# 12.540 Principles of the Global Positioning System Lecture 14

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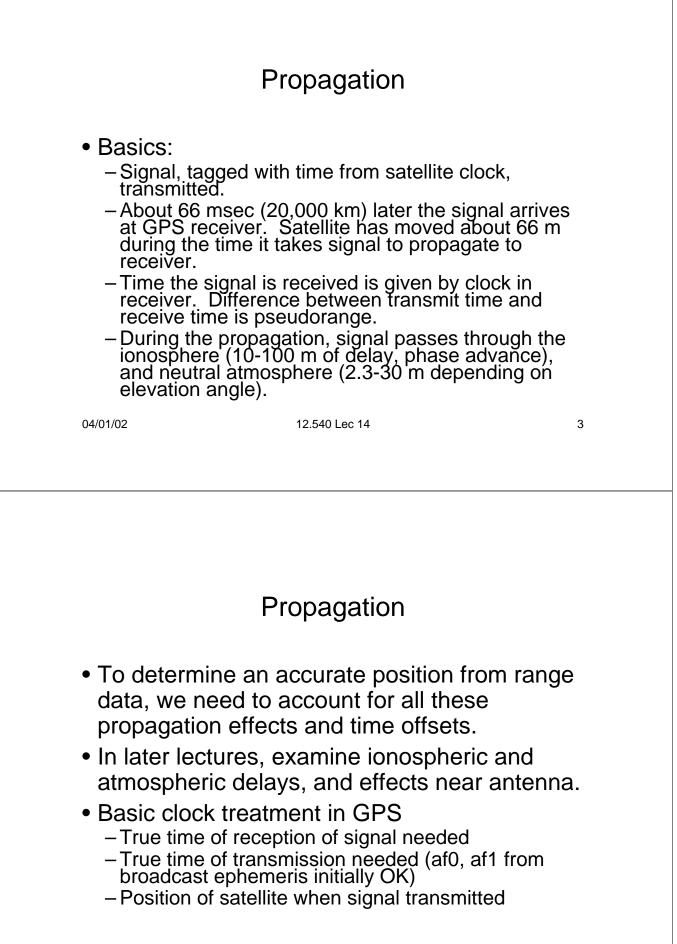
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#### **Propagation Medium**

- Summary
  - Paper topics from those not yet submitted
  - -Review homework #1
  - -<u>Homework #2</u>. Due April 17, 2002
- Propagation:
  - -Signal propagation from satellite to receiver
  - -Light-time iteration
  - -Basic atmospheric and ionospheric delays
  - Propagation near receiving antenna

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#### Times

- RINEX data files, tag measurements by reception time as given by the receiver clock. The error in the receiver time must be determined iteratively
- For linearized least squares or Kalman filter need to establish non-linear model and then estimator determines adjustments to parameters of model (e.g. receiver site coordinates) and initial clock error estimates that "best" match the data.

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Non-linear model

- Basics of non-linear model:
  - Rinex data file time tags give approximate time measurement was made.
  - Using this time initially, position of satellite can be computed
  - Range computed from receiver and satellite position
  - Difference between observed pseudorange and computed ranges, gives effects of satellite and receiver clock errors. In point positioning, satellite clock error is assumed known and when removed from difference, error in receiver clock determined.
  - With new estimate of receiver clock, process can be iterated.
  - If receiver position poorly known, then whole system can be iterated with updated receiver coordinates.

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### Sensitivities

- Satellites move at about 1km/sec, therefore an error of 1 msec in time results in 1 m satellite position (and therefore in range estimate and receiver position).
- For pseudo-range positioning, 1 msec errors OK. For phase positioning (1 mm), times needed to 1 μsec.
- (1 μsec is about 300 m of range.
  Pseudorange accuracy of a few meters in fine).

