COB44470
C COB44480
C COB44490
C COB44500
C COB44510
C COB44520
C COB44530
C COB44540
C COB44550

```

LOGICAL GRID

REAL KIJ, KF, KKF, KCLAD, KFUEL

```

COMMON /COBRA3/ MA ,MC ,MG ,MH ,MR ,MS ,MX
1  $$$ ,SA ,SAA ,SAC ,SALPHA ,SAN ,SANSWE ,SD
2  SCD ,SCHFR ,SCON ,SCP ,SDC ,SDFDX
3  SDHX ,SDHYD ,SDHYDN ,SDIST ,SDPK ,SDUR ,SDR ,SF
4  SFACTO ,SFDIV ,SFINLE ,SFLUX ,SFMULT ,SFOLD ,SFSP ,SFSPLI ,SFXALO ,SFBAD
5  SIDFUE ,SIDGAP ,SIK ,SJOBOL ,SJK ,SLENGT ,SLOCA ,SLR
6  SMCFR ,SMCFRC ,SMCFRR ,SMIYPE ,SMWRAP ,SMWRPS ,SP ,SPERIM ,SPH
7  SPMI ,SPRNTC ,SPRNTA ,SPRNTM ,SPM ,SPWRF ,SOC ,SOF ,SOPRIM ,COB44190
8  SQUAL ,SRADIA ,SRHD ,SRHDL ,SSP ,ST ,STDUMY ,STINLE ,STROD ,COB44200
9  SU ,SUH ,SUSAVE ,SUSTAR ,SV ,SVISC ,SVISCH ,SVP ,SVPA
A  $H ,SHULD ,SNP ,SNSAVE ,$X ,$XCROS ,$$A ,$$B ,$XPOLD

```

COMMON DATA(1)

```

LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))

```

C W-3 CORRELATION INCLUDING, SPACER FACTOR, UNHEATED WALL CORRECTION,

```

C AXIAL FLUX FACTOR
C REFERENCE, LS TONG, BOILING CRISIS AND CRITICAL HEAT FLUX
C AEC CRITICAL REVIEW SERIES, TID-25887(1972).

```

```

DE=4.*DATA(SA+1)/DATA(SPERIM+1)
DH=4.*DATA(SA+1)/DATA(SPERI+1)
RU = 1.-DE/DH
XX=(DATA(SH+1)+MC*(J-1))-HF/HFG

```

C W-3 CORRELATION USING EQUILIBRIUM STEAM QUALITY

```

CHF2 = ((2.022 - 0.0004302*PREF) + (0.1722 - 0.0000984*PREF)
1 *EXP(18.2 - 0.004129*PREF)*XX)
2 *(0.1484-1.596*XX+1.729*XX*ABS(XX))*DATA(SF+1)+MC*(J-1))/
DATA(SA+1)

```

3 \*.0036 + 1.037

4 \*(1.157 - 0.869\*XX)

5 \*(0.2664 + 0.8357\*EXP(-37.812\*DH))

6 \*(0.8258+0.000794\*(HF-DATA(SHINLE+1)))/.0036

C UNHEATED WALL CORRECTION

```

IF(RU.GT.0.) CHF2 = CHF2*(1. - RU*(13.76-1.372*EXP(1.78*XX)
1 -4.732/(DATA(SF+1)+MC*(J-1))/DATA(SA+1)*0.0036)*.0535
2 -11.101*DH*.1077)
3 -0.0619*(PREF*.001)*.14

```

C SPACER FACTOR CORRECTION

```

C USER SHOULD SELECT PROPER VALUE OF TDC
TDC = .019
IF(NGRID.GT.0) CHF2 = CHF2

```

CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

```

1 1*(1+.03*DATA(SF+I*MC*(J-1))/DATA(SA+I))*0.036*(TDC/.019)**.35)
C AXIAL FLUX PROFILE CORRECTION
FAXIAL = 1.
IF(J.LE.IDAT(SUBOIL+1)) GO TO 10
C=1.8*(1.-XX)**4.31/(DATA(SF+I*MC*(J-1))/DATA(SA+I))*0.036**0.478
SUM = 0.
JS=IDAT(SUBOIL+1)+1
CE=C/2.
DO 5 J,J=J,S
5 SUM=SUM+DATA(SFLUX+N*MR*(J-1))*(EXP(CE*DATA(SX+JU)))+
1EXP(CE*DATA(SX+JU-1))*(EXP(CE*DATA(SX+JU))-EXP(CE*DATA(SX+JU-1)
2)))
FAXIAL=SUM*EXP(-CE*DATA(SX+J))/DATA(SFLUX+N*MR*(J-1))//
1(1.-EXP(-C*(DATA(SX+J)-DATA(SX+JS-1))))
FAXIAL=FAXIAL*EXP(-CE*DATA(SX+J))
10 CHF2 = CHF2/FAXIAL
RETURN
END
FUNCTION CHF3(N,I,J)
C
C
C IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX ,
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , HF ,
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KI , J ,
4 NAFAC , NARAMP , NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , RPROP ,
6 NRAMP , NRFD , NSCBC , HV , NVISCM , PI , PITCH , POWER , PREF ,
7 OAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VF , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFAC(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
3 IGRID(10) , KCLAD(2) , KEUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4 PPI(30) , RCLAD(2) , RFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)
C
C LOGICAL GRID
REAL KI , J , KF , MKF , KCLAD , KFUEL
C
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MK
1 $$$ , $A , $AA , $AC , $ALPHA , $AN , $ANSWE , $B ,
2 $SCHAR , $CD , $CHFR , $CON , $COND , $CP , $D , $DC , $DFDX ,
3 $DHDX , $DHYD , $DHYDN , $DIST , $DPDX , $DPK , $DUR , $DR , $F ,
4 $FACT , $FDIV , $FINLE , $FLUX , $FRULT , $FOLD , $FSP , $FSPLI , $FAFLO ,
5 $IDFUE , $IDGAP , $IK , $SUBOIL , $JK , $SLC , $LENGT , $LOCA , $LR ,
6 $MCHFR , $MCFRC , $MCFRR , $NTYPE , $NWRAP , $NWRPS , $P ,
7 $PHI , $PRNTC , $PRNTR , $PRNTN , $PW , $PMRF , $QC , $QF , $OPRIM ,
8 $QUAL , $RADIA , $RHO , $RHOOOL , $SP , $T , $TDUMY , $TINLE , $TROD ,
9 $SU , $SUH , $USAVE , $USTAR , $V , $VISC , $VISCM , $VPA , $VPA

```

COB44560  
COB44570  
COB44580  
COB44590  
COB44600  
COB44610  
COB44620  
COB44630  
COB44640  
COB44650  
COB44660  
COB44670  
COB44680  
COB44690  
COB44700  
COB44710  
COB44720  
COB44730  
COB44740  
COB44750  
COB44760  
COB44770  
COB44780  
COB44790  
COB44800  
COB44810  
COB44820  
COB44830  
COB44840  
COB44850  
COB44860  
COB44870  
COB44880  
COB44890  
COB44900  
COB44910  
COB44920  
COB44930  
COB44940  
COB44950  
COB44960  
COB44970  
COB44980  
COB44990  
COB45000  
COB45010  
COB45020  
COB45030  
COB45040  
COB45050  
COB45060  
COB45070  
COB45080  
COB45090  
COB45100

```

C
A SW ,SWOLD ,SWP ,SWSAVE ,SX ,SXCRDS ,SSA ,SSB ,SXPOLD
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),JDAT(1),LDAT(1))
C
C MENCH-LEVY CORRELATION FOR CRITICAL HEAT FLUX
C
C
Q=DATA($F+I+MC*(J-1))*0.036/DATA($A+I)
XE=(DATA($H+I+MC*(J-1))-HF)/HFG
IF(DATA($FLUX+NR*(J-1)).LE.0.) GO TO 10
XC1=0.273-0.212*(TANH(3.*G))**2
XC2=0.5-0.269*(TANH(3.*G))**2+0.0346*(TANH(2.*G))**2
IF(XE.GE.XC2) Q=0.6-0.7*XE-0.09*(TANH(2.*G))**2
IF(XE.GT.XC1.AND.XE.LT.XC2) Q=1.9-3.3*XE-0.7*(TANH(3.*G))**2
IF(XE.LT.XC1) Q=1.0
Q=Q+1.E6
Q=Q*(1.1-0.1*((PREF-600.)/400.))**1.25)
Q=Q/3600
CHF3=Q
RETURN
C
C 10 CHF3=10.*DATA($FLUX+NR*(J-1))
RETURN
END
FUNCTION CHF4(N,I,J)
C
C
IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA ,AFLUX ,ATOTAL,BBETA ,DIA ,DT ,DX
1 ELEV ,FERROR,FLO ,FTM ,GC ,GRID ,HSURF ,HF
2 HFG ,HG ,I2 ,I3 ,IERROR,IQP3 ,ITERAT ,JI ,J2
3 J3 ,J4 ,J5 ,J6 ,J7 ,KDEBUG,KF ,KIJ
4 NAFACT,NARAMP,NAX ,NAXL ,NBBC ,NCHAN ,NCHF ,NDX ,NF
5 NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGXL ,NK ,NODES ,NODESF ,NPROP
6 NRAMP ,NRDOD ,NSCBC ,NV ,NVISCH ,PI ,PITCH ,POWER ,PREF
7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK
8 UF ,VF ,VFG ,VG ,VZ
COMMON /COBRA2/ AA(4) ,AF(7) ,AFACT(10,10) ,AV(7) ,AXIAL(30)
1 AXL(10) ,BB(4) ,BK(30) ,CC(4) ,CCLAD(2) ,CFUEL(2) ,DFUEL(2)
2 GAPXL(10) ,GFACT(9,10) ,GRIDXL(10) ,HGAP(2) ,HHF(30) ,HMG(30)
3 IGRIDI(10) ,KCLAD(2) ,KFUEL(2) ,KKF(30) ,NCH(10) ,NGAP(9)
4 PFI(30) ,RCLAD(2) ,RFUEL(2) ,SSIGMA(30) ,TCLAD(2) ,UUF(30)
5 VVF(30) ,VVG(30) ,XQUAL(30) ,Y(30) ,YT(30)
C
C
LOGICAL GRID
REAL KIJ ,KF ,KKF ,KCLAD ,KFUEL
COMMON /COBRA3/ MA ,MC ,MG ,MN ,MR ,MS ,MX
1 $$$ ,SA ,SAA ,SAC ,SALPHA,SAN ,SANSUE,SB ,COB45640
,COB45630
,COB45620
,COB45610
,COB45590
,COB45580
,COB45570
,COB45560
,COB45550
,COB45540
,COB45530
,COB45520
,COB45510
,COB45500
,COB45490
,COB45480
,COB45470
,COB45460
,COB45450
,COB45440
,COB45430
,COB45420
,COB45410
,COB45400
,COB45390
,COB45380
,COB45370
,COB45360
,COB45350
,COB45340
,COB45330
,COB45320
,COB45310
,COB45300
,COB45290
,COB45280
,COB45270
,COB45260
,COB45250
,COB45240
,COB45230
,COB45220
,COB45210
,COB45200
,COB45180
,COB45170
,COB45160
,COB45150
,COB45140
,COB45130
,COB45120
,COB45110

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1  SCCHAN,SCD ,SCHFR ,SCON ,SCORD ,SCP ,SD ,SDC ,SDFDX ,COB45660
2  SDMDX ,SDHYD ,SDHYDN ,SDIST ,SDPDX ,SDPK ,SDUR ,SDR ,SF ,COB45670
3  $FACTO,$FDIV ,$FINLE,$FLUX ,$FNULT,$FOLD ,$FSP ,$FSPLI,$FKFLO ,COB45680
4  $GAP ,$GAPN ,$GAPS ,$H ,$HFILM,$HINLE,$HOLD ,$HPERI,$IDARE ,COB45690
5  $IDYUE,$IDGAP,$IK ,SUBOIL,$JK ,SLC ,$LENGT,$LOCA ,SLR ,COB45700
6  $MCHFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP ,SNWRP,$P ,COB45710
7  $PHI ,$PRNTC,$PRNTR ,$PRNTN ,$PW ,$PWRP ,SOC ,$OF ,$QPRIM ,COB45720
8  $QUAL ,$RADIA,$RHO ,SRHOL,$SP ,SY ,$TDUMY,$TINLE,$TROM ,COB45730
9  $U ,SUH ,SUSAVE ,SUSTAR ,SV ,SVISC ,SVISCH,$VPA ,COB45740
A  $W ,SWOLD ,SWP ,$WSAVE,$X ,$XCROS,$XA ,$XB ,$XPOLD ,COB45750
C  COB45760
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C C THE CISE CORRELATION IS USED TO ESTIMATE
C CRITICAL POWER
C C
C IF(J,L,E.IDAT($JBOIL+I)) GO TO 100
C
C XLBL=.304B*DX*FLOAT(J-IDAT($JBOIL+I))
G=4.8B*DATA(SF+INC*(J-1))/DATA($A+I)
C1=(1.-PREF/3206.)
GSTAR=3375.*C1**3
DH=.304B*DATA($HYD+I)
A=C1/(G*.001)**.333
IF(G,LT,GSTAR) A=1./(1.+1.481E-4*C1**(-3)*G)
B=0.190*(3206./PREF-1.)**0.4*G*DH**1.4
XCR=(DATA($HPERI+I)*A*XLDL)/(DATA($PERIM+I))*(XLBL*B)
XE=(DATA($H+INC*(J-1))-HF)/HFG
HSUB=H1-DATA($H+I)
CPR=(XCR*HFG+HSUB)/(XE*HFG+HSUB)
CHF4=CPR
RETURN
C 100 CHF4=10.
RETURN
END
SUBROUTINE DIVER(T,J)
C THIS PROCEDURE CONTAINS THE COMMON AND TYPE STATEMENTS SHARED BY THE COB46070
C MAJOR SUBROUTINES OF COBRA-IIIIC.
C C
C IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1 ELEV , FERROR , FLO , FTM , GC , GRID , HSURF , HF , COB46120
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , J1 , J2 , COB46130
3 J3 , J4 , J5 , J6 , J7 , KOEBUG , KF , KI , J , COB46140
4 NAFACT , NARAMP , NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF , COB46150
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGKL , NK , NODES , NODESF , NPROP , COB46160
6 NRAMP , NROD , NSCBC , NV , NVISCH , PI , PITCH , POWER , PREF , COB46170
7 QAX , RHO , RHO , RHO , SIGMA , SL , TF , TFLUID , THETA , THICK , COB46180
8 UF , VF , VFG , VG , VZ , COB46190
COB46200

```



```

MID = (MS+1)/2
DO 310 K=1,NK
DO 290 L=1,LMAX
290 DATA(SAAA+K+NK*(L-1))=0.
II=IDAT($IK+K)
JJ=IDAT($JK+K)
C TRANSVERSE MOMENTUM PARAMETER IN NEXT EQUATION
1 DATA($B+K)=(DATA($SPK+MG*(J-1))-(DATA($DPK+II)-DATA($DPK+JJ))*
+ DX)*SL*FACSL(K)*DATA($FACTO+K)+DATA($USAVE+K)*
2 DATA($W+K+NG*(JM-1))/
DXGC+DATA($WOLD+K+NG*(J-1))/DTGC
3 SAVE=ABIT(1, DATA($U+II), DATA($USTAR+K), DATA($A+II), DATA($DPK+II),
1 DATA($F+II+MC*(JM-1)), DATA($F+II+MC*(J-1)))
2 +ABIT(1, DATA($U+JJ), DATA($USTAR+K), DATA($A+JJ), DATA($DPK+JJ),
3 DATA($F+JJ+MC*(JM-1)), DATA($F+JJ+MC*(J-1)))
IF (IPILE.GT.0) GO TO 7213
NBOUND=IDAT($LOCA+K+MG*13)
GO TO 7214
7213 NBOUND=IDAT($LOCA+K+MG*7)
7214 DO 300 LL=1,NBOUND
L = IDAT($LOCA+K+MG*(LL-1))
IF (LL.EQ.1) GO TO 295
I2 = 1
IF (L,LT.0) I2=-1
L = IABS(L)
IJ = JJ
IF (IJ.EQ.IDAT($IK+L),OR.
1 IJ.EQ.IDAT($JK+L)) IJ=II
SAVE = ABIT(I2,DATA($U+IJ),DATA($USTAR+L),DATA($A+IJ),
1 DATA($DPK+IJ),DATA($F+IJ+MC*(JM-1)),DATA($F+IJ+MC*(J-1)))
295 L = MID - K + L
C TRANSVERSE MOMENTUM PARAMETER IN NEXT EQUATION
300 DATA($AA+K+HK*(L-1))=SAVE*SLDX*FACSL(K)/GC*DATA($FACTO+K)
C TRANSVERSE MOMENTUM PARAMETER IN NEXT EQUATION
DATA($AAA+K+HK*(MID-1))=
1 DATA($AAA+K+HK*(MID-1))*SL*FACSLK(K)*CJ(K,J)*DATA($FACTO+K)+
2 DATA($USTAR+K)/DXGC+1./DTGC
310 CONTINUE
IF(J6.LT.1) GO TO 105
C
C MODIFY SIMULTANEOUS EQUATIONS TO ACCOUNT FOR SPECIFIED VALUES OF
C CROSSLFLOW GIVEN IN SUBROUTINE FORCE
C
DO 90 K=1,NK
IF(LDAT($FDIV+K)) GO TO 90
DO 85 L=1,NK
LL=MID-K+L
IF(LL.EQ.MID) GO TO 85
IF(LL.GT.LMAX.OR.LL.LT.1) GO TO 85
IF(LDAT($FDIV+L))
1 DATA($B+K)=DATA($B+K)-DATA($B+K)+DATA($AAA+K+HK*(LL-1))*
85 CONTINUE
90 CONTINUE
DO 100 K=1,NK

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COB46760

COB46770

COB46780

COB46790

COB46800

COB46810

COB46820

COB46830

COB46840

COB46850

COB46860

COB46870

COB46880

COB46890

COB46900

COB46910

COB46920

COB46930

COB46940

COB46950

COB46960

COB46970

COB46980

COB46990

COB47000

COB47010

COB47020

COB47030

COB47040

COB47050

COB47060

COB47070

COB47080

COB47090

COB47100

COB47110

COB47120

COB47130

COB47140

COB47150

COB47160

COB47170

COB47180

COB47190

COB47200

COB47210

COB47220

COB47230

COB47240

COB47250

COB47260

COB47270

COB47280

COB47290

COB47300

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COB47310
COB47320
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COB47340
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COB47480
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COB47650
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COB47680
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COB47900
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COB47920
COB47930
COB47940
COB47950
COB47960
COB47970
COB47980
COB47990
COB48000

IF(.NOT.LDAT($FDIV+K)) GO TO 100
DO 95 L=1,LMAX
DATA($AAA+K+MK*(L-1)) = 0.0
LL = MAXO(1,(L+K-MID))
LL=MINO(LL,NK)
MPICU=MID*K-LL
95 DATA($AAA+LL+MK*(MPICU-1))=0.0
DATA($AAA+K +MK*(MID-1)) = 1.0
DATA($B+K)=DATA($W+K+MG*(J-1))
CONTINUE
100 CONTINUE
105 IF(INDEBUT.LT.1) GO TO 110
WRITE(13,2) ((DATA($AAA+K+MK*(L-1)),L=1,LMAX),DATA($B+K),K=1,MNK)
2 FORMAT(1H0, 1P1E15.4)
110 CALL DECOMP(NK,ERROR,LMAX,MID,DATA($AAA+1),DATA($B+1),DATA($SWE+1),
1 DATA($B+1),NK)
IF(ERROR.GT.1) GO TO 1000
CALL SOLVE(NK,LMAX,MID,DATA($AAA+1),DATA($SWE+1),DATA($B+1),NK)
DO 150 K=1,NKK
150 DATA($W+K+MG*(J-1))=DATA($SWE+K)
RETURN
1000 WRITE(13,1)
1 FORMAT(24H ERROR IN DECOMP, DIVERT )
1 ERROR = 3
RETURN
END
SUBROUTINE INDAT(INIT,NOPRIN)
C
C
IMPLICIT INTEGER ($)
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL,BBETA , DIA , DT , DX ,
1 ELEV , FERROR,FLO , FTM , GC , GK , GRID , HSURF , HF ,
2 HFG , HG , I2 , I3 , IERROR,IOP3 , ITERAT , J1 , J2 ,
3 J3 , J4 , J5 , J6 , J7 , KDEBUG,KF , KIJ ,
4 NAFAC , NARAMP , NAX , NAXL , NBBC , NCHAN , NCHF , NDX , NF ,
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , NPROP ,
6 NRAM , NRCD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID,THETA , THICK ,
8 UF , VF , VFG , VG , VZ ,
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , MHF(30) , MHG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4 PPI(30) , RCLAD(2),RFUEL(2),SSIGMA(30), TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , Y(30) , Y(30)
C
C
LOGICAL GRID,PRINT
REAL MIJ , KF , KKF , KCLAD , KFUEL
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MX ,
1 $$$ , SA , $AAA , $AC , $ALPHA,$AN , $ANSWE,$B
1 $CCHAN,$CD , $CHFR , $CON , $COND , $CP , $D , $DC , $DFDX ,
2 $DHDX , $DHYD , $DHYDN , $D1ST , $DPOX , $DPK , $DUR , $DR , $F

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3 SFAC20,SF DIV ,SFINLE,SFLUY ,SFMULT,SFOLD,SFSP ,SFSPLI,SFIFLO,C0B478C0
4 SGAP ,SG:PN ,SGAPS ,SH ,SHELM,SHEINLE,$HOLD,$HPERT,$IDARE,C0B47B70
5 $IDFUE,$IDGAP,$IK ,SUBOIL,$JK ,SLC ,SLENGT,$LOCA,$LR ,C0B478B0
6 $MCHFR,$MCFRC,$MCFRR,$NTYPE,$NHRAP,$NHRPS,$P ,SPEPRIM,$PH ,C0B47890
7 $PHI ,SPRINTC,$PRNTR,$PRNTN,$PW ,SPWRF,$SOC ,SOF ,SOPRIM,C0B47900
8 $QUAL,$RADIA,$RHO,$RHOOL,$SS ,ST ,STOUMY,$TINLE,$TROD,C0B47910
9 $U ,SUH ,SUSAVE,$USTAR,$V ,SVISC,$SVISCM,$VPA ,C0B47920
A $W ,SWOLD,$SWP ,SWSAVE,$X ,$XCROS,$XA ,$XB ,SXPOLD ,C0B47930
C0B47940
C0B47950
C0B47960
C0B47970
C0B47980
C0B47990
C0B48000
C0B48010
C0B48020
C0B48030
C0B48040
COMMON/LINK2/CROSS(6),DATE(2),FG(30),FH(30),FP(30),FO(30),FM(9),
1 JMI(9),OUTPUT(10),PRINT(12),TEXT(17),TIME(3),Y(30),YH(30),YP(30),
2 YQ(30)
COMMON/LINK3/DTX,ETIME,GIN,HIN,IB,IG,IN,ISAVE,JUMP,KASE,KT,MAKT,
1 NDT,NDXPI,NFUELT,NG,NH,NJUMP,NOUT,NP,NPCHAN,NPNODE,NPROD,NQ,NR,
2 NSKIPT,NSKIPIX,NTRIES,PEXIT,PHTOT,SAVEDT,TIN,TIME,ZZ
CC
COMMON/LINK4/IFRM,IHTM,IPROP,NCC,NCF,NDM1,NDS,NGP
CC
COMMON/LINK9/ENEH(400)
C READ I/I INPUT DATA (MAIN 5365-8830)
IF (INIT.EQ.2) GO TO 990
C
C THE UNIVAC 1108 SETS THE CORE TO ZERO AT THE START OF EACH JOB
C THE INITIALIZATION BELOW IS TO INITIALIZED FOR OTHER MACHINES
C UNITS 12,13, AND 18 ARE THE INPUT, OUTPUT, AND SAVE TAPE UNITS
CC BEGINNING OF VARIABLE BLOCK
12=5
13=6
13=6
READ(12,68) MC,MG,MN,MR,MX
WRITE(13,3000) MC,MG,MN,MR,MX
3000 FORMAT('1',T50,'PROBLEM SIZE',T50,'MC=',T5/
1 T50,'MG=',T5/T50,'MN=',T5/T50,'MR=',T5/T50,'MX=',T5//)
MX=MX+1
CALL CORE
C ALL VALUES INITIALIZED TO ZERO BETWEEN HERE AND 930 COULD PROBABLY
C BE LEFT OUT SINCE NOW INITIALIZED IN CORE. HOWEVER LEFT IN FOR
C TIME BEING FOR SAFETY AS NO TIME TO CHECK.
PI = 355./113.
10=8
GC = 32.2
NAXL = 0
NGXL = 0
NGRID = 0
NAX = 0
NAX = 0
ERROR = 0

```

```

NGAPS = 0
NAFACT = 0
NSCBC = 0
NBBC = 0
J5 = 0
J6 = 0
NOPRIN=0
J7=0
NGRIDT = 0
JUMP = 0
NJUMP = 0
NRD = 0
NRAMP = 1
NODESF = 0
NFUELT = 0
NOUF = 0
NCHAN = 0
NPRDDE = 0
NARAMP = 1
IG = 0
ISAVE = 0
IN = 0
IFRM=0
IHTM=0
IPROP=0
GRID = .FALSE.
DO 900 I=1,MC
DATA($HINLE+I)=0.
DATA($FINLE+I)=0.
DATA($OPRIM+I)=0.
DO 905 K=1, MG
FACSL(K)=1.
FACSL(K)=1.
DATA($MP +K)=0.
LDAT($FDIV +K)=.FALSE.
DO 930 J=1,MX
DO 910 I=1,MC
DATA($P +I+MC*(J-1))=0.
DATA($H +I+MC*(J-1))=0.
DATA($F+I+MC*(J-1))=0.
DATA($IHO +I+MC*(J-1))=0.
DATA($HOLD +I+MC*(J-1))=0.
DATA($FOLD +I+MC*(J-1))=0.
DATA($RHOOOL+I+MC*(J-1))=0.
DO 920 N=1,MR
DATA($FLUX +N+MR*(J-1))=0.
IDAT($CCHAN+N+MR*(J-1))=0
DO 918 L=1,MH
DATA($TROD+L+MN*(N-1+MR*(J-1)))=0.
900 CONTINUE
930 CONTINUE
READ (12,52) MAXT
IF(MAXT.LT.1) MAXT = 1000

```

CC FUEL ROD AND HEAT TRANSFER MODEL INDICATORS INITIALIZED AS ZERO

```

COB48410
COB48420
COB48430
COB48440
COB48450
COB48460
COB48470
COB48480
COB48490
COB48500
COB48510
COB48520
COB48530
COB48540
COB48550
COB48560
COB48570
COB48580
COB48590
COB48600
CC948610
COB48620
COB48630
COB48640
COB48650
COB48660
COB48670
COB48680
COB48690
COB48700
COB48710
COB48720
COB48730
COB48740
COB48750
COB48760
COB48770
COB48780
COB48790
COB48800
COB48810
COB48820
COB48830
COB48840
COB48850
COB48860
COB48870
COB48880
COB48890
COB48900
COB48910
COB48920
COB48930
COB48940
COB48950

```

```

C READ CASE CONTROL CARD
990 READ(12,2) IPILE,KASE,J1,TEXT
J7 = IPILE
  ERROR = 0
  ISAVE = 0
  DO 991 I = 1,11
    PRINT(1) = .FALSE.
  IF(J1.EQ.1) PRINT(1) = .TRUE.
991 CONTINUE
C CHECK FOR CONTINUATION OF CALCULATIONS
  IF(KASE.LT.1) STOP
  DO 915 J=1,MX
  DO 914 K=1,MC
    DATA$COND+K) = 0.0
    DATA$W +K+MG*(J-1))=0.
    DATA$SP +K+MG*(J-1))=0.
914 DATA$WOLD+K+MG*(J-1))=0.
    DATA$QC +K+MC*(J-1))=0.
    DATA$QF +K+MC*(J-1))=0.
915 CONTINUE
  IDAT($IK+1) = 1
  IDAT($JK+1) = 1
  CALL DOY(DATE)
  CALL TOD(TIME)
  WRITE(13,3) KASE,TEXT,DATE,TIME
  IF(IPILE.EQ.0) WRITE(13,1000)
  IF(IPILE.EQ.1) WRITE(13,1001)
  IF(IPILE.EQ.2) WRITE(13,1002)
C READ GROUP CONTROL CARD
995 READ(12,1) NGROUP,N1,N2,N3,N4,N5,N6
  IF(INGROUP.EQ.20) GO TO 230
  IF(INGROUP.LT.1) GO TO 250
  IF(INGROUP.GT.12) GO TO 240
  IF(INGROUP.LT.0) GO TO 240
  GO TO (110,120,130,140,150,160,170,180,190,200,210,220),NGROUP
C INPUT FOR CARD GROUP 1, PROPERTY TABLE
110 CALL CARDS1(PP,TT,VVF,VVG,HMF,HMG,UUF,KF,SSIGMA,N1,12)
  HPROP = N1
  IF(J1.LE.1) PRINT(1)=.TRUE.
  GO TO 995
C INPUT FOR CARD GROUP 2, FRICTION FACTOR AND TWO-PHASE FLOW CORRELATIO
120 READ (12,5) (AA(I),BB(I),CC(I),I=1,4)
  J2 = N1
  J3 = N2
  J4 = N3
  NVISCM = N4
  IF(J3.GT.4) READ(12,41) NV,AV
  IF(J4.GT.4) READ(12,41) NF,AF
  IF(J1.LE.1) PRINT(2) = .TRUE.
  GO TO 995
C

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COB48960
COB48970
COB48980
COB48990
COB49000
COB49010
COB49020
COB49030
COB49040
COB49050
COB49060
COB49070
COB49080
COB49090
COB49100
COB49110
COB49120
COB49130
COB49140
COB49150
COB49160
COB49170
COB49180
COB49190
COB49200
COB49210
COB49220
COB49230
COB49240
COB49250
COB49260
COB49270
COB49280
COB49290
COB49300
COB49310
COB49320
COB49330
COB49340
COB49350
COB49360
COB49370
COB49380
COB49390
COB49400
COB49410
COB49420
COB49430
COB49440
COB49450
COB49460
COB49470
COB49480
COB49490
COB49500

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C INPUT FOR CARD GROUP 3, AXIAL HEAT FLUX TABLE
130 IF (N1.GT.1) GO TO 135
    IOP3 = N1
    GO TO 995
135 READ(12,5) (Y(I),AXIAL(I),I=1,N1)
    MAX = N1
    IF(J1.LE.1) PRINT(3) = .TRUE.
    GO TO 995

C INPUT FOR CARD GROUP 4, CHANNEL LAYOUT AND DIMENSIONS
140 IF(IPILE.EQ.0) GO TO 1405
    COMBINE CARD GROUPS 4, 7, 9 FOR PHR AND BWR.
    CALL CARDS4(DATA($AC+1),DATA($GAPS+1),
    1 DATA($DR+1),DATA($GAPS+1),
    1 IDAT($LC+1),MA,NG,N1,N2,NCHF,NFUELT, DATA($PH+1),
    2 PHOT,PRINT,DATA($PW+1),MC)
    IF (ERROR.GE.1) GO TO 240
    CALL CORE3
    GO TO 995
1405 DO 141 J=1,N1
    READ(12,7) N,I,DATA($AC+1),DATA($PW+1),DATA($PH+1),
    1 IDAT($LC+1+MC*(L-1)),DATA($GAPS+1+MG*(L-1)),
    2 DATA($DIST+1+MC*(L-1)),L=1,4)
    IDAT($NTYPE+1)=N
    IF(N.LE.1)
    11DAT($NTYPE+1)=1
141 CONTINUE
142 PHOT = 0.
    ATOTAL = 0.
    K=0
    NCHANL = N2
    DO 147 I=1,NCHANL
    DO 146 L=1,4
    IF(1DAT($LC+1+MC*(L-1))) 144,146,143
    J= IDAT($LC+1+MC*(L-1))
    IF(J.LE.1) GO TO 146
    K=K+1
    DATA($FACTO+K)=1.
    GO TO 145
144 J=-IDAT($LC+1+MC*(L-1))
    IF(J.LE.1) GO TO 146
    K=K+1
    DATA($FACTO+K)=0.5
145 IDAT($JK+K)=J
    IDAT($IK+K)=I
    DATA($JAPN +K)=DATA($GAPS +1+MG*(L-1))/12.
    DATA($GAP +K)=DATA($GAPN +K)
    DATA($LENGT+K)=DATA($DIST +1+MC*(L-1))/12.
146 CONTINUE
    DATA($PERIM+1)=DATA($PW+1)/12.
    DATA($HPERI+1)=DATA($PH+1)/12.
    DATA($AN +1)=DATA($AC+1)/144.
    DATA($A +1)=DATA($AN+1)
    DATA($DC +1)=DATA($AC+1)+4./DATA($PW+1)
    DATA($DNVD +1)=DATA($DC +1)/12.
COB49510
COB49520
COB49530
COB49540
COB49550
COB49560
COB49570
COB49580
COB49590
COB49600
COB49610
COB49620
COB49630
COB49640
COB49650
COB49660
COB49670
COB49680
COB49690
COB49700
COB49710
COB49720
COB49730
COB49740
COB49750
COB49760
COB49770
COB49780
COB49790
COB49800
COB49810
COB49820
COB49830
COB49840
COB49850
COB49860
COB49870
COB49880
COB49890
COB49900
COB49910
COB49920
COB49930
COB49940
COB49950
COB49960
COB49970
COB49980
COB49990
COB50000
COB50010
COB50020
COB50030
COB50040
COB50050

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```

DATA($DHYDN+1)=DATA($DHYD +I)
PHTOT=PHTOT+DATA($HPERI +I)
147 ATOTAL=ATOTAL+DATA($SAN+I)
NK=K
CALL ACOL(2, IDAT($IK+1), IDAT($JK+1), NMAX, IDAT($LOCA+1), MA, MS, NK,
1 MG, IPILE)
IF(J1.LE.1) PRINT(4) = .TRUE.
CALL CORE3
GO TO 995

C INPUT FOR CARD GROUP 5, CHANNEL AREA VARIATION TABLE
150 DO 151 I=1, NCHAHL
151 IDAT($IDARE+I)=0
NAXL = N2
NRAMP = N3
IF(NRAMP.LE.0) NRAMP = 1
IF(N2.LT.1) GO TO 995
READ(12,5) (AXL(I), I=1, N2)
NAFACT=N1
DO 152 J=1, N1
READ(12,8) I, (AFACT(J,L), L=1, N2)
IDAT($IDARE+I)=J
152 NCH(J) = I
IF(J1.LE.1) PRINT(5) = .TRUE.
GO TO 995

C INPUT FOR CARD GROUP 6, GAP SIZE VARIATIONS TABLE
160 DO 161 K=1, NK
161 IDAT($IDGAP+K)=0
NGXL = N2
IF(N2.LT.1) GO TO 995
READ(12,5) (GAPXL(L), L=1, NGXL)
NGAPS = N1
DO 162 LL=1, NGAPS
READ(12,1) K
IDAT($IDGAP+K)=LL
NGAP(LL) = K
READ (12, 5) (GFACT(LL,L), L=1, NGXL)
162 CONTINUE
IF(J1.LE.1) PRINT(6) = .TRUE.
GO TO 995

C INPUT FOR CARD GROUP 7, SPACER DESIGN INFORMATION
170 IF(IPILE.EQ.0) GO TO 1705
WRITE(13,1704) IPILE, NGROUP
1704 FORMAT(' IPILE=', I2, ' CARD GROUP', I2,
1 ' INCORRECTLY ENTERED . CHECK DATA')
ERROR = 1
GO TO 240
1705 J6 = N1
NRAMP = N4
IF(NRAMP.LT.1) NRAMP = 1
GRID = .FALSE.
NGRID = C
IF(J6.EQ.0) GO TO 995

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COB50060  
COB50070  
COB50080  
COB50090  
COB50100  
COB50110  
COB50120  
COB50130  
COB50140  
COB50150  
COB50160  
COB50170  
COB50180  
COB50190  
COB50200  
COB50210  
COB50220  
COB50230  
COB50240  
COB50250  
COB50260  
COB50270  
COB50280  
COB50290  
COB50300  
COB50310  
COB50320  
COB50330  
COB50340  
COB50350  
COB50360  
COB50370  
COB50380  
COB50390  
COB50400  
COB50410  
COB50420  
COB50430  
COB50440  
COB50450  
COB50460  
COB50470  
COB50480  
COB50490  
COB50500  
COB50510  
COB50520  
COB50530  
COB50540  
COB50550  
COB50560  
COB50570  
COB50580  
COB50590  
COB50600

```

IF(J6.EQ.1) GO TO 171
IF(J6.EQ.2) GO TO 176
GO TO 995
171 READ(12,42) PITCH,DIA,THICK
    PITCH = PITCH/12.
    DIA = DIA/12.
    THICK = THICK/12.
    NUIMP = NS
    DO 172 M=1,NK
    READ(12,64) K,DUM,CROSS
    DATA($DUR+K) = DUM
    DO 172 L=1,6
    DATA($XCROSS+K+MG*(L-1)) = CROSS(L)
172 READ(12,68) (IDAT($NWRAP+I),I=1,NCHANL)
    DO 173 I=1,NCHANL
    IDAT($NWRPS+I) = IDAT($NWRAP+I)
173 IF(J1.LE.1) PRINT(7) = .TRUE.
    IF(NUIMP.EQ.3) JUMP = 3
    IF(NUIMP.NE.3) GO TO 995
    REWIND 18
    READ(18) ((DATA($M+I+MG*(J-1)),I=1,MC),J=1,MX),
    ((DATA($RHO+I+MC*(J-1)),I=1,MC),J=1,MX),
    ((DATA($RHO+I+MC*(J-1)),I=1,MC),J=1,MX),
    ((DATA($F +I+MC*(J-1)),I=1,MC),J=1,MX)
    REWIND 18
    GO TO 995
176 NGRID = N2
    NGRIDT = N3
    READ(12,66) (GRIDXL(I),IGRID(I),I=1,NGRID)
    DO 178 I=1,NGRIDT
    DO 177 K=1,NK
    DATA($XFLO+K+MG*(I-1)) = 0.
177 DO 178 II=1,NCHANL
178 READ(12,67) J,DATA($CD+J+MC*(I-1)),K,DATA($XFLO+K+MG*(I-1))
    IF(J1.LE.1) PRINT(7) = .TRUE.
    GO TO 995
C
C INPUT FOR CARD GROUP 8, ROD LAYOUT, DIMENSIONS, AND POWER FACTORS
180 IF(IPILE.EQ.0) GO TO 1805
    WRITE(13,1704) IPILE,NGROUP
    IERROR = 1
    GO TO 240
1805 NROD = N2
    DO 181 J=1,N1
    READ (12,11) N,I,DATA($DR+I),
    DATA($RADIA+I), (IDAT($LR+I+NR*(L-1)),
    1 IDAT($IDFUE+I)=N
    1 IF(N.LT.1) IDAT($IDFUE+I)=1
181 CONTINUE
    DO 182 I=1,MC
    DO 182 J=1,NR
182 DATA($PHRF+I+MC*(J-1)) = 0.
    DO 185 I=1,NROD
    DO 184 L=1,6

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COB50610  
COB50620  
COB50630  
COB50640  
COB50650  
COB50660  
COB50670  
COB50680  
COB50690  
COB50700  
COB50710  
COB50720  
COB50730  
COB50740  
COB50750  
COB50760  
COB50770  
COB50780  
COB50790  
COB50800  
COB50810  
COB50820  
COB50830  
COB50840  
COB50850  
COB50860  
COB50870  
COB50880  
COB50890  
COB50900  
COB50910  
COB50920  
COB50930  
COB50940  
COB50950  
COB50960  
COB50970  
COB50980  
COB50990  
COB51000  
COB51010  
COB51020  
COB51030  
COB51040  
COB51050  
COB51060  
COB51070  
COB51080  
COB51090  
COB51100  
COB51110  
COB51120  
COB51130  
COB51140  
COB51150

```

183 IF(IDAT($IR+I+NR*(I-1))) 184,184,183
K = IDAT($LR+I+NR*(I-1))
DATA($PRF+MC*(I-1))=DATA($PHI+I+NR*(I-1))
184 CONTINUE
185 DATA($D+I)=DATA($DR+I)/12.
IF(J1.LE.1) PRINT(8) = .TRUE.
NODESF = N3
MFUELT = N4
NCHF = N5
IF(NODESF.EQ.0) GO TO 995
READ (12,79) (KFUEL(I), CFUEL(I), RFUEL(I), DFUEL(I),
1 KCLAD(I), CCLAD(I), RCLAD(I), TCLAD(I), HGAP(I), I=1,NFUELT)
DO 187 I = 1,NFUELT
KFUEL(I) = MFUEL(I)/3600.
KCLAD(I) = KCLAD(I)/3600.
DFUEL(I) = DFUEL(I)/12.
TCLAD(I) = TCLAD(I)/12.
HGAP(I) = HGAP(I)/3600.
187 CONTINUE
GO TO 995
C
C INPUT FOR CARD GROUP 9, CALCULATION VARIABLES
190 READ(12,14) K1J,FTM,Z,THETA,NDX,NDT,TTIME,NTRIES,FERROR,SL
IF(SL.LT.1.E-5) SL = .5
ELEV = COS(THETA*PI/180.)
IF(NTRIES.LT.1) NTRIES=20
IF(FERROR.LE.0) FERROR = 1.E-3
NDXP1 = NDX + 1
NSKIPX = N1
NSKIPT = N1
KOEBUG = N3
IF(NSKIPT.LT.1) NSKIPT = 1
IF(NSKIPX.LT.1) NSKIPX = 1
ZZ = Z
Z = Z/12.
IF(Z.LE.0.) GO TO 240
IF(NDX.LT.1) GO TO 240
DX = Z/FLOAT(NDX)
DT = 0.
IF(NDT.GT.0 .AND. TTIME.LE.0.) NDT = 0
IF(NDT.GT.0) DT = TTIME/FLOAT(NDT)
SAVEDT = DT
DXX = DX*12.
IF(J1.LE.1) PRINT(9) = .TRUE.
GO TO 995
C
C INPUT FOR CARD GROUP 10, MIXING PARAMETERS
200 IF(IPILE.LT.2) GO TO 205
WRITE(13,1704) IPILE, NGROUP
GO TO 995
205 NSCBC = N1
IF (NSCBC.NE.4) READ(12,5) ABETA,BBETA
DO 206 I=1,MG
208 ENEH(I)=1.0
NBBC =N2

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COB51160

COB51170

COB51180

COB51190

COB51200

COB51210

COB51220

COB51230

COB51240

COB51250

COB51260

COB51270

COB51280

COB51290

COB51300

COB51310

COB51320

COB51330

COB51340

COB51350

COB51360

COB51370

COB51380

COB51390

COB51400

COB51410

COB51420

COB51430

COB51440

COB51450

COB51460

COB51470

COB51480

COB51490

COB51500

COB51510

COB51520

COB51530

COB51540

COB51550

COB51560

COB51570

COB51580

COB51590

COB51600

COB51610

COB51620

COB51630

COB51640

COB51650

COB51660

COB51670

COB51680

COB51690

COB51700

```

J5 = N3
IF(N2.GE.2) READ(12,5) (XQUAL(I),BX(I),I=1,N2)
IF(J5.EQ.0) GK = 0.
IF(J5.EQ.1) READ(12,5) GK
IF(J1.LE.1) PRINT(10) = .TRUE.
GO TO 995

C INPUT FOR CARD GROUP 11, OPERATING CONDITIONS AND TRANSIENT FORCING
210 READ(12,9) PEXIT,MIN,GIN,AFLUX
    PREF = PEXIT
    CALL PROP(1,1)
    IF(IEORR.GT.1) GO TO 240
    IN = N1

C FOR N1=0, MIN IS THE INLET H. FOR N1=1, MIN IS THE INLET T.
C FOR N1=2, READ IN CHANNEL H. FOR N1=3, READ IN CHANNEL T.
    IF(N1.GE.2) GO TO 214
    IF(N1.EQ.1) GO TO 211
    TIN = IF
    IF(HIN.LT.HF) CALL CURVE(TIN,MIN,TT,MHF,NPROP,IEORR,1)
    IF(IEORR.GT.1) GO TO 240
    GO TO 212

211 TIN = HIN
    CALL CURVE(HIN,TIN,MHF,TT,NPROP,IEORR,1)
    IF(IEORR.GT.1) GO TO 240

212 DO 213 I=1,NCHANL
213 DATA($HINLE+I)=HIN
    GO TO 216

214 READ(12,10) (DATA($HINLE+I),I=1,NCHANL)
    IF(HI.LE.2) GO TO 216
    CALL CURVE(DATA($HINLE+I),DATA($HINLE+1),MHF,TT,NPROP,IEORR,1)
    IF(IEORR.GT.1) GO TO 240

215 CONTINUE
216 DO 2160 I=1,NCHANL
    DATA($HINLE+I)=TF
    IF(DATA($HINLE+1).LT.MHF)
1CALL CURVE(DATA($HINLE+1),DATA($HINLE+1),TT,MHF,NPROP,IEORR,1)
    IF(IEORR.GT.1) GO TO 240

2160 CONTINUE
    IG = N2

C FOR N2=0, GIN IS THE INLET G FOR EACH CHANNEL. FOR N2=1, GIN IS THE
C AVERAGE G BUT THE CHANNEL FLOWS ARE SPLIT TO GIVE EQUAL OP/DX, FOR
C INDIVIDUAL CHANNEL TOTAL FLOW FRACTION IS READ AS INPUT
    FLO = GIN/.0036*ATOTAL
    DO 217 I=1,NCHANL
217 DATA($FINLE+I)=GIN*DATA($AN+I)/.0036
    IF(N2.EQ.1) CALL SPLIT
    IF(IEORR.GT.1) GO TO 240
    IF(N2.LT.2) GO TO 219
    READ(12,10) (DATA($FSPLI+I),I=1,NCHANL)
    DO 218 I=1,NCHANL
218 DATA($FINLE+I)=GIN*DATA($AN+I)+DATA($FSPLI+I)/.0036
219 NP = N3
    IF(NP.GT.1) READ(12,10) (YP(I),FP(I),I=1,NP)
    NH = N4

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COB51710
COB51720
COB51730
COB51740
COB51750
COB51760
COB51770
COB51780
COB51790
COB51800
COB51810
COB51820
COB51830
COB51840
COB51850
COB51860
COB51870
COB51880
COB51890
COB51900
COB51910
COB51920
COB51930
COB51940
COB51950
COB51960
COB51970
COB51980
COB51990
COB52000
COB52010
COB52020
COB52030
COB52040
COB52050
COB52060
COB52070
COB52080
COB52090
COB52100
COB52110
COB52120
COB52130
COB52140
COB52150
COB52160
COB52170
COB52180
COB52190
COB52200
COB52210
COB52220
COB52230
COB52240
COB52250

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IF(NH.GT.1) READ(12,10) (YH(I),FH(I),I=1,NH)
NG = N5
IF(NG.GT.1) READ(12,10) (YG(I),FG(I),I=1,NG)
NQ = N6
IF(NQ.GT.1) READ(12,10) (YQ(I),FQ(I),I=1,NQ)
IF(J1.LE.2) PRINT(11) = .TRUE.
GO TO 995

C
C INPUT FC: CARD GROUP 12, OUTPUT OPTIONS FOR CALCULATIONS
220 NCUT = N1
  NPCHAN = N2
  IF(N2.LT.1) GO TO 221
  READ(12,17) (IDAT($PRNC+I),I=1,N2)
  221 NPROD = N3
  NPHDE = N4
  IF(N3.LT.1) GO TO 222
  READ 17, (IDAT($PRNTR+I),I=1,N3)
  222 IF(N4.LT.1) GO TO 225
  READ 17, (IDAT($PRNTN+I),I=1,N4)
  225 GO TO 995
C CARD GROUP 20 READ DATA VIA ITHO
230 NUPRIN=N1
  CALL CARD20(NOPRIN)
  IF((ERROR.GT.0) GO TO 240
  GO TO 995

C
C INPUT DATA ERROR MESSAGE
240 WRITE(13,54)
  STOP

C
C END OF INPUT
C
C 250 RETURN
C
  1 FORMAT(7I5)
  2 FORMAT(11, 14, 15, 17A4)
  3 FORMAT(15HINPUT FOR CASE
    19H DATE 2A4,7H TIME 2A4,A1 )
  5 FORMAT(12F5.3)
  7 FORMAT(11,14,3E5.2,4(15,2E5.2))
  8 FORMAT ( 15/(12F5.3))
  9 FORMAT (6F10.0)
 10 FORMAT(12E5.0)
 11 FORMAT(11,14,2E5.2,6(15, E5.2))
 14 FORMAT(4E5.2,215,E5.2,15,4E5.2)
 17 FORMAT (36I2)
 41 FORMAT (15,7E10.5)
 42 FORMAT(8E10.5)
 52 FORMAT (15,6E12.6)
 54 FORMAT(//' INPUT DATA ERROR, THIS RUN STOPPED, CHECK INPUT')
 64 FORMAT(15,10E5.2)
 66 FORMAT (6( E5.2,15))
 67 FORMAT (15,E5.2,15,E5.2)
 68 FORMAT (10I5)
 79 FORMAT ( 9E5.2)

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COB52260
COB52270
COB52280
COB52290
COB52300
COB52310
COB52320
COB52330
COB52340
COB52350
COB52360
COB52370
COB52380
COB52390
COB52400
COB52410
COB52420
COB52430
COB52440
COB52450
COB52460
COB52470
COB52480
COB52490
COB52500
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COB52540
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COB52660
COB52670
COB52680
COB52690
COB52700
COB52710
COB52720
COB52730
COB52740
COB52750
COB52760
COB52770
COB52780
COB52790
COB52800

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1000 FORMAT(/, ' NORMAL COBRA INPUT DATA PRESENTATION',/)
1001 FORMAT(/, ' SIMILAR CHANNELS ALL CONNECTED EG.PWR',/)
1002 FORMAT(/, ' SIMILAR CHANNELS ALL SEPARATED EG.BWR',/)
END
SUBROUTINE MODEL(IPART,CARD,IPILE)
C
C
      IMPLICIT INTEGER (5)
      COMMON /COBRA1/ ABETA , AFLUX , ATOTAL, BBETA , DIA , DT , DX
1     ELEV , FERROR, FLO , FTN , GC , GK , GRID , HSURF , HF
2     HFG , HG , L2 , L3 , TERROR, IQP3 , ITERAT, J1 , J2
3     J3 , J4 , J5 , J6 , J7 , KOBBUG, KF , KIJ
4     NAFACT, NARAMP, NAX , NAXL , NBBC , NCHANL, NCHF , NDX , NF
5     NGAPS , NGRID , NGRIDT, NGTYPE, NGXL , NK , NODES , NODESF, NPROP
6     NRAMP , NRCD , NSCBC , NV , NVISCH, PI , PITCH , POWER , PREF
7     QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID, THETA , THICK
8     UF , VF , VFG , VG , Z
C
      COMMON /COBRA2/ AA(4) , AF(7) , AFAC(10,10) , AV(7) , AXIAL(30)
1     AX(10) , BB(4) , BK(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(30)
2     GAPXL(10) , GFACT(9,10) , GRIDXL(10) , HGAP(2) , HMF(30) , HHG(30)
3     IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9)
4     PI(30) , RCLAD(2) , RFUEL(2) , SSIGNA(30) , TCLAD(2) , UUF(30)
5     VVF(30) , VVG(30) , XQUAL(30) , Y(30) , Y(30) , Y(30)
C
C
      LOGICAL GRID
      REAL KIJ , KF , KKF , KCLAD , KFUEL
C
C
      COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MX
1     SS , SA , SAA , SAC , SALPHA, SAN , SANSWE, SB
2     SCCHAN, SCD , SCHFR , $CON , $COND , $CP , $D , $DC , $DFDX
3     SDHX , $DHYD , $DHYDN, $DIST , $DPOX , $DPK , $DUR , $DR , $F
4     $FACTO, $FDIV , $FINLE, $FLUX , $FMULT, $FOLD , $FSP , $FSPLI, $FAYLO, COBS3150
5     $IDFUE, $IDGAP, $IK , $JBDIL, $JK , $L , $LENGT, $LOCA , $LR , COBS3170
6     $MCHFR, $MCFRC, $MCFRR, $NTYPE, $NWRAP, $NWRPS, $P , $PERIM, $PH , COBS3180
7     $PHI , $PRITC, $PRNTR, $PRNTN, $PM , $PWR , $QC , $QF , $QPRIM , COBS3190
8     $QUAL , $RADIA, $RHO , $RHODL, $SP , $T , $TDUMY, $TINLE, $TROD , COBS3200
9     SU , $UH , $USAVE, $USTAR, $V , $VISC , $VISCW, $VP , $VPA , COBS3210
      A $W , $WOLD , $WP , $WSAVE, $X , $XCROS, $XA , $XB , $XPOLD
C
      COMMON DATA (1)
      LOGICAL LDAT(1)
      INTEGER IDAT(1)
      EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
C
C
      COMMON /LINKS/ DXA , ETIME , GIN , HIN , IB , IG , IN , ISAVE , JUMP , KASE , KT , MAKT
1     NDT , NDXP1 , NFUELT , NG , NH , NJUMP , NOUT , NP , NPCMAN , NPNODE , NP ROD , NQ , NR
2     NSKIPT , NSKIPTX , NTRIES , PEXIT , PHOTOT , SAVEDT , TIN , TTIME , ZZ
      COMMON /LINK9/ ENEH(400)
      COMMON /SAVMD/ N1 , N2 , N3 , N4 , N5 , N6 , N7 , N9
      DIMENSION CARD(20) , TAG(2)

```

COBS2810

COBS2820

COBS2830

COBS2840

COBS2850

COBS2860

COBS2870

COBS2880

COBS2890

COBS2900

COBS2910

COBS2920

COBS2930

COBS2940

COBS2950

COBS2960

COBS2970

COBS2980

COBS2990

COBS3000

COBS3010

COBS3020

COBS3030

COBS3040

COBS3050

COBS3060

COBS3070

COBS3080

COBS3090

COBS3100

COBS3110

COBS3120

COBS3130

COBS3140

COBS3150

COBS3160

COBS3170

COBS3180

COBS3190

COBS3200

COBS3210

COBS3220

COBS3230

COBS3240

COBS3250

COBS3260

COBS3270

COBS3280

COBS3290

COBS3300

COBS3310

COBS3320

COBS3330

COBS3340

COBS3350

C DATA TAG /4HW/GS, 4HW/GD /  
 C IPART=1 SET HYDRAULIC MODEL  
 C IPART=2 PRINT HYDRAULIC MODEL  
 C SAME AS NEXIN CODING  
 C  
 C PRESET MODEL IS CODED FIRST AND IS USED IF ALL N1-N7=0  
 C INDIVIDUAL PARTS OF MODEL MAY BE CHANGED BY  
 C SETTING ANY OF N1-N7 POSITIVE NON-ZERO  
 C  
 C IF (IPART.EQ.2) GO TO 30  
 C (N1) MIXING MODEL (CARD GROUP 10)  
 NSCBC=1  
 NBRC=1  
 J5=0  
 GK=0.0  
 ABETA=0.02  
 OBETA=0.0  
 C IF (IPILE.EQ.2) ABETA=0.0  
 C (N2) SINGLE PHASE FRICTION (CARD GROUP 2)  
 DD 4 J=1,4  
 AA(1)=0.184  
 BB(1)=-0.2  
 4 CC(1)=0.0  
 NVISCW=0  
 C (N3) TWO PHASE FRICTION (CARD GROUP 2)  
 J4=0  
 C (N4) VOID FRACTION (CARD GROUP 2)  
 J2=0  
 J3=0  
 C (N5) FLOW DIVISION AT INLET (CARD GROUP 11)  
 IG = 0  
 C (N6) CONSTANTS (CARD GROUP 9)  
 MCHF = 0  
 KIJ=0.5  
 FTM=0.0  
 SL=0.5  
 THETA=0.0  
 ELEV=1.0  
 C (N7) ITERATION (CARD GROUP 9)  
 NTRIES=20  
 FERROR=0.001  
 C (N8) PHYSICAL PROPERTIES (CARD GROUP 1)  
 NPROP=0  
 C (N9) COUPLING PARAMETER FOR ENTHALPY EXCHANGE  
 DO 3201 K=1,NK  
 3201 ENEH(K)=1.0  
 C  
 C READ(12,1001) CARD,N1,N2,N3,N4,N5,N6,N7,NPROP,N9  
 C WRITE(13,1009) CARD  
 C IF((N1+N2+N3+N4+N5+N6+N7+NPROP+N9).EQ.0) RETURN  
 C  
 C IF(N1.EQ.0) GO TO 6  
 C IF (N1.EQ.2) NSCBC=2

COB53360  
 COB53370  
 COB53380  
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 COB53890  
 COB53900

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IF (N1.EQ.3) NSCBC=4
IF (IPILE.EQ.2) WRITE(13,1010)
IF (N1.LT.3) READ(12,1002) CARD,ABETA,BBETA
IF (N1.EQ.1) WRITE(13,1011) CARD
6 IF (N2.EQ.0) GO TO 8
  READ(12,1003) CARD,NVISCW,(AA(1),BB(1),CC(1),I=1,4)
  WRITE(13,1012) CARD
8 IF (N3.EQ.0) GO TO 10
  READ(12,1001) CARD,J4
  WRITE(13,1013) CARD
IF (N4.LE.4) GO TO 10
  READ(12,1003) CARD,NF,AF
  WRITE(13,1014) CARD
10 IF (N4.EQ.0) GO TO 12
  READ(12,1001) CARD,J2,J3
  WRITE(13,1015) CARD
IF (J3.LE.4) GO TO 12
  READ(12,1003) CARD,NV,AV
  WRITE(13,1014) CARD
12 IF (N5.EQ.0) GO TO 16
  READ(12,1001) CARD,IG
  WRITE(13,1016) CARD
IF (IG.LE.1) GO TO 16
  CALL READINI(1,RCHANL,DATA($FINLE+1),CARD,CARD,1)
16 IF (N6.EQ.0) GO TO 18
  READ(12,1003) CARD,NCHF,K1J,FTM,SL,THETA
  WRITE(13,1017) CARD
  ELEV=COS(THETA*PI/180.0)
18 IF (N7.EQ.0) GO TO 20
  READ(12,1003) CARD,NTRIES,ERROR
  WRITE(13,1018) CARD
20 IF (NPRDP.EQ.0) GO TO 22
  READ(12,1004) CARD,NPROP,N,PH,PP(2)
  WRITE(13,1019) CARD
  PP(1) = PH
  IF (N.LE.1) GO TO 22
  PP(1) = 10.0
  IF (PH.LT.200.0) GO TO 22
  R = 0.01*PH
  PP(1) = 6.0*R*R*(R-1.35)/(R-0.35)
22 CONTINUE
  M=1
  IF (N9.EQ.0) GO TO 3206
3204 NM=MINO((M+13),NK)
  READ(12,3202) CARD,(ENEH(K),K=M,NM)
3202 FORPAT(20A4,T1,14E5.0)
  WRITE(13,3203) CARD
3203 FORMAT(' COUPLING FACTOR NH',10X,'***',20A4,'*** MODEL')
  M=MNH+1
3206 CONTINUE
  IF (M.LE.NK) GO TO 3204
  RETURN
C
C IPART = 2. PRINT MODEL
C 30 WRITE(13,1061)