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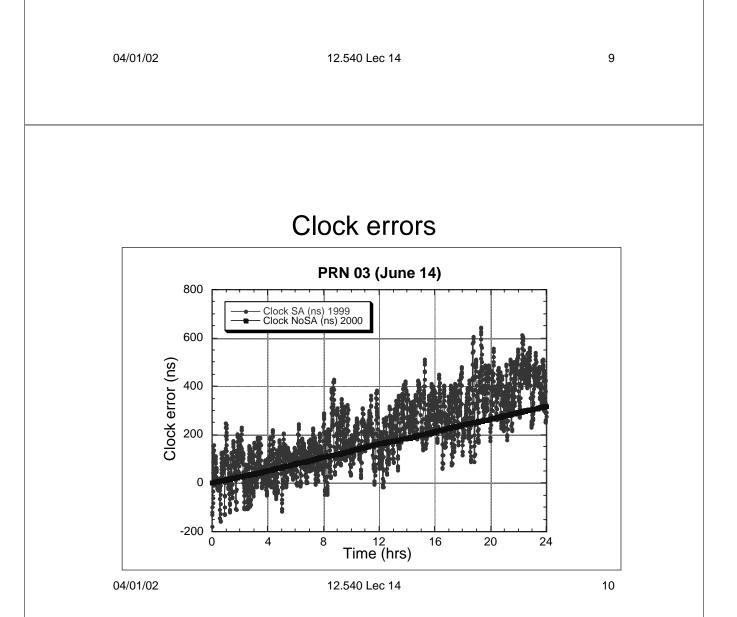
"Light-time-iteration"

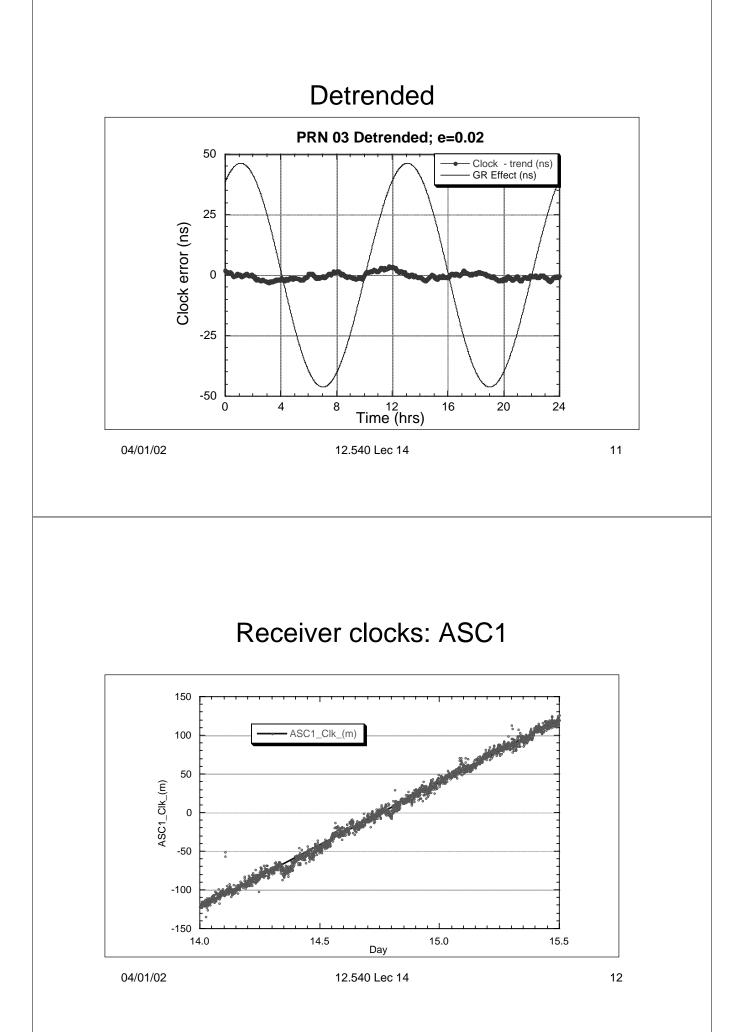
- To compute theoretical range; two basic methods used
  - (a) "Doppler shift corrections" ie. Account for rate of change of range during propagation time
  - (b) "Light-time-iteration" Method most commonly used.
- Light time iteration: Basic process is to compute range using simple Cartesian geometry but with position of receiver at receive time and position of transmitter at transmit time.

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## Light-time-iteration

- Light time iteration must be computed in a nonrotating frame
- Reason: Consider earth-fixed frame: one would simply compute Earth fixed coordinates at earlier time. In non-rotating frame, rotation to inertial coordinates would be done at two different time (receiver when signal received; transmitted when signal transmitted).
- Difference is rotation of Earth on ~60 msec. Rotation rate ~460 m/sec; therefore difference is about 30 meters.





# Receiver Clock: HOB2 Hydrogen Maser