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COB53910  
COB53920  
COB53930  
COB53940  
COB53950  
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COB53990  
COB54000  
COB54010  
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COB54030  
COB54040  
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COB54350  
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COB54370  
COB54380  
COB54390  
COB54400  
COB54410  
COB54420  
COB54430  
COB54440  
COB54450





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6 $MCHFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP,$NHRPS,$P,$PERM,$PH,$COD555G0
7 $PHI,$PRNTC,$PRNTR,$PRNTN,$PW,$PWRP,$Q,$SQF,$QPRM,$QPRIM,$COD55570
8 $QUAL,$RADIA,$RHO,$RHOOOL,$SSP,$ST,$STDUY,$STINLE,$STROD,$COD55580
9 $SU,$SUH,$SUSAVE,$SUSTAR,$SV,$SVISC,$SVISCH,$SVP,$SVA,$COD55590
A $W,$WOLD,$WP,$WSAVE,$X,$XCROS,$XA,$XBB,$XAPOLD,$COD55600
      $COD55610
C      COMMON DATA(1)
      LOGICAL LDAT(1)
      INTEGER IDAT(1)
      EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
C
C      COMMON/LINK2/CROSS(6),DATE(2),FG(30),FH(30),FP(30),FQ(30),FM(9),
1 JMI(9),OUTPUT(10),PRINT(12),TEXT(17),TIME(3),VG(30),VH(30),VP(30),
2 YQ(30)
C      COMMON/LINK3/DXX,ETIME,GIN,MIN,I8,IG,IN,ISAVE,JUMP,KA$E,KT,MAXT,
1 IOT,NDAP1,NFUELT,NG,NH,NJUMP,NOUT,NP,NPCHAN,NPNODE,NPROD,NQ,NR,
2 NSKIPT,NSKIPIX,NTRIES,PEXIT,PHOTOT,SAVEDT,TIN,TTIME,ZZ
      DIMENSION CARD(20)
C
C      SET PRINTING PARAMETERS
C      FOR INPRIN
C      IF (J1.GT.1) GO TO 4
      DD 2 I=1,11
2      PRINT(1) = .TRUE.
      PRINT(5) = .FALSE.
      PRINT(6) = .FALSE.
C
C      FOR CALC (CARD GROUP 9)
4      READ (12,1001) CARD, KDEBUG
      WRITE (13,1002) CARD
C
C      FOR EXPRIN (CARD GROUPS 9, 12)
C      READ (12,1001) CARD,NSKIPIX, NSKIPT, NOUT, NPCHAN, NPROD, NPNODE
      WRITE (13,1003) CARD
C      NSKIPIX. EVERY NSKIPIX AXIAL STEP PRINTED. (0 = 1)
C      NSKIPT. EVERY NSKIPT TIME STEP PRINTED. (0 = 1)
C      NOUT = 0-3 FOR PRINTING (0) CHANNEL ONLY, (1) CHAN + CROSS FLOWS,
C      (2) CHAN + FUEL TEMP, (3) CHAN + C-F + FUEL TEMP
C      NPCHAN = 0, ALL CHAN PRINTED. .GT.0 READ CHANS REQD.
C      NPROD, NPNODE AS NPCHAN BUT FOR RODS AND NODES.
C      IF (NSKIPIX.LT.1) NSKIPIX = 1
C      IF (NSKIPT.LT.1) NSKIPT = 1
C      IF (NPCHAN.LT.1) GO TO 6
      MROSI=1
7209 MMJAVI=MING((MROSI+13),NPCHAN)
      READ(12,1001) CARD,(IDAT($PRNTR+1),I=MROSI,MMJAVI)
      WRITE(13,1004) CARD
      MROSI=MMJAVI+1
      IF(MROSI.LE.NPCHAN) GO TO 7209
6      IF(NPROD.LT.1) GO TO 8
      MROSI=1
8209 MMJAVI=MING((MROSI+13),NPROD)
      READ (12,1001)CARD,(IDAT($PRNTR+1),I=MROSI,MMJAVI)
      WRITE (13,1006) CARD

```





CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

```

DBHTDUMY ,BHINLET ,BHITROD ,BHUSCH ,BHUSAVE ,BHUSAVE ,COB56600
GONUSTAR ,BHV ,BHVISC ,BHVISCW ,BHVP ,BHVPA ,COB56670
FBHW ,BHWOLD ,BHWP ,BHWSAVE ,BHX ,BHXCROSS ,COB56680
GBHA ,BHB ,BXPOLD ,COB56690
INTEGER STYPE(97) /7*1,2,18*1,3,15*1,7*2,1,2*2,1,5*2,4*1,3*2,
1 32*1/
END
SUBROUTINE CORE
IMPLICIT INTEGER ($)
COMMON DATA(1)
COMMON /COBRA3/ MA,MC,MG,MN,MR,MS,MX,SS$,SORG(97)
INTEGER SLX(97)
COMMON /COBRA5/ $NAMES,SLX,STYPE
DIMENSION STYPE(97)
REAL*8 $NAMES(97)
MA = 1
MS = 1
IF (MG.LE.0) MG=1
IF (MN.LE.0) MN=1
C $*****
$$$=97
DO 100 I=1, $$$
100 SLX(I)=MC
SLX( 2)=1
SLX( 6)=MG
SLX( 7)=MG
SLX( 8)=MR*MX
SLX( 9)=MC*5
SLX(10)=MR*MX
SLX(12)=MG
SLX(14)=MR
SLX(20)=MC*4
SLX(23)=MG
SLX(24)=MR
SLX(25)=MC*MX
SLX(26)=MG
SLX(27)=MG
SLX(29)=MR*MX
SLX(31)=MC*MX
SLX(34)=MG*5
SLX(35)=MG
SLX(36)=MG
SLX(37)=MG*4
SLX(38)=MC*MX
SLX(41)=MC*MX
SLX(44)=MR
SLX(45)=MG
SLX(46)=MG
SLX(48)=MG
SLX(49)=MC*4
SLX(50)=MG
SLX(51)=MG*14
SLX(52)=MR*6
SLX(53)=MX
SLX(54)=MX
COB56700
COB56710
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COB57200

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$IX(55)=MX
$IX(59)=MC*MX
$IX(62)=MR*6
$IX(64)=MR
$IX(65)=MN
$IX(67)=MC*MR
$IX(68)=MC*MX
$IX(69)=MC*MX
$IX(72)=MR
$IX(73)=MC*MX
$IX(74)=MC*MX
$IX(75)=MG*MX
C
PROVIDE SPACE FOR SP IN BWR ITERATION.
IF ($IX(75) .LT.3*MC) $IX(75) = 3*MC
$IX(77)=MN
$IX(79)=MN*MR*MX
$IX(82)=MG
$IX(83)=MG
$IX(89)=MG*MX
$IX(90)=MG*MX
$IX(91)=MG
$IX(92)=MG
$IX(93)=MX
$IX(94)=MG*6
$IX(95)=3*MR
$IX(96)=MN
C *****
$IX(97)=MC*MX
SORG(1)=1
SLXA=0
DO 110 I=1, $$$
SLXA=$IX+$IX(I)
IF(I.GT.1) SORG(1)=SORG(1-1)+$IX(I-1)
110 CONTINUE
KS=1
CC
CC KMAX IN SUBROUTINE CORE EQUALS
CC LENGTH OF DATA ARRAY GIVEN BELOW
CC
KMAX=80000
KFREE=KS
KTOP = KS + KMAX - 1
KS=KS+MOD(KS+1,2)
IF(KMAX.LT.$LXA) GO TO 902
DO 300 K=KS, KTOP
300 DATA(K) = 0.0
DO 400 N=1, $$$
400 SORG(N)=SORG(N)+KS-1
RETURN
C
ENTRY CORE2(MSP, MKP)
NK=NKP
MS=MSP
MA=NK*HS
$IX(2)=MA

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COB57740  
COB57750



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7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK , COB58310
8 UF ,VFG ,VG ,Z , COB58320
C COB58330
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
1 AXL(10), BB(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2 CAPXL(10), GFACT(9,10), GRIDXL(10), HGAP(2), HHF(30), HHG(30),
3 IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9),
4 PP(30), RCLAD(2), RFUEL(2), SSGMA(30), TCLAD(2), UUF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30)
C COB58380
C COB58390
C COB58400
C COB58410
C COB58420
C COB58430
C COB58440
C COB58450
COMMON /COBRA3/ MA ,MC ,MG ,MN ,MR ,MS ,MK ,
1 $$S ,SA ,$AAA ,$AC ,$ALPHA,$AN ,$ANSWE,$B ,
1 SCCHAN,$CD ,$CHFR ,$CON ,$COND ,$CP ,SD ,SDC ,$DFDX ,
2 $DHDX ,SDH/D ,SDHYDN,$DIST ,SDPK ,SDUR ,SDR ,SF ,
3 SFACD,$FDIV ,$FINLE,$FLUX ,$FMULT,$FOLD ,$FSP ,$FSPLI,$FKFLO ,
4 SGAP ,$GAPN ,$GAPS ,SH ,SHFILM,$HINLE,$HOLD ,$HPERI,$IDARE ,
5 $IDFUE,$IDGAP,$IK ,SUBOIL,$UK ,SLC ,$LENGT,$LOCA ,$LR ,
6 $MCHFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP,$NWRPS,$P ,
7 $PHI ,$PRNTC,$PRNTR,$PRNTN,$PW ,$PWRF ,SOC ,$OF ,$OPRIM ,
8 $QUAL ,$RADIA,$RHO ,$RHODL,$SP ,ST ,$TDUMY,$TINLE,$TROD ,
9 $U ,SUH ,$USAVE ,$USTAR,$V ,VISC ,$VISCW,$VP ,
A $W ,$WOLD,$WP ,$WSAVE,$X ,$XCROSS,$XA ,$$B ,$XPOLD
COMMON/LINK9/ENEH(400)
C COB58580
C COB58590
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C COB58620
C COB58630
C COB58640
C COB58650
C COB58660
C COB58670
C COB58680
C COB58690
C COB58700
C COB58710
C COB58720
C COB58730
C COB58740
C COB58750
C COB58760
C COB58770
C COB58780
C COB58790
C COB58800
C COB58810
C COB58820
C COB58830
C COB58840
C COB58850
PART 1, CALCULATE DH/DX FOR STEADY STATE AT K AND T.
100 DO 120 I=1,NCHANL
120 DATA($DHDX+I)=0.
IF (I*FILE.EQ.2) GO TO 185
DO 160 K=1,NK
I=IDAT($JK+K)
L=IDAT($JK+K)
MV=(DATA($H+L+MC*(J-1))-DATA($H+L+MC*(J-1)))
IF (DATA($W+K+MG*(J-1)).LT.0.) GO TO 140
HNL = 0.0
HNL= DATA($W+K+MG*(J-1)) * MV
GO TO 160
140 HNL= DATA($W+K+MG*(J-1)) * MV
HNL = 0.0

```

```

160 CONTINUE
  DATA(SDRDX+I)=DATA(SDRDX+I)+HWI-WV+DATA(SVP+K)/ENEH(K)-(DATA(ST+I)
1  -DATA(ST+L))+DATA(SCOND+K)
  DATA(SDHD+L)=DATA(SDHD+L)+HML+VW+DATA(SWP+K)/ENEH(K)+(DATA(ST+I)
1  -DATA(ST+L))+DATA(SCOND+K)
180 CONTINUE
185 DO 190 I=1,NCHANL
190 DATA(SDHD+I)=(DATA(SDRDX+I)+DATA(SQPRIM+I)+DATA(SQC+I+MC+J))/DX
1  /DATA(SF+I+MC*(J-1))
  GO TO 500
C
C PART 2, CALCULATE DF/DX FOR STEADY STATE AT X AND Y
200 DO 220 I=1,NCHANL
220 DATA(SDFDX+I)=0.
  IF (IPILE.EQ.2) GO TO 500
  DO 240 K=1,NK
  I = IDAT(SIK+K)
  L = IDAT(SJK+K)
  DATA(SDFDX+I)=DATA(SDFDX+I)-DATA(SW+K+MG*(J-1))
240 DATA(SDFDX+L)=DATA(SDFDX+L)+DATA(SW+K+MG*(J-1))
  GO TO 500
C
C PART 3, CALCULATE DF/DX WITHOUT W
300 DO 302 I=1,NCHANL
302 DATA(SDPDX+I)=0.
  IF (FIRAMP.LE.0) GO TO 306
  IF (IPILE.EQ.2) GO TO 306
  DO 304 K=1,NK
  I=IDAT(SIK+K)
  L=IDAT(SJK+K)
  KW=(DATA(SU+I)-DATA(SU+L))+DATA(SWP+K)
  DATA(SDPDX+I)=DATA(SDPDX+I)+WV
  DATA(SDPDX+L)=DATA(SDPDX+L)-WV
304 CONTINUE
306 DO 390 I=1,NCHANL
  SAVE=0.5*DATA(SFSP+I)+DATA(SFMULT+I)+DATA(SV+I)/DATA(SRHO+I)
1  +(DATA(SVP+I)/DATA(SA+I)-DATA(SVPA+I))*DATA(SA+I)/DX
  IF(.NOT.GRID) GO TO 310
  IF(NRAMP.LE.0) GO TO 1000
  DUMY = FLOAT(ITERAT)/FLOAT(NRAMP)
  IF(DUMY.GT.1.) DUMY = 1.
  SAVE=SAVE+.5*DUMY*DATA(SCD+I+MC*(NGTYPE-1))+DATA(SVP+I)/DX
310 DATA(SDPK+I)=SAVE/(DATA(SA+I)+DATA(SA+1))
  JJ = J/I
  IF (J.GT.1) GO TO 382
  JJ = 1
382 FLOWSQ=ABS(DATA(SF+I+MC*(JJ-1)))*
  DATA(SF+I+MC*(JJ-1))
C
C JK INSERT
  IF(IPILE.EQ.2) FLOWSQ= DATA(SF+I+MC*(J-1))+2
  DATA(SDPDX+I)=DATA(SDPK+I)+FLOWSQ/GC-DATA(SRHO+I+MC*(J-1))*
1  ELEV-DATA(SDPDX+I)+FTM/(DATA(SA+I)*GC)
  IF(DT.GT.100.) GO TO 390
  T R INSERT
C
RHODIF=DATA(SRHO+I+MC*(J-1))-DATA(SRHOOL+I+MC*(J-1))

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COB50860  
COB50870  
COB50880  
COB50890  
COB50900  
COB50910  
COB50920  
COB50930  
COB50940  
COB50950  
COB50960  
COB50970  
COB50980  
COB50990  
COB59100  
COB59110  
COB59120  
COB59130  
COB59140  
COB59150  
COB59160  
COB59170  
COB59180  
COB59190  
COB59200  
COB59210  
COB59220  
COB59230  
COB59240  
COB59250  
COB59260  
COB59270  
COB59280  
COB59290  
COB59300  
COB59310  
COB59320  
COB59330  
COB59340  
COB59350  
COB59360  
COB59370  
COB59380  
COB59390  
COB59400

```

C
RHDDT=RHDDIF/DT
JK INSERT
IF (IPILE.NE.2) GO TO 385
DATA(SOPDX+I)=DATA(SOPDX+I)+RHDDT/GC*2.*DATA(SU+I)
1 +(DATA(SFOLD+I+MC*(J-1))-DATA(SF+I+MC*(J-1)))/DATA(SA+I)/DT/GC
GO TO 390
385 DATA(SOPDX+I)=DATA(SOPDX+I)+RHDDT/GC*(2.*DATA(SU+I)+DX/DT
1 +DATA(SOPK+I)-ABS(DATA(SF+I+MC*(JMI-1))+DATA(SF+I+MC*(J-1))))*
2 DATA(SA+I)+DX)
3 (DATA(SFOLD+I+MC*(J-1))-DATA(SF+I+MC*(JMI-1))
390 CONTINUE
GO TO 500
C
C PART 4, CALCULATE DP/DX WITH M
400 IF (J.EQ.1) GO TO 500
DO 410 I=1,NCHANL
410 DATA(DHDX+I)=0.
IF (IPILE.EQ.2) GO TO 425
DO 420 K=1,NK
I=IDAT ($IK+K)
L=IDAT ($JK+K)
DATA(DHDX+I)=DATA(DHDX+I)+((2.*DATA(SU+I)-DATA(SUSTAR+K)+DX/DT)
1 /DATA(SA+I)+DATA(SOPK+I)-ABS(DATA(SF+I+MC*(JMI-1))+
2 DATA(SF+I+MC*(J-1)))+DX)*DATA($H+K+MG*(J-1))
DATA(DHDX+I)=DATA(DHDX+L)-((2.*DATA(SU+L)-DATA(SUSTAR+K)+DX/DT)
1 /DATA(SA+L)+DATA(SOPK+L)-ABS(DATA(SF+L+MC*(JMI-1))+
2 DATA(SF+L+MC*(J-1)))+DX)*DATA($H+K+MG*(J-1))
420 CONTINUE
425 DO 430 I=1,NCHANL
430 DATA(SOPDX+I)=DATA(SOPDX+I)+DATA(SDHDX+I)/GC
C
500 CONTINUE
RETURN
1000 IERROR = 2
RETURN
END
SUBROUTINE HEAT(J)
CALCULATE THE HEAT INPUT TO EACH SUBCHANNEL AT POSITION J.
C IF NODES GREATER THAN ZERO, CALCULATE HEAT INPUT USING THERMAL
C CONDUCTION. OTHERWISE HEAT INPUT IS DEFINED BY HEAT GENERATION.
C POWER = AVERAGE INTERNAL HEAT GENERATION.
C
C
C IMPLICIT INTEGER ($)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , HF
2 HFG , HG , I2 , I3 , IERROR , IQP3 , ITERAT , J1 , J2
3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ
4 NFACT , NARAMP , MAX , MAXL , NABC , NCHAN , NCHF , NDX , NF
5 NGAPS , NGRID , NGRID , NGTYPE , NGXL , NK , NODES , NODESF , NPROP
6 NRAMP , NROD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
8 UF , VF , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) .

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1 AX(10), BX(4), BX(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2), COB59960
2 GAPXL(10), GFAC(9,10), GRIDXL(10), HGAP(2), HMF(30), MHG(30), COB59970
3 IGRID(10), KCLAD(2), KFUEL(2), KKF(30), NCH(10), NGAP(9), COB59980
4 PP(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30), COB59990
5 VVF(30), VVG(30), XQUAL(30), Y(30), Y(30), TT(30) COB60000
C
C
LOGICAL GRID
REAL KIJ, KF, KMF, KCLAD, KFUEL
COB60010
COB60020
COB60030
COB60040
COB60050
COB60060
COMMON /COBRA3/ MA, MC, MG, MN, MR, MS, MX
1 $$$, SA, $AAA, $AC, $ALPHA, $AN, $ANSWE, $B, COB60070
1 SCCHAN, $CD, $CHFR, $CON, $COND, $CP, $D, $DC, $DFDX, COB60080
2 $DHDX, $DHYD, $DHYDN, $DIST, $DPDX, $DPK, $DUR, $DR, $F, COB60090
3 $FACTO, $FDIV, $FINLE, $FLUX, $FMULT, $FOLD, $FSP, $FSPLI, $FXFLO, COB60110
4 $GAP, $GAPN, $GAPS, $H, $HFILM, $HINLE, $HOLD, $HPERI, $IDARE, COB60120
5 $IDFUE, $IDGAP, $IK, $JBOIL, $JK, $LC, $LENGT, $LOCA, $LR, COB60130
6 $MCHFR, $MCFRC, $MCFRR, $NTYPE, $NWRAP, $NWRPS, $P, $PERIM, $PH, COB60140
7 $PHI, $PRITC, $SPRTR, $SPRTN, $SPW, $SPWF, $SOC, $OF, $OPRIM, COB60150
8 $QUAL, $RADIO, $RHO, $RHOO, $SP, $ST, $STUUMY, $TINLE, $TROD, COB60160
9 $U, $UH, $USAVE, $USTAR, $V, $VISC, $VISCW, $VP, $VPA, COB60170
A $W, $WOLD, $WP, $WSAVE, $X, $XCROS, $$A, $$B, $XPOLD, COB60180
COB60190
C
COMMON /LINK3/ DIX, ETIME, GIN, HIN, I0, IG, IN, ISAVE, JUMP, KASE, KT, MAXT,
1 NDT, NDAP1, NFUEL, NG, NH, NJUMP, NOUT, NP, NPCHAN, NPNODE, NPROD, NQ, NR,
2 NSKIPT, NSKIPIX, NTRIES, PEXIT, PHTOT, SAVEDT, TIN, TTIME, ZZ
C
COMMON /ERDATA/ BURN, CPR, E, FEB, EPSF, EXPR, F, PRESS, FPUO2, FRAC, FTD,
1 GRHX(4), GRGH, PGAS, RADR, RDEL, THC, THG
C
COMMON /LINK4/ IFRM, INTM, IPROP, NCC, NCF, NDM1, NDS, NGP
C
COMMON /TIMEST/ NT
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1), IDAT(1), LDAT(1))
C
EQUIVALENCE (NCHAN, NCHAN)
C
IPILE = J7
NPI = NODESF+1
C BYPASS THE HEAT FLUX CALCULATION IF BEYOND THE FIRST ITERATION AND
C IF FUEL TEMPERATURES ARE NOT TO BE CALCULATED.
C IF (ITERAT.GT.1 .AND. NODESF.LT.1) GO TO 60
C BYPASS THE HEAT FLUX CALCULATION USING THE FUEL TEMPERATURE MODEL
C IF BEYOND THE FIRST ITERATION, AND IF FUEL TEMPERATURES HAVE BEEN
C CALCULATED AND IF A TRANSIENT CALCULATION IS BEING PERFORMED.
C IF (ITERAT.GT.1 .AND. NODESF.GT.0 .AND. DT.LT.100.) GO TO 60
C IF (IOP3.LE.1) GO TO 170
CALL CURVE(QAX, (DATA($X+J)-DX*0.5)/Z, AXIAL, Y, MAX, IERROR, 1)
170 CONTINUE
C DETERMINE THE HEAT FLUX FROM EACH ROD.

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COB60510
COB60520
COB60530
COB60540
COB60550
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COB60670
COB60680
COB60690
COB60700
COB60710
COB60720
COB60730
COB60740
COB60750
COB60760
COB60770
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COB60950
COB60960
COB60970
COB60980
COB60990
COB61000
COB61010
COB61020
COB61030
COB61040
COB61050

DO 50 N=1,NROD
IF(IOP3.LE.1) GO TO 160
C CALCULATE FORCED HEAT FLUX FROM EACH ROD.
DATA($FLUX+N*MR*(J-1))=AFUX+DATA($RADIA+N)*QAX*POWER/.0038
GO TO 150
160 K-IDAT($DFUE+N)
IF(K.EQ.1) DATA($FLUX+N*MR*(J-1))=DATA($QF+N*MC*(J-1))
1 / (DATA($HPERI+H)*DX)
C IF(K.EQ.2) DATA($FLUX+N*MR*(J-1))=DATA($QF+N*MC*(J-1))
1 / (DATA($HPERI+H)*DX)
150 CONTINUE
IF(MODESF.LT.1) GO TO 50
C CORRECT HEAT FLUX FOR THERMAL CAPACITY USING TRANSIENT FUEL MODEL.
C CALCULATE AVERAGE FLUID TEMPERATURE, HEAT TRANSFER COEFFICIENT.
CC START OF LOOP FOR OBTAINING STEADY STATE FUEL ROD TEMPERATURES.
DO 40 INN=1,50
SAVE = 0.
TFLUID = 0.
HSURF = 0.
IF (IPILE.EQ.0) GO TO 6
TFLUID=DATA($T+N)
CALL HTRAM(N,N,J-1,HSURF,TFLUID,IHTM,NT)
IF (IERROR.GT.1) RETURN
GO TO 7
6 DO 9 L=1,6
IF(IDAT($LR+N*MR*(L-1))) 9,9,10
I=IDAT($LR+N*MR*(L-1))
DUMY=DATA($PHI+R+MR*(L-1))
SAVE = SAVE + DUMY
TFLUID=TFLUID+DATA($T+I)+DUMY
CALL HTRAM(N,I,J-1,HTC,DATA($T+I),IHTM,NT)
HSURF = HSURF + DUMY*HTC
IF(IERROR.GT.1) RETURN
9 CONTINUE
IF(SAVE.LE.0.) GO TO 1000
TFLUID = TFLUID/SAVE
HSURF = HSURF/SAVE
C CALCULATE FUEL TEMPERATURE
DO 8 I=1,NP1
DATA($TDUMY+I)=DATA($TROD+I+MN*(N-1+MR*(J-1)))
IF(IFRM.EQ.0) GO TO 20
QP=DATA($FLUX+N*MR*(J-1))+4.*DATA($D+N)/(DFUEL(1)**2)
CALL TEMFR(DATA($TDUMY+I),DT,N,TFLUID,HGAP(1),HSURF,OP,INN,NT)
GO TO 22
20 CALL TEMPI(DATA($TDUMY+I),DT,N,J,DATA($SA+I),DATA($SB+I))
IF(IERROR.GT.1) RETURN
22 DO 24 I=1,NP1
24 DATA($TROD+I+MN*(N-1+MR*(J-1)))=DATA($TDUMY+I)
IF (IHTM.EQ.0.AND.IPROP.EQ.0) GO TO 45
IF (INT.GT.1) GO TO 45
IF(INI.LT.2) GO TO 40
IF (ABS(DATA($TDUMY+I))-FOLD).GT.EPSF) GO TO 40

```



```

GO TO 45
40 FTLD=DATA(STDUNY+1)
WRITE(13,55) N,J
55 FORMAT(1H,' FUEL TEMPERATURES FAILED TO CONVERGE IN FUEL ROD ',
1 13,' AT AXIAL LEVEL ',13,'. MAXIMUM ITERATIONS = 50.')
```

GO TO 1000

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45 DATA($FLUX+N*MR*(J-1))-HSURF*(DATA($TROD+NP1+MN*(N-1+MR*(J-1)))
1 -TFLUID)
50 CONTINUE
60 IF (IPILE.EQ.0) GO TO 70
IF (NODESF.LI.1) GO TO 66
DO 65 I=1,NCHANL
JK INSERT
C
CALL HTRAN(I,I,J-1,HSURF,DATA($T+I),IHTM,NT)
65 DATA($OPRIM+I)=DATA($PWRFI+MC*(I-1))*PI*DATA($D+I)
1 *HSURF*(DATA($TROD+NP1+MN*(I-1+MR*(J-1)))-DATA($T+I))
RETURN
66 DO 68 I=1,NCHANL
68 DATA($OPRIM+I)=DATA($PWRFI+MC*(I-1))*PI*DATA($D+I)*
1 DATA($FLUX+I+MR*(J-1))
RETURN
C
CALCULATE HEAT INPUT TO EACH CHANNEL.
70 DO 100 I=1,NCHANL
SAVE = 0.
DO 90 N=1,NROD
DUMY=DATA($PWRFI+MC*(N-1))
IF(DUMY.GT.0.) SAVE=SAVE+DUMY*DATA($FLUX+N*MR*(J-1))*PI*DATA($D+N*COBE+1,20
1 )
90 CONTINUE
100 DATA($OPRIM+I)=SAVE
RETURN
1000 ERROR = 14
RETURN
END
SUBROUTINE HTRAN(N,I,JJ,MTC,TLIQ,IHTM,NT)
C
C CALCULATES ROD-TO-COOLANT HEAT TRANSFER COEFFICIENT, HTC
C
IMPLICIT INTEGER(S)
COMMON/PSAVE/P,ROV,ROL,TSAT
C
COMMON /COBRA1/ ABETA , AFLUX , ATOTAL,BBETA , DIA , DT , DX
1 ELEV , FERROR,FLO , FTM , GC , GK , GRID ,HSURF , HF
2 HFG , HG , I2 , I3 , IERROR,IQ3 , ITERAT , J1 , J2
3 J3 , J4 , J5 , J6 , J7 , KDEBUG,KF , KIJ ,
4 NAFCT,NARAMP,NAX , NAXL , NBBC , NCHAN ,NCHF ,NDX , NF
5 NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGXL ,NK , NODES ,NODESF ,NPROP
6 NRAMP ,NRD ,NSCBC ,NV , NVISCH,PI , PITCH ,POWER ,PREF , COB61530
7 OAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID,THETA , THICK
8 UF , VF , VFG , VG , Z
C
C
C
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MX

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1   $$$ ,SA ,SAA ,SAC ,SALPHA,SAN ,SANSWE,$B ,COB61610
1   $CCHAN,$CD ,SCHFR,$CON ,SCOND ,SCP ,SD ,SDC ,SDFDA ,COB61620
2   $DHDX,$DHYD,$SHYDN,$DIST,$DPK,$DUR,$DR,$SF ,COB61630
3   $FACTO,$FDIV,$FINLE,$FLUX,$FMULT,$FOLD,$FSP,$FPLI,$FAFLO,COB61640
4   $GAPN,$GAPN,$GAPS,$H,$HFILM,$HINLE,$HOLD,$HPERI,$IDARE,COB61650
5   $IDFUE,$IDGAP,$IK,$JBOIL,$JK,$LC,$LENGT,$LOCA,$LR ,COB61660
6   $MCFR,$MCFRC,$MCFRR,$NTYPE,$NWRAP,$NWRPS,$P,$PERIM,$PH ,COB61670
7   $PHI,$SPRNTC,$SPRNTN,$SPM,$SPWRF,$SQ,$SQF,$SQPRIM,COB61680
8   $QUAL,$RADIA,$RHO,$RHODL,$SP,$ST,$STDUWY,$TINLE,$STROD ,COB61690
9   $SU,$SUH,$SUSAVE,$SUSTAR,$V,$VISC,$VISCM,$VVP ,COB61700
A   $W,$WOLD,$WP,$WSAVE,$X,$XCROS,$XA,$XB,$XPOLD ,COB61710
C   COMMON DATA(1) ,COB61720
    LOGICAL LDAT(1) ,COB61730
    INTEGER IDAT(1) ,COB61740
    COB61750
C   CHOICE BETWEEN OLD AND NEW HEAT TRANSFER MODELS MADE HERE
    IF (IHTM.EQ.0) GO TO 300
    NP1=NODESF+1
CC  VALUES CONVERTED TO SI UNITS FOR USE BY HTCOR
CC
C   TW=TCON( DATA($TROD+NP1*MM*(N-1+MR*(JU))) )
C   LOW WALL TEMP. INDICATES THAT ROD TEMP. NOT YET
C   CALCULATED - SO OLD HEAT TRANSFER MODEL USED.
    IF (TW.LT.280.) GO TO 300
    TL=TCON(TL1Q)
    TV=TL
    XK=DATA($QUAL+1)
    ALP=DATA($ALPHA+1)
    IF (XX.LE.0.) VL=.304B*DATA($F+I+MC*(JU-1))/
    1 (DATA($RHO+I+MC*(JU-1))*(1.-ALP)+DATA($A+I))
    IF (XX.GT.0.) VL=.304B*((DATA($F+I+MC*(JU-1))
    1 (1.-XX))/(RHOF*(1.-ALP)+DATA($A+I)))
    VV=VL
    IF (XX.GT.0.) VV=.304B*DATA($F+I+MC*(JU-1))*XX/
    1 (RHOF+ALP*DATA($A+I))
    HD=.304B*DATA($DHYD+1)
    ROV=DCDN(RHOG)
    ROL=DCDN(RHOF)
C   CONVERT PRESSURE FROM PSI TO N/M**2
C
C   P=6.893E3*(PREF)
C
C   NO CHF CHECK IN HTCOR IF NT AND ITERAT BOTH EQUAL ONE
C   BECAUSE START OF BOILING INDICATORS WILL NOT
C   BE SET YET IN THIS CASE
C
    NHTM=IHTM
    IF (NT.EQ.1.AND.ITERAT.EQ.1) NHTM=1
C
    CALL HTCOR(IDUM1,QV,QL,HVFC,HLNB,HLFC,TW,TL,TV,P,ALP,XX,
    COB62130
    COB62140
    COB62150

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3 SFACD,SFDDV,SFINLE,SFLUX,SFMULT,SFOLD,SFSP,SFSPLI,SFXFLO,C0B62710
4 SGAP,SGAPN,SGAPS,SH,SHFILM,SHINLE,SHOLD,SHPERI,SIDARE,C0B62720
5 SIDFUE,SIDGAP,SIK,SUBOIL,SJK,SLENGT,SLOCA,SLR,C0B62730
6 SMCFR,SMCFRFC,SMCFR,SNTYPE,SNWRAP,SNWRPS,SP,SPERIM,SPH,C0B62740
7 SPHI,SPRINTC,SPRINTR,SPRINT,SPW,SPWRF,SOC,SOF,SOPRIM,C0B62750
8 SQUAL,SRADIA,SRHO,SRHOOL,SSP,ST,STDUMY,STINLE,STRD,C0B62760
9 SU,SUH,SUSAVE,SUSTAR,$V,$VISC,$VISCH,$VPA,C0B62770
A SW,SWOLD,SWP,SWSAVE,$X,$XCROSS,$SA,$SB,$XPOLD,C0B62780
C0B62790
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
EQUIVALENCE (NCHAN,NCHANL)
C
DIMENSION AAA(1)
1 FORMAT('ERROR DETECTED IN SUBROUTINE SCHEME AT NODE',I3,
1,X',E10.5,FEET',/,'CALCULATION FOR THIS CASE STOPPED')
2 FORMAT(' NODE',I3,X',E10.5)
3 FORMAT(' I',H(I,J),F(I,J),P(I,J),H(I,J-1)
1(I,J-1),P(I,J-1))
4 FORMAT(' I',QUAL(I),ALPHA(I),RHO(I,J),VP(I)
1 V(I),FMULT(I))
5 FORMAT(' K',W(K,J-1),W(K,J),WP(K),USTAR(K)
1(K,J-1),SP(K,J))
6 FORMAT(' I',DHDX(I),UH(I),DPDX(I),QPRIM(I)
1(I,J),RHOOLD(I,J))
16 FORMAT(3I5,4E12.6)
52 FORMAT(15,6E12.6)
C
MEKIN, IPILE = 0,1,2 FOR STANDARD COBRA, PWR, BWR
IPILE = J7
NCHANL = NCHANL
FMIN = .0001
NDXP1 = NDX+1
IF(JUMP.EQ.3) GO TO 400
JUMP = 2
C
C BEGIN STEPPING THROUGH CHANNEL
400 DO 450 J=1,NDXP1
CALL PRINTIR (3)
JPI = J+1
JMI = J-1
IF(J.GT.1) GO TO 405
C SET CONDITIONS AT START OF CHANNEL
DO 401 I=1,NCHANL
401 DATA($QPRIM+I)=0.
IF(IEGERR.GT.1) GO TO 440
CALL AREA(1)
IF(IEGERR.GT.1) GO TO 440
CALL PROP(2,1)
IF(IEGERR.GT.1) GO TO 440
CALL VOID(1)
IF(IEGERR.GT.1) GO TO 440

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GO TO 428
405 IF(JUMP.EQ.3) GO TO 420
   IF(NGRID.LT.1) GO TO 410
   GRID = .FALSE.
   DO 408 I=1,NGRID
     ZG = GRIDXL(I)*Z
     IF(ZG.GT.DATA(SX+JMI).AND.
1     ZG.LE.DATA(SX+J)) GO TO 409
408 CONTINUE
   GO TO 410
409 NGTYPE = IGRID(I)
   GRID = .TRUE.
C CALCULATE PARAMETERS TO BE SAVED FROM PREVIOUS SPACE
410 DO 411 I=1,NCHARL
   DATA(SVFA+I)=DATA(SVP+I)/DATA(SA+I)
411 CONTINUE
420 CALL HEAT(J)
   IF(IEORR.GT.1) GO TO 440
   IF (IPILE.EQ.2) GO TO 423
   CALL MIX(JMI)
   IF(IEORR.GT.1) GO TO 440
423 CALL DIFFER(1,JMI)
   IF(IEORR.GT.1) GO TO 440
C CALCULATE ENTHALPY AND ESTIMATE FLOW AT X.
C DO 425 I=1,NCHANL
C JK INSERT
   IF(ITRAT.EQ.1.AND.JUMP.NE.3.OR.(PILE.EQ.2) DATA(SF+I+MC*(J-1))
1   DATA(SH+I+MC*(J-1))=DATA(SH+I+MC*(JMI-1)
   DATA(SF+I+MC*(JMI-1))
2   DATA(SHOLD+I+MC*(J-1))+DX*DATA(SHDX+I)/(1.0+DX/DT)
425 CONTINUE
   IF(JUMP.EQ.3) GO TO 450
   CALL FORCE(J)
   IF(IEORR.GT.1) GO TO 440
   CALL AREA(J)
   IF(IEORR.GT.1) GO TO 440
   CALL PROP(2,J)
   IF(IEORR.GT.1) GO TO 440
   CALL VOID(J)
   IF(IEORR.GT.1) GO TO 440
   CALL DIFFER(3,J)
   IF(IEORR.GT.1) GO TO 440
   IF (IPILE.NE.2) GO TO 4255
   CALL SEPRAT(1,J,JUMP)
   IF(IEORR.GT.1) GO TO 440
   GO TO 435
4255 DO 426 K=1,NK
   DATA(SWAVE+K)=DATA(SW+K+MC*(J-1))
426 CONTINUE
C CALCULATE THE DIVERSION CROSSFLOW AT X.
   CALL PRINTM(4)
   CALL DIVERT(J)
   CALL PRINTM(5)

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IF(IEORR.GT.1) GO TO 440
C CALCULATE THE FLOW AT X AND CHECK FOR CONVERGENCE.
CALL DIFFER(2,J)
IF(IEORR.GT.1) GO TO 440
DO 4270 I=1,NCHANL
  FSAVE=DATA($F+I+MC*(J-1))
  T R INSERT
  RHODIF=DATA($RHOD+I+MC*(J-1))-DATA($RHODL+I+MC*(J-1))
  IF(DT.LT.0.001.AND.ABS(RHODIF).LT.0.001) RHODIF=0.0
  DATA($F+I+MC*(J-1))=DATA($F+I+MC*(JMI-1))+DX*DATA($DFDX+I)-DX/DT*
  1 RHODIF*DATA($A+I)
C THE FOLLOWING STATEMENT PROVIDES DAMPING TO ASSIST IN MORE RAPID
C CONVERGENCE, ESPECIALLY WHEN USING THE SUBCOOLED VOID OPTION.
C USERS MAY WISH TO TRY OTHER COMBINATIONS OF CONSTANTS.
DATA($F+I+MC*(J-1))=0.2*FSAVE+0.8*DATA($F+I+MC*(J-1))
IF(ABS(DATA($F+I+MC*(J-1))-FSAVE)/FSAVE.GT.FERROR) JUMP=1
IF(DATA($F+I+MC*(J-1)).LT.FMIN) DATA($F+I+MC*(J-1))=FMIN
4270 CONTINUE
C CALCULATE SP AT X-DX.
CALL DIFFER(4,J)
IF(IEORR.GT.1) GO TO 440
C THE FACTOR DAMPING WAS ADDED AFTER PUBLICATION. A VALUE OF ZERO WAS
C USED FOR THE SAMPLE PROBLEMS. A VALUE OF 0.5 HAS BEEN FOUND TO SPEED
C CONVERGENCE FOR MANY PROBLEMS. USERS MAY WISH TO TRY OTHER VALUES.
DAMPNG = 0.
DO 430 K=1,NK
  II=IDAT($IK+K)
  JJ=IDAT($JK+K)
  DATA($P+K+MG*(JMI-1))=DAMPNG*DATA($SP+K+MG*(JMI-1))+{1.-DAMPNG}*
  1 (DATA($P+K+MG*(J-1))-(DATA($DPDX+II)-DATA($DPDX+JJ))*DX)
430 CONTINUE
435 DO 427 I=1,NCHANL
427 DATA($F+I+MC*(J-1))=DATA($P+I+MC*(JMI-1))+DX*DATA($OPDX+I)
428 CONTINUE
IF(KDEBEG.LT.1) GO TO 450
GO TO 445
440 WRITE(13,1) J,DATA($K+J)
445 WRITE(13,2)J,DATA($K+J)
446 WRITE(13,3)
  1 WRITE(13,52) (I,DATA($H +I+MC*(J -1)),DATA($F +I+MC*(J -1)),
  1 DATA($P +I+MC*(J -1)),DATA($H +I+MC*(JMI-1)),
  2 DATA($F +I+MC*(JMI-1)),DATA($P +I+MC*(JMI-1)),
  1 I=1,NCHANL)
  1 WRITE(13,4)
  1 WRITE(13,52) (I,DATA($QUAL +I
  1 DATA($RHOD +I+MC*(J -1)),DATA($ALPHA+I
  2 DATA($V +I
  1 I=1,NCHANL)
  1 WRITE(13,5)
  1 WRITE(13,52) (K,DATA($W +K+MG*(JMI-1)),DATA($W +K+MG*(J -1)),
  2 DATA($WP +K
  1 DATA($USTAR+K
  2 DATA($SP +K+MG*(JMI-1)),DATA($SP +K+MG*(J -1)),
  1 K=1,NK)
  1 WRITE(13,6)

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WRITE(13,52) (I,DATA(SDHX +I
1      DATA(SDPOX +I
2      DATA(SFOLD +I+MC*(J
1      I=1,NCHANL)
IF(IEORHOR.GT.1) RETURN
450 CONTINUE
IF(JUMP.EQ.3) RETURN
C CORRECT SUBCHANNEL PRESSURES TO ZERO EXIT PRESSURE.
C PRESSURE P(I,J) IS THE PRESSURE ABOVE THE EXIT REFERENCE PRESSURE.
DO 460 I=1,NCHANL
PEK11=DATA($P+I+MC*(NDXP1-1))
DO 460 J=1,NDXPI
460 DATA($P+I+MC*(J-1))-DATA($P+I+MC*(J-1)) - PEK11
IF (IPILE.NE.2) RETURN
CALL SEPRAT(2,J,JUMP)
RETURN
END
FUNCTION SCQUAL(I,J)
C LEVY SUBCOOLED MODEL. CALCULATES TRUE QUALITY AS A CORRECTION TO
C THE EQUILIBRIUM QUALITY.
C
C
IMPLICIT INTEGER ($)
COMMON /COBRAT/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , HF
2 HFG , HG , I2 , I3 , IERROR , IOP3 , ITERAT , JI , J2
3 J3 , J4 , J5 , J6 , J7 , KOEBUG , KF , KI , J
4 HAFACT , NARAMP , NAX , MAXL , NDBC , NCHANL , NCHF , NDX , NF
5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , NPROP ,
6 NRAMP , NROD , NROD , NSDC , NV , NVISCM , PI , PITCH , POWER , PREF ,
7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK ,
8 UF , VV , VFG , VG , Z
C
COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
2 GAPAL(10) , GFACT(9,10) , GRIDAL(10) , HGAP(2) , HMF(30) , HMG(30) ,
3 IGRID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
4 PP(30) , RCLAD(2) , RFUEL(2) , SIGMA(30) , TCLAD(2) , UUF(30) ,
5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , TT(30)
C
LOGICAL GRID
REAL KIJ , KF , KKF , KCLAD , KFUEL
C
COMMON /COBRA3/ MA , MC , MG , MN , MR , MS , MX
1 $$$ , SA , SAA , SAC , SALPHA , SAN , SANSWE , SB
2 SCCHAN , SCD , SCHFR , SCON , SCOND , SCP , SD , SDC , SDFDX
3 SDHX , SDHYD , SDHYDN , SDI , SDPX , SDPK , SDUR , SDR , SF
4 SFACTO , SFDIV , SFINLE , SFLUX , SFMULT , SFOLD , SFSP , SFSPLI , SFXFLO ,
5 SIDVLE , SIDGAP , SIK , SUBOIL , SJK , SLC , SLENGT , SLOCA , SLR
6 SIMCHFR , SIMCFRC , SIMCFRR , SNTYPE , SNWRAP , SNWRPS , SP
7 SPHI , SPINTC , SPINTR , SPRTN , SPW , SPWRF , SOC , SOF , SOPRIM ,
8 SQUAL , SRADIA , SRHO , SRHOOL , SSP , ST , STDUWY , STJNLE , STROD ,

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9 SU      ,SUH      ,SUSAVE,SUSTAR,SV      ,SVISC ,SVISCM,SVP      ,SVPA  ,COB64910
A $W     ,SWOLD   ,SWP      ,SWSAVE,$X    ,XACRUS,$SA  ,SSB  ,XPOLO  ,COB64920
C
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
C
XP=DATA(SQUAL+1)
DATA($XPOLD+1)=Q.
SQQUAL = XP
IF(DATA(SOPRIM+1).LE.0.) RETURN
CNC = 0.015
JJ=J
C ***** THE FOLLOWING CARDS CORRECT THE LEVY MODEL *****
YB=GNC/UF *3600. *SORT( SIGMA *GC*DATA(SOHYD+1)/VF)
TAUH= DATA($FSP+1)*.125*VF *(DATA($F+MC*(J-1)+1)/
1 DATA($A+1))*2/GC
PR=DATA(SCP+1)*UF/KF
Q= DATA(SOPRIM+1)/(DATA($HPERI+1)/VF *DATA(SCP+1)*
1 SORT(TAUN*GC*VF
))
JK INSERT
RE=DATA($F+I+MC*(J-1))/DATA($A+1)*DATA(SOHYD+1)/DATA(SVISC+1)
IF(RE.LT.2000.) RE=2000.
HTC=DATA(SCOH+1) /DATA(SOHYD+1)*.023*RE**4
DELTA=DATA(SOPRIM+1)/DATA($HPERI +1)/HTC
C *****
IF(YB.GE.0. AND. YB.LT.5.) DELTAT = DELTAT - Q*PR*YB
IF(YB.GE.5. AND. YB.LT.30.) DELTAT = DELTAT - 5.*Q*(PR+ALOG(1.+5.*PR
1 YB*.2-1.))
IF(YB.GE.30.) DELTAT = DELTAT - 5.*Q*(PR+ALOG(1.+5.*PR
1 + .5*ALOG(YB/30.))
XD=-DATA(SCP+1)*DELTAT/HFG
ARG=DATA(SQUAL+1)/XD-1.
IF (ARG.LT. -15.0) GO TO 140
IF(ARG.GT.0.) ARG = 0.
XP =DATA(SQUAL+1)-XD*EXP(ARG)
C ***** THE FOLLOWING CARDS CORRECT THE LEVY MODEL *****
IF(DATA(SQUAL+1).LT.XD) XP=0.
IF(J7.EQ.2) GO TO 130
IF(ITERAT.EQ.1) DATA($XPOLD+I+MC*(J-1))=XP
DUMMY=DATA($XPOLD+I+MC*(J-1))
XP=.99*XP+.01*DUMMY
130 IF(JJ.EQ.1) JJ=2
XP=AMAX1(XP,DATA($XPOLD+I+MC*(JJ-2)))
DATA($XPOLD+I+MC*(J-1))=XP
140 SQQUAL = XP
RETURN
END
SUBROUTINE SEPRAT(IPART,J,JUMP)
C
C FLOW ITERATION FOR SEPARATED CHANNELS (EG BWR)
C CALLED FROM SCHEME
C SP USED FOR (1) DM/DP (2) DM (3) DP

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10  PMIN=10000.0
    PMAX=-1000.0
    DO 12 I=1,NCHANL
      WV=DATA($P+I)
      IF (WV.LT.PMIN) PMIN=WV
      IF (WV.GT.PMAX) PMAX=WV
12  CONTINUE
    IF (ABS(1.-PMIN/PMAX).LT.FERROR) RETURN
      JUMP=1
      IF(ITERAT.GT.1) GO TO 16
      FTOT=0.0
      DO 14 I=1,NCHANL
        FDOT=FDOT+DATA($F+I)
      IF (DATA($SP+I).GT.0.0) GO TO 14
      DATA($SP+I)=0.7*DATA($F+I)/(DATA($P+I)-DATA($RHO+I+MC*NDX)*
      ELEV*Z)
14  CONTINUE
      GO TO 20
16  DO 18 I=1,NCHANL
      DELTAP= (DATA($P+I) -DATA($SP+I+2*MC))
      IF (ABS(DELTAP).LT..001) GO TO 18
      DATA($SP+I) = ( DATA($F+I)-DATA($SP+I+MC))/DELTAP
18  CONTINUE
      SUM1=0.0
      SUM13=0.0
      DO 22 I=1,NCHANL
        SUM1=SUM1+DATA($SP+I)
        SUM13=SUM13+DATA($SP+I)+DATA($P+I)
        DATA($SP+I+MC)=DATA($F+I)
        DATA($SP+I+MC*2)=DATA($P+I)
22  JK INSERT
      P113=SUM13/SUM1
      IF (ITERAT.EQ.1.AND.DT.GT.1000.) GO TO 23
      IF (P113.LE.0..OR.P113.GT.2*PO) P113=PO
      PO=P113
      IF (PO.LT.0.) PO=ABS(PO)
      SUMF=0.0
      DO 24 I=1,NCHANL
        DATA($F+I)=DATA($F+I)+DATA($SP+I)*(PO-DATA($P+I))
        SUMF=SUMF+DATA($F+I)
24  RETURN
      DATA($F+I)=DATA($F+I)+FTOT/SUMF
      END
      SUBROUTINE VOID (J)
      C
      C
      IMPLICIT INTEGER (S)
      COMMON /COBRA1/ ABETA ,AFLUX ,ATOTAL,BBETA ,DIA ,DT ,DX
      1  ELEV ,FERROR,FLO ,FTM ,GC ,GK ,GRID ,HSURF ,HF
      2  HFG ,HG ,I2 ,I3 ,IERROR,IQPS ,ITERAT ,JI ,J2
      3  J3 ,J4 ,J5 ,J6 ,J7 ,KDEBUG,NF ,KIJ
      4  NAFAC,NARAMP,NAX ,NAXL ,NBBC ,NCHAN ,NCHF ,NDX ,NF
      5  NGAPS ,NGRID ,NGRIDT ,NGTYPE ,NGXL ,NK ,NODES ,NODESF ,NPROP
      6  NRAMP ,NROD ,NSCBC ,NV ,NVISCH,PI ,PITCH ,POWER ,PREF ,

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CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

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7 QAX ,RHOF ,RHOG ,SIGMA ,SL ,TF ,TFLUID,THETA ,THICK ,COB66560
8 UF ,VFG ,VG ,Z ,COB66570
COB66580
COMMON /COBRA2/ AA(4), AF(7), AFACT(10,10), AV(7), AXIAL(30),
1 AXL(10), BB(4), BK(30), CC(4), CCLAD(2), CFUEL(2), DFUEL(2),
2 GAPXL(10), GFAC(9,10), GRIDXL(10), HGAP(2), HHF(30), HHG(30),
3 IGRID(10), KCLAD(2), KFUEL(2), KMF(30), NCH(10), NGAP(9),
4 PP(30), RCLAD(2), RFUEL(2), SSIGMA(30), TCLAD(2), UUF(30),
5 VVF(30), VVG(30), XQUAL(30), Y(30), TT(30)
COB66640
COB66650
COB66660
COB66670
COB66680
COB66690
COB66700
COMMON /COBRA3/ MA ,MC ,MG ,MH ,MI ,MJ ,MK ,ML ,MM ,MN ,MO ,MP ,MQ ,MR ,MS ,MT ,MX
1 $$$ ,SA ,SAAA ,SAC ,SALPHA,$AN ,SANSWE,$D ,SANSWE,$D
2 SCCHAN,$CD ,SCHFR,$CON ,SCOND,$CP ,SD ,SDC ,SDFDX ,COB66720
3 SDHX ,SDHYD ,SDHYDN,$DST ,SDPDX ,SDPK ,SDUR ,SDR ,SDF ,COB66730
4 SFACTO,$FDIV ,SFINLE,$FLUX ,SFMULT,$FOLD ,SFSP ,SFSPLI,$FXFLO,COB66740
5 SGAP ,SGAPN ,SGAPS,$H ,SHFILM,$HINLE,$HOLD ,SHPERI,$IDARE,COB66760
6 SIDFUE,$IDGAP,$IK ,SUBOIL,$JK ,SLC ,SLENGT,$LOCA,$LR ,COB66770
7 SICHFR,$ICFRC,$ICFRF,$INTYPE,$NWRAP,$NWRPS,$P ,SPERIM,$PH ,COB66780
8 SDOUM ,SPRNTC,$PRNTR,$PRNTN,$PW ,SPRNF,$QC ,SQF ,SQPRIM,COB66790
9 SQUAL ,SRADIA,$RHO ,SRHOOL,$SP ,ST ,STDUIM,$TINLE,$TROD ,COB66800
A SU ,SUH ,SUSAVE,$USTAR,$V ,SVISC ,SVISCH,$VP ,SVPA ,COB66810
A $W ,SWOLD,$WP ,SWSAVE,$X ,SXCROS,$XA ,SXPOLD ,COB66820
COB66830
COB66840
COMMON /PPSV/ PPI
COB66850
COMMON DATA(1)
LOGICAL LDAT(1)
INTEGER IDAT(1)
EQUIVALENCE (DATA(1),IDAT(1),LDAT(1))
C
EQUIVALENCE (NCHAN,NCHANL)
$PHI=$FMULT
DO 200 I=1,NCHAN
PSI = 0.
DPSIDH = 0.
IF (J3.EQ.0) GO TO 40
DATA($H+MC*(J-1))=DATA($H+MC*(J-1))-1
DATA($QUAL+I) =(DATA($H+MC*(J-1))-HF)/HFG
IF (J2.EQ.1) DATA($QUAL+I)=SQUAL(I,J)
IF (DATA($QUAL+I).LE.0.) DATA($QUAL+I)=0.
DATA($ALPHA+I) = BVOID(I,J)
PSI=RHO*DATA($QUAL+I)*(1.-DATA($ALPHA+I))-RHO*DATA($ALPHA+I)*
1 (1.-DATA($QUAL+I))
DATA($H+MC*(J-1))=DATA($H+MC*(J-1))+1
DATA($QUAL+I) =(DATA($H+MC*(J-1))-HF)/HFG
IF (J2.EQ.1) DATA($QUAL+I)=SQUAL(I,J)
IF (DATA($QUAL+I)-LE.0.) GO TO 150
XP= DATA($QUAL+I)
DATA($ALPHA+I)=BVOID(I,J)
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C CALCULATE TWO-PHASE DENSITY.
C ***** THE FOLLOWING CARDS CORRECT THE CALCULATION OF RHO AND VP
DATA(SRHO+I+MC*(J-1))=RHOG*DATA(SALPHA+I)+I./DATA(SV+I)*((1.-
1 DATA(SALPHA+I))
C CALCULATE TWO-PHASE SPECIFIC VOLUME FOR MOMENTUM.
DATA(SVP+I)=DATA(SV+I)*((1.-XP)**2/((1.-DATA(SALPHA+I))+VG*XP**2/
1 DATA(SALPHA+I))
C JK INSERT
IF(J7.NE.2) GO TO 3
IF(J.EQ.1) GO TO 3
RHODIF = DATA(SRHO+I+MC*(J-1))-DATA(SRHOOL+I+MC*(J-1))
DATA(SF+I+MC*(J-1))= DATA(SF+I+MC*(J-2))-DX/DT*RHODIF*DATA(SA+I)
3 CONTINUE
C TWO-PHASE FRICTIONAL PRESSURE GRADIENT MULTIPLIERS.
DATA(SPHI+I)=1.
IF(J4.EQ.0) DATA(SPHI+I)=RHOF/DATA(SRHO+I+MC*(J-1))
GRV = 3600.0*DATA(SF+I+MC*(J-1))/DATA(SA+I)
IF (J4.EQ.2) CALL BAROC(2,PREF,XP,GRV,DATA(SPHI+I),PPI)
IF(J4.NE.1) GO TO 50
DATA(SPHI+I)=1.
XA=DATA(SALPHA+I)
IF(XA.GT.0.0.AND.XA.LE.0.6)XXX=(1.-XP)**2/((1.-XA)**1.42
IF(XA.GT.0.6.AND.XA.LE.0.9)XXX=.478*(1.-XP)**2/((1.-XA)**2.2
IF(XA.GT.0.9.AND.XA.LE.1.0)XXX=1.73*(1.-XP)**2/((1.-XA)**1.64
DATA(SPHI+I)=XXX
50 IF(J4.NE.5) GO TO 140
DATA(SPHI+I)=AF(1)
XX =DATA(SQUAL+I)
DO 130 K=2,NF
DATA(SPHI+I)=DATA(SPHI+I)+AF(K)*XX
130 XX =DATA(SQUAL+I)*XX
140 DATA(SU+I)=DATA(SF+I+MC*(J-1))/DATA(SA+I)*DATA(SVP+I)
IF(J3.EQ.0) GO TO 145
DPSIDH=-10.*(PSI-RHOF*DATA(SQUAL+I))*((1.-DATA(SALPHA+I))+RHOG*
1 DATA(SALPHA+I))*((1.-DATA(SQUAL+I)))
145 DATA(SUH+I)=DATA(SF+I+MC*(J-1))/DATA(SA+I)/(DATA(SRHO+I+MC*(J-1))
1 -HFG-DPSIDH)
GO TO 200
C TWO-PHASE FLOW PARAMETERS WITHOUT BOILING.
150 DATA(SALPHA+I)=0.0
DATA(SRHO+I+MC*(J-1))=1.0/DATA(SV+I)
C JK INSERT
IF(J7.NE.2) GO TO 4
IF(J.EQ.1) GO TO 4
RHODIF = DATA(SRHO+I+MC*(J-1))-DATA(SRHOOL+I+MC*(J-1))
DATA(SF+I+MC*(J-1))= DATA(SF+I+MC*(J-2))-DX/DT*RHODIF*DATA(SA+I)
4 CONTINUE
DATA(SVP+I)=DATA(SV+I)
DATA(SU+I)=DATA(SF+I+MC*(J-1))/DATA(SA+I)*DATA(SVP+I)
DATA(SUH+I)=DATA(SU+I)
DATA(SPHI+I)=1.0
DATA(SQUAL+I)=0.0
200 CONTINUE
RETURN
END

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COB67600  
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```

C      DATA CC,CCI,CCM /1.3, .76923, 0.3/
C      DATA RL0,RL1,RL2 /1.E3, -2.E-5, -.15E-9/
C      DATA RL22 /-.3E-9/
C      DATA CL21 /0.657E-6/
C      FOR EL IF TL < 300 DEG C
C      DATA SL0,SL1,SL2,SL3 /-1.4655677D+06, 6.9269554D+03,
1      -7.7423067E0, 7.2803006D-03/
C      SLO IS CHOSEN SO THE JUMP IN EL AT 300 DEG C IS AS
C      SMALL AS POSSIBLE
C      DATA SL22,SL33 /-15.484613, 2.1840901E-2/
C      SL22 = 2.* SL2, SL33 = 3.* SL3
C      FOR EL IF TL > 300 DEG C
C      DATA SH0,SH1,SH2,SH3 /-0.9, 2.3639439E+04,
1      -7.7434017E+01, 7.0215574E-02/
C      DATA SH22,SH33 /-1.5486803E2, 2.1084672E-1/
C      SH22 = 2.* SH2, SH33 = 3.* SH3
C      FOR VAPOR
C      DATA A11,A12,A13 /1.2959E-3, 593.59, 1.6847E-3/
C      DATA HALF,ZERO,ONE,TWO /0.5, 0., 1., 2./
C      -----
C      CHECK THAT P, TL, TV, ARE WITHIN RANGE OF FITS
C      IF (P.GE.1.0E+3.AND.P.LE.190.0E+5) GO TO 5
C      IERR = 1
C      RETURN
C      IF (TL.GE.280.0.AND.TL.LE.647.0) GO TO 10
C      IERR = 2
C      RETURN
C      IF (TV.GE.280.0) GO TO 20
C      IERR = 3
C      RETURN
C      IERR = 0
C      CALCULATE SATURATION PROPERTIES
C      1. TSAT SATURATION TEMPERATURE
C      2. DTSDP DERIVATIVE OF TSAT WRT PRESSURE
C      3. ES SATURATION INTERNAL ENERGY
C      4. DPES DERIVATIVE OF ES WRT PRESSURE
C      5. GAMS GAMMA SUB S
C      6. DPGAMS DERIVATIVE OF GAMS WRT PRESSURE
C      7. CPS C SUB PS
C      8. DPCPS DERIVATIVE OF CPS WRT PRESSURE
C      9. GAMSM GAMS-ONE
C      TSAT = TSC1* P**TSEXP
C      PINV = ONE/ P
C      DTSDP = TSAT*TSEXP*PINV

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COB68210
COB68220
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COB68680
COB68690
COB68700
COB68710
COB68720
COB68730
COB68740
COB68750

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C      TSAT = TSAT + TSC2
      T1 = ONE - TSAT*TCRINV
      CPS = CPS1 + T1*CFSEXP
      DPCS = CPS2 + CPS/T1 + OTSDP
C
      IF (P.GT.P20B) GO TO 150
      T2 = ONE/ (G13+P)
      T1 = T2*G12
      ES = G11 + T1
      DPES = -T1*T2
      GAMS = G14 + P*(G15 + P*G16)
      DPGAMS = G15+G17*P
      GO TO 200
150 CONTINUE
      ES = G21+(G23*P+G22)*P
      DPES = G22+G27*P
      GAMS = G24+(G26*P+G25)*P
      DPGAMS = G25 + G28*P
      GAMS*P = GAMS - ONE
200 GAMS*P = GAMS - ONE
C
      CALCULATE LIQUID PROPERTIES
C
      1. INTERNAL ENERGY AND ITS DERIVATIVES
      DELDP = 0.
      IF (TL.GE.573.15) GO TO 220
      EL = SLO + TL*(SL1 + TL*(SL2 + TL*SL3))
      DELDT = SL1 + TL*(SL22 + TL*SL33)
      GO TO 240.
220 CONTINUE
      EL = SH0 + TL*(SH1 + TL*(SH2 + TL*SH3))
      DELDT = SH1 + TL*(SH22 + TL*SH33)
240 CONTINUE
C
      2. DENSITY AND ITS DERIVATIVES
      ROL = RLO + EL*(RL1 + EL*RL2) + P*CL21
      DRLOP = CL21
      DRLDE = RL1 + EL*RL2
      DRLDT = DRLDE*DELDT
C
      CALCULATE VAPOR PROPERTIES
C
      DT = TV-TSAT
      IF (DT.LE.ZERO) GO TO 250
      SUPERHEATED VAPOR
C
      1. BETA A WORKING PARAMETER
      2. CAPK A WORKING PARAMETER
      3. DBETAP DERIVATIVE OF BETA WRT PRESSURE
      4. DCAPKP DERIVATIVE OF CAPK WRT PRESSURE

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COB68760  
 COB68770  
 COB68780  
 COB68790  
 COB68800  
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 COB68980  
 COB68990  
 COB69000  
 COB69010  
 COB69020  
 COB69030  
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 COB69130  
 COB69140  
 COB69150  
 COB69160  
 COB69170  
 COB69180  
 COB69190  
 COB69200  
 COB69210  
 COB69220  
 COB69230  
 COB69240  
 COB69250  
 COB69260  
 COB69270  
 COB69280  
 COB69290  
 COB69300

CONVERSATIONAL MONITOR SYSTEM

FILE: COBRA3C FORTRAN A

```

C      5, DEVDT
C      6, DEVDP
C      7, ROV
C      8, DRVDE
C      9, DRVDP
C
T1 = ONE/(A11*CPS-ONE)
T1SQ = T1*T1
BETA = TSAT*TSAT*(ONE - T1SQ)
T2 = TSAT*T1
DE = A12*(DT+SQRT(TV*TV-BETA))-T2
EV = ES + DE
CAPK = A13*DE+TSAT+T2
DBETAP = TWO*(BETA*DTSDP+T2*T2+T2*A11*DPCPS)/TSAT
DCAPKP = -A13*DPES + (ONE + T1)*DTSDP
1  -TSAT*A14*T1SQ*DPCPS
T3 = ONE-BETA/(CAPK*CAPK)
DEVDT = ONE/(HALF*T3*A13)
DEVDP = -HALF*(T3*DCAPKP+DBETAP/CAPK)*DEVDT
T4 = ONE/(GAMSM*ES+CCM*DE)
ROV = P*T4
DRVDE = -ROV*CCM*T4
DRVDT = DRVDE*DEVDT
DRVDP = ROV*(PINV-(ES*DPGAMS+(GAMSM-CCM)*DPES)*T4)
1  + DRVDE*DEVDP
GO TO 300
250 CONTINUE
C
C      SUBCOOLED VAPOR
C
DEVDT = CPS + CCI
DE = DT * DEVDT
EV = ES + DE
T1 = ONE/ CPS
DEVDP = -(DTSDP -CC*T1*(DPES +DE*DPCPS*T1) )*DEVDT
T2 = ONE/ GAMSM
ROV = P *T1*T2
DRVDE = -ROV *T2
DRVDT = DRVDE * DEVDT
DRVDP = ROV *(PINV - DPGAMS*T1) + DRVDE*DEVDP
C      300 CONTINUE
RETURN
END
SUBROUTINE TEMFR(TDUMY,DT,N,TFLUID,HGAP,HSURF,QP,III,NT)
C OVERSEES NEW FUEL ROD MODEL CALCULATIONS
C      IMPLICIT INTEGER($)
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COB69310  
COB69320  
COB69330  
COB69340  
COB69350  
COB69360  
COB69370  
COB69380  
COB69390  
COB69400  
COB69410  
COB69420  
COB69430  
COB69440  
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COB69690  
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COB69720  
COB69730  
COB69740  
COB69750  
COB69760  
COB69770  
COB69780  
COB69790  
COB69800



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C
C COMMON/LINK4/IFRM,INTM,IPROP,NCC,NCF,NOM1,NDS,NGP
C
C DIMENSION TDUMY(1)
C
CC
CC OP IS VOLUMETRIC HEAT GENERATION RATE IN FUEL(BTU/SEC-FT**3)
C
C DO 20 JJ=1,NCF
C 20 OPPP(JJ)=QP
C   RDEL=1./DT
C   IF(INT.EQ.1) RDEL=0.
C   IF(INT.EQ.1.AND.III.EQ.1) GO TO 30
C   IF (IPROP.EQ.0) GO TO 30
C   CALL RTEMP(TDUMY(1),NCF,NGP,NOM1,HGAP,IPROP)
C   RETURN
C END
C SUBROUTINE INITRC
C
C C INITIALIZE ARRAYS FOR NEW FUEL ROD MODEL
C
C IMPLICIT INTEGER($ )
C REAL KCLAD,KFUEL
C
C COMMON /COBRA1/ ABETA , AFLUX , ATOTAL , BBETA , DIA , DT , DX
C 1 ELEV , FERROR , FLO , FTM , GC , GK , GRID , HSURF , MF
C 2 HFG , HG , I2 , I3 , TERROR , IQP3 , ITERAT , J1 , J2
C 3 J3 , J4 , J5 , J6 , J7 , KDEBUG , KF , KIJ
C 4 NAFAC , NARAMP , MAX , MAXL , NBBC , NCHAN , NCHF , NDX , NF
C 5 NGAPS , NGRID , NGRIDT , NGTYPE , NGXL , NK , NODES , NODESF , NPROP
C 6 NRAINP , NROD , NSCBC , NV , NVISCM , PI , PITCH , POWER , PREF
C 7 QAX , RHOF , RHOG , SIGMA , SL , TF , TFLUID , THETA , THICK
C 8 UF , VF , VFG , VG , VZ
C
C COMMON /COBRA2/ AA(4) , AF(7) , AFACT(10,10) , AV(7) , AXIAL(30) ,
C 1 AXL(10) , BB(4) , BX(30) , CC(4) , CCLAD(2) , CFUEL(2) , DFUEL(2) ,
C 2 GAPL(10) , GFAC(9,10) , GRIDXL(10) , HGAP(2) , HHF(30) , HHG(30) ,
C 3 IGR ID(10) , KCLAD(2) , KFUEL(2) , KKF(30) , NCH(10) , NGAP(9) ,
C 4 PP(30) , RCLAD(2) , RFUEL(2) , SSIGMA(30) , TCLAD(2) , UUF(30) ,
C 5 VVF(30) , VVG(30) , XQUAL(30) , Y(30) , YI(30)
C
C COMMON /COBRA3/ NA , NC , NG , NR , NS , NR , NS , NR
C 1 $$$ , SA , SAAA , SAC , SALPHA , SAN , SANSWE , SB
C 2 SCCHAN , SCD , SCHFR , SCON , SCOND , SCP , SD , SDC , SDFDX
C 3 SDHDX , SDHYD , SDHYDN , SDIST , SDPDX , SDPK , SDUR , SDR , SF
C 4 SFACD , SFDIV , SFINLE , SFLOX , SFMULT , SFOLD , SFSP , SFSPLI , SFXFLO , COB70330
C 5 SGAP , SGAPN , SGAPS , SH , SHFILM , SHINLE , SHOLD , SHPERI , SIDARE , COB70340
C 6 SIDFUE , SIDGAP , SIK , SUBOIL , SJK , SLENGT , SLOCA , SLR , COB70350
C 7 SNCHFR , SNCFRC , SNCFRR , SNTYPE , SNWRAP , SNWRPS , SP , SPERIM , SPH , COB70360
C 8 SPHI , SPRNTC , SPRNTR , SPRNTN , SPW , SPWRF , SOC , SOF , SOPRIM , COB70370
C 9 SQUAL , SRADIA , SRHO , SRHOOL , SSP , ST , STDUMY , STINLE , STROD , COB70380
C 0 SUH , SUSAVE , SUSTAR , SV , SVISC , SVISCM , SVP , SVPA , COB70390
C A SW , SWOLD , SWP , SWSAVE , SX , SXCR05 , SXS , SXOLD , COB70400

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C
COMMON /MCOND/ CND(22),RCP(22),RAD(22),RRDR(22),
1 VM(22),VP(22),QPPP(22)
C
COMMON /FRDATA/ BURN, CPR, EFB, EPSF, EXPR, FPRESS, FPUO2, FRAC, FTD,
1 GMIX(4), GRGH, PGAS, RADR, RDELT, THC, THG
C
COMMON /LINK4/ IFRM, IHTM, IPROP, MCC, NCF, NDM1, NDS, NGP
C
COMMON DATA(1)
C
C INITIALIZE ROD CONDUCTION ARRAYS
C AND MAKE INITIALIZING CALL TO GAP CONDUCTANCE SUBROUTINE
C
C GEOMETRY ARRAYS
RADR=DATA(SD+1)/2.
THC=TCCLAD(1)
NDM1=NODESF
NDS=NODESF+1
NGP=NGF+1
DRF=0.5*DFUELL(1)/NCF
DRC=THC/NCC
RAD(1)=0.0
DO 10 K=1,NCF
RAD(K)=K*DRF
RAD(NGP+1)=RAD(NCF+1)+THG
DO 20 K=1,NCC
RAD(NGP+1+K)=RAD(NGP+1)+K*DRC
DO 30 K=1,NDM1
IF(K.EQ.NGP)RRDR(K)=.5*(RAD(K+1)+RAD(K))
IF(K.NE.NGP)RRDR(K)=.5*(RAD(K+1)+RAD(K))/(RAD(K+1)-RAD(K))
30 CONTINUE
VM(1)=0.0
VP(1)=DRF*DRF/8.0
DO 40 K=2,NDM1
RP=0.5*(RAD(K+1)+RAD(K))
RM=0.5*(RAD(K)+RAD(K-1))
VP(K)=0.5*(RP*RP-RAD(K)*RAD(K))
VM(K)=0.5*(RAD(K)*RAD(K)-RM*RM)
RM=0.5*(RADR+RADI(NDM1))
VM(NDS)=0.5*(RADR+RADR-RM*RM)
VP(NDS)=0.0
C ASSUME NO HEAT GENERATED IN GAP OR CLADDING
DO 105 K=NGP,NDM1
105 QPPP(K)=0.
C
C MATERIAL PROPERTY ARRAYS
C
DO 110 K=1,NCF
CND(K)=KFUEL(1)
RCP(K)=CFUEL(1)*RFUEL(1)
CND(NGP)=HGAP(1)
RCP(NGP)=0.0
110

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COB70410  
COB70420  
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COB70890  
COB70900  
COB70910  
COB70920  
COB70930  
COB70940  
COB70950

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DO 120 K=1,NCC
  CND(NGP+K)=KCLAD(1)
  RCP(NGP+K)=CCLAD(1)*RCLAD(1)
120
C INITIALIZE GAP CONDUCTANCE DATA
C
  IF(1PROP.LT.2)GO TO 205
  CALL MPG(TRUE,BURN,EFFB,FRAC,D3,D4,D5,GRGH,THG,RAD(NGP),
  1 D6,D7,D8,D9,D10,D11)
205 CONTINUE
  RETURN
  END
  SUBROUTINE RTEMPF (TR,ROD,RADR,HSURF,TFLUID,NODES,NDM1)
C GAUSSIAN SOLUTION OF TRIANGULAR TEMPERATURE PROBLEM IN FUEL ROD
C
  COMMON /MCOND/ CND(22),RCP(22),RAD(22),RRDR(22),
  1 VM(22),VP(22),QPPP(22)
C
  DIMENSION A1(23),A2(22),A3(22),B(22),TR(1)
  FSS=1.
  FIR=1.-FSS
  RDEL=ROD
C SET UP COEFFICIENTS OF TRIANGULAR MATRIX
C
  A1(1)=0.0
  A2(1)=RRDR(1)*CND(1)+RDEL*VP(1)*RCP(1)
  B(1)=VP(1)*QPPP(1)+RDEL*VP(1)*RCP(1)*TR(1)
  DO 100 K=2,NDM1
    A1(K)=-RRDR(K-1)*CND(K-1)
    A2(K)=-A1(K)+RRDR(K)*CND(K)+RDEL*(VP(K)*RCP(K)+VM(K)*RCP(K-1)+
    1 RCP(K-1))*TR(K)
  100 CONTINUE
    A1(NODES)=-RRDR(NDM1)*CND(NDM1)
    A2(NODES)=-A1(NODES)+RDEL*VM(NODES)*RCP(NDM1)+
    + RADR*FSS*HSURF
    B(NODES)=VM(NODES)*QPPP(NDM1)+
    + RDEL*VM(NODES)*RCP(NDM1)*TR(NODES)+
    + RADR*HSURF*(TFLUID-FTR*TR(NODES))
    A1(NODES+1)=0.0
C FORWARD ELIMINATION
C
  A2(1)=1./A2(1)
  A3(1)=A1(2)*A2(1)
  B(1)=B(1)-A2(1)
  DO 200 K=2,NODES
    A2(K)=1./(A2(K)-A1(K)*A3(K-1))
    A3(K)=A1(K+1)+A2(K)
    B(K)=B(K)-A1(K)*B(K-1)+A2(K)
  200 CONTINUE
CC BACKWARD SUBSTITUTION

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COB70960  
COB70970  
COB70980  
COB70990  
COB71000  
COB71010  
COB71020  
COB71030  
COB71040  
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COB71060  
COB71070  
COB71080  
COB71090  
COB71100  
COB71110  
COB71120  
COB71130  
COB71140  
COB71150  
COB71160  
COB71170  
COB71180  
COB71190  
COB71200  
COB71210  
COB71220  
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COB71440  
COB71450  
COB71460  
COB71470  
COB71480  
COB71490  
COB71500

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CC      TR(NODES)=B(NODES)
      DO 250 K=1,NDMI
      KK = NODES-K
      250 TR(KK)=B(KK)-TR(KK+1)*A3(KK)
C
C      RETURN
      END
C      SUBROUTINE RPROP(TRN,NCF,NGP,NDMI,HGAP,IPROP)
C      GET MATERIAL AND GAP PROPERTIES FOR ROD CONDUCTION CALCULATION
C
      COMMON /MCOND/ CND(22),RCP(22),RAD(22),RRDR(22),
      1 VM(22),VP(22),QPPP(22)
C
      COMMON /FRDATA/BURN,CPR,EFFB,EPFS,EXPR,FPRESS,FPUO2,FRAG,FTD,
      1 GMIX(4),GRGH,PGAS,RADR,RDELT,THC,TMG
C      DIMENSION TRN(1)
C      COMPUTE FUEL PROPERTIES
C
      DO 100 K=1,NCF
      ATEMP=0.5*(TRN(K+1)+TRN(K))
      CALL MPF(ATEMP,FTD,FPUO2,RCP(K),CND(K))
      100 CONTINUE
C      COMPUTE CLAD PROPERTIES
C
      KSTART=NGP+1
      DO 200 K=KSTART,NDMI
      ATEMP=0.5*(TRN(K+1)+TRN(K))
      CALL MPC(ATEMP,RCP(K),CND(K))
      200 CONTINUE
C      CALCULATE GAP HEAT TRANSFER COEFFICIENT
C
      IF(IPROP.LT.2) GO TO 300
      TGAP=(TRN(NGP)+TRN(NGP+1))*0.5
      CALL MPC(.FALSE.,BURN,EFFB,FRAC,FPRESS,CPR,EXPR,GRGH,TMG,
      1 RAD(NGP),PGAS,TGAP,GMIX,TRN(NGP),TRN(NGP+1),HGAP)
      300 CONTINUE
      CND(NGP)=HGAP
      RCP(NGP)=0.0
      305 RETURN
      END
C      SUBROUTINE MPF (TFUEL, FTD, FPUO2, RCP, COND)
C      CALCULATES HEAT CAPACITY AND CONDUCTIVITY OF UO2 AND PUO2 FUELS AS
C      FUNCTIONS OF TEMPERATURE, FRACTION OF THEORETICAL DENSITY, AND
C      PLUTONIUM CONTENT
C      ARGUMENTS      TFUEL      TEMPERATURE (DEG F)
C      INPUT

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COB71510
COB71520
COB71530
COB71540
COB71550
COB71560
COB71570
COB71580
COB71590
COB71600
COB71610
COB71620
COB71630
COB71640
COB71650
COB71660
COB71670
COB71680
COB71690
COB71700
COB71710
COB71720
COB71730
COB71740
COB71750
COB71760
COB71770
COB71780
COB71790
COB71800
COB71810
COB71820
COB71830
COB71840
COB71850
COB71860
COB71870
COB71880
COB71890
COB71900
COB71910
COB71920
COB71930
COB71940
COB71950
COB71960
COB71970
COB71980
COB71990
COB72000
COB72010
COB72020
COB72030
COB72040
COB72050

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C      FTD      FRACTION OF THEORETICAL DENSITY
C      FPUO2    PLUTONIUM FRACTION BY VOLUME
C      RCP      HEAT CAPACITY (BTU/FT**3-DEG F)
C      COND     CONDUCTIVITY (BTU/SEC-FT-DEG F)
C      COB72060
C      COB72070
C      COB72080
C      COB72090
C      COB72100
C      COB72110
C      COB72120
C      COB72130
C      COB72140
C      COB72150
C      COB72160
C      COB72170
C      COB72180
C      COB72190
C      COB72200
C      COB72210
C      COB72220
C      COB72230
C      COB72240
C      COB72250
C      COB72260
C      COB72270
C      COB72280
C      COB72290
C      COB72300
C      COB72310
C      COB72320
C      COB72330
C      COB72340
C      COB72350
C      COB72360
C      COB72370
C      COB72380
C      COB72390
C      COB72400
C      COB72410
C      COB72420
C      COB72430
C      COB72440
C      COB72450
C      COB72460
C      COB72470
C      COB72480
C      COB72490
C      COB72500
C      COB72510
C      COB72520
C      COB72530
C      COB72540
C      COB72550
C      COB72560
C      COB72570
C      COB72580
C      COB72590
C      COB72600

C-----
C      DIMENSION RC(4), RCM(4), CN(3), CNM(3)
C      DATA RC /1.78E6, 3.62E3, -2.61, 6.59E-4/
C      DATA RCM /1.61E6, 3.72E3, -2.57, 6.13E-4/
C      DATA CN /10.8, -8.84E-3, 2.25E-6/
C      DATA CNM /9.88, -8.44E-3, 2.25E-6/
C      DATA CVTC,CVTRC/1.61E-4, 1.49E-5/
C-----
C      TEM=.5556*(TFUEL+459.7)
C      IF (FPUO2.GT.1.E-7) GO TO 20
C      UO2 FUEL
C      10 RCP = FTD*( RC(1)+ TEM*(RC(2) +TEM*(RC(3) +TEM*(RC(4)))) )
C      BT = 2.74 - TEM * 5.8E-4
C      POR = 1.- BT*(1.- FTD)
C--THE FACTOR /((1.-BT)*(1.-.95)) IS INCORPORATED IN THE FIT CN(3)
C      COND = POR*( CN(1)+ TEM*(CN(2)+ TEM*(CN(3))) )
C      GO TO 100
C      MIXED OXIDE FUEL
C      20 RCP = FTD *(1.+0.045*FPUO2) *
C      * (RCM(1)+ TEM*(RCM(2)+ TEM*(RCM(3)+ TEM*(RCM(4)))) )
C      BT = 2.74 - TEM * 5.8E-4
C      POR = FTD / (1.+ BT*(1.-FTD))
C      THE FACTOR (1.+BT*(1.-.96))/1.96 IS INCORPORATED IN CNM(3)
C      COND = POR*( CNM(1)+ TEM*(CNM(2)+ TEM*(CNM(3))) )
C
C      100 CONTINUE
C      COND CONVERTED FROM (W/M-DEG K) TO (BTU/SEC-FT-DEG F)
C      COND=COND*CVTC
C      RCP CONVERTED FROM (J/M**3-DEG K) TO (BTU/FT**3-DEG F)
C      RCP=RCP*CVTRC
C      RETURN
C      END
C      SUBROUTINE MPG (INIT, BURN, EFFB, FRAC, PRESS, CPR, EXPR, GRGH,
C      THG, RADFU, PG, TG, GMIX, TF, TC, HGAP)
C      CALCULATES GAP HEAT TRANSFER COEFFICIENT, IN THREE PARTS:
C      1. OPEN GAP COMPONENT, BASED ON CONDUCTIVITY OF A MIXTURE OF FOUR

```

```

C      NOBLE GASES; A SMALL GAP CORRECTION IS APPLIED IF PGAS > 0.
C      CONTRIBUTION FROM PARTIAL FUEL-CLAD CONTACT
C      2. CONTRIBUTION FROM PARTIAL FUEL-CLAD CONTACT
C      3. RADIATION COMPONENT
C      4. CLOSED GAP LAW = CPR + (PRESS**EXPR)
C      4. CLOSED GAP LAW = CPR + (PRESS**EXPR)
C      PARTS 1 & 2 ARE BASED ON TREE-NUREG-1005, APPENDIX C, WITH CRACKED
C      PELLET MODEL; PART 4 IS USER-SUPPLIED.
C      MPG IS CALLED WITH INIT = .TRUE. TO PERFORM INITIALIZATION
C      NORMAL CALLS HAVE INIT = .FALSE.
C      ARGUMENTS: INIT = .TRUE.
C      BURN      BURRIUP (MMD/MTU)
C      GRGH      ROOT MEAN SQUARE OF FUEL PELLETT AND CLADDING
C      GRGH      SURFACE ROUGHNESSES (FT)
C      THG       GAP THICKNESS (FT)
C      GRGH      IF GRGH = 0 ON INPUT, A DEFAULT VALUE OF
C      GRGH      1.34E-6 FEET IS RETURNED
C      EFFB       FRACTIONAL EFFECT OF BURNUP, USED IN PARTIAL
C      EFFB       FUEL-CLAD CONTACT MODEL
C      FRAC       FRACTION OF FUEL PERIMETER IN LIGHT CONTACT
C      FRAC       WITH CLAD
C      ARGUMENTS: INIT = .FALSE. (NORMAL ENTRY)
C      FRAC       FRACTION OF FUEL PERIMETER TOUCHING CLAD
C      PRESS      PRESSURE OF FUEL AGAINST CLAD FOR CLOSED GAP
C      CPR        COEFFICIENT OF PRESS
C      EXPR       EXPONENT OF PRESS
C      GRGH       RMS OF FUEL AND CLAD GRGH NESSES (FT)
C      THG        GAP THICKNESS (FT)
C      PG         PRESSURE OF GAS MIXTURE IN GAP, FOR SMALL GAP
C      PG         CORRECTION FACTOR (PSIA)
C      TG         TEMPERATURE OF GAS MIXTURE IN GAP (DEG F)
C      GMIX       FOUR MOLE FRACTIONS OF NOBLE GASES
C      GMIX       1. HELIUM
C      GMIX       2. ARGON
C      GMIX       3. KRYPTON
C      GMIX       4. ZENON
C      THE FOUR ELEMENTS OF GMIX MUST SUM TO 1
C      TF         TEMPERATURE OF FUEL PELLETT SURFACE (DEG F)
C      TC         TEMPERATURE OF INNER CLAD SURFACE (DEG F)
C      HGAP       GAP HEAT TRANSFER COEFFICIENT (BTU/FT**3-DEG F)
C      RETURN
C      LOGICAL INIT
C      DIMENSION GMIX(4)
C      DIMENSION AIR(4,4), BM(4,4)
C      COMBINING FACTORS WHICH ARE FUNCTIONS ONLY OF THE MOLECULAR
C      WEIGHTS OF THE FOUR NOBLE GASES
C      DATA AIR / 0., .295, .232, .194,
C      DATA BM / 0., .362, 0., .309, .332,
C      DATA BM / .413, .235, 0., .286,
C      DATA BM / .435, .260, .232, 0. /
C      DATA BM / 0., 1.78, 2.14, 2.39,

```

```

COB72610
COB72620
COB72630
COB72640
COB72650
COB72660
COB72670
COB72680
COB72690
COB72700
COB72710
COB72720
COB72730
COB72740
COB72750
COB72760
COB72770
COB72780
COB72790
COB72800
COB72810
COB72820
COB72830
COB72840
COB72850
COB72860
COB72870
COB72880
COB72890
COB72900
COB72910
COB72920
COB72930
COB72940
COB72950
COB72960
COB72970
COB72980
COB72990
COB73000
COB73010
COB73020
COB73030
COB73040
COB73050
COB73060
COB73070
COB73080
COB73090
COB73100
COB73110
COB73120
COB73130
COB73140
COB73150

```

```

2      .563, 0., 1.20, 1.35,
3      .467, .831, 0., 1.12,
4      .418, .743, .894, 0. /
DIMENSION CC(4), EE(4), CON(4), CSR(4)
DATA CC / 3.366E-3, 3.421E-4, 4.029E-5, 4.726E-5 /
DATA EE / .668, .701, .872, .923 /
C-----
C
CC CONVERT TO DGAP (M)
  DGAP = THG*.3048
CC TEMPERATURES CONVERTED FROM (DEG F) TO (DEG K)
  TCLAD = .5556*(TC+459.67)
  TGAS = .5556*(TG+459.67)
  TFUEL = .5556*(TF+459.67)
  TGRS = .5556*(TG+459.67)
CC CONVERT TO PGAS (M/M**2)
  PGAS = PG*6.893E3
CC CONVERT TO ROUGH(M)
  ROUGH = GRGH*.3048
C
C IF (INIT) GO TO 200
C
C NOBLE GAS CONDUCTIVITIES
C
CON(1) = 0.
DO 10 I = 1, 4
  IF (GMIX(I).LT.1.E-6) GO TO 10
  CON(I) = CC(I) *(TGAS**EE(I))
  CSR(I) = SQRT(CON(I))
10 CONTINUE
C SMALL GAP CORRECTION FOR HELIUM:
  GAP = AMAX1 (ROUGH, DGAP)
  FAC = PGAS * GAP
  IF (FAC.LT.1.E-9) GO TO 15
  CON(1) = CON(1) / ((1.+ CON(1))*-.2103*SQRT(TGAS)/FAC)
  CSR(1) = SQRT(CON(1))
15 CONTINUE
C MIXTURE CONDUCTIVITY
  GCND = 0.
DO 30 J = 1, 4
  IF (GMIX(J).LT.1.E-6) GO TO 30
  XSUM = GMIX(J)
  DO 20 I = 1, 4
    IF (J.EQ.I) GO TO 20
    IF (GMIX(I).LT.1.E-6) GO TO 20
  15 = CSR(J) + CSR(I)*BM(I,J)
  XSUM = XSUM + GMIX(J)*AM(I,J)*TS*TS/CON(J)
20 CONTINUE
  GCND = GCND + CON(1)*GMIX(I)/XSUM
30 CONTINUE
C
  HGAP = GCND / (DGAP + ROUGH)

```

COB73160  
COB73170  
COB73180  
COB73190  
COB73200  
COB73210  
COB73220  
COB73230  
COB73240  
COB73250  
COB73260  
COB73270  
COB73280  
COB73290  
COB73300  
COB73310  
COB73320  
COB73330  
COB73340  
COB73350  
COB73360  
COB73370  
COB73380  
COB73390  
COB73400  
COB73410  
COB73420  
COB73430  
COB73440  
COB73450  
COB73460  
COB73470  
COB73480  
COB73490  
COB73500  
COB73510  
COB73520  
COB73530  
COB73540  
COB73550  
COB73560  
COB73570  
COB73580  
COB73590  
COB73600  
COB73610  
COB73620  
COB73630  
COB73640  
COB73650  
COB73660  
COB73670  
COB73680  
COB73690  
COB73700

```

C      C      PARTIAL FUEL-CLAD CONTACT MODEL
C      C      HGAP = (1.-FRAC)*HGAP + FRAC*GCOND/ROUGH
C      C      RADIATION HEAT TRANSFER CONTRIBUTION
C      C      REMISF = AMAX1(1.1485, AMINI(2.451, -.154+TFUEL*1.3025E-3 ))
C      C      REMISC = 1.33
C      C      RFVIEW = REMISF + (REMISC-1.)*RADFU/(RADFU+THG)
C      C      HGAP = HGAP +
C      C      + 5.279E-8*(TFUEL+TCLAD)*(TFUEL+TFUEL+TCLAD+TCLAD)/RFVIEW
C      C      CONVERT HGAP FROM (W/M**2-DEG K) TO (BTU/SEC-FT**2-DEG F)
C      C      HGAP=HGAP*4.89E-5
C      C
C      C      CLOSED GAP CONTACT HEAT TRANSFER
C      C      IF (DGAP .GE. ROUGH) RETURN
C      C      HGAP = HGAP + CPR * (PRESS **EXPR)
C      C
C      C      RETURN
C      C
C      C      INITIALIZATION OF MPG, CALLED ONLY ONCE
C      C      200 IF (GRGH.LE.0.) GRGH = 1.34E-6
C      C
C      C      FRACTION OF FUEL IN LIGHT CONTACT WITH CLAD, A FUNCTION OF BURNUP
C      C      C--FRACTIONAL EFFECT OF BURNUP, INDEPENDENT OF FUEL RADIUS
C      C      IF (BURN-600.) 210,210,220
C      C      210  EFFB = 0.
C      C      GO TO 230
C      C      220 CONTINUE
C      C      TS = .001*BURN - .6
C      C      TS = TS*TS
C      C      EFBF = 1.- 1./((TS*TS + 1.))
C      C      230 CONTINUE
C      C      C--FRACTION OF CIRCUMFERENCE OF FUEL IN LIGHT CONTACT WITH CLAD
C      C      A1 = 100. - 98.*EFB
C      C      A2 = 4. - .5*EFB
C      C      FRAC = 1./ (A1*(100.*DGAP/RADFU)**A2 + 1.42857) + .3
C      C      RETURN
C      C      END
C      C      SUBROUTINE MPC (TCL, RCP, COND)
C      C      CALCULATES HEAT CAPACITY AND CONDUCTIVITY OF ZIRCALOY AS A FUNCTION
C      C      OF TEMPERATURE
C      C      C      ARGUMENTS
C      C      C      INPUT      TCL      TEMPERATURE (DEG F)
C      C      C      RETURN    RCP      HEAT CAPACITY (BTU/FT**3-DEG F)

```

```

COB73710
COB73720
COB73730
COB73740
COB73750
COB73760
COB73770
COB73780
COB73790
COB73800
COB73810
COB73820
COB73830
COB73840
COB73850
COB73860
COB73870
COB73880
COB73890
COB73900
COB73910
COB73920
COB73930
COB73940
COB73950
COB73960
COB73970
COB73980
COB73990
COB74000
COB74010
COB74020
COB74030
COB74040
COB74050
COB74060
COB74070
COB74080
COB74090
COB74100
COB74110
COB74120
COB74130
COB74140
COB74150
COB74160
COB74170
COB74180
COB74190
COB74200
COB74210
COB74220
COB74230
COB74240
COB74250

```



```

C      COND      CONDUCTIVITY (BTU/SEC-FT-DEG F)      COB74260
C      COB74270
C      COB74280
C THIS SUBROUTINE IS BASED ON DATA IN TREE-NUREG-1005, APPENDIX B.
C CONDUCTIVITY IS USED UNCHANGED. HEAT CAPACITY HAS BEEN FIT
C LINEARLY IN THE ALPHA PHASE (TEM < 1190), BY A CONSTANT IN THE
C BETA PHASE (TEM > 1254), AND BY AN INVERTED VEE IN THE TRANSITION.
C ERROR IS 5 PER CENT IN THE ALPHA PHASE, 300 < TEM < 1190 DEG K.
C      COB74300
C      COB74310
C      COB74320
C      COB74330
C      COB74340
C      COB74350
C      COB74360
C      COB74370
C      COB74380
C      COB74390
C      COB74400
C      COB74410
C      COB74420
C      COB74430
C      COB74440
C      COB74450
C      COB74460
C      COB74470
C      COB74480
C      COB74490
C      COB74500
C      COB74510
C      COB74520
C      COB74530
C      COB74540
C      COB74550
C      COB74560
C      COB74570
C      COB74580
C      COB74590
C      COB74600
C      COB74610
C      COB74620
C      COB74630
C      COB74640
C      COB74650
C      COB74660
C      COB74670
C      COB74680
C      COB74690
C      COB74700
C      COB74710
C      COB74720
C      COB74730
C      COB74740
C      COB74750
C      COB74760
C      COB74770
C      COB74780
C      COB74790
C      COB74800

```

```

C THIS SUBROUTINE IS BASED ON DATA IN TREE-NUREG-1005, APPENDIX B.
C CONDUCTIVITY IS USED UNCHANGED. HEAT CAPACITY HAS BEEN FIT
C LINEARLY IN THE ALPHA PHASE (TEM < 1190), BY A CONSTANT IN THE
C BETA PHASE (TEM > 1254), AND BY AN INVERTED VEE IN THE TRANSITION.
C ERROR IS 5 PER CENT IN THE ALPHA PHASE, 300 < TEM < 1190 DEG K.

```

```

DIMENSION CN(4)
DATA CN /7.51, 2.09E-2, -1.45E-5, 7.67E-9 /
DATA CVTC,CVTRC/1.61E-4, 1.49E-5/

```

```

C HEAT CAPACITY

```

```

CC CONVERT TO TEM (DEG K)
TEM = .5556*(TCL+459.67)
IF (TEM.GT.1090.) GO TO 20
C ALPHA PHASE: (0 < TEM < 1090 DEG K, USUAL CASE)
RCP = 1673456. + TEM * 721.6
GO TO 50

```

```

C 20 IF (TEM.GE.1254.) GO TO 30
RCP = 5346400. - 36080.*ABS(TEM-1170.)
GO TO 50

```

```

30 RCP = 2315680.

```

```

C 50 CONTINUE

```

```

C CONDUCTIVITY
COND = CN(1) + TEM*(CN(2) + TEM*(CN(3) + TEM*CN(4)))

```

```

CC CONVERT COND FROM (W/M-DEG K) TO (BTU/SEC-FT-DEG F)
COND = COND*CVTC

```

```

CC CONVERT RCP FROM (J/M**3-DEG K) TO (BTU/FT**3-DEG F)
RCP = RCP*CVTRC
RETURN
END

```

```

1 SUBROUTINE HTCOR(LHTR,QV,QL,HVFC,HLNB,HLFC,TM,TL,TV,P,ALP,X,
ROV,ROL,VV,VL,ND,IHTM,CHFR,TSAT,FLUX,NCHF,NN,II,JJ,13)

```

```

C THIS ROUTINE COMPUTES HEAT TRANSFER COEFFICIENTS AND/OR HEAT
C FLUXES

```

```

C THE TOTAL HEAT FLUX IS ASSUMED TO BE OF THE FORM:
C Q=QV+QL+HVFC(TM-TV)+HLNB(TM-TSAT)+HLFC(TM-TL)

```

```

C NORMALLY QV AND QL WILL BE ZERO AND ONE OR MORE OF THE HEAT
C TRANSFER COEFFICIENTS HVFC, HLNb, AND HLFC WILL BE NON-ZERO.
C IN TRANSITION BOILING, HOWEVER, THE HEAT TRANSFER COEFFICIENTS ARE
C ZERO AND Q=QV+QL.
C Nomenclature:

```

C QV HEAT FLUX TO VAPOR (W/M\*\*2) COB74810  
 C QL HEAT FLUX TO LIQUID (W/M\*\*2) COB74820  
 C HVFC CONVECTION HEAT TRANSFER COEFFICIENT TO VAPOR (W/M\*\*2 K) COB74830  
 C HLNB NUCLEATE BOILING HEAT TRANSFER COEFFICIENT (W/M\*\*2 K) COB74840  
 C HLFC CONVECTION HEAT TRANSFER COEFFICIENT TO LIQUID (W/M\*\*2 K) COB74850  
 C TW WALL TEMPERATURE (K) COB74860  
 C TL LIQUID TEMPERATURE (K) COB74870  
 C TV VAPOR TEMPERATURE (K) COB74880  
 C P PRESSURE (P) COB74890  
 C ALP VAPOR VOLUME FRACTION COB74900  
 C ROV VAPOR DENSITY (KG/M\*\*3) COB74910  
 C ROL LIQUID DENSITY (KG/M\*\*3) COB74920  
 C WV VAPOR VELOCITY (M/S) COB74930  
 C VL LIQUID VELOCITY (M/S) COB74940  
 C HD HYDRAULIC DIAMETER (M) COB74950  
 C TSAT SATURATION TEMPERATURE (K) COB74960  
 C COB74970  
 C COB74980  
 C COB74990  
 C COB75000  
 C COB75010  
 C COB75020  
 C COB75030  
 C COB75040  
 C COB75050  
 C COB75060  
 C COB75070  
 C COB75080  
 C COB75090  
 C COB75100  
 C COB75110  
 C COB75120  
 C COB75130  
 C COB75140  
 C COB75150  
 C COB75160  
 C COB75170  
 C COB75180  
 C COB75190  
 C COB75200  
 C COB75210  
 C COB75220  
 C COB75230  
 C COB75240  
 C COB75250  
 C COB75260  
 C COB75270  
 C COB75280  
 C COB75290  
 C COB75300  
 C COB75310  
 C COB75320  
 C COB75330  
 C COB75340  
 C COB75350

C NOTE: THE FOLLOWING QUANTITIES ARE AVAILABLE AND,  
 IF DESIRED, COULD BE ADDED TO THE ARGUMENT LIST OF  
 HFCOR AND THE CORRESPONDING CALL STATEMENT:

C TCHF TEMPERATURE AT CRITICAL HEAT FLUX  
 C TMSFB MINIMUM STABLE FILM BOILING TEMPERATURE  
 C QCHF CRITICAL HEAT FLUX  
 C QMSFB HEAT FLUX AT TMSFB

C COMMON/HYSAVE/BETAV,BETAL,CPV,CPL,MFG,SPVV,SPVL,  
 1 ROVS,ROLS,EV,EL,DTSDP,DELOP,DEVOP,DELDT,DEVDT,  
 2 DRLOP,DRVDP,DRLODT,DRVDT

C COMMON/CHFSV/CHSAVE(20,20,31)

C DATA GCON/9.8066/  
 HVFC=0.0  
 HLFC=0.0  
 HLNB=0.0  
 CHFR=1.0  
 QV=0.0  
 QL=0.0  
 ITR=0  
 VVA=ABS(VV)  
 VLA=ABS(VL)  
 RHD=1./HD  
 IF(JJ.GT.1.OR.II.GT.1) GO TO 4

C PROPERTIES CALCULATED ONCE EACH TIME STEP AND SAVED

C OBTAIN FLUID PROPERTIES  
 C (RUNNING TIME COULD BE SHORTENED BY REPLACING THE  
 C FOLLOWING CALL TO STATE AND THE SUBSEQUENT COMPUTATION OF  
 C HFC, BETAV, BETAL, CPV, AND CPL BY APPROPRIATE FITS TO  
 C THESE QUANTITIES)

C PROPERTIES OBTAINED FROM STATE AT SATURATION TEMP. CORRESP.

```

C TO PRESSURE P.
C
  TSAT1 = 9.0395*POW(P,.223EO) + 255.2
  CALL STATE(P,TSAT1,TSAT1,ROVS,ROLS,EV,EL,TSAT,DTSDP,DELDP,
1  DEVDP,DELDT,DEVDI,DRLOP,DRVDP,DRLOI,DRVDT,2,IEAR)
  SPVV = 1./ROVS
  SPVL = 1./ROLS
  HFG = EV*P*SPVV -EL*P*SPVL
  BETAV = -DRVDT*SPVV
  BETAL = -DRLOI*SPVL
  CPV = DEVDI -P*DRVDT*SPVV*SPV
  CPL = DELDT -P*DRLOI*SPVL*SPVL
4 CONTINUE
  VISV = VISVP(TV)
  VISL = VISLQ(TL)
  CNDV = CONDV(P,TV)
  CNDL = CONDL(P,TL)
  SIG = SURTT(TL)
C
  GV = ALP*ROV*VVA
  GL = (1.-ALP)*ROL*VLA
  G = GV + GL
10 CONTINUE
C
  ... DETERMINE HEAT TRANSFER REGIME ...
C TEST QUALITY
C
  IF(X.GE.0.99)GO TO 300
C TEST FOR COLD WALL
C
  IF(TW.LE.TSAT)GO TO 200
C
  IF(IHTH.LT.2)GO TO 30
C
C COMPUTE MINIMUM STABLE FILM BOILING TEMPERATURE
C
  IF (P.GT.68.96E5) GO TO 20
  THN = 581.5 + .01676*SQRT( AMAX1(P-1.0345E5,(0.)) )
  GO TO 25
20 THN = 630.37 + .00432*SQRT(P-68.96E5)
25 CONTINUE
  PSI=0.0
  IF (P.LT.4.827E5) PSI = 127.3 - 26.37E-5*P
  CALL MPC(TW,RCP,COND)
  RRKCPW = 1./ (RCP*COND)
C INVERSE OF RCP OF ZIRCALOY TIMES CONDUCTIVITY OF OXIDE
  RRKCPW = 3.1E-7 - 1.3E-10*TW
  RKCP = ROL*CNDL*CPL
C
  TMSFB = THN + (THN-TL)*POW(RKCP/RRKCPW,.5E0) - PSI
C TEST WHETHER TWALL EXCEEDS TMSFB
C

```

COB75360  
 COB75370  
 COB75380  
 COB75390  
 COB75400  
 COB75410  
 COB75420  
 COB75430  
 COB75440  
 COB75450  
 COB75460  
 COB75470  
 COB75480  
 COB75490  
 COB75500  
 COB75510  
 COB75520  
 COB75530  
 COB75540  
 COB75550  
 COB75560  
 COB75570  
 COB75580  
 COB75590  
 COB75600  
 COB75610  
 COB75620  
 COB75630  
 COB75640  
 COB75650  
 COB75660  
 COB75670  
 COB75680  
 COB75690  
 COB75700  
 COB75710  
 COB75720  
 COB75730  
 COB75740  
 COB75750  
 COB75760  
 COB75770  
 COB75780  
 COB75790  
 COB75800  
 COB75810  
 COB75820  
 COB75830  
 COB75840  
 COB75850  
 COB75860  
 COB75870  
 COB75880  
 COB75890  
 COB75900

```

C      IF(TW.LT.TMSFB)GO TO 30
C      COB75910
C      COB75920
C      COB75930
C      COB75940
C      COB75950
C      COB75960
C      COB75970
C      COB75980
C      COB75990
C      COB76000
C      COB76010
C      COB76020
C      COB76030
C      COB76040
C      COB76050
C      COB76060
C      COB76070
C      COB76080
C      COB76090
C      COB76100
C      COB76110
C      COB76120
C      COB76130
C      COB76140
C      COB76150
C      COB76160
C      COB76170
C      COB76180
C      COB76190
C      COB76200
C      COB76210
C      COB76220
C      COB76230
C      COB76240
C      COB76250
C      COB76260
C      COB76270
C      COB76280
C      COB76290
C      COB76300
C      COB76310
C      COB76320
C      COB76330
C      COB76340
C      COB76350
C      COB76360
C      COB76370
C      COB76380
C      COB76390
C      COB76400
C      COB76410
C      COB76420
C      COB76430
C      COB76440
C      COB76450

IF(TW.LT.TMSFB)GO TO 30
C COMPUTE FILM BOILING HEAT TRANSFER COEFFICIENT
C
CALL FILM(HVFC,ALP,ROV,ROL,VVA,VLA,HD,RHD,TL,TV,TW,TSAT,MFG,
1  CPV,CPL,P,VISV,VISL,BETAV,SIG,INTR,X)
GO TO 1000
C
C 30 CONTINUE
C
C DETERMINE HEAT TRANSFER COEFFICIENTS USING CHEN CORRELATION
C
RVISL = 1./VISL
XTTI=POW(X/(1.-X),.9EO) *SQR(ROL/ROV) *POW(VISV*RVISL,.1EO)
F=1.0
GX = G
IF(TL.LT.TSAT) GO TO 32
IF(XTTI.GT.0.1)F=2.35*POW(XTTI+.213EO,.736EO)
GX = GL
32 PRL = VISL*CPL/CIIDL
REL = GX*HD*RVISL
HLF = .023*F*CNDL*RHD* POW(REL,.8EO) *POW(PRL,.4EO)
RETP = REL *POW(F,1.25EO)*1.E-4
S=.1
IF(RETP.LT.70.0.AND.RETP.GE.32.5) S=1./((1+.42*POW(RETP,.78EO))
IF(RETP.LT.32.5) S=1./((1+.12*POW(RETP,1.14EO))
HS = .0012*S*SQR(CNDL*CPL/(SIG*GCON)) *POW(PRL,.29EO) *
* POW(ROL,.25EO) *POW(CPL*RDL/(MFG*ROV),.24EO)
PWALL = (.11062558*(TW-255.2))**4.4843049
HLN = HS*POW(TW-TSAT,.24EO)*POW(PWALL-P,.75EO)
C COMPUTE HEAT FLUX AS PREDICTED BY CHEN'S CORRELATION AND
C COMPARE AGAINST THE CRITICAL HEAT FLUX
C
QCHEN = HLF*(TW-TL) + HLN*(TW-TSAT)
IF(IHTM.LT.2) GO TO 400
C CALCULATE CRITICAL HEAT FLUX
C
CC BTU/S-FT**2 = 11400. W/M**2
C
CVTHF=11400.
IF (NCHF.EQ.5.AND.(NN.GT.20.OR.II.GT.20.OR.JJ.GT.30)) GOTO 2000
IF (NCHF.EQ.1) QCHF=CVTHF*CHF1(NN,II,JJ+1)
IF (NCHF.EQ.2) QCHF=CVTHF*CHF2(NN,II,JJ+1)
IF (NCHF.EQ.3) QCHF=CVTHF*CHF3(NN,II,JJ+1)
IF (NCHF.EQ.4) QCHF=CVTHF*CHF4(NN,II,JJ+1)*FLUX
IF (NCHF.EQ.5) CALL CHF5(QCHF,ALP,ROV,ROL,G,P,X,HD,MFG,SIG)
CHSAVE (NN,II,JJ)=QCHF/CVTHF
C
IF(QCHEN.LE.QCHF) GO TO 400
C
C

```

```

C SOLVE THE EQUATION
C HLFC=(TCHF-TL)+HLNB*(TCHF-TSAT)**1.24*(PWALL-P)**.75 = QCHF
C FOR TCHF USING NEWTON'S ITERATION
C
    TCHF=AMAX1(TL,TSAT+.1)
    DO 35 K=1,10
        TCS=AMAX1(TCHF-TSAT,(0.))
        PWALL=(.11062558*(TCHF-255.2))**.4+.4843049
        DQ = QCHF-HLF*(TCHF-TL)-HS*POW(TCS,1.24E0)*POW(PWALL-P,.75E0)
        DDOT = HLF + HS*POW(TCS,.24E0)*POW(PWALL-P,.75E0)
        * ((1.24 + 3.3632287*TCS*PWALL/((TCHF-255.2)*(PWALL-P)))
        DTCHF = DQ/DDOT
        TCHF = TCHF + DTCHF
        IF(ABS(DTCHF).LE.0.1)GO TO 40
    35 CONTINUE
    40 CONTINUE
    GO TO 500

C
C ... INDIVIDUAL CORRELATIONS FOLLOW ...
C
C CONVECTION TO SINGLE PHASE LIQUID
C MAX OF SIEDER-TATE AND MCADAMS CORRELATIONS
C
C NOTE: MCADAMS SHOULD EVALUATE PROPERTIES AT A LIQUID FILM TEMP
C
    200 CONTINUE
    T1=ROL*ROL*GCON*BETAL*CPL*ABS(TM-TL)/(VISL*CNDL)
    HMA=.13*CNDL*POW(T1,.333333E0)
    REL=ROL*VLA*HD/VISL
    PRL=VISL*CPL/CNDL
    VISW=VISLQ(TW)
    HST=.023*CNDL*RHD*POW(REL,.8E0)*POW(PRL,.33E0)*
    1 POW(VISL/VISW,.14E0)
    HLFC=AMAX1(HMA,HST)
    CHFR=100.0
    IHR=1
    IF(HMA.GT.HST) IHR=2
    GO TO 1000

C
C CONVECTION TO SINGLE PHASE VAPOR
C MAX OF SIEDER-TATE AND MCADAMS CORRELATIONS
C
C NOTE: MCADAMS SHOULD EVALUATE PROPERTIES AT A VAPOR FILM TEMP
C
    300 CONTINUE
    T1=ROV*ROV*GCON*BETAV*CPV*ABS(TM-TV)/(VISV*CNDV)
    HMA=.13*CNDV*POW(T1,.333333E0)
    REV=ROV*VVA*HD/VISV
    PRV=VISV*CPV/CNDV
    VISW=VISVP(TW)
    HST=.023*CNDV*RHD*POW(REV,.8E0)*POW(PRV,.33E0)*
    1 POW(VISV/VISW,.14E0)
    HVFC=AMAX1(HMA,HST)
    IHR=9

```

COB76460  
COB76470  
COB76480  
COB76490  
COB76500  
COB76510  
COB76520  
COB76530  
COB76540  
COB76550  
COB76560  
COB76570  
COB76580  
COB76590  
COB76600  
COB76610  
COB76620  
COB76630  
COB76640  
COB76650  
COB76660  
COB76670  
COB76680  
COB76690  
COB76700  
COB76710  
COB76720  
COB76730  
COB76740  
COB76750  
COB76760  
COB76770  
COB76780  
COB76790  
COB76800  
COB76810  
COB76820  
COB76830  
COB76840  
COB76850  
COB76860  
COB76870  
COB76880  
COB76890  
COB76900  
COB76910  
COB76920  
COB76930  
COB76940  
COB76950  
COB76960  
COB76970  
COB76980  
COB76990  
COB77000

```

IF(HMA,GT,HST) IHTR=10
GO TO 1000
C
C SUBCOOLED OR SATURATED NUCLEATE BOILING
C CHEN CORRELATION
C
400 CONTINUE
HLFC = HLF
HLNB = HLN
IHTR=4
IF(TL,LT,TSAT) IHTR=3
GO TO 1000
C
C TRANSITION BOILING
C
500 CONTINUE
CALL FILM(HVTB,ALP,ROV,ROL,VVA,VLA,HD,RHD,TL,TV,TMSFB,TSAT,MFG,
1 CPV,CPL,P,VISV,VISL,BETAV,SIG,IHTR,X)
RDIMC = 1./((TMSFB-TW)*QCHF)
EPS = (TMSFB-TW)*RDIMC
EPS2 = EPS*EPS
QMSFB=HVTB*(TMSFB-TV)
QV=(1.-EPS2)*QMSFB
QL=EPS2*QCHF
DOLDTW = -2.*EPS*QCHF*RDIMC
DOVDTW = 2.*EPS*QMSFB*RDIMC
HLFC = DOLDTW
QL = QL + DOLDTW*(TL-TW)
HVFC = DOVDTW
QV = QV + DOVDTW*(TV-TW)
IHTR=5
C
C
C
1000 CONTINUE
RETURN
2000 WRITE(13,2020)
2020 FORMAT(1H,' ERROR DETECTED IN SUBROUTINE HTCOR. ATTEMPT TO USE',
1 ' NCHF=5 OPTION FOR TOO LARGE A PROBLEM.')
CALL EXIT
END
SUBROUTINE FILM(H,ALP,ROV,ROL,VVA,VLA,HD,RHD,TL,TV,TW,TSAT,MFG,
1 CPV,CPL,P,VISV,VISL,BETAV,SIG,IHTR,X)
DATA GCON,PI2/9.8066,6.2831853/
C
C NOTE: IN PROBLEY'S AND MCADAMS' CORRELATIONS VAPOR PROPERTIES
C ARE EVALUATED AT BULK VAPOR TEMPERATURE AND NOT
C AT VAPOR FILM TEMPERATURE.
C IN GROENEVELD'S CORRELATION THE VAPOR PRANDTL NUMBER
C IS EVALUATED AT BULK VAPOR TEMPERATURE AND NOT
C AT WALL TEMPERATURE.
C
C HIGH FLOW FILM BOILING
C GROENEVELD 5.7 OR MODIFIED DITTUS-BOELTER (FOR LOW PRESSURE)

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COB77010  
COB77020  
COB77030  
COB77040  
COB77050  
COB77060  
COB77070  
COB77080  
COB77090  
COB77100  
COB77110  
COB77120  
COB77130  
COB77140  
COB77150  
COB77160  
COB77170  
COB77180  
COB77190  
COB77200  
COB77210  
COB77220  
COB77230  
COB77240  
COB77250  
COB77260  
COB77270  
COB77280  
COB77290  
COB77300  
COB77310  
COB77320  
COB77330  
COB77340  
COB77350  
COB77360  
COB77370  
COB77380  
COB77390  
COB77400  
COB77410  
COB77420  
COB77430  
COB77440  
COB77450  
COB77460  
COB77470  
COB77480  
COB77490  
COB77500  
COB77510  
COB77520  
COB77530  
COB77540  
COB77550

```

C
CNDV = CONDVP(P,TV)
REV = HD*ROV*(VLA*ALP*(VVA-VLA))/VISV
PRV = VISV*CPV/CNDV
IF(P.LT.1.33E6) GO TO 10
Y = 1.-1*POW((1.-X)*((ROL/ROV)-1.),.4E0)
HGDB = .052*CNDV*RHD +POW(REV,.688E0)*POW(PRV,1.26E0)*
1 POW(Y,-1.06E0)
IHDR = 6
GO TO 20
10 HGDB = .023*CNDV*RHD +POW(REV,.800E0)*POW(PRV,0.40E0)
20 CONTINUE
H = HGDB
C
C TEST FOR LOW OR HIGH FLOW
C
C AUG = ALP +ROV*VVA/SQRT(GCON*HD*ROV*(ROL-ROV))
AJF = (1.-ALP)*ROL*VLA/SQRT(GCON*HD*ROL*(ROL-ROV))
AJ = SORT(AJG)*SQRT(AJF)
IF(AJ.GE.2.0) RETURN
C
C LOW FLOW FILM BOILING
C
C BROWLEY PLUS MAX OF MCADAMS AND FORCED CONVECTION(AS FOR HIGH FLOW)
C
CLAM = PI2*SQRT(SIG/(ROL-ROV))
HFGP = HFG+0.5*CPV*(TW-TSAT)
T1 = GCON*(ROL-ROV)*ROV*(CNDV*.3)*HFGP/(CLAM*VISV*(TW-TSAT))
HMB = .62*POW(T1,.25E0)
C
T1 = ROV*ROV*GCON*BETAV*CPV*ABS(TW-TV)/(VISV*CNDV)
HMA = .13*CNDV*POW(T1,.333333E0)
C
H = (1.-ALP)*HMB + ALP*HMAX1(HGDB,HMA)
IHDR = 8
C
RETURN
END
SUBROUTINE CHF5(QCHF,ALP,ROV,ROL,G,P,X,HD,HFG,SIG)
C DETERMINES CRITICAL HEAT FLUX
C
DATA GCON/9.8066/
DATA EE /2.7182818/
C
PBAR=1.0E-5+P
GHI=1350.0
GLO=27.0
IF(PBAR.GE.83.0.AND.X.GE.0.5)GHI=270.0
IF(S.LT.GLO)GO TO 20
C
C BIASI CORRELATION FOR HIGH FLOW
C
EN=-0.4
IF(HD.LT.0.01)EN=-0.6

```

COB77560  
COB77570  
COB77580  
COB77590  
COB77600  
COB77610  
COB77620  
COB77630  
COB77640  
COB77650  
COB77660  
COB77670  
COB77680  
COB77690  
COB77700  
COB77710  
COB77720  
COB77730  
COB77740  
COB77750  
COB77760  
COB77770  
COB77780  
COB77790  
COB77800  
CU577810  
COB77820  
COB77830  
COB77840  
COB77850  
COB77860  
COB77870  
COB77880  
COB77890  
COB77900  
COB77910  
COB77920  
COB77930  
COB77940  
COB77950  
COB77960  
COB77970  
COB77980  
COB77990  
COB78000  
COB78010  
COB78020  
COB78030  
COB78040  
COB78050  
COB78060  
COB78070  
COB78080  
COB78090  
COB78100

```

C      GT=AMAX1(G,GHI)
C      Q10=0.0
C      IF(GT.LT.300.0)GO TO 10
C      F=.7249 + .099*PBAR*POW(EE,-(.032)*PBAR)
C      GG=POW(GT,(-.166667))
C      Q10=2.764E7*POW(100.E0*HD,EN)*GG*(1.468*F*G6-X)
10 CONTINUE
C      H=-1.159 + .149*PBAR*POW(EE,-.019E0*PBAR) + 8.99*PBAR/
C      1 (10.+PBAR*PBAR)
C      Q11=15.04BE7*H*POW(100.E0*HD,EN)*POW(GT,(-.6))*(1.0-X)
C      QB=AMAX1(Q10,Q11)
C      QCHF=QB
C
C      IF(G.GE.GHI)GO TO 100
20 CONTINUE
C
C      CHF=VOID CORRELATION FOR LOW FLOW
C
C      T1=SIG*GCON*GCON*(ROL-ROV)*ROV*ROV
C      QVC=.1179*(1.-ALP)*HFC*POW(T1, (.25))
C      QCHF=QVC
C
C      IF(G.LE.GLO)GO TO 100
C
C      LINEAR INTERPOLATION BETWEEN BIASI AND CHF=VOID
C
C      WT=(G-GLO)/(GHI-GLO)
C      QCHF=WT*QB+(1.-WT)*QVC
C
C      100 CONTINUE
C      RETURN
C      END
C      FUNCTION POW(A,B)
C
C      THIS FUNCTION IS CALLED WHENEVER A LOW ACCURACY EXPONENTIATION
C      WOULD BE ADEQUATE
C
C      POW=A**B
C      RETURN
C      END
C      FUNCTION CONDL (P, TL)
C
C      THERMAL CONDUCTIVITY OF LIQUID WATER
C      W/M DEG K      FUNCTION OF      PASCAL,  DEG K
C
C      ERROR OF APPROXIMATION < 5 PERCENT FOR 273 < TL < 573 DEG K
C      C VALUE AT 150 BAR, 300 DEG C = .55
C
C      TS = TL - 415.
C      CONDL = .686 - 5.87E-6*TS*TS + 7.3E-10*P
C      RETURN
C      END
C      FUNCTION CONDV (P, TV)

```

COB78110  
COB78120  
COB78130  
COB78140  
COB78150  
COB78160  
COB78170  
COB78180  
COB78190  
COB78200  
COB78210  
COB78220  
COB78230  
COB78240  
COB78250  
COB78260  
COB78270  
COB78280  
COB78290  
COB78300  
COB78310  
COB78320  
COB78330  
COB78340  
COB78350  
COB78360  
COB78370  
COB78380  
COB78390  
COB78400  
COB78410  
COB78420  
COB78430  
COB78440  
COB78450  
COB78460  
COB78470  
COB78480  
COB78490  
COB78500  
COB78510  
COB78520  
COB78530  
COB78540  
COB78550  
COB78560  
COB78570  
COB78580  
COB78590  
COB78600  
COB78610  
COB78620  
COB78630  
COB78640  
COB78650



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C COB78660
C COB78670
C COB78680
C COB78690
C COB78700
C COB78710
C COB78720
C COB78730
C COB78740
C COB78750
C COB78760
C COB78770
C COB78780
C COB78790
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C COB79110
C COB79120
C COB79130
C COB79140
C COB79150
C COB79160
C COB79170
C COB79180
C COB79190
C COB79200

C THERMAL CONDUCTIVITY OF DRY STEAM
C W/M DEG K FUNCTION OF PASCAL, DEG K
C ERROR OF APPROXIMATION < 10 PERCENT FOR 373 < TV < 623 AND
C P IN SUPERHEATED REGION
C FOR LOW P, CONDV DEPENDS MORE ON TV, FOR P > 50 BAR CONDV DEPENDS
C MORE ON P.
C VALUE AT SATURATION FOR 70 BAR = .061
      CONDV = -.0123 + P*(7.8E-9 + P*2.44E-16) +
      + 1.25E-11*TV*(80.ES - P)
      RETURN
      END
      FUNCTION VISLQ (TL)
C VISCOSITY OF SATURATED LIQUID WATER
C KG/M SEC FUNCTION OF DEG K
C ERROR OF APPROXIMATION = 6 PERCENT FOR 273 < TL < 623 DEG K
C MAY ALSO BE USED FOR NON-SATURATED CONDITIONS AT SAME TL
C THIS FIT HAS A SINGULARITY AT TL = 251 DEG K
C VALUE AT 250 DEG C = .107E-3
      VISLQ = 25.3 / (-8.58E4 + TL*(91.+ TL))
      RETURN
      END
      FUNCTION VISVP (TV)
C VISCOSITY OF SATURATED STEAM
C KG/M SEC FUNCTION OF DEG K
C ERROR OF APPROXIMATION = 3 PERCENT FOR 373 < TV < 623 DE K
C MAY ALSO BE USED FOR NON-SATURATED CONDITIONS AT SAME TV
C THIS FIT HAS A SINGULARITY AT TV = 822 DEG K
C VALUE AT 250 DEG C = .174E-4
      IF (TV.GT.623.) GO TO 50
      VISVP = 11.4 / (1.37E6 - TV*(844.+ TV))
      RETURN
50 VISVP = 4.07E-8*TV-3.7E-7
      RETURN
      END
C FUNCTION TCON(T)
C CONVERTS FROM F TO K
      TCON = 5./9.*(T-32.)*273.15
      RETURN
      END
C FUNCTION DCON(RHO)
C CONVERTS FROM LB/FT**3 TO KG/M**3
      DCON = RHO*16.0185
      RETURN
      END
      FUNCTION SURT (TL)

```

C SURFACE TENSION OF LIQUID WATER  
C KG(F)/M FUNCTION OF DEG K  
C ( 1 KG(F) = 9.80665 KG W/SEC\*\*2 )  
C ALSO EQUAL TO SURFACE TENSION / GRAVITATIONAL ACCELERATION CONSTANT  
C IN UNITS OF KG/M  
C  
C ERROR OF APPROXIMATION = 2 PERCENT FOR 373 < TL < 623 DEG K  
C VALUE AT 250 DEG C = .0026  
C  
C SURT = (80.72 - TL\*.126) / (5140.+ TL)  
C IF(SURT.LT.0.0) SURT=0.0  
C RETURN  
C END

COB79210  
COB79220  
COB79230  
COB79240  
COB79250  
COB79260  
COB79270  
COB79280  
COB79290  
COB79300  
COB79310  
COB79320  
COB79330  
COB79340

## APPENDIX O

### Sample Input and Output for the Improved Version of COBRA-IIIC/MIT

Sample input and output for the improved version of COBRA-IIIC/MIT is presented in this section. Sample output is given for the PWR and BWR transient test case described in Section IV.C. Both sample output decks select the new fuel rod modeling option. Sample output obtained from the BWR transient test case sample input is given. The sample output was shortened by removing the pages of output for predictions between 0.0 and 2.5 seconds.

Sample Input Deck for BWR Transient Test Case-Page 1 of 1

```

2      1      10      2      40 ← first card
2000
2      1      2      BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL
20
1      2      1
26.1512
0. .430.0208 .470.0625 .550.1042 .640.1458 .740.1875 .850.2292 .970
.2708 1.10.3125 1.21.3542 1.29.3958 1.34.4375 1.38.4792 1.40.5208 1.39
.5630 1.36.6042 1.30.6458 1.23.6875 1.15.7292 1.08.7708 1.01.8125 .930
.8542 .840.8958 .740.9375 .600 .979 .430 1.00 .350
1.40 1.04
150. 20 50 2.5
2 2 9 3 6 0 1 0 1
1.0E-02
1 1 1. 62. .48315.82118.394.08
33. 1. 10.
1 1 1. 62. .48315.82118.394.08
33. 1. 9.0
2
0.01 1.0714 2.2143 2.3571 2.5000 2.6429 2.7857 2
.9289 2 0.99 3
2. .08 640. .4100 8.80 .076 405. .0320500.9
4 1.0045
.95 0.
1 1 1 1
2
1 2
1
3 0.5 0.5
0528.6 1.101031.
3 3 5
0.0 1.0 2.01.165 2.51.165
0.0 1.0 2.0 0.7 5.0 0.5
0.0 1.0 0.8 1.5 1.5 2.31 2.0 1.0 2.5 0.25
5 2

```

\* \* END OF CARD DECK

Sample Input Deck for PWR Transient Test Case-Page 1 of 2

9 18 10 15 20 ← first card  
2000  
0 1 .1 MAINE YANKEE - 3 PUMP LOF TRANSIENT NEW FR MODEL  
20  
3 6 6  
0 8 8 8 8 8  
0 2 1 0 8 8  
6 4 3 5 5 8  
8 0 7 0 6 8  
8 8 8 8 8 8  
9 9 9 9 9 9  
21-1821  
0. .100 .05 .175 .10 .250 .15 .350 .20 .450 .25 .575 .30 .700  
.35 .900 .40 1.10 .45 1.25 .50 1.40 .55 1.52 .601.640 .651.660  
.701.680 .751.590 .801.500 .851.275 .901.050 .95 .710 1.0 .35  
136.7 20 20 5.  
0 5 9 3 7 0 1 0 1  
1.0E-02  
1 1 1. 1. .44.18431.3821.392  
1.105.46051.015  
1 1 1. 0.85 .44.23091.6951.178  
1.105.46051.015  
2 E  
1 1 1. 0.40 .44.0918.9083.5496  
1.105.46051.015  
4  
1 1 1. 152.0 .4433.00251.0210.1  
1.105.46051.015  
8  
1 1 1. 4418. .44895.86813.6107.  
1.105.46051.015  
9  
.0050 1.0877 2.2194 2.3511 2.4828 2.6144 2.7461 2  
.8778 2 .995 3  
15 15 7 1  
1 1 .441.475 1 .2554 8 .7692  
1 2 .441.475 1 .2564 8 .5128  
1 3 .441.475 2 .3089 8 .7166  
1 4 .441.475 1 .2564 3 .2867 7 .2564 8 .2564  
1 5 .441.611 1 .2442 2 .2942 3 .2730  
1 6 .441.475 3 .2857 5 .2564 7 .2564 8 .2564  
1 7 .441.475 3 .2857 4 .2339 5 .2554 6 .3069  
1 8 .441.475 5 .2564 8 .7692  
1 9 .441.475 5 .2564 6 .2564 8 .5128  
1 10 .441.475 6 .3089 8 .7166  
1 11 .441.475 7 .2564 8 .7692  
1 12 .441.475 7 .2564 8 .7692  
1 13 .441.264 8 168.0  
1 14 .44.9495 9 4716.  
1 15 .441.711 4 .1943  
1.5 0.08 650 .3675 8.8 .078 410. .028 600.  
5 1.0075  
.95 0.  
.2796 .280 .140.1396 .140 .280.1396 .140 .140.2796 .140 .140 .4207.280  
2 1 0 1 1 1 0 1  
.0062 -.10  
1 .184 -0.2  
1 0  
2

Sample Input Deck for PWR Transient Test Case-Page 2 of 2

```
.95 .95 .95 .95 .95 .95 .95 .951.002
2 0.5 0.2413 0.
30 1 600.2600.
1 546. 2.292200.
      7 7
0.0 1.0 1.0 0.95 2.0 0.89 3.0 0.84 4.00.805 5.00.755 6.00.730
0.0 1.0 2.9 1.0 4.0 0.67 4.4 0.49 4.6 0.37 4.8 0.19 5.0 0.14
      2 3 2 3
3 4
5 14 15
```

\*\*\* END OF CARD DECK \*\*\*

Sample Output for BWR Transient Test Case

PROBLEM SIZE  
MC= 2  
MG= 1  
MN= 10  
MR= 2  
MX= 40

INPUT FOR CASE 1 BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL DATE 9/24/80 TIME 14:48:31

SIMILAR CHANNELS ALL SEPARATED EG.BWR





3-LOGICAL

INDEX	NAME	LENGTH	ORL..N	TYPE
1	A	2	1	1
2	AAA	1	3	1
3	AC	2	4	1
4	ALPHA	2	6	1
5	AN	2	8	1
6	ANSWER	1	10	1
7	B	1	11	1
8	CCHANL	82	12	2
9	CD	10	94	1
10	CHFR	82	104	1
11	CON	2	186	1
12	COND	1	188	1
13	CP	2	189	1
14	D	2	191	1
15	DC	2	193	1
16	DFDX	2	195	1
17	DHDX	2	197	1
18	DHYD	2	199	1
19	DHYDN	2	201	1
20	DIST	8	203	1
21	DPDX	2	211	1
22	OPK	2	213	1
23	DUR	1	215	1
24	DR	2	218	1
25	F	82	218	1
26	FACTOR	1	300	1
27	FDIV	1	301	3
28	FINLET	2	302	1
29	FLUX	82	304	1
30	FNULT	2	386	1
31	FOLD	82	388	1
32	FSP	2	470	1
33	FSPLIT	2	472	1
34	FAPLOW	5	474	1
35	GAP	1	479	1
36	GARN	1	480	1
37	GAPS	4	481	1
38	H	82	485	1
39	HFILM	2	567	1
40	HINLET	2	568	1
41	HOLD	82	571	1
42	HPERIM	2	653	1
43	IDAREA	2	655	2
44	IDFUEL	2	657	2
45	IDGAP	1	659	2
46	IK	1	660	2
47	JBOIL	2	661	2
48	JK	1	663	2
49	LC	8	664	2
50	LENGTH	1	672	1
51	LCCA	14	673	2
52	LR	12	687	2
53	MCHFR	41	699	1
54	MCHFRFC	41	740	2
55	MCHFRR	41	781	2
56	NTYPE	2	822	2
57	NWRAP	2	824	2
58	NWRAPS	2	826	2
59	P	82	826	1
60	PERIM	2	910	1
61	PH	2	912	1
62	PHI	12	914	1

63	PRINTC	2	928	2
64	PRINTR	2	3	2
65	PRINTN	10	JO	2
66	PW	2	940	1
67	PWRF	4	942	1
68	QC	82	946	1
69	QF	82	1028	1
70	QPRIM	2	1110	1
71	QUAL	2	1112	1
72	RADIAL	2	1114	1
73	RHO	82	1116	1
74	RHOOLD	82	1198	1
75	SP	41	1200	1
76	T	2	1221	1
77	TDUMY	10	1323	1
78	TINLET	2	1333	1
79	TROD	820	1335	1
80	U	2	2155	1
81	UH	2	2157	1
82	SAVE	1	2159	1
83	USTAR	1	2160	1
84	V	2	2161	1
85	VIJC	2	2163	1
86	VISCW	2	2165	1
87	VP	2	2167	1
88	VPA	2	2169	1
89	W	41	2171	1
90	WOLD	41	2212	1
91	WP	1	2253	1
92	WSAVE	1	2254	1
93	X	41	2255	1
94	XCROSS	6	2298	1
95	A	30	2302	1
96	B	10	2332	1
97	XPOLD	82	2342	1

DYNAMIC ALLOCATION OF CORE GOT 80000 WORDS

DYNAMIC STORAGE REQUIRED = 2423 WORDS

REGION SIZE ON JCL CARD COULD HAVE BEEN REDUCED BY 303 K

PROCESSED INPUT DATA

\* - SET IN NEUTRONICS (CARD20)

OPERATING CONDITIONS

PRESSURE (PSIA) = 1031.00  
 AV. INLET MASS VELOCITY (MLB/SOFT-HR) = 1.1000  
 IN-0 INLET ENTHALPY (BTU/LB) = 528.800  
 \*CHANNEL LENGTH (IN) = 150.00  
 \*NO. OF AXIAL INTERVALS = 20  
 \*NO. OF TIME STEPS = 80  
 \*TOTAL TIME OF TRANSIENT (SEC) = 2.80

FORCING FUNCTION FOR PRESSURE

TIME PRESSURE  
 (SEC) FACTOR  
 0.0 1.0000  
 2.0000 1.1650  
 2.5000 1.1650

FORCING FUNCTION FOR INLET FLOW

TIME INLET FLOW  
 (SEC) FACTOR  
 0.0 1.0000  
 2.0000 0.7000  
 5.0000 0.5000

FORCING FUNCTION FOR HEAT FLUX

TIME HEAT FLUX  
 (SEC) FACTOR  
 0.0 1.0000  
 0.8000 1.5000  
 1.5000 2.3100  
 2.0000 1.0000  
 2.5000 0.2500

CHANNEL, ROD AND GRID DATA

(1=PMR, 2=BR)

REACTOR TYPE = 2  
 \*NO. FUEL ASSEMBLIES = 2  
 \*NO. ASSEMBLY TYPES = 2  
 \*NO. GRIDS = 9  
 \*NO. GRID TYPES = 3  
 \*NO. FUEL NODES = 6  
 \*NO. FCD FLOW TYPES = 0  
 FUEL ROD MODEL IND. = 1  
 HEAT TRANSFER MODEL IND. = 0  
 FUEL ROD PROP. IND. = 1

CHANNEL DATA

\*CHANNEL NUMBERING MAP

TYPE	CHANNEL NUMBERS
1	2
2	2

TYPE	FRIC	AREA SQ FT	WT PER FT	HT PER FT	NO. RODS	ROD DIA IN	GAP IN
1	1	0.10986	9.858	7.840	62.	0.4830	0.0
2	1	0.10986	9.858	7.840	62.	0.4830	0.0

GRID DATA

NO. GRIDS	NO. GRID TYPES	NO. GRID TYPES TYPE AT X/L	GRID COEFF FOR GRID TYPES 1 - 3
3	3	0.9900	1 0.0100 2 0.0714 2 0.2143 2 0.3571 2 0.5000 2 0.6429 2 0.7857 2 0.9289

ASSY. TYPE	GRID COEFF FOR GRID TYPES 1 - 3
1	33.0000 1.0000 10.0000
2	33.0000 1.0000 9.0000

THERMAL PROPERTIES FOR FUEL MATERIAL 6 RADIAL FUEL NODES

FUEL PROPERTIES				CLAD PROPERTIES					
TYPE	COND. NO. (B/HR-FT-F)	SP. HEAT (B/LB-F)	DENSITY (LB/FT <sup>3</sup> )	DIA. (IN.)	COND. (B/HR-FT-F)	SP. HEAT (B/LB-F)	DENSITY (LB/FT <sup>3</sup> )	THICK. (IN.)	GAP COND. (B/HR-FT <sup>2</sup> -F)
1	2.00	0.0800	640.0	0.4100	8.80	0.0760	405.0	0.0320	500.90

NEW FUEL ROD MODEL  
 -----  
 NUMBER OF FUEL PELLETT NODES = 4  
 NUMBER OF CLAD NODES = 1  
 GAP THICKNESS(IN) = 0.45000E-02

FUEL AND CLAD PROPERTIES WILL BE CALCULATED USING FUEL ROD TEMPERATURES.  
 FRACTION THEORETICAL DEN(FUEL) = 0.95000E+00  
 FRACTION PUO2 = 0.0

(1) MIXING  
 -----

MIXING COEFFICIENT (W/GS) = 0.020\* (RE\*\* 0.0)  
 TWO-PHASE MIXING SAME AS SINGLE PHASE (NBBC=1)  
 NO THERMAL CONDUCTION (GK=0.0)

(2) SINGLE-PHASE FRICTION F = A\*(RE\*\*B) + C  
 -----

WVISC = 0 (=0 FOR NO WALL VISCOSITY CORRECTION, =1 FOR INCLUSION)

FRIC TYPE	A	B	C
1	0.1840	-0.2000	0.0
2	0.1840	-0.2000	0.0
3	0.1840	-0.2000	0.0
4	0.1840	-0.2000	0.0

(3) TWO-PHASE FRICTION  
 -----

J4 = 2 (J4=0 HOMOGENEOUS, -1 ARMAND, -2 BAROCZY, -5 POLYNOMIAL IN QUALITY)

(4) VOID FRACTION  
 -----

J2 = 1 (J2=0 NO SUBCOOLED VOID, -1 LEVY MODEL)  
 J3 = 2 (J3=0 SLIP RATIO = 1, -1 ARMAND, -2 SMITH, -5 SLIP POLYNOMIAL, -8 VOID = F(QUAL))

(5) FLOW DIVISION AT INLET  
 -----

IG = 1 (IG=0 SAME Q, -1 SAME DP/DX, -2 QIN/QAV RATIO GIVEN)

(6) CONSTANTS  
 -----

CRITICAL HEAT FLUX (NCHF) = 3  
 CROSS-FLOW RESISTANCE (K1J) = 0.500  
 MOMENTUM TURBULENT FACTOR (FTM) = 0.0  
 TRANSVERSE MOMENTUM FACTOR (S/L) = 0.500  
 CHANNEL ANGLE FROM VERTICAL = 0.0 DEGREES

(7) ITERATION  
 -----

MAX. ALLOWABLE NO. ITERATIONS = 20  
 FLOW CONVERGENCE FACTOR = 1.000E-03

(8) COUPLING PARAMETER FOR THE MIXING TERM  
 -----

PHYSICAL PROPERTIES

P	T	VF	VG	HF	HG	VISC	KF	SIGMA
706.7	504.23	0.02052	0.64875	492.85	1201.01	0.24742	0.34494	0.00158
723.7	506.90	0.02058	0.63261	496.05	1200.59	0.24588	0.34381	0.00156
740.8	509.53	0.02064	0.61718	499.20	1200.15	0.24438	0.34269	0.00154
757.8	512.11	0.02071	0.60242	502.31	1199.69	0.24292	0.34159	0.00151
774.9	514.64	0.02077	0.58827	505.37	1199.21	0.24149	0.34049	0.00149
791.9	517.13	0.02083	0.57471	508.40	1198.71	0.24010	0.33940	0.00147
809.0	519.58	0.02090	0.56170	511.39	1198.20	0.23874	0.33832	0.00145
826.0	522.00	0.02096	0.54921	514.34	1197.67	0.23742	0.33725	0.00142
843.1	524.37	0.02102	0.53720	517.26	1197.13	0.23613	0.33619	0.00140
860.1	526.71	0.02109	0.52565	520.14	1196.57	0.23486	0.33513	0.00138
877.2	529.01	0.02115	0.51453	522.99	1196.00	0.23363	0.33408	0.00136
894.2	531.28	0.02121	0.50382	525.80	1195.42	0.23242	0.33304	0.00134
911.3	533.52	0.02128	0.49349	528.59	1194.82	0.23123	0.33201	0.00132
928.3	535.73	0.02134	0.48353	531.35	1194.21	0.23008	0.33099	0.00130
945.4	537.90	0.02140	0.47391	534.07	1193.59	0.22894	0.32997	0.00128
962.4	540.05	0.02147	0.46462	536.77	1192.96	0.22783	0.32896	0.00126
979.5	542.16	0.02153	0.45564	539.45	1192.32	0.22674	0.32795	0.00124
996.5	544.25	0.02159	0.44696	542.09	1191.66	0.22567	0.32696	0.00123
1013.6	546.31	0.02166	0.43855	544.72	1191.00	0.22463	0.32597	0.00121
1030.6	548.35	0.02172	0.43042	547.31	1190.32	0.22360	0.32498	0.00119
1047.7	550.36	0.02178	0.42253	549.89	1189.64	0.22259	0.32400	0.00117
1064.7	552.34	0.02185	0.41489	552.44	1188.94	0.22160	0.32303	0.00115
1081.8	554.30	0.02191	0.40748	554.97	1188.23	0.22063	0.32206	0.00114
1098.8	556.24	0.02197	0.40029	557.47	1187.52	0.21967	0.32110	0.00112
1115.9	558.16	0.02203	0.39331	559.96	1186.79	0.21873	0.32014	0.00110
1132.9	560.05	0.02210	0.38654	562.43	1186.05	0.21780	0.31919	0.00109
1150.0	561.92	0.02216	0.37995	564.88	1185.30	0.21690	0.31824	0.00107
1167.0	563.77	0.02222	0.37354	567.30	1184.55	0.21600	0.31730	0.00106
1184.1	565.60	0.02229	0.36732	569.72	1183.78	0.21512	0.31637	0.00104
1201.1	567.41	0.02235	0.36126	572.11	1183.00	0.21425	0.31543	0.00103

FUEL ROD TEMP. CONVERGENCE CRITERIA = .10000E-01

MASS BALANCE --  
MASS FLOW IN 0.67137E+02 LB/SEC  
MASS FLOW OUT 0.67137E+02 LB/SEC  
MASS FLOW ERROR 0.0

ENERGY BALANCE --  
FLOW ENERGY IN 0.35489E+05 BTU/SEC  
ENERGY ADDED 0.82320E+04 BTU/SEC  
FLOW ENERGY OUT 0.45523E+05 BTU/SEC  
ENERGY ERROR 0.18027E+04 BTU/SEC

CHANNEL ENTHALPY TEMPERATURE DENSITY EQUIL VOID FLOW MASS FLUX  
(NO.) (BTU/LB) (DEG-F) (LB/FT3) QUALITY FRACTION (LB/SEC) (MLB/HR-FT2)

1	711.35	548.39	13.27	0.255	0.750	31.5056	1.0324
2	648.63	548.39	17.59	0.158	0.651	35.6318	1.1676



CHANNEL RESULTS  
CASE 1

BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:46:34

BUNDLE AVERAGED RESULTS

DISTANCE (IN.)	DELTA-P (PSI)	ENTHALPY (BTU/LB)	TEMPERATURE (DEG-F)	DENSITY (LB/CU-FT)	EQUIL QUALITY FRACTION	VOID FRACTION	FLOW (LB/SEC)	MASS FLUX (MLB/HR-FT <sup>2</sup> )
0.0	23.82	528.60	533.53	47.00	0.0	0.0	67.1373	1.1000
7.5	16.45	532.17	536.39	46.82	0.0	0.0	67.1373	1.1000
15.0	15.97	536.49	539.82	45.82	0.0	0.005	67.1373	1.1000
22.5	15.70	541.65	543.90	43.04	0.0	0.069	67.1373	1.1000
30.0	15.42	547.76	548.39	39.09	0.001	0.159	67.1373	1.1000
37.5	14.84	554.92	548.39	35.20	0.012	0.248	67.1373	1.1000
45.0	14.55	563.23	548.39	31.76	0.025	0.327	67.1373	1.1000
52.5	14.24	572.46	548.39	28.90	0.039	0.392	67.1373	1.1000
60.0	13.55	582.30	548.39	26.55	0.054	0.446	67.1373	1.1000
67.5	13.23	592.53	548.39	24.56	0.070	0.491	67.1373	1.1000
75.0	12.44	602.98	548.39	22.87	0.087	0.530	67.1373	1.1000
82.5	12.10	613.36	548.39	21.44	0.103	0.563	67.1373	1.1000
90.0	11.75	623.40	548.39	20.24	0.118	0.590	67.1373	1.1000
97.5	10.84	632.86	548.39	19.22	0.133	0.614	67.1373	1.1000
105.0	10.48	641.64	548.39	18.36	0.147	0.633	67.1373	1.1000
112.5	10.11	649.77	548.39	17.62	0.159	0.650	67.1373	1.1000
120.0	9.09	657.26	549.39	16.99	0.171	0.665	67.1373	1.1000
127.5	8.71	664.02	548.39	16.45	0.181	0.677	67.1373	1.1000
135.0	8.33	669.93	548.39	16.01	0.191	0.687	67.1373	1.1000
142.5	7.25	674.73	548.39	15.66	0.198	0.695	67.1373	1.1000
150.0	0.0	678.06	548.39	15.43	0.203	0.700	67.1373	1.1000

CHANNEL RESULTS  
CASE 1

BWR TURBINE TRIP W/D BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:46:34

TIME = 0.0 SECONDS DATA FOR CHANNEL 1

DISTANCE (IN.)	DELTA-P (PSI)	ENTHALPY (BTU/LB)	TEMPERATURE (DEG-F)	DENSITY (LB/CU-FT)	EQUIL QUALITY FRACTION	VOID FRACTION	FLOW (LB/SEC)	MASS FLUX (MLB/HR-FT <sup>2</sup> )
0.0	23.82	528.60	533.53	47.00	0.0	0.0	31.5056	1.0324
7.5	17.32	532.97	537.02	46.78	0.0	0.0	31.5056	1.0324
15.0	16.66	538.25	541.21	44.97	0.0	0.024	31.5056	1.0324
22.5	16.59	544.56	546.19	40.61	0.0	0.124	31.5056	1.0324
30.0	16.31	552.03	548.39	36.15	0.007	0.226	31.5056	1.0324
37.5	15.75	560.79	548.39	32.26	0.021	0.315	31.5056	1.0324
45.0	15.46	570.95	548.39	28.91	0.037	0.392	31.5056	1.0324
52.5	15.16	582.23	548.39	26.17	0.054	0.455	31.5056	1.0324
60.0	14.47	594.26	548.39	23.92	0.073	0.506	31.5056	1.0324
67.5	14.14	606.77	548.39	22.02	0.092	0.549	31.5056	1.0324
75.0	13.35	619.55	548.39	20.40	0.112	0.586	31.5056	1.0324
82.5	13.00	632.23	548.39	19.04	0.132	0.618	31.5056	1.0324
90.0	12.63	644.51	548.39	17.88	0.151	0.644	31.5056	1.0324
97.5	11.70	656.08	548.39	16.90	0.169	0.667	31.5056	1.0324
105.0	11.32	666.81	548.39	16.08	0.186	0.685	31.5056	1.0324
112.5	10.93	676.75	548.39	15.37	0.201	0.702	31.5056	1.0324
120.0	9.88	685.02	548.39	14.77	0.215	0.715	31.5056	1.0324
127.5	9.49	694.17	548.39	14.25	0.228	0.727	31.5056	1.0324
135.0	9.09	701.40	548.39	13.83	0.240	0.737	31.5056	1.0324
142.5	7.96	707.27	548.39	13.50	0.249	0.744	31.5056	1.0324
150.0	0.0	711.35	548.39	13.27	0.255	0.750	31.5056	1.0324

CHANNEL RESULTS  
CASE 1

BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:48:34

TIME = 0.0 SECONDS DATA FOR CHANNEL 2

DISTANCE (IN.)	DELTA-P (PSI)	ENTHALPY (BTU/LB)	TEMPERATURE (DEG-F)	DENSITY (LB/CU-FT)	EQUIL QUALITY FRACTION	VOID FRACTION	FLOW (LB/SEC)	MASS FLUX (MLB/HR-FT <sup>2</sup> )
0.0	23.82	528.60	533.53	47.00	0.0	0.0	35.6318	1.1676
7.5	15.57	531.47	535.83	46.85	0.0	0.0	35.6318	1.1676
15.0	15.07	534.94	538.59	46.68	0.0	0.0	35.6318	1.1676
22.5	14.81	539.08	541.87	45.47	0.0	0.013	35.6318	1.1676
30.0	14.52	543.99	545.74	42.03	0.0	0.092	35.6318	1.1676
37.5	13.93	549.74	548.39	38.14	0.004	0.181	35.6318	1.1676
45.0	13.63	556.41	548.39	34.61	0.014	0.261	35.6318	1.1676
52.5	13.33	563.83	548.39	31.64	0.026	0.329	35.6318	1.1676
60.0	12.63	571.72	548.39	29.18	0.038	0.388	35.6318	1.1676
67.5	12.31	579.94	548.39	27.11	0.051	0.433	35.6318	1.1676
75.0	11.53	588.34	548.39	25.34	0.064	0.473	35.6318	1.1676
82.5	11.20	596.67	548.39	23.85	0.077	0.508	35.6318	1.1676
90.0	10.87	604.73	548.39	22.59	0.089	0.536	35.6318	1.1676
97.5	9.98	612.33	548.39	21.54	0.101	0.561	35.6318	1.1676
105.0	9.63	619.38	548.39	20.64	0.112	0.581	35.6318	1.1676
112.5	9.28	625.91	548.39	19.87	0.122	0.599	35.6318	1.1676
120.0	8.30	631.93	548.39	19.21	0.132	0.614	35.6318	1.1676
127.5	7.94	637.55	548.39	18.65	0.140	0.626	35.6318	1.1676
135.0	7.57	642.10	548.39	18.19	0.147	0.637	35.6318	1.1676
142.5	6.64	645.95	548.39	17.83	0.153	0.645	35.6318	1.1676
150.0	0.0	648.63	548.39	17.59	0.158	0.651	35.6318	1.1676

TIME = 0.0 SECONDS TEMPERATURE DATA FOR ROD 1, FUEL TYPE 1

DISTANCE (IN.)	FLUX (MBTU/HR-FT <sup>2</sup> )	CHFR	CHANNEL	TEMPERATURE (F)							
				T (1)	T (2)	T (3)	T (4)	T (5)	T (6)	T (7)	
0.0	0.0	0.0	0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7.5	0.1012	0.0	0	1233.0	1205.6	1126.0	1000.5	838.2	602.8	572.1	572.1
15.0	0.1221	8.108	1	1413.8	1377.4	1272.4	1110.0	904.4	620.3	583.5	583.5
22.5	0.1461	6.780	1	1593.8	1545.8	1409.2	1202.2	947.1	607.3	563.1	563.1
30.0	0.1729	5.726	1	1853.7	1708.8	1606.3	1337.5	1017.8	615.5	563.2	563.2
37.5	0.2028	4.883	1	2185.3	2096.1	1849.1	1497.2	1096.8	625.1	563.9	563.9
45.0	0.2352	4.210	1	2603.6	2402.6	2147.8	1683.9	1183.0	635.8	565.0	565.0
52.5	0.2612	3.791	1	2973.2	2828.7	2416.4	1844.4	1251.7	644.1	565.5	565.5
60.0	0.2784	3.433	1	3221.0	3066.3	2607.7	1956.2	1296.9	649.3	565.7	565.7
67.5	0.2896	3.080	1	3379.7	3221.6	2738.3	2032.3	1326.4	652.8	565.8	565.8
75.0	0.2959	2.795	1	3467.1	3308.2	2813.5	2076.3	1343.1	654.7	565.9	565.9
82.5	0.2936	2.597	1	3434.7	3276.0	2785.3	2059.7	1336.7	653.7	565.5	565.5
90.0	0.2842	2.464	1	3302.6	3145.8	2673.7	1994.4	1311.5	650.4	565.1	565.1
97.5	0.2678	2.395	1	3065.9	2916.9	2486.2	1884.9	1267.8	644.9	564.4	564.4
105.0	0.2485	2.361	1	2787.1	2653.3	2279.2	1762.6	1216.6	638.5	563.6	563.6
112.5	0.2301	2.331	1	2530.8	2415.1	2095.5	1651.4	1167.7	632.4	563.1	563.1
120.0	0.2121	2.309	1	2296.2	2198.4	1928.4	1547.3	1119.9	626.5	562.4	562.4
127.5	0.1911	2.342	1	2046.9	1968.0	1740.4	1431.4	1064.1	619.4	561.6	561.6
135.0	0.1672	2.458	1	1791.7	1730.8	1559.4	1305.1	1000.2	611.2	560.5	560.5
142.5	0.1359	2.805	1	1497.8	1455.5	1334.1	1148.2	916.4	600.2	559.0	559.0
150.0	0.0945	3.815	1	1163.8	1139.2	1067.3	953.5	805.2	585.3	556.5	556.5

TIME = 0.0 SECONDS TEMPERATURE DATA FOR ROD 2, FUEL TYPE 1

DISTANCE (IN.)	FLUX (MBTU/HR-FT <sup>2</sup> )	CHFR	CHANNEL	TEMPERATURE(F)								
				T (1)	T (2)	T (3)	T (4)	T (5)	T (6)	T (7)		
0.0	0.0	0.0	0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7.5	0.0752	0.0	0	1029.5	1011.3	957.7	871.8	757.8	582.9	560.0	560.0	560.0
15.0	0.0907	0.0	0	1149.5	1126.0	1057.5	948.6	806.3	595.3	567.7	567.7	567.7
22.5	0.1085	9.126	2	1295.1	1264.8	1176.8	1038.9	861.9	609.5	576.7	576.7	576.7
30.0	0.1285	7.708	2	1438.5	1399.7	1288.1	1116.2	899.9	601.0	562.0	562.0	562.0
37.5	0.1506	6.574	2	1634.4	1584.0	1440.3	1273.7	958.3	607.9	562.2	562.2	562.2
45.0	0.1747	5.668	2	1871.2	1805.0	1619.2	1346.1	1022.0	615.6	562.7	562.7	562.7
52.5	0.1940	5.103	2	2082.2	2000.7	1774.4	1448.0	1073.2	621.8	563.1	563.1	563.1
60.0	0.2068	4.789	2	2232.5	2139.6	1882.9	1518.7	1106.8	625.7	563.3	563.3	563.3
67.5	0.2151	4.603	2	2335.9	2235.1	1956.9	1565.6	1128.8	628.3	563.4	563.4	563.4
75.0	0.2198	4.470	2	2396.1	2290.7	1999.8	1592.4	1141.2	629.0	563.4	563.4	563.4
82.5	0.2181	4.311	2	2373.5	2269.7	1983.6	1582.2	1136.4	629.0	563.2	563.2	563.2
90.0	0.2111	4.259	2	2284.6	2107.7	1920.1	1542.2	1117.6	626.5	562.8	562.8	562.8
97.5	0.1989	4.326	2	2136.8	2051.2	1813.8	1474.0	1085.0	622.3	562.1	562.1	562.1
105.0	0.1846	4.468	2	1974.5	1900.8	1695.3	1396.5	1046.9	617.4	561.5	561.5	561.5
112.5	0.1709	4.631	2	1829.8	1766.4	1588.0	1324.7	1010.5	612.8	561.0	561.0	561.0
120.0	0.1576	4.830	2	1697.1	1642.5	1487.9	1256.2	974.8	608.3	560.5	560.5	560.5
127.5	0.1470	5.165	2	1552.6	1507.0	1376.8	1179.7	933.2	602.9	559.8	559.8	559.8
135.0	0.1242	5.710	2	1398.9	1362.2	1256.4	1092.7	885.6	598.6	558.9	558.9	558.9
142.5	0.1009	6.833	2	1213.2	1186.2	1107.6	983.7	823.1	588.2	557.5	557.5	557.5
150.0	0.0702	9.632	2	990.0	973.4	924.3	845.3	740.1	576.9	555.4	555.4	555.4

TIME = 0.0 SECONDS

H-L CRITICAL HEAT FLUX SUMMARY

DISTANCE	FLUX	NCHFR	ROD	CHANNEL
0.0	0.0	0.0	0	0
7.5	0.0	0.0	0	0
15.0	0.122	8.108	1	1
22.5	0.146	6.780	1	1
30.0	0.173	5.726	1	1
37.5	0.203	4.883	1	1
45.0	0.235	4.210	1	1
52.5	0.261	3.791	1	1
60.0	0.278	3.433	1	1
67.5	0.290	3.080	1	1
75.0	0.296	2.795	1	1
82.5	0.294	2.597	1	1
90.0	0.284	2.464	1	1
97.5	0.268	2.395	1	1
105.0	0.249	2.361	1	1
112.5	0.230	2.331	1	1
120.0	0.212	2.309	1	1
127.5	0.191	2.342	1	1
135.0	0.167	2.458	1	1
142.5	0.136	2.805	1	1
150.0	0.094	3.815	1	1

ITERATIONS = 3

CHANNEL RESULTS  
CASE 1

BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:47:30

TIME = 2.50000 SECONDS DATA FOR CHANNEL 1

DISTANCE (IN.)	DELTA-P (PSI)	ENTHALPY (BTU/LB)	TEMPERATURE (DEG-F)	DENSITY (LB/CU-FT)	EQUIL QUALITY FRACTION	VOID FRACTION	FLOW (LB/SEC)	MASS FLUX (MLB/HR-FT <sup>2</sup> )
0.0	13.85	528.60	533.53	47.00	0.0	0.0	20.4713	0.6708
7.5	10.98	536.00	539.43	46.62	0.0	0.0	20.4706	0.6708
15.0	10.68	544.82	546.40	46.17	0.0	0.0	20.4708	0.6701
22.5	10.46	554.83	554.20	44.26	0.0	0.012	20.4491	0.6691
30.0	10.24	567.05	563.58	37.86	0.0	0.164	20.4198	0.6691
37.5	9.91	581.34	567.41	32.26	0.015	0.298	20.4174	0.6695
45.0	9.70	597.89	567.41	27.85	0.042	0.402	20.4318	0.6703
52.5	9.48	616.20	567.41	24.44	0.072	0.484	20.4554	0.6724
60.0	9.09	635.65	567.41	21.75	0.104	0.548	20.4849	0.6713
67.5	8.86	655.82	567.41	19.56	0.137	0.600	20.5186	0.6736
75.0	8.40	676.36	567.41	17.73	0.171	0.644	20.5554	0.6749
82.5	8.15	696.66	567.41	16.21	0.204	0.680	20.5949	0.6762
90.0	7.90	716.25	567.41	14.94	0.236	0.710	20.6368	0.6777
97.5	7.34	734.63	567.41	13.88	0.266	0.735	20.6809	0.6792
105.0	7.07	751.62	567.41	13.00	0.294	0.756	20.7272	0.6808
112.5	6.81	767.26	567.41	12.26	0.319	0.774	20.7758	0.6825
120.0	6.15	781.57	567.41	11.63	0.343	0.799	20.8209	0.6842
127.5	5.88	794.34	567.41	11.10	0.364	0.801	20.8609	0.6861
135.0	5.61	805.37	567.41	10.67	0.382	0.812	20.9383	0.6881
142.5	4.90	814.10	567.41	10.35	0.398	0.819	20.9995	0.6903
150.0	0.0	819.82	567.41	10.14	0.405	0.824	21.0654	0.6903

CHANNEL RESULTS  
CASE 1

BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:47:30

TIME = 2.50000 SECONDS DATA FOR CHANNEL 2

DISTANCE (IN.)	DELTA-P (PSI)	ENTHALPY (BTU/LB)	TEMPERATURE (DEG-F)	DENSITY (LB/CU-FT)	EQUIL QUALITY FRACTION	VOID FRACTION	FLOW (LB/SEC)	MASS FLUX (MLB/HR-FT <sup>2</sup> )
0.0	13.85	528.60	533.53	47.00	0.0	0.0	24.2069	0.7958
7.5	9.90	533.23	537.23	46.76	0.0	0.0	24.2069	0.7958
15.0	9.56	538.76	541.62	46.48	0.0	0.0	24.2876	0.7959
22.5	9.34	545.28	546.76	46.15	0.0	0.0	24.2900	0.7959
30.0	9.11	552.68	552.53	45.70	0.0	0.0	24.3937	0.7961
37.5	8.76	561.54	559.37	41.81	0.0	0.070	24.3951	0.7961
45.0	8.52	571.77	567.16	36.54	0.0	0.196	24.3264	0.7971
52.5	8.29	583.13	567.41	32.40	0.018	0.294	24.3712	0.7986
60.0	7.88	595.21	567.41	29.12	0.038	0.372	24.4280	0.8005
67.5	7.65	607.73	567.41	26.47	0.058	0.435	24.4915	0.8026
75.0	7.19	620.48	567.41	24.28	0.079	0.488	24.5586	0.8048
82.5	6.95	633.09	567.41	22.49	0.100	0.530	24.6283	0.8070
90.0	6.71	645.24	567.41	21.01	0.120	0.566	24.6997	0.8094
97.5	6.17	656.63	567.41	19.78	0.138	0.595	24.7723	0.8118
105.0	5.92	667.14	567.41	18.77	0.156	0.619	24.8456	0.8142
112.5	5.67	676.80	567.41	17.91	0.171	0.639	24.9195	0.8166
120.0	5.06	685.04	567.41	17.10	0.196	0.657	24.9937	0.8190
127.5	4.80	693.52	567.41	16.57	0.199	0.671	25.0683	0.8215
135.0	4.54	700.31	567.41	16.08	0.210	0.683	25.1437	0.8239
142.5	3.88	705.69	567.41	15.70	0.219	0.692	25.2205	0.8264
150.0	0.0	709.21	567.41	15.46	0.224	0.698	25.2990	0.8290



TIME = 2.50000 SECONDS TEMPERATURE DATA FOR ROD 1, FUEL TYPE 1

DISTANCE (IN.)	FLUX (MBTU/HR-FT <sup>2</sup> )	CHFR	CHANNEL	TEMPERATURE(F)								
				T(1)	T(2)	T(3)	T(4)	T(5)	T(6)	T(7)		
0.0	0.0	0.0	0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7.5	0.1108	0.0	0	1312.7	1284.0	1198.9	1059.7	875.0	622.5	589.5	589.5	589.5
15.0	0.1327	7.036	1	1510.6	1472.7	1361.1	1181.6	948.1	645.6	606.3	606.3	606.3
22.5	0.1523	6.130	1	1711.5	1662.5	1520.6	1297.9	1015.6	667.7	623.0	623.0	623.0
30.0	0.1869	4.994	1	1993.0	1926.3	1734.5	1437.5	1069.6	639.6	583.7	583.7	583.7
37.5	0.2193	4.276	1	2348.3	2257.9	2000.9	1613.1	1150.5	647.4	582.1	582.1	582.1
45.0	0.2522	3.702	1	2787.1	2667.9	2326.3	1818.9	1239.0	657.7	582.4	582.4	582.4
52.5	0.2791	3.345	1	3164.7	3026.5	2615.8	1996.4	1309.4	666.0	582.8	582.8	582.8
60.0	0.2957	2.847	1	3412.9	3267.7	2819.4	2120.4	1355.5	671.4	583.0	583.0	583.0
67.5	0.3093	2.410	1	3570.4	3423.2	2956.9	2205.0	1385.6	674.8	583.1	583.1	583.1
75.0	0.3148	2.030	1	3656.7	3509.3	3035.4	2254.0	1402.6	676.8	583.1	583.1	583.1
82.5	0.3124	1.717	1	3624.8	3477.5	3006.0	2235.5	1396.1	675.8	582.9	582.9	582.9
90.0	0.3027	1.444	1	3494.1	3347.5	2889.2	2162.9	1370.5	672.6	582.4	582.4	582.4
97.5	0.2858	1.204	1	3258.1	3116.6	2690.5	2041.5	1326.0	667.0	581.8	581.8	581.8
105.0	0.2659	1.142	1	2975.9	2846.1	2468.5	1906.1	1273.7	660.6	581.2	581.2	581.2
112.5	0.2468	1.162	1	2711.8	2597.2	2259.8	1783.3	1223.5	654.5	580.7	580.7	580.7
120.0	0.2280	1.190	1	2466.0	2367.5	2087.7	1660.3	1174.1	648.4	580.1	580.1	580.1
127.5	0.2061	1.250	1	2200.9	2120.4	1890.5	1540.3	1116.2	641.2	579.4	579.4	579.4
135.0	0.1807	1.359	1	1926.1	1863.4	1682.4	1400.6	1049.3	632.7	578.5	578.5	578.5
142.5	0.1472	1.605	1	1606.2	1562.1	1433.0	1226.6	960.5	621.3	577.0	577.0	577.0
150.0	0.1022	2.250	1	1238.5	1212.6	1135.9	1009.7	841.2	609.7	574.8	574.8	574.8

TIME = 2.50000 SECONDS TEMPERATURE DATA FOR ROD 2, FUEL TYPE 1

DISTANCE (IN.)	FLUX (MBTU/HR-FT <sup>2</sup> )	CHFR	CHANNEL	TEMPERATURE (F)								
				T (1)	T (2)	T (3)	T (4)	T (5)	T (6)	T (7)		
0.0	0.0	0.0	0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7.5	0.0828	0.0	0	1088.4	1069.1	1011.3	915.3	785.2	595.9	570.9	570.9	570.9
15.0	0.0993	9.403	2	1220.7	1196.0	1122.4	1001.3	839.1	611.8	582.0	582.0	582.0
22.5	0.1180	7.913	2	1380.7	1348.9	1254.9	1101.9	900.2	629.9	594.7	594.7	594.7
30.0	0.1353	6.901	2	1541.9	1502.1	1385.6	1199.3	957.9	647.6	607.5	607.5	607.5
37.5	0.1630	5.727	2	1755.2	1702.9	1551.0	1311.2	1006.6	631.6	582.8	582.8	582.8
45.0	0.1887	4.949	2	2011.9	1943.9	1748.4	1446.2	1072.8	638.2	581.8	581.8	581.8
52.5	0.2092	4.462	2	2238.5	2155.5	1919.0	1559.4	1125.6	643.5	580.8	580.8	580.8
60.0	0.2225	4.196	2	2398.5	2304.6	2037.9	1616.6	1160.4	647.5	580.9	580.9	580.9
67.5	0.2312	4.037	2	2507.8	2406.5	2118.8	1608.4	1183.2	650.2	581.0	581.0	581.0
75.0	0.2361	3.798	2	2571.1	2465.6	2165.6	1718.0	1196.1	651.7	581.0	581.0	581.0
82.5	0.2343	3.555	2	2547.3	2433.4	2147.9	1706.8	1171.7	650.9	580.8	580.8	580.8
90.0	0.2270	3.399	2	2453.7	2356.1	2078.6	1662.6	1151.1	648.4	580.4	580.4	580.4
97.5	0.2142	3.332	2	2296.9	2209.9	1962.3	1587.4	1137.9	644.1	579.9	579.9	579.9
105.0	0.1992	3.317	2	2123.2	2047.8	1832.2	1501.7	1098.2	639.1	579.3	579.3	579.3
112.5	0.1947	3.311	2	1967.3	1902.0	1714.0	1422.2	1060.0	634.3	578.9	578.9	578.9
120.0	0.1704	3.325	2	1823.5	1767.0	1603.4	1346.3	1022.4	629.8	576.4	576.4	576.4
127.5	0.1538	3.425	2	1656.0	1618.6	1480.4	1203.3	978.4	624.0	577.7	577.7	577.7
135.0	0.1346	3.656	2	1497.7	1459.3	1346.8	1164.7	927.7	617.5	578.9	578.9	578.9
142.5	0.1093	4.252	2	1293.1	1264.8	1180.8	1043.3	860.5	608.7	575.7	575.7	575.7
150.0	0.0755	5.918	2	1045.5	1028.0	975.6	888.6	770.4	596.6	573.7	573.7	573.7

CASE 1 BWR TURBINE TRIP W/O BYPASS NEW FUEL ROD MODEL

DATE 9/24/80 TIME 14:47:30

TIME = 2.50000 SECONDS

H-L CRITICAL HEAT FLUX SUMMARY

DISTANCE	FLUX	INCHFR	ROD	CHANNEL
0.0	0.0	0.0	0	0
7.5	0.0	0.0	0	0
15.0	0.133	7.036	1	1
22.5	0.152	6.130	1	1
30.0	0.187	4.994	1	1
37.5	0.218	4.276	1	1
45.0	0.252	3.702	1	1
52.5	0.279	3.345	1	1
60.0	0.297	2.847	1	1
67.5	0.308	2.410	1	1
75.0	0.315	2.030	1	1
82.5	0.312	1.717	1	1
90.0	0.303	1.444	1	1
97.5	0.286	1.204	1	1
105.0	0.266	1.142	1	1
112.5	0.247	1.162	1	1
120.0	0.228	1.190	1	1
127.5	0.200	1.250	1	1
135.0	0.181	1.359	1	1
142.5	0.147	1.605	1	1
150.0	0.102	2.250	1	1

ITERATIONS = 2

APPENDIX P  
 INPUT FOR BYPASS ANALYSIS

COBRA-IIIC/MIT Nineteen Channel Case One Input

```

19 38 1 10 200 ← first "card"
2000
0 1 1 BWR Bypass Analysis Test Case - No Control Rods Inserted 9-80 1.00 P
1 0
1 1 500. 1500. 20
2 0 0 0 0
.184 -.2
3 21
0..2653 .05.5491 .10.7268 .15.8541 .20.9363 .25.9814
.30.9841 .35.9523 .40.9045 .45.8674 .50.8753 .55.9496
.601.096 .651.302 .701.517 .751.671 .801.674 .851.451
.90.9867 .95.3899 1.0 0.00
4 19 19
12.1795.4385.438 2.3752 5.3752
23.19510.8810.88 1.3752 3.3752 6.3752
32.1795.4385.438 2.3752 7.3752
41.0525.4385.438 5.1870 11.1870
54.79416.3116.31 1.3752 4.1870 6.3752 10.7520
64.20721.7521.75 2.3752 5.3752 7.3752 9.3752
73.19510.8810.88 3.3752 6.3752 8.3752
82.1795.4385.438 7.3752 9.3752
94.79416.3116.31 6.3752 8.3752 10.7520 16.1870
108.40821.7521.75 5.7520 9.7520 11.7520 15.7520
114.79416.3116.31 4.1870 10.7520 12.3752 14.3752
122.1795.4385.438 11.3752 13.3752
133.19510.8810.88 12.3752 14.3752 19.3752
144.20721.7521.75 11.3752 13.3752 15.3752 18.3752
154.79416.3116.31 10.7520 14.3752 16.1870 17.3752
161.0525.4385.438 9.1870 15.1870
172.1795.4385.438 15.3752 18.3752
183.19510.8810.88 14.3752 17.3752 19.3752
192.1795.4385.438 13.3752 18.3752
7
8 10 10
1 6.921.158 1 .33 2 .25 5 .25 6 .17
2 6.921.174 2 .25 3 .33 6 .17 7 .25
3 6.92 1.00 4 .17 5 .25 10 .33 11 .25
4 6.92 1.00 5 .25 6 .17 9 .25 10 .33
5 6.921.158 6 .17 7 .25 8 .33 9 .25
6 6.921.158 11 .25 12 .33 13 .25 14 .17
7 6.92 1.00 10 .33 11 .25 14 .17 15 .25
8 6.92 1.00 9 .25 10 .33 15 .25 16 .17
9 6.921.174 13 .25 14 .17 18 .25 19 .33
10 6.921.158 14 .17 15 .25 17 .33 18 .25
9 3 0
0.5 0.167.8 0. 60 0 0. .094
10 0 1
.02
11 0 1
1035. 527.1 .1750 .0060
12 1 7
1 2 3 4 5 6 10
  
```

/\* end of file

COBRA-IIIC/MIT Four Channel Case Input

```

      4      3      1      2      200      first "card"
2000
0  1      1      Four Channel BWR Bypass Test Case 10-80
  1      -1
  1      500.      2000.      30
  2      0      0      0      0
.184  -.2
  3      21
  0. .2653 .05.5491 -.10.7268 .15.8541 .20.9363 .25.9814
.30.9841 .35.9523 .40.9045 .45.8674 .50.6753 .55.9496
.601.096 .651.302 .701.517 .751.671 .801.674 .851.451
.90.9867 .95.3899 1.0 0.00
  4      4      4
  11.0902.7192.719      2 .375
  23.19510.8810.88      3 .375      4 .375
  31.0902.7192.719
  41.0525.4385.438
  8      2      2      0      0
1  1  6.92  1.  1 .167  2 .25  4 .083
i  2  6.92  1.  2 .25  3 .167  4 .083
  9      3
  0.5  0.167.8  0.  60  0  0.  .094
  10     0      1      0
.02
  11     0      1
  1035.      527.1      .3290      .0132
  12     1

```

end of input deck

THERMIT Four Channel Case Input

```

1      first "card"
BWR Bypass Analysis - Four Channel Case 10-80
4,2,4,10,1,1
0,1,0,0,1,2,0,1,1,0,0, 0,0,0,0,0,-9.81,0.03,5.0
1.3,20,1.e-7,1.e-7
9.81e4,0.,0.,8.794e-2,1.e-3,1.e-3,1.e6,1.,0.,1.,1.,1.,1.e-4,1.e6,1.,0.,0.,0.,0.
3 1 $ ncr
0 1 $ indent
30(19.350e-3) 10(6.370e-3)$ arx
2( 10(6.601e-3) 10(19.35e-3) ) $ ary
11(7.032e-4) 11(2.061e-3) 11(7.032e-4) 11(6.787e-4)$ arz
10(2.997e-4) 10(8.784e-4) 10(2.997e-4) 10(2.892e-4) $ vol
1.549e-2 4.540e-2 1.549e-2 4.540e-2 $ dx
3(4.540e-2) 1.495e-2 $ dy
12(0.4262) $ dz
4.073e-2 2.984e-2 4.073e-2 1.965e-2 $ hdz
3(0.) 0.00953 0.00953 0. 0.00953 0. 0.00953 3(0.) 0. 0.00953 2(0.) $ sij
11(01) $ ifwz
48(7.135e6)$p
48(0.) $ alp
48(551.11)$ tv
11(0.3) 11(0.3) 11(0.3) 11(0.3) $ vvz
1 2 3 4 $ icr
4.073e-2 2.984e-2 4.073e-2 1.965e-2 $ hdh
40(551.11)$ tw
.407 .7905 .9585 .968 .886 .9125 1.199 1.594 1.54625 .6885 $ qz
1.333 1. 1.333 0.666 $ qt
3(1.) $ qr
4(1.) $ rn
0.125 0.5 0.125 0.250 $ fracp
0,0,0,0
40.,0.01,0. 10,10.,40.,1.0,4
-1
0.....
end of deck

```

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